

Dynamic Behavior of an Ordinary Chondrite: the Effects of Microstructure on Strength, Failure and Fragmentation

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Abstract

Knowledge of the relationships between microstructure, stress-state and failure mechanisms is important in the development and validation of numerical models simulating large-scale impact events. In this study, we investigate the effects of microstructural constituent phases and defects on the compressive and tensile strength, failure, and fragmentation of a stony meteorite (GRO 85209). In the first part of the paper we consider the effect of defects on the strength and failure. Strengths are measured and linked with detailed quantification of the important defects in this material. We use the defect statistic measurements in conjunction with our current understanding of rate-dependent strengths to discuss the uniaxial compressive strength measurements of this ordinary chondrite with those of another ordinary chondrite, with a different defect population. In the second part of the paper, we consider the effects of the microstructure and defects on the fragmentation of GRO 85209. Fragment size distributions are measured using image processing techniques and fragments were found to result from two distinct fragmentation mechanisms. The first is a mechanism that is associated with relatively

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smaller fragments arising from individual defect grains and the coalescence of fractures initiating from microstructure defects. This mechanism becomes more dominant as the strain-rate is increased. The second mechanism is associated with larger fragments that are polyphase and polygrain in character and is dependent on the structural failure mechanisms that are activated during load. In turn, these are dependent on (for example) the strain-rate, stress state, and specimen geometry. The implications of these results are briefly discussed in terms of regolith generation on airless bodies.

Keywords: brittle fragmentation; brittle failure; strength; planetary materials; catastrophic disruption;

1. Introduction

Understanding the dynamic behavior of planetary materials in well-controlled laboratory experiments is important when interpreting large-scale impacts (e.g., Hörz and Cintala, 1997) and developing sophisticated numerical models for such dynamic events (e.g., Michel et al., 2003). During impacts, the colliding bodies will experience a range of stress states (compression, tension, shear) and a wide range of strain rates. The stress-state, strain-rate and loading histories determine the failure mechanisms that are activated during impact (Ramesh et al., 2015). Ultimately, we are interested in developing simple physics-based models for strength and fragmentation of planetary materials that take into account these stress-state and strain-rate dependent mechanisms. These simple models can be used to understand the effect of target material and impact conditions on regolith formation and catastrophic disruption, aspects that have been well-studied experimentally (Durda and Flynn, 1999; Flynn and Durda, 2004; Cintala and Hörz, 2008)

Planetary materials are typically quasi-brittle and inhomogeneous, and may be comprised of mineral and metal grains and amorphous clasts, each with varying crystal

16 properties (e.g., structure, size, and shape) and mechanical properties (e.g., density,
17 strength, fracture toughness). Failure in such heterogeneous materials is initiated from
18 internal defects, and this may include grain boundaries, pores, inclusions, and, at much
19 larger scales, faults. The types of defects that are activated are dependent on the load-
20 ing history (e.g., stress-state and strain-rate) of the event of interest. Commonly, the
21 failure strength of planetary materials exhibits a dependence on strain rate (Kimberley
22 and Ramesh, 2011). Below a critical strain rate, the strength remains nearly constant,
23 but when loaded above this transition strain rate, the strength increases more rapidly
24 for increasing strain rate. The size of the body, the defect density ($\#/m^3$), the degree
25 to which defects are distributed throughout the body, and defect size and orientation
26 distributions have been shown to be important in governing the failure strength of brit-
27 tle materials (Housen and Holsapple, 1999; Paliwal et al., 2008; Holsapple, 2009; Hu
28 et al., 2014). For example, materials with larger defects have lower strengths than those
29 with smaller defects (for those with the same defect densities). Similarly, materials with
30 more defects are weaker than those with fewer defects (Kimberley et al., 2013). Re-
31 cently, Kimberley et al. (2013) developed a scaling relationship for the rate-dependent
32 compressive strength of brittle solids based on defect population, average flaw size and
33 flaw density, given some material properties (such as fracture toughness and Young's
34 modulus). We use insights from this scaling law to explore our experimental results on
35 dynamic strength measurements of ordinary chondrites.

36 During failure, fractures will grow and coalesce and this results in a distribution of
37 fragment sizes across many length scales, ranging from the structural scale (e.g., order
38 of the body-size) down to the micro-scale (e.g., spacing between defects). Measure-
39 ments of the fragment size and shapes can offer insight into important physical failure
40 processes. For example, larger fragments (boulders) on Eros have been used to con-

41 strain its collisional history (Dombard et al., 2010). The extent to which "microstruc-
42 ture" (including material composition and defects) has an effect on the fragmentation of
43 planetary materials is not yet well understood. This is also explored here.

44 In this paper we investigate the role of microstructure, strain-rate and stress-state on
45 the strength, failure and fragmentation of an L6 chondrite meteorite (GRO 85209). This
46 study is partially motivated by the work of Ryan (2000) on asteroid fragmentation, and
47 more recently, the review papers by Zhang and Zhao (2013) on the dynamic behaviour
48 of rocks and Ramesh et al. (2015) on the failure of brittle materials. In this manuscript,
49 we study the material's behavior in uniaxial compression and in indirect tension (using
50 the Brazilian disk technique). We begin by introducing the GRO 85209 microstructure,
51 and describe our methods for quantifying initial defect populations and fragments result-
52 ing from our experiments. We then explore two critical areas related to dynamic failure:
53 strength and fragmentation. First, we investigate the relationship between defects and
54 the rate dependent strength of GRO 85209 using defect statistic measurements and the
55 recently developed scaling relation of Kimberley et al. (2013). In particular, we explain
56 differences between the strength measurements of ordinary chondrites GRO 85209 and
57 MAC 88118, the latter of which was studied by Kimberley and Ramesh (2011). Second,
58 we explore the role of defects and microstructure on the fragmentation of GRO 85209,
59 identifying the relative contributions of each GRO 85209 constituent phase to fragmen-
60 tation. We then seek to use the strength, failure, and fragmentation results to provide
61 insights into regolith formation on airless bodies and catastrophic disruption.

62 **2. Methods and Materials**

63 The strength, failure and fragmentation of an ordinary chondrite meteorite are stud-
64 ied in uniaxial compression for quasi-static and dynamic conditions, and in dynamic

65 indirect tension using the dynamic Brazilian disk technique. Descriptions of the mate-
66 rial characteristics and testing methods used in this study follow.

67 *2.1. Material Characteristics*

68 The ordinary chondrite meteorite studied here is Grosvenor Mountains (GRO) 85209,
69 an Antarctic find that is currently held at the Smithsonian Institute. It is an L6 chondrite
70 consisting primarily of low-Ca pyroxene and iron nickel, with some olivine (Grossman,
71 1994). GRO 85209 also contains chondrules (> 3mm in size), but these were not com-
72 monly found at the scale of our tested samples due to their relative size. A polarized thin
73 section image of the GRO 85209 microstructure is shown in Figure 1a. Linda Welzen-
74 bach of the Smithsonian Institute is credited with analysis of the thin section. Analysis
75 of the thin sections indicates no major fractures, shock veins or brecciation. Olivine
76 (<300 μm and circular), pyroxene (<150 μm and darker in shade), and iron-nickel (<
77 500 μm , blocky and dark in color in polarized light) grains are highlighted in Figure 1a.
78 Large olivine grains show only irregular fractures and no undulatory extinction, suggest-
79 ing a shock stage of S1. An optical microscope image of the GRO 85209 microstructure
80 taken in a reflected light mode is shown in Figure 1b. Here, the iron-nickel grains appear
81 white due to their high reflectivity, and are imbedded in a matrix comprised primarily
82 of the low-Ca pyroxene. No internal fractures are visible, nor are any large olivine
83 grains. The Young's modulus of GRO 85209 is 14 GPa (measured during quasi-static
84 tests using digital image correlation, and compared with strain gage measurements), and
85 it has a density of approximately 3,350 kg/m^3 (measured via Archimedes method) and
86 a porosity of 7%.

87 *2.2. Testing Methods*

88 Cuboidal specimens approximately 3.5 mm x 4 mm and 5.3 mm (loading direction)
89 in dimension were used for the uniaxial compression experiments. The dimensions of
90 the Brazilian disk specimens were 10 mm in diameter and 1.5 mm thick. Quasi-static
91 uniaxial compression experiments were performed with an MTS servo-hydraulic test
92 machine under displacement control at strain rates ranging from 10^{-3} to 10^0 s⁻¹. The
93 dynamic uniaxial compression and Brazilian disk tests were performed using a Kolsky
94 bar apparatus to achieve a range of strain rates from 10^1 to 10^3 s⁻¹. Both the MTS
95 machine and Kolsky bar devices used in this study were also used by Kimberley and
96 Ramesh (2011) in their study on the compressive strength of ordinary chondrite MAC
97 88118, and details of the experimental setup are discussed therein. MAC 88118 is a
98 stony meteorite found in MacAlpine Hills, Antarctica and also currently held at the
99 Smithsonian Institute. The Brazilian disk test is an indirect technique to measure the
100 tensile strength of brittle materials (Li and Wong, 2013). A schematic of the Brazilian
101 Disk setup is shown in Figure 2. The tensile stress, σ_y , in the specimen is calculated as:

$$\sigma_y = \frac{2P}{\pi Dt} \quad (1)$$

102 where P is the load (N), D is the diameter of the disk (m), and t is the thickness (m).
103 While we take the peak stress measurement from the Brazilian disk test as its "tensile"
104 strength, we note that the stress-state in the disk is actually quite complex (Ruiz et al.,
105 2000; Swab et al., 2011), and thus the specimen undergoes non-uniform deformation.
106 This will be taken into consideration in the interpretation of our fragmentation results.
107 Please refer to the recent review by Li and Wong (2013) for additional details on the
108 Brazilian disk technique.

109 Two different high-speed cameras were used to visualize the deformation and failure
110 processes during dynamic experiments. A Kirana (Specialized Imaging) Ultra High-
111 Speed Video Camera filming at 2 Mfps with a 110 ns exposure time captured time-
112 resolved images of the Brazilian disk experiments, while an Ultra 8 (Hadland Imaging)
113 camera was used to capture images of the uniaxial compression experiments at frame
114 rates up to 1 Mfps with exposure times of 200 ns. The use of cuboidal specimens and
115 the flat face of the Brazilian disk allows us to visualize failure.

116 After each test, fragments were collected and imaged using a Zeiss optical micro-
117 scope with an AxioCam MRC camera. An example image showing the fragments is
118 shown in Figure 1c. These images of fragments are converted to monochrome (Fig-
119 ure 1d) using a thresholding procedure, where fragments now appear as white features.
120 An image-processing routine was used to determine the major axis dimension¹ (ℓ), pro-
121 jected area (A) and perimeter (P) of individual fragments. This is the same procedure
122 used in Hogan et al. (2014). The images in Figure 1c were taken using the differ-
123 ential interference contrast setting (a bright-field mode) to have the fragments appear
124 dark. Additional dark-field images with a suitable exposure were also taken to have
125 the fragments appear in greyscale. In the greyscale images, transparent minerals such
126 as pyroxene will show up as bright, and iron-nickel grains will show up dark. After
127 a thresholding operation for the greyscale images, all fragments appear as white, and
128 each white feature consists of tens of pixels. We were then able to relate the coordi-
129 nates of the pixels in the monochrome to the greyscale image, and this allows us to
130 compute the average greyscale intensity across all pixels for a given fragment in the
131 original set of greyscale images. In turn, we relate the greyscale values to fragment

¹The major axis dimension is taken as the largest spanning dimension of the fragment.

132 composition, where, again, the iron-nickel have low greyscale intensities (they are dark)
133 and the pyroxene have high greyscale intensities (they appear bright). This operation
134 allows for correlations between fragments size, shape and composition to be investi-
135 gated. Image processing and analysis techniques was also used to determine the major
136 axis size, number density ($\#/m^2$), and spacing between adjacent iron-nickel grains in
137 the initial GRO 85209 microstructure (Figure 1b). The sizes and number densities were
138 used when exploring or strength results, while the spacing distributions are compared
139 with fragmentation size measurements. A Tescan Mira3 GM Scanning Electron Mi-
140 croscope (SEM) is also used to investigate fracture surfaces, while Energy Dispersive
141 Spectroscopy (EDS) capabilities are used to identify composition of constituent phases.

142 **3. Dynamic Failure Experiments**

143 In this section we examine the stress-time history and associated time-resolved high-
144 speed photography of the dynamic failure experiments. We then examine failure sur-
145 faces of fragments using optical microscopy and scanning electron microscopy tech-
146 niques to determine likely sites for crack initiation, and the dominant modes of crack
147 propagation.

148 *3.1. Characterization of Strength and Failure Processes*

149 Initially we consider the time-resolved failure of a Brazilian disk experiment in Fig-
150 ure 3. The stress-time curve is shown as the solid black curve, which increases nearly
151 linearly with time to a peak stress of 36 MPa at 30 μs after loading begins. This is
152 its dynamic tensile strength at that corresponding strain rate. After the peak stress is
153 reached, the stress decreases to zero over the next 100 μs . The stress rate $\dot{\sigma}$ is taken
154 as the slope of a linear fit to the rising portion of the stress–time curve between 10 and

155 90 % of the peak stress, illustrated by the dashed-line (red underneath). For this ex-
156 periment the stress rate is calculated to be 1.6 MPa/ μ s. The nominal strain rate may be
157 estimated by dividing the stress rate by the Young's modulus, $\dot{\epsilon} = 114 \text{ s}^{-1}$ for this exper-
158 iment. Due to the non-uniform deformation of the Brazilian disk samples, we cannot
159 directly measure the strain rate using the classic Kolsky bar equations, and so we re-
160 port the stress rate and estimate the strain rate using the initial modulus. Corresponding
161 time-resolved high-speed video images selected at 12 μ s intervals (t_1 to t_6) are shown on
162 the right of Figure 3 so as to allow the fracture to be visualized throughout the duration
163 of the failure process. The loading direction (horizontal) is also defined for the images
164 on the right. Red arrows are used in the images to denote fracture initiation sites that
165 are determined by tracking the observed fractures backward in time. Often, the regions
166 of fracture initiation appear as localized bright spots which are likely due to reflection
167 of light from a highly reflective grain in the material that is rotating out-of-plane. In
168 GRO 852909, bright regions are believed to be the metallic iron-nickel grains in the mi-
169 crostructure, which we also believe serve as the most common fracture initiation sites.
170 Additional evidence that fracture initiates from the iron-nickel grains is shown later in
171 optical microscope images in Figure 5a, as well as in fragmentation results presented
172 later in Figure 9.

173 At peak stress (t_1) in Figure 3, no fractures are visible on the specimen surface.
174 At 12 μ s post-peak stress (t_2), there is a fracture that initiates near the middle of the
175 sample (right arrow) and is visible spanning the middle of the disk between both arrows.
176 The stress in the sample collapses as a result of fracturing. At 24 μ s after peak stress
177 (t_3), additional fractures are initiated from iron-nickel grains to the right of the original
178 fracture and, as a result, the stress continues to collapse. At t_4 , fractures are observed
179 to fully span the disk surface. The spanwise propagation of the crack across the entire

180 length of the sample may possibly relate to the hump we observe in the stress-time curve
181 on the left just prior to t_4 . It is challenging to observe additional larger fractures on the
182 surface at later times (t_5 and t_6), although they likely occur. The stress in the sample
183 continues to collapse to 0 MPa at these later times. The average speed of the first few
184 cracks measured across multiple experiments is 500 ± 90 m/s, and this is measurement
185 by tracking the displacement of the crack tip over multiple camera frames across many
186 experiments.

187 Next we consider a dynamic uniaxial compression experiment in Figure 4. The
188 stress-time curve is shown on the left, and time-resolved high-speed camera images (t_1
189 to t_6) are shown on the right. The numbered grey dots in the stress-time plot indicate
190 the times and values of stress corresponding to each numbered image. Here, images
191 are shown at $8 \mu\text{s}$ intervals. The loading of the material in the high-speed video images
192 occurs from left to right, and red arrows are used to highlight the location of cracks
193 in the specimen. In this experiment, the compressive strain rate was measured to be
194 $1,000 \text{ s}^{-1}$ as calculated from the standard Kolsky bar analysis (Ramesh, 2008). The
195 stress time-plot indicates that the stress rises in a nearly linear manner to a peak value
196 of 294 MPa (the uniaxial compressive strength) at $\sim 20 \mu\text{s}$ after loading. Just before
197 peak stress (image t_1), we see no cracks on the surface of the specimen. After the peak
198 stress is reached cracks grow, reducing the stress in the sample. At t_2 , we see one crack
199 near the lower right surface. In subsequent images this crack continues to extend, and
200 other cracks can be seen on the surface as indicated by the red arrows. At the time
201 corresponding to t_4 , the stress in the sample has fallen to zero, and subsequent images
202 show that the specimen continues to fracture and expand in the vertical direction. Here
203 we highlight bright regions that intersect the cracks using the red arrows. The average
204 speed of cracks across multiple images has been measured to be 139 ± 64 m/s. Images

205 of t_5 and t_6 show a multitude of cracks that have aligned with the axis of loading, as well
206 as some additional bright regions intersecting these cracks that are believed to be iron-
207 nickel grains. It has been documented elsewhere, that these axial cracks form columns,
208 and at later times, these columns buckle and transverse fracturing (i.e., perpendicular to
209 the axial cracks) occurs (Ashby and Hallam, 1986; Hogan et al., 2014). Interestingly,
210 we also observe bright features on the surface of high-speed camera images in Fig. 10
211 of Kimberley and Ramesh (2011) and we see fracture intersecting these bright features.
212 The bright spots are iron-nickel grains in MAC 88118 (also confirmed with our recent
213 post-experiment analysis of the fragments). We believe a similar failure process occurs
214 in our experiments of GRO 85209, and that iron-nickel grains serve as initiation sites
215 for failure.

216 *3.2. Fracture Initiation and Fragment Characterization*

217 Micro-scale modes of failure are determined by examining fragments (collected
218 from each experiment) in both optical and scanning electron microscopes. Figure 5
219 shows, as an example, images of fragments collected from a dynamic uniaxial com-
220 pressive experiment. First we examine an optical microscope image of internal fracture
221 features inside of a polyphase fragment (Figure 5a). To acquire this image, some frag-
222 ments are mounted in resin and systematically polished through their cross-section. In
223 the example shown in Figure 5a, exposed iron-nickel grains (highlighted by white ar-
224 rows) are observed protruding from the fracture surface and there is evidence of internal
225 fractures intersecting regions of high angularity of the iron-nickel grains (highlighted
226 by yellow arrows). This suggests that fracture of GRO 85209 initiates from iron-nickel
227 grains, and that compressive failure of this material results in the generation of internal
228 damage. Stiffness and hardness mismatches between the iron-nickel grains and adja-

229 cent pyroxene grains (the lighter grain materials in Figure 5a) are believed to promote
230 fracture at these boundaries. Here, the stiffness for pyroxene is 95 ± 3 GPa, and for the
231 iron-nickel 178 ± 6 GPa (measured using nano-indentation).

232 Next, we examine a representative portion of fragments in the image in Figure 5b. In
233 this image, there are two different features of fragments: 1. optically bright fragments
234 and 2. optically dark fragments. The brighter fragments are primarily comprised of
235 pyroxene (confirmed with scanning electron microscopy and electron dispersive spec-
236 troscopy). The brightness of the fragments appears to be dependent on their size, with
237 smaller ones being much brighter and the larger ones being grainy. Examples of both are
238 highlighted. They appear grainy because their surface morphology is highly variable,
239 and they are comprised of multiple pyroxene and iron-nickel grains. We also observe
240 iron-nickel grains protruding from the surface of the larger fragments, which is also
241 shown in Figure 5a. The individual darker fragments in Figure 5b are iron-nickel grains
242 (confirmed with electron dispersive spectroscopy). We see that the darker fragments are
243 mainly around $150\ \mu\text{m}$ in size, with only a few smaller ones.

244 We explore the surface morphology of a large fragment further in the scanning elec-
245 tron microscope image in Figure 5c. The fragment is mainly comprised of pyroxene
246 and the surface is quite jagged. This is believed to be a result of transgranular fracture
247 (Figure 5d) of the pyroxene mineral, which is believed to be the dominant failure mode of
248 this mineral due to its relative weak cleavage plane. Shown in Figure 5d is a magnified
249 image of a region on the pyroxene fragment. In the centre of this image is an iron-nickel
250 grain that is protruding from the surface, much like in Figure 5a and b. All together,
251 these results highlight the character (composition) of different fragment sizes, as well
252 as provide further evidence that iron-nickel grains play an important role in the failure
253 process in GRO 85209.

254 **4. Connections Between Defects and Strength**

255 *4.1. The Scaling of Rate Dependent Strength*

256 In this section, the results of the strength measurements obtained in the previous
257 section are compared with the rate dependent strength model of Kimberley et al. (2013).
258 This model describes the rate dependent strength of brittle materials by incorporating
259 fundamental physics related to crack initiation, growth, and interaction. The model is
260 sensitive to key microstructural (e.g. flaw size) and material parameters (e.g. Young's
261 modulus), and takes the following form:

$$\frac{\sigma_c}{\sigma_0} = 1 + \left(\frac{\dot{\epsilon}}{\dot{\epsilon}_0} \right)^{2/3}. \quad (2)$$

262 Here, σ_c is the compressive strength, and $\dot{\epsilon}$ is the applied strain rate. σ_0 is a character-
263 istic compressive strength term (taken as the quasi-static compressive strength) which
264 depends on the internal flaw distribution:

$$\sigma_0 = \alpha \frac{K_{Ic}}{\bar{s}\eta^{1/4}}, \quad (3)$$

265 where K_{Ic} is the mode I fracture toughness (Pa \sqrt{m}), \bar{s} is the average flaw size (m) and η
266 is the areal flaw density (m⁻²). The term α is a dimensionless proportionality constant.
267 The corresponding characteristic compressive strain rate, $\dot{\epsilon}_0$, is defined as:

$$\dot{\epsilon}_0 = \alpha \frac{v_c K_{Ic} \eta^{1/4}}{\bar{s}E} \quad (4)$$

268 where v_c is a limiting crack growth speed (m/s), and E is the Young's modulus (Pa).
269 Kimberley et al. (2013) have shown that this model captures the behavior of a large

270 number of brittle materials, including engineered ceramics and geological materials.

271 To compare the results of the unconfined compression experiments with this model,
272 values of σ_0 and $\dot{\epsilon}_0$ were fit to the experimental data using the functional form of Equa-
273 tion (2), and are presented in Table 1. The normalized strength data are plotted in
274 Figure 6 along with the normalized strength data for meteorite MAC 88118 (Kimberley
275 and Ramesh, 2011). Here we see that the both sets of experimental show little variation
276 in strength for low rates, but show a significant (2-4X) increase in strength at elevated
277 strain rates, agreeing well with the model predictions.

278 Kimberley et al. (2013) also showed that the model presented in Equations (2 – 4)
279 can be applied to tensile loading, although the characteristic stresses and strain rates
280 will take on different values under tensile vs. compressive loading (because different
281 flaw distributions are exercised). Thus the results of the indirect tension tests on GRO
282 samples can also be compared with this model as shown in Figure 7. As no quasi-static
283 indirect Brazilian disk data were available, nor were any direct tension data, we take
284 the tensile quasi-static strength to be 1/10 of the quasi-static compressive strength. The
285 choice of a ratio of 1/10 is motivated by tensile and compressive strengths found in the
286 handbook by Charles (2001). The quasi-static tensile strengths are those points for $\dot{\epsilon}/\dot{\epsilon}_0$
287 < 1 . The normalized experimental data agree well with the model, and shows an even
288 more significant increase in strength at higher normalized rates when compared with the
289 compressive results. This dramatic increase in strength is reflected in the lower value of
290 characteristic strain rate shown in Table 1.

291 Since the characteristic stress (Equation (3)) and characteristic strain rate (Equation
292 (4)) are expressed in terms of material properties and flaw distribution parameters, the
293 best fit values of Table 1 can be compared with the values that would be predicted
294 based upon measured material properties and flaw statistics. This is explored in the next

295 subsection.

296 4.2. Relations Between Flaw Population and Strength

297 Flaw population statistics are calculated for the iron-nickel grains, which we believe
298 to be the most important defect for our testing conditions. Note that the iron-nickel
299 grain morphology does not resemble the slit-like flaw geometry that formed the basis of
300 Equations (2 – 4). However, the distribution of iron-nickel grains is relevant if failure
301 is controlled by cracks extending from these grains, as observed in Figure 5. We are
302 interested in the average defect size \bar{s} and the number of defects in a given area, de-
303 noted as η ($\#/m^2$). We use these measured flaw statistics to explore our strength results
304 for GRO 85209, as well as compare our strength measurements with existing measure-
305 ments for another ordinary chondrite (MAC 88118) used by Kimberley and Ramesh
306 (2011). MAC 88118 is an L5 ordinary chondrite meteorite that was studied in Kim-
307 berley and Ramesh (2011), and it contains a different microstructure than GRO 85209.
308 For reference, we show the MAC 88118 microstructure in Figure 8a. Again, the lighter
309 features are iron-nickel grains and the darker matrix material is primarily comprised of
310 pyroxene. Examining the microstructure of MAC 88118 (Figure 8a) there appear to
311 be more defects (iron-nickel grains) and some of which are much larger in size than
312 observed in the GRO 85209 material (Figure 1b). The image processing routine previ-
313 ously outlined was also used to determine the size and total number of defects per area
314 ($\#/m^2$) for MAC 88118, allowing of the differences in microstructure to be quantified.
315 In Figure 8c we plot the areal density of defects larger than the corresponding defect
316 size. We use a power-law fit in the form of:

$$PL(x) = Cs^{-n} \quad (5)$$

317 where C and n are fitted coefficients to experimentally measured defect density data.
 318 Over the range of characterized defects (here we consider defects larger than $5 \mu\text{m}$),
 319 GRO 85209 has fewer defects per unit area than MAC 88118, and the rate at which
 320 the defect density decreases (i.e., the magnitude of n) is greater for GRO 85209. This
 321 confirms the qualitative observation that MAC has higher flaw density.

322 We also characterize the defect half-sizes, defined as half of the longest spanning
 323 dimension of the iron-nickel grains in each material. Shown in Figure 8c is a probabil-
 324 ity plot of the defect half-size (s) for GRO 85209 as measured from figures similar to
 325 those in Figure 1b. The corresponding defect size distribution for the iron-nickel grains
 326 in MAC 88118 (measured from figures similar to those in Figure 8a) are shown in Fig-
 327 ure 8d. Note here that we are only considering defects larger than $5 \mu\text{m}$ in the probability
 328 plot based on our ability to resolve the features in the optical microscope image. The
 329 probability plot is used for assessing whether or not an empirical data set (here it is the
 330 defect size) follows a given reference distribution (e.g., lognormal, normal). In a prob-
 331 ability plot, the y-axis is scaled accordingly to make the selected reference distribution
 332 appear as a line. Differences between the reference line and the data set indicate a lack
 333 of fit. Mathematically: consider an ordered sets of data:

$$\bar{x}_{(1)}, \bar{x}_{(2)}, \dots, \bar{x}_{(m)} \quad (6)$$

334 with probability distribution functions of $g(\bar{x})$. The cumulative distribution function,
 335 $G(x)$, is given as:

$$G(x) = \int_0^x g(\bar{x}) d\bar{x} \quad (7)$$

336 where $G(x)$ ranges between 0 and 1. From this, we are then able to compute percentile
 337 values of $G(x)$ (the 35th percentile occurs when $G(x)=0.35$). If $F(y)$ is the cumulative

338 distribution of a reference distribution (e.g., lognormal or normal) then we are able to
 339 contrast expected percentiles for both the data ($G(x)$) and reference distribution ($F(y)$).
 340 In Figures 8c and d, the defect sizes are compared against a lognormal distribution in
 341 the form:

$$f(x)_{\log} = \frac{1}{x\sigma_{\log}\sqrt{2\pi}} e^{-\frac{(\log(x)-\mu_{\log})^2}{2\sigma_{\log}^2}} \quad (8)$$

342 where μ_{\log} and σ_{\log} are the mean and standard deviation of the data's logarithm, respec-
 343 tively. The defect sizes are adequately described using a lognormal distribution for sizes
 344 $> 10 \mu\text{m}$, with the corresponding values of μ_{\log} and σ_{\log} denoted in each subfigure. For
 345 reference, the mean defect size for GRO 85209 with standard error is $27.7 \pm 0.4 \mu\text{m}$,
 346 and $51.2 \pm 1.9 \mu\text{m}$ for MAC 88118. The standard error is calculated by dividing the
 347 standard deviation by the square root of the number of size measurements. We pro-
 348 vide a summary of the defect statistics and material properties in Table 2. Note that the
 349 reported crack growth speeds v_c were measured in several dynamic uniaxial compres-
 350 sion experiments for each material (the measured tensile growth speeds for GRO were
 351 larger). Also note that defect densities are computed by taking the average areal density
 352 across 100+ images for both materials. We report the average of those values and the
 353 standard deviation in the last column of Table 2.

354 With the measured properties and flaw statistics for both MAC 88118 and GRO
 355 85209, we can compare the changes in characteristic stress and strain rate with the
 356 experimentally determined best fit values shown in Table 1. Equation (3) allows for the
 357 ratio of characteristic stresses under compression for GRO 85209 and MAC 88118 to
 358 be expressed as:

$$\frac{\sigma_{0G}}{\sigma_{0M}} = \frac{K_{IcG}}{K_{IcM}} \frac{\bar{s}_M}{\bar{s}_G} \left(\frac{\eta_M}{\eta_G} \right)^{1/4}. \quad (9)$$

359 Here the subscripts G and M correspond to the properties of GRO 85209 and MAC

360 88118, respectively. Using the properties listed in Table 2 and assuming that the frac-
 361 ture toughness of the two materials are equal (there exist no measurements of fracture
 362 toughness for stony meteorites) we calculate $\sigma_{0G}/\sigma_{0M} = 1.49$. This compares reason-
 363 ably with the ratio of the experimental best fit values, 1.82.

364 The ratio of characteristic rates is derived from Equation (4) to be

$$\frac{\dot{\epsilon}_{0G}}{\dot{\epsilon}_{0M}} = \frac{K_{IcG} c_G E_M \bar{s}_M}{K_{IcM} c_M E_G \bar{s}_G} \left(\frac{\eta_G}{\eta_M} \right)^{1/4}. \quad (10)$$

365 Using the values listed in Table 2 we find the ratio of characteristic strain rates $\dot{\epsilon}_{0G}/\dot{\epsilon}_{0M} =$
 366 0.52, which compares well with the ratio of observed values, 0.75. If we assume that the
 367 ratio of fracture toughnesses is $K_{IcG}/K_{IcM} = 1.22$ then the ratio of characteristic stresses
 368 can be forced into agreement. Our predicted ratio of characteristic rates would then
 369 equal 0.63, in closer agreement with our best fit ratio. The above calculations indicate
 370 that the strength model presented in Equations (2–4) is capable of capturing the exper-
 371 imentally observed trends (i.e., higher characteristic strength, and lower characteristic
 372 rate in GRO 85209) based upon difference in microstructure and material properties.

373 **5. The Role of Microstructure on Dynamic Fragmentation**

374 In this section we explore the effect of the microstructure’s composition and defect
 375 population on the fragmentation of GRO 85209 for both stress-states we previously
 376 studied. This is mainly accomplished using the image processing techniques previously
 377 outlined to measure fragment size and mean greyscale intensity, as well as the spacing
 378 between the iron-nickel grains.

379 *5.1. Fragment Size and Defect Spacing Distributions*

380 Cumulative distributions of fragment sizes from GRO 85209 are shown in Figure 9
381 for both the quasi-static (red dashed line) and dynamic (red solid line) uniaxial com-
382 pression experiments, and a lower rate and higher rate for the Brazilian disk experi-
383 ments (blue lines). Note that some uncertainty exists for fragments $<30 \mu\text{m}$ as these
384 fragments are challenging to collect after the experiments. For this reason, we do not
385 include them the analysis. Initially, we discuss the quasi-static uniaxial compression
386 experiment (dashed red line in Figure 9). The strain rate here is 10^{-3} s^{-1} . Most of the
387 fragments are between $30 \mu\text{m}$ and 1 mm in size, and the cumulative distribution shows
388 an inflection at around $120 \mu\text{m}$. We believe that this suggests that two different fragmen-
389 tation mechanisms may be present. As the eCDF represents the relative frequency, we
390 note that about 33 % of the fragments generated by quasi-static uniaxial compression
391 are less than $120 \mu\text{m}$ in size. The cumulative distribution of fragment sizes for the dy-
392 namic uniaxial compression case is shown using the solid red curve. The strain rate here
393 is 10^{+3} s^{-1} . The curve is shifted to the left compared to the uniaxial case, indicating that
394 increasing the strain rate produces smaller fragments. This is due to additional internal
395 strain energy at the peak stress (due to higher rate), and this energy is subsequently con-
396 verted into more fractures (hence smaller fragments). Additionally, more defects may be
397 activated at higher rates, thus facilitating increased fracturing and fragmentation. Also
398 note that the bump at around $120 \mu\text{m}$ in size persists, but is less prominent. We see that
399 about 57 % of the fragments generated in dynamic uniaxial compression are less than
400 $120 \mu\text{m}$ in size. We divide the distributions by fragment size at a size of $120 \mu\text{m}$, with
401 the domain $\ell_i < 120 \mu\text{m}$ called fragmentation Regime 1, and that with $\ell_i > 120 \mu\text{m}$ called
402 Regime 2.

403 Next, we examine the cumulative distribution of fragment sizes for two Brazilian

404 disk experiments, one at a strain rate of 45 s^{-1} and one at 285 s^{-1} . Note again, that the
405 deformation in the Brazilian disk tests is quite non-uniform (not pure tension), and this
406 likely results in a different sequence of events leading to the eventual fragmentation of
407 the sample. As before, the curves shift to left as the strain rate is increased, with Regime
408 1 representing 37 % of the total fragment population for the 45 s^{-1} case and 60 % for
409 the 285 s^{-1} experiment. The inflection at around $120 \mu\text{m}$ exists for these materials as
410 well.

411 Why are there two regimes of fragmentation and what is the significance of the in-
412 flection at around $120 \mu\text{m}$? In Figure 9, we also plot the cumulative distribution of spac-
413 ings between the iron-nickel grains (solid black curve) and observe that the maximum
414 defect spacing appears to coincide with the inflection at $120 \mu\text{m}$. There are a total of
415 6,200 measurements taken, so we believe the curve is representative of the actual data.
416 We believe that this suggests that fragments $< 120 \mu\text{m}$ (Regime 1) are controlled by the
417 microstructure defect spacing. In this mechanism, fractures initiated at the iron-nickel
418 grains may coalesce with fractures initiated from adjacent iron-nickel grains.

419 *5.2. Fragment Composition*

420 In this last subsection, we investigate measurements of the mean greyscale intensity
421 (GI) of fragments in the scatter plots of GI plotted against fragment sizes in Figure 10.
422 In order to estimate a mean greyscale intensity, pixel values in monochrome images
423 obtained through thresholding are related to pixel values in the original greyscale im-
424 ages, and the values are then averaged over the entire fragment. We normalize the mean
425 greyscale intensity by the maximum to have the values range between 0 and 1. We
426 show the Brazilian disk experiment at a strain rate of 285 s^{-1} in Figure 10a. Again, we
427 only consider fragments $> 30 \mu\text{m}$. In Figure 10a, we divide concentrations of points

428 in the scatter plot into three sub-regions: Sub-region A, with fragments less than 120
429 μm in size and > 0.4 in GI that are primarily optically bright, which are believed to
430 be comprised of pyroxene fragments (Figure 5b); sub-Region B, with fragments larger
431 than $120 \mu\text{m}$ and GI larger than 0.4, which are polygrain and polyphase in nature and
432 contain multiple grains of iron-nickel and pyroxene grains (Figure 5a); sub-Region C,
433 with fragments with GI less than 0.4, i.e., dark in color, which are believed to be indi-
434 vidual iron-nickel grains (Figure 5b). Although the bounds are drawn for all fragments
435 sizes for sub-Region C, there appears to be a cluster of fragments that are between 70
436 and $180 \mu\text{m}$. Iron-nickel grains larger than $70 \mu\text{m}$ represent 25 % of the total iron-nickel
437 grain population. Similar sub-region trends are observed for the dynamic uniaxial com-
438 pression case, which is shown in Figure 10b, although with different concentrations for
439 each sub-region.

440 The total percentage (%) -representation of each sub-region to the total population is
441 computed in Table 3. There is only one set of measurements for each of the strain rates
442 (low and high) and stress states (Brazilian disk and uniaxial compression). We associate
443 the uncertainty with each %-population measurement with the choice of boundaries. As
444 examples, the bounds for the % population in sub-Region A is obtained by considering
445 the population for all fragments less than 110 (minimum), or less $130 \mu\text{m}$ (maximum).
446 For sub-regions B and C, we vary the size bounds between 110 and 130, and the color
447 bounds between 0.35 and 0.45, and compute the associated % population in Table 3.
448 The associated uncertainty is about $\pm 2 \%$ for any of the measurements. From Table 3,
449 there is an increase in the total %-representation of sub-Region A (pyroxene fragments)
450 for increasing strain rate, and that the %-representation of sub-Region A for the Brazil-
451 ian disk experiments is greater than for the uniaxial compression experiments. The
452 %-representation of sub-Region C (iron-nickel grains) decreases as the strain rate is in-

453 creased, and, again, the Brazilian Disk experiments have a greater %-representation for
454 sub-Region C than the uniaxial compression experiments.

455 **6. Summary and Implications**

456 During failure, stored strain energy is converted to kinetic energy and surface energy,
457 tensile stresses are created, and fragmentation ensues via crack growth and coalescence.
458 In this study we have shown that the resulting fragment characteristics (sizes and com-
459 position) are dependent on where cracks are initiated, how cracks grow and branch,
460 and how they coalesce. In GRO 85209, we believe that fractures are commonly initi-
461 ated at the iron-nickel grain boundaries during dynamic compressive loading, likely a
462 result of the stiffness mismatches between iron-nickel and pyroxene. Here, the stiff-
463 ness for pyroxene is 95 ± 3 GPa, and for the iron-nickel 178 ± 6 GPa (measured using
464 nano-indentation).

465 As these cracks grow, their paths will be dependent on the material composition and
466 grain boundary relationships, and the strain-rate and stress-state. In GRO 85209, scan-
467 ning electron and optical microscopy identified transgranular fracture as the dominant
468 fracture mechanism in pyroxene, while intergranular fracture was observed to dominate
469 in the metallic iron-nickel phases. As the strain rate is increased, we expect more small
470 fragments to be formed due to added strain energy associated with the strength increase
471 (see Figure 6), and this manifests as a shift to the left in the cumulative distribution
472 curves. We may also expect less intergranular fracture and more transgranular fracture.
473 We observe this trend in the fragmentation composition results, where pyroxene rep-
474 resents a larger number in the total fragment population as the strain rate is increased
475 for both materials. Similarly, as the stress-state becomes more multidimensional and
476 non-uniform, we expect there to be more transgranular fracture than intergranular frac-

477 ture. The comparison between the compression and Brazilian disks tests indicate that
478 pyroxene represents more of the population in Brazilian disk case, perhaps a result of
479 increased transgranular fracture. All together, fragmentation appears to be dependent
480 on the strain-rate and stress-state, the time-history of fractures (i.e., what happens first),
481 and the mineral compositions being fractured.

482 After growth at a finite crack speed, the fractures will eventually coalesce. Crack
483 propagation speeds for the Brazilian disk test were observed to be more than 3x greater
484 than in the uniaxial compression case (500 ± 90 m/s vs. 139 ± 64). This is, perhaps, a
485 result of less tortuous paths experienced by the crack during loading in the Brazilian disk
486 experiment. Fracture coalescence is also different for both stress-states studied here. In
487 the Brazilian disk experiment, fractures are first observed to grow horizontally across the
488 disk, and this partitions the disks into two hemispheres. At later times, the edges of the
489 disk in contact with the platens break and this creates the fragments. In the compression
490 case, many axial cracks are observed to propagate across the sample. At later times,
491 these axial cracks span the entire sample, creating columns, and these columns buckle
492 at later times, resulting in transverse fracturing (Hogan et al., 2014). The differences in
493 structural failure likely manifest in the different fragmentation results.

494 After cracks are formed during structural failure, additional abrasion between sur-
495 faces is believed to generate the smaller fragments (pyroxene in composition) and indi-
496 vidual iron-nickel grain fragments. With this sequence of events in mind, we summarize
497 the key fragmentation regions as follows:

- 498 1. Sub-Region A: This fragmentation mechanism is associated with the initiation,
499 propagation and coalescence of fractures between iron-nickel grains. Fragments
500 in this regime are less than $120 \mu\text{m}$ in size, and are primarily comprised of py-
501 roxene fragments, which is the weakest mineral phase and the matrix material

502 in GRO 85209. Weaker materials are preferentially comminuted during high-
503 rate events (Spray, 2010). We associate the increase in %-representation of the
504 smaller pyroxene fragments (sub-Region A) for increasing strain rates (independ-
505 ent of stress-state) as a result of the additional strain energy in the sample and the
506 subsequent increase in the number of activated defects. Smaller fragments being
507 composed in the matrix material was also noted by Durda and Flynn (1999) and
508 Flynn and Durda (2004), and the has implications in the collection of interplane-
509 tary dust particles (believed to originate from the asteroid belt), where they note
510 that the smallest fragments composition may not be wholly representative of the
511 parent material (e.g., in terms of volatile content).

512 2. Sub-Region B: Fragments in this region are greater than 120 μm in size, and
513 polyphase and polygrain in character. These are developed through failure mech-
514 anisms that occur during the structural failure of the sample. The mechanisms ac-
515 tivated during failure are dependent on the geometry, loading history, stress-state,
516 and strain-rate. In uniaxial compression, failure occurs through the coalescence of
517 axial and transverse cracks (Horii and Nemat-Nasser, 1985). In the Brazilian disk
518 experiment, multiple horizontal cracks first grow across the sample to partition
519 the disk into two hemi-disks, and at later times the hemi-disks fracture.

520 3. Sub-Region C: Fragments in this regime are comprised of individual iron-nickel
521 grains that are between 60 and 200 μm in size. We believe these fragments are
522 too ductile to easily fracture again.

523 Similar links between the microstructure and fragmentation have been noted by Durda
524 and Flynn (1999), wherein fragment sizes were noted to be related to matrix materials or
525 individual grains. In their case, the chondrules dominate the failure sites during impact

526 in a meteorite materials, whereas in our case its the iron-nickel grains.

527 Lastly, the boundary between both the microstructure- and structural-dependent
528 fragmentation regimes ($\approx 120 \mu\text{m}$) is less distinct in this material than in the advanced
529 ceramic boron carbide (Hogan et al., 2014), where a clear separation exists for frag-
530 ments between 70 and 100 microns in size. This can be possibly explained as follows:
531 the lower bound of the structural controlled fragmentation (Regime 2) is related to the
532 energy that is available to fragment the body through axial and transverse cracking (in
533 the case of compression). One could essentially view this regime as those described by
534 the rate-dependent brittle fragmentation models that exist in the literature (e.g., Grady
535 (2006), Glenn and Chudnovsky (1986), Zhou et al. (2006a), and Levy and Molinari
536 (2010)). On the other hand, the upper bound of the microstructure regime is believed
537 to correspond to the upper bound of the spacing distribution. Certainly, energy is also
538 required to create these fragments as well, but the mechanism is different. As the strain
539 rate is increased, more and more defects are probed, so more of the spacing distribu-
540 tion is probed. For some sufficiently high strain rate, the microstructure-dependent and
541 structural-dependent mechanisms will begin to overlap over a certain size range as they
542 compete against each other. For boron carbide, the characteristic strain rate in compres-
543 sion is approximately 10^4s^{-1} , whereas the characteristic strain rate in compression for
544 GRO 85209 is 150s^{-1} . Thus, at a comparable testing strain rate ($\approx 10^3\text{s}^{-1}$), one would
545 expect the failure and fragmentation of the GRO 85209 to be more catastrophic (since
546 the rate is well-beyond where it exhibits strain-rate dependence), and thus perhaps more
547 overlap between the regimes is to be expected.

548 6.1. Implications

549 Insights gained from the experimental strength and fragmentation results have im-

550 plications in the generation of regolith and catastrophic disruption. The fragmentation
551 results and past studies indicate that fragment sizes in an impact event will be dependent
552 on defect populations (links made here) and mechanical properties (Grady, 1982). The
553 crack speed is also influential in the fragmentation process. Crack speeds in this ordi-
554 nary chondrite (GRO 85209: 139 ± 64) are lower than those reported by Hogan et al.
555 (2015) for basalt (650 ± 100), mainly due to the added porosity and increased plasticity
556 in GRO 85209. This will have consequences in fragmentation outcomes, where the rel-
557 atively lower crack speeds would result in smaller fragments for the ordinary chondrite
558 than a basaltic material (since fragment size is proportional to crack speed multiplied
559 by a time). This has two implications: (1) finer regolith would be generated on bodies
560 composed of ordinary chondrite (compared to basalt), and the ordinary chondrite would
561 be harder to disrupt than a basaltic (since it is more challenging to yield larger enough
562 fragments). Experimental evidence to both is found in Cintala and Hörz (2008). Addi-
563 tional consideration of the influence of crack speed, mechanical properties, and defect
564 populations may also be important in interpreting regolith formation on planets (e.g.,
565 lunar mare vs. highlands). For the lunar mare (mainly basaltic), results from this paper
566 and Hogan et al. (2015) will provide insight into its failure and fragmentation mecha-
567 nisms. The interpretation of fragmentation from the highlands may be different, since
568 it is mainly comprised of anorthosite (a monophase material), and the key defects and
569 crack speeds are not yet well understood in this material.

570 Additionally, the functional form of Equation (4) yields some insights into what
571 happens during repeated impacts. Equation (4) predicts that the characteristic strain rate
572 is proportional to the quarter root of the areal flaw density, and inversely proportional
573 to the flaw size. Under the action of many impacts that are not severe enough to cause
574 fragmentation, some cracks within the body will be activated, grow and eventually link

575 up, resulting in an increase in the flaw sizes in the body (and perhaps a net decrease in the
576 number of flaws as a result of the linkup). A net increase in defect size and defect density
577 would result in a decrease in the characteristic strain rate of the material (according to
578 equation (4), and thus the material would be harder to disrupt. The consequence of
579 repeated impacts can also evolve the net defect population (identified as microstructural
580 heterogeneities *as well as* the newly introduced cracks) in other ways (both in terms
581 of net defect density and the shape of the distribution. In turn, the characteristic strain
582 rate may change leading perhaps to changes in the effective dynamic strength of the
583 material.

584 Lastly, the direct link between microstructure (e.g., composition, defect spacing)
585 and fragmentation also has implications for developing analytical models to predict the
586 size of regolith on airless bodies through mechanical fragmentation (e.g., fragmentation
587 models by Grady (1982); Glenn and Chudnovsky (1986); Zhou et al. (2006b); Levy
588 and Molinari (2010)). In the current study we showed that there are two fragmentation
589 mechanisms associated with the failure of GRO 85209: a mechanism associated with the
590 structural failure, and a mechanism associated with the microstructure. Our results indi-
591 cate that the microstructure-controlled mechanism becomes more important for increas-
592 ing rate and for increasingly complicated stress-states. We hypothesize that the rates and
593 stress-states that manifest during impact would tend to favor microstructure-dominated
594 fragmentation. In the recent review paper by Ramesh et al. (2015), the current (essen-
595 tially structural) models by Grady (1982); Glenn and Chudnovsky (1986); Zhou et al.
596 (2006b); Levy and Molinari (2010) were shown to *over-estimate* the average fragment
597 size in uniaxial compression tests of basalt. This suggests that these analytical models
598 may not be sufficient to predict the fragmentation size outcomes of impact events (e.g.,
599 regolith generation) on basaltic bodies like the moon. **New models are needed that in-**

600 clude microstructure, allowing one to more fully understand how fragmentation evolves
601 during impact for different materials (e.g., lunar mare vs. highlights). In addition, we
602 also note that the analytical models of Grady (1982); Glenn and Chudnovsky (1986);
603 Zhou et al. (2006b); Levy and Molinari (2010) are developed for a tensile stress-state,
604 and new models are also needed for compression. Further challenges exist in the de-
605 scription and prediction of size distribution shapes (like those shown in Figure 9), and
606 incorporation of the activation of additional failure mechanisms into the fragmentation
607 models that depend on the loading history (Ramesh et al., 2015)

608 **7. Concluding Remarks**

609 We have examined microstructure and stress-state effects on the strength, failure
610 and fragmentation of an ordinary chondrite (GRO 85209). The iron-nickel grains have
611 been identified as sites for fracture initiation during dynamic uniaxial compression and
612 indirect dynamic tension (using the Brazilian Disk technique). The size and number per
613 area of the iron-nickel grains were then quantified, and in conjunction with a recently
614 developed scaling relation, they were used to explore our experimental results, as well
615 as the differences in strength between the ordinary chondrite in this paper and another
616 study by Kimberley and Ramesh (2011). Fragments were also collected after the ex-
617 periments, and measurements of their size, shape, and greyscale intensity were used
618 to inform us about inherent failure and microstructural lengths that are probed during
619 our loading conditions. Two fragmentation mechanisms were identified: one associated
620 with the structural failure of material and one associated with inherent microstructure
621 length scales (i.e., size and spacing of defects). Understanding the role of defects on the
622 strength, failure and fragmentation of representative extra-terrestrial materials is central
623 for developing improved numerical models of naturally-occurring planetary and space

624 science phenomena (e.g., regolith generation).

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729 belled. In this study we are interested in linking the size and number per
730 unit area of the iron-nickel grains, and the spacing between these grains
731 with strength, failure and fragmentation results. (c) Optical microscope
732 image of GRO 85209 fragments and (d) converted monochrome image
733 with fragment statistics defined. 37

734 2 Schematic of Brazilian Disk experiment with tensile stress (σ_y) labelled. 37

735 3 Stress-time history of dynamic Brazilian disk experiment of GRO 85209
736 (left) with time-resolved high-speed video images showing mesoscale
737 failure mechanisms (right). The hashed line in the stress-time plot is the
738 linear fit of 10 and 90 % of the peak stress and this corresponds to a
739 stress rate of $\dot{\sigma}=1.6 \text{ MPa}/\mu\text{s}$ 39

740 4 Stress-time history of dynamic uniaxial compression of GRO 85209
741 (left) with time-resolved high-speed video images showing mesoscale
742 failure mechanisms (right). 40

743 5 GRO 85209 optical microscope images of: (a) internal fracture features
744 inside of fragments, (b) the character of fragments showing optically
745 bright (pyroxene) and dark (iron-nickel) phases, as well as combinations
746 (polyphase and polygrain). (c) Scanning electron microscope image of
747 a pyroxene fragment with (d) highlighted region of an iron-nickel grain
748 protruding from the surface of the pyroxene fragment. 41

Table 1: Estimates for the characteristic stress (σ_0) and the characteristic strain-rate ($\dot{\epsilon}_0$) that provide the best fit of the experimental data to the strength model.

Material	Stress state	σ_0 (MPa)	$\dot{\epsilon}_0$ (s^{-1})
MAC 88118	Compression	50	200
GRO 85209	Compression	91	150
GRO 85209	Tension	10	35

749	6	Normalized uniaxial compressive strength data for MAC 88118 and	
750		GRO 85209 samples compared with the strength model of Kimberley	
751		et al. (2013).	42
752	7	Normalized indirect tensile strength data for GRO 85209 samples com-	
753		pared with the strength model of Kimberley et al. (2013).	43
754	8	(a) Optical microscope image of MAC 88118 microstructure, (b) Iron-	
755		nickel defect areal number density ($\#/m^2$) for all defects larger than the	
756		corresponding size on the x-axis. Probability plot of defect size (s) com-	
757		paring experimentally measured sizes with exponential distribution for	
758		(c) GRO 85209 and (d) MAC 88118.	44
759	9	GRO 85209: cumulative distribution of fragment major axis sizes and	
760		spacing between iron-nickel grains for uniaxial compression and in-	
761		direct tension using the Brazilian disk tests. Strain rates are labelled. . .	45
762	10	GRO 85209: mean greyscale color intensity (CGI) for: (a) Brazilian	
763		Disk experiment at a strain rate of $285 s^{-1}$ and (b) uniaxial compression	
764		experiment at a strain rate of $10^{+3} s^{-1}$	45

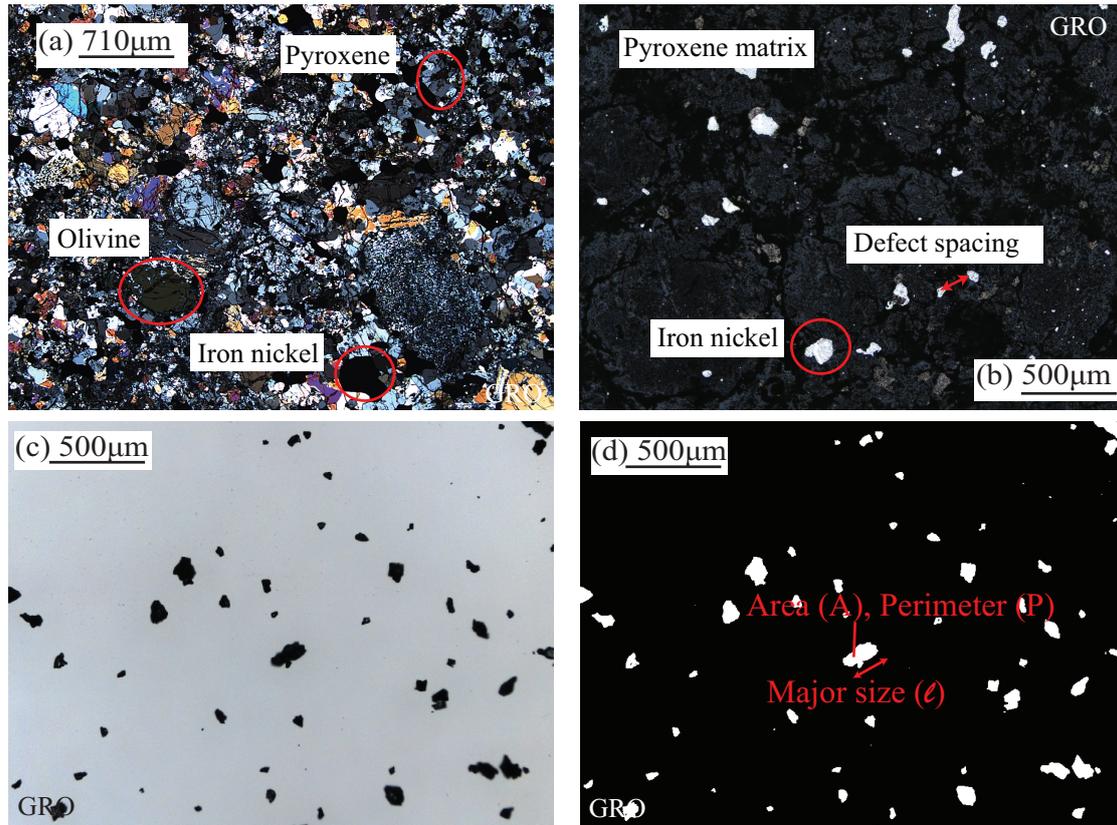


Fig. 1: (a) Polarized optical microscope image of GRO 85209 thin section (credit: Linda Welzenbach, Smithsonian Institute) and (b) non-polarized surface image of GRO 85209 microstructure, both with constituent phases labelled. In this study we are interested in linking the size and number per unit area of the iron-nickel grains, and the spacing between these grains with strength, failure and fragmentation results. (c) Optical microscope image of GRO 85209 fragments and (d) converted monochrome image with fragment statistics defined.

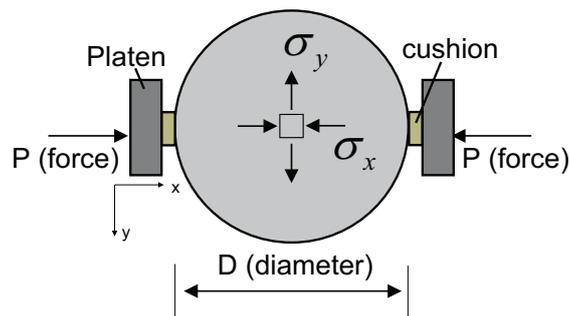


Fig. 2: Schematic of Brazilian Disk experiment with tensile stress (σ_y) labelled.

Table 2: Material properties and defect characteristics for GRO 85209 and MAC 88118, including: Young’s modulus (E : Pa), density (ρ : kg/m³), crack speed (v_c : m/s), average defect size with standard error (\bar{s} : μm), and defect density with standard deviation (η : #/m²).

Material	E GPa	ρ kg/m ³	v_c m/s	\bar{s} μm	η ($s > 5 \mu\text{m}$) #/m ²
GRO 85209	14	3,350	139 ± 64	27.7 ± 0.4	$3.5 \pm 1.1 \times 10^7$
MAC 88118	3.2	3,240	136 ± 60	51.2 ± 1.9	$7.8 \pm 2.5 \times 10^7$

Table 3: Percentage-contribution of sub-Regions A-C for uniaxial (UC) and indirect tension using Brazilian disk testing (BD). The uncertainty in each measurement is about $\pm 2\%$

sub-Region	UC at 10^{-3} s^{-1}	UC at 10^{+3} s^{-1}	BD at 45 s^{-1}	BD at 285 s^{-1}
A: Pyroxene-dominated	33	57	37	60
B: Polyphase and Polygrain	45	25	27	8
C: Iron-Nickel-dominated	22	18	36	32

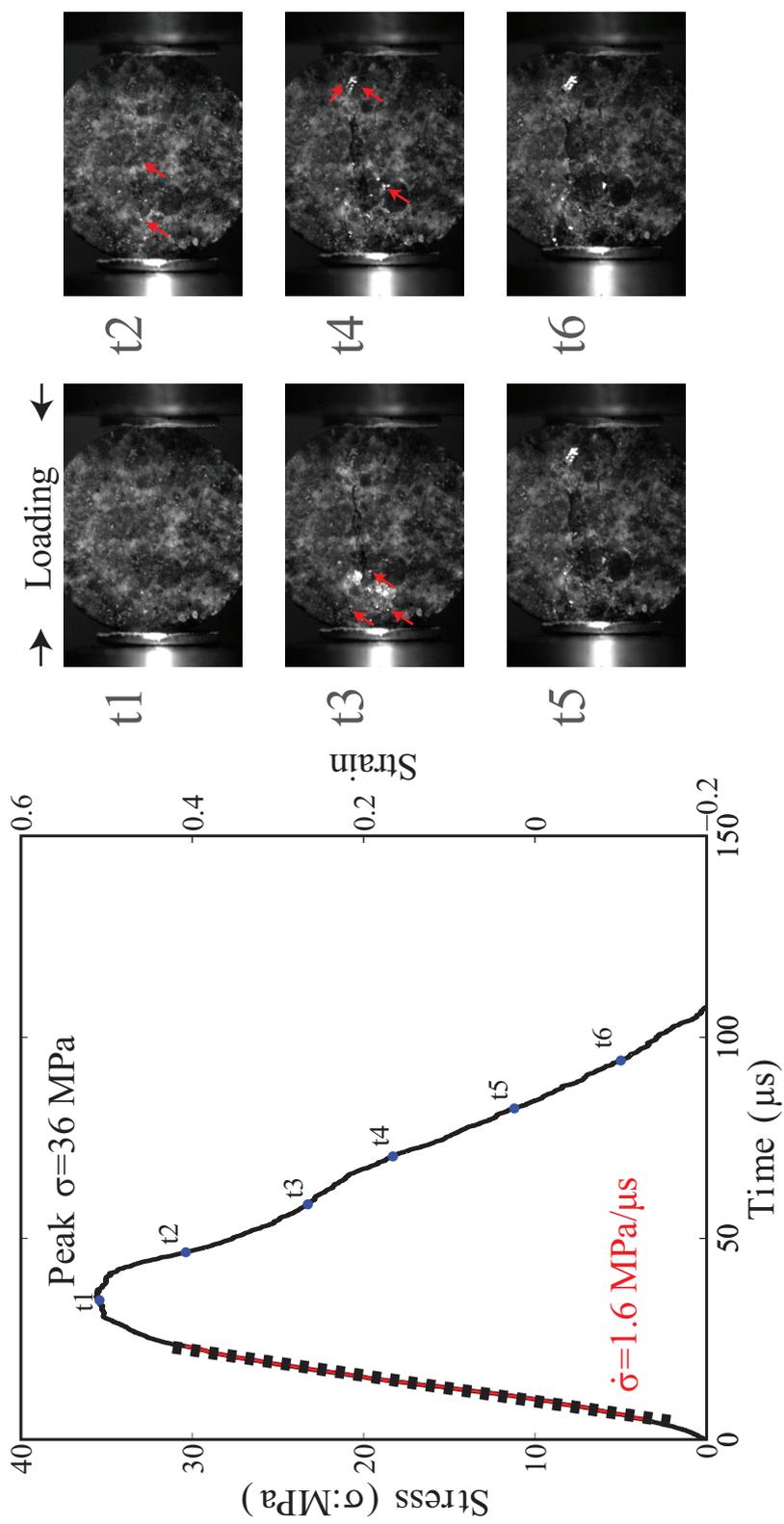


Fig. 3: Stress-time history of dynamic Brazilian disk experiment of GRO 85209 (left) with time-resolved high-speed video images showing mesoscale failure mechanisms (right). The hashed line in the stress-time plot is the linear fit of 10 and 90 % of the peak stress and this corresponds to a stress rate of $\dot{\sigma} = 1.6 \text{ MPa}/\mu\text{s}$.

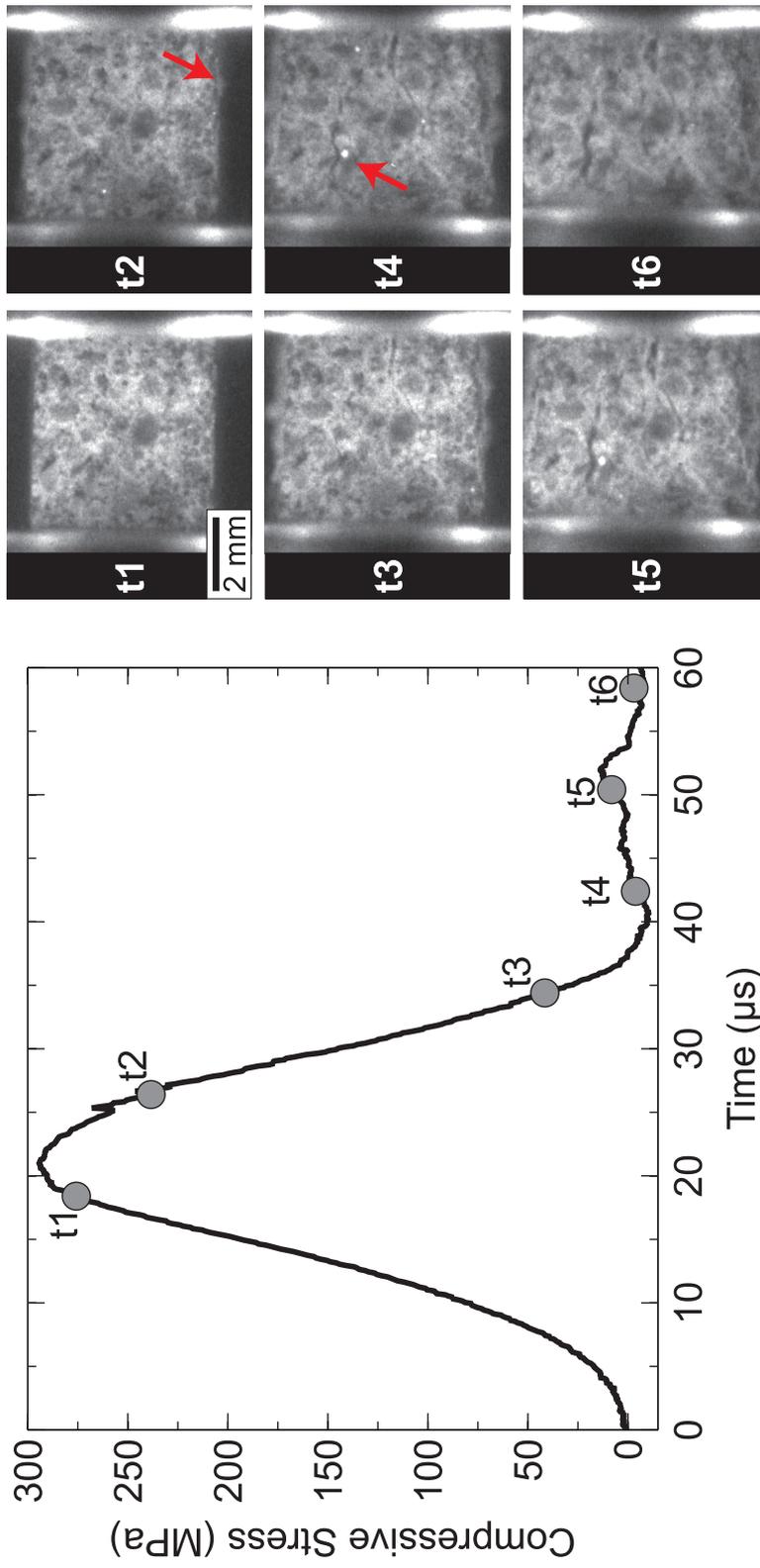


Fig. 4: Stress-time history of dynamic uniaxial compression of GRO 85209 (left) with time-resolved high-speed video images showing mesoscale failure mechanisms (right).

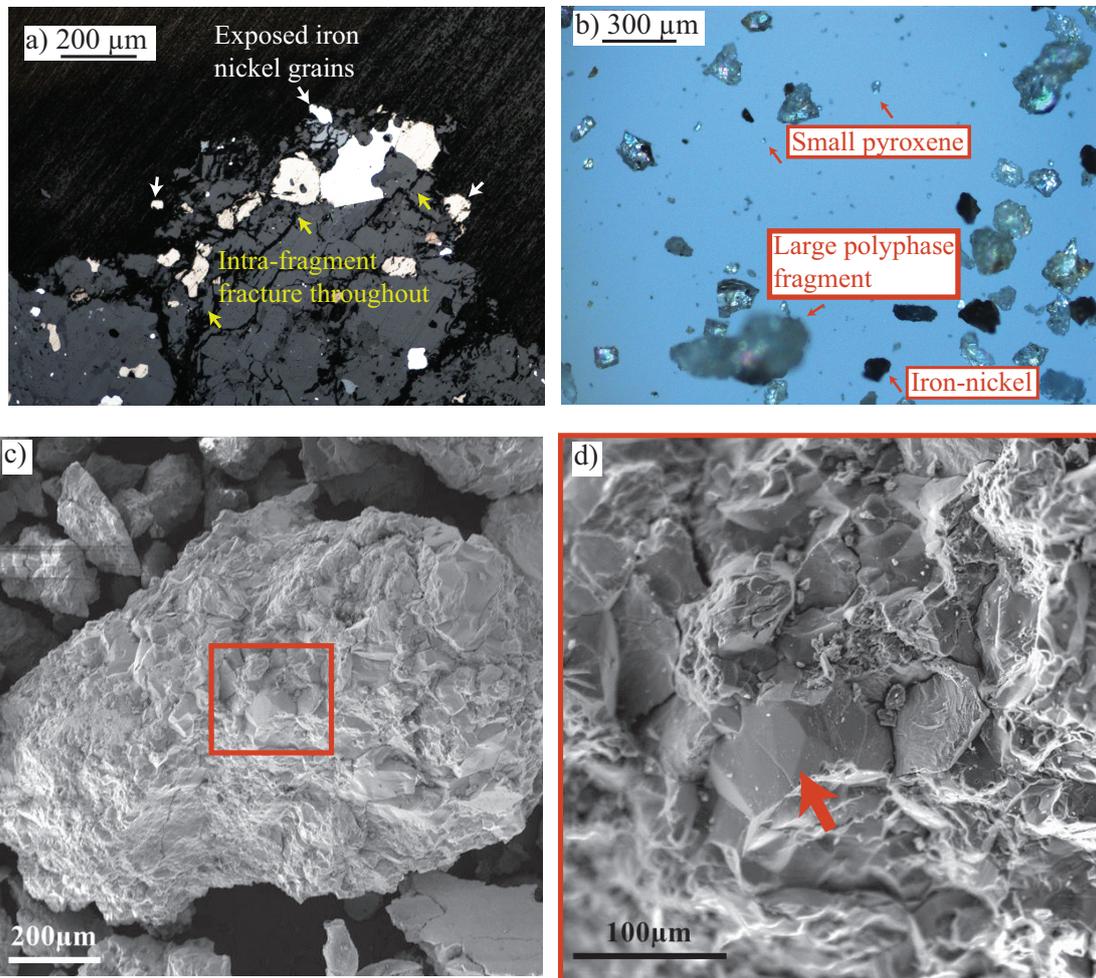


Fig. 5: GRO 85209 optical microscope images of: (a) internal fracture features inside of fragments, (b) the character of fragments showing optically bright (pyroxene) and dark (iron-nickel) phases, as well as combinations (polyphase and polygrain). (c) Scanning electron microscope image of a pyroxene fragment with (d) highlighted region of an iron-nickel grain protruding from the surface of the pyroxene fragment.

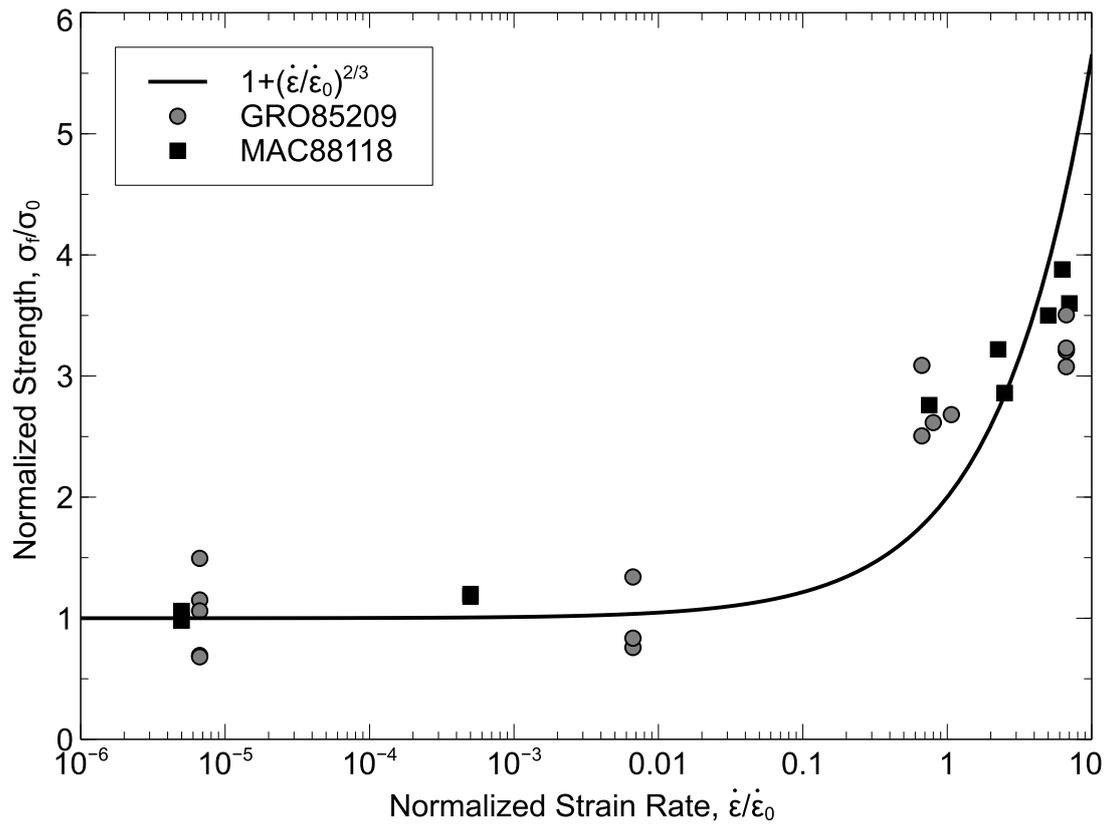


Fig. 6: Normalized uniaxial compressive strength data for MAC 88118 and GRO 85209 samples compared with the strength model of Kimberley et al. (2013).

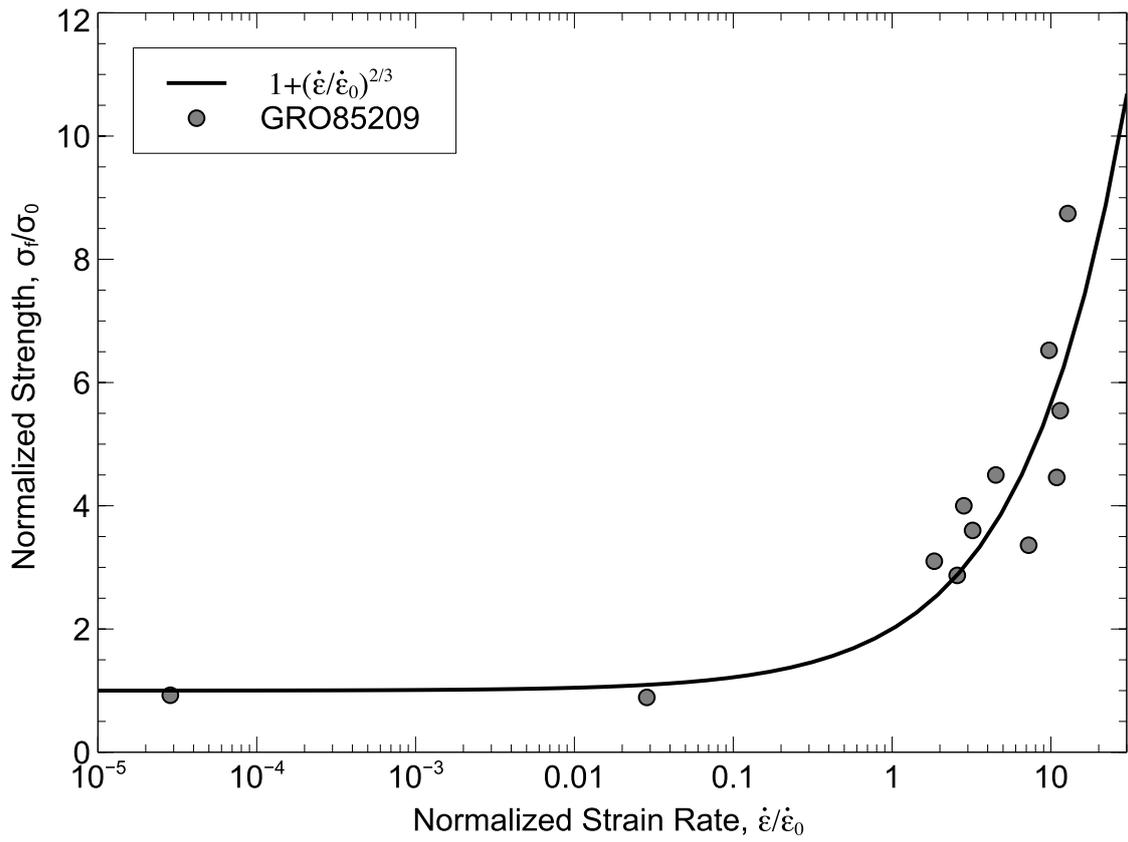


Fig. 7: Normalized indirect tensile strength data for GRO 85209 samples compared with the strength model of Kimberley et al. (2013).

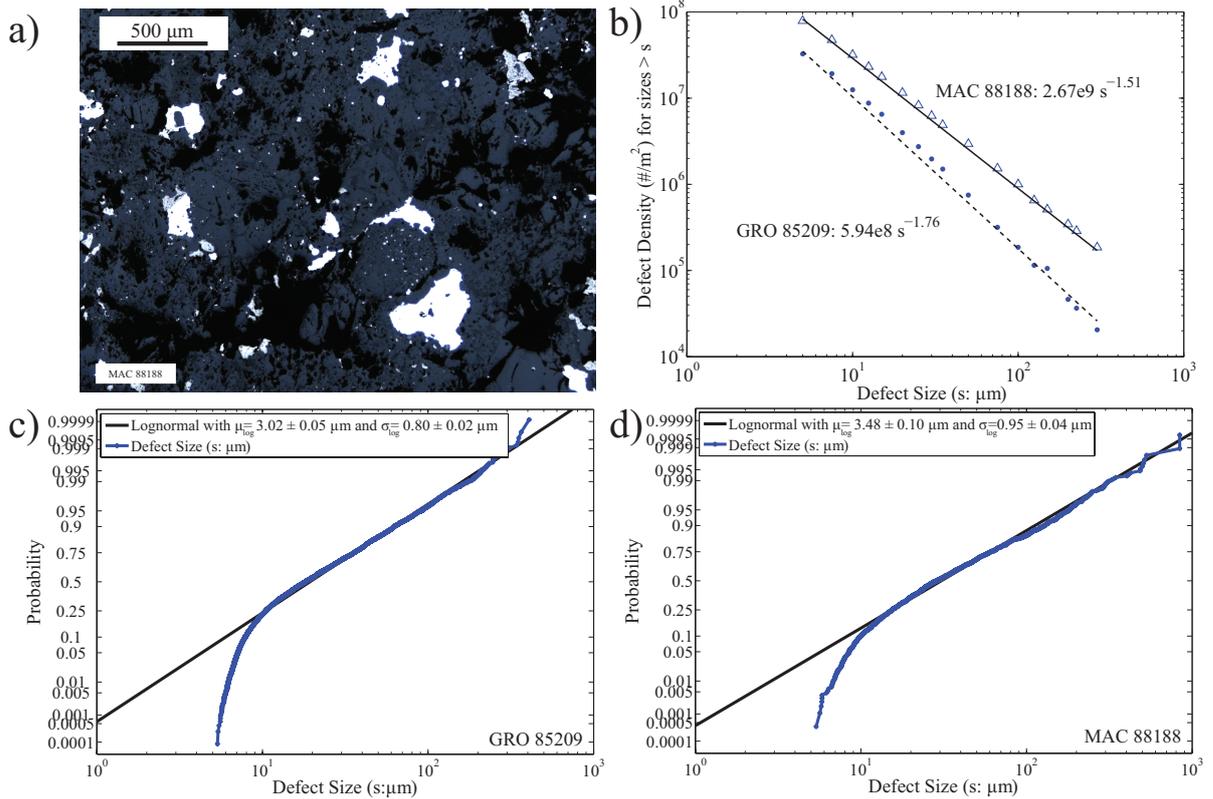


Fig. 8: (a) Optical microscope image of MAC 88118 microstructure, (b) Iron-nickel defect areal number density ($\#/m^2$) for all defects larger than the corresponding size on the x-axis. Probability plot of defect size (s) comparing experimentally measured sizes with exponential distribution for (c) GRO 85209 and (d) MAC 88118.

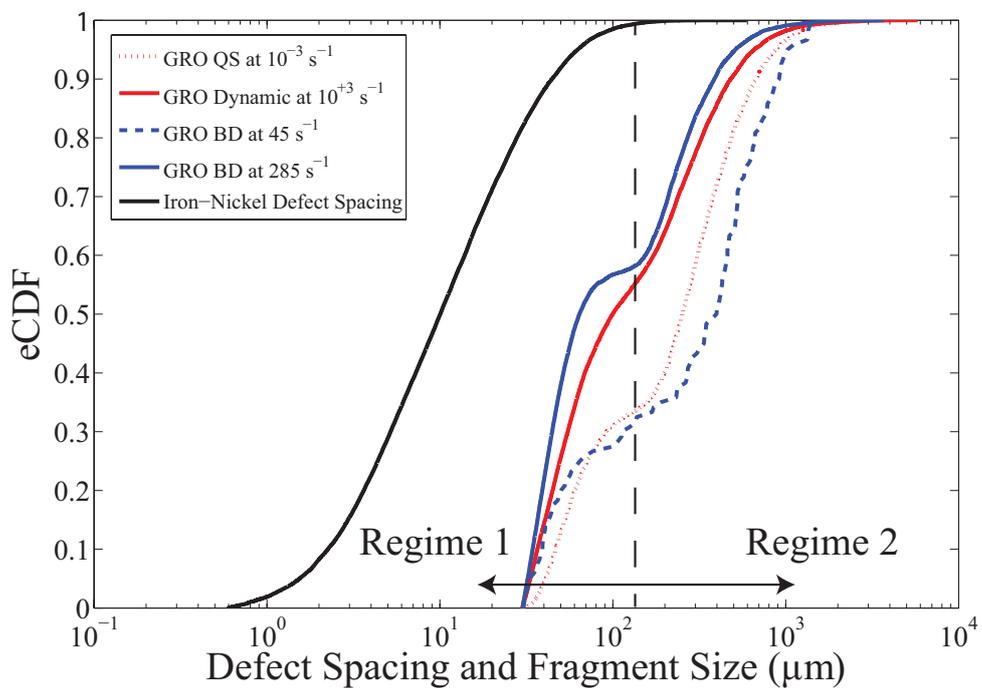


Fig. 9: GRO 85209: cumulative distribution of fragment major axis sizes and spacing between iron-nickel grains for uniaxial compression and in-direct tension using the Brazilian disk tests. Strain rates are labelled.

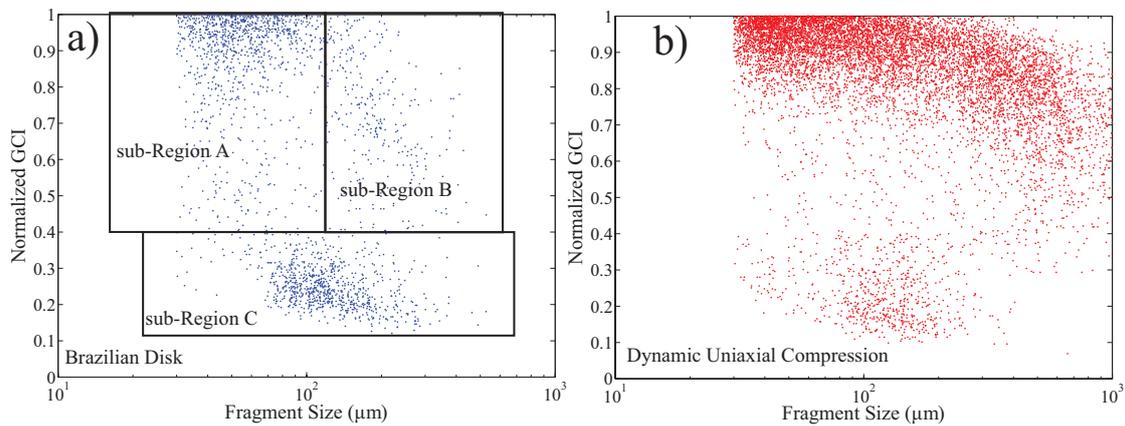


Fig. 10: GRO 85209: mean greyscale color intensity (CGI) for: (a) Brazilian Disk experiment at a strain rate of 285 s^{-1} and (b) uniaxial compression experiment at a strain rate of 10^{+3} s^{-1} .