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THE UNIVERSITY OF ALBERTA

PRELIMINARY STUDY OF THE USE OF LASER DOPPLER ANAEMOMETRY IN  
PUMP SUMP DESIGN

by

© SIDNEY J. LODEWYK

A THESIS

SUBMITTED TO THE FACULTY OF GRADUATE STUDIES AND RESEARCH  
IN PARTIAL FULFILMENT OF THE REQUIREMENTS FOR THE DEGREE  
OF MASTER OF SCIENCE

DEPARTMENT OF CIVIL ENGINEERING

EDMONTON, ALBERTA

FALL, 1986

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## ABSTRACT

The hydraulic design of pump sumps is a common and difficult task. The flow patterns in a sump are too complex for theoretical analysis, and empirical rules will not guarantee success. A hydraulic model study is required to ensure that the installed sump will perform adequately. However, even a model study is not completely reliable, as there is no firm consensus as to what criteria should be used to size the model.

This thesis first reviews the literature on the design of pump sumps. The major aspects of good and poor sump designs are discussed, and the arguments for and against the use of Reynolds and Froude criteria in modelling sumps are presented. The paper goes on to present an experimental study of the flow structure in a pump sump. The many merits of laser Doppler anaemometry in the hydraulic design of sump models are brought forward as a superior alternative to other design methods.

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### List of Symbols

- $U$  - Average velocity in the longitudinal, or x direction.
- $V$  - Average velocity in the transverse, or y direction.
- $\overline{u'^2}$  - Turbulent fluctuations of the x direction velocity.
- $\overline{v'^2}$  - Turbulent fluctuations of the y direction velocity.
- $\overline{u'v'}$  - Turbulent shear stress.
- $xsect1$  - The section in the main sump where data was taken, refer to figures 16 and 47 for locations of this and the following sections.
- $xsect2$  - The centre main sump section where data was taken.
- $xsect3$  - The third section in the main sump where data was taken, this section is nearest the entrance.
- $xback$  - A section near the back wall where data was taken.
- $xbbels$  - A section where data was taken between the bells.
- $xsside$  - The section on the south side of the bell.
- $xnside$  - The section on the north side of the bell.
- $xubels$  - The area under the north bell consisting of five sections, one through five with section one closest to the backwall.

NOTE: Where other author's equations are used, variables in the equations are defined in the text.

## 1. PART ONE: LITERATURE REVIEW

### 1.1 INTRODUCTION

Pump sumps are a necessary feature in most operations that use fluids. The hydraulic design of a pumping station involves the flow through the pump, the discharge system, as well as the channelling of the fluid into the pump (the pump sump). The trend for pumping stations in recent years has been towards larger capacity pumps. As pumps become larger, the casing wall and support structure rigidity does not increase in proportion to the size, making large facilities more susceptible to vibration. To prevent vibration problems and achieve adequate performance in large pumps, better entrance and sump conditions are required. This increased need for high performance in sumps has led to the intense study of flow problems in pump sumps. The first part of this thesis presents a summary of the work done in this area to date.

### 1.2 General Aspects of Sump Design

#### 1.2.1 Why Modelling is Important in Pump Sump Design

The complex nature of the flow patterns in pump sump limits the method of design to the use of a hydraulic model. Analytical solutions are impossible due to the complex nature of the problem, and empirical rules cannot eliminate all possible hydraulic problems. The model must reproduce

any flow separation, air entrainment, and development of vortices and eddies that occur in the prototype under scaled conditions. The model should allow for the conditions where one of a series of several pumps in a sump is shut down or operating at a lower capacity. The entrance conditions as well as the sump itself have a large influence on the overall performance and should be modelled carefully.

### 1.2.2 Applications of Pump Sumps

Pump sumps are used in too many areas to be listed here, but a few applications in the field of Civil Engineering are listed below.

*Irrigation* - In this application water is pumped from a river or reservoir into a system of pipes or canals. The water is usually untreated.

*Water Supply* - This is similar to the Irrigation application except that the water may or may not be treated.

*Storm Water* - This involves the pumping of untreated water from a culvert or sump into another culvert.

*Sewage* - This involves the transferring of water to and from sewage works.

*Dewatering* - This is important in ship drydocks, farming and navigation locks.

*Cooling Water* - This is very important in both Nuclear and Thermoelectric power generation, and involves the pumping of water into the cooling system of the plant.

The importance of reliable supplies of cooling water in the Nuclear power industry is a large factor behind the recent interest in sump design. Because nuclear power plants cannot shut down quickly, the need for a trouble free source of cooling water is paramount.

### 1.2.3 Hydraulic Problems in Pump Sumps

Following Tullis (1979), the problems associated with the operation of sumps can be separated into six categories, but almost all have to do with fluid rotation in the sump itself. This fluid rotation may result from unsymmetrical sump layouts or sharp corners that cause separation of the flow. Solutions to these problems will be discussed in the third section.

1. *Surface Vortices* - When severe enough, these may draw air from the surface into the pump causing imbalanced impeller loading, vibration and cavitation. The strength of a vortex is often rated as shown in figure 3.
2. *Subsurface Vortices* - These emanate from the floor, side walls or back walls and enter the pump causing vibration and cavitation.
3. *Prerotation of flow entering Intake* - This will change the angle of attack on the impeller blades and reduce efficiency.
4. *Uneven Distribution of Flow entering the Pump Throat* -

This will result in imbalanced loading of the impeller blades and vibration.

5. *Fluctuations In Velocities at Pump Throat* - This may cause vibration and cavitation.
6. *Separation of Flow from the Intake Lip* - This is more a problem of pump design, but can also lead to severe vibration and cavitation.

The above list covers the basic problems in sump design. A more detailed list is given by Prosser, 1977. These problems can be very serious, and correcting them may entail plant shut-down.

#### 1.2.4 Hydraulic Features of Pump Sumps

Prosser describes a typical pump sump as consisting of four separate zones. The first zone, the inlet, is where the water is channelled into the sump either through a pipe, or as free surface flow. There may be several inlet channels. The final entry into the sump is often through a control structure such as a weir.

The second zone, the approach flow area, contains items such as screens to remove debris, flow straighteners, and dividing walls that direct the flow into separate sumps.

— The third zone is the sump itself. The sump is generally rectangular with a flat floor, and may serve more than one pump. The intake for the pump may be suspended, or

built either partially or completely into the wall or floor. The primary purpose of the sump is to provide stored fluid for the pump, and to damp out any flow problems originating in the second zone.

The fourth zone is the portion of pipe between the intake and the pump impeller. This zone must always be full of water to ensure that no cavitation and unbalanced impeller loading occurs in the pump. If this zone is too long, a large pressure drop may exist between the pump and the intake, increasing the risk of cavitation. These zones are illustrated in figure 1.

### 1.3 Modelling flow in Pump Sumps

The question of what criteria should be used when modelling the complex flows of pump sumps can be clarified using some theory. Hecker (1984) and others have shown, from dimensional analysis or consideration of the Reynolds equations, that the formation of a vortex of a certain type (VT) is dependent on five variables. These five variables are the relative submergence ( $s/r$ ), the circulation number ( $\Gamma N$ ), the Froude number ( $F$ ), the Reynolds number ( $R$ ), and the Weber number ( $W$ ).

$$VT = f(s/r, \Gamma N, F, R, W) \text{ where,}$$

$$\Gamma N = \Gamma r / Q$$

$$F = Q / (rs) / (gs)^{0.5}$$

$$R = Q / (\nu s)$$

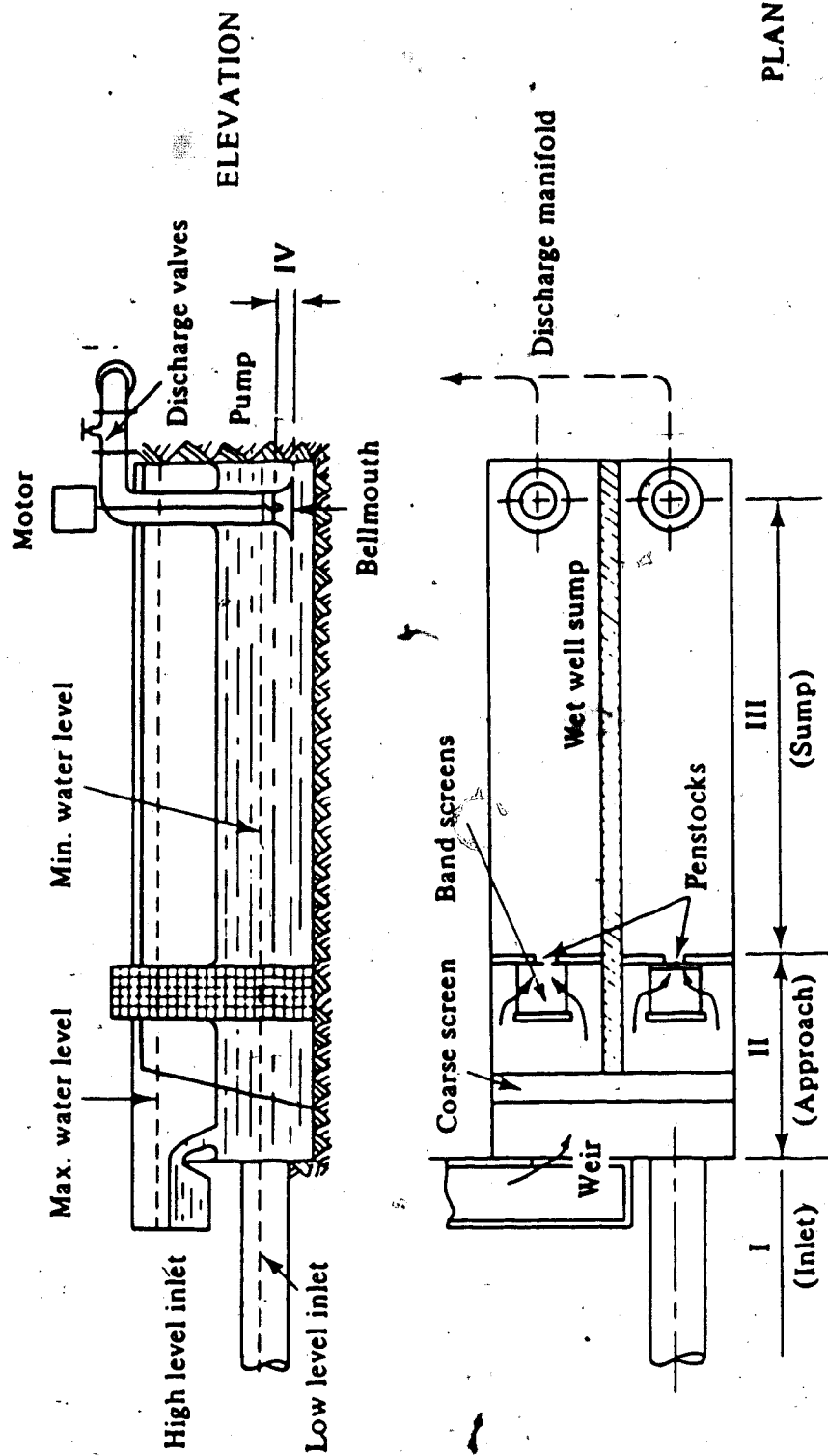


Figure 1. Definition Sketch of Pump Sump. (adapted from Prosser, 1977)



$$W = \rho u^3 d / \sigma$$

$$\text{or } W_2 = \rho s Q^2 / (A^3 \sigma)$$

$f$  = a function dependant on the geometry and roughness of the intake.

And,  $s$  = submergence of intake.

$r$  = radius of intake pipe.

$d$  = diameter of intake pipe.

$\sigma$  = surface tension.

$\Gamma$  = circulation.

$Q$  = discharge.

$\rho$  = fluid density.

$\nu$  = kinematic viscosity.

$g$  = gravitational acceleration.

$u$  = fluid velocity.

These equations are useful in that they define the important parameters, but the presence of vorticies and other flow problems cannot be predicted from them.

Dynamic similarity requires that all five of these parameters have the same value in the model and prototype. To reduce the physical scale in the model several of these parameters must be neglected, since the only way complete similarity can be achieved using water is at full scale. For free surface models, the most important requirement is the similarity of the Froude number. This is also the case in

pump sump models; but the swirling flows and air core vortices cause other parameters to be important as well. The two parameters most often neglected or 'relaxed' are the Weber number and the Reynolds number, for reasons that will be examined in the following text.

#### 1.4 Relaxing Surface Tension Similarity

The similarity of surface tension phenomena is achieved when the Weber numbers are the same in the model and prototype. There is no real consensus on how well flow is modelled when the similarity of surface tension is neglected, or on how to allow for the scale effects. Daggett and Keulegan (1974) used different fluids to investigate the effects of surface tension on model size. They found that surface tension did not affect scaled vortex flow significantly, and that the scaled critical intake submergence was modeled correctly. Other researchers will not go this far in neglecting surface tension.

Guidelines are given by some researchers on how to take scale effects into account. Anwar (1965) states that while the air core size is not modeled correctly with the Froude criteria, the prototype vortex size may be calculated from the correctly scaled circulation. He later found (1977) that as long the Weber number ( $W_2$ ) was  $1 \times 10^4$  or larger there were no surface tension scale effects. Jain et al (1978) found that as long as the Weber number  $W$  was greater than 120 no scale effects were present. McCorquodale (1968) also

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found that vortex sizes were not modeled correctly by Froude criteria and that the relative size of the air core was smaller in small models due to the stronger surface tension. He found that the scale effects became significant when the intake orifice was less than 2 inches in diameter, but were negligible when the orifice was larger.

Odgaard (1984) is another proponent of limiting the values of certain parameters to justify neglecting surface tension scale effects. He developed two equations of surface tension parameters from theory. When these 'surface tension factors'  $S_1$  and  $S_2$  are below a limiting value, surface tension effects become negligible.

$$S_1 = 0.335\sigma(g/(cH^2 \nu^2))^{0.2} / (\gamma H)$$

$$S_2 = 147\sigma(cH^2 \nu^2/g)^{0.2} / (\rho \Gamma^2)$$

where,  $\sigma$  = surface tension.

$\nu$  = kinematic viscosity.

$H$  = air core depth.

$c$  = shear drag coefficient.

$\Gamma$  = circulation.

$\gamma$  = unit weight.

$\rho$  = fluid density.

$g$  = acceleration due to gravity.

Yildirim and Jain (1981) indicate that surface tension effects may be more important than is generally thought.

They state that the effects of surface tension increase with decreasing depths and decreasing circulation. Hecker (1981) found that in Froude models of a fairly large size surface tension effects were significant for deep air core vortices.

To define a general guide from the above research is not easy since enough is not yet known about the problem. It seems that if a model is operated at a Weber number  $W$  larger than 120 and is large enough so that the intake orifice is larger than 2 inches, the model will be free from surface tension scale effects except for inside the vortex core.

#### 1.4.1 Reynolds Criteria Relaxation

The similarity of viscous phenomena in fluid flows is governed by the Reynolds number. Some researchers believe that when the Reynolds number is larger than a certain limiting value, there are no scale effects in Froude scaled flows. This limiting Reynolds number is approximately  $3 \times 10^4$ . Daggett and Keulegan (1974) found this number to be the limit for the pipe Reynolds number. Anwar (1978) found this number to hold for the radial Reynolds number, and Padmanabhan and Hecker (1984) found this number to be the limit for the approach flow Reynolds number as well. Following this idea, as long as the model is operated above this limiting Reynolds number, the actual value of the Reynolds numbers do not have to match in model and prototype.

The Reynolds number is the ratio of inertial to viscous forces in fluid flows. Sweeny and Rockwell (1982) recommend the location of the maximum bell diameter as a representative location for the definition of the Reynolds number, but many definitions are used. For the study of flow in pump sumps, three major formulations of the Reynolds number have been used. These are:

Pipe Reynolds number -  $R_1 = vd/\nu$

Radial Reynolds number -  $R_2 = Q/(\nu s)$

Approach flow Reynolds number -  $R_3 = (s_3 v_3)/\nu$

where,  $d$  = diameter of pump intake pipe.

$s$  = the submergence of the intake.

$s_3$  = depth of approach flow.

$v_3$  = velocity of approach flow.

$v$  = velocity at the location of

Froude scaling is not accepted by everyone however, and many of the detractors believe in speeding up the model above Froude scaled speeds to overcome scale effects. Fraser, Denny, and Young have all proposed the use of model intake velocities equal to that of the prototype. Dhillon *et al* (1981) however, have shown that the use of this equal velocity concept can be misleading. His work shows that the use of equal intake velocities can produce highly exaggerated vortices that do not represent the prototype

situation. Babb and Mih (1984) indicate, in apparent contradiction, that increased intake velocities are desirable to limit scale effects, but should not be so large that surface waves are created since this will inhibit vortex formation. Durgin and Hecker (1978) give an approach to increasing velocities above Froude scaled values, and a technique to allow the application of the results to the prototype. Padmanabhan and Hecker (1984) conclude from their model - prototype comparisons that there is no need for higher than Froude scaled velocities. Models should be operated at Froude scaled flows or just slightly higher, if the latter researcher's work is followed.

Following the arguments of the majority of recent researchers then, a model may be said to be free of viscous scale effects if it is operated at Froude scaled flows and the Reynolds number is larger than  $3 \times 10^4$ . The speed of the flow may be increased to bring on vortices suppressed by scale effects, but should not be speeded up a large amount for general model observation and measurement.

#### 1.4.2 Modelling using the Froude Criteria

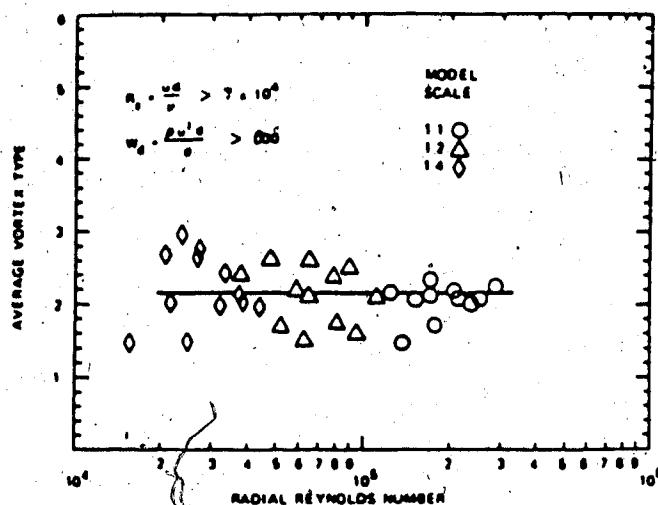
After the relaxation of the Reynolds and Weber number constraints as indicated above, the flow in a pump sump, model is a function of the Froude number, the circulation number, and the submergence only. Padmanabhan and Hecker (1984) found that Froude models correctly model the persistence of a vortex, the length of time it exists, and

the circulation in the sump. They also noted an increase in vortex type for similar Froude numbers when the flow near the entrance was partially blocked. This should be expected however, since the blocking will produce high local velocities and increased local Froude numbers. The Froude numbers used may not be comparable with those used elsewhere, depending on the factors used to define the Froude number and the type of blocking. Haindl (1956) found that Froude similarity did not hold for vortex phenomena, but he was more concerned with the core of very strong, long filament vortices.

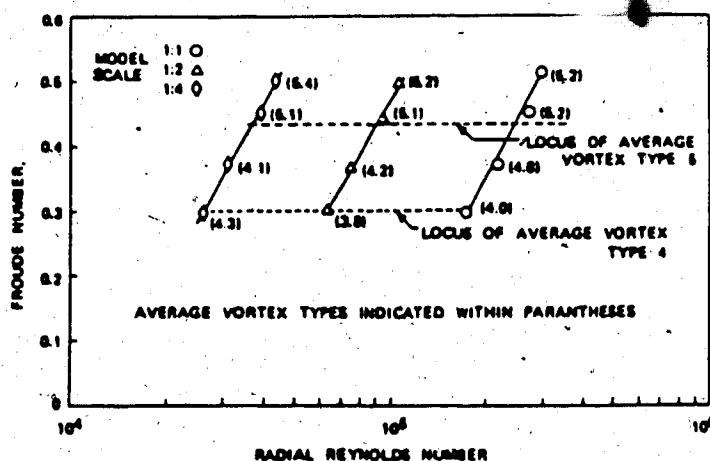
Anwar (1965) recommends a maximum scale of 1:20, and Dhillon found this to be valid as well. This limit would likely be covered by the 2 inch minimum intake diameter, except in models of very large sumps. The independence of the vortex type from the Reynold's number, and its dependance on the Froude number is shown in figure 2. Since the flow is a function of the Froude number alone, pump sumps should be scaled according to Froude criteria.

Wijdieks (1984) gives some guidelines for pump sump modelling when the Weber and Reynolds number limits cannot be met due to size considerations.

### 1.5 Factors to Consider in Pump Sump Design



Average Vortex Types for Various Radial Reynolds Numbers, Uniform Approach Flow, Both Pipes Operating



Average Vortex Types on Froude Number versus Radial Reynolds Number Plot, Data of Screen Blockage Tests

Figure 2. Variation of Vortex Type with Reynolds and Froude Numbers. (adapted from Hecker, 1984)



### 1.5.1 Evaluating Pump Sump Performance

The performance of a pump sump is most often evaluated qualitatively by the type of flow patterns that can be seen inside it. The size and strength of vortices in the sump is often the criteria for the design of a successful facility. Vortices are divided into several categories according to strength, and a limit as to the strongest allowable vortex governs sump design. An example of such a system of vortex designation is shown in figure 3. The persistence of a vortex, the percent of the time a vortex of a given strength exists, is another parameter used to identify vortex problems. Sweeny et al (1982) measured the persistence of a vortex class with a multi-button recorder, where an observer pushed a button corresponding to the vortex type present for the length of time that it was present.

Swirling flow inside the intake is often indicated by a straight vaned propeller located inside the intake pipe. Wijdieks (1984) gives a safe value for the swirl in the intake pipe using the measured rotation of such a propeller. He also measured velocity profiles in the intake and developed several statistical criteria for the required uniformity of the velocity profile. The presence or absence of vortices and swirling flows is used to evaluate performance because these are the major sources of hydraulic problems.

Measured parameters can also be used to evaluate the performance of a sump. Daggett and Keulegan (1974) plotted

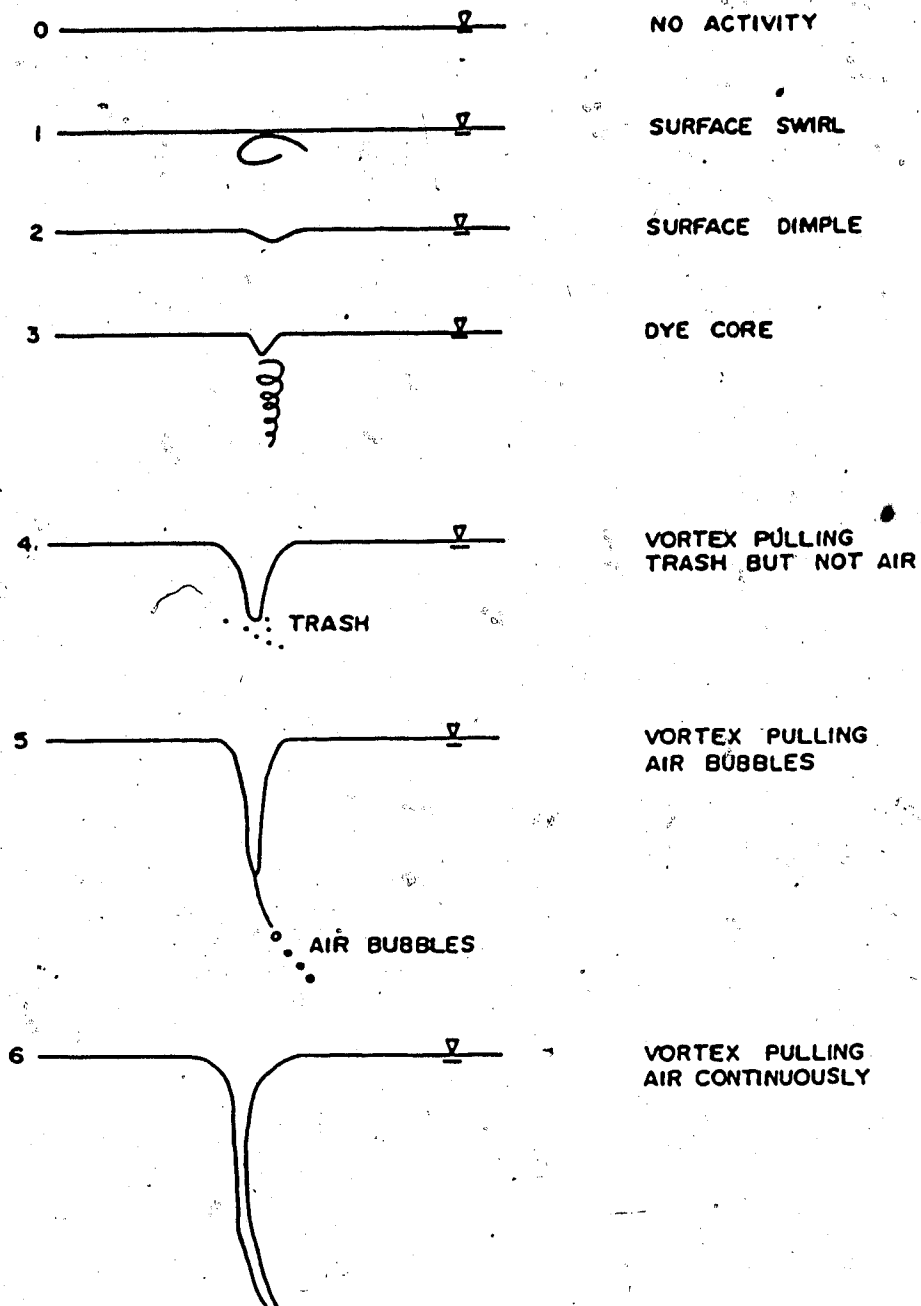


Figure 3. Vortex Classification System. (adapted from Kells et al, 1984)

the circulation number against the relative submergence and defined regions where the sump performed adequately. This method requires the measurement of the kinematic viscosity, the density, the circulation and the submergence.

Measurement of the circulation can be done by measuring the tangential velocity at a specific distance from the air core.

The critical submergence is another parameter used to describe flow in sumps. This 'critical submergence' is the intake submergence at which air begins to be drawn into the intake. This parameter was used extensively to determine the presence of scale effects and to set the limits on the Reynolds and Weber numbers. Venkataratna *et al* (1981) found that this parameter depends on the velocity in the intake in a highly varied manner. Jain *et al* (1978) found that in cylindrical tanks with a central drain on the floor, the critical submergence varied with the Froude number alone in a highly predictable manner. Apparently, the turbulence and curved vortex filaments in pump sumps lead to more varied and unstable conditions.

The above procedures can be used to evaluate the performance of both models and prototypes. To design a sump that performs well hydraulically, a model must be built. Both evaluation procedures require physical measurements and/or observations. No standard design can be used since the shape of a sump is usually dictated by the site.

### 1.5.2 Minimum Sump Volumes

The minimum volume that a sump must contain depends on the maximum number of starts of the suction pump in a certain period. The maximum number of starts depends on the motor heat generated per start-up and the maximum allowable temperature of the motor running the pump. For a fixed speed pump, the following must be taken into consideration:

1. Required discharge rate.
2. Number of pumps operating from the same sump.
3. The capacity of each pump.
4. The operating level for each pump (NPSH).
5. Minimum time between start-ups.

The required volume of the sump for a fixed speed pump can be found from the simple formula derived by Prosser (1977).

$$T = 4V/Q$$

where,  $V$  = Sump volume.

$T$  = Time between starts.

$Q$  = the pumping rate.

For a frequency of 10 starts per hour for example,  $T = 6$  min., the sump would have to be one and a half times the discharge rate  $Q$  (units of volume/ minute).

For a variable speed pump, the number of starts per hour becomes insignificant, and the minimum sump volume is governed by the control system that controls the speed of the pump.

### 1.5.3 Aspects of Good Sump Design

Prosser (1977) gives a list of general rules that should be followed in the design of pump sumps. This list is reproduced below with some minor modifications.

1. The flow approaching the pump intake, whether a horizontal or vertical bellmouth, should be uniform across the channel.
2. The kinetic energy associated with changes in level or crosssection should be dissipated well away from the final approach to the pumps.
3. Obstructions, such as supporting pillars, should be streamlined to prevent flow separation and swirl generation near the intake. This does not hold for baffles that are intended to increase turbulence to allow fast expansion of flow from pipe into sump.
4. Areas where stagnant water may occur should be filled in.
5. Average velocities should be kept low: about 0.6m/s for flow entering the sump, and 0.3m/s maximum for the approach flow to the bellmouth.
6. Trash racks should serve a double purpose as flow straightening devices whenever possible.

Figure 5 shows examples of good sump design, and figure 4

shows a typical bellmouth detail designed for good, non-separating, flow into intakes.

#### 1.5.4 Aspects of Poor Sump Design

The object of a pump sump is to provide storage and good flow conditions to the pump. A summary of unwanted hydraulic conditions in sumps are listed below.

**Jets** - Jets cause swirl as they expand, and if the jet impinges on walls or support columns, large unsteady wakes form adding to the swirl in the sump.

**Separated Flow** - Regions of separated flow are sources of vortices and unnecessary sump swirl.

**High velocity Flows** - These cause problems similar to jets.

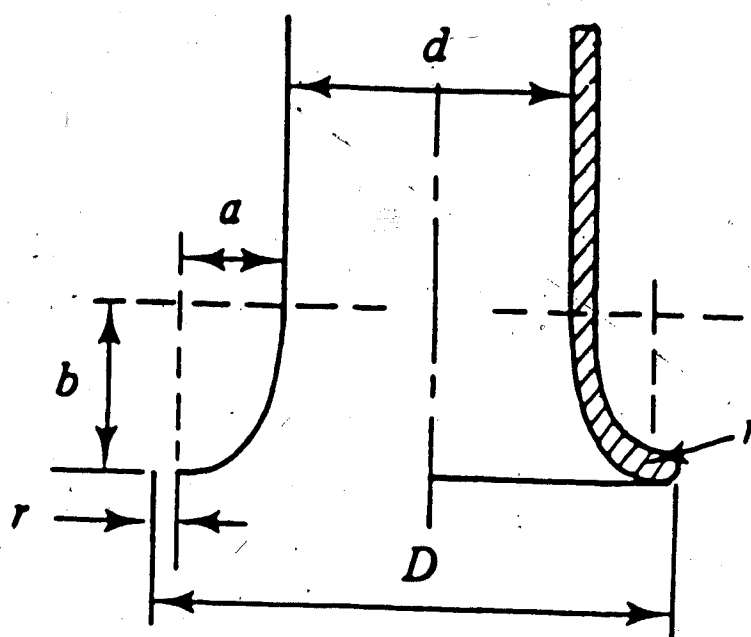
**Unsteady Flow** - This is a cause of unsteady pump loading and vibration.

**Surface Waves** - Large surface waves are a sign that either the velocities are too high, or the flow is unsteady.

**Free-Falling Fluid** - This causes aeration that reduces pump efficiency.

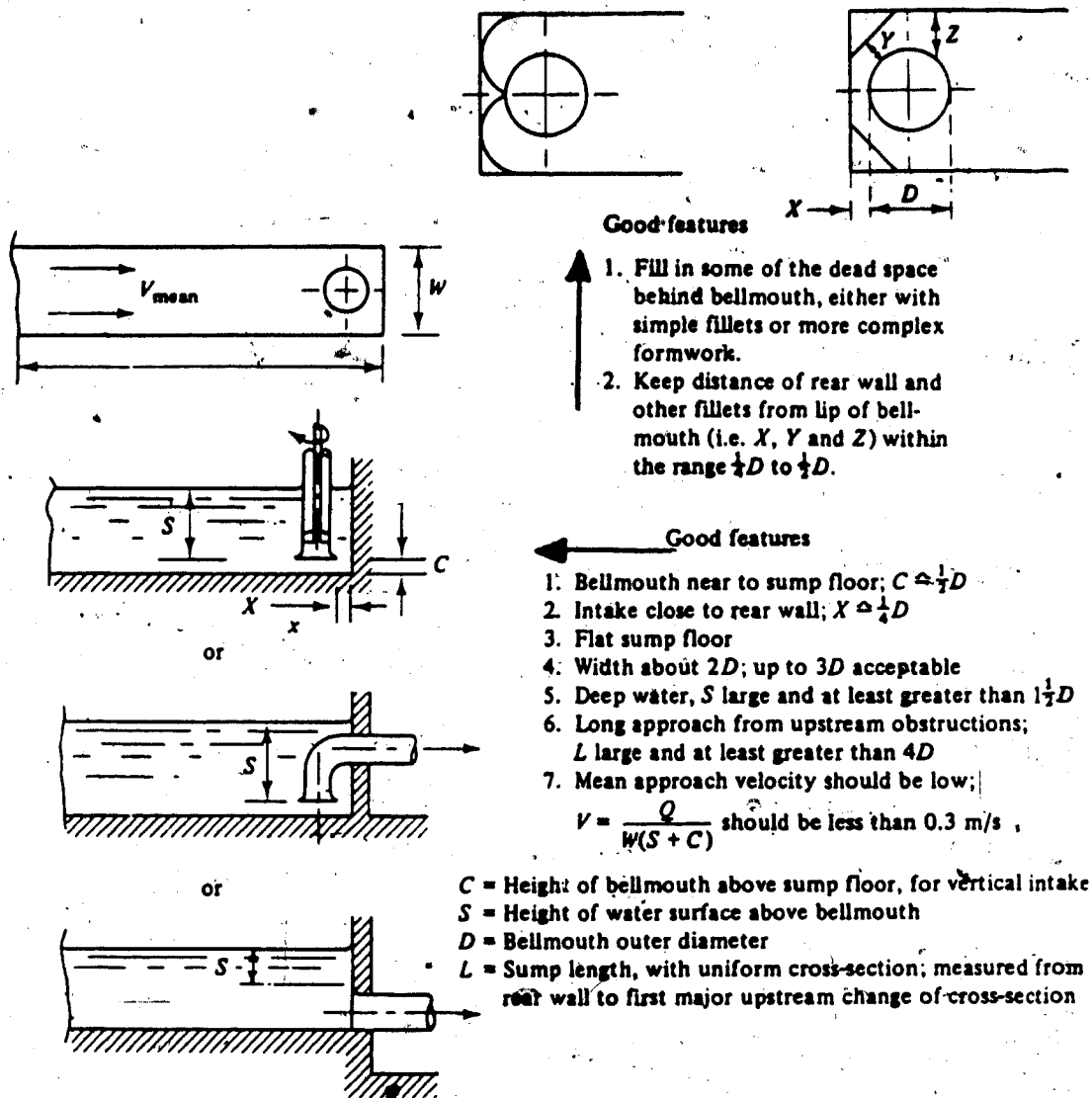
These poor hydraulic conditions are the result of poor sump configurations. A general list of poor design practices is given below, and illustrated in figure 6.

1. Undersized penstocks, control gates or valves - These will cause jets and/or high velocity flows in the sump.
2. Sharp corners - These are areas of vortex generation.



Typical bellmouth design  
 based on quarter ellipse  
 (major axis =  $b$   
 minor axis =  $a$ )  
 Bellmouth diameter  
 $D = d + 2a + 2r$

Figure 4. Typical Intake Bellmouth Detail. (adapted from  
 Prosser, 1977)



*The main aspects of a good single cell sump*

The clearance of the bellmouth from the floor should be in the range  $\frac{1}{4}D$  to  $\frac{1}{2}D$ . If  $C$  is less than  $\frac{1}{4}D$ , the flow area under the lip of the bell is less than the flow area into the bell and the resulting deceleration causes unsteady flow in the bellmouth. If  $C$  is greater than  $D$  there is a tendency for the upward component of flow into the bellmouth to become unstable and promote swirling flow. The proximity of the end and side walls to the bellmouth inhibits the production of swirling flow and vortex formation.

Figure 5. Good Sump Designs. (adapted from Prosser, 1977)



3. Rapidly diverging passages - This leads to flow separation and may cause surging in the sump.
4. Steep slopes - This may produce aeration through hydraulic jumps and may cause unsteady flow in the sump.
5. Weirs with no energy dissipating devices - This causes aeration and surging.
6. Blunt pillars, piers or guide vanes - Cause unnecessary amounts of swirl in the sump due to large wakes.
7. Asymmetric Designs - Although they can be made to work, they often increase swirl in the sump.
8. Insufficient submergence - This leads to intense vortex activity, and ingestion of air into the pump.

#### 1.5.5 Sources of Vorticity

De Siervi et al (1982) have proposed two separate origins for the vorticity in pump sumps. The first source of the vorticity in sumps is the intensification of the ambient vorticity, generated far upstream. Three dimensional drift of vertical vortex filaments into the inlet results in a strong concentration of these filaments around the stagnation streamline near the inlet. This phenomena does not depend on the viscous properties of the fluid, and may actually be calculated for simple sumps with the potential flow equations.

The second source of vorticity is the variation of circulation around the length of the inlet cylinder that

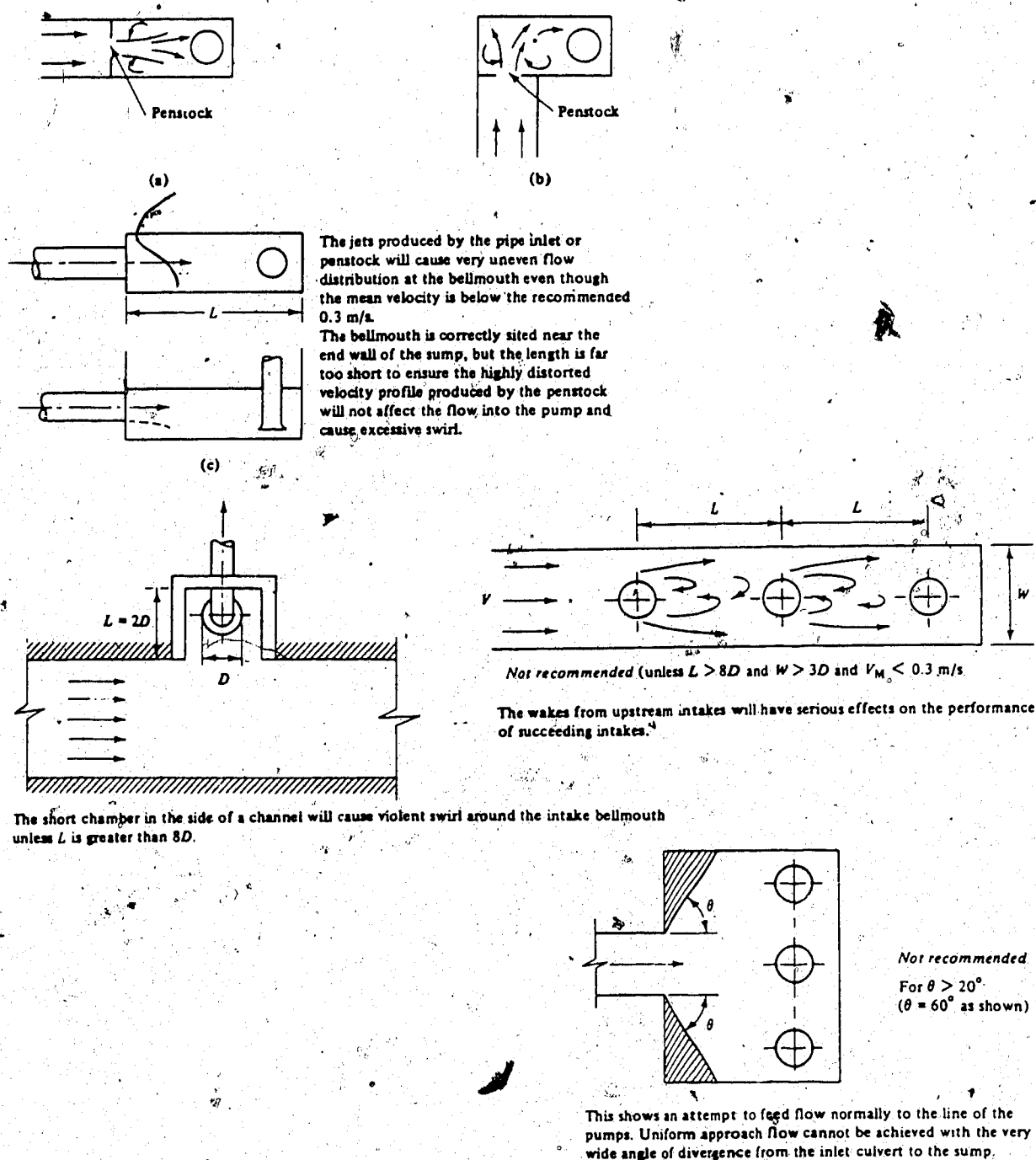


Figure 6. Poor Sump Designs. (adapted from Prosser, 1977)

goes along with vortex shedding. Viscous effects are important here since the circulation distribution around the inlet depends on the separation of the boundary layers on the outer surface of the inlet. Calculation of this flow is much more complicated, and requires knowledge of the lines of separation and the pressure field on the cylinder.

At zero degrees of yaw, when the flow is directly into the mouth of the inlet, the inlet vortex is created by the intensification of the ambient vorticity. At ninety degrees of yaw, when the flow is at right angles to the inlet mouth, boundary layer phenomena create the inlet vortex.

#### 1.5.6 Improving Poor Sump Designs

There are a number of methods available for improving the flow in pump sumps, or allowing the use of shorter sumps. If a sump is long enough, flow disturbances generated at the entrance are suppressed before they reach the intake. A sump this long would not be economical. Shorter pump sumps may also have an adequate hydraulic performance, but require special flow modifiers. Floor beams, corner fillets, backwall splitters, sump dividing walls and anti-vortex beams have all been tried at one time or another.

##### 1.5.6.1 Surface Vortex Elimination

Surface vortices are generally the most severe problems encountered in sump design. The only really effective way to eliminate these vortices is to keep the intake under enough water to prevent their

formation. Prosser (1977) recommends a minimum intake submergence of  $1.5D$  for this purpose, where  $D$  is the bellmouth diameter. One device that is effective in suppressing surface vortices is the anti-vortex beam. This beam spans the entire width of the wet pit and intersects the water surface. This beam acts as a sluice gate and directs the flow along the floor of the sump.

The effects of the anti-vortex beam are not all beneficial, however. Floor and sidewall vortices are often intensified, the probability of separation at the suction bell increases, and the prerotation of the flow in the intake pipe also increases in the presence of an anti-vortex beam.

#### 1.5.6.2 Suppression of Floor Vortices

Floor vortices are common to almost all sump installations and, according to Tullis (1979), are the most difficult to eliminate. The floor vortex is usually strong and stable with a constant direction of rotation and is concentrated near the centreline of the pump. Floor vortices are a potential source of cavitation as well as a source of prerotation of flow in the intake pipes. They should be restricted in strength, if not completely eliminated.

One method of preventing floor vortices is to place a cone under the suction bell. This method has its drawbacks however, as it often increases the

strength of wall vortices. The floor vortex may also form anyway and attach to the tip of the cone.

Floor blocks placed under the suction bell are a more effective method of eliminating floor vortices. These blocks increase the turbulence near the floor of the sump, and this turbulence breaks up the vortex. The major drawback of this method is that the increased turbulence is drawn into the pump and may cause vibration and cavitation.

Floor beams and turning vanes placed under the suction bell are the most effective preventer of floor vortices. These devices can also be useful in eliminating vortices emanating from the walls.

#### 1.5.6.3 Supression of Wall vortices

Side wall vortices can generally be eliminated by changing the distance between the walls and the intake pipe. Prosser (1977) recommends wall clearances of  $0.5D$  between the sump and the wall, where  $D$  is suction bell diameter. Backwall splitters and increased wall roughness can also eliminate wall vortices. Kells *et al* state that these backwall splitters should not extend above the water surface since they generate less intense vortices when submerged.

The use of a confined approach to the suction bell that is similar to a draft tube is effective in eliminating surface wall and floor vortices. This method is put forward by Tullis who cautions that it

has only been used on one project to date. This complicated design would also likely prove more costly than conventional designs and would still require a model test. Figure 7 shows a detail of this design.

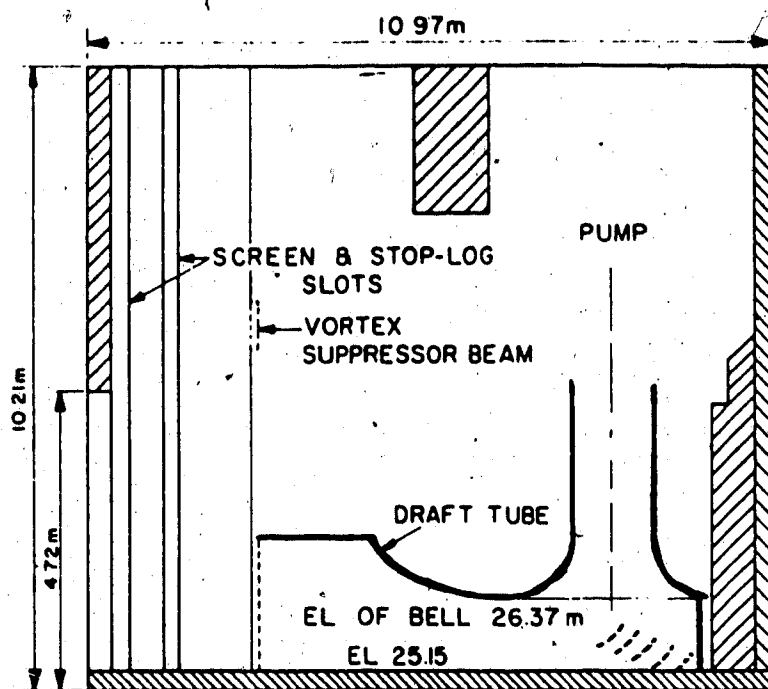
#### 1.5.6.4 Sump Dividing Walls

The use of sump dividing walls in multipump facilities is not recommended since these walls often generate swirl in the sump. If the flow from the entrance is not completely straight, the dividing walls split the flow at an angle, causing more problems than they correct. Kells *et al* (1985) found that dividing walls did not improve flow in their two pump sump, and in fact added to the vortexing problem. Figures 8 and 9 illustrate some of these flow improvement devices.

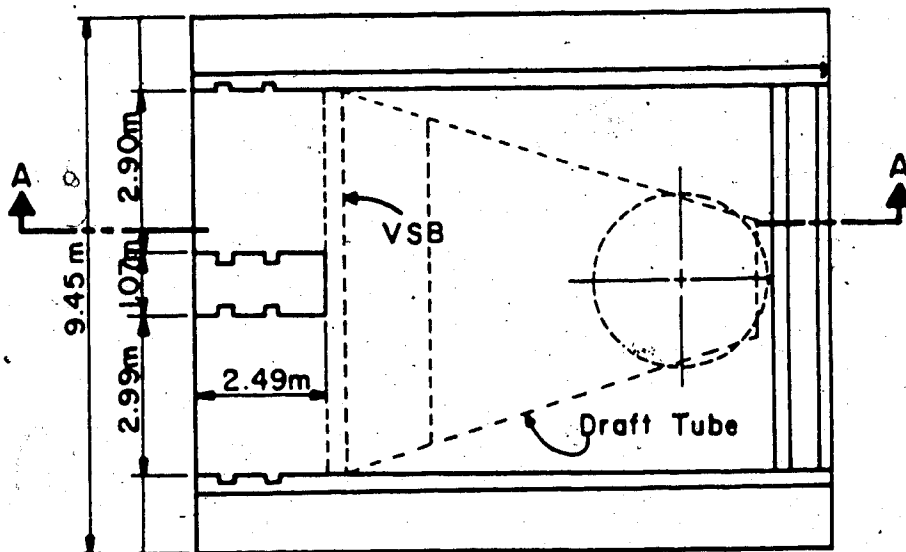
### 1.6 Conclusions

The Design of Pump sumps is a field about which little is known. Because of this lack of knowledge, even sumps that are minutely different from designs that work well must be designed using a hydraulic model. Pump sump models should be designed so that:

- The Froude numbers are the same in model and prototype.
- Intake pipes are larger than 2 inches, and the scale is larger than 1:20.
- The model Weber number ( $W$ ) is larger than 120.
- The model Reynolds number is larger than  $1 \times 10^4$ .



SECTION A-A



PLAN

Figure 7. Draft Tube Sump Intake. (adapted from Tullis, 1979)

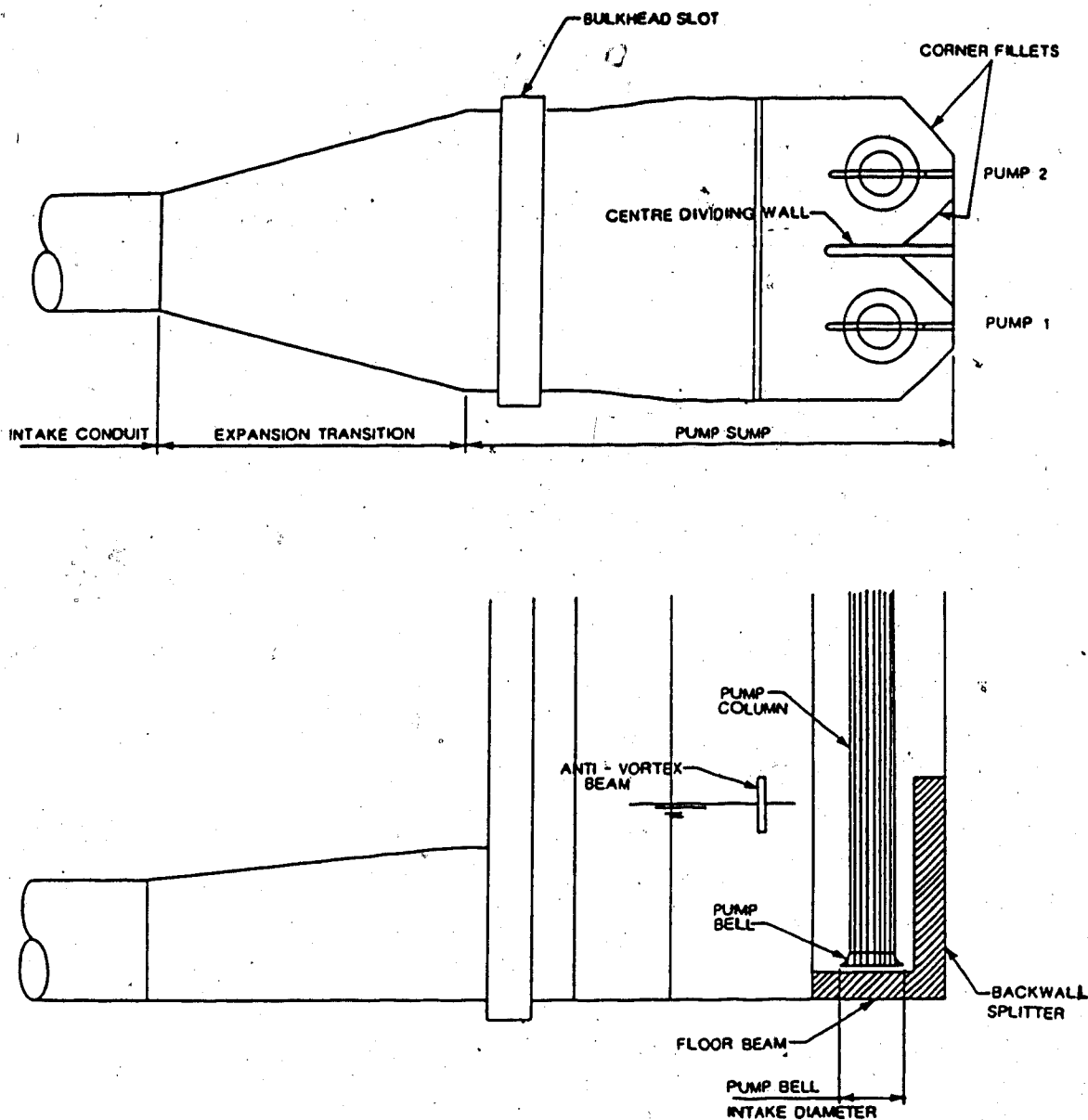


Figure 8. Devices that Improve the Hydraulic Performance of Sumps. (adapted from Kells et al, 1983)



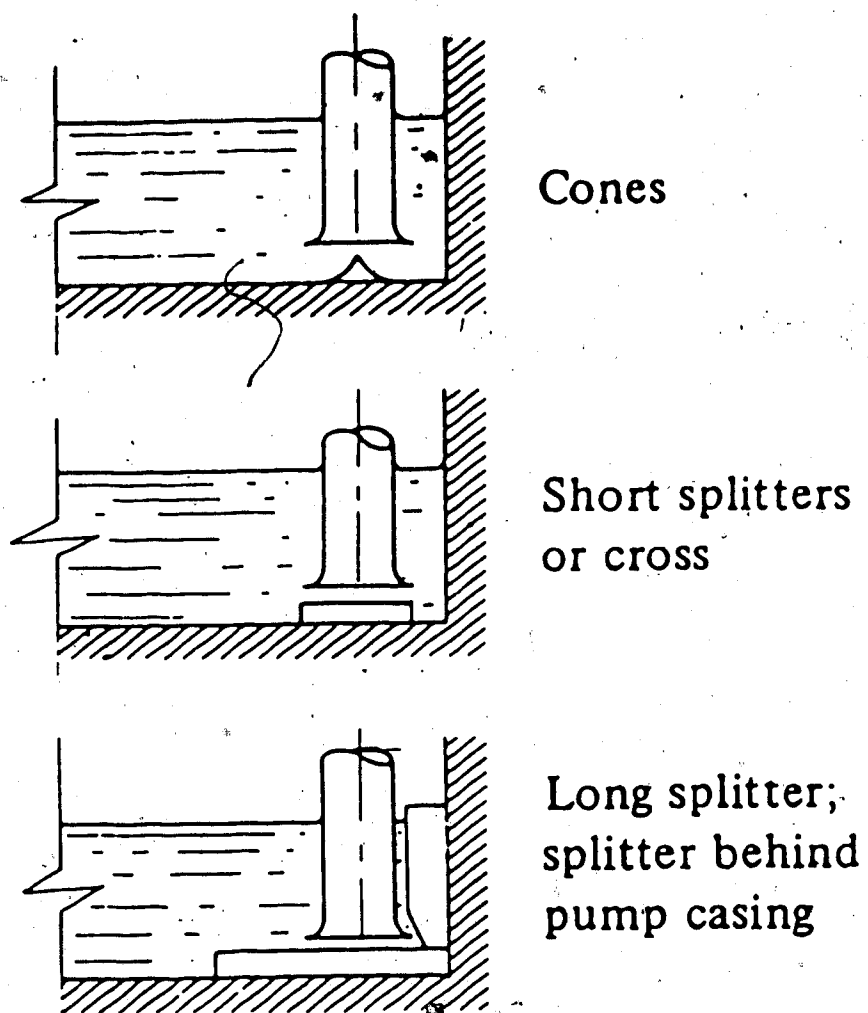


Figure 9. Flow Improvement Devices. (adapted from Prosser, 1977)

If a sump cannot be made long enough to damp out disturbances created at the entrance, flow modifying devices such as anti-vortex beams and floor beams may be used to improve the flow condtions.

## 2. PART TWO: EXPERIMENTAL STUDY

### 2.1 INTRODUCTION

Because of the lack of any detailed study of the flow structure in pump sumps, it was thought that a general study of such a flow structure would be of value. The study was made feasible by the availability of the Laser Doppler Anaemometer (LDA) in the Graduate Hydraulics laboratory of the University of Alberta, along with the donation of a sump model from Northwest Hydraulic Consultants. The sump had been used in a traditional model study for the Forty Mile Coulee project in southern Alberta. This study is described by Kells *et al.* in the Transactions of the CSCE annual conference of 1985.

Due to time considerations and the limited availability of the LDA equipment, it was decided to study the flow structure of two sump designs. Both designs were based on the sump plan recommended for construction by Northwest Hydraulics Consultants. The first design had no flow improvement devices. The second plan included the recommended flow improvement devices. The sump was operated at the minimum recommended submergence and the maximum recommended discharge for both designs. It was felt that these operating conditions would provide realistic conditions and would be most likely to reveal troublesome flow characteristics.

## 2.2 Laser Doppler Anaemometry

Laser Doppler Anaemometry (LDA) is a new method of fluid velocity measurement that uses the Doppler shift in light beam frequency to measure the velocity of small inert particles suspended in the flow. The LDA system allows measurement of instantaneous velocities without requiring the presence of physical devices in the flow, devices that often change the flow patterns. Other advantages of the LDA system are the ability to measure turbulence and the small volumes in which the velocities are measured. The LDA method requires optical access to a clear fluid containing some suspended particles. The only disadvantage to the LDA system is that the depth of penetration into the flow is governed by the focal length of the lens.

### 2.2.1 Some Physics of LDA Flow Measurement

When two coherent light beams, such as those produced by a laser, cross, an interference pattern of maxima and minima fringes is set up. The LDA determines the velocity of a particle passing through the area of beam intersection by measuring the time that the particle takes to cross a known number of interference fringes. The accuracy of a measured velocity is ultimately determined by the number of fringes in the region of beam intersection.

The LDA system depends on the presence of particles of the right size and shape to reflect enough light to analyse. The particles should be neutrally bouyant and small enough

to closely follow the flow. The number of particles present greatly affects the rate at which data is taken. Seeding the flow with particles thus reduces the time required to make velocity measurements. The addition of too many particles however, will weaken the beams and increase the time required to take data.

### 2.2.2 LDA Equipment Used in Tests

Following is a description of the equipment that comprises the LDA setup available at the T. Blench Graduate Hydraulics Laboratory at the University of Alberta. The laser used in the experiment was a four watt Innova 90 Argon-Ion laser manufactured by Coherent. By means of a beamsplitter, the single laser beam is separated into three beams: a blue beam with a wavelength of 488 nanometres, a green beam with a wavelength of 514.5 nanometres, and a blue/green mixed beam. These three beams form two perpendicular planes such that velocities can be found in two directions. Both the green and blue beams are shifted by forty megahertz with the acoustic-optic or Bragg cell.

The velocity data were measured on a planar grid using a two dimensional computer controlled traversing system. This system consists of a mirror that deflects the beams by ninety degrees and a 160mm focal length front lens through which the beams penetrate vertically into the flow. With the beams orientated vertically, the velocities on horizontal planes may be measured. The focal length of the front lens

limits the distance that the beam intersection point can move into the flow.

The light scattered off the particles is reflected back through the optics and collected by the two photomultiplier tubes. These tubes convert the the light signal into an electrical signal. This electrical signal is resolved into the two components of velocity by the two counter processors. After a check has been made to ensure that the measurement of the two orthogonal velocity components is coincidental to within 1000 microseconds, these velocities are transferred to the computer through the buffer. The optics system as well as the electronics were made by Disa Electronics, now Dantec Electronics. This LDA system is represented schematically in figure 10, and photographically in plates 1 and 2.

Twenty thousand instantaneous velocities were taken at each point in the two dimensional grid. The Masscomp computer performed a statistical analysis procedure on this data to find:

$U$  - the mean longitudinal velocity.

$V$  - the mean transverse velocity.

$u'^2$  - the mean longitudinal turbulence.

$v'^2$  - the mean transverse turbulence.

$u'v'$  - the covariance of the velocity fluctuations, or the shear stress.

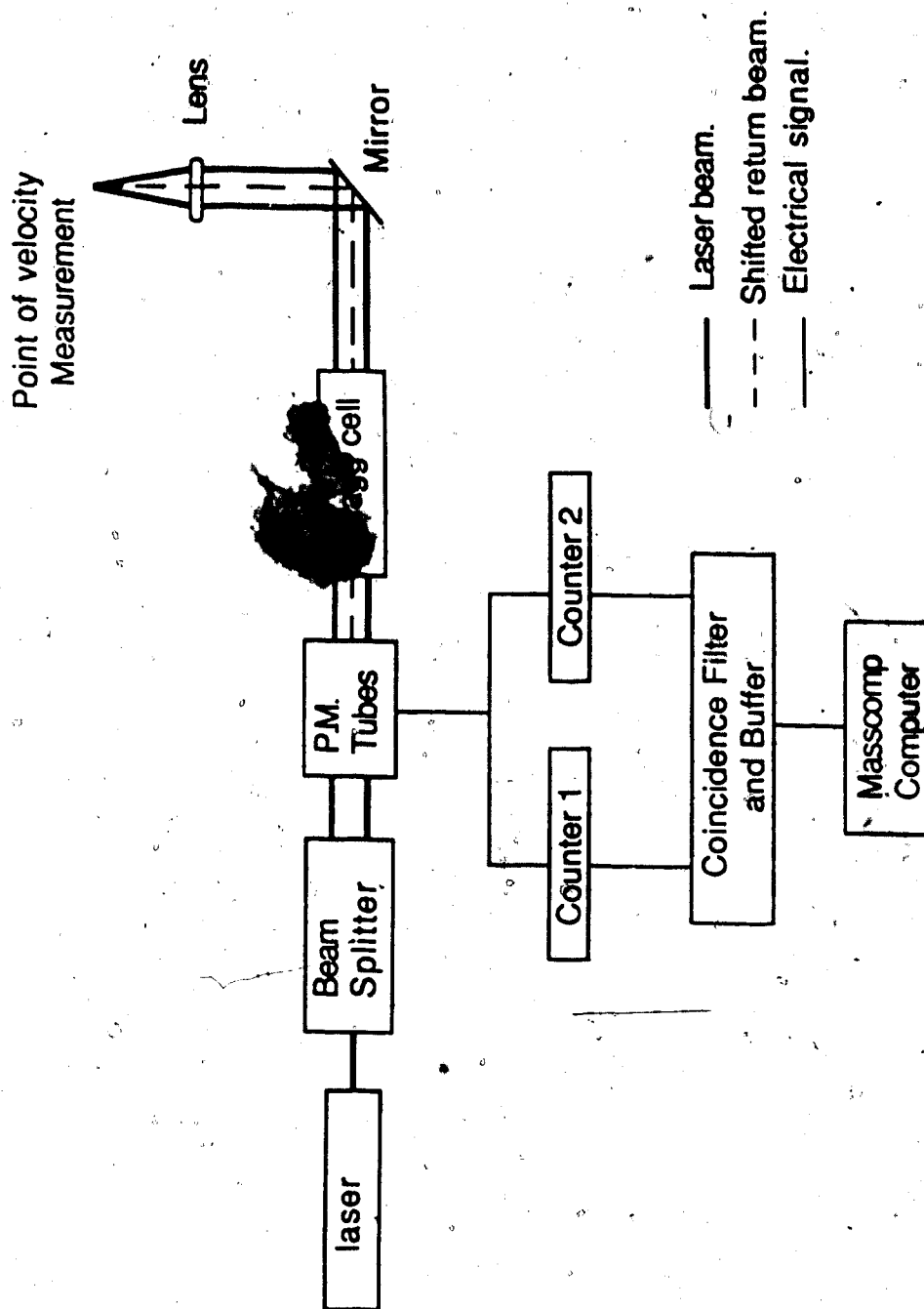


Figure 10. Schematic diagram of LDA data acquisition system.

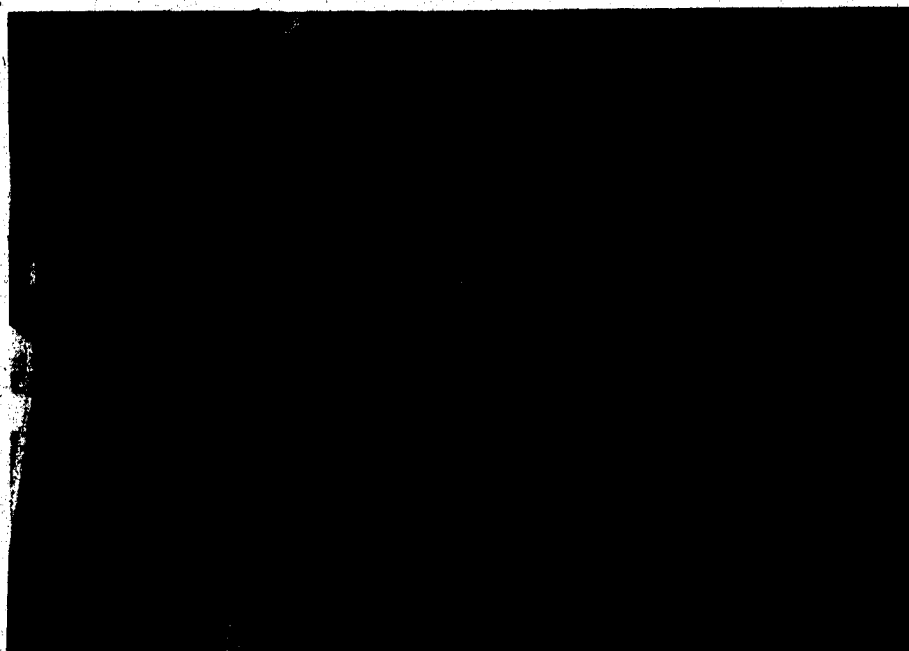


Plate 1. Laser Doppler anaemometer set up under the sump showing traverse, laser source (lower left), and front lens (top).



Plate 2. View of laser beams penetrating the floor of the sump. The location where the velocity is read is at the intersection of the beams.



### 2.3 Experimental Arrangement

The overall arrangement of the sump and related equipment is illustrated in figure 11. The water is pulled up out of the sump through two intake bells by a pair of four inch pumps. These pumps push the water into a head tank from where it flows back into the sump through a ten inch pvc pipe. The discharge in the system is measured with two magnetic flow meters located just downstream of the pumps.

The model was constructed according to the Froude criteria and has a length scale of 1:8.44. For the two designs tested, the outside walls of both the entrance and the sump were identical. Figure 12 gives the dimensions of the unimproved sump, and figure 13 shows the improved sump with the dimensions of all changes. The sump itself was suspended two metres above the floor to allow the laser to operate beneath it.

The laser beams of the LDA penetrated the bottom of the sump through a clear plate installed for this purpose. With this set up the LDA was able to measure velocities in planes parallel to the floor only, vertical velocities could not be measured.

### 2.4 Experimental Procedure

For both designs the sump was operated at one discharge, 12.56 l/s, and at one submergence, LWL(2). These conditions represent the worst case permitted in the study by Northwest Hydraulic Consultants. It was thought that

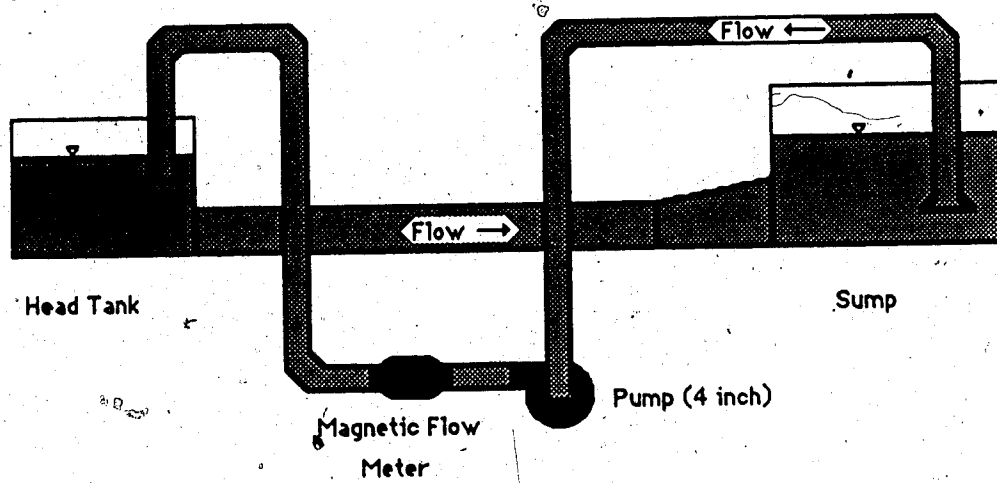


Figure 11. Overall view of the Sump set up.



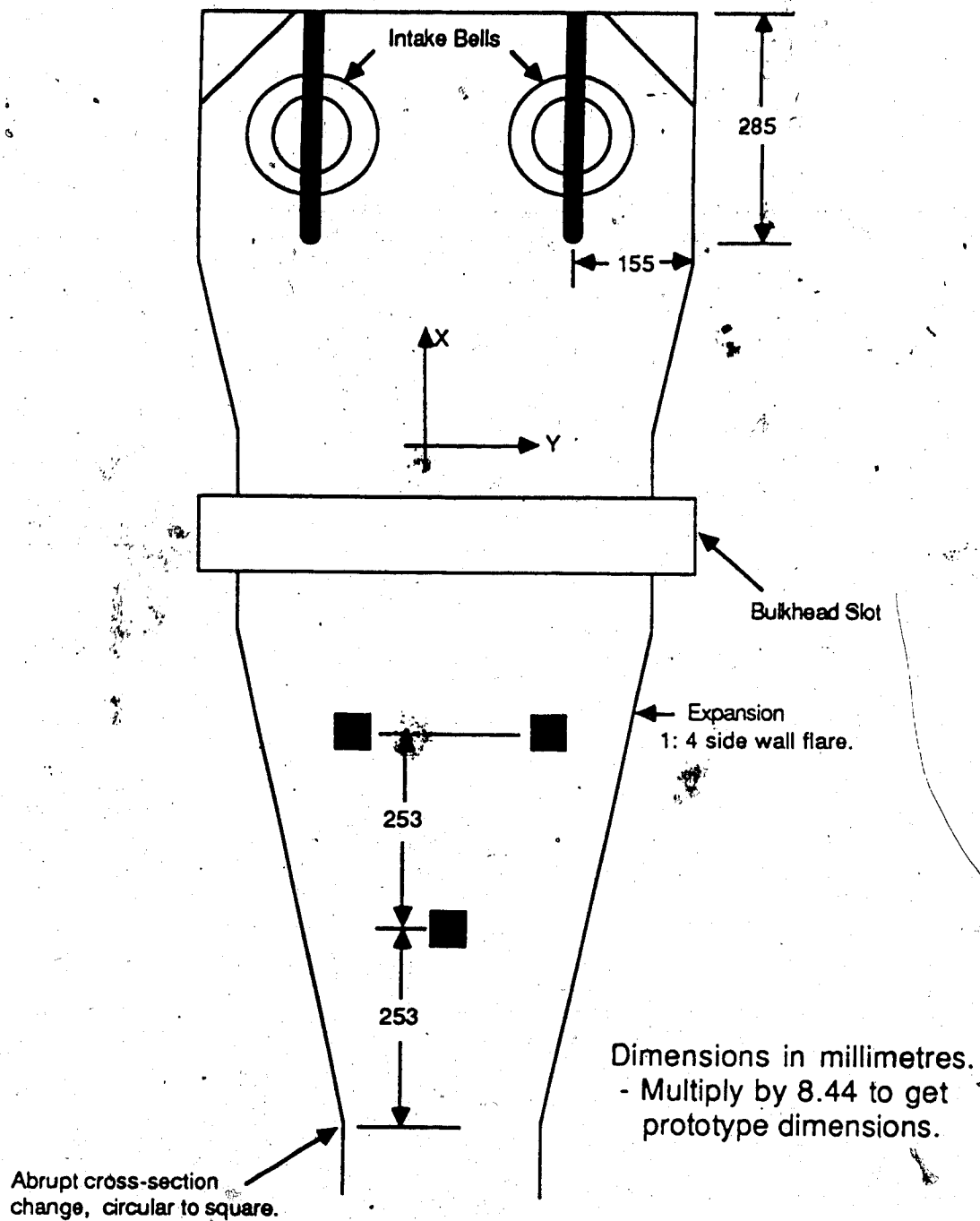


Figure 13. Plan view of improved sump showing dimensions of the improvements.

these extreme conditions would best illustrate the structure of flow in a pump sump. The submergence of the intakes is illustrated in the side view of the sump given in figure 14.

A dye test was conducted first in order to determine the general patterns of flow in the sump. Dye was injected with a needle at various locations in the sump to find the direction of flow. From these point tests, a rough map of the streamlines in the sump was constructed.

From the dye tests, important areas were identified for investigation with the LDA. For these areas, a two dimensional grid of points was programmed into the computer. The computer then moved the point of laser beam intersection to these locations and took a set of data. The laser took all the required data for one height before moving up to the next height as this required fewer adjustments to the voltage fed to the photomultiplier tubes.

#### 2.4.1 Measuring Flow Velocities in the Sump

The laser Doppler anaemometer took twenty thousand readings at each specified point in the flow and averaged them to arrive at the horizontal velocities at that point. The time required to take one set of readings varied with the depth of beam penetration into the flow and the number of particles in the flow. A complete set of data might be taken near the bed in under one minute, but twelve centimetres into the flow a data set might take four to five minutes to complete.

Good data was taken where the flow was steady, but problems may exist with data taken in unsteady regions. In the unsteady areas, an eddy might last for half a minute or so, completely changing the direction of flow for that length of time. The magnitude of the velocities will not be representative in these regions since velocities in both directions will be included. These large scale eddies will also affect the values of turbulence and shear stress. The regions of unsteady flow include under the centreline of the bells, downstream of the intake bells, and the areas of return flow in the unimproved sump.

The LDA was able to take data in the horizontal directions, U along the length of the sump, and V across the sump as shown in the figures. Data was taken at levels very near the floor and at 30, 60, 90, and 120 millimetres above the lowest level. Data was taken at these levels for all areas except directly under the sump where other increments were used.

#### 2.4.2 Percent of Flow in the Study Zone

The depth of flow that can be studied with the laser Doppler anaemometer is limited by the focal length of the front lens. For the equipment used in this study, the focal length was 160 mm. However, with the thickness of both the floor and lens, the range was limited to about 120 mm above the floor.

An estimate of the amount of flow passing through the areas where tests were conducted in the main sump was made by multiplying an average velocity on a certain cross section with the area of that section. The percentage of flow through the 120 mm test area ranged from 30 to 50 percent for sections three to one of the unimproved sump, and 40 to 60 percent for the same sections in the improved sump.

Approximately half of the flow moves through less than one third of the water depth, the area where the tests are conducted. Near the intakes all of the flow must pass through the area of measurement since the intake bells are located near the floor.

#### 2.4.3 Checks on the LDA Velocities

Two independant checks were performed to ensure that the LDA was taking correct velocities. The first was an order of magnitude check using the average velocity in the return flow pipe. This average velocity can be calculated for the known discharge as,

$$V = 2Q/A = 0.70 \text{ m/s}$$

where,

V = the average velocity in the return flow pipe.

Q = the discharge of each pump.

A = the cross sectional area of the 10 inch diameter pipe.

If the LDA measured velocities are of the same order as this average value, then they are correct measurements. The actual values of the velocity in the main sump were slightly smaller, but the sump area is greater than that of the pipe requiring smaller velocities for the same discharge.

As a further check on the validity of the LDA readings, a propeller probe was used to measure the fluid velocities in the sump. The data from this probe are included in Appendix A. The measurements taken with the propeller correspond almost exactly to the LDA measured velocities.

#### 2.4.4 Visualizing the Flow

The LDA equipment generates a large amount of data that must be rendered understandable. The data for this study is plotted as the variation of velocity with either a horizontal or vertical distance, a velocity profile. Several of these velocity profiles are drawn on one axis to show the variation of the velocity in the third direction. For example, in the main sump cross-sections the velocity at a given height is plotted against the distance across the sump, as in figure 17. To show the variation in the vertical direction, velocity profiles for different heights are plotted on the same axis. The use of three dimensional plots was considered, but the coarseness of the grid used would not allow the construction of a clear plot, and would make



the extraction of actual velocity values difficult. Local coordinates were used for each section or area where data was taken.

## 2.5 Vorticity Meter

One of the intakes was equipped with a straight-vaned propeller used to show rotation of the fluid in the intake pipes. The experience with the unimproved sump has led me to question the validity of any readings from such a device located near the mouth of the intake.

With the clear floor of the sump it was possible to closely observe the interaction of the vorticity meter and the floor vortex. When filament of the floor vortex was attached to the hub of the vorticity meter, the propeller would spin at a fast rate. However, the floor vortex would often attach to one of the vanes of the vorticity meter and stop it dead. The floor vortex could also extend past the vanes of the vorticity meter in which case the vorticity meter would also not move.

The rotation, or vorticity, of the flow entering the intake is concentrated along one line, the core of the floor vortex. Unless the hub of the vorticity meter coincides with this moving core, any data taken from it will be of little value.

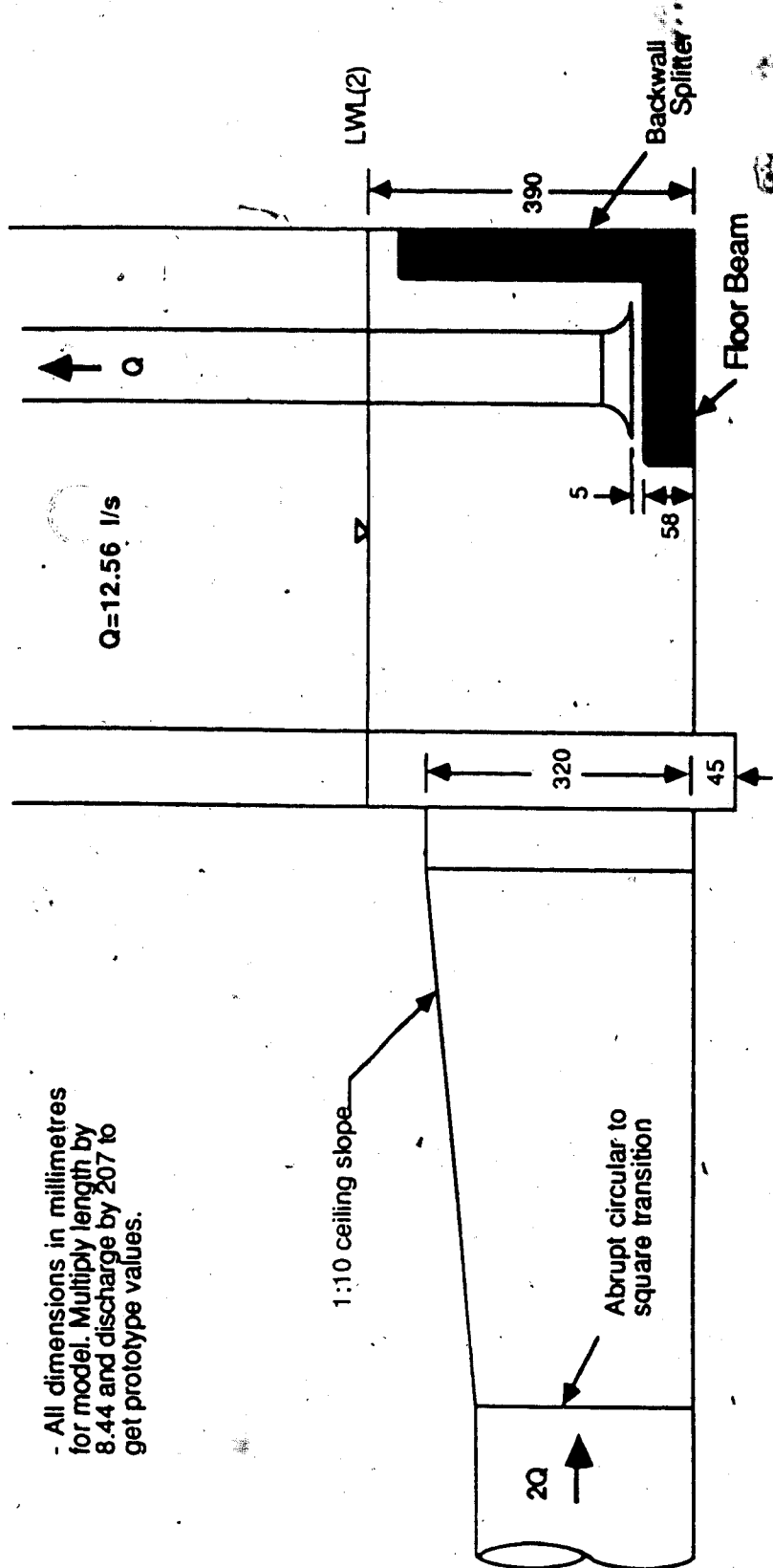
As the fluid continues up the intake pipe, the centre of rotation will slowly align itself with the centre of the pipe due to the frictional effects of the walls. Vorticity

meters will only give true readings away from the entrance, where the centre of the fluid rotation coincides with the centre of the pipe. The vorticity meter was not used in this study.

## 2.6 Structure of Flow in the Unimproved Sump

The overall flow patterns were initially observed with a dye tracer, and the results are shown in figure 15. The dimensions of the unimproved design are shown in figure 13, and the locations of the cross-sections where the LDA data was taken are shown in figure 16. For this sump configuration the flow is very asymmetric. The jet emerging from the pipe clings to one wall in the short expansion area of the sump. This initial asymmetry contributes to a large circulation pattern in the sump. The jet from the expansion region moves along the wall to one intake, and around the backwall to the other intake. After the jet passes the second intake, its momentum is dissipated in large scale eddies, some of which are shed from the intake cylinders.

The pattern of flow shown in the figure is fairly steady, but the jet does sometimes move from one side of the sump to the other, inverting the flow pattern. The jet prefers to cling to the south wall of the sump, but can be switched to the north side wall by deflecting the jet with a small board. Severe floor vortices were observed at the design flowrate and submergence, but no surface vortices stronger than class three were in evidence.



- All dimensions in millimetres for model. Multiply length by 8.44 and discharge by 207 to get prototype values.

Figure 14. Side view of sump showing operating conditions.

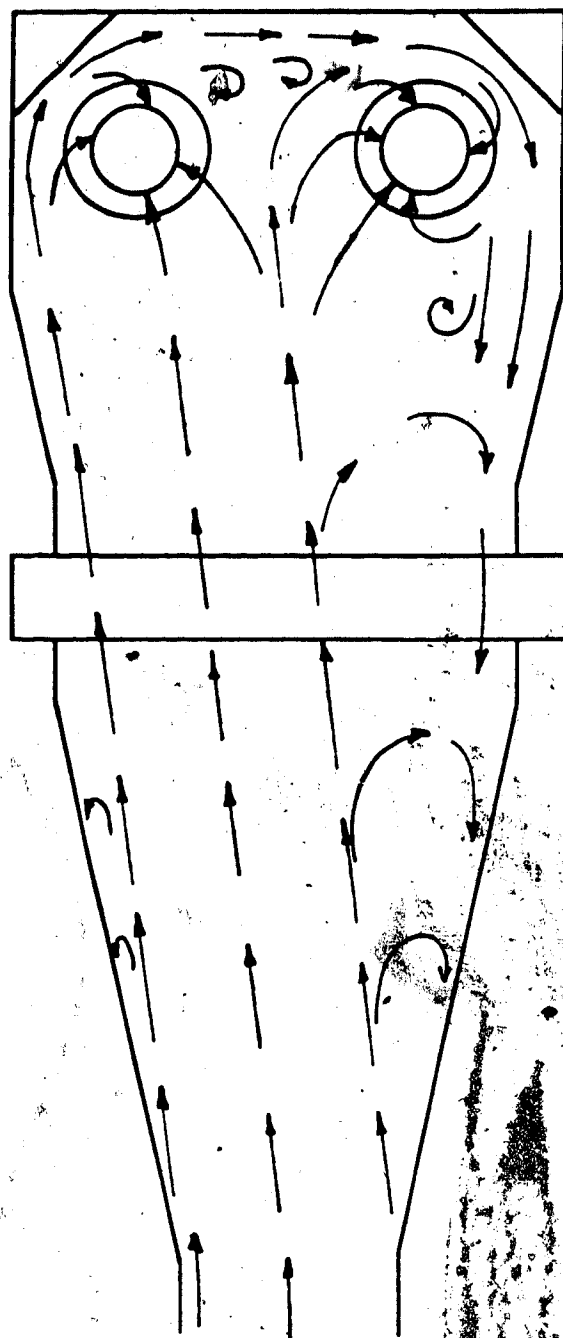


Figure 15. Patterns of flow in the unimproved sump.

Figure 16. Plan view of the glass floor section of the sump indicating the locations of the cross-sections where data was taken. These cross-sections are centred in the sump or on the bell centre, except for the sections between the intake bells and the sump wall. These two sections are centred between the bell and the wall.

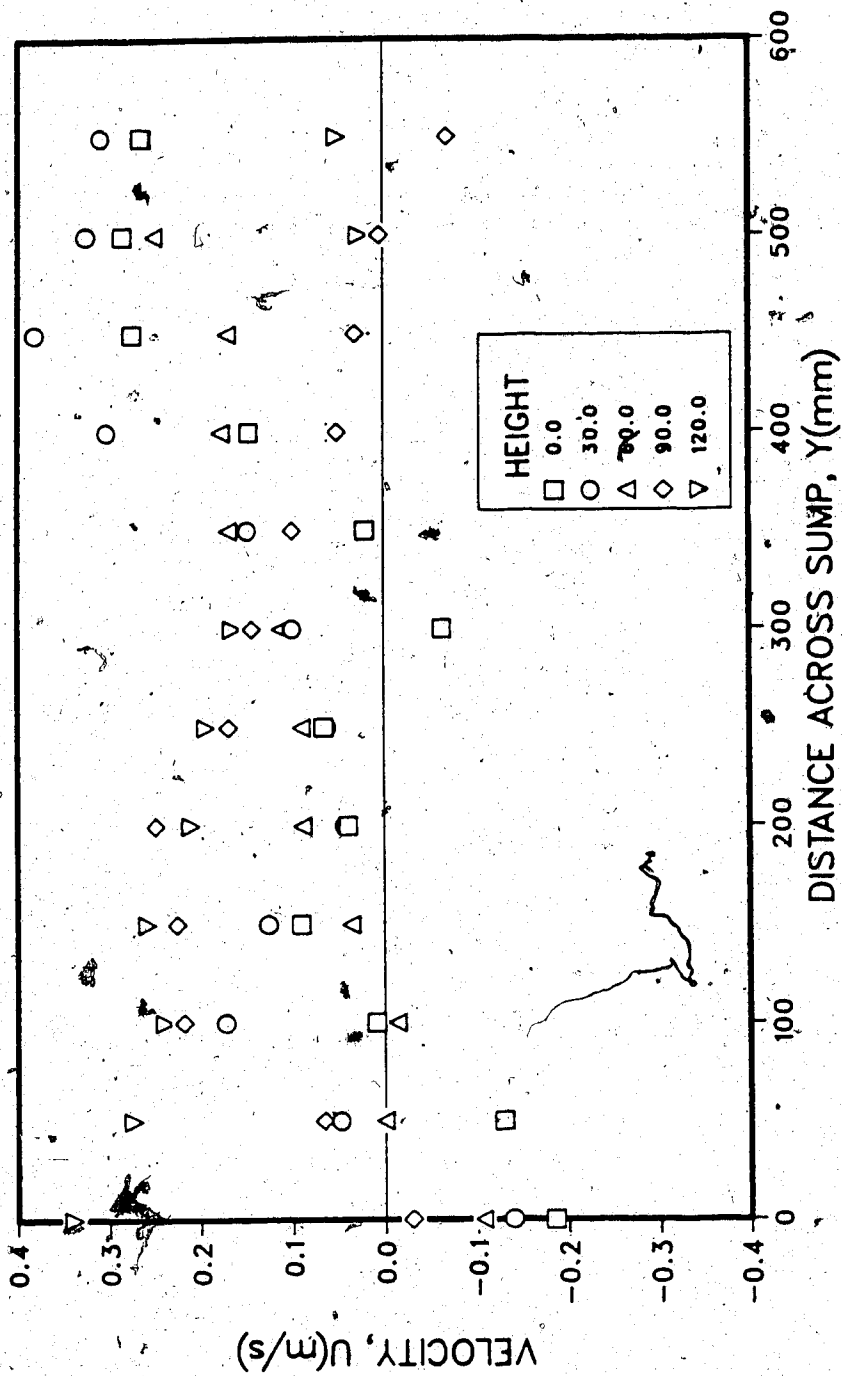


Figure 17. Longitudinal velocity profiles for the first section in the unimproved sump, xsect#1 in figure 16.

### 2.6.1 The Main Sump

The asymmetry of the flow in the sump can be seen in the velocity profiles. Figures 17 through 19 show the velocity profiles across the entire sump. The second and third cross sections show the jet from the pipe as an area of positive flow on one side of the sump, and the returning flow as an area of negative flow on the opposite side of the sump. Although this area of returning flow seems to be of a much more unsteady nature than the flow in the jet, the turbulence is no higher on this side of the sump. The strong forward flow of the jet as it passes beside the south intake pipe is shown in figure 20. The return flow of the jet is illustrated in a vertical velocity profile in figure 21.

Figure 19, on cross section 3 illustrates the unsteady nature of the jet. For this section, as the data was being taken, the jet moved from one side of the sump to the other. The data for heights very near the floor (ie. height = 0 on plots), at 30mm, and at 60mm were all taken with the jet clinging to one wall. As the LDA moved up to the next level, the jet switched to the opposite wall. This shows up as an inverted velocity profile for levels 90mm and 120mm above the floor.

Cross section number one, figure 17, is a complex mixture of velocity profiles. This section may also have been subject to an inversion as described in the above paragraph. The nearness of the intake bells can be seen to influence the velocity profiles of the heights 0mm and 30mm

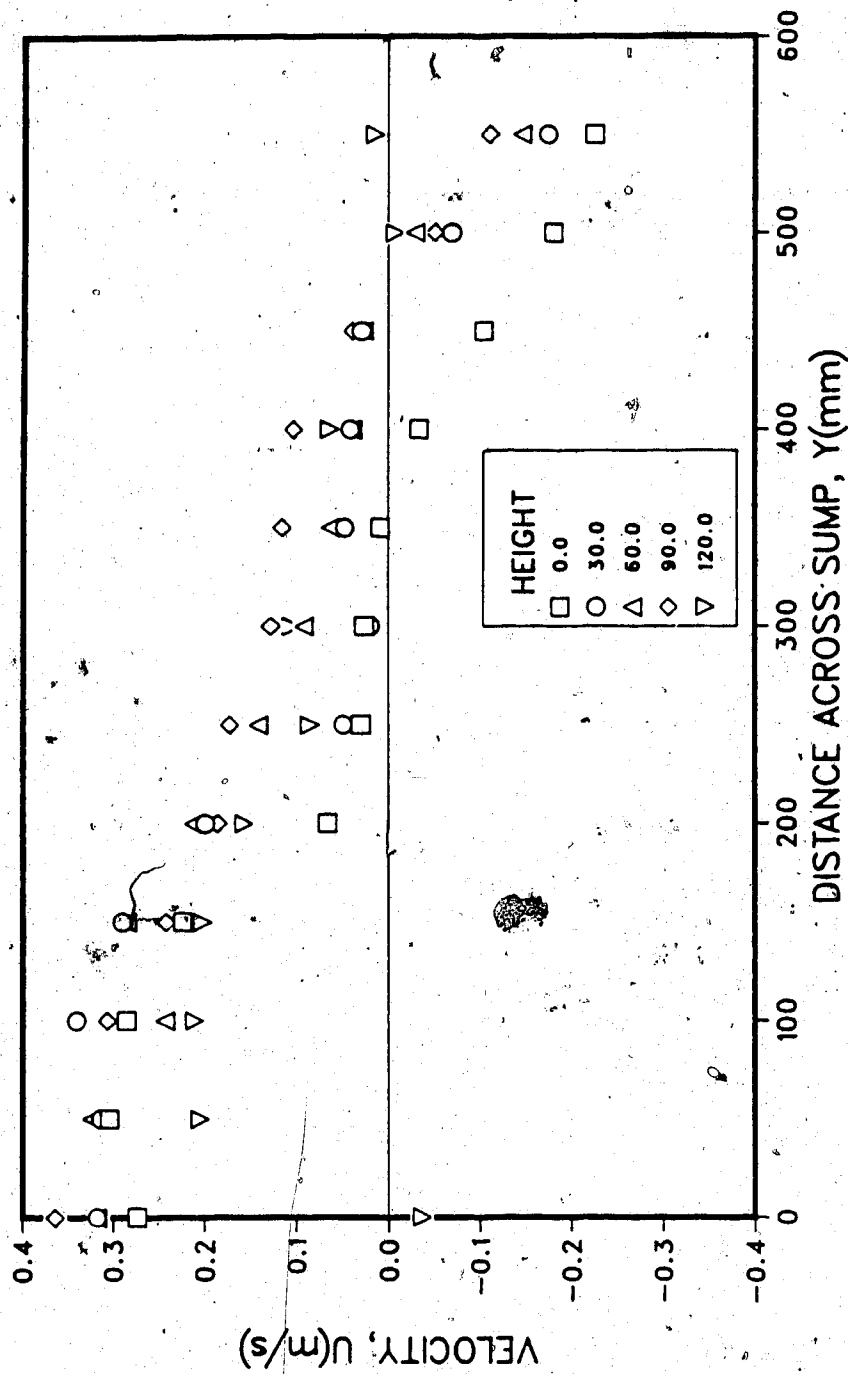


Figure 18. Longitudinal velocity profiles for the second section in the unimproved sump, xsect#2 in figure 16.



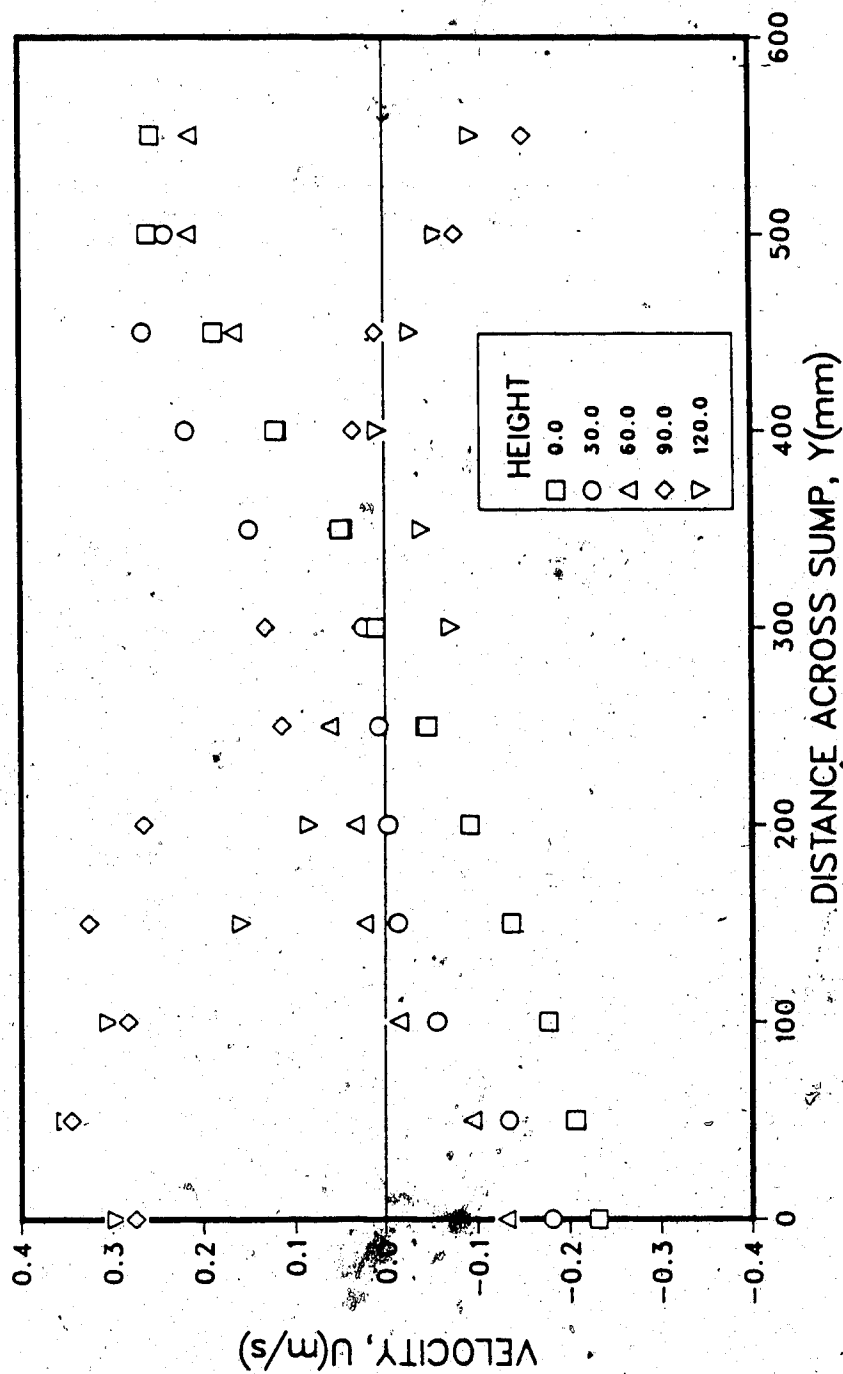


Figure 19. Longitudinal velocity profiles for the third section in the unimproved sump, xsect#3 in figure 16.

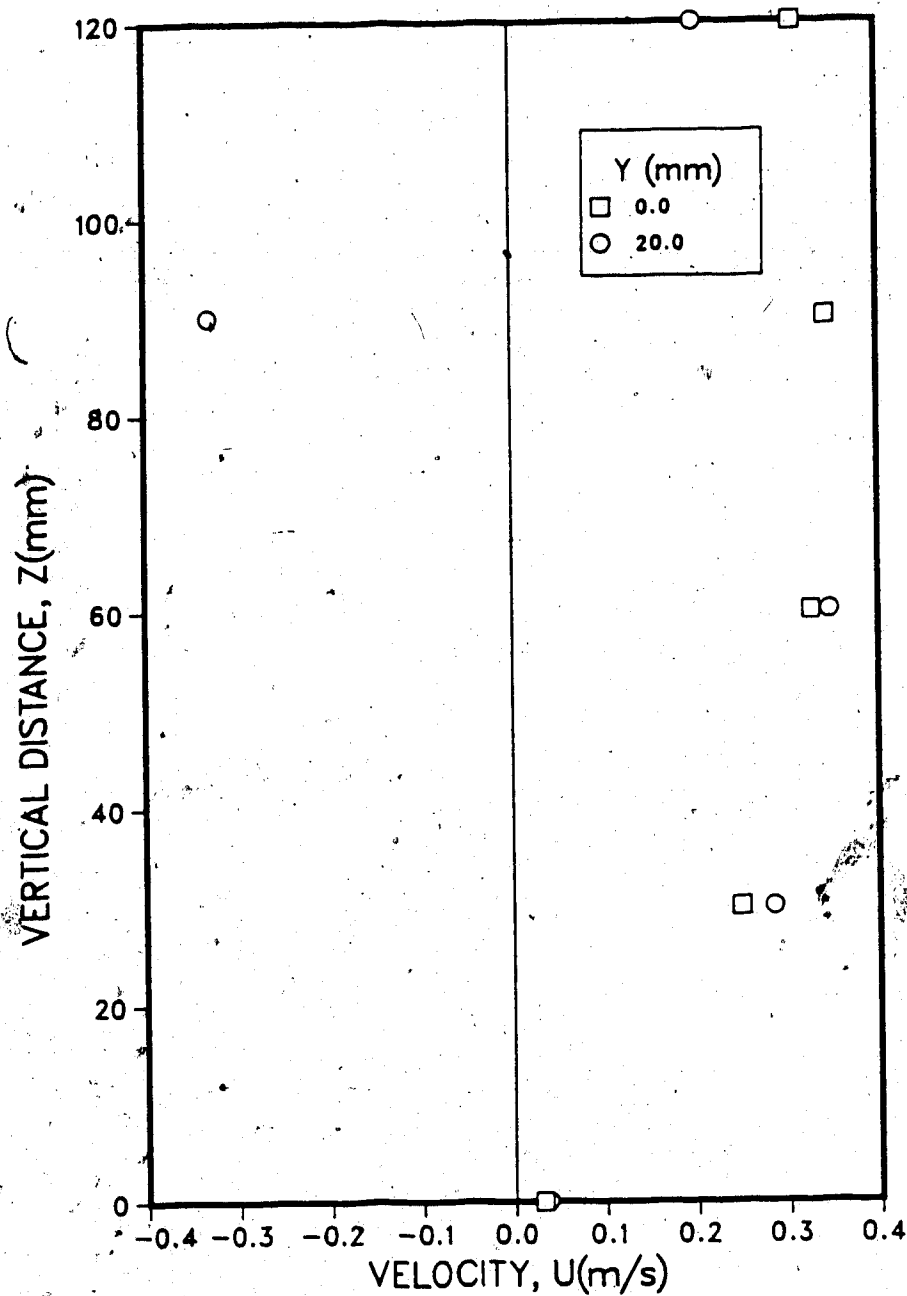


Figure 20. Longitudinal velocity profiles for the area on the south side of the south intake in the unimproved sump, xsside in figure 16.

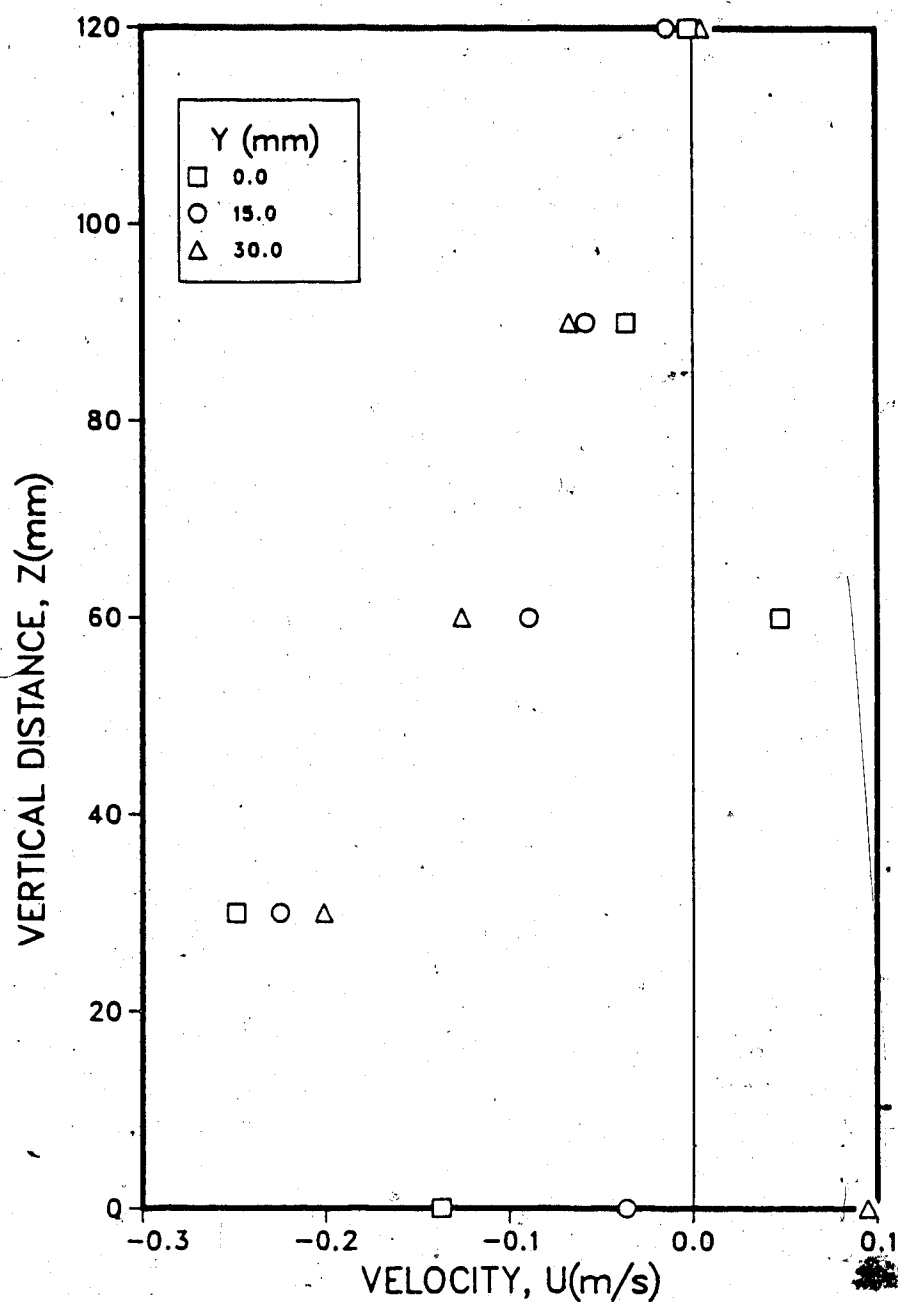


Figure 21. Longitudinal velocity profiles for the area on the north side of the north sump in the unimproved sump, inside in figure 16.

above the floor. For these two velocity profiles there are two definite humps in the velocity profile at  $x = 100\text{mm}$  and  $x = 450\text{mm}$ , areas where the fluid comes under the direct influence of the 'sinks' that are the intake bells.

The cross section between the bells, figure 22, has a positive flow in the upper regions and a negative flow in the lower areas. This flow distribution indicates the tendency for vortices joining the bells to form. The upper flows move to the backwall, descend and return along the floor forming a circulation pattern around a horizontal axis connecting the two bells. At higher flows, or smaller submergences, this pattern became so intense that a cavitation core vortex filament appeared. An example of this phenomena appears in plate 3.

In the velocity profile of the back wall, shown in figure 23, the influence of the intake bells can again be seen. In the regions very near the floor, two distinct humps in the negative direction are evident in the areas where the bells are located. The turbulence profile for this area, figure 28, is fairly constant across the back wall of the sump. There is a slight tendency for the turbulence to increase with height. This tendency is likely due to a greater abundance of large scale eddies further from the damping effects of the walls. Turbulence is produced near the wall due to the action of the boundary layer, but is small in comparison to the turbulence that already exists. Figures 24 through 28 show the turbulence profiles

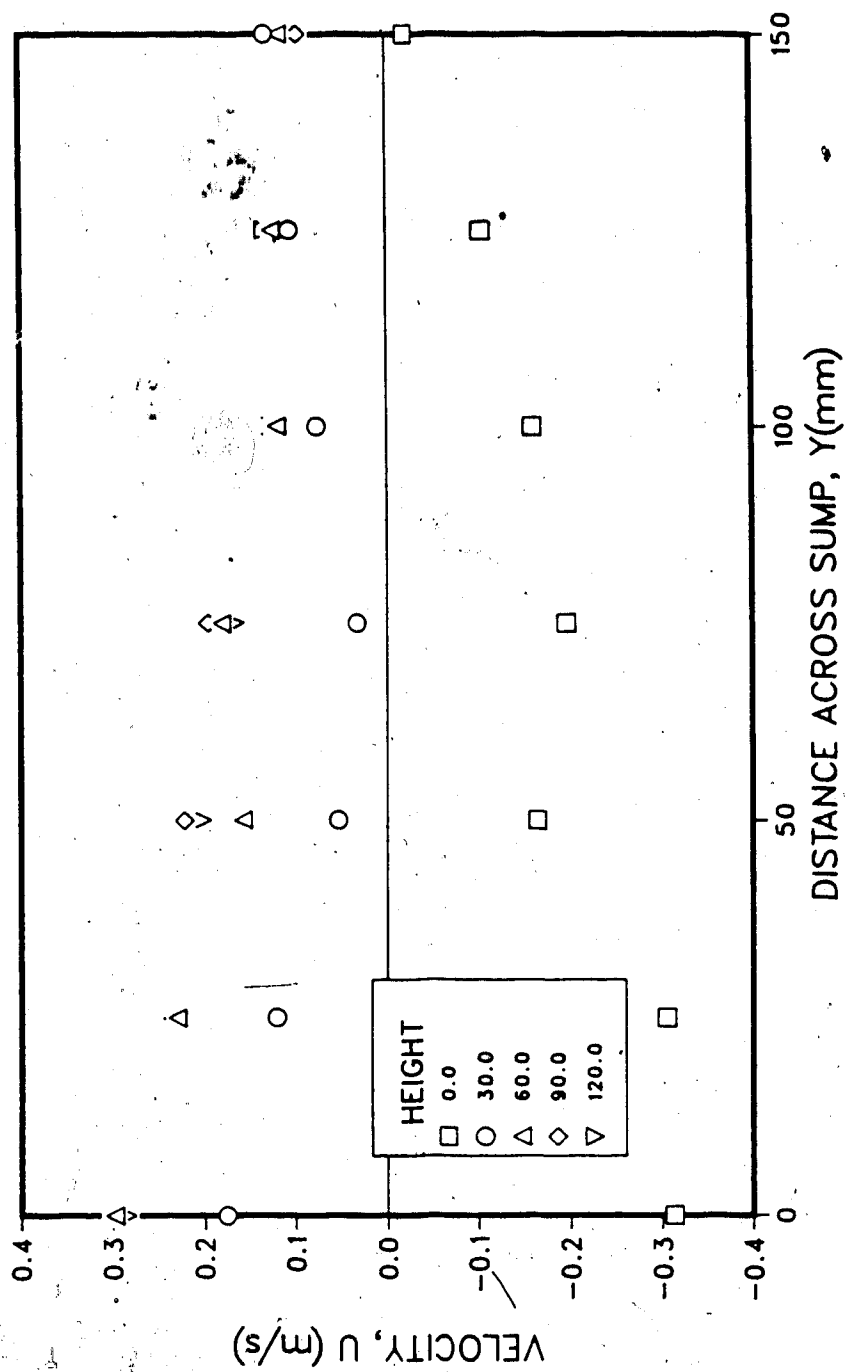


Figure 22. Longitudinal velocity profiles for the section between the bells of the improved sump, xbbels in figure 16.

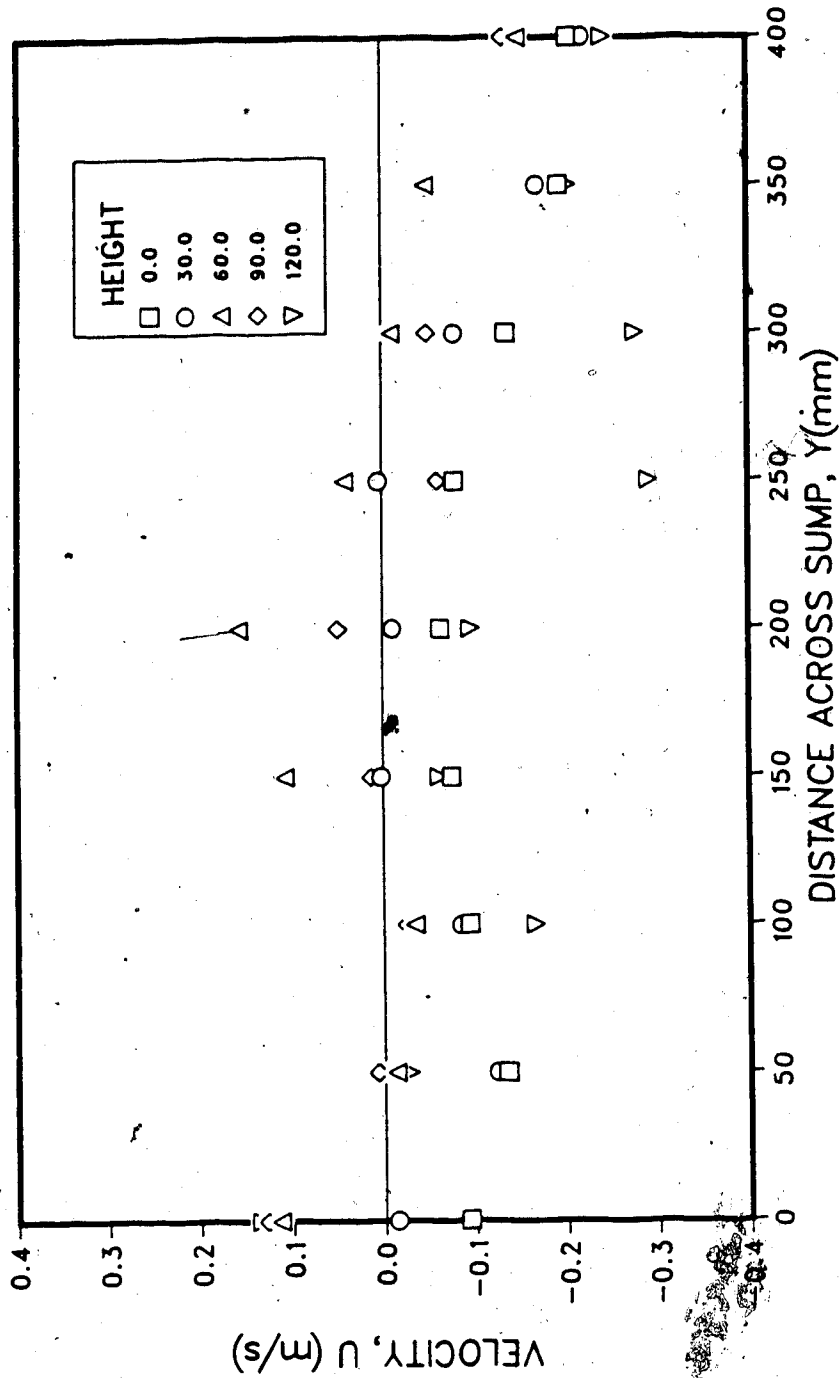


Figure 23. Longitudinal velocity profiles for the section near the back wall of the unimproved sump, xback in figure 16.

for the sections in the main sump.

### 2.6.2 Structure of Flow Under the Intake Bell

Figures 29 through 33 show the variation of the U component of the velocity. At sections one, two and five the flow is directed quite uniformly towards the centre of the intake bell. Sections three and four contain areas of positive and negative flow on opposite sides of the intake. These profiles illustrate the intense floor vortices seen constantly below the intake bells in the unimproved sump. An example of these vortices is shown in plate four. The velocity profile of the third cross section is typical of what would be expected across a vortex. High velocities of opposing sign exist on opposite sides of the vortex centre, with an area of lower velocities near the centre. The area of low velocities near the centre is much larger than it would be for an ideal vortex. Some of the lower velocities in the centre area may be low due to the averaging of high velocities with opposing sign.

For the third and fourth sections, the areas where the floor vortex was active, the turbulence was very much higher than for the other sections. The turbulence is highest in the centre, the area of smallest velocity. High turbulence is to be expected in these areas since this is where the vortex core will pass most frequently. The vortex was very intense as evidenced by the cavitation core that was usually present. As the vortex core passes the point where the

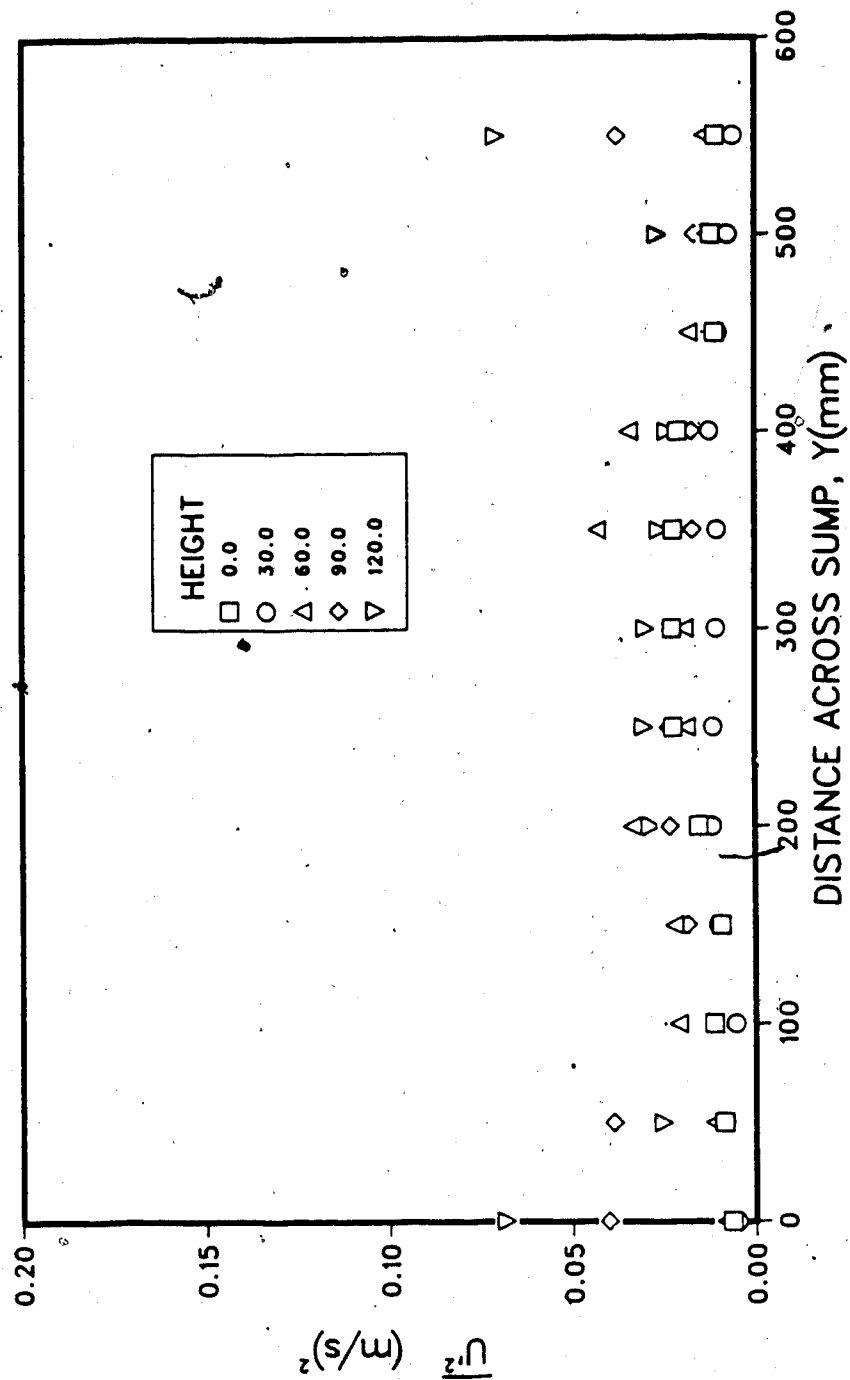


Figure 24. Longitudinal turbulent velocity fluctuations for the first section in unimproved sump, xsect#1 in figure



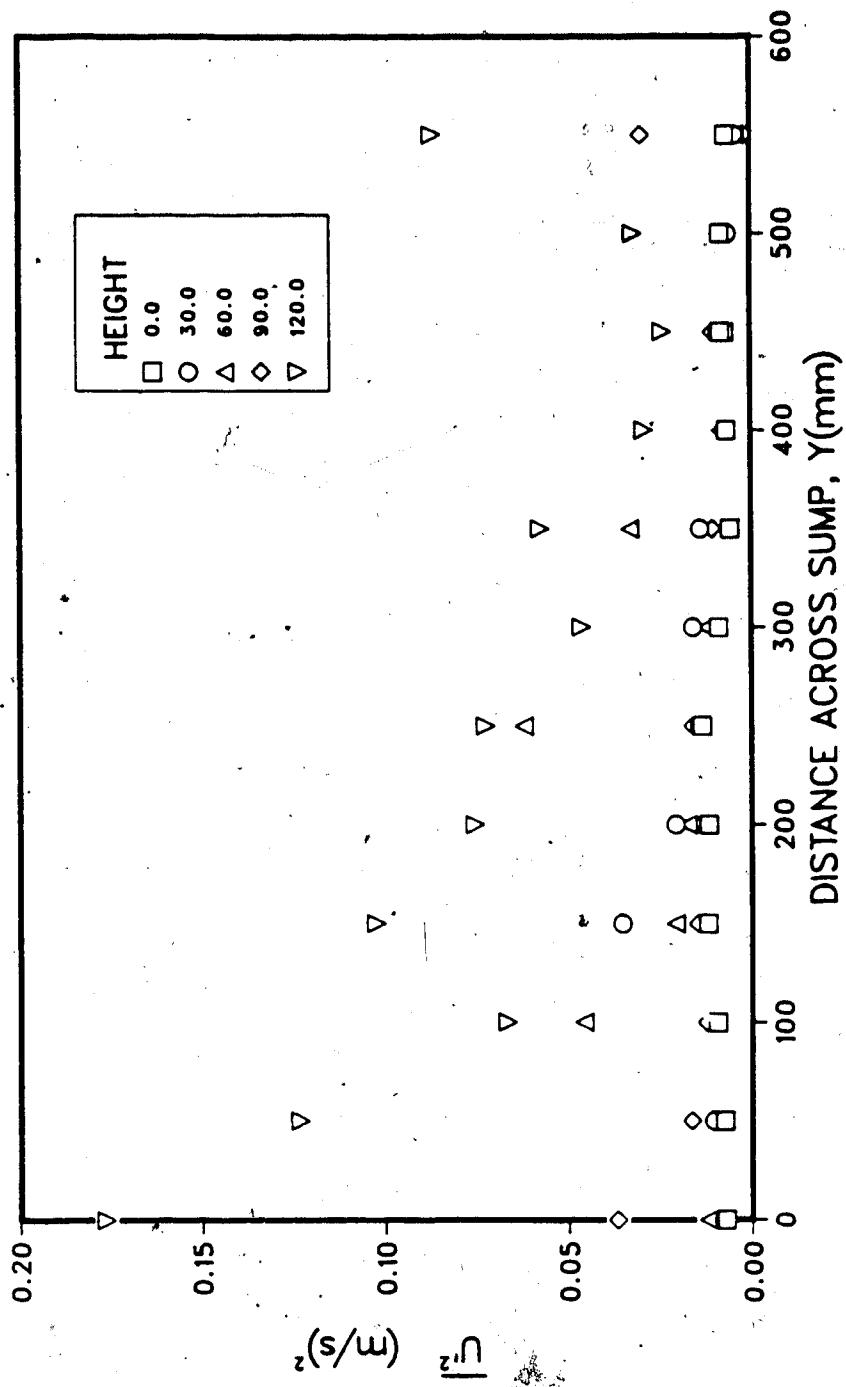


Figure 25. Longitudinal turbulent velocity fluctuations for the second section in unimproved sump, xsect#2 in figure 16.

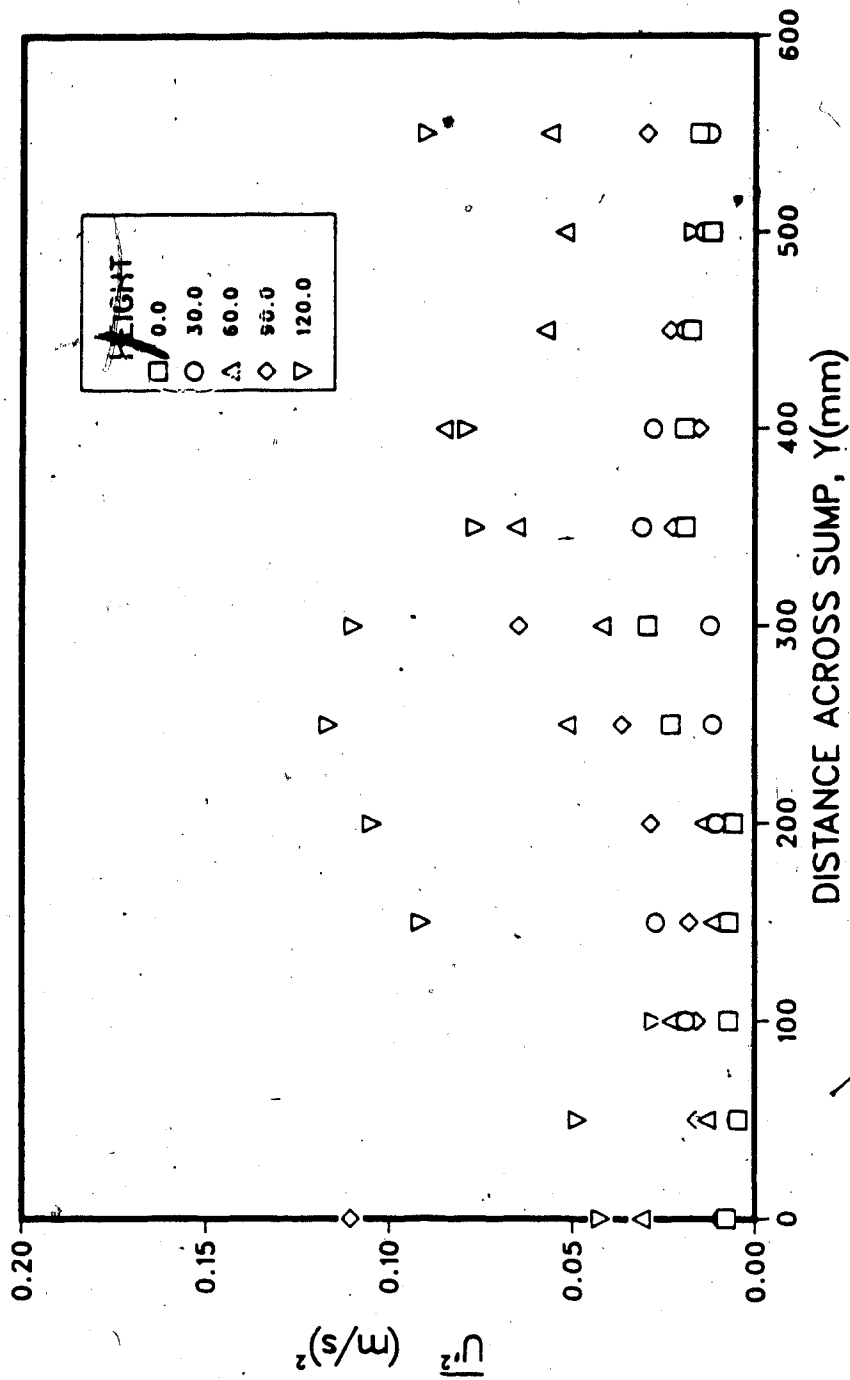


Figure 26. Longitudinal turbulent velocity fluctuations for the third section in unimproved sump, xsect#3 in figure

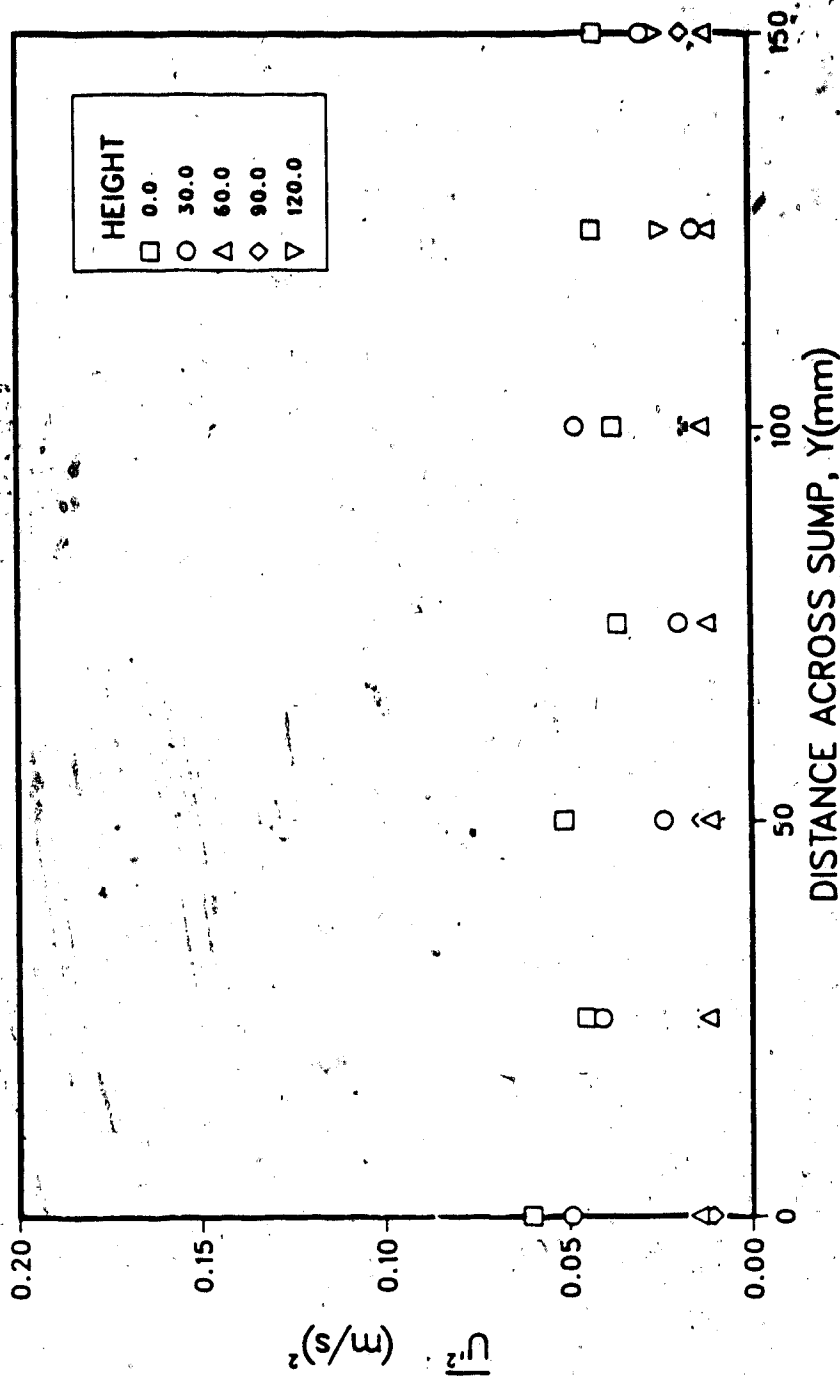


Figure 27. Longitudinal turbulent velocity fluctuations for the section between the bells of the unimproved sump, xbbels in figure 16.

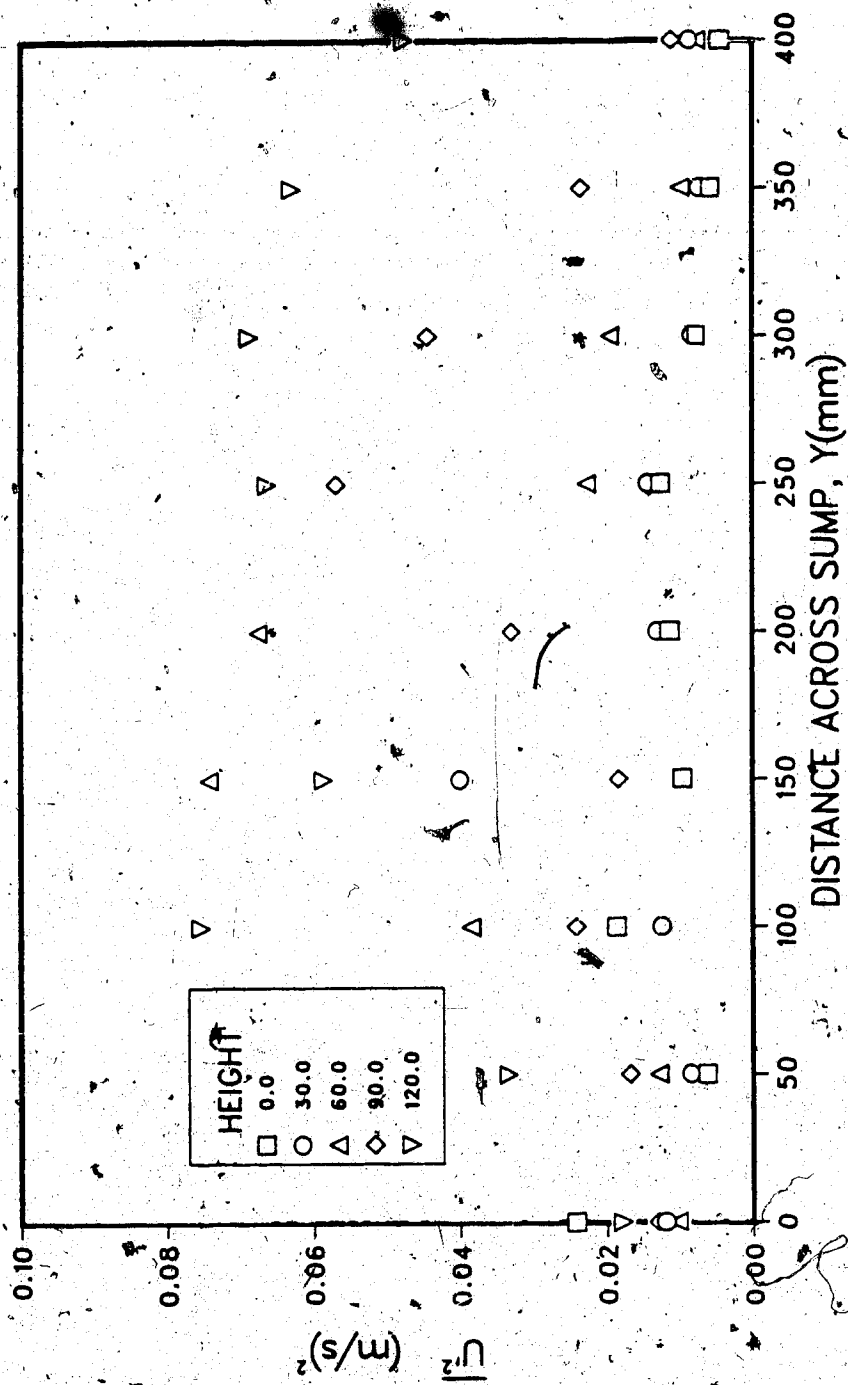


Figure 28. Longitudinal turbulent velocity fluctuations for the section near the back wall of the unimproved sump, xback in figure 16.

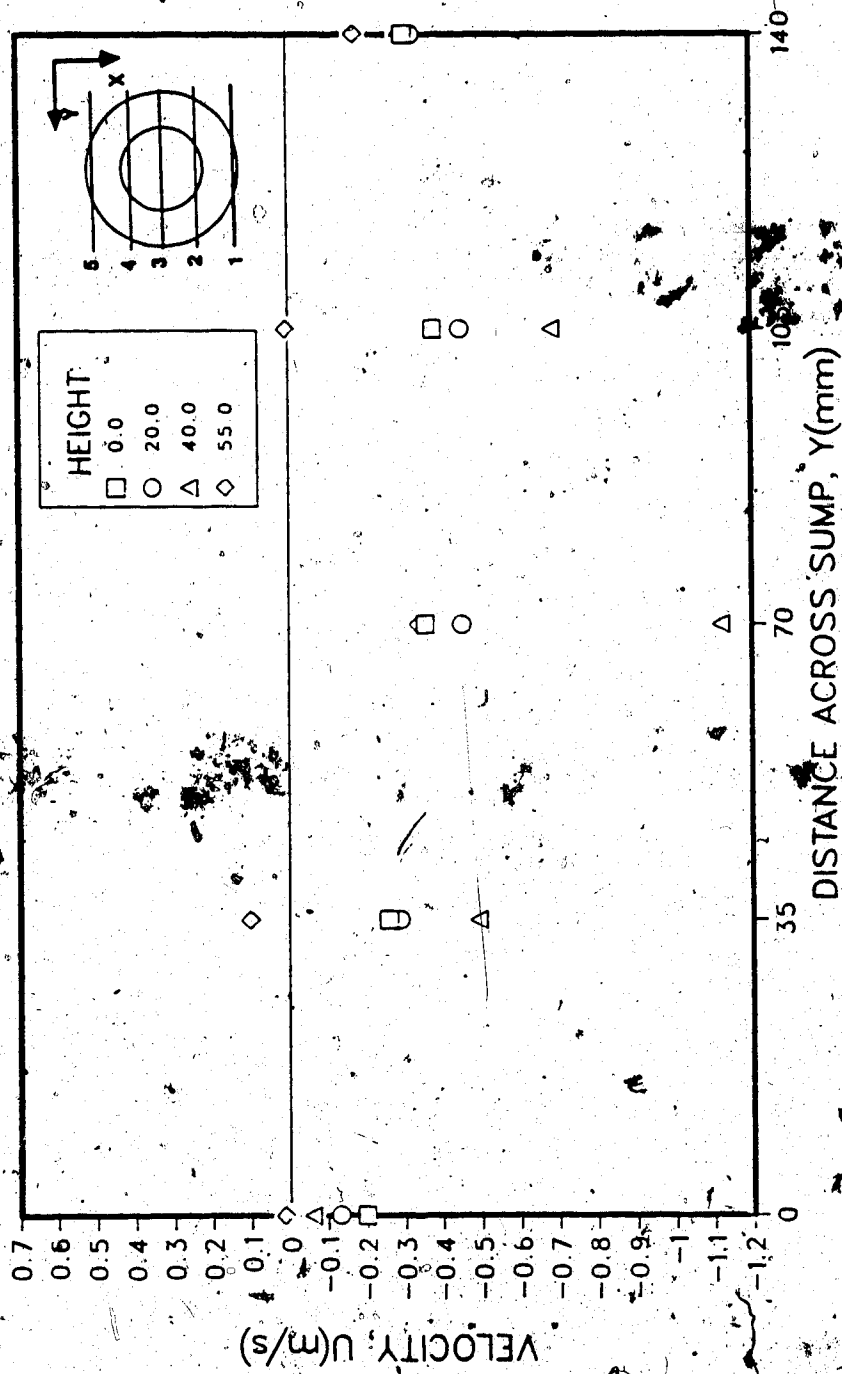


Figure 29. Longitudinal velocity profiles for the area under the north bell, section 1 in the area xubels of figure 16. Location shown in small inset figure.

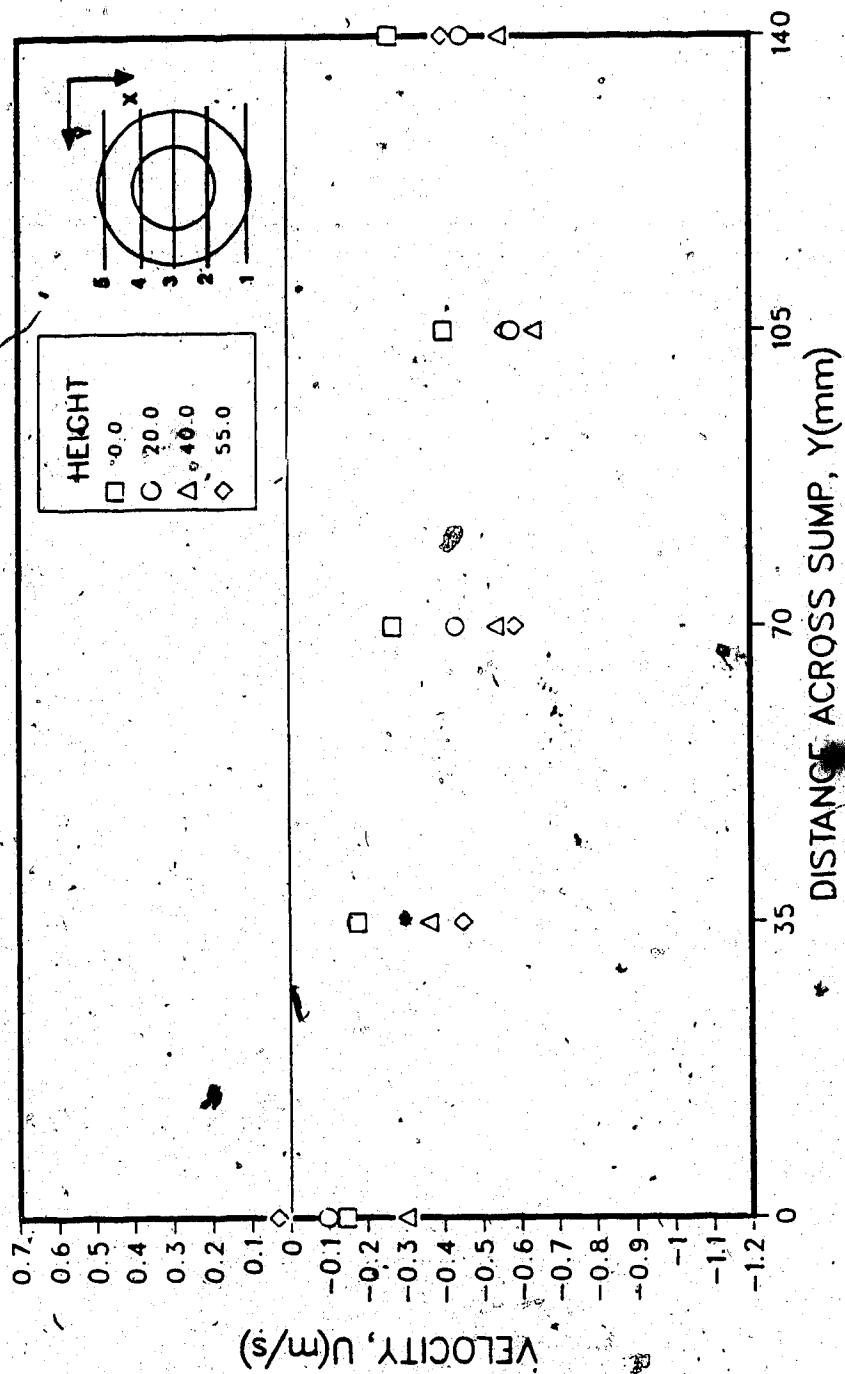


Figure 30. Longitudinal velocity profiles for the area under the north bell, section 2 in the area subplots of figure 16. Location shown in small inset figure.

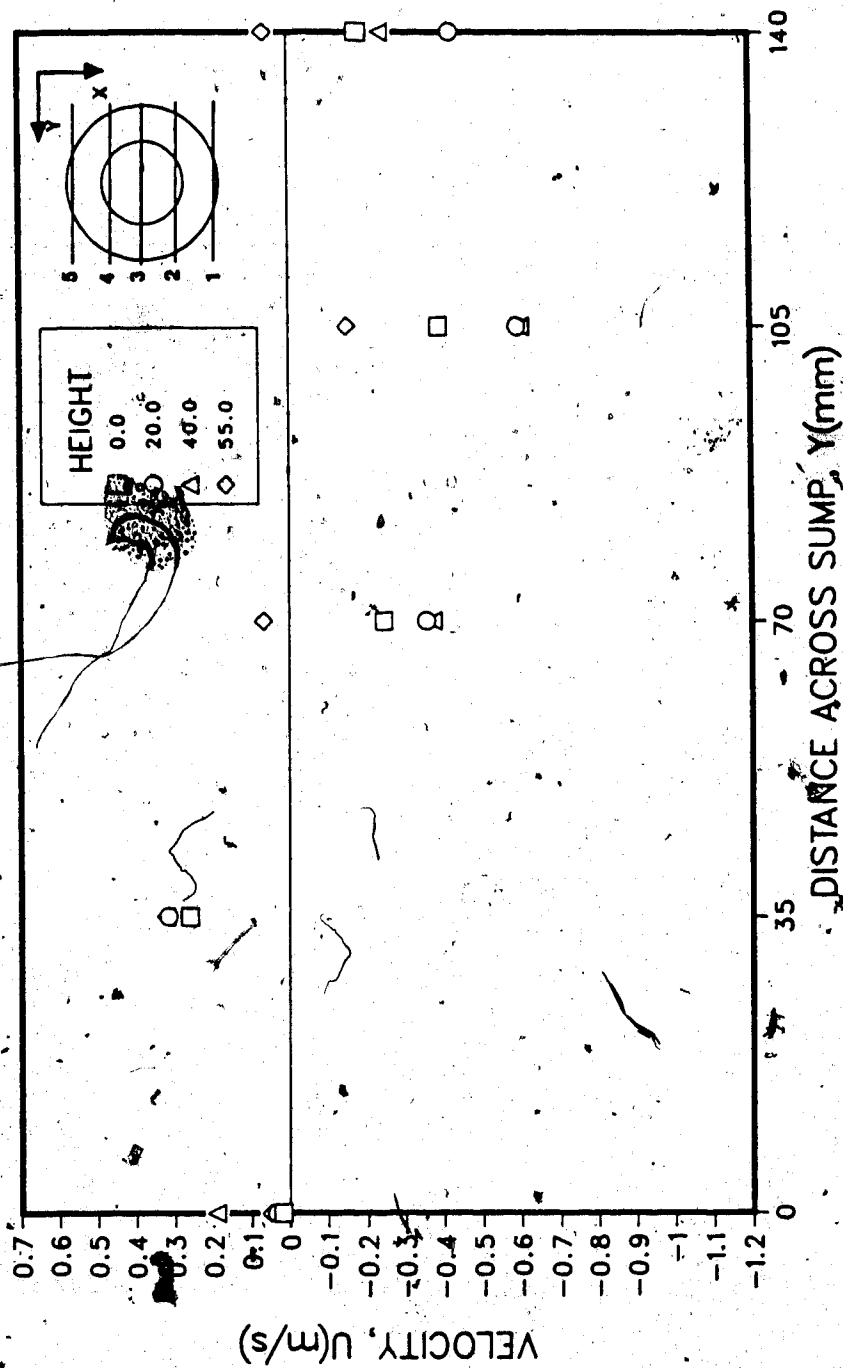


Figure 31. Longitudinal velocity profiles for the area under the north bell, section 2, in the area subels of figure 16. Location shown in small inset figure.

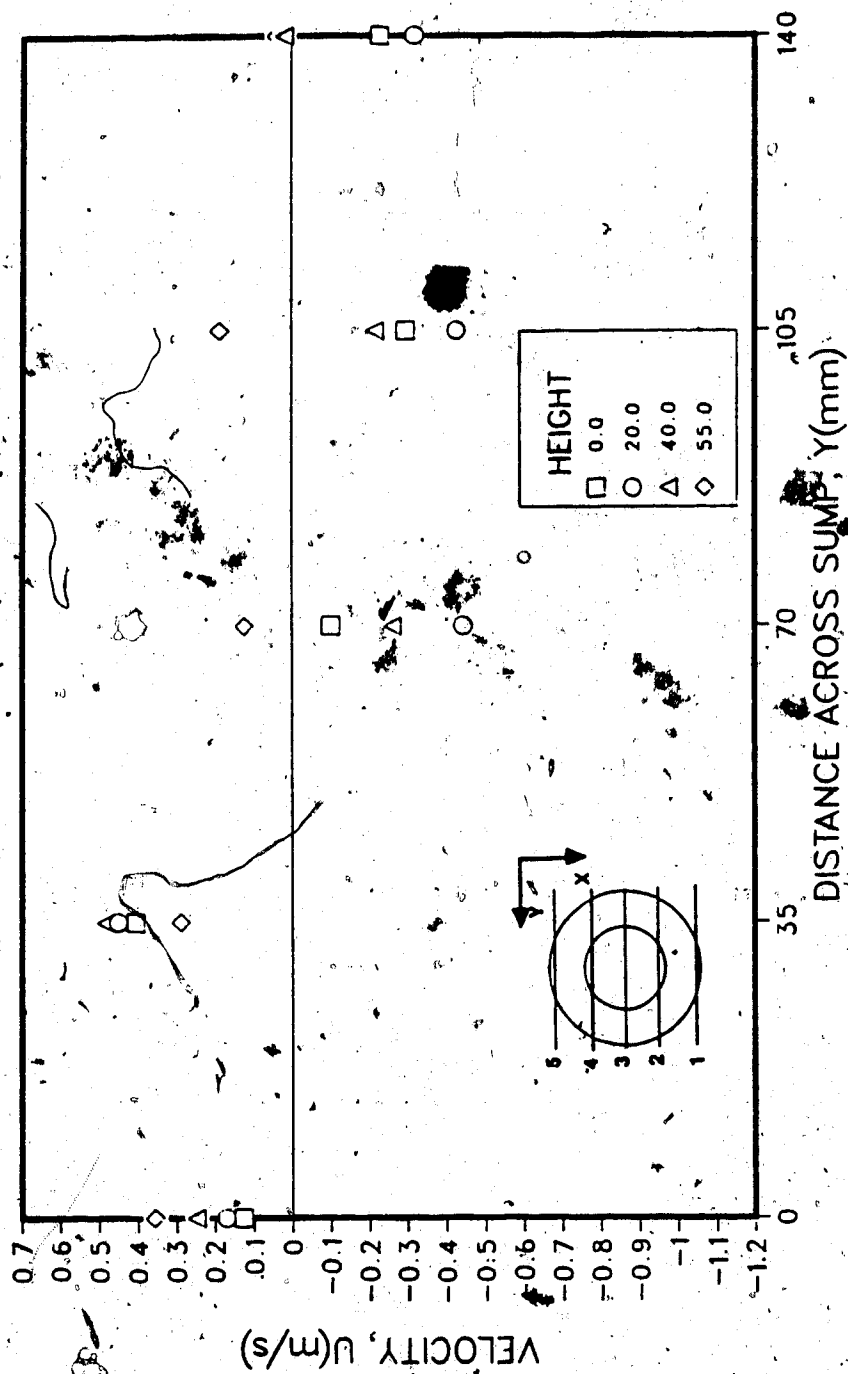


Figure 32. Longitudinal velocity profiles for the area under the north bell, section 4 in the area xubels of figure 16. Location shown in small inset figure.



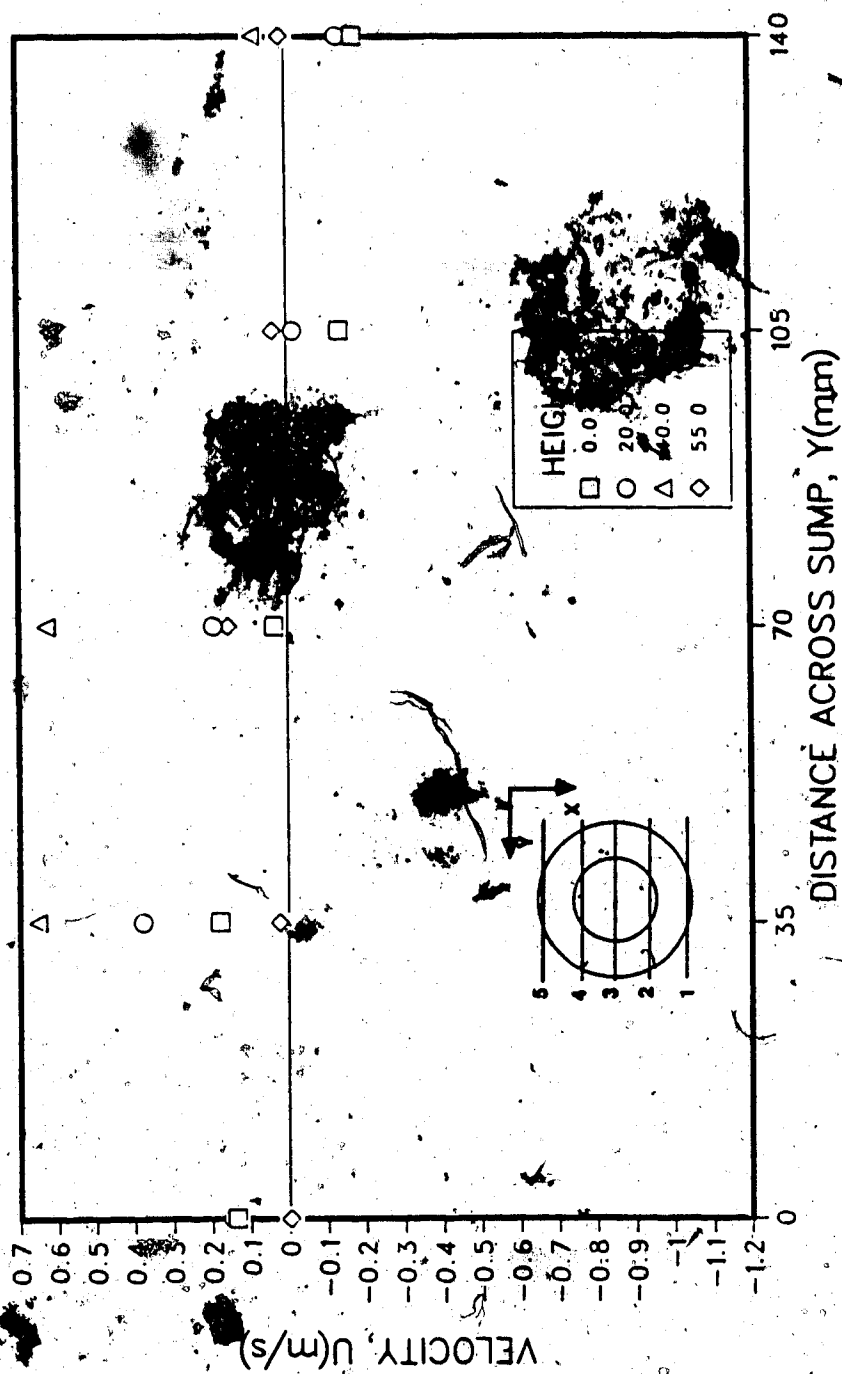


Figure 33. Longitudinal velocity profiles for the area under the north bell, section 5 in the area xubels of figure 16. Location shown in small inset figure.

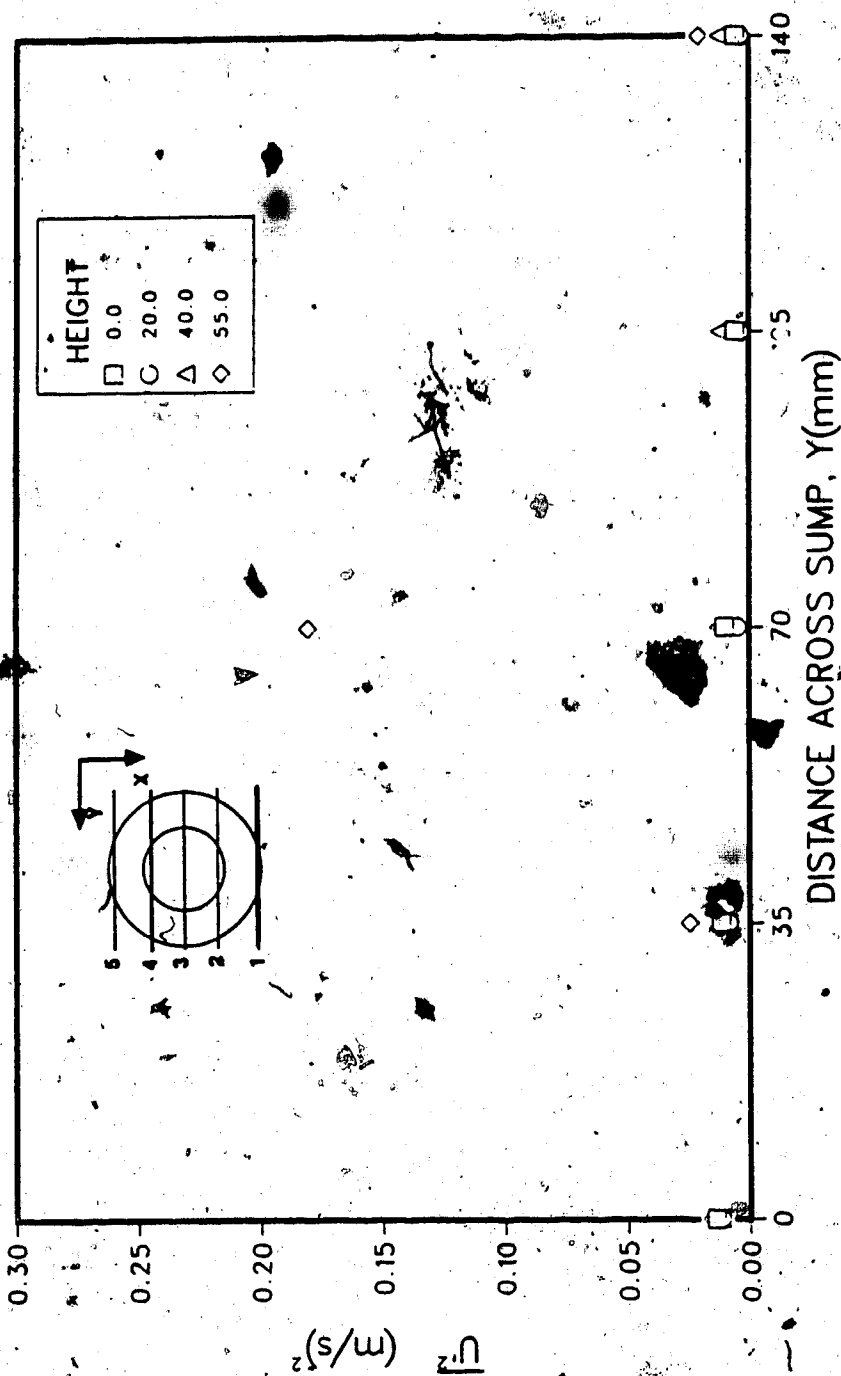


Figure 34. Longitudinal turbulent velocity fluctuations for the area under the north bell, section 1 in the area subplots of figure 16. Location shown in small inset figure.

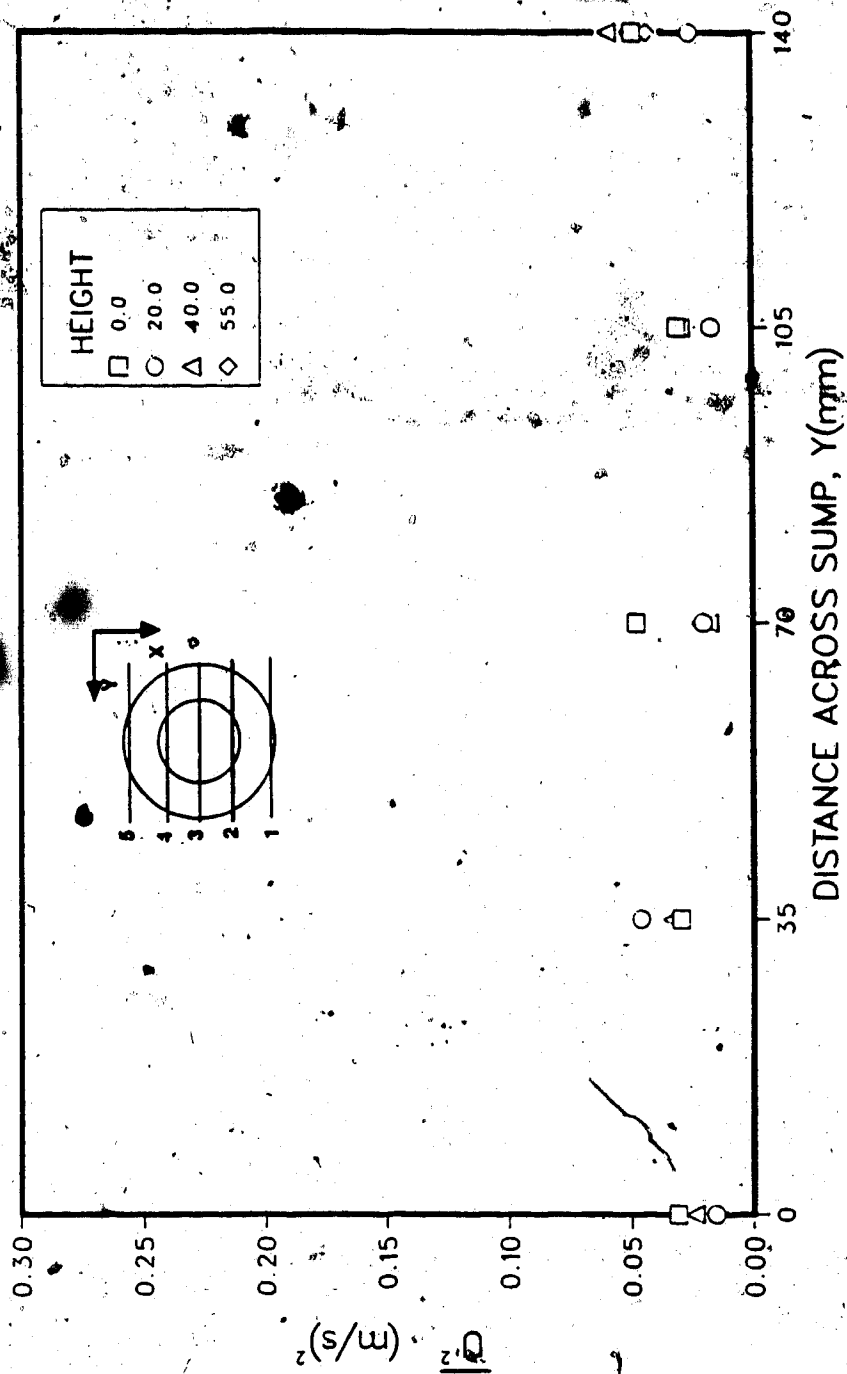


Figure 35. Longitudinal turbulent velocity fluctuations for the area under the north bell, section 2 in the area xubels of figure 16. Location shown in small inset figure.

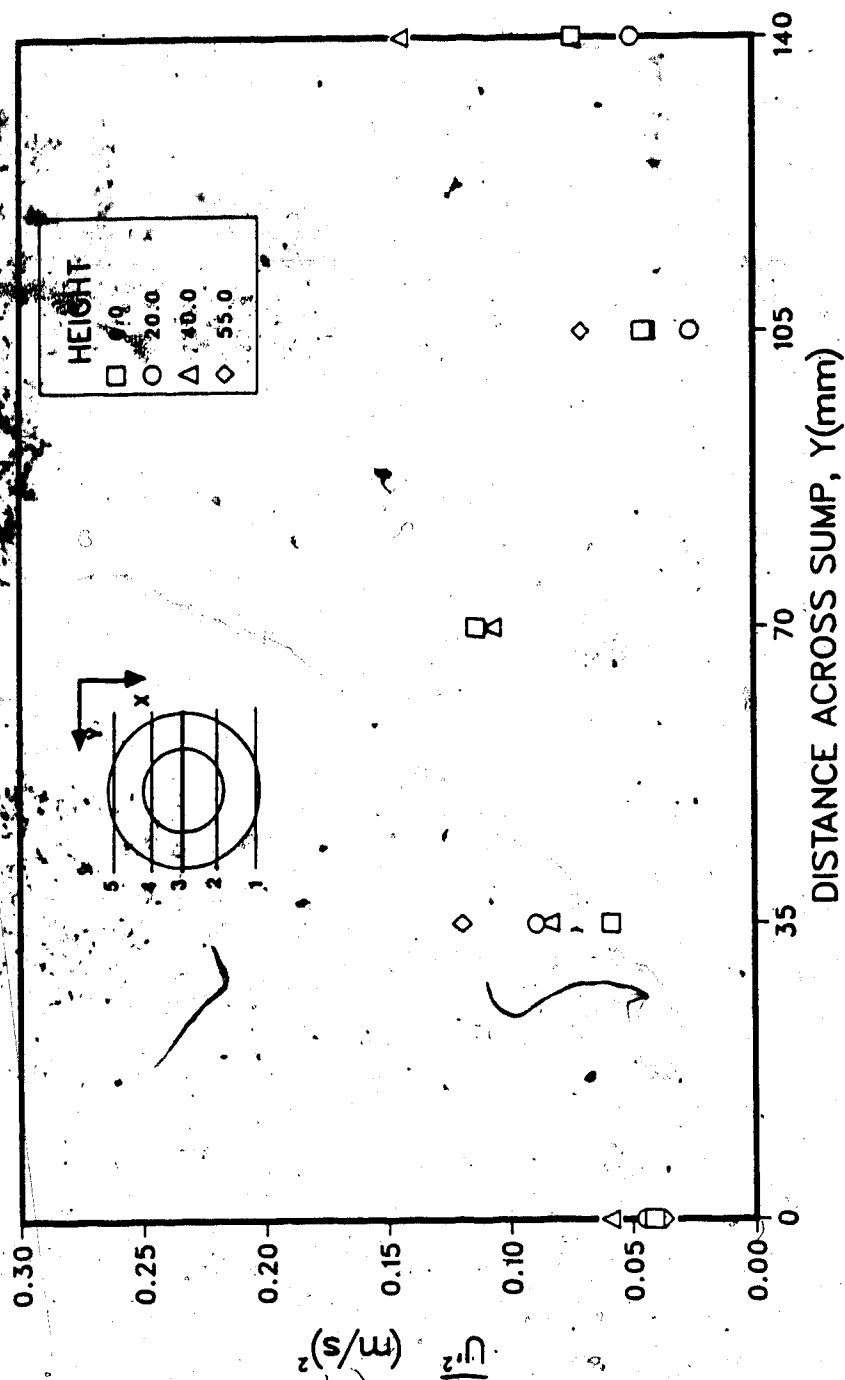


Figure 36. Longitudinal turbulent velocity fluctuations for the area under the north bell, section 3 in the area xubels of figure 16. Location shown in small inset figure.

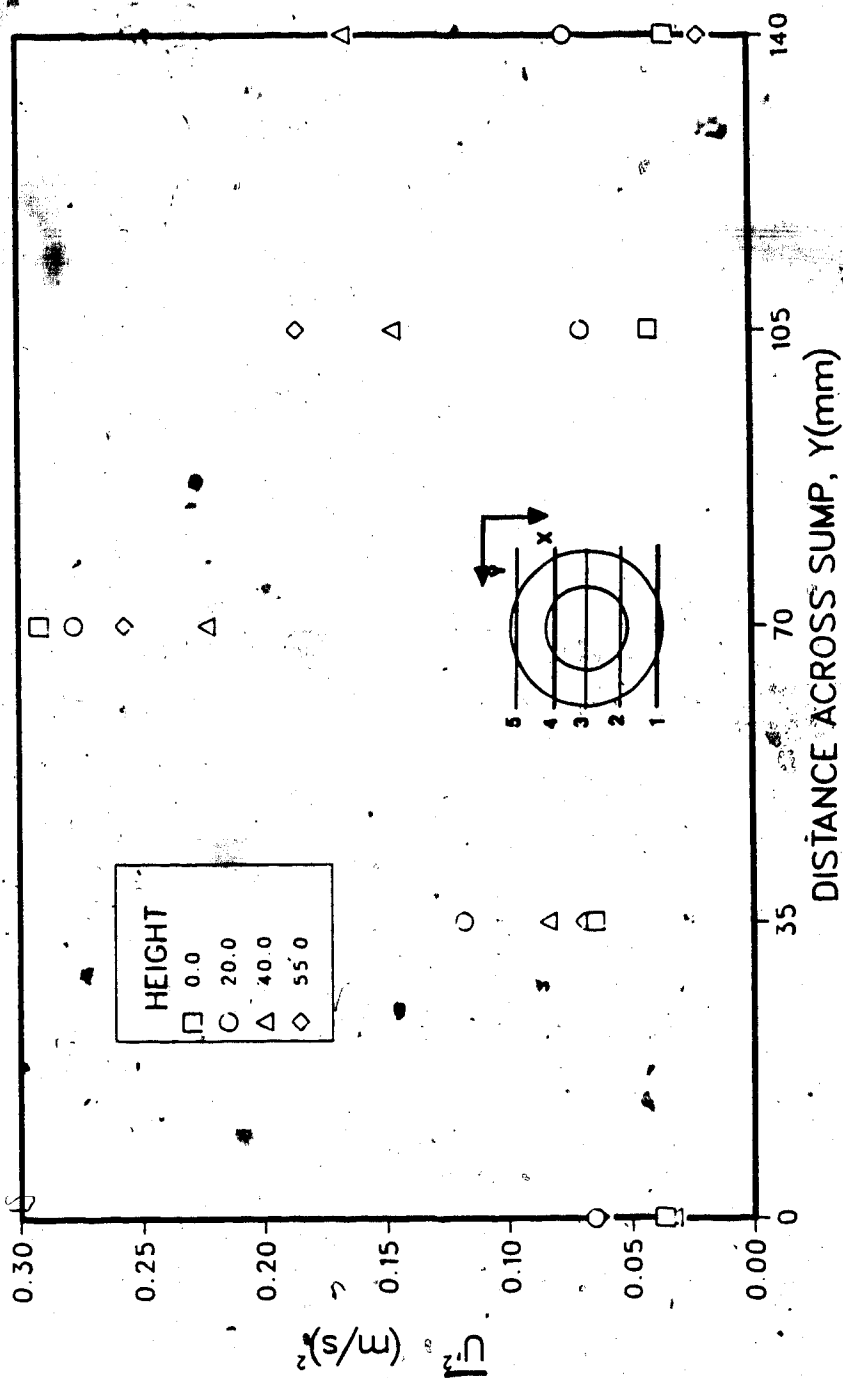


Figure 37. Longitudinal turbulent velocity fluctuations for the area under the north bell, section 4 in the area subplots of figure 16. Location shown in small inset figure.

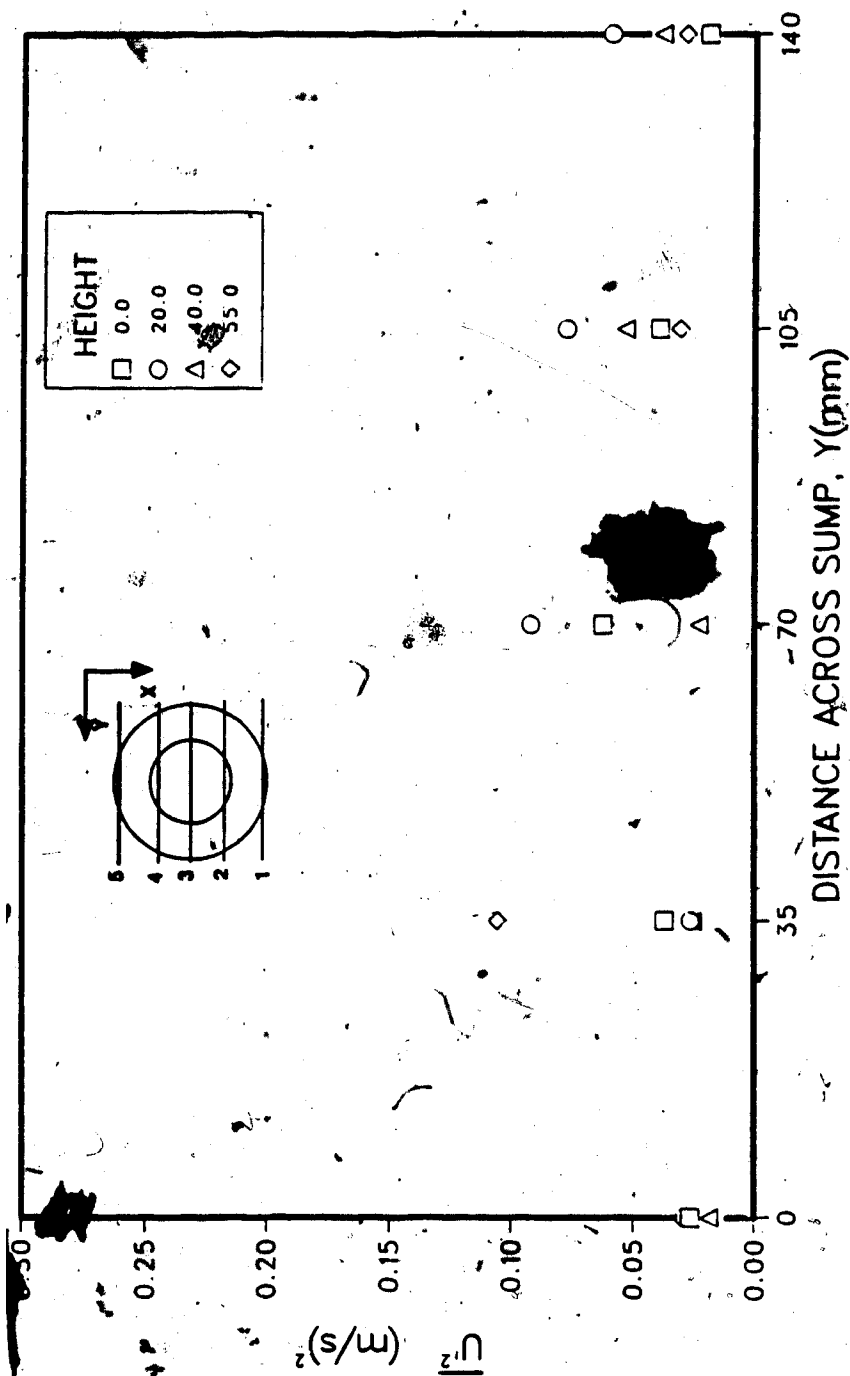


Figure 38. Longitudinal turbulent velocity fluctuations for the area under the north bell, section 5 in the area subplots of figure 16. Location shown in small inset figure.

velocity is measured, the velocity will change dramatically from positive to negative, or negative to positive depending on the direction of rotation. Figures 34 through 38 show the turbulence profiles for this area.

The transverse velocity profiles, figures 39 through 43, show the same features of the flow as did those in the longitudinal direction. The velocity profiles at  $y$  values of 0, 35, 105, and 140 millimetres show a steady flow towards the intake bell. No new trends can be found from these velocity profiles. The transverse velocity profile for  $y = 70\text{mm}$ , figure 41, again illustrates the vortexing motion under the bells. The velocities near the back wall are negative and those farther away are positive showing the tendency for the fluid in this area to circulate. This tendency does not show up as clearly here as it had in the  $U$  velocity profiles. The velocities for the centre of this profile are the highest, which is opposite to the  $U$  velocity trends. This could only be due to the vortex core staying on one side of the data collection point for the majority of the time. Plates 5 and 6 illustrate more vortexing problems.

### 2.6.3 Inside, the Intake Pipes

Figures 44 and 45 show the vertical variation of both the transverse ( $V$ ) and longitudinal ( $U$ ) velocities at a distance of one quarter of the pipe diameter inside the intake pipe. This area of the sump contains the results of all the nonuniformities of the flow in the main sump. The

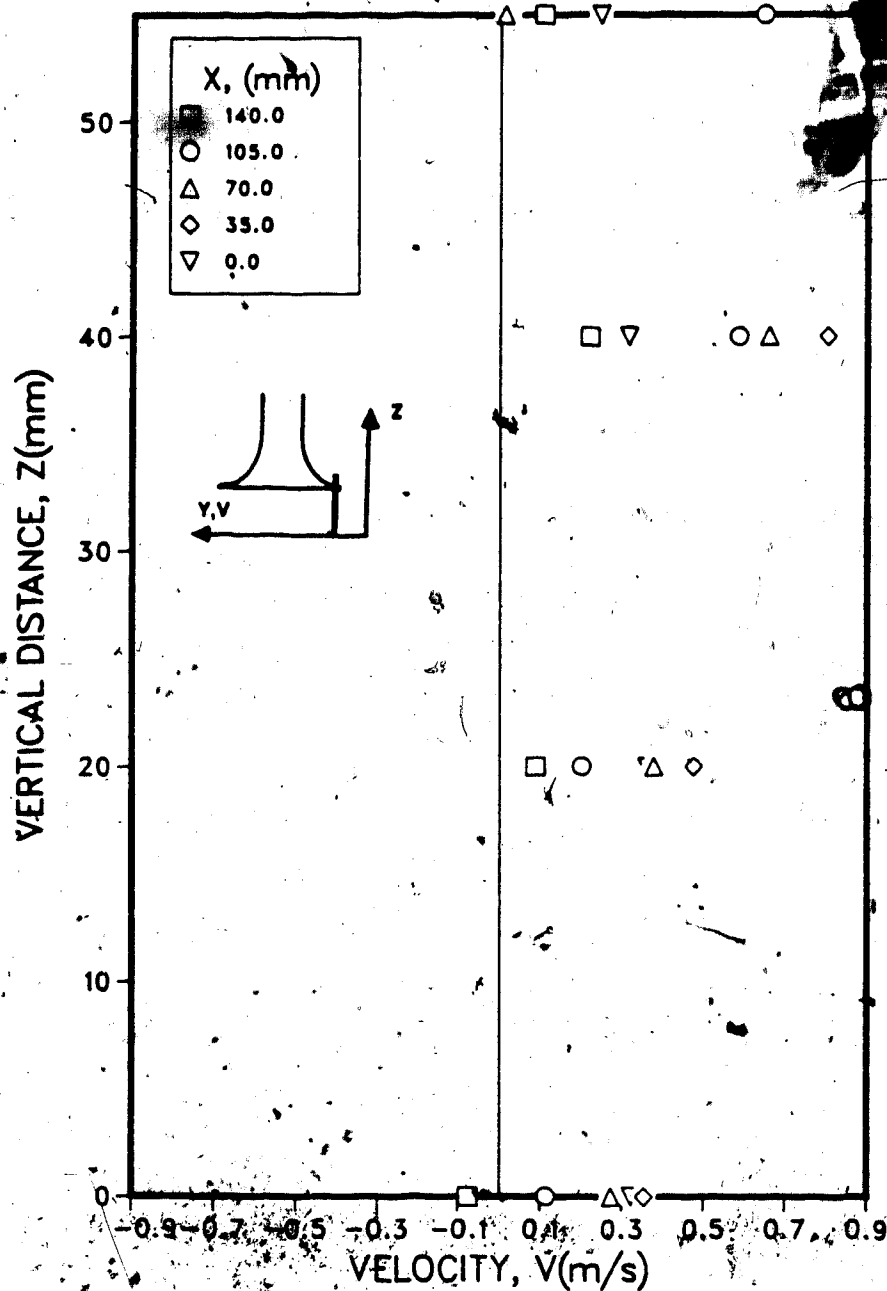


Figure 39. Vertical profiles of the transverse velocity under the north bell of the unimproved sump, profiles at  $y = 0$  in the area xubels. Location shown in small inset figure.



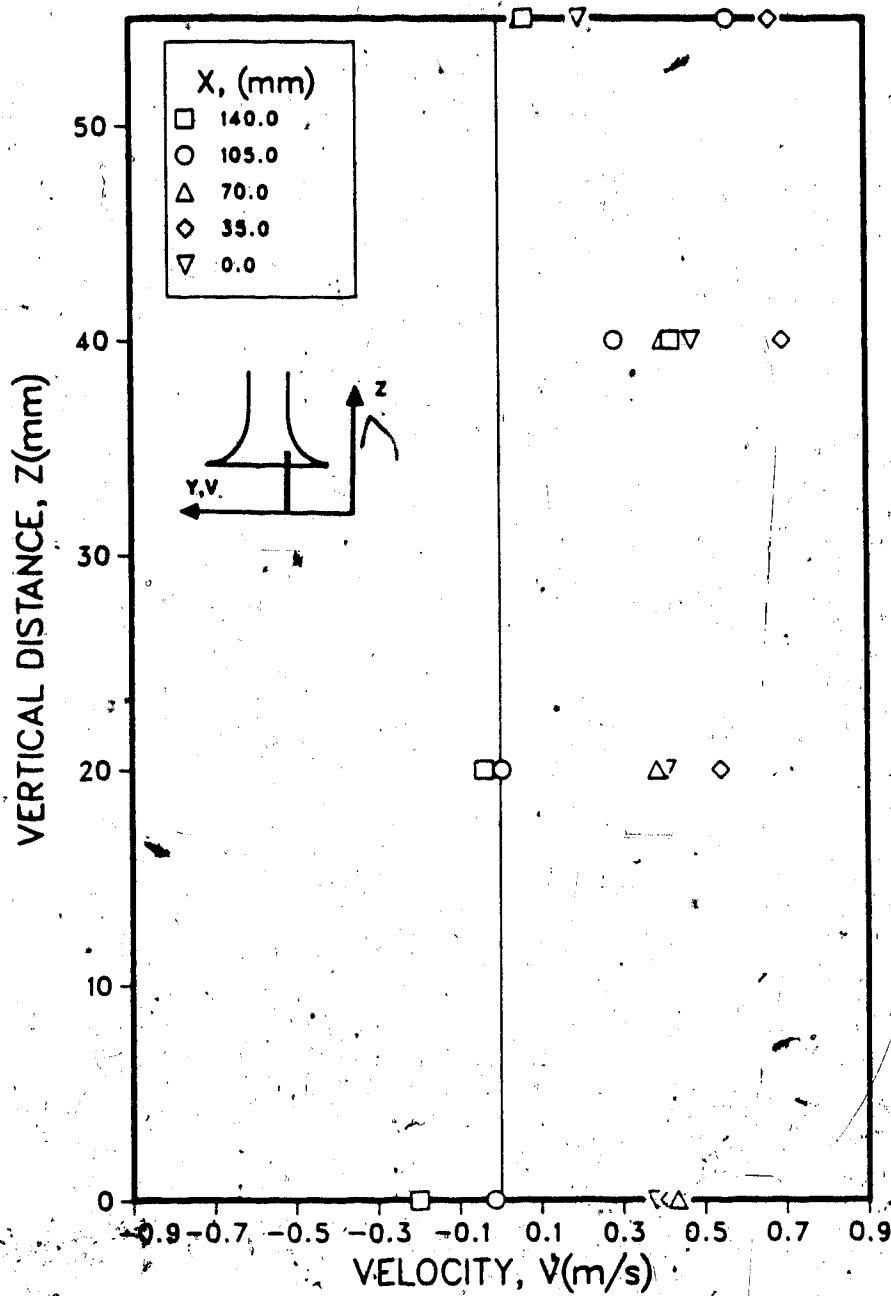


Figure 40. Vertical profiles of the transverse velocity under the north bell of the unimproved sump, profiles at  $y = 35$  in the area xubels. Location shown in small inset figure.