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Full Name of Author — Nom complet de l'auteur

AGONICHUKWU SUNDAY CHUKWUEMEKA EZELIDEMBAH

Date of Birth — Date de naissance

February 4, 1945

Country of Birth — Lieu de naissance

NIGERIA

Permanent Address — Résidence fixe

c/o St James' Church, NANKA, Via Aguata L.G.A.  
ANAMBRA STATE, NIGERIA

Title of Thesis — Titre de la thèse

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Prof. P.M. Dranchuk

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STEADY STATE RADIAL GAS FLOW THROUGH POROUS MEDIA



by  
AGODICHUKWU S. C. EZEUEDEMBAH

A THESIS

SUBMITTED TO THE FACULTY OF GRADUATE STUDIES AND RESEARCH  
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OF MASTER OF SCIENCE

IN  
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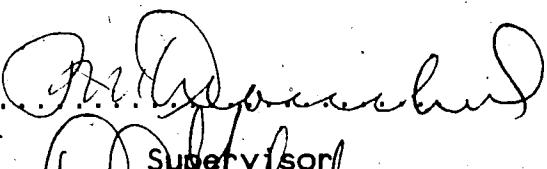
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for acceptance, a thesis entitled STEADY STATE RADIAL GAS  
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EZEUDEMBAH in partial fulfilment of the requirements for the  
degree of MASTER OF SCIENCE in PETROLEUM ENGINEERING.

Prof. P.M. Dranchuk

  
Supervisor

Dr. D.L. Flock

  
Dr. R.G. Bentsen

  
Dr. N. Rajaratnam

Date October 30, 1980

To the memory of my father

Daniel Ogbonnaya EZEUDEMBAH

March 15, 1915 - December 27, 1979

## Abstract

This work deals with the possible causes of anomalous gas flow behavior which has been observed on both Klinkenberg and visco-inertial plots at higher flow rates and pressures. In pursuance of this study, experiments were carried out at higher flow rates and pressures, and at lower flow rates and pressures, with the following objectives:

1. evaluation and re-arrangement of flow case studies for use in the experiments;
2. re-evaluations of the existing visco-inertial flow model equation which uses Forchheimer's quadratic equation;
3. consideration of stress and strain on the core during the course of radial flow under uniaxial confinement;
4. a check on the possibility of hysteresis effect.

The plot profiles, obtained by conducting the experimental runs at higher flow rates and pressures, suggest a deviation from expected profiles which fit the visco-inertial flow model equation. By analysis, and model fitting, these profiles are found to fit Forchheimer's cubic equation. The log-log plot of the friction factor against the Reynolds number suggests an existence of some term beyond the visco-inertial flow regime. The plot profiles, obtained by conducting the runs at lower flow rates and pressures, conform to the visco-inertial flow model equation. Similar anomalous flow behavior, in which the apparent gas permeability increases for increasing flow

rate, is found to be dependent on both the flow method, and the type of radial confinement used. Application of stress-strain analysis has proved to be a vital tool in the study of confining pressure effects on flow behavior. Hysteresis has been observed to be significant in the parameter estimated values.

Both Forcheimer's quadratic and cubic equations have been derived by considering Navier-Stokes equations and dimensional analysis for the quadratic case, and kinetic energy equation of mean flow and dimensional analysis for the cubic case. The cubic term is regarded as an extension to the quadratic equation at higher flow rates. The physical existence of its main parameter, gamma, together with the flow regime in which it operates, is established by consideration of the boundary layer theory to the flow problem. Properties of gamma, obtained from experiment, agree with those obtained from theory.

Similar runs were made under triaxial confinement, using a triaxial overburden radial cell. The results, however, are not conclusive due to design problems.

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Cordial and understanding atmosphere within the family contributed greatly to the completion of this work. To this effect, the author gives credit to his wife, Onyemere, his son, Udoka, and his other family members in Nigeria, who, despite the present trying period within the family, gave him full support.

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## Nomenclature

- a = Hydraulic resistivity in the Kozeny-Carman form,  
 $K^{-1}$ , in Equations 2.5, 2.22
- A = Cross-sectional Area
- A = Arbitrary constant vector used in Appendix B
- A' = Area open to flow in Appendix A
- $A_1, A_2$  = Arbitrary constants used in Chapter 5
- b = Slippage coefficient,  $[ML^{-2}T^{-2}]$  in Equation 2.7
- B = Arbitrary constant vector used in Appendix B
- CC1 = Coefficient of the viscous term of the general cubic  
flow equation, Chapter 4.
- CC2 = Coefficient of the inertial term of the general  
cubic flow equation, Chapter 4.
- CC3 = Coefficient of the cubic term of the general cubic  
flow equation, Chapter 4.
- Cd = Orifice drag coefficient in Equations 2.30, 2.31
- $C_L$  = Laminar energy loss coefficient, in Equations 2.30,  
2.31
- CP = Confining pressure
- $C_t$  = Turbulent energy loss coefficient, Equations 2.30,  
2.31
- $C_1-C_3$  = Arbitrary flow physical constants in the general  
cubic flow equation, Chapter 4
- $C_1-C_6$  = Arbitrary flow physical constants in the general  
cubic flow equation, Appendix A
- d = Boundary layer thickness in Equation A-26
- dc = Characteristic hydraulic matrix mean pore size

diameter, Equations 2.30, 2.31

dm = Mean grain diameter in Equation 2.5

D = Skin frictional coefficient, dimensionless, in  
Appendix A

Da = Average diameter of the particles, ft, in Equations  
2.26, 2.27

D,DT = Arbitrary constant vector and its transpose used in  
Appendix B

E = Young's Modulus in Chapter 5

E = Arbitrary constant vector used in Appendix B

f = Friction factor, in Equations 2.15, 2.16, 2.27

Fb = Inertial resistance coefficient, Beta,  $\text{ft}^{-1}$ ,  $\text{m}^{-1}$

g = Acceleration due to gravity

G = Gas Gravity in Equations 2.26, 2.27

Gs = Superficial mass velocity,  $\text{lbmass}/\text{ft}^2\text{sec}$ , in  
Equations 3.1, 3.2, 3.3

h = Height, [L]

J = Hydraulic gradient, in Equation 2.22

K = Absolute gas permeability at infinite mean pressure  
in Equation 2.7,  $[\text{L}^2]$ , md,  $\mu\text{m}^2$

Ka = Apparent gas permeability, calculated from Darcy's  
law in Equation 2.7,  $[\text{L}^2]$ , md,  $\mu\text{m}^2$

K = Hydraulic conductivity which measures the  
permeability of the porous medium, in Equation  
2.5

L = Length scale in Appendix A

Le = Overall length of the porous media, ft, in Equation  
3.1

$m, n$  = Exponential values for porosity, in Equation 2.26,  
2.27

$m(P)$  = Real gas pseudo-pressure in Equation 2.12

$M$  = Molecular weight of gas

$P$  = Pressure at ambient condition, atm, kPa, [ML<sup>-2</sup>T<sup>-2</sup>]

$P_b$  = Base pressure in Equation 2.12

$P_e, P_2$  = Outer pressure, atm, kPa, [ML<sup>-2</sup>T<sup>-2</sup>]

$\beta_{EW}$  = Coefficient of the slippage term of the general  
cubic flow equation, Chapter 4.

$P_{hc}$  = Hydrostatic rock confinement pressure, psi

$P_m$  = Arithmetic mean flowing fluid pressure =  $(P_e + P_w)/2$

$P_{nc}$  = Net rock confinement pressure, psi

$P_w, P_1$  = Pressure, psi, kPa

$q$  = Gas flow rate, Q/Area

$Q$  = Gas flow rate, Mscf/day.

$Q_o$  = Gas flow rate, cm<sup>3</sup>/sec, m<sup>3</sup>/day, at standard  
conditions

$r_e, r_2$  = Outer radius, [L]

$r_w, r_1$  = Inner radius, [L]

$Re$  = Reynolds number, dimensionless.

$S$  = Shape factor which is dependent on the shape of the  
grains and their structure in Equation 2.5

$S'$  = Shape factor for increasing inertial effect, in  
Equation 2.23

$S_r$  = Principal stress in the radial direction

$S_t$  = Principal stress in the tangential direction

- $S_z$  = Principal stress in the axial direction
- $T$  = Temperature, °R
- $T_a$  = Average temperature of flowing gas, °R
- $u$  = Velocity notation in Appendix A
- $U$  = Flow rate of displacement in Chapter 5
- $U$  = Velocity scale in Appendix A
- $U'$  = Fluctuation's velocity scale
- $v$  = Superficial gas velocity, ft/sec, Equations 2.18,  
2.19, 2.20, 2.30
- $w$  = Arbitrary constant vector used in Appendix B
- $W_i, W_n$  = Arbitrary constants used in Appendix B
- $W_o, W_n$  = Arbitrary constants used in Appendix B
- $W_o - W_5$  = Arbitrary constants used in Appendix B
- $W_s$  = Weight of flowing gas, lb/sec, in Equations 2.26,  
2.27
- $x$  = Distance between plates of orifice model, in  
Equations 2.30, 2.31
- $X$  = Arbitrary constant vector used in Appendix B
- $Y$  = Arbitrary constant vector used in Appendix B
- $Z$  = Compressibility factor of gas
- $Z_a$  = Average compressibility factor
- $\alpha$  = Alpha, Viscous coefficient,  $[L^{-2}]$ ,  $K^{-1}$ ,  $ft^{-2}$ , in  
Equations 2.20, 2.28
- $\beta$  = Beta, Inertial coefficient,  $[L^{-1}]$
- $\gamma$  = Gamma, Parameter in the cubic term,  $[LTM^{-1}]$ ,  
 $atm \cdot cm^2 \cdot sec^3 / gm^2 = cm \cdot sec/gm, m \cdot sec/kg$
- $\lambda$  = Mean free path of the flowing gas.

- $\mu$  = Gas viscosity, [ $ML^{-1}T^{-1}$ ], Ib/ft-sec, cp,  $\mu$ Pa.s  
 $\nu$  = Kinematic viscosity, [ $L^{-1}$ ], in Equation 2.5,  
 $atm.sec^2/gm = cm^{-1}$   
 $\nu$  = Poisson's ratio in Chapter 5.  
 $\epsilon_r$  = Principal strain in the radial direction  
 $\epsilon_t$  = Principal strain in the tangential direction  
 $\epsilon_z$  = Principal strain in the axial direction  
 $\rho$  = Density of gas, [ $ML^{-3}$ ]  
 $\rho_0$  = Density of gas at standard condition, [ $ML^{-3}$ ]  
 $\Phi$  = Porosity

### Conversion Factors

- $T_0 = 288.150 ^\circ K$ , at standard condition  
 $P_0 = 1 atm = 101.325 kPa$ , at standard condition  
 $Z_0 = \text{approximately unity}$ , at standard condition  
 $g_C = 32.17 \text{ lbm.ft/lbf.sec}^2$   
 $R = \text{Gas constant,}$   
 $R = 1545. \text{Ibforce.ft}^3/\text{ft}^2.\text{Ibmole} \cdot R$   
 $R = 82.056 \text{ atm.cm}^3/\text{gmmole} \cdot K$   
 $R = 8.3143 \text{ kPa.m}^3/\text{kgmole} \cdot K$   
 $1 \text{ cp} = 10^3 \mu\text{Pa.s}$   
 $1 \text{ darcy} = 0.9869 \mu\text{m}^2$   
 $1 \text{ cm}^3/\text{sec} = 86400 \times 10^{-6} \text{ m}^3/\text{day}$   
 $1 \text{ atm.sec}^2/gm = 101.325 \times 10^6 \text{ m}^{-1}$   
 $1 \text{ atm.cm}^3/\text{gmmole} \cdot K = 0.101325 \text{ kPa.m}^3/\text{kgmole} \cdot K$   
 $1 \text{ atm.sec}^3.\text{cm}^2/\text{gm}^2 = 101.325 \times 10^5 \text{ msec/kg}$

## 1. INTRODUCTION

Considerable effort has been made to explain the phenomenon of gas flow through porous media, using Darcy's law as a fundamental equation. Problems encountered in interpretation and understanding of this phenomenon brought some necessary modifications to this law. These include a correction for gas slippage at lower flow rates, using Klinkenberg's equation. For increasing inertial effects due to higher flow rates, modification has been made by using Forchheimer's quadratic equation.

This study is an attempt to explain some of the observations made by both Piplapure(1) and Senturk(2) on Klinkenberg and visco-inertial plots at higher flow rates and pressures. They were of the general opinion that inadequate confinement pressure could be contributive to some of the observed deviations. Senturk, in addition, felt that these deviations are related to the production method of experimentation. The existing model flow equation is due to Piplapure who pointed out that it is a simplified form of a more complete equation.

It becomes, therefore, necessary that the fundamental theory of steady state radial gas flow through porous media be reviewed with the following objectives.

### 1.1 Objectives

The main objective is to attempt an explanation of the deviations observed on the Klinkenberg and visco-inertial plots. To effect this, experimentation is done

1. with different flow methods,
2. to span both lower and higher flow rates and pressures,
3. to study the significance of both the flow methods and hysteresis on gas flow behavior.

Secondly, the effect of confinement pressure is to be studied by analysing the stress and strain on the core sample as flow progresses.

Thirdly, there is need for reassessing the existing visco-inertial flow model equation and reviewing it for higher, increasing flow rates and pressures. In order to accomplish this, it may be necessary to study the applicability of both Forchheimer's quadratic and cubic equations to the flow data.

## 2. LITERATURE REVIEW

### 2.1 Viscous Flow Equations

As quoted by Hubbert(3), Darcy(4) determined that when water flows vertically downwards through a packed sand filter, the volumetric flux density is found to be proportional to the differential pressure gradient. This relationship, which subsequently became known as Darcy's law, is expressed by the following equation:

$$q = -K \frac{h_2 - h_1}{L} \quad (2.1)$$

where K is the constant coefficient which depends on the permeability of the sand. Numerous attempts have been made to theoretically derive this law using variables with more general physical meaning.

In 1956, Hall(5) derived the general form of Darcy's law for non-isotropic porous media from Newton's basic laws of motion and viscosity. Assuming negligible forces of inertia in the system, and continuous liquid films, he obtained the following equation

$$q = -\left(\frac{1}{\mu}\right) K \cdot (\vec{\nabla}p + \rho \vec{\nabla}gz) \quad (2.2)$$

by using a representative volume element. The permeability, K, is considered to be a tensor and gz is the potential of the gravitational force.

Hubbert(6) used the Navier-Stokes equations of motion of viscous fluids to derive Darcy's law for macroscopically homogeneous and isotropic porous medium in the form of:

$$q = Nd^2(\rho/\mu) [\vec{g} - \frac{1}{\rho} \text{grad } P] \quad (2.3)$$

This is preceded by his earlier work in 1940.(7)

Irmay(8), in 1958, used both a hydraulic model and statistical methods on the Navier-Stokes equations of motion to study a one-dimensional flow of an incompressible fluid through a homogeneous and isotropic medium. At low values of Reynolds number, he derived Darcy's law in the form of:

$$q = KJ \quad (2.4)$$

where

$$K = \frac{1}{a} = \frac{gd_m^2}{Sv} \cdot \frac{\phi^3}{(1-\phi)^2} \quad (2.5)$$

He further emphasized that Darcy flow through porous media is such that the total kinetic energy of flow, which is proportional to the square of the flow rate, is minimum.

The book by Scheidegger(9) gives a detailed review of various theories, models, and derivations of Darcy's law. This law is used extensively to describe the flow of fluids through porous media for the viscous laminar flow regime.

### 2.1.1 Gas Slippage Phenomena

Flowing gas through porous media, especially at lower pressures, yields higher permeability values than those values obtained by flowing liquid. From theory and through extensive experimentation, Klinkenberg(10) in 1941 showed this to be due to gas slippage which occurs at the fluid-solid interface. He observed that the permeability to a gas

- a) is a function of the mean free path of the gas molecules.
- b) depends on pressure, temperature and the nature of the gas.
- c) is both dependent on the property of the porous rock and on the mean pressure at which the gas flows.

Darcy's law, for horizontal plane radial flow, and Klinkenberg's equation are, respectively:

$$-\frac{dp}{dr} = \frac{\mu}{k} q \quad (2.6)$$

$$ka = k \left(1 + \frac{b}{p_m}\right) \quad (2.7)$$

where

b = Slippage coefficient, expressed as

$$b = \frac{4c\lambda}{r} P_m \quad (2.8)$$

Combination of Equations 2.6 and 2.7 shows that both the viscous and the slippage terms are active in Darcy gas flow.

The work of Klinkenberg opened up more investigations to determine the validity and theory of gas slippage. According to this equation, a plot of  $k_a$  versus  $1/P_m$ , usually called the Klinkenberg plot, should be linear with intercept  $K$  and slope  $bK$ .

Calhoun and Yuster(11) in 1946 validated Klinkenberg's equation but pointed out that the relationship between the apparent permeability,  $k_a$ , and the reciprocal of the mean pressure is not exactly linear. They concluded that permeability measurements at different pressure differentials, as stated by Klinkenberg, gave the same value of permeability as long as the mean pressure is constant.

Yuster(12), made the following observations:

1. The plot of apparent gas permeability against the reciprocal of the mean pressure is usually linear, but appreciable curvature at high mean pressures may be noticeable.
2. Using several gases on one core for the Klinkenberg plot, all the lines extrapolated to the same point at infinite pressure.
3. Apparent gas permeability increases with increase in

temperature, but its value at an infinite mean pressure is essentially independent of temperature.

Using the basic concept of kinetic theory of gases, Rose(13) validated Klinkenberg's interpretation of gas slip flow phenomena. He expressed the mean free path of the flowing gas, measured at mean pressures, in terms of temperature,T, gas constant,R, and molecular weight,M, of the gas with the following relation:

$$\lambda = 2.13 \frac{\mu}{P_m} \sqrt{\frac{RT}{M}} \quad (2.9)$$

By statistical averaging, Heid, McMahon, Nielsen, and Yuster(14) obtained a relationship between the absolute gas permeability,k, and the gas slippage coefficient,b, as:

$$b = 0.777k^{-0.38} \quad (2.10)$$

where b is in atmospheres and k is in millidarcies. They observed that, at higher permeabilities, small errors in permeability measurements can cause large errors in the values of b obtained, and unusually low porosity tends to give lower than average values of b.

Dranchuk and Sadiq(15) observed that permeability obtained in a Klinkenberg extrapolation technique is invalid if the data used are not taken in the viscous flow region. They noticed a departure from linearity at high mean pressures on a Klinkenberg plot. Subsequently, they

suggested that a back-pressure curve be plotted first on log-log paper in order to delineate the viscous and the visco-inertial flow regions.

Jones(16) obtained the following relation for gas slip coefficient,b(in psi), in terms of permeability,k(in md):

$$b = 6.9k^{-0.36} \quad (2.11)$$

Casse and Ramey,Jr.(17) observed that the slip coefficient varies linearly with temperature.

### 2.1.2 Real Gas Pseudo-Pressure

The dependence of the gas compressibility factor and viscosity on pressure was studied by Al-Hussainy, Ramey and Crawford(18). They developed the real gas pseudo-pressure as:

$$m(P) = 2 \int_{P_b}^P \frac{P}{\mu(P)Z(P)} dP \quad (2.12)$$

where  $P_b$  is the base pressure. Using this potential, they proposed the following equation for steady state, isothermal, radial gas flow in the viscous region:

$$q = \frac{\pi k h T_0 [m(r_e) - m(r_w)]}{T P_0 \ln \frac{r_e}{r_w}} \quad (2.13)$$

Mackett(19) considered the isothermal variation of compressibility factor and viscosity with pressure at their

reduced conditions with the following integral

$$XKI = \int_0^{P_r} \frac{P_r}{Z\mu_r} dP_r \quad (2.14)$$

He then concluded that under the experimental conditions in which he worked, the values obtained by using average fluid properties do not differ very appreciably from the values obtained by using the above integral.

## 2.2 Reynolds Number Criterion

Increasing both flow rate and flowing pressure leads to a deviation from Darcy's law, and this leads, subsequently, to departure from laminar flow. Earlier investigators(20,21,22) attributed this phenomenon to turbulent flow. Recent reasoning, which has been facilitated mainly by the study of Navier-Stokes equations of motion, attribute it to inertial effects(23,24).

Wright(25) in 1968 explained that, as flow rate increases, deviation from Darcy's law is initially due to inertial effects and at higher flow rates turbulent effects may be observed. He designated the flow regimes as: viscous, steady-inertial, transition to turbulence and full turbulence. Using this explanation, the transition from purely laminar, Darcy-type flow to fully turbulent flow, covers a wide range of flow rates.

For pipe flow, the Reynolds number is used to determine the onset of turbulence and the departure from laminar flow.

Applying this concept to flow through porous media, a need arises for a suitable cut-off value of Reynolds number which characterizes the departure from both Darcy's law and laminar flow.

Hubbert(26), using unconsolidated porous medium, came to the following conclusions:

1. For deviation from Darcy's law, Reynolds number is greater than 1.
2. Darcy's law fails when velocity is great enough to make the inertial force significant.
3. Incidence of turbulence in Darcy flow occurs at much higher velocities corresponding to very high Reynolds numbers.

Subsequently, he defined the Reynolds number as the ratio of the inertial forces to the viscous drag forces.

Linquist(27) in 1933 concluded that Darcy's law is true for Reynolds number up to 4 and, for Reynolds number between 4 and 180, he postulated the equation

$$fRe = 40Re + 2500 \quad (2.15)$$

On the basis of his results, King(28) in 1940 concluded that Darcy's law holds for Reynolds number up to 10. For Reynolds number between 10 and 300, he postulated

$$f = 94/Re^{0.16} \quad (2.16)$$

Collins(29) pointed out that, in sands and sandstones, the transition from laminar to turbulent flow occurs gradually at Reynolds numbers between one and ten. He added that true turbulence occurs only at higher Reynolds numbers.

Scheidegger(30) reviewed the correlations of Reynolds number and friction factors with flow regimes. He noted that there are two critical Reynolds numbers at which the flow regime changes, even though they may not be universally defined. The first change occurs when inertial effects in laminar flow become important. The second change signals real turbulence. He further stated that correlations are valid, and useful, only for application to particular systems. Caution must therefore be exercised if they are to be applied to other conditions for which they were not originally obtained.

### 2.3 Forchheimer's Equations

Forchheimer(31), in 1901, suggested that Darcy's law be modified for high flow rates by including a second-order term in the flow rate as follows:

$$-\frac{dP}{dx} = aq + bq^2 \quad (2.17)$$

where  $a$  and  $b$  are polynomial constants. The extra term takes care of the increasing inertial effect.

By dimensional considerations, Green,Jr. and Duwez(32) observed that for a low Reynolds number, corresponding to a

low velocity, the following relation is obtained:

$$-\frac{dp}{dx} = \text{Const} \cdot \frac{\mu v}{\delta^2} \quad (2.18)$$

Identifying this equation as Darcy's law, they obtained the following relation for higher velocity:

$$-\frac{dp}{dx} = \text{Const} \cdot \frac{\rho v^2}{\delta} \quad (2.19)$$

They reasoned that

- a) the losses due to the inertia of the fluid become progressively more important with increasing velocity;
- b) the gradual transition from Darcy flow regime is marked by losses due to both the viscous shear in creeping flow, and to inertial effects.

Therefore, they combined the two equations to obtain a Forchheimer-type of equation

$$\frac{dp}{dx} = \alpha \mu v + \beta \rho v^2 \quad (2.20)$$

Beta, in this equation, was previously known as the turbulence factor, but is now known as the inertial resistance coefficient; Fb. Kolada(33) expressed it as:

$$F_b = \frac{7.56 \times 10^8}{\phi^{1.67} k^{1.12}} \quad (2.21)$$

Senturk(34) concluded from his studies that this parameter

is both a property of the flowing gas and of the porous medium.

Irmay(35) derived Forchheimer's equation for a one dimensional steady-state flow through a homogeneous isotropic medium of average grain diameter as:

$$J = aq + bq^2 \quad (2.22)$$

where

$$b = \frac{s'(1-\phi)^2}{\phi^3} \cdot \frac{1}{gd_m^2} \quad (2.23)$$

He stated that while coefficient, a, depends on both viscosity and temperature, coefficient, b, does not. For Reynolds number greater than 100, the flow becomes turbulent.

Schiedegger(36), reported that following further research, Forcheimer added a third-order term thus:

$$-\frac{dp}{dx} = aq + bq^2 + cq^3 \quad (2.24)$$

where a, b, and c are polynomial constants. He added that this equation fitted experimental data better for higher flow rates.

Basak and Soni(37), reviewing the flow relations due to both Darcy and Forchheimer, observed that while Darcy's law failed beyond Reynolds number one, the velocity gradient

response is adequately represented by Forchheimer's relations. Analysing high-velocity gas flow through porous media, Firoozabadi and Katz(38) noted that there are cases where Forchheimer's quadratic equation failed to correctly fit the experimental data. They therefore suggested for such cases the use of Forchheimer's cubic equation in the following form

$$-\frac{dP}{dx} = \frac{\mu}{k} q + \beta \rho q^2 + \gamma \rho^2 q^3 \quad (2.25)$$

#### 2.4 Radial Visco-Inertial Flow Equations

Elenbaas and Katz(39) developed a radial turbulent gas flow equation, by assuming average values of gas viscosity, temperature and compressibility factor. For laminar and turbulent flow through unconsolidated particles, they gave, respectively, the following equations:

$$P_2^2 - P_1^2 = 8.12 \times 10^{-4} \frac{\mu Z_a T_a W_s}{D_a^2 \phi^{n-m} h G} \ln \frac{r_2}{r_1} \quad (2.26)$$

$$P_2^2 - P_1^2 = 2.02 \times 10^{-6} \frac{W_s^2 T_a Z_a f}{D_a \phi^n h^2 G} \left( \frac{1}{r_1} - \frac{1}{r_2} \right) \quad (2.27)$$

Furthermore, they observed that a back-pressure curve of  $(Pe^2 - Pw^2)$  versus  $Q$  for a given gas well can be used to designate the different flow regimes present. Deviation of

the slope of the curve at higher flow rates from unity marks the point of deviation from Darcy's law.

Using the linear equation presented by Green and Duwez(40), Cornell and Katz(41) gave a corresponding equation for radial geometry cases:

$$\begin{aligned} P_2^2 - P_1^2 &= 1.5 \times 10^{-11} \frac{\alpha \mu ZT Q_0}{h} \ln \frac{r_2}{r_1} \\ &\quad + 1.09 \times 10^{-13} \frac{\beta M Q_0^2 ZT}{h^2} \left( \frac{1}{r_1} - \frac{1}{r_2} \right) \end{aligned} \quad (2.28)$$

Tek, Coats and Katz(42) combined both Forchheimer's quadratic equation and the modified gas law to obtain the following radial steady state turbulent gas flow equation:

$$\begin{aligned} P_2^2 - P_1^2 &= \frac{1424 \mu ZT Q_0}{hk} \ln \frac{r_2}{r_1} \\ &\quad + \frac{3.161 \times 10^{-12} \beta G Q_0^2 ZT}{h^2} \left( \frac{1}{r_1} - \frac{1}{r_2} \right) \end{aligned} \quad (2.29)$$

By using a force momentum balance on a capillary-orifice model, Crafton(43) obtained a quadratic equation similar to that of Forchheimer:

$$-\frac{dP}{dr} = \left( \frac{C_L R_e}{\pi r \phi d_c} \right) \mu v + \left( \frac{\rho C_d}{2x} + \frac{C_t}{\pi r \phi} \right) v^2 \quad (2.30)$$

where the  $F_b$  term in Forchheimer's equation is given as :

$$F_b = \frac{\rho C_d}{2x} + \frac{C_t}{\pi r \phi} \quad (2.31)$$

$F_b$  = matrix term + "radial controlled" fluid term

Piplapure(44) obtained the following modified equation, similar to Equation 2.29, which expresses the parameters  $k$ ,  $b$ ,  $F_b$  simultaneously:

$$(P_e^2 - P_w^2) + 2b(P_e - P_w) = - \left[ \frac{1424 \mu Z T Q_0}{h k} \ln \frac{r_e}{r_w} + 3.1602 \times 10^{-12} \frac{F_b G Z T Q_0 |Q_0|}{h^2} \left( \frac{1}{r_w} - \frac{1}{r_e} \right) \right] \quad (2.32)$$

He identified them as viscous, slippage, and inertial terms respectively.

In deriving this equation, he transformed the mean pressure term in Klinkenberg's equation into its instantaneous pressure value. Subsequently, he assumed the value

$$\left[ F_b \left( \frac{P_0 m}{R T} \right)^2 \frac{Z R T}{m} \cdot \frac{|Q_0| Q_0}{(2 \pi h)^2} \right] \times \frac{b}{P} \quad (2.33)$$

is sufficiently small so that it can be neglected.

### 3. DEVELOPMENTS FROM PREVIOUS EXPERIMENTS

#### 3.1 Klinkenberg and Visco-Inertial Plots

It is standard practice to study and predict gas flow behavior in a porous medium under viscous and visco-inertial flow conditions by conducting a multiple-point flow test on a sample of the medium. The parameters - absolute gas permeability,  $k$ , gas slippage factor,  $b$ , and inertial resistance coefficient,  $F_b$ , - are evaluated, graphically or numerically, by using the flow equations in data analysis. Equation 2.7 yields the Klinkenberg plot for flow within the viscous region where Darcy's law is valid. The absolute gas permeability is the extrapolated value at infinite mean pressure and the gas slippage coefficient,  $b$ , is obtained from the slope. This plot departs from predicted linearity and curves downwards at higher pressures corresponding to higher flow rates.

##### 3.1.1 Linear Geometry

Equation 2.20 represents the more general form of the second-order Forchheimer equation (Equation 2.17). Cornell and Katz(45) modifying this equation for linear geometry cases, presented it as:

$$\frac{(P_2^2 - P_1^2)Mg_c}{2L e_s G_s ZRT\mu} = \alpha + \beta \frac{G_s}{\mu} \quad (3.1)$$

In this form, a plot of the left hand side versus  $G_s/\mu$

yields the visco-inertial plot. This is a linear plot with the intercept as the reciprocal of the apparent permeability,  $k_a$ , and the slope as the inertial resistance coefficient,  $F_b$ . Dranchuk and Sadiq(46) observed that this plot departs from linearity with a characteristic downward curve at lower flow rates. This, they explained, is due to gas slippage at the gas-solid interface.

From the foregoing, two methods exist for the determination of permeability of a porous medium. While the resulting values from these two methods are expected to be equal, experiments(47,48) have shown them to be different. The visco-inertial plot yields the higher value, and depending on the gas slippage coefficient, the difference can be very large. Dranchuk and Kolada(49) inferred that possibly some or all of the data may be affected by gas slippage. They therefore suggested that, in order to account for gas slippage, Equation 3.1 be modified to:

$$\frac{(P_2^2 - P_1^2)Mg_c}{2L e^{ZRT\mu} G_s} = \frac{1}{K(1 + \frac{b}{P_m})} + \beta \frac{G_s}{\mu} \quad (3.2)$$

or

$$\frac{(1 + \frac{b}{P_m})(P_2^2 - P_1^2)Mg_c}{2L e^{ZRT\mu} G_s} = \frac{1}{k} + \beta \frac{G_s}{\mu} (1 + \frac{b}{P_m}) \quad (3.3)$$

Equation 3.3 suggests a linear plot between the left hand side term and  $(G_s/\mu)(1+b/P_m)$ . The intercept yields the reciprocal of the absolute permeability while the slope

yields the inertial resistance coefficient. This plot, known as the modified visco-inertial plot, requires that the gas slippage coefficient be known first. Dranchuk and Kolada recommended that the value of  $b$  be obtained through trial and error. This is done by repeatedly plotting both the back-pressure and the Klinkenberg plots and checking the values of  $b$  for good agreement. The final reliable value is used in Equation 3.3 to construct a modified visco-inertial plot for all the data points except those which have been found to be in the viscous region.

By expressing the unknown parameters,  $b$ ,  $k$ ,  $F_b$ , simultaneously, as in Equation 3.3, some possible numerical procedure for estimating these parameters may be found. Kolada, using the Newton-Raphson approximation technique, pointed out that this method requires an initial estimate of the parameters and also quasilinearization in the derivatives containing the parameters. He added that convergence is dependent on the goodness of the initial estimates, and on the algorithm used.

### 3.1.2 Radial Geometry

Piplapure(50) replaced both the conventional visco-inertial equation (Equation 3.1) and the modified visco-inertial equation (Equation 3.3), with the following equations:

$$-\left[ \frac{(P_e^2 - P_w^2)h}{1424\mu T Q_0} \right] = \frac{\ln \frac{r_e}{r_w}}{k} + \frac{2.2192 \times 10^{-15} F_b G \left( \frac{1}{r_w} - \frac{1}{r_e} \right) \frac{|Q_0|}{\mu}}{h} \quad (3.4)$$

$$\frac{[(P_e^2 - P_w^2) + 2b(P_e - P_w)]h}{1424\mu T Q_0} = \frac{\ln \frac{r_e}{r_w}}{k} + \frac{2.2192 \times 10^{-15} F_b G \left( \frac{1}{r_w} - \frac{1}{r_e} \right) \frac{|Q_0|}{\mu}}{h} \quad (3.5)$$

Equation 3.5 is not strictly an exact counterpart of Equation 3.3 because, as Piplapure pointed out, it is a simplified form of the complete equation. Nevertheless, the three parameters are simultaneously represented and all the treatments using Equation 3.3 are assumed to apply. Dranchuk and Piplapure(51) indicated that the viscous region can be delineated on the basis of Reynolds number which can be calculated only when  $K$  and  $F_b$  are known.

### 3.2 Anomalies

Some anomalous behavior in both Klinkenberg and visco-inertial plots has been reported. Kolada(52) plotted some reconstituted data of Cornell(53) and noticed that the apparent permeability,  $k_a$ , decreases with mean pressure,  $P_m$ , and flow rate,  $Q_0$ , at lower flow rates, and increases with them at higher flow rates. Piplapure(54) obtained a profile with an upward curve on the Klinkenberg plot, instead of a profile with the usual downward curve. The nature of these plots, he reported, caused less effective prediction of both

permeability and slippage coefficient to be made. Though the runs produced peculiar Klinkenberg plots, he noticed that the resulting visco-inertial plots appeared to be normal in every respect. The runs were taken at constant upstream pressures for varying downstream pressures.

In another set of runs, taken at both constant downstream and upstream pressures, and at higher flow rates, he again reported an upward curve on the Klinkenberg plots. This meant an increase in apparent permeability with increasing mean pressure. He explained that graphical parameter estimation was still possible, but both visco-inertial plots and numerical estimations yielded negative values of the inertial resistance coefficient. Despite this, he added, conventional visco-inertial plots were smooth, and modified visco-inertial plots yielded expected linear plots. He remarked that these peculiarities may be due to the effect of confining pressure.

Similar anomalies on the Klinkenberg plots were observed by Senturk(55) on runs taken at both constant downstream and constant upstream pressures, and at constant predefined net confinement pressures. He stressed that they were found in runs on low permeability cores, and at higher mean flowing gas pressures. These anomalies, he explained, may be due to inadequate definition of the net rock confinement pressure, lack of radial confinement pressures, or the manner in which the flow tests were conducted.

### 3.3 Effect of Confinement Pressure

Senturk investigated more closely the effect of confinement pressure on the parameters. He defined the net confinement pressure in terms of both the hydrostatic rock confinement pressure and mean flowing fluid pressure as:

$$P_{nc} = P_{hc} - P_m \quad (3.6)$$

Subsequently, he concluded that for increasing net rock confinement pressure, absolute permeability decreased, and both gas slippage and inertial resistance coefficients increased.

### 3.4 Evaluating the Anomalies Using the Current Theory

By Equation 2.8, gas slippage phenomenon varies inversely with the mean pressure, and can be expressed as

$$\frac{4c\bar{\lambda}}{r} = \frac{b}{P_m} \quad (3.7)$$

where b, the proportionality constant, is the gas slippage coefficient. In conjunction with Equation 2.7, gas slippage varies directly with apparent gas permeability,  $k_a$ . By standard experimental procedure and from the back-pressure relationship, mean pressure increases as flow rate increases. Therefore, both mean pressure and flow rate increase for decreasing gas slippage and apparent

permeability,  $k_a$ . Similarly, from theory, both mean pressure and flow rate increase with increasing inertial effects.

These considerations are combined in the analysis of three data cases from the literature. The trends of  $k_a$ , gas slippage, and inertial effects, are respectively followed with reference to both the flow rate,  $Q_o$ , and the mean pressure,  $P_m$ . Since  $P_m$  is an independent variable in the Klinkenberg plot, and  $Q_o$  is part of a dependent variable in the visco-inertial plot, evaluation of these trends may be made with respect to the viscous and the visco-inertial regions of both plots.

Case 1 (Constant  $P_w$ , and Decreasing  $P_e$ ):

In Piplapure's data for Sample 2 Run 1, (56),  $P_w$  is held constant at atmospheric pressure while  $P_e$  is steadily decreased at constant confining pressure,  $C_P$ , of 3447.4 kPa. This results in decreasing  $(P_e^2 - P_w^2)$ , decreasing  $P_m$ , and decreasing flow rate,  $Q_o$ . With reference to  $P_m$ ,  $1/P_m$  increases,  $k_a$  increases,  $1/k_a$  decreases, slip increases, and inertial effects decrease. With reference to  $Q_o$ , which decreases,  $k_a$  increases,  $1/k_a$  decreases, slip increases, and inertial effects decrease. The overall trend in both regions of flow of both plots is such that for decreasing  $P_m$  and  $Q_o$ ,  $k_a$  increases, gas slippage increases, and inertial effects decrease. Both plots should yield a profile with a downward curve and a positive slope. Figures 3.1 and 3.2 show, respectively, the Klinkenberg and visco-inertial plots from

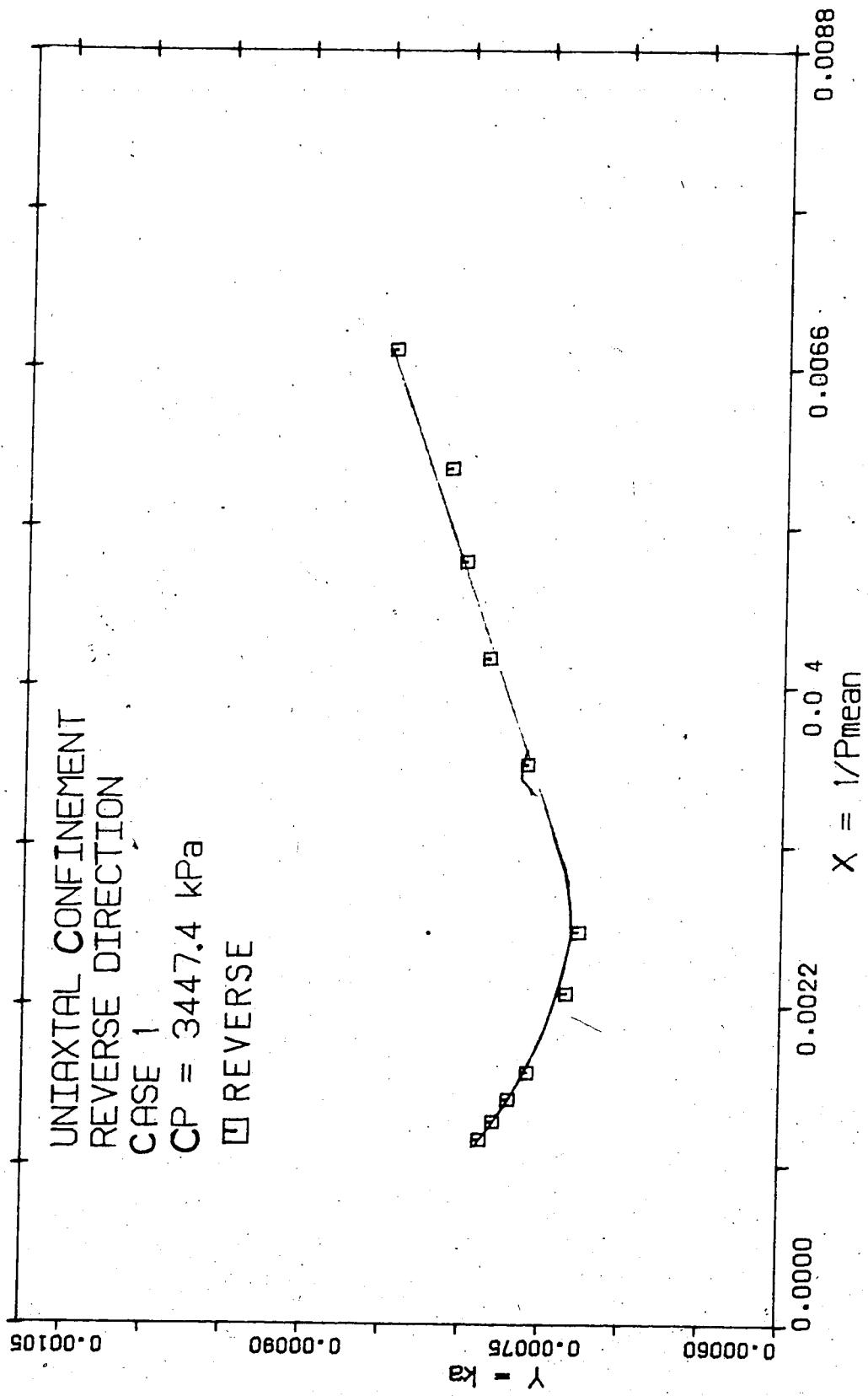


FIGURE 3-1: PIRAPURE'S DATA KLINKENBERG PLOT

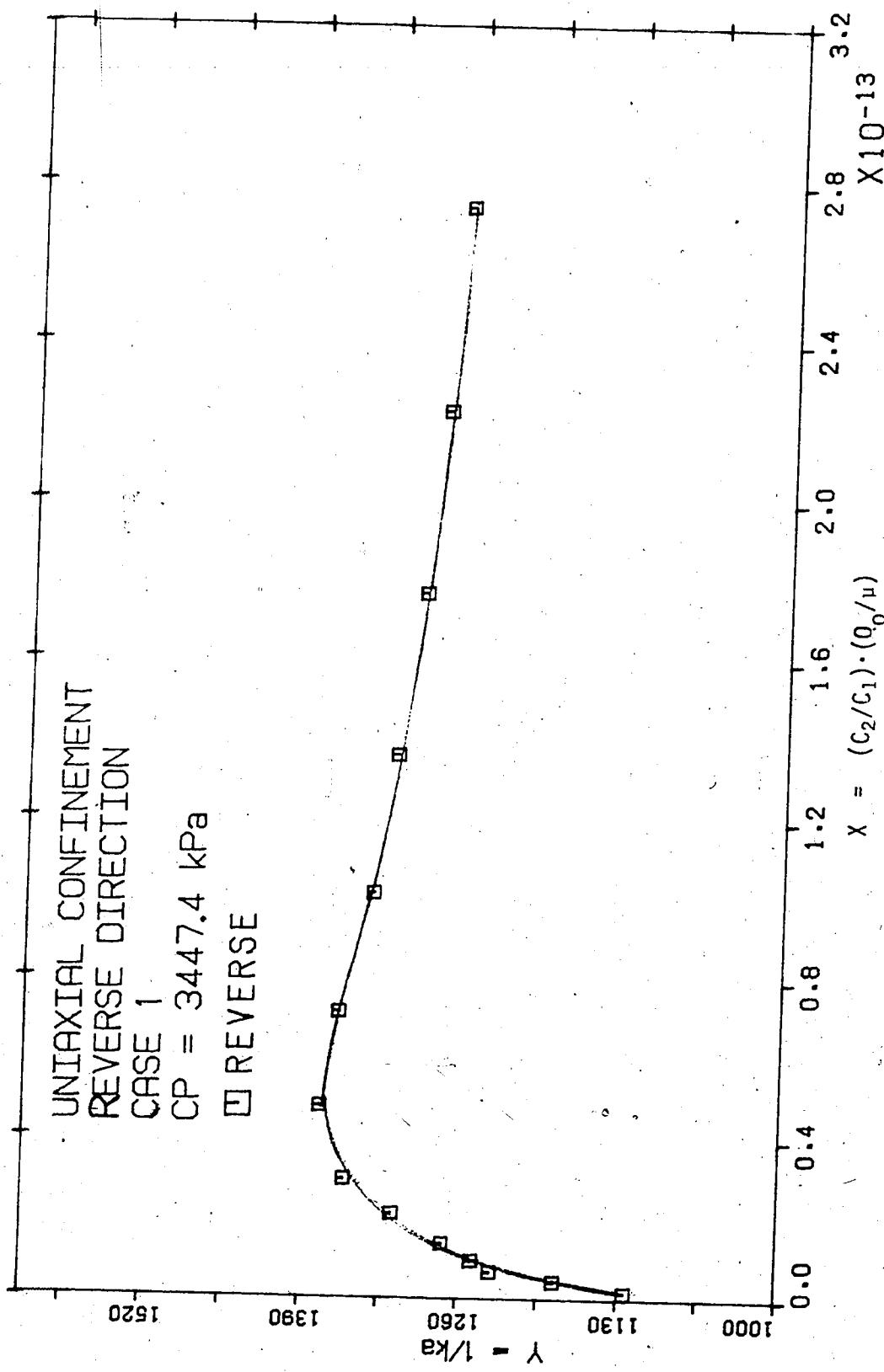


Figure 3-2: PIPLAPURE'S DATA VISCO-INERTIAL PLOT

Piplapure's data. The values of  $k_a$  decrease initially, and then increase for decreasing  $P_m$  and  $Q_o$ . This gives the Klinkenberg plot a concave upward profile while the visco-inertial plot gives a concave downward profile. Figure 3.3, which is the back-pressure plot, shows a decreasing slope at higher flow rates instead of an increasing slope.

Case 2 (Constant  $P_w$  and Increasing  $P_e$ ):

This is Senturk's data for Sample SS-1-A, Run 2,(57) at constant confining pressure,  $C_P$ , of 4136.9 kPa. The value of  $P_w$  is held constant at atmospheric pressure while  $P_e$  is steadily increased. This results in increasing  $(P_e^2 - P_w^2)$ , increasing  $P_m$ , and increasing  $Q_o$ . With reference to  $P_m$ ,  $1/P_m$  decreases,  $k_a$  decreases,  $1/k_a$  increases, slip decreases, and inertial effects increase. With reference to  $Q_o$ , which increases,  $k_a$  decreases,  $1/k_a$  increases, slip decreases, and inertial effects increase. Therefore, for increasing  $P_m$  and  $Q_o$  the overall trend in both regions of both plots is such that  $k_a$  decreases, gas slippage decreases, and inertial effects increase. This acts in the opposite sense to the expected trend in Case 1, even though the profiles should be similar. Figure 3.4 gives a trend similar to that of Figure 3.1 while Figure 3.5 gives a trend of decreasing  $k_a$  with increasing  $Q_o$  as has been predicted. The back-pressure plot in Figure 3.6 gives a lower slope at higher flow rates instead of the expected higher value.

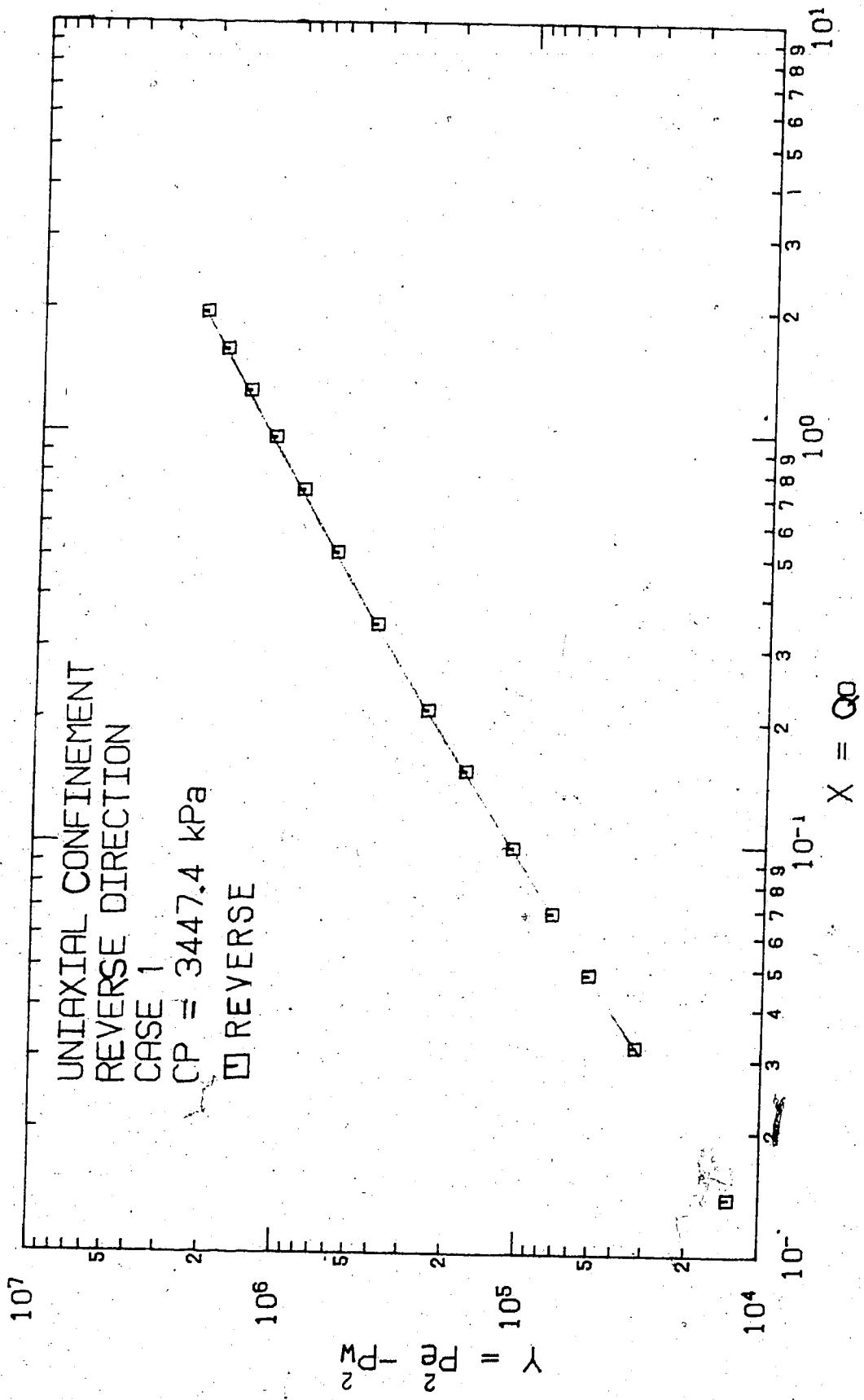


Figure 3-3: PIPLAPURE'S DATA BACK PRESSURE PLOT

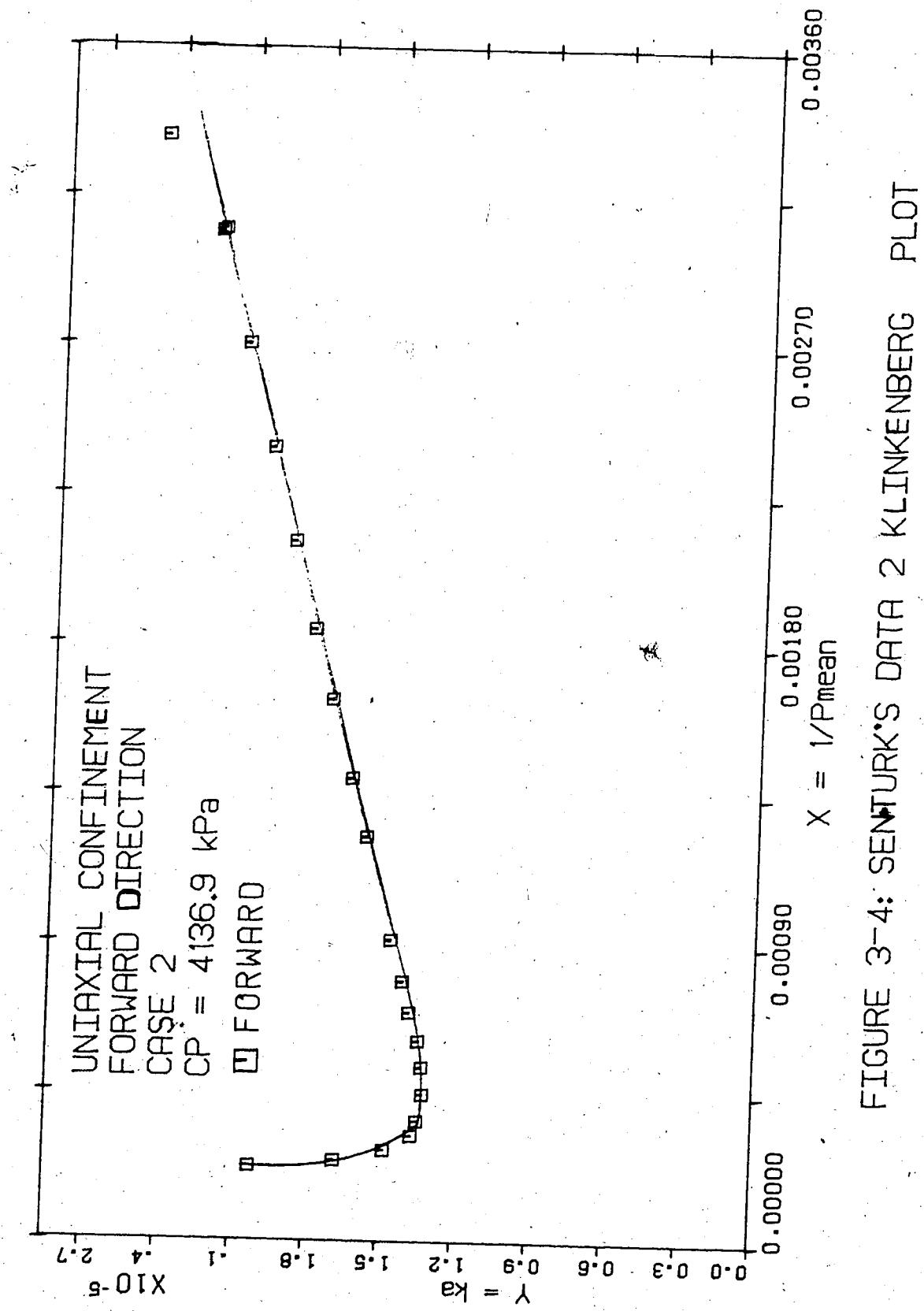


FIGURE 3-4: SENTURK'S DATA 2 KLINKENBERG PLOT

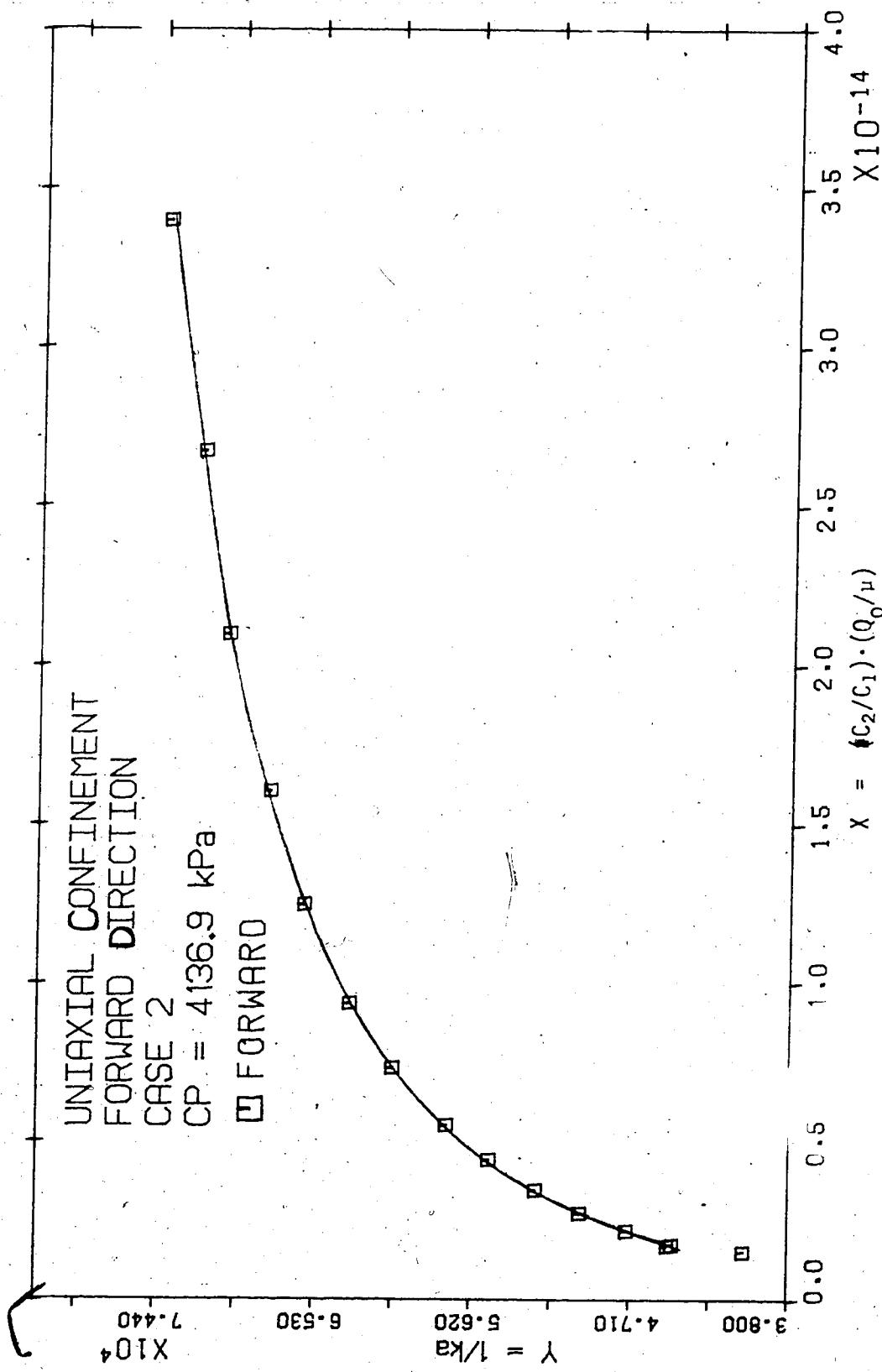


Figure 3.5: SENTURK'S DATA < ISCO-INERTIAL PLOT

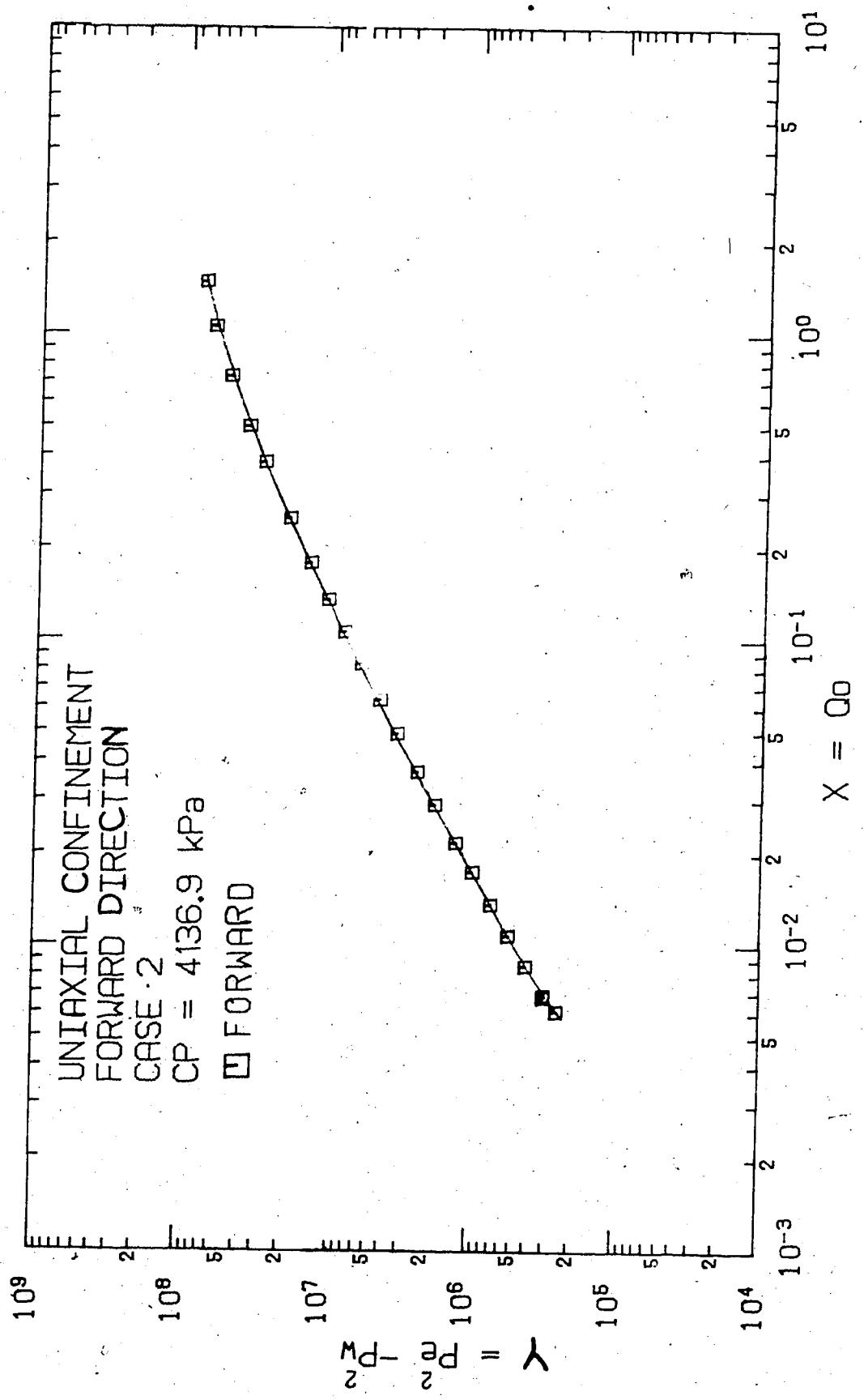


Figure 3-6: SENTURK'S DATA 2 BACK PRESSURE PLOT

Case 3 (Increasing Pw and Constant Pe):

This is taken from Senturk's data for Sample LS1, Run 5, (58) at confining pressure, CP, of 5515.8 kPa. The external pressure, Pe, is held constant at 2068.43 kPa and Pw is steadily increased. This results in decreasing  $(Pe^2 - Pw^2)$ , decreasing Qo, and increasing Pm. With reference to Pm,  $1/Pm$  decreases, ka decreases,  $1/ka$  increases, slip decreases, and inertial effects increase. With reference to Qo, which decreases, ka increases,  $1/ka$  decreases, slip increases, and inertial effects decrease. Since the flow condition results in increasing Pm, but decreasing Qo, the trends of ka, gas slippage, and inertial effects in both regions of both plots are expected to be opposite. Depending on the magnitude of the flow rates and pressures, the produced profile may be such that ka decreases with decreasing Qo and Pm. Figure 3.7 shows a decreasing ka for decreasing Qo and Pm, while Figure 3.8 shows a decreasing ka for increasing Qo and Pm. Figure 3.9 gives a back-pressure plot with higher slope at higher flow rate.

If, in Case 3, Pw decreases instead of increasing, for constant Pe,  $(Pe^2 - Pw^2)$  increases, Qo increases, and Pm decreases. This new case, designated as Case 4, suggests that the trends of ka, gas slippage, and inertial effects in both regions of both plots are opposite to those obtained in Case 3. Table 3.1 summarizes the effect of the flow methods on flow profiles for the cases considered.

The foregoing analysis shows that:

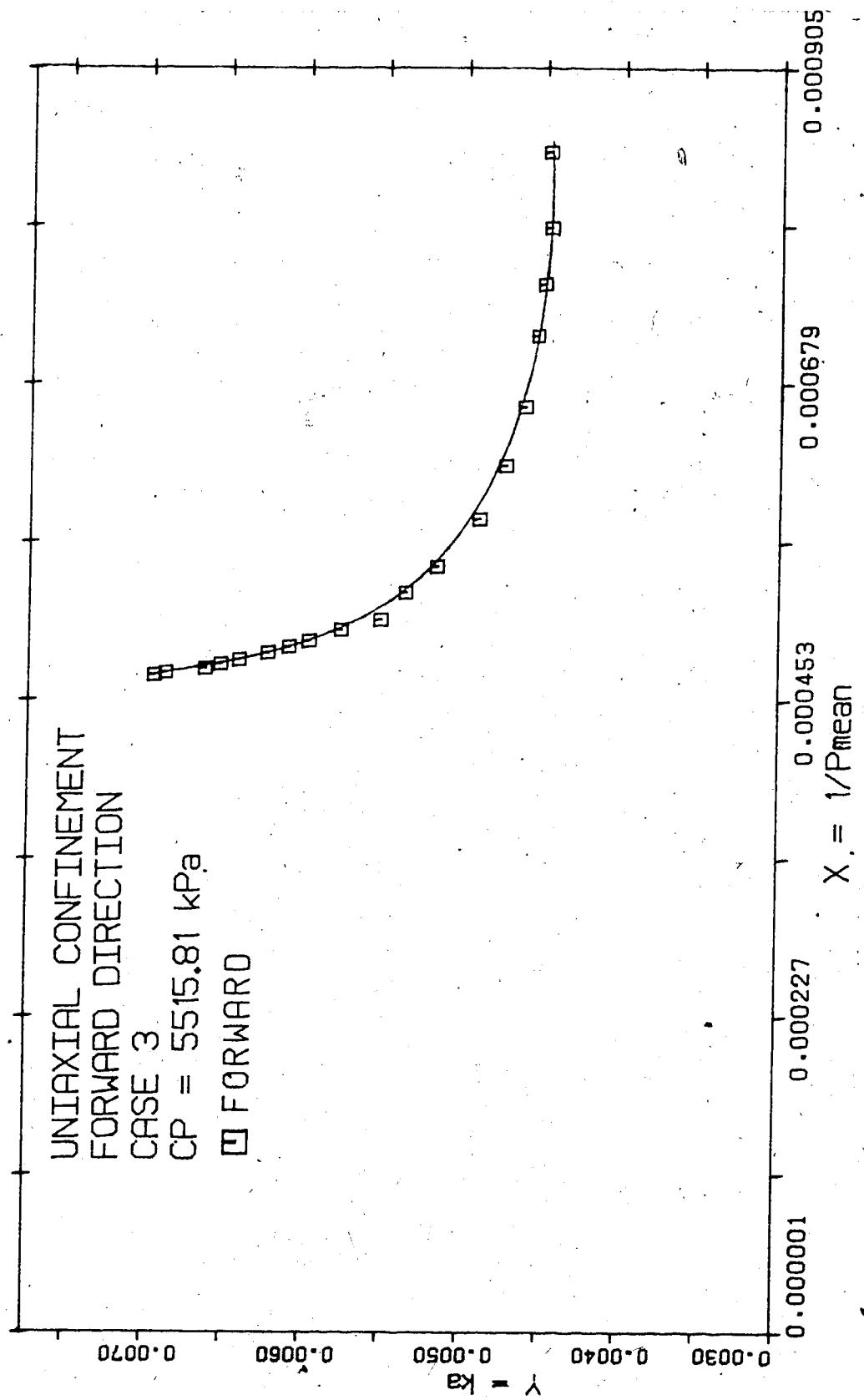


FIGURE 3-7: SENTURK'S DATA 3 KLINKENBERG PLOT

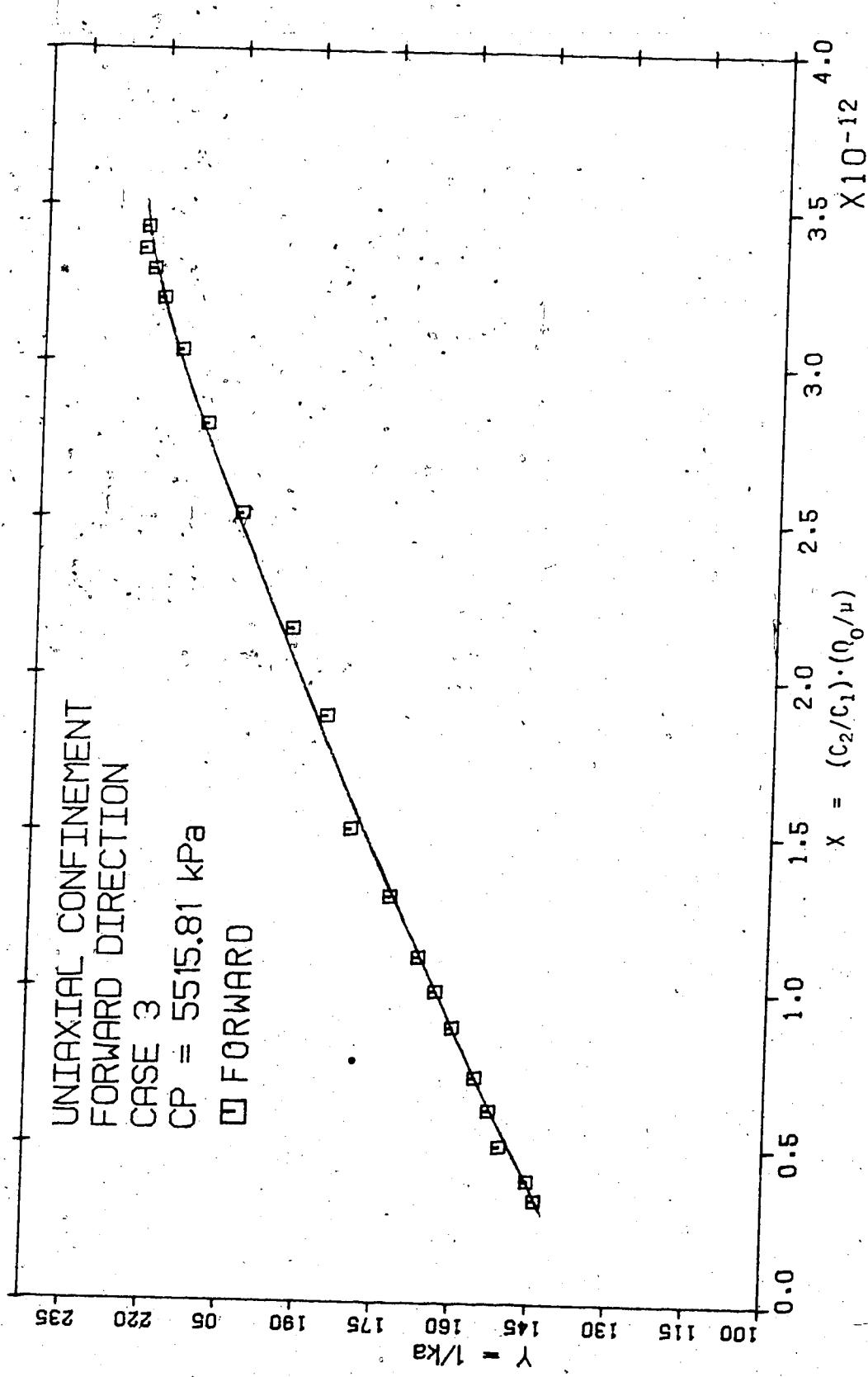


Figure 3-8: SENTURK'S DATA 3 VISCO-INERTIAL PLOT

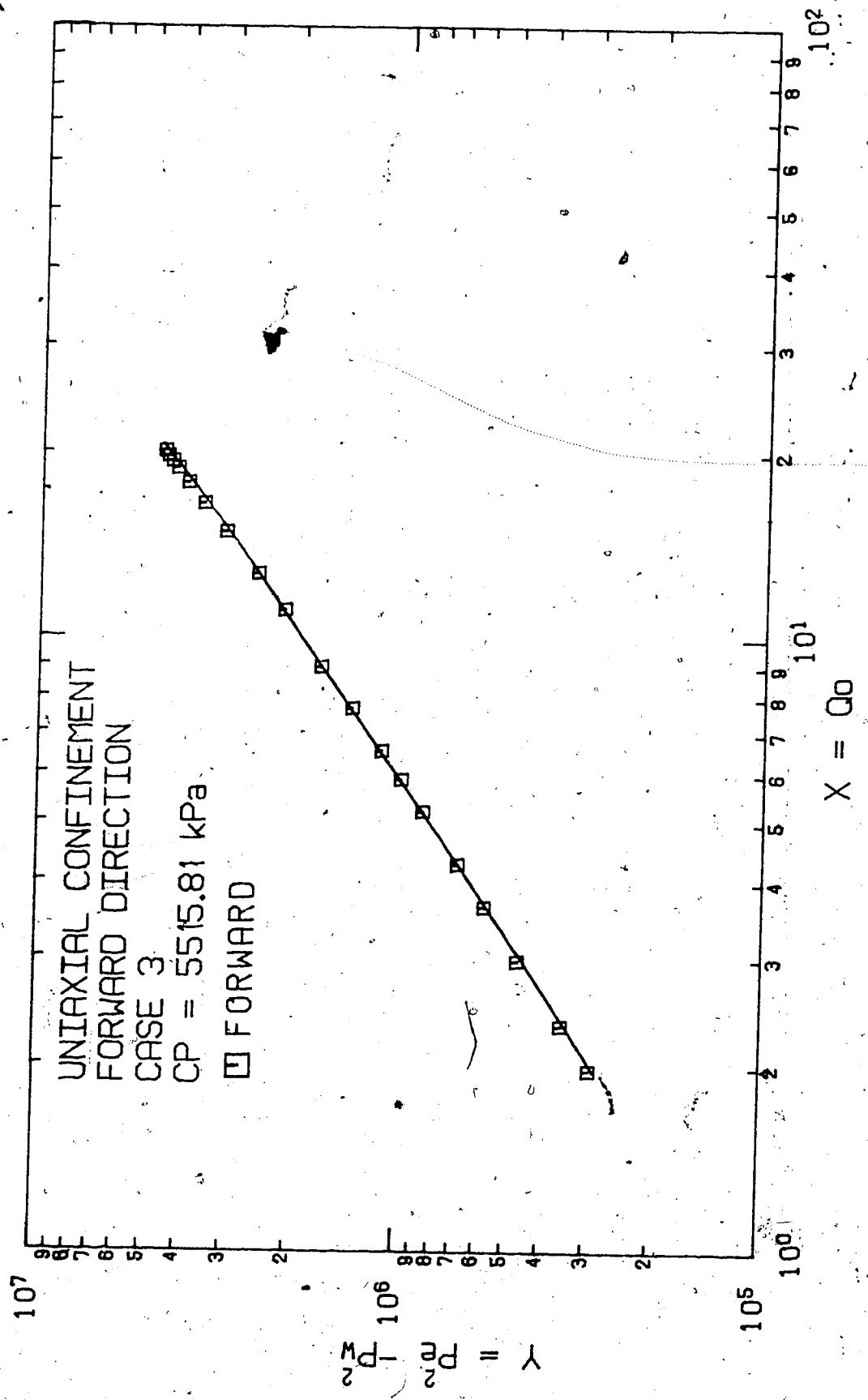


Figure 3-9: SENTURK'S DATA 3 BACK PRESSURE PLOT

TABLE 3-1: EFFECT OF FLOW METHODS ON FLOW PROFILES

| PARAMETER                         | CASE 1                 | CASE 2                 | CASE 3                 | CASE 4                 |
|-----------------------------------|------------------------|------------------------|------------------------|------------------------|
| Flow Method                       | Pw = Const<br>Pe = Dec | Pw = Const<br>Pe = Inc | Pe = Const<br>Pw = Inc | Pe = Const<br>Pw = Dec |
| Pe <sup>2</sup> - Pw <sup>2</sup> | Decrease               | Increase               | Decrease               | Increase               |
| Qo                                | Decrease               | Increase               | Decrease               | Increase               |
| Pm                                | Decrease               | Increase               | Increase               | Decrease               |
| 1/Pm                              | Increase               | Decrease               | Decrease               | Increase               |
|                                   | Using:<br>Pm : Qo      | Using:<br>Pm : Qo      | Using:<br>Pm : Qo      | Using:<br>Pm : Qo      |
| ka                                | Inc : Inc              | Dec : Dec              | Dec : Inc              | Inc : Dec              |
| 1/ka                              | Dec : Dec              | Inc : Inc              | Inc : Dec              | Dec : Inc              |
| Slip                              | Inc : Inc              | Dec : Dec              | Dec : Inc              | Inc : Dec              |
| Inertia                           | Dec : Dec              | Inc : Inc              | Inc : Dec              | Dec : Inc              |

where: Inc = Increase, Dec = Decrease, Const = Constant

Slip = Gas Slippage, Inertia = Inertial Effects

1. the flow method and the direction of flow influence the flow behavior.
2. abnormal behavior at higher flow rates may be due to
  - a. normal behavior which depends on the flow method as depicted in Case 3.
  - b. lower (Pe<sup>2</sup> - Pw<sup>2</sup>) at higher flow rates as in Cases 1 and 2 and depicted especially in the back-pressure plots.

The back-pressure flow relationship may be expressed as:

$$(P_e^2 - P_w^2) = 2P_m(P_e - P_w) \propto Q_0 \quad (3.8)$$

This means that the flow rate is a function of both the mean pressure and the pressure drawdown. Keeping either of them constant, the variation of the other with respect to the flow rate can be analysed. Both pressure terms contribute to the magnitude of the flow rate, but the drawdown pressure, in addition, gives both direction and the driving force. If the drawdown pressure is greater than the mean pressure, increased compaction of the sample may result. If this condition continues without change, this may possibly lead to cracking or fracturing which in itself leads to increasing values of  $k_a$  for increasing mean pressures.

Therefore, investigation of these problems may be carried out by:

1. obtaining experimental data which cover both lower and higher flow rates and pressures, and by studying the significance of flow method and hysteresis on gas flow behavior,
2. reevaluating the confinement and fluid pressure relationship in terms of the stress-strain relationship generated during the flowing procedure under uniaxial confinement,
3. reassessing the existing visco-inertial model flow

equation. Since these problems have been noticed at higher flow rates and pressures, it becomes necessary to explore the possibility of using higher order equations.

#### 4. STEADY STATE GAS FLOW MODEL AND PARAMETER ESTIMATION

The use of Forchheimer's quadratic equation, as discussed in the preceding pages, has facilitated the understanding of visco-inertial gas flow through porous media. It has been derived in Appendix A from consideration of both the Navier-Stokes equations of motion and dimensional analysis for laminar flow with negligible Reynolds stresses. For higher flow rates, the flow condition changes, and local irregularities become increasingly significant in the laminar mainstream such that Reynolds stresses begin to form.

As an aid to understanding the aspect of this study which involve higher flow rates and pressures, it has become necessary to derive the third order equation of Forchheimer as shown in Equation A-23 for 1-dimensional, linear, horizontal, incompressible fluid flow under the conditions of no-slip. The cubic term in this equation is regarded as a modification to Forchheimer's quadratic equation for data obtained at higher flow rates and pressures under the conditions of no-slip. It has been derived in Appendix A by considering both the kinetic energy equation of mean flow and dimensional analysis.

#### 4.1 Derivation of a General Radial Gas Flow Equation

For 1-dimensional, horizontal, steady state, plane radial incompressible flow, Equation A-23 is written as

$$-\frac{dp}{dr} = \frac{\mu}{k} q + F_b p q^2 + \gamma p^2 q^3 \quad (4.1)$$

where the linear distance  $x$  is replaced by the radial distance  $r$ . It is adapted to gas flow with the following basic assumptions:

1. Flow is steady, compressible and horizontal such that
  - a. gravitational forces are neglected,
  - b. density is not constant but depends on temperature and pressure as given by the equation of state for real gases

$$\rho = \frac{PM}{ZRT} = \frac{28.96GP}{ZRT} \quad (4.2)$$

- c. the mean momentum of fluctuations is zero,
- d. Kinetic energy equation of mean flow is valid.
2. Slippage condition exists and the permeability term in Equation 4.1 is changed to its apparent value,  $k_a$ . Slippage effects are corrected by using the Klinkenberg equation (Equation 2.7).
3. Continuity equation for steady state condition is given as:

$$\rho_0 q_0 = \rho q = \text{Constant} \quad (4.3)$$

4. All flows are geometrically similar so that unique velocity and length scales can be obtained which satisfy them geometrically.
- a. The system is of plane radial geometry with uniform thickness.
  - b. For 1-dimensional, horizontal cases, the flow rate per cross-sectional area at any radial distance,  $r$ , from a source or sink, is given as:

$$q = \frac{Q}{A} = \frac{Q}{2\pi rh} \quad (4.4)$$

5. Porous medium is both homogeneous and isotropic.
6. Both velocity and velocity fluctuation scales can be represented by a single unique velocity scale.
7. Viscosity, compressibility factor, and density are obtained at mean temperatures and pressures.
8. A single phase gas with constant composition saturates and flows through the porous medium.

Equation 4.1 can therefore be written as:

$$-\frac{dp}{dr} = \frac{\mu}{k_a} q + F_b \rho q^2 + \gamma \rho^2 q^3 \quad (4.5)$$

which, when multiplied across with the density term, yields

$$-\rho \frac{dp}{dr} = \frac{\mu}{k_a} (\rho q) + F_b (\rho q)^2 + \gamma (\rho q)^3 \quad (4.6)$$

By combining Equations 4.2 and 4.4 according to the continuity requirement of Equation 4.3, and substituting them into Equation 4.6, the following expression, after further simplification, is obtained:

$$\begin{aligned} -\frac{1}{2} \frac{dp^2}{dr} &= \frac{\mu T Z P_0 Q_0}{2\pi k_a h T_0} \cdot \frac{1}{r} + \frac{28.96 G P_0^2 Q_0^2 Z T F_b}{R T_0^2 (2\pi h)^2 r^2} \\ &+ \frac{28.96^2 G^2 P_0^3 Q_0^3 Z T Y}{R^2 T_0^3 (2\pi h)^3 r^3} \end{aligned} \quad (4.7)$$

Since  $P_m$  is inversely proportional to  $k_a$ , and for any given set of boundary conditions  $P_m$  is known,  $k_a$  is also known and can be regarded as being constant. Hence, integrating between  $P_e$  and  $P_w$ , and  $r_e$  and  $r_w$ , the following equation is obtained:

$$\begin{aligned} (P_e^2 - P_w^2) &= \frac{\mu T Z P_0 Q_0}{\pi k_a h T_0} \ln \frac{r_e}{r_w} \\ &+ \frac{28.96 G P_0^2 Q_0^2 Z T F_b}{R T_0^2 (2\pi h)^2} \left( \frac{1}{r_w} - \frac{1}{r_e} \right) \\ &+ \frac{28.96^2 G^2 P_0^3 Q_0^3 Z T Y}{R^2 T_0^3 (4\pi^2) h^3} \left( \frac{1}{r_w^2} - \frac{1}{r_e^2} \right) \end{aligned} \quad (4.8)$$

By convention, flow is taken as positive in the direction of increasing radius, and negative in the direction of decreasing radius. For production, flow is in the direction of decreasing radius such that  $(P_e^2 - P_w^2)$  is always positive. Since production is being considered, it is convenient to reverse the convention so that flow is taken as positive. Therefore, Equation 4.8 becomes

$$\begin{aligned} P_e^2 - P_w^2 = & \frac{\mu Z T P_0 Q_0}{\pi k_a h T_0} \ln \frac{r_e}{r_w} + \frac{28.96 G P_0^2 Q_0^2 Z T F_b}{R T_0^2 (2\pi) h^2} \left( \frac{1}{r_w} - \frac{1}{r_e} \right) \\ & + \frac{28.96^2 G^2 P_0^3 Q_0^3 \gamma Z T}{R^2 T_0^3 (4\pi^2) h^3} \left( \frac{1}{r_w^2} - \frac{1}{r_e^2} \right) \end{aligned} \quad (4.9)$$

This conforms to the the convention commonly found in the literature. Equation 4.9 is valid in cgs units where

$P_e, P_w, P_0$ , are in atmospheres pressure,

$k_a, \mu$ , are in darcy and centipoise respectively,

$h, r_e, r_w$ , are in cm;

$Q_0$  is in  $\text{cm}^3/\text{sec}$ . (at standard conditions),

$T_0, T$  are in °K,

$F_b$  is in  $\text{atm.sec}^2/\text{gm} = \text{cm}^{-1}$ ,

$\gamma$  is in  $\text{atm.cm}^2.\text{sec}^3/\text{gm}^2 = \text{cmsec/gm}$

R = Universal gas constant, in  $\text{atm.cm}^3/\text{gmmole : K}$ .

In SI units, it becomes

$$\begin{aligned} P_e^2 - P_w^2 = & 1.2955 \times 10^{-3} \frac{\mu T Z Q_0}{k_a h} \ln \frac{r_e}{r_w} \\ & + 2.9229 \times 10^{-15} \frac{G Z T Q_0^2 F_b}{h^2} \left( \frac{1}{r_w} - \frac{1}{r_e} \right) \\ & + 6.5946 \times 10^{-21} \frac{G^2 Z T Q_0^3 \gamma}{h^3} \left( \frac{1}{r_w^2} - \frac{1}{r_e^2} \right) \end{aligned} \quad (4.10)$$

where

$P_e, P_w, P_o$  are in kPa,  $T, T_o$  are in °K,  
 $\mu$  is in  $\mu\text{Pa.s}$ ,  $K, k_a$  are in  $\mu\text{m}^2$ ,  
 $h, r_e, r_w$  are in m,  $Q_o$  is in  $\text{m}^3/\text{day}$ ,  
 $\gamma$  is in msec/kg,  $F_b$  is in  $\text{m}^{-1}$ ,  
 $R$  is in  $\text{kPa.m}^3/\text{kgmole}^\circ\text{K}$ .

With the following relations

$$C_1 = \frac{1.2955 \times 10^{-3}}{h} \ln \frac{r_e}{r_w} \quad (4.11a)$$

$$C_2 = 2.9229 \times 10^{-15} \frac{G}{h^2} \left( \frac{1}{r_w} - \frac{1}{r_e} \right) \quad (4.11b)$$

$$C_3 = 6.5946 \times 10^{-21} \frac{G^2}{h^3} \left( \frac{1}{r_w^2} - \frac{1}{r_e^2} \right) \quad (4.11c)$$

Equation 4.10 is rewritten as,

$$P_e^2 - P_w^2 = \frac{C_1 \mu T Q_o}{k_a} + C_2 Z T Q_o^2 F_b + C_3 Z T Q_o^3 \gamma \quad (4.12)$$

Correcting for slippage by substituting for  $k_a$  with the relation in Equation 2.7, Equation 4.12 becomes

$$\begin{aligned} P_e^2 - P_w^2 &= \frac{C_1 \bar{T} Q_0}{k} - 2b(P_e - P_w) + C_2 \bar{T} Q_0^2 F_b \\ &\quad + \frac{C_2 \bar{T} Q_0}{P_m} (b \cdot F_b) + C_3 \bar{T} Q_0^3 \gamma + \frac{C_3 \bar{T} Q_0^3}{P_m} (b \cdot \gamma) \end{aligned} \quad (4.13)$$

For simplicity, let

$$CC1 = C_1 \bar{T} Q_0 \quad (4.14a)$$

$$CC2 = C_2 \bar{T} Q_0^2 \quad (4.14b)$$

$$CC3 = C_3 \bar{T} Q_0^3 \quad (4.14c)$$

$$PEW = 2(P_e - P_w) \quad (4.14d)$$

$$P_m = (P_e + P_w)/2 \quad (4.14e)$$

Therefore Equations 4.12 and 4.13 become respectively.

$$P_e^2 - P_w^2 = \frac{CC1}{k_a} + CC2 \cdot F_b + CC3 \cdot \gamma \quad (4.15)$$

$$\begin{aligned} P_e^2 - P_w^2 &= \frac{CC1}{k} - b \cdot PEW + CC2 \cdot F_b + \frac{CC2}{P_m} \cdot (b \cdot F_b) \\ &\quad + CC3 \cdot \gamma + \frac{CC3}{P_m} \cdot (\gamma \cdot b) \end{aligned} \quad (4.16)$$

Using the quadratic form, Equation 4.15 reduces to

$$P_e^2 - P_w^2 = \frac{CC1}{k_a} + CC2 \cdot F_b \quad (4.17)$$

This is the SI version of Equation 2.29 of Tek et al for which slippage is not corrected. When corrected for slippage, it becomes

$$\frac{P_e^2 - P_w^2}{k} = \frac{CC_1}{k} - b \cdot PEW + CC_2 \cdot F_b + \frac{CC_2}{P_m} (b \cdot F_b) \quad (4.18)$$

If the flow rate is such that the flow condition obeys Darcy's law, Equation 4.15 further reduces to

$$\frac{P_e^2 - P_w^2}{k} = \frac{CC_1}{k} - b \cdot PEW \quad (4.19)$$

Correcting for slippage using Equation 2.7, and rearranging, this equation becomes

$$k_a = \frac{CC_1}{\frac{P_e^2 - P_w^2}{k} - b \cdot PEW} = k \left(1 + \frac{b}{P_m}\right) \quad (4.20)$$

## 4.2 Parameter Estimation

### 4.2.1 Graphical Parameter Estimation

A linear Klinkenberg plot of  $k_a$  versus  $1/P_m$  can be obtained using Equation 4.20. The intercept at infinite mean pressure yields the absolute gas permeability,  $k$ , while the slope yields the gas slippage coefficient,  $b$ , as a product of  $bk$ .

Equation 4.17 can be rearranged into

$$\frac{\frac{P_e^2 - P_w^2}{k} - b \cdot PEW}{CC_1} = \frac{1}{k_a} + \frac{CC_2}{CC_1} \cdot F_b \quad (4.21)$$

while Equation 4.18 can be written in the form of

$$\frac{(P_e^2 - P_w^2) + b \cdot PEW}{CC1} = \frac{1}{k} + \frac{CC2}{CCT} \cdot F_b \left(1 + \frac{b}{P_m}\right) \quad (4.22)$$

Both forms yield linear visco-inertial plots. The expression for the ordinate in the visco-inertial plot, using Equation 4.21, is its reciprocal for the Klinkenberg plot, using Equation 4.20. The intercept on the first plot using Equation 4.21 gives the reciprocal of the apparent gas permeability,  $k_a$ , while the slope gives the inertial resistance coefficient,  $F_b$ . For the second plot, using Equation 4.22, the intercept yields the reciprocal of the absolute gas permeability,  $k$ , and the slope gives the inertial resistance coefficient,  $F_b$ , provided the slippage coefficient,  $b$ , is known from the Klinkenberg plot. Equations 4.21 and 4.22 are, respectively, the radial SI version of Equations 3.1 and 3.3, which are due to Kolada. The plots obtained from them are known, respectively, as the "conventional visco-inertial plot" and the "modified visco-inertial plot".

A back-pressure curve of  $(P_e^2 - P_w^2)$  versus  $Q$  is plotted on log-log paper to delineate the plot points in the viscous region. The points that lie on the 45 degree line are taken to be in the viscous region. They are used in the Klinkenberg plot to obtain both the gas permeability,  $k$ , and the gas slippage coefficient,  $b$ .

The points on the linear profile of both the

Klinkenberg and the visco-inertial plots are used to find the best straight lines by the method of least squares. The parameter values obtained are checked with those obtained by graphical means.

#### 4.2.2 Numerical Parameter Estimation

Numerical methods permit the use of both the quadratic and the cubic model flow equations in estimating the parameters simultaneously. The model equations are fitted to the experimental data, and plotted to check for the best fit.

#### 4.2.3 Direct Multiple Linear Regression Method

Equation (4.15) is linear in its coefficients even though it is itself nonlinear. If the combined variables in Equation 4.16 can be represented by single variables, linearity in its coefficients may be achieved. In these forms, both equations can be adapted to a multiple linear regression technique for parameter estimation. The same case exists if Equation 4.16 is further simplified by neglecting the combined variables.

Linearity in the coefficients can similarly be achieved for the quadratic case when Equation 4.18 is used. By neglecting its combined variable, this equation is further simplified to the SI version of Equation 2.32 which is due to Piplapure.

#### 4.2.4 Model Fitting

The cubic equation (Equation 4.15) can be rearranged in terms of  $k_a$ ,  $F_b$ , gamma, as

$$\frac{P_e^2 - P_w^2}{CC1} = \frac{1}{k_a} + \frac{CC2}{CCT} \cdot F_b + \frac{CC3}{CCT} \cdot \gamma \quad (4.23)$$

or, in terms of its polynomial in  $Q_0$ , as

$$P_e^2 - P_w^2 = \left( \frac{C_1 \bar{\mu} \bar{T} \bar{Z}}{k_a} \right) \cdot Q_0 + (C_2 \bar{Z} \bar{T} F_b) \cdot Q_0^2 + (C_3 \bar{Z} \bar{T} \gamma) \cdot Q_0^3 \quad (4.24)$$

For the quadratic case, Equation 4.24 reduces to

$$P_e^2 - P_w^2 = \left( \frac{C_1 \bar{\mu} \bar{T} \bar{Z}}{k_a} \right) \cdot Q_0 + (C_2 \bar{Z} \bar{T} F_b) \cdot Q_0^2 \quad (4.25)$$

while Equation 4.23 becomes Equation 4.21.

Fitted visco-inertial plots are made by plotting both the fitted values of the left-hand side of Equations 4.21 and 4.23, and experimental data, respectively on the ordinate against  $CC2/CC1$  on the abscissa. The best fit, cubic or quadratic, is sought for the experimental data. Similarly, fitted back-pressure plots on log-log paper are made by plotting both the fitted values of the left-hand side of Equations 4.24 and 4.25, and experimental data, respectively on the ordinate against  $Q_0$ . The best fit, beyond the linear section which obeys Darcy's law, is determined.

The formal adaptation of these model equations for

computer application is given in Appendix B.

## 5. CONFINEMENT AND FLUID PRESSURE CONSIDERATIONS

Following the discussions in Chapter 3, a need arises for a re-evaluation of confinement and fluid pressure considerations with respect to the partial uniaxial confining pressure imposed on the core. The geometry of the core is cylindrical; therefore it suggests a thick-walled cylindrical model. If the core is assumed to possess elastic properties, analysis can be made using the theory of elasticity. This assumption, therefore, furnishes a good method for investigating the state of stress and strain (deformation) as gas flows into the well bore from the external boundary.

### 5.1 Stress-Strain Analysis

By definition (59), principal stresses at any point within a stressed body, are stresses which are normal to three mutually perpendicular planes in which there are no shearing stresses. Hence, the state of stress at any point within the core in the cell may be expressed, respectively, in terms of the principal stresses - axial,  $S_z$ , radial,  $S_r$ , and tangential,  $S_t$ . The sum of the radial and tangential stresses gives the horizontal stresses. They can either be compressive or tensile. By convention, a negative sign denotes compressive stress, and a positive sign indicates tensile stress.

Similarly, principal strains have no shearing strains associated with them. Their extensions are in the directions of the principal stresses with which they are related according to the generalized Hooke's law(60)

$$\epsilon_z = \frac{1}{E} (-\nu s_r - \nu s_t + s_z) \quad (5.1)$$

$$\epsilon_r = \frac{1}{E} (s_r - \nu s_t - \nu s_z) \quad (5.2)$$

$$\epsilon_t = \frac{1}{E} (-\nu s_r + s_t - \nu s_z) \quad (5.3)$$

where E is the Young's modulus and the Poisson's ratio is given as

$$\nu = \left| \frac{\text{Lateral Strain}}{\text{Axial Strain}} \right| \quad (5.4)$$

The strains are caused by uniaxial stress only and both the lateral and axial strains are of opposite sign for uniaxial stress. The ratio, which is an elastic constant, is usually a positive number which varies for different materials, according to Jaeger(61), in the range of 0.10 and 0.50.

Assuming Poisson's relation, Jaeger established that this ratio simplifies to 0.25. Also assuming a solid, incompressible material, he established the ratio to be 0.5. Furthermore, the ratio for sedimentary materials(62) is

found to lie between 0.115 and 0.300. Several investigators(63,64,65) have used 0.25, and for this work, the value of 0.25 is chosen as an approximation to the true value of the ratio.

For uniaxial confining stress, the principal axial strain is assumed to be zero and Equation 5.1 reduces to

$$S_z = v(S_r + S_t) \quad (5.5)$$

Substituting this into Equations 5.2 and 5.3, and solving for the radial and tangential stresses, the following expressions are obtained:

$$S_r = \frac{E}{(1+v)(1-2v)} [(1-v)\epsilon_r + v\epsilon_t] \quad (5.6a)$$

$$S_t = \frac{E}{(1+v)(1-2v)} [(1-v)\epsilon_t + v\epsilon_r] \quad (5.6b)$$

The strain-displacement relationships in cylindrical co-ordinates due only to the radial flow rate of displacement U, are given by

$$\epsilon_r = \frac{(U + \frac{dU}{dr} dr) - U}{dr} = \frac{dU}{dr} \quad (5.7a)$$

$$\epsilon_t = \frac{(r+U)d\theta - r d\theta}{rd\theta} = \frac{U}{r} \quad (5.7b)$$

Since flow is assumed to be radial in the direction of the principal radial stress, the additional portion of the tangential strain due to incremental tangential displacement is neglected.

Equations 5.7a and 5.7b are substituted into Equations 5.6a and 5.6b to obtain

$$S_r = \frac{E}{(1+v)(1-2v)} [(1-v) \frac{dU}{dr} + v \frac{U}{r}] \quad (5.8a)$$

$$S_t = \frac{E}{(1+v)(1-2v)} [(1-v) \frac{U}{r} + v \frac{dU}{dr}] \quad (5.8b)$$

If an infinitesimal element in static equilibrium within the core is considered, then by three dimensional force balance, the equation of equilibrium in radial direction only is given by

$$\frac{\partial S_r}{\partial r} + \frac{1}{r} \frac{\partial \tau_{r\theta}}{\partial \theta} + \frac{\partial \tau_{rz}}{\partial z} + \frac{1}{r} (S_r - S_t) + R = 0 \quad (5.9)$$

By assumption, shear stresses are zero. For a plane horizontal radial core, gravitational forces  $R$  are assumed zero. Therefore Equation 5.9 reduces to

$$\frac{\partial S_r}{\partial r} + \frac{1}{r} (S_r - S_t) = 0 \quad (5.10a)$$

From inspection, Equation 5.10a suggests that the radial stress is dependent only on the radius  $r$ . Therefore, it can

be written as

$$\frac{dS_r}{dr} + \frac{1}{r} (S_r - S_t) = 0 \quad (5.10b)$$

Substituting Equations 5.8a and 5.8b into Equation 5.10b and simplifying, yields the following second order differential equation:

$$\frac{d^2U}{dr^2} + \frac{1}{r} \frac{dU}{dr} - \frac{U}{r^2} = 0 \quad (5.11)$$

By integrating this, the following general equation is obtained:

$$U = A_1 r + \frac{A_2}{r} \quad (5.12)$$

Both the inner and the outer pressures are compressive on the radial core at their respective radii. Therefore the boundary conditions for the core are

$$\begin{array}{ll} r = r_w & S_r = -P_w \\ r = r_e & S_r = -P_e \end{array} \quad (5.13)$$

Substituting Equations 5.12 and 5.13 into Equation 5.8a, and solving for the constants of integration, yields

$$A_1 = \frac{(1+v)(1-2v)}{E} \cdot \frac{P_w r_w^2 - P_e r_e^2}{r_e^2 - r_w^2} \quad (5.14a)$$

$$A_2 = \frac{(1+v)}{E} \cdot \frac{r_e^2 r_w^2 (P_w - P_e)}{r_e^2 - r_w^2} \quad (5.14b)$$

These equations, together with Equation 5.12, are substituted into Equations 5.8a and 5.8b to yield the general radial and tangential stresses in terms of the boundary conditions as follows:

$$S_r = \frac{P_w r_w^2 - P_e r_e^2}{r_e^2 - r_w^2} - \frac{(P_w - P_e) r_w^2 r_e^2}{(r_e^2 - r_w^2) r^2} \quad (5.15a)$$

$$S_t = \frac{P_w r_w^2 - P_e r_e^2}{r_e^2 - r_w^2} + \frac{(P_w - P_e) r_w^2 r_e^2}{(r_e^2 - r_w^2) r^2} \quad (5.15b)$$

From these equations, which are originally due to Lame(66), both the radial and the tangential stresses are compressive for cases where  $P_e$  is greater than or equal to  $P_w$ .

With these equations, the stresses at any radial point  $r$  are easily evaluated. This suggests that the externally applied stresses may be regarded as continuations of the internal stresses. For uniaxial confinement pressure condition, these equations are substituted into Equation 5.5 to get

$$S_z = 2v \left[ \frac{P_w r_w^2 - P_e r_e^2}{r_e^2 - r_w^2} \right] \quad (5.16)$$

The result from Brant's investigation(67) suggests that only 85% of the internal fluid pressures within the pores react against the imposed confining pressure. Fatt(68), added that this percentage is pressure dependent and does vary from rock to rock. Therefore with this correction, Equation 5.16 yields the following equation:

$$S_z = 1.7\nu \left[ \frac{P_w r_w^2 - P_e r_e^2}{r_e^2 - r_w^2} \right] \quad (5.17)$$

and using, for a limiting case, 0.25 as the value of the Poisson's ratio, Equation 5.17 becomes

$$S_z = 0.425 \left[ \frac{P_w r_w^2 - P_e r_e^2}{r_e^2 - r_w^2} \right] \quad (5.18)$$

## 6. EXPERIMENTAL EQUIPMENT AND PROCEDURE

### 6.1 Experimental Apparatus

The apparatus consists of a Boyles' law porosimeter (Figure 6-1) and a flow-test equipment (Figure 6-2). The Boyles' law porosimeter is used to measure the porosity of the core samples usually prior to starting the experimental runs. It consists of a vacuum pump, a constant volume bottle connected to a radial chamber, a gauge and a manometer. The flow test equipment is used to conduct the experimental runs. It consists of a radial cell, a hydraulic pump, a monitoring console for pressure and flow rate readings, and a copper constantan thermocouple.

Two radial cells, designed to house the core sample, are connected, respectively, to the flow-test equipment for uniaxial confinement (Figure 6-3), and for triaxial confinement (Figure 6.4). The experimental set-up, which contains the radial cell for uniaxial confinement, is similar to the set-up that Senturk used(69). This radial cell, together with the hydraulic pump, provides uniaxial loading by confining the core in the direction of its axial height. Similarly, the triaxial overburden radial cell, together with the hydraulic pump, is expected to provide a uniform confining pressure in both the horizontal and the normal directions. This radial cell is designed in such a way that gas enters into the core sample through a sintered screen which ensures a uniform radial flow. The rubber

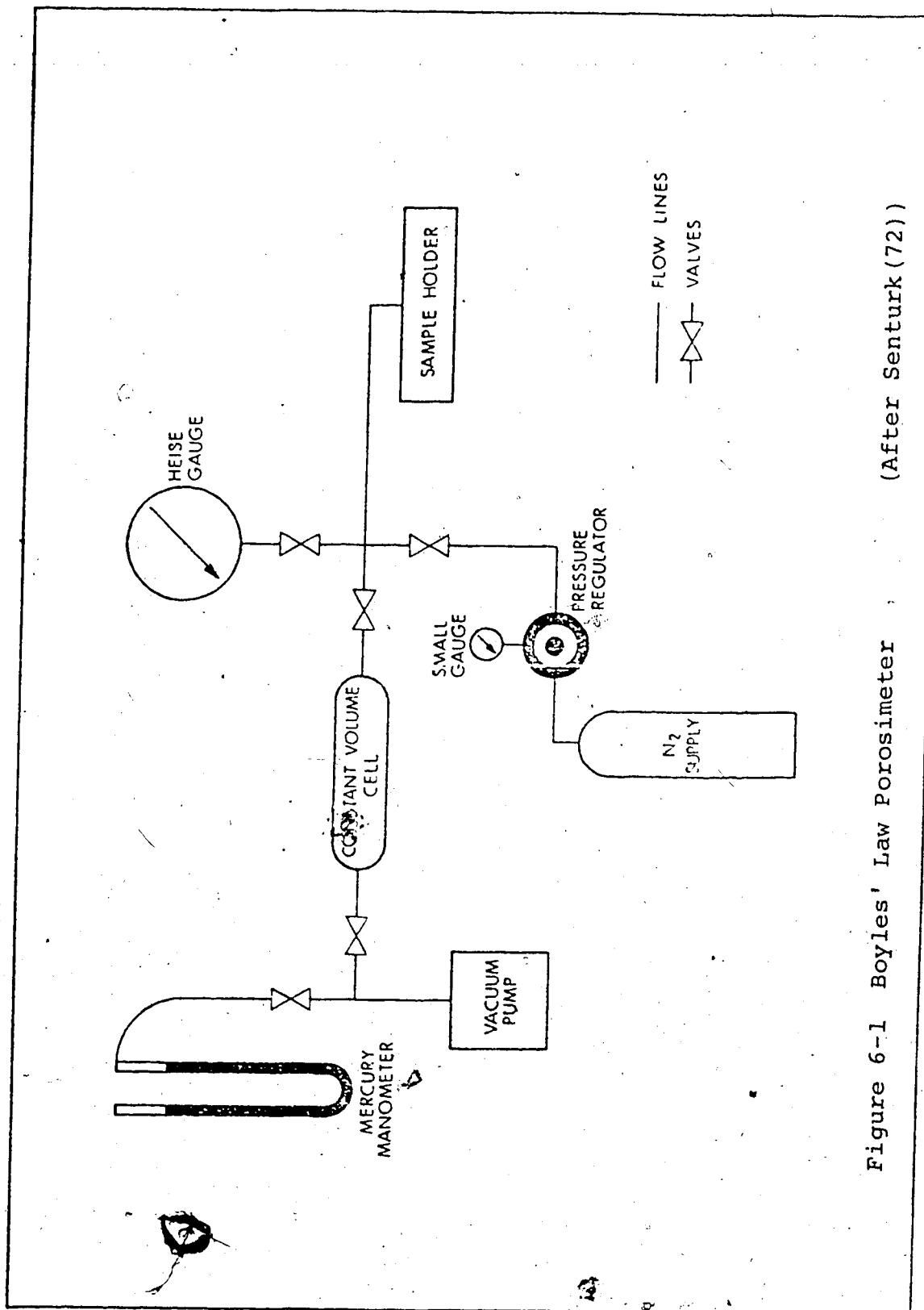


Figure 6-1 Boyles' Law Porosimeter  
(After Senturk (72))

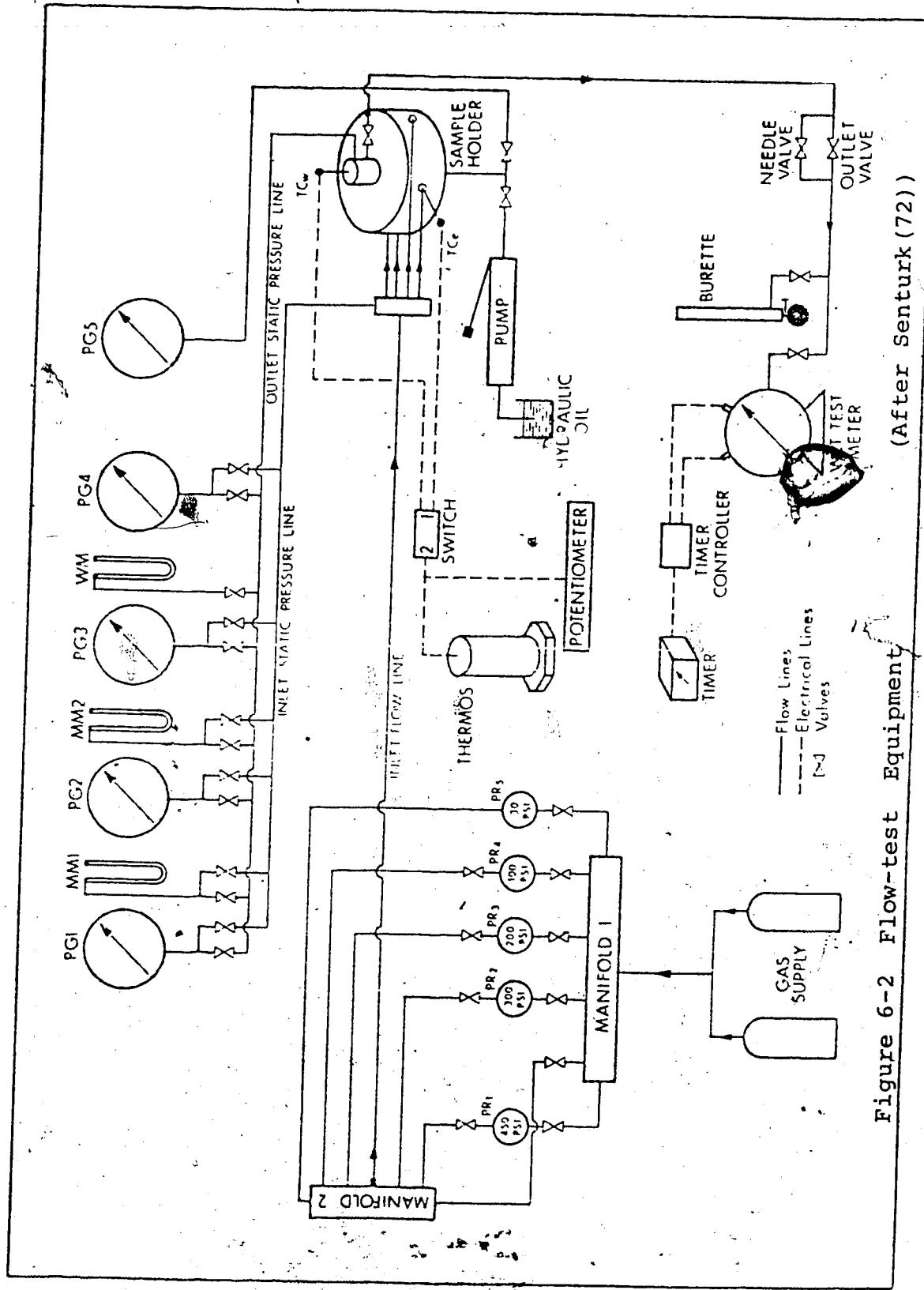


Figure 6-2 Flow-test Equipment (After Senturk (72))

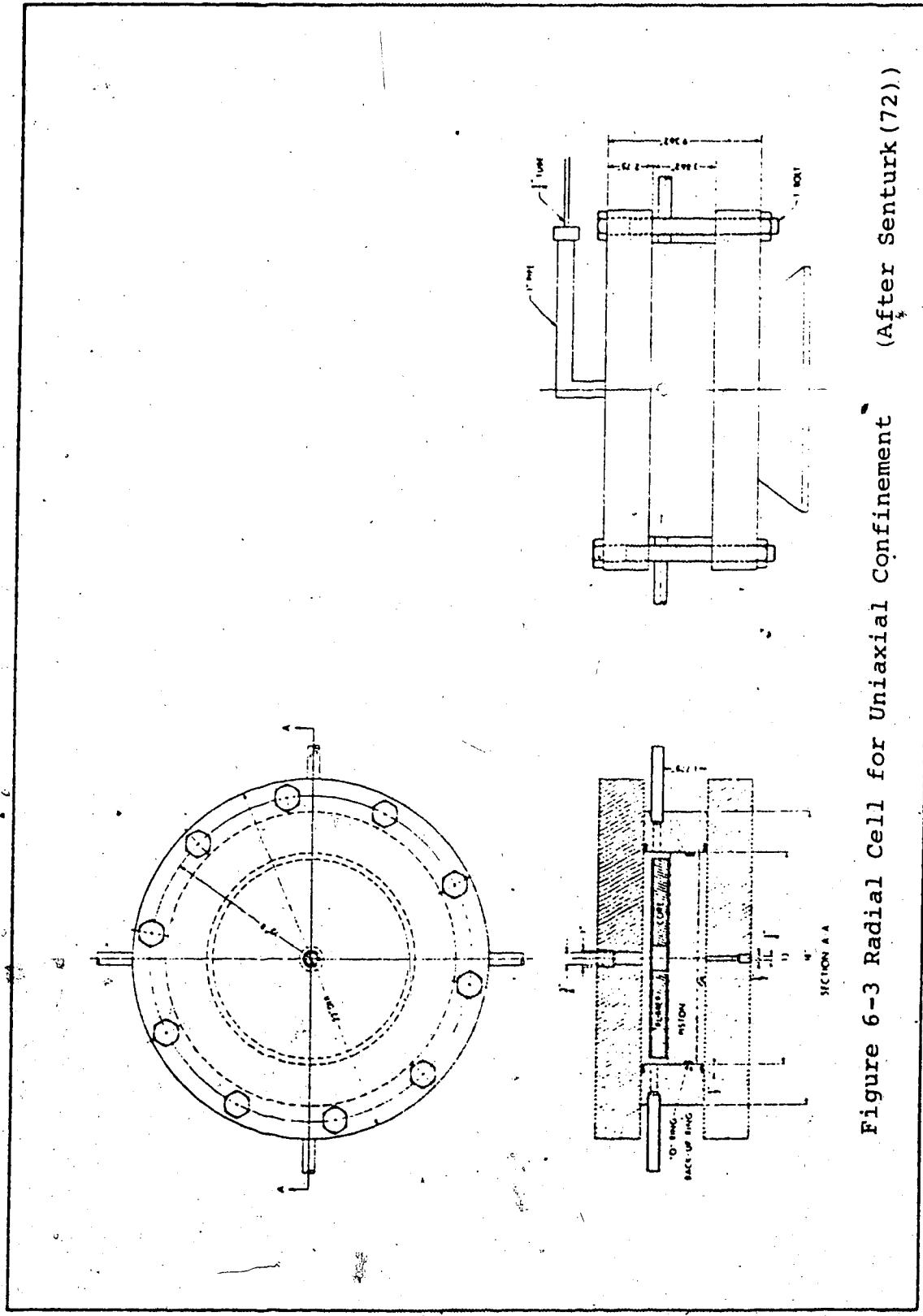


Figure 6-3 Radial Cell for Uniaxial Confinement  
(After Senturk (72))

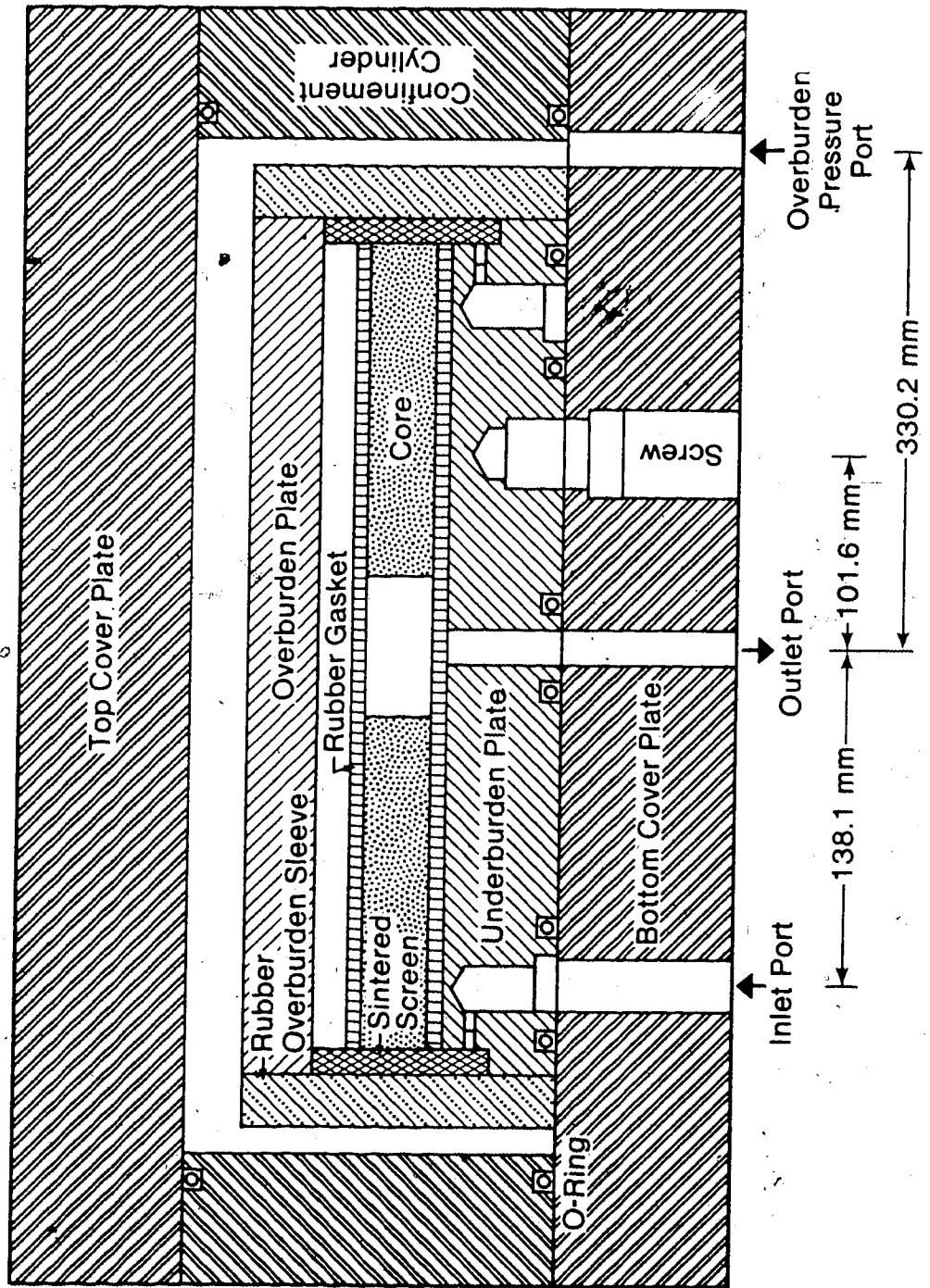


Figure 6-4. Radial Tri-axial Overburden Cell

overburden sleeve is wrapped tightly round the core sample, the sintered screen, and both the overburden and the underburden plates. This is then enclosed in the confinement cylinder. When pressurized by using the hydraulic pump, a uniform, all-sided (triaxial) confinement is expected on the core sample. However, the rubber overburden sleeve in this cell is loose, and to ensure a triaxial confinement, the sleeve was held tight with a wire. With either cell, varying confining pressures,  $P_C$ , on the core can be provided as desired during the course of the runs.

## 6.2 Porosity Determination

Estimation of effective porosities can be made with the Boyles' law porosimeter which is operated by the gas expansion method. With this method, the stabilized pressure,  $P_2$ , of the gas, which occupies the constant volume bottle of volume  $V_2$ , is measured. By allowing the gas to expand through a connector valve into the radial chamber of volume  $V_3$ , which contains solid aluminium blanks of volume  $V_1$ , the new equalizing pressure,  $P_3$ , is recorded. Since both volumes,  $V_2$  and  $V_3$ , are constant, the equalizing pressure varies with the volume of the solid blanks in the radial chamber. Plotting  $P_3$  against  $V_1$ , and  $P_2/P_3$  against  $V_1$ , (Figure 6-5), serves as a calibration for the porosimeter.

If the volume of the solid blanks represents the effective grain volume of a core sample, then, replacing the

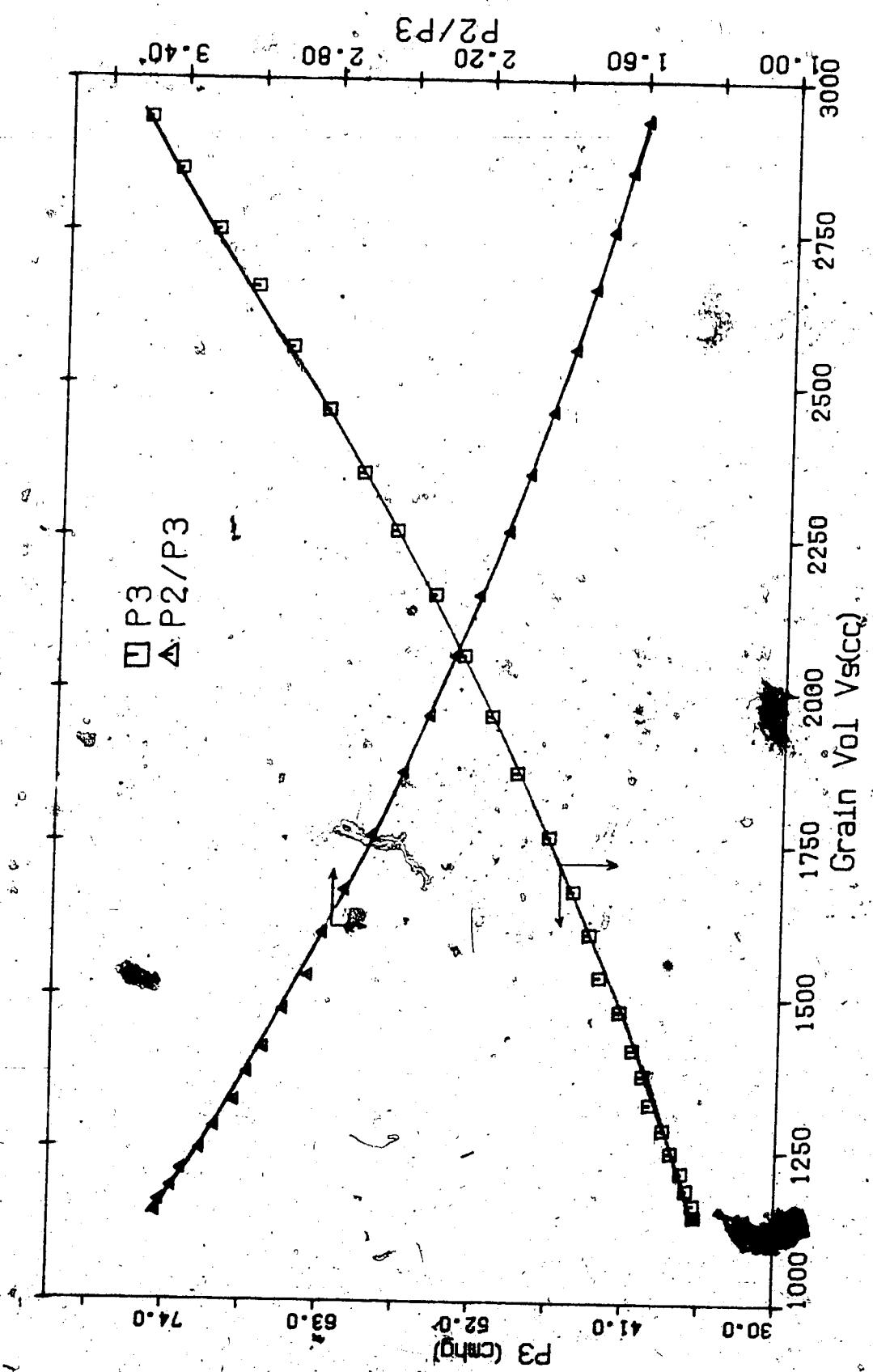


Figure 6-5 CALIBRATION OF BOYLE'S LAW POROSIMETER

blanks with a core sample, and repeating the experiment, its effective grain volume can be obtained from the plots. From the dimensions of the core samples, which were previously measured, the bulk volume is calculated, and porosity is determined from the following definition:

$$\phi = (\text{Pore volume/Bulk Volume}) \times 100 \% \quad (6.1)$$

### 6.3 Flow Test Methods

The four flow cases considered in Chapter 3 are rearranged to give:

- Flow Method 1, where both  $P_e$  and  $P_w$  increase (forward direction), or both decrease (reverse direction),
- Flow Method 2, where  $P_w$  is kept constant while  $P_e$  increases (forward direction), or while  $P_e$  decreases (reverse direction),
- Flow Method 3, where  $P_e$  is kept constant while  $P_w$  increases (forward direction), or while  $P_w$  decreases (reverse direction).

Keeping either  $P_w$  or  $P_e$  constant while the other is varied, results, by Equation 3.8, in larger drawdown pressures, which induce radial compaction. Also, by increasing or decreasing both  $P_w$  and  $P_e$ , higher mean pressures and lower drawdown pressures result. Therefore, the expected differences between Flow Method 1 and Flow Methods 2 and 3 may be given as:

- a) The mean pressure is larger but the drawdown pressure is much less in Flow Method 1 than in Flow Methods 2 and 3.
- b) Depending on the magnitude of  $P_w$  and  $P_e$ ,  $(P_e^2 - P_w^2)$  may increase more rapidly for increasing  $Q_0$  in Flow Methods 2 and 3 than in Flow Method 1.
- c) As flow progresses, the generated confinement pressure, by Equation 5.17, may increase more slowly in Flow Method 1 than in Flow Methods 2 and 3.

Conducting the experimental runs in both forward and reverse directions serves to check for hysteresis.

#### 6.4 Experimentation

The core sample is mounted into the radial cell and the flow-test equipment is pressure-tested for leaks. Gas is passed into the core sample several hours prior to starting any experimental run. For each run, and for each step within a run, the inlet pressure,  $P_e$ , is pre-selected and measured, as gas passes through the inner line to the outer periphery of the core. This is repeated for different pre-defined confining pressures,  $C_P$ .

Using the timer, the flow rate is obtained as the ratio of one complete revolution of the wet-test meter dial to the time it takes to make a complete revolution. When this time is constant, the flow is assumed to be steady and all readings are taken at this condition. These include:

- the outlet pressure,  $P_w$ , at the inner periphery,
- the confining pressure,  $C_P$ ,

- the galvanometer readings,  $T_{ce}$  and  $T_{cw}$ , for the inlet and outlet temperatures respectively,
- the flow temperature on top of the wet-test meter,
- the ambient conditions of mean temperature and barometric pressure.

## 7. TREATMENT OF DATA

Readings and measurements are taken in English units. They are converted into the current SI units with standard reference condition of temperature and pressure being 288.150 °K and 101.325 kPa, respectively.

### 7.1 Temperature Conversion

The anticipated optimum temperature range, between 15°C and 30°C for corresponding millivolt (mV) readings, is obtained from conversion charts and curve fitted. The ensuing linear relation for the temperature range is

$$^{\circ}\text{C} = (\text{mV} + 0.021456) / 0.040465$$

### 7.2 Physical and Fluid Properties

The physical properties of the core samples are summarized in Table 7.1.

Evaluation of the fluid properties - compressibility factor and gas viscosity - is made at arithmetic mean temperature and pressure, using nitrogen. The compressibility factors for nitrogen at given temperatures and pressures are determined, from the curve-fitted data of Hilsenrath et al(70) using a Lagrangian interpolation technique. The viscosity of nitrogen is obtained from the

TABLE 7-1: PHYSICAL PROPERTIES OF THE CORE SAMPLES

| Core sample        | 1C      | 4A      | 2       |
|--------------------|---------|---------|---------|
| Inner Radius, m    | 0.00319 | 0.00820 | 0.00815 |
| Outer Radius, m    | 0.15239 | 0.15256 | 0.15225 |
| Height, m          | 0.02568 | 0.02600 | 0.02803 |
| Porosity, fraction | 0.10620 | 0.22714 | 0.23432 |

curve-fitted equation of Kestin and Wang(7.1):

$$\mu = 10^5 \{ 0.778 \times 10^{-3} [1 + 0.8958 \times 10^{-3} (P_{\text{atm}} - 1) \\ + 0.612 \times 10^{-6} (P_{\text{atm}} - 1)^2 + 0.3927 \times 10^{-7} (P_{\text{atm}} - 1)^3] \\ + 0.455 \times 10^{-6} (T - [273.15 + 25]) \} \quad (7.2)$$

$\mu = \text{Pa}\cdot\text{s}$

Both treatments for the compressibility factor and viscosity are given in computer subroutine LAGINT.

## 8. EXPERIMENTAL RESULTS AND DISCUSSION

Flow Method 1 has been used on Core Samples 1C and 4A for 24 experimental runs in both forward and reverse directions under uniaxial confinement. High flow rates and pressures were reached for various predefined confining pressures, CP. Runs 7, 8, and 9, were made on Core 4A for lower flow rates and pressures, using Flow Method 2. Core Sample 2 was obtained after the radial cell for uniaxial confinement was replaced with the triaxial overburden radial cell. Hence runs on Core 2 could not be made under uniaxial confinement. Instead, four runs were made on it under triaxial confinement, and by using both flow methods. The predefined confining pressure, CP, on Core 2, was kept constant at 4136.9 kPa.

### 8.1 Plot Profiles with Flow Method 1

Figures 8-1, 8-2, and 8-3 show, respectively, the Klinkenberg, visco-inertial, and back-pressure plots, for Core 1C, Run 7, in the forward direction. Linearity, in the Klinkenberg plot exists at the lower ranges of flow rates and pressures, after which deviation starts in a downward manner. The visco-inertial plot has a surprising profile in which three stages are observed as  $(P_e^2 - P_w^2)/CC_1$ , (which is  $1/ka$ ), increases for increasing flow rate,  $Q_o$ . After the

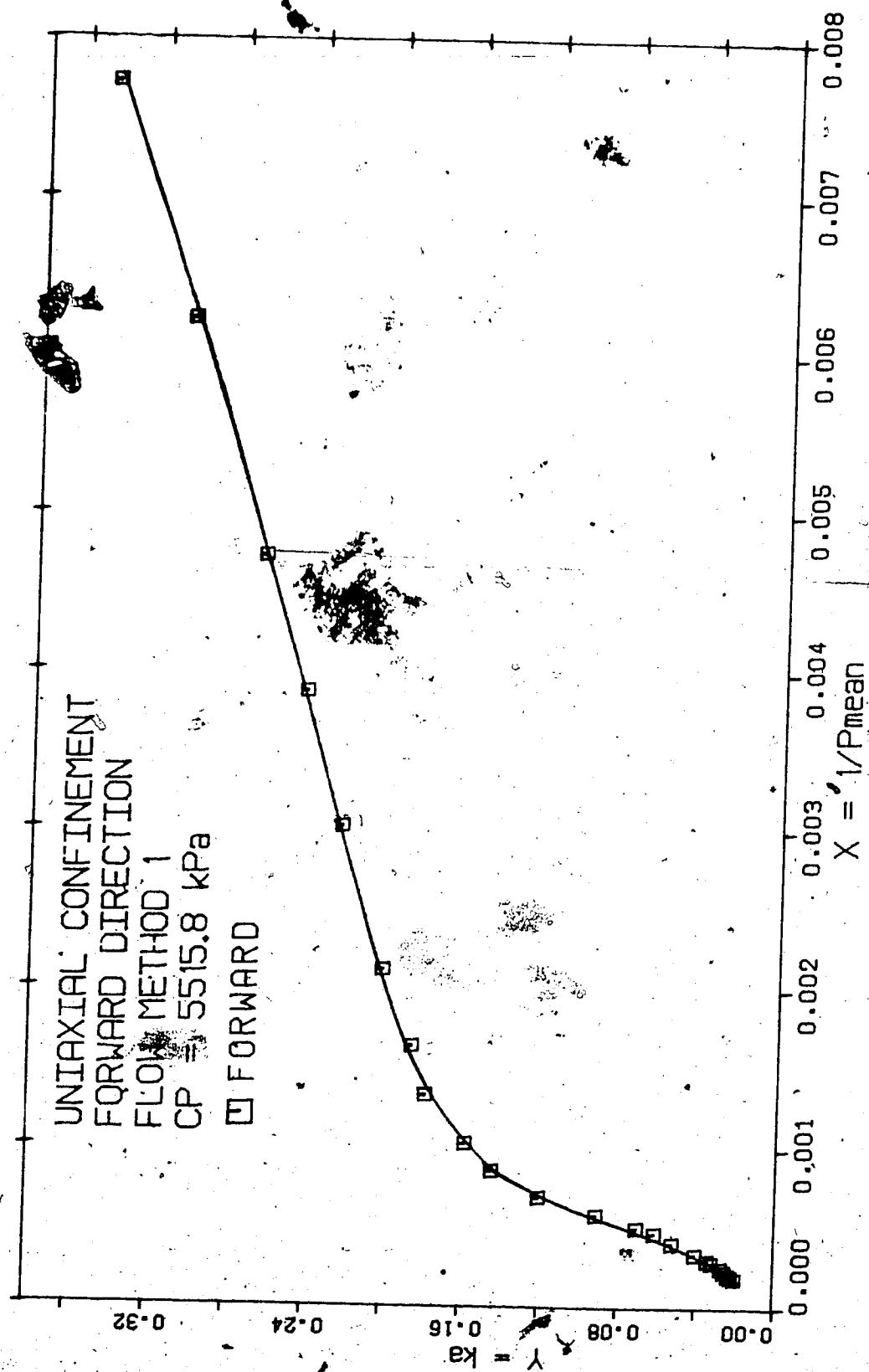


Fig. 8-1: CORE 1C KLINKENBERG PLOT RUN 7

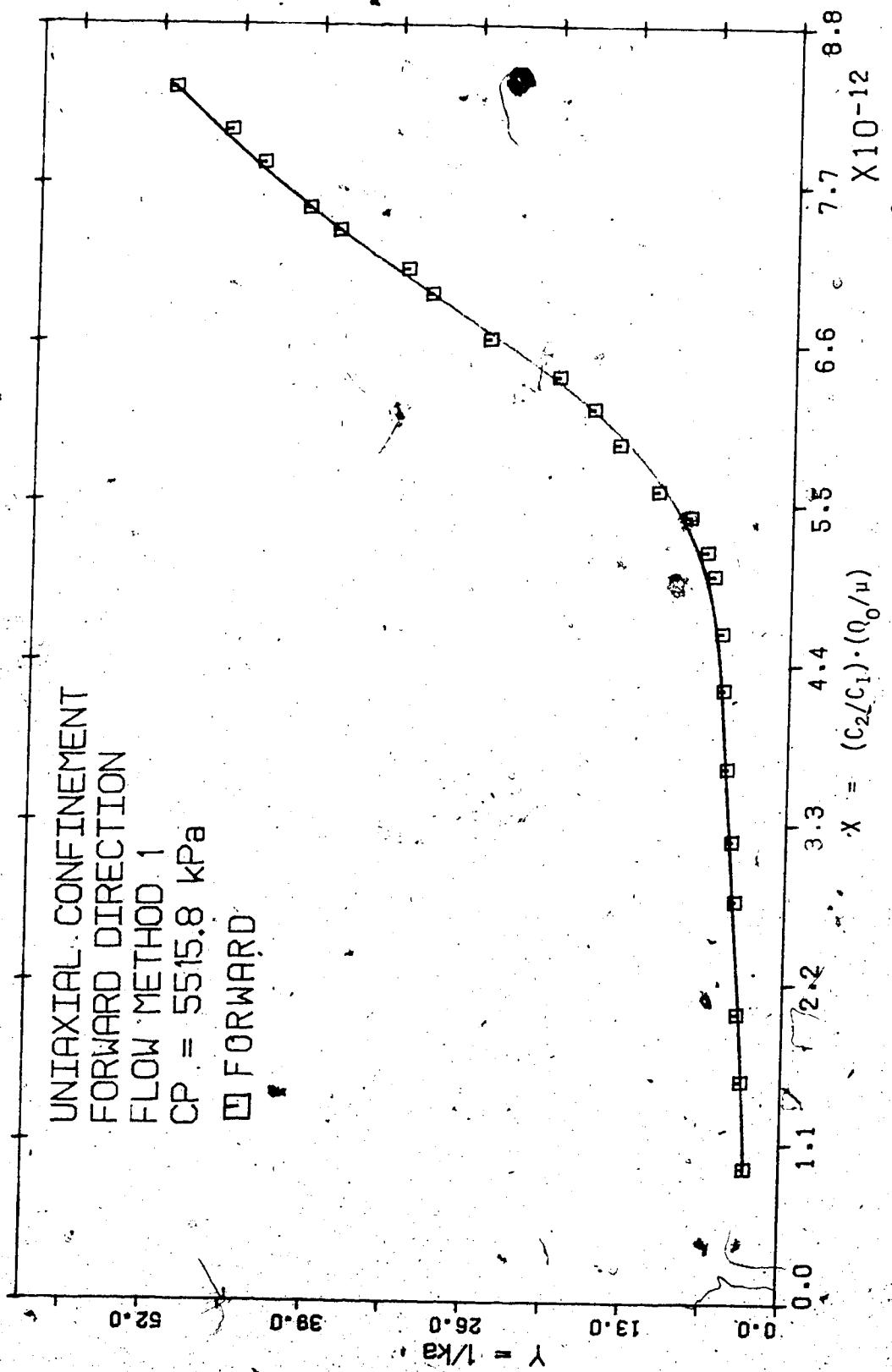


Fig.8-2: CORE 1C VISCO-INERTIAL PLOT RUN 78

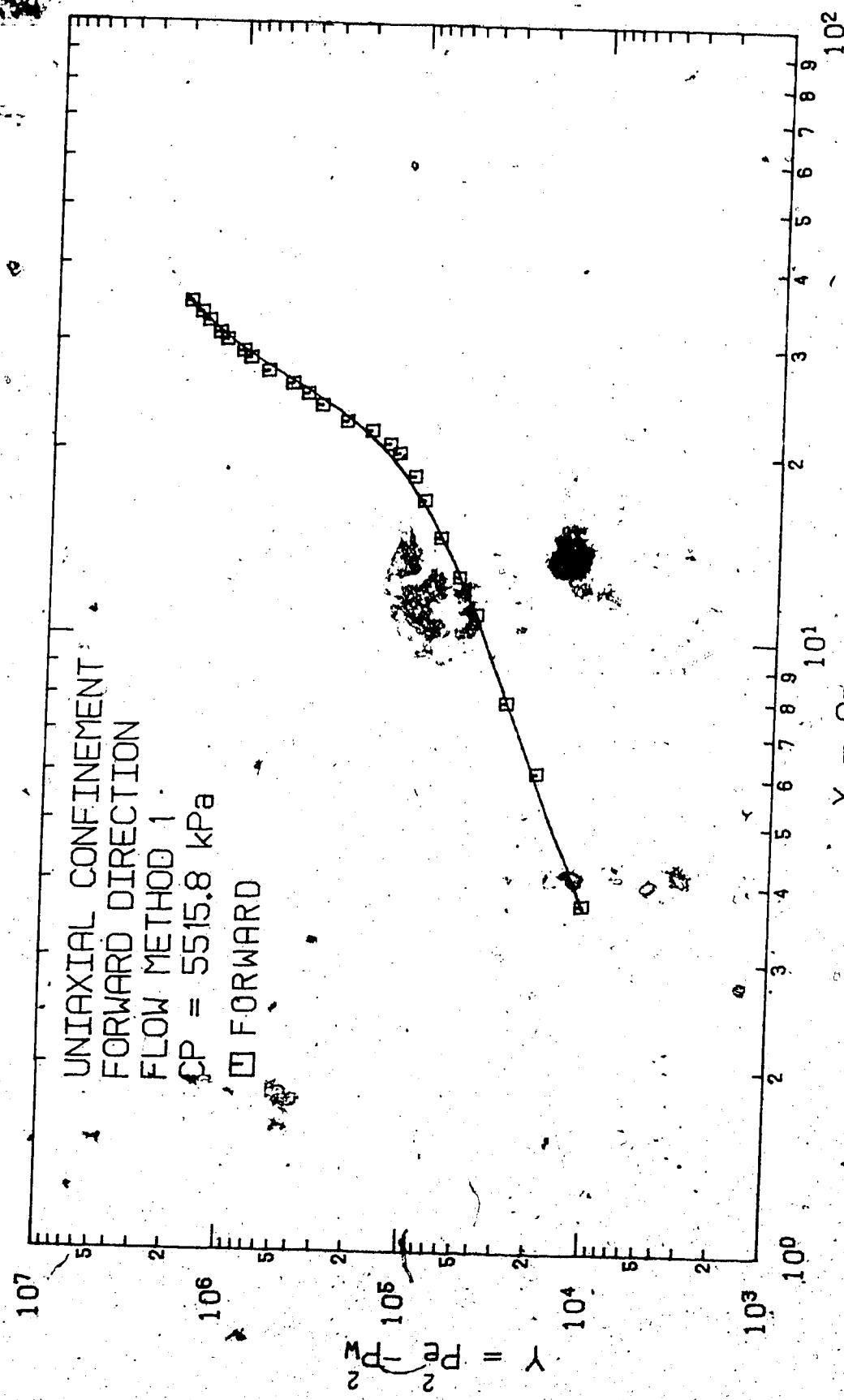


Fig.8-3: CORE 1C BACK PRESSURE PLOT RUN 7

first stage,  $1/ka$  sharply increases with steeper slopes as  $Q_o$  further increases. In the third stage, the steepness of the slope reduces as  $1/ka$  continues to increase for further increases in  $Q_o$ . The back-pressure plot on log-log paper shows three corresponding regions as  $(Pe^2 - Pw^2)$  increases with increasing  $Q_o$ . Its first region is linear with slope angle of  $45^\circ$  after which, the slope increases in its second stage, but decreases in the third stage. The general profile for these plots, therefore, is such that  $(Pe^2 - Pw^2)$  increases, while  $ka$  decreases, with increasing  $Q_o$  and  $P_m$ .

From the model Equation 4.21 and the literature, the visco-inertial plot is expected to slope down at lower flow rates and pressures, and become linear as flow rates and pressures increase. Accordingly, the back-pressure plot is expected to be linear with slope of  $45^\circ$  and increase in slope for higher  $Q_o$ . By Flow Method 1 for the forward direction, both  $Pe$  and  $Pw$  increase for increasing  $Q_o$  and, therefore,  $(Pe^2 - Pw^2)$  increases with increasing  $P_m$  and increasing  $Q_o$ . This is equivalent to Case 2 of Table 3.1. With analysis similar to that used for Case 2 in Chapter 3,  $ka$  is expected to decrease for decreasing  $1/P_m$ , (Klinkenberg plot), and  $1/ka$  is expected to increase with increasing  $Q_o$ , (visco-inertial plot). These predicted profiles, arising from the flow method used, agree with observed experimental plot profiles.

The anomalies, observed in the later stages, and which involve the variation of  $(Pe^2 - Pw^2)$  and  $ka$  with  $Q_o$ , lead to

the following observations:

- a) Linearity of Model Equation (4.21) is obtained only in the first region with moderate flow rates and pressures. Beyond this,  $(Pe^2 - Pw^2)$ , increases rapidly with increasing  $Q_o$ , as can be seen from the steepness of the slope.
- b) In the third region,  $Q_o$  appears to increase faster than it should be for increasing  $(Pe^2 - Pw^2)$ .

The first observation suggests a higher-order model equation, while the second poses a problem similar to that observed from Figures 3.3 and 3.6, in which lower  $(Pe^2 - Pw^2)$  values were obtained for increasing  $Q_o$ .

## 8.2 Plot Profiles with Flow Method 2

Surprising profiles have been observed in Figures 8.4 to 8.9, which correspond, respectively, to the Klinkenberg, the visco-inertial and the back-pressure plots of Runs 8 and 9 using Core Sample 4A. In both the Klinkenberg and the visco-inertial plots,  $k_a$  values are observed to decrease initially with increasing  $P_m$  and  $Q_o$ , and then increase with increasing  $P_m$  and  $Q_o$ . In the back-pressure plots,  $(Pe^2 - Pw^2)$  increases with increasing  $Q_o$ . The slope angle is initially 45°, after which it decreases with increasing  $Q_o$ . In this later region, the plots indicate lower  $(Pe^2 - Pw^2)$  for increasing  $Q_o$ . These profiles are similar to those of Figures 3.1 to 3.6, and, therefore, they pose similar problems.

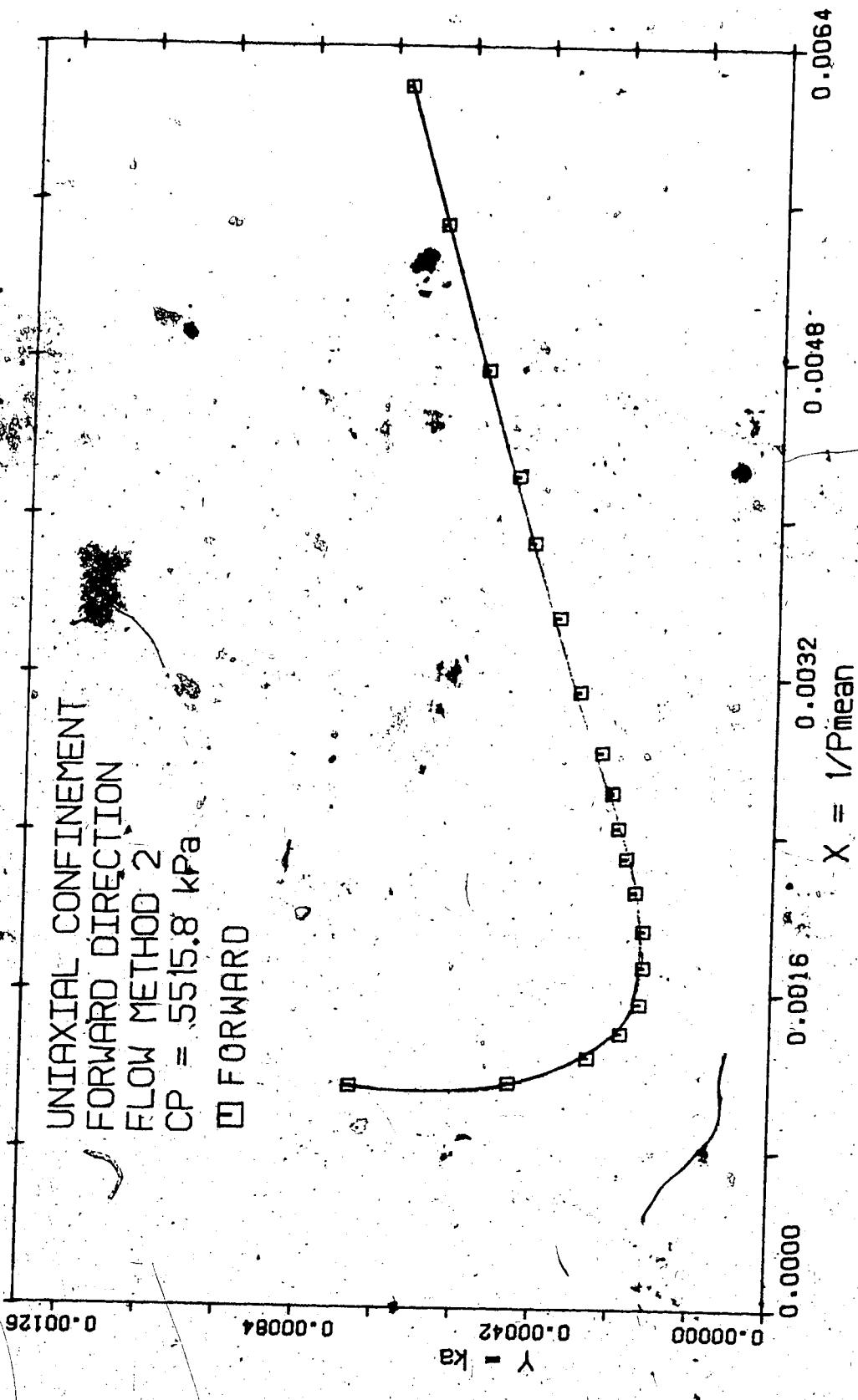


Fig. 8-4: CORE 4A KLINKENBERG PLOT RUN 8

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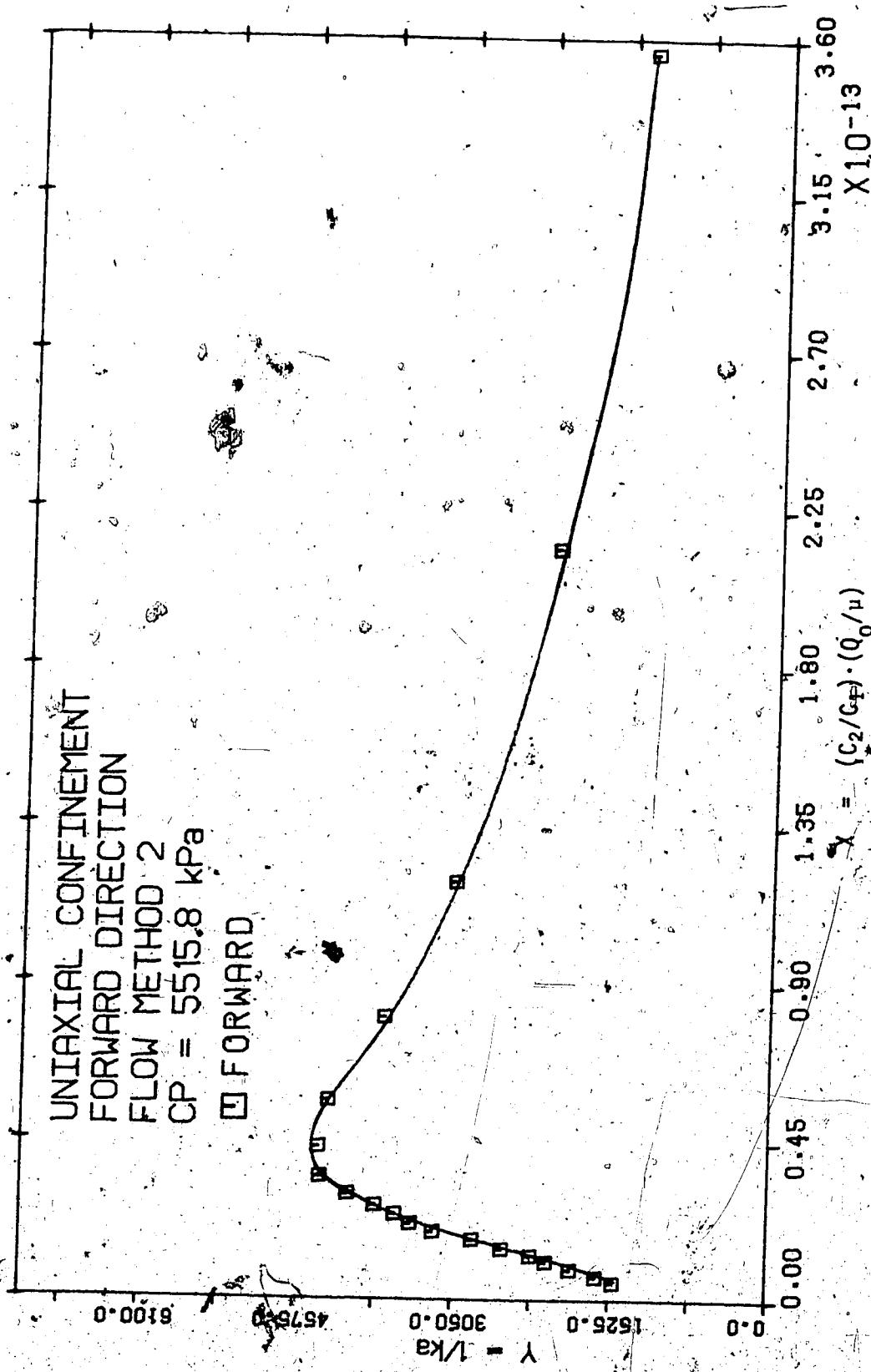


Fig.8-5: CORE 4A- VISCO-INERTIAL PLOT RUN 8

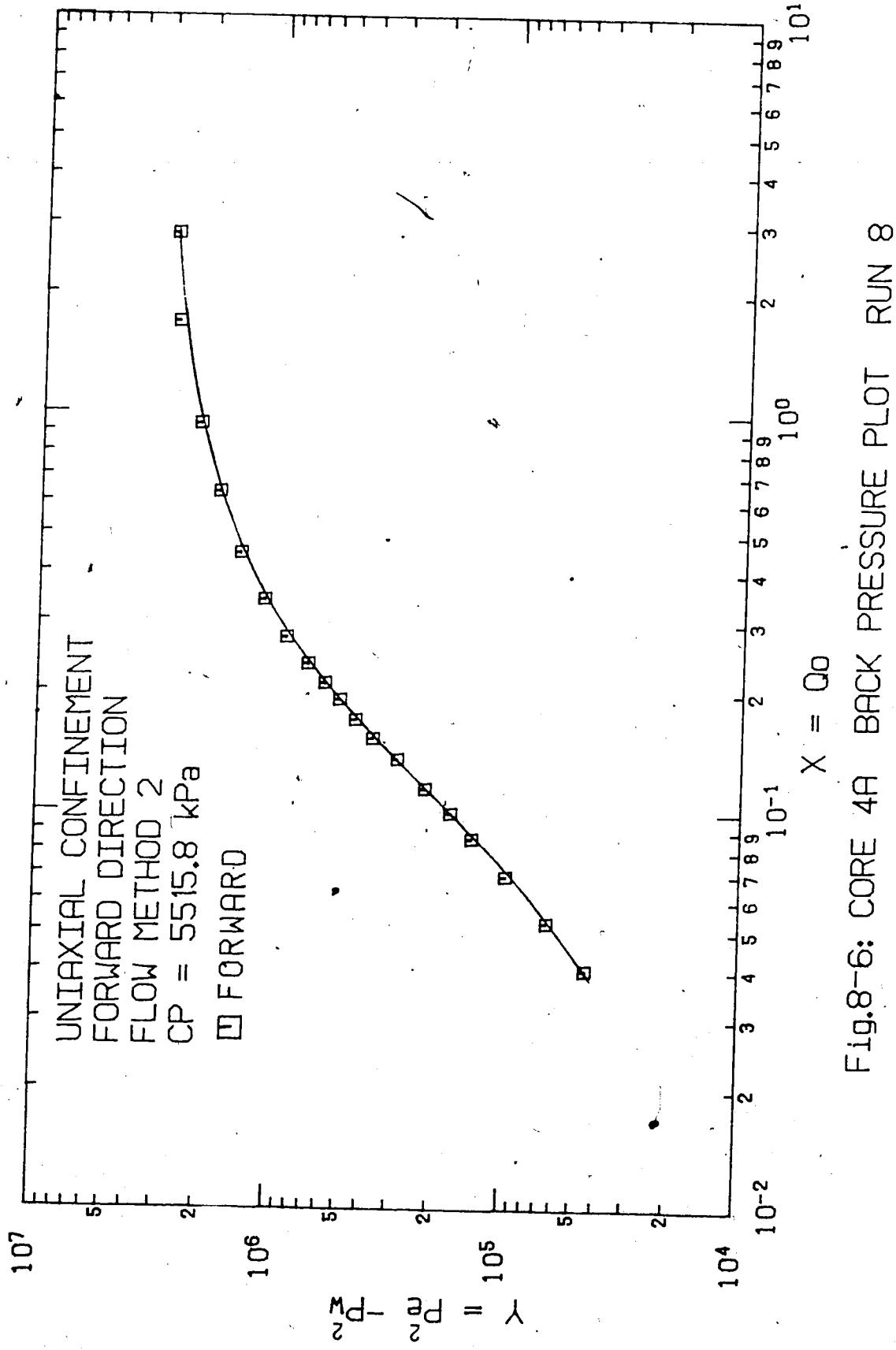
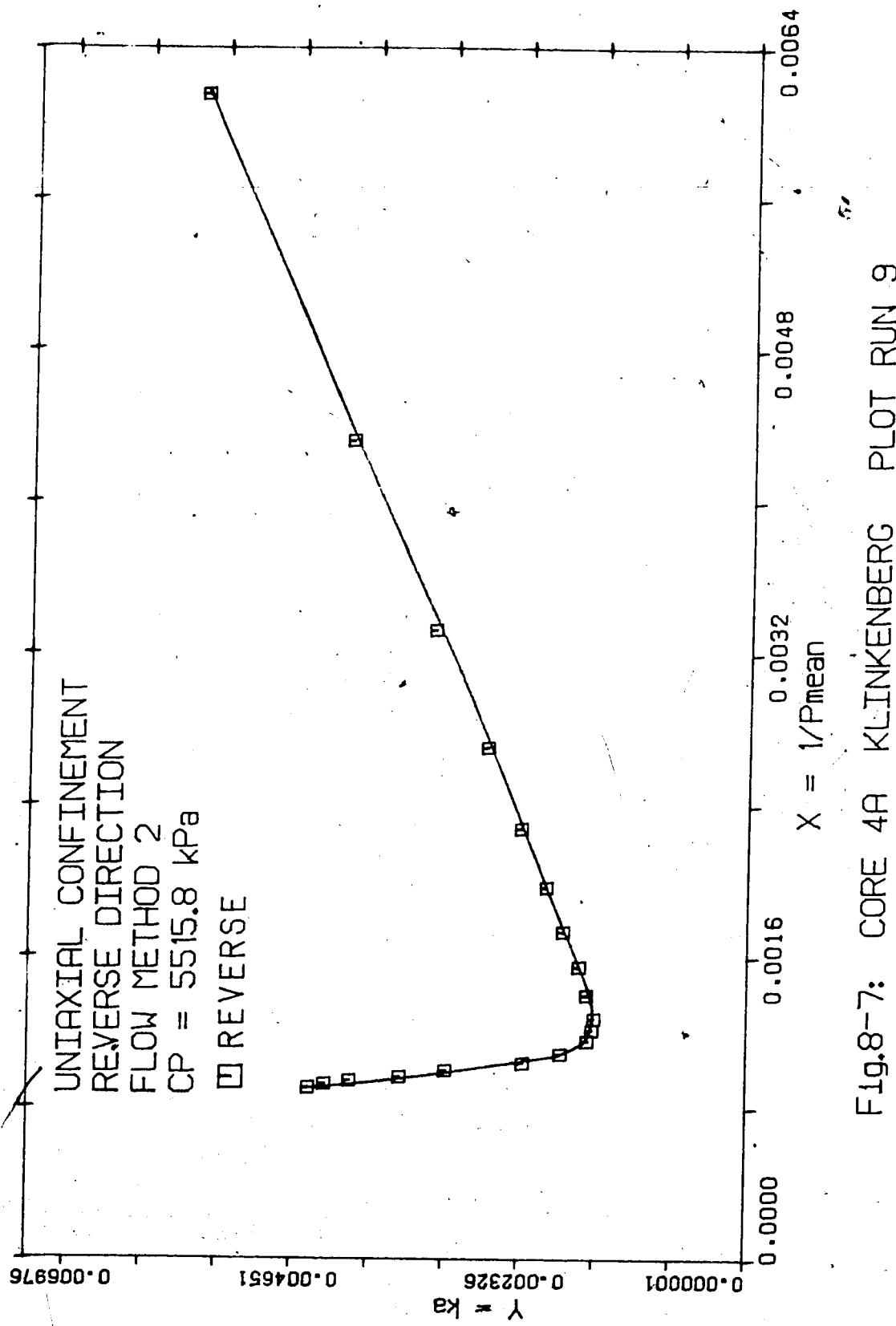


Fig.8-6: CORE 4A BACK PRESSURE PLOT RUN 8



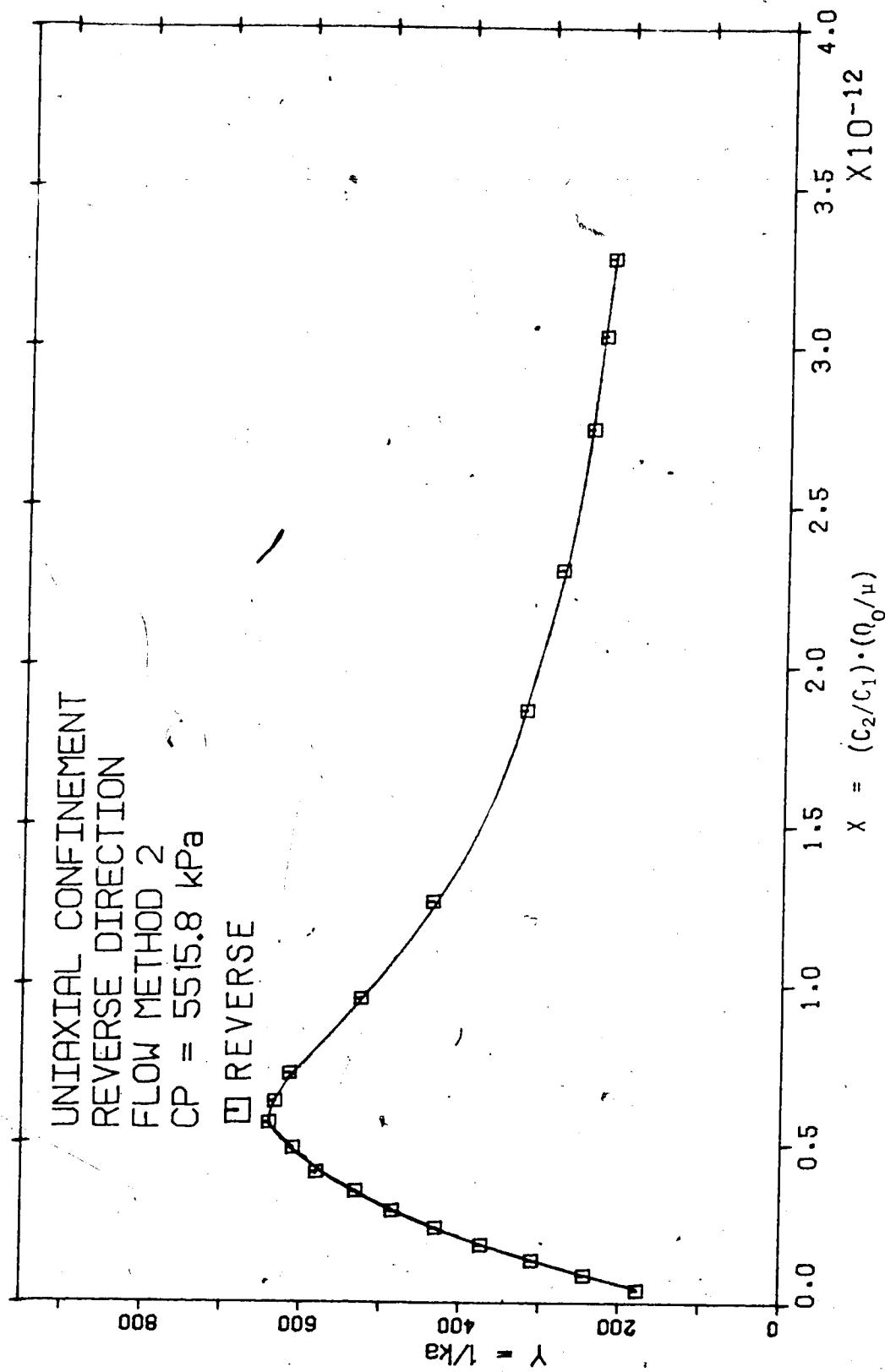


Fig.8-8: CORE 4A VISCO-INERTIAL PLOT RUN 9

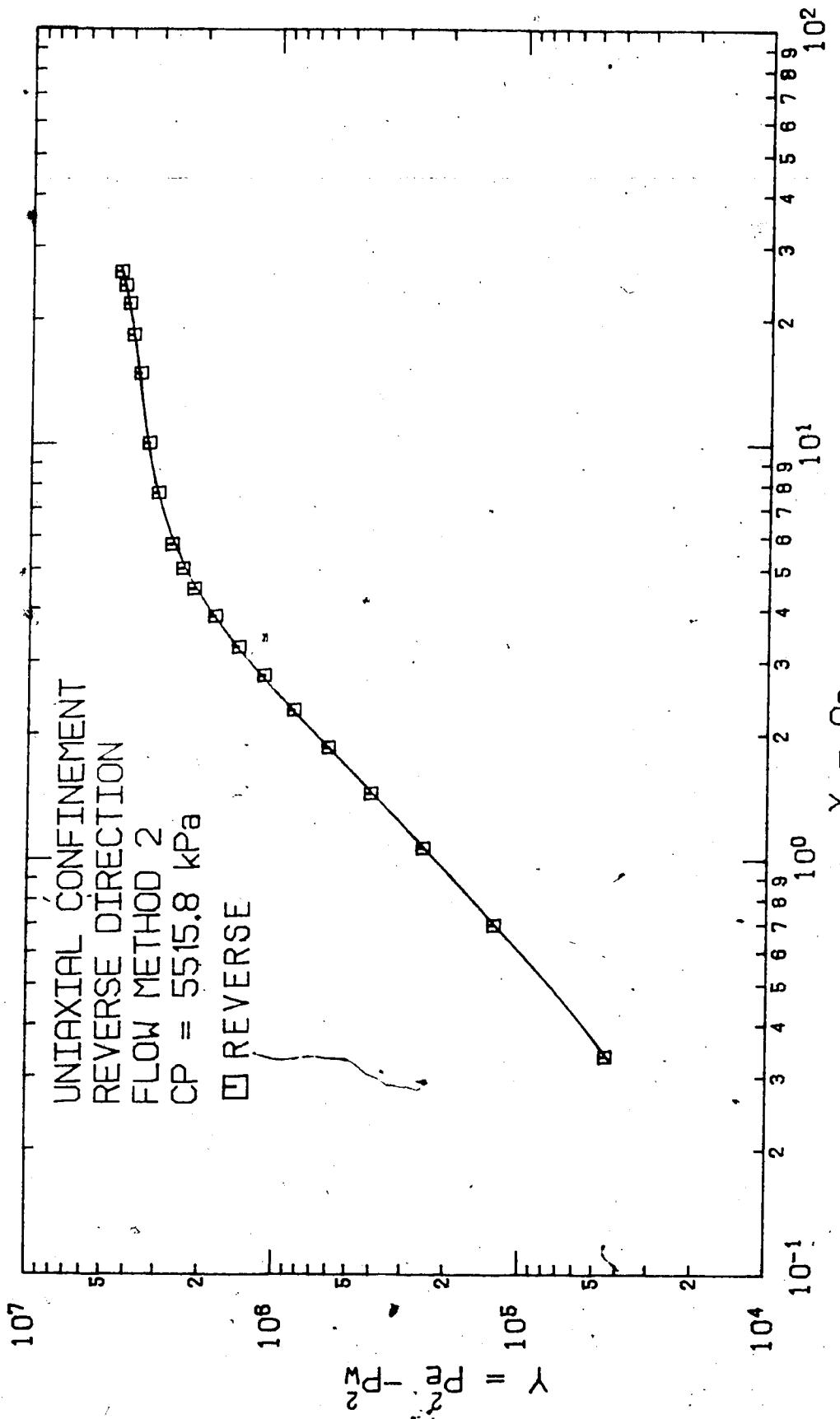


Fig.8-9: CORE 4A BACK PRESSURE PLOT RUN 9

In Run 8,  $P_w$  was kept constant at atmospheric pressure while  $P_e$  was increased. Run 9 is in the reverse direction as  $P_e$  decreased instead of being increased. The conditions for these runs are similar to Cases 2 and 1, respectively, in Table 3.1 of Chapter 3. If a similar analytical procedure is followed in both the Klinkenberg and the visco-inertial plots,  $k_a$  is expected to decrease for Run 8, and to increase for Run 9, with respect to both  $Q_o$  and  $P_m$ . Also, by using a similar analysis for the back-pressure plot,  $(P_e^2 - P_w^2)$  should increase with increasing  $Q_o$  for Run 8, and decrease with decreasing  $Q_o$  for Run 9. Generally, the plot profiles agree with predicted profiles at the earlier stage, but the prediction fails at the later stage where  $k_a$  increases with increasing  $Q_o$  and  $P_m$ .

### 8.3 Fitted Visco-Inertial and Back-Pressure Plots

In order to study further the observations obtained from the plot profiles, especially in the region of higher flow rates and pressures, fitted visco-inertial and back-pressure plots are made on the experimental data by using both cubic and quadratic model equations. The cubic equation, by having an additional term in gamma for increasing  $Q_o$ , is regarded as a modification to the quadratic equation.

Both the model cubic equation (Equation 4.15) and the model quadratic equation (Equation 4.17) are adapted, respectively, to Equations 4.23 and 4.21, and are used to

give fitted visco-inertial plots. Equations 4.23 and 4.21 are, respectively, quadratic and linear, and expressions on the left of these equations, which are the same for both equations, are used to calculate the experimental y values.

Fitted y values are calculated, respectively, from Equations 4.23 and 4.21 by using a multiple linear regression fit.

Both experimental and fitted y values are plotted on the same graph against CC2/CC1. For the fitted back-pressure plots, experimental data are, similarly, treated with Equations 4.24 and 4.25, which are, respectively, cubic and quadratic in Qo. Fitted y values are calculated by using a polynomial fit on cubic and quadratic model equations, and both experimental and fitted y values are plotted on the same log-log plot against Qo.

Figure 8-10, shows the fitted visco-inertial plot, for Core 1\$ Run 7 in the forward direction. Comparing the experimental data with the fitted plots obtained with both the cubic and the quadratic model flow equations, the experimental data have a closer fit with the cubic equation than with the quadratic equation. Specifically, the cubic fit is closer in the region of higher flow rates and pressures than in the region with lower flow rates and pressures.

The fitted back-pressure plots were generally poor. Erratic behavior was obtained on the fit with both model equations in the region of lower flow rates. Specifically, the quadratic equation yielded unreliable fitted values in

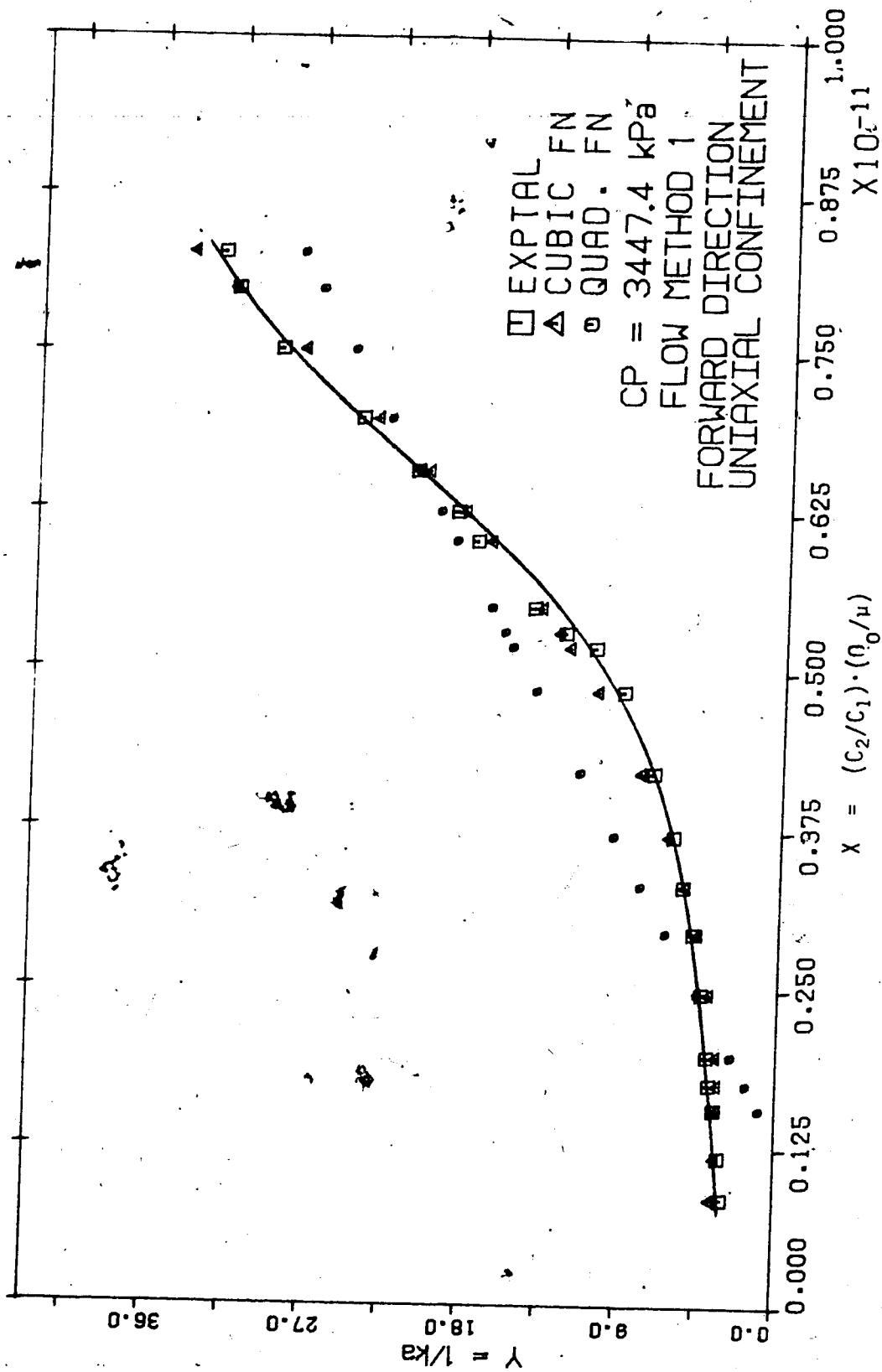


Figure 8-10 CORE 1C RUN 4 VISCO-INERTIAL FIT PLOT

the lower region. This made plotting in this section unreliable. Comparing the fits with the experimental data in the region of higher flow rates where both fits are reliable, the cubic equation gave a closer fit. Reasons for the poor fit in the region of lower flow rates could be attributed to the following:

- a) This is the viscous region where, by Darcy's law,  $(Pe^2 - Pw^2)$  has a linear relationship with  $Q_o$ . Therefore using higher order equations to fit this linear section should not be expected to produce good results.
- b) In the regression procedure, using Equations 4.24 and 4.25, known variable is  $Q_o$ , while the rest of the experimentally determined variables, as well as the parameters,  $k_a$ ,  $F_b$ , and gamma, which are yet to be estimated, are all lumped as the unknown variable. By having large unknowns in a system that is not very stable in its matrix formation, (problem of ill-conditioning, which is treated in Appendix B), erratic behavior could be the case. This is reflected in the determined values of the determinants, and the error sum of squares.

Figure 8.11, which is the fitted back-pressure plot for Core 1C, Run 3 in the forward direction, shows the experimental data and the fitted cubic equation. More plots are shown in Appendix C.

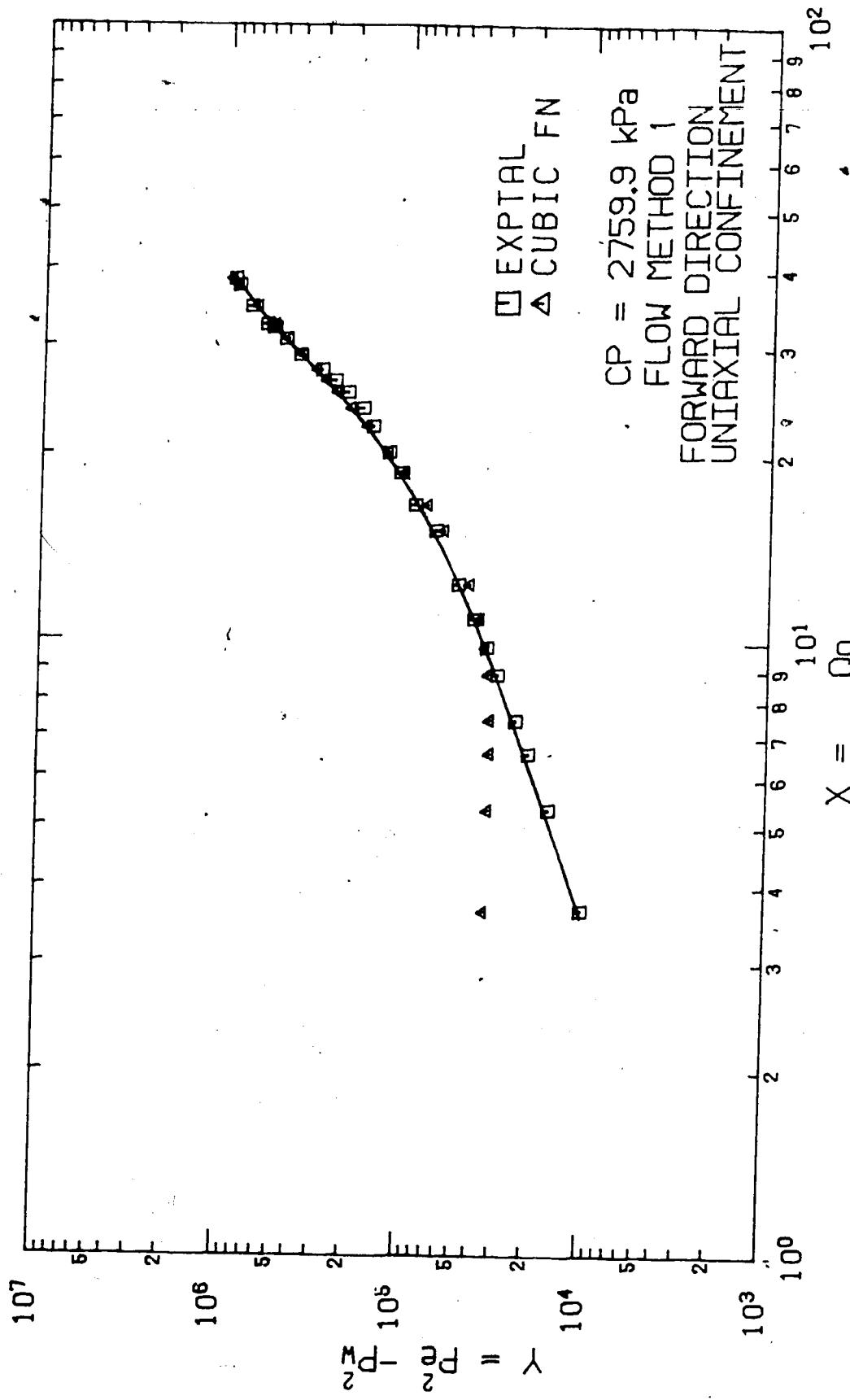


Figure 8-11 CORE 1C RUN 3 BACK PRESSURE FIT PLOT

#### 8.4 Effect of Stress and Flow Methods on Plot Profiles

In the preceding sections, lower  $(Pe^2 - Pw^2)$  values were observed in the plot profiles for increasing  $Q_o$  and  $k_a$ . In this section, a possible explanation to this phenomenon is attempted by analysing the back-pressure flow relationship as given by Equation 3.8. By this equation,  $(Pe^2 - Pw^2)$  is proportional to  $Q_o$ , and is a product of  $P_m$  and  $(Pe - P_w)$ . While  $P_m$  provides magnitude,  $(Pe - P_w)$  provides both magnitude and direction, and therefore, these effects manifest themselves according to the respective magnitudes of  $Pe$  and  $P_w$ .

By Flow Method 1,  $P_m$  increases at a faster rate than  $(Pe - P_w)$ , and by Flow Method 2,  $(Pe - P_w)$  increases at a faster rate than  $P_m$ . With increasing pressures, which corresponds to increasing  $Q_o$ , the effect of  $P_m$  increases more for Flow Method 1, while the effect of  $(Pe - P_w)$  increases more for Flow Method 2. Both effects lead to increased stresses on the core material. The direct effect of obtaining large drawdown pressures may lead to radial compaction. For the same pressure conditions, such stress may be attained faster, and possibly be more acute, with Flow Method 2 than with Flow Method 1. Depending on the strength of the core samples, this stressed condition may generate cracks and fractures which in turn create increased permeability with resultant increases in  $Q_o$ . Therefore, the overall effect is to cause increases in  $k_a$  for increasing  $Q_o$  and lower  $(Pe^2 - Pw^2)$ . This predicted behavior is similar to what was

observed at higher pressures while using Flow Method 1, and at moderate pressures while using Flow Method 2.

The anomalous behavior from these effects suggests a probable confinement pressure problem. Figure 8.12 of Core 1C Run 3, shows, for Flow Method 1, the variation of  $k_a$  with generated confining pressure, CP, which was obtained from Equation 5.18. The plot shows a decreasing  $k_a$  profile with increasing CP, which is similar to the findings of Fatt(72) and Senturk(73). Figure 8.13 shows, for Core 4A, Run 9, the variation of  $k_a$  with CP using Flow Method 2. The value of  $k_a$  decreases with generated CP after which  $k_a$  increases for further increases in CP..

After Run 9 on Core 4A, the radial cell was dismantled and a crack was noticed on the core sample as shown in Plates 1 and 2 . No crack was noticed on Core 1C, and on Core 4A prior to making runs on it with Flow Method 2. It is significant to note that both the anomalous plot profile in Figure 8.13 and the crack were observed with respect to the use of Flow Method 2 in Run 9, Core 4A. It is, therefore, probable that the use of Flow Method 2 could be responsible for the crack.

#### 8.5 Plot Profiles using a Tri-axial Radial Cell

Runs 1 and 2 were made using Flow Method 1 in forward and reverse directions, respectively. Run 3 in the forward direction, and Run 4 in the reverse direction, were similarly made using Flow Method 2. The plot profiles show

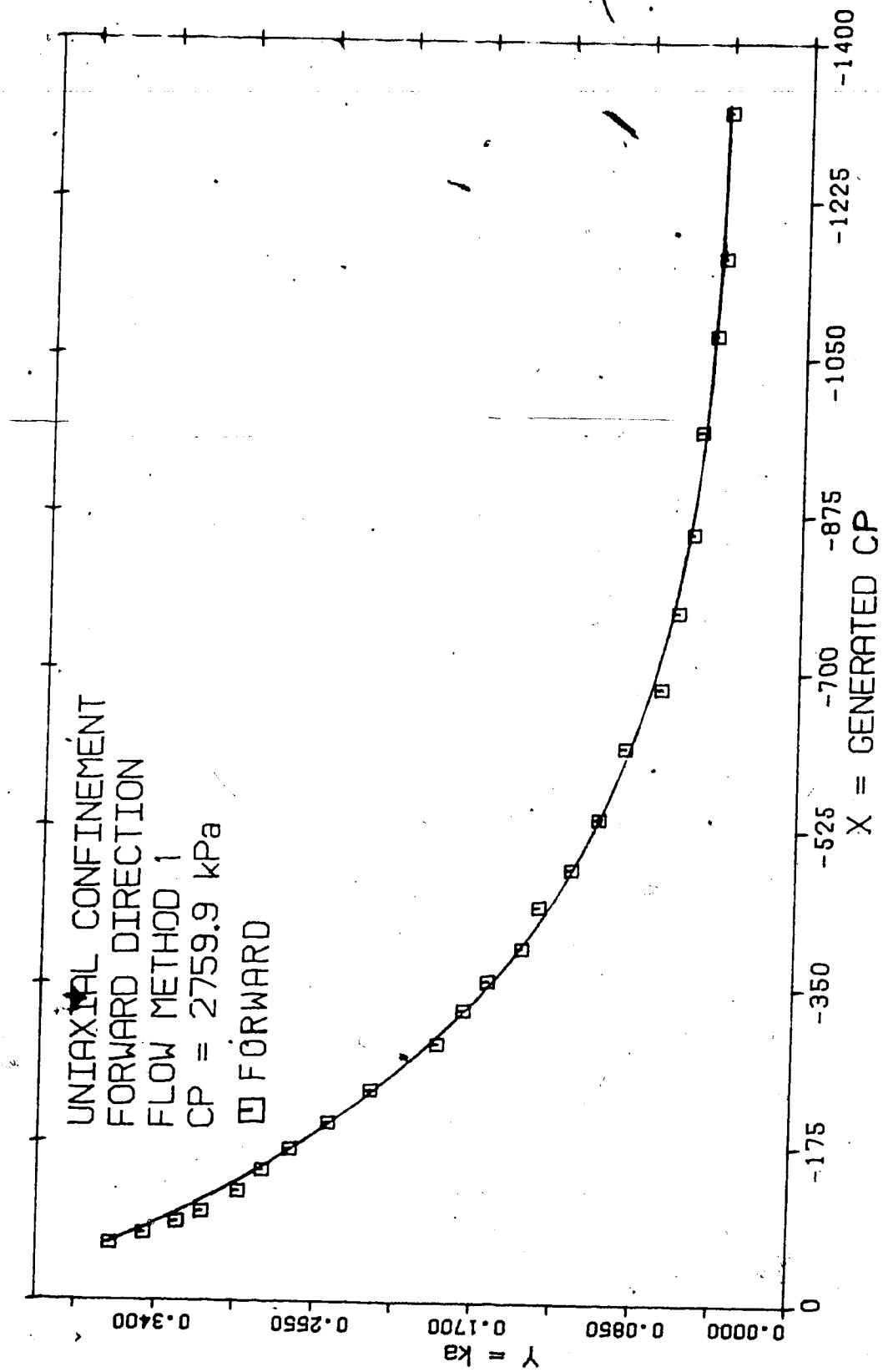


FIGURE 8-12: CORE 1C GENERATED CP VERSUS ka RUN 3

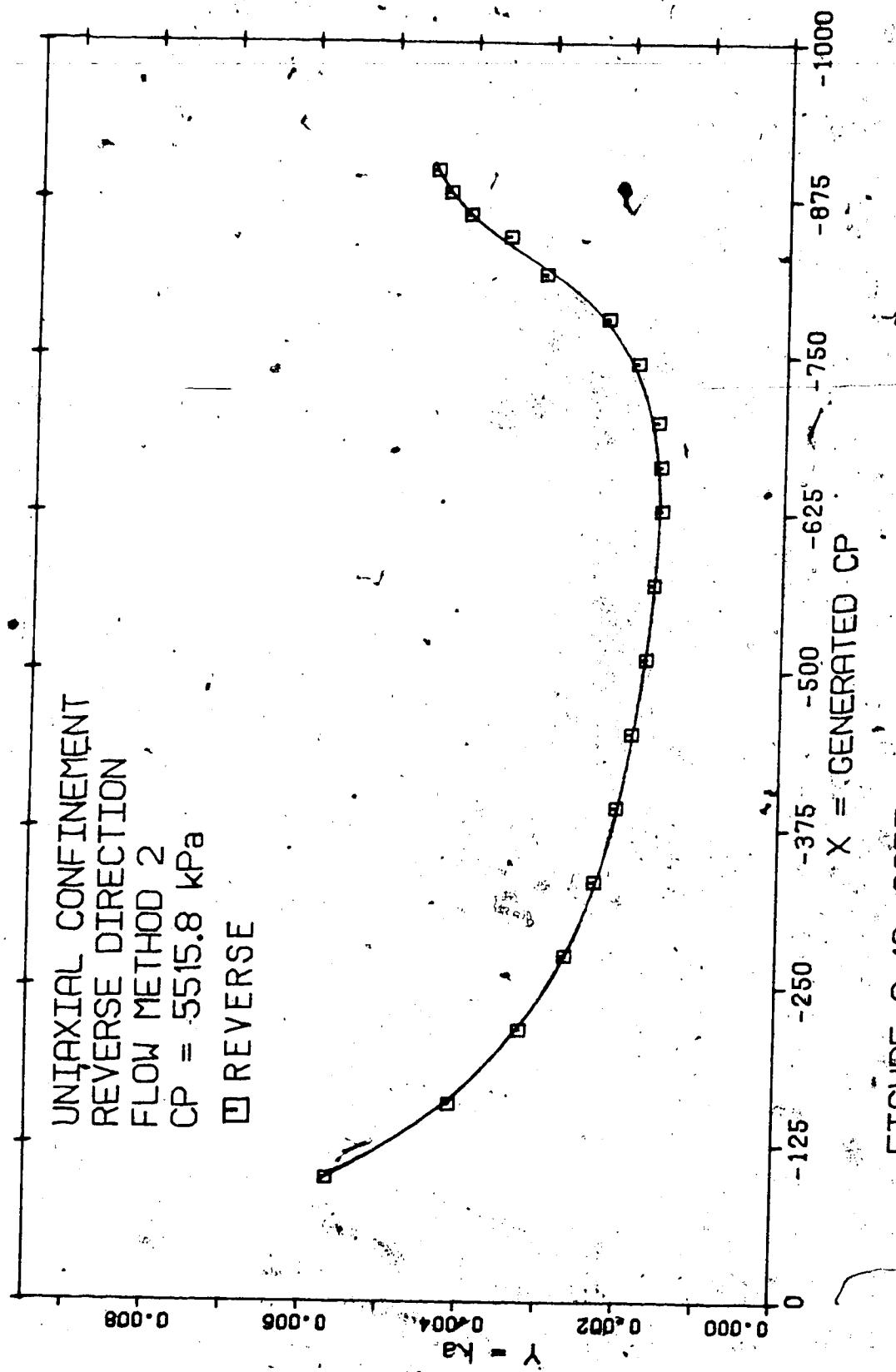


FIGURE 8-13: CORE 4A GENERATED CP WITH  $k_a$  RUN 9

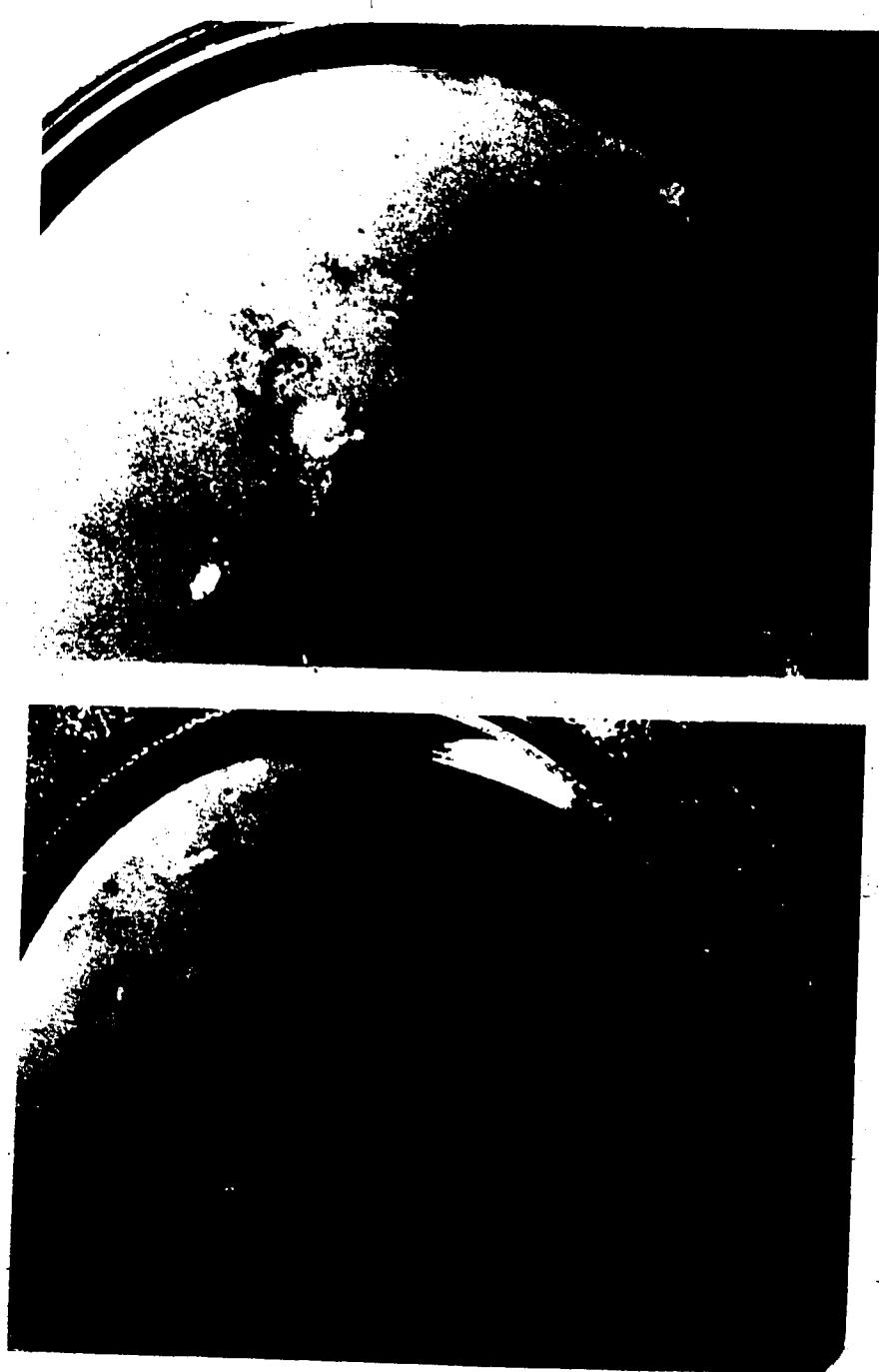


Plate 8:1 Photographs of Core 4A showing Fracture

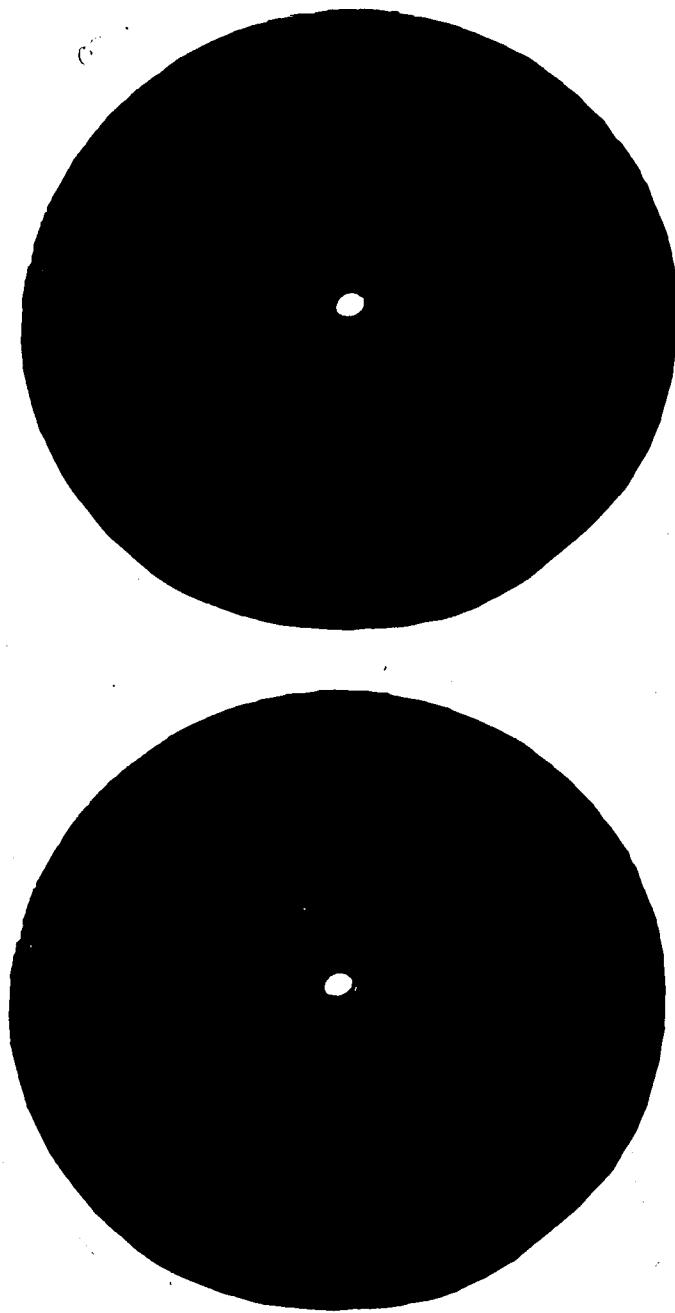


Plate 8:2 Photographs of Rubber Gaskets showing  
marks of fracturing of Core 4A

distinct differences between the flow methods used. Figures 8-14 and 8-15 show, respectively, the Klinkenberg and the visco-inertial plot profiles for Run 1 using Flow Method 1. They have anomalous behavior similar to that obtained previously with Flow Method 2, when a uniaxial radial cell was used. The cause of this is unknown, but it is suspected that, possibly, the loose rubber overburden sleeve could not give adequate confinement at higher pressures. Anomalous behavior started occurring between 2800.0 kPa and 2940.0 kPa. The predefined CP, which was kept constant at 4136.9 kPa, could possibly have been inadequate for complete confinement at higher fluid pressures, and as the fluid pressures approached its predefined value.

Figures 8-16 and 8-17 show respectively the Klinkenberg and visco-inertial plot profiles for Run 3, using Flow Method 2. The plots have expected profiles.

## 8.6 Parameter Estimation Results

### 8.6.1 Graphical and Linear Fit Methods

Estimated parameter values from both graphical and linear fit methods are summarized in Tables 8-1, 8-2, 8-3, 8-4. All runs are with Flow Method 1, except Run 7 on Core 4A, and Run 1SEN from Senturk's data(74), which are both with Flow Method 2.

The graphical method is limited in the choice of the best fit line. Proper delineation of viscous plot points from both Klinkenberg and back-pressure plots by visual

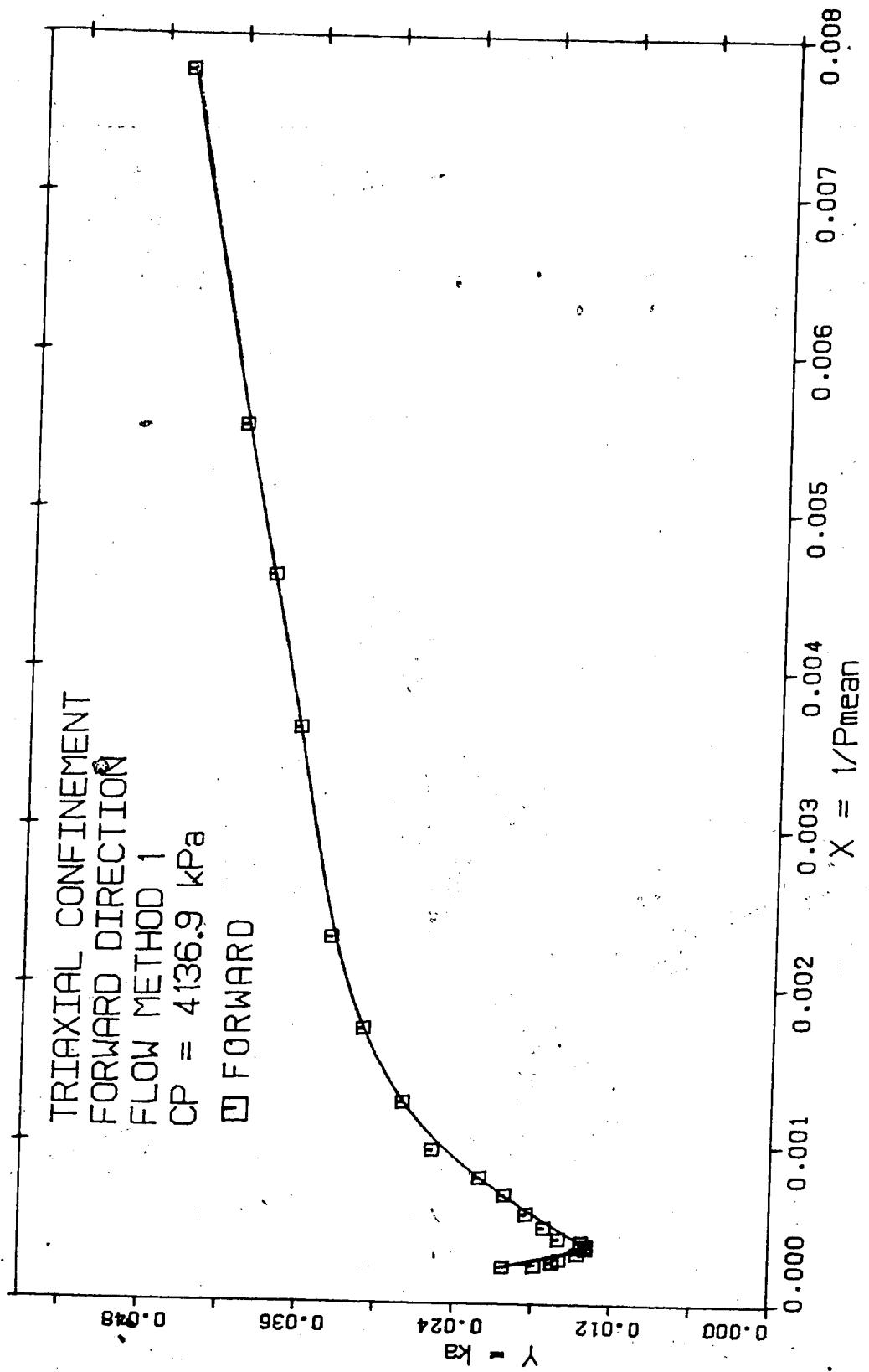


Figure 8-14: CORE 2 KLINKENBERG PLOT RUN 1

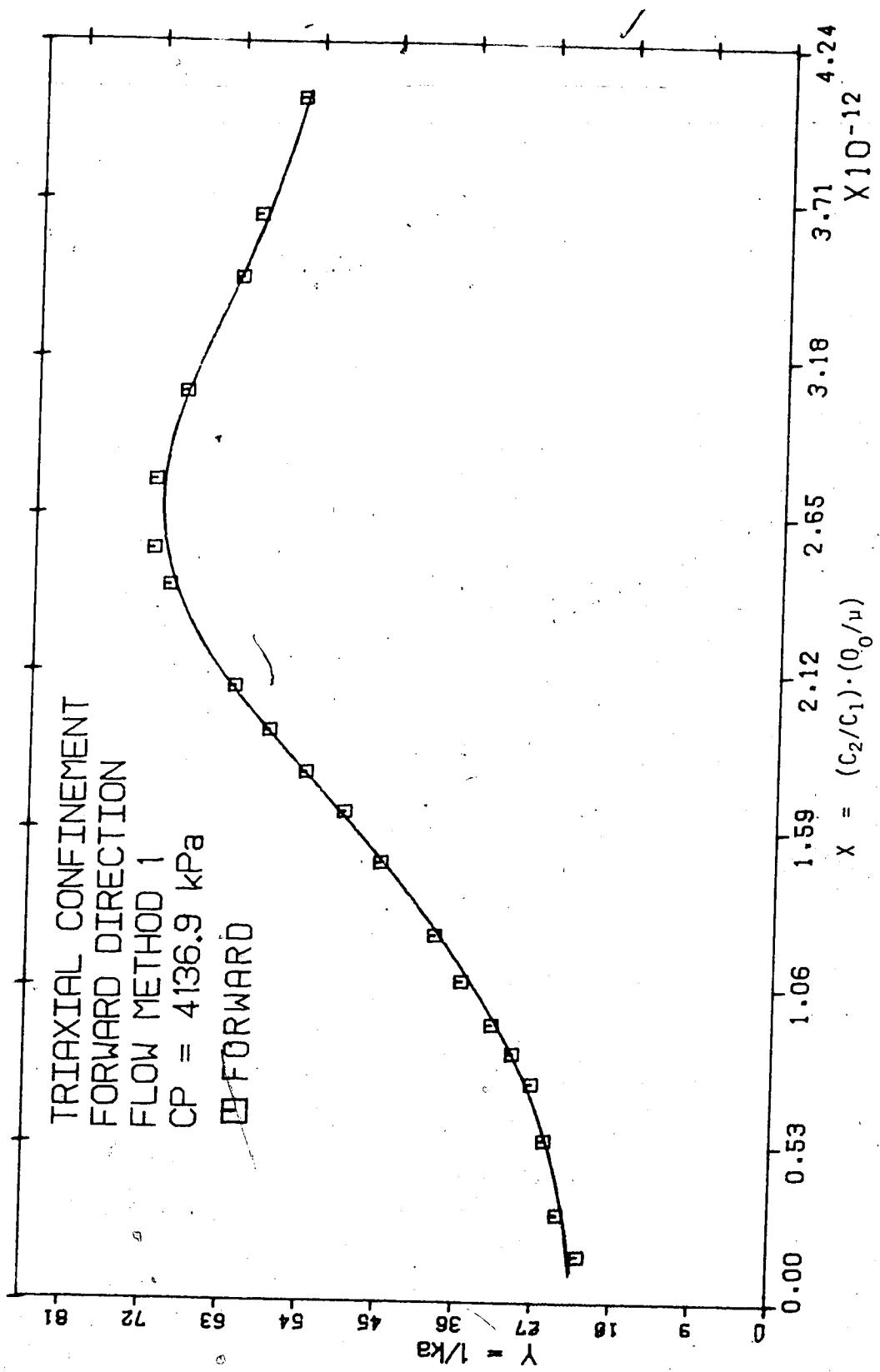


Figure 8-15: CORE 2 VISCO-INERTIAL PLOT RUN 1

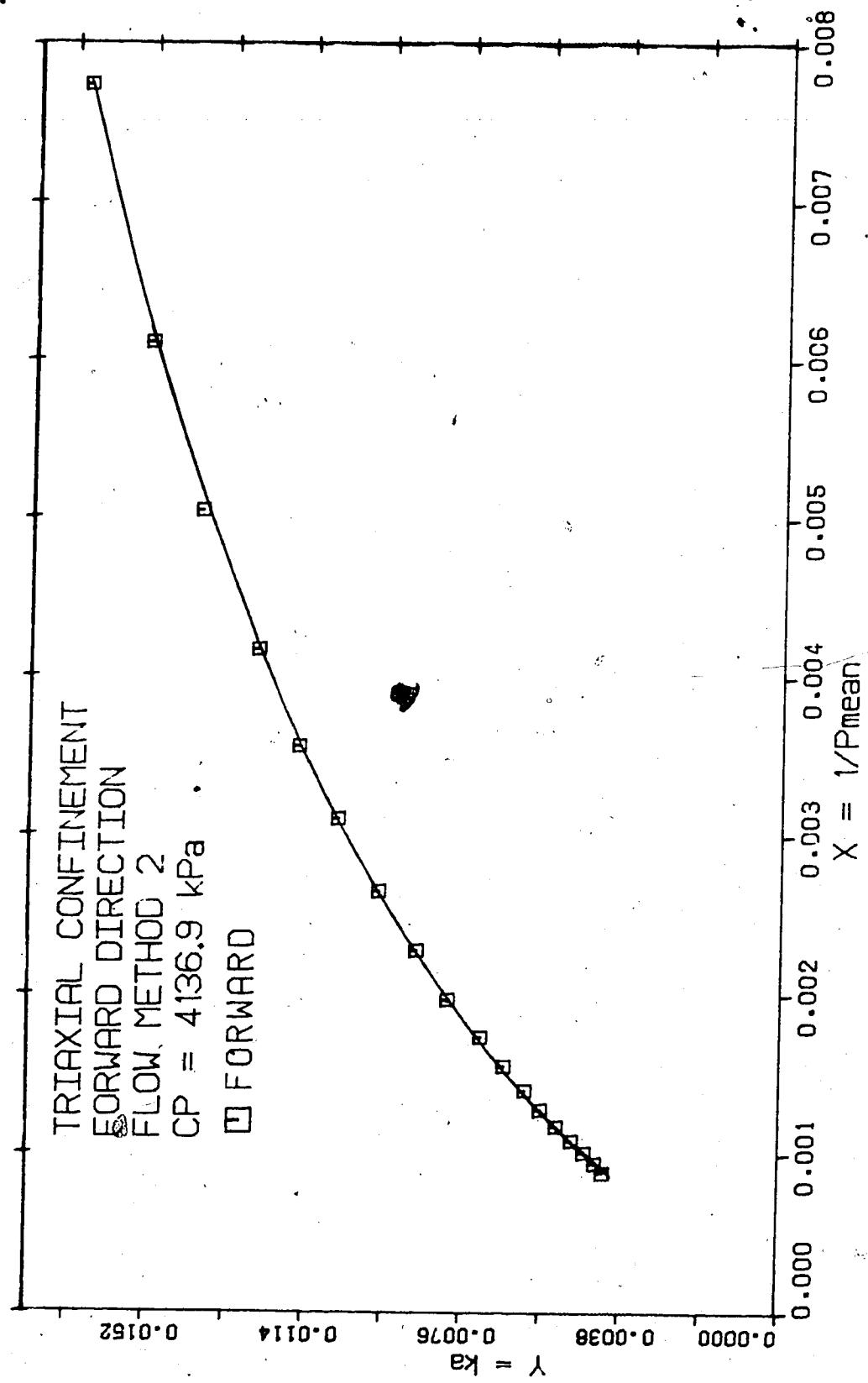


Figure 8-16: CORE 2 KLINKENBERG PLOT RUN 3

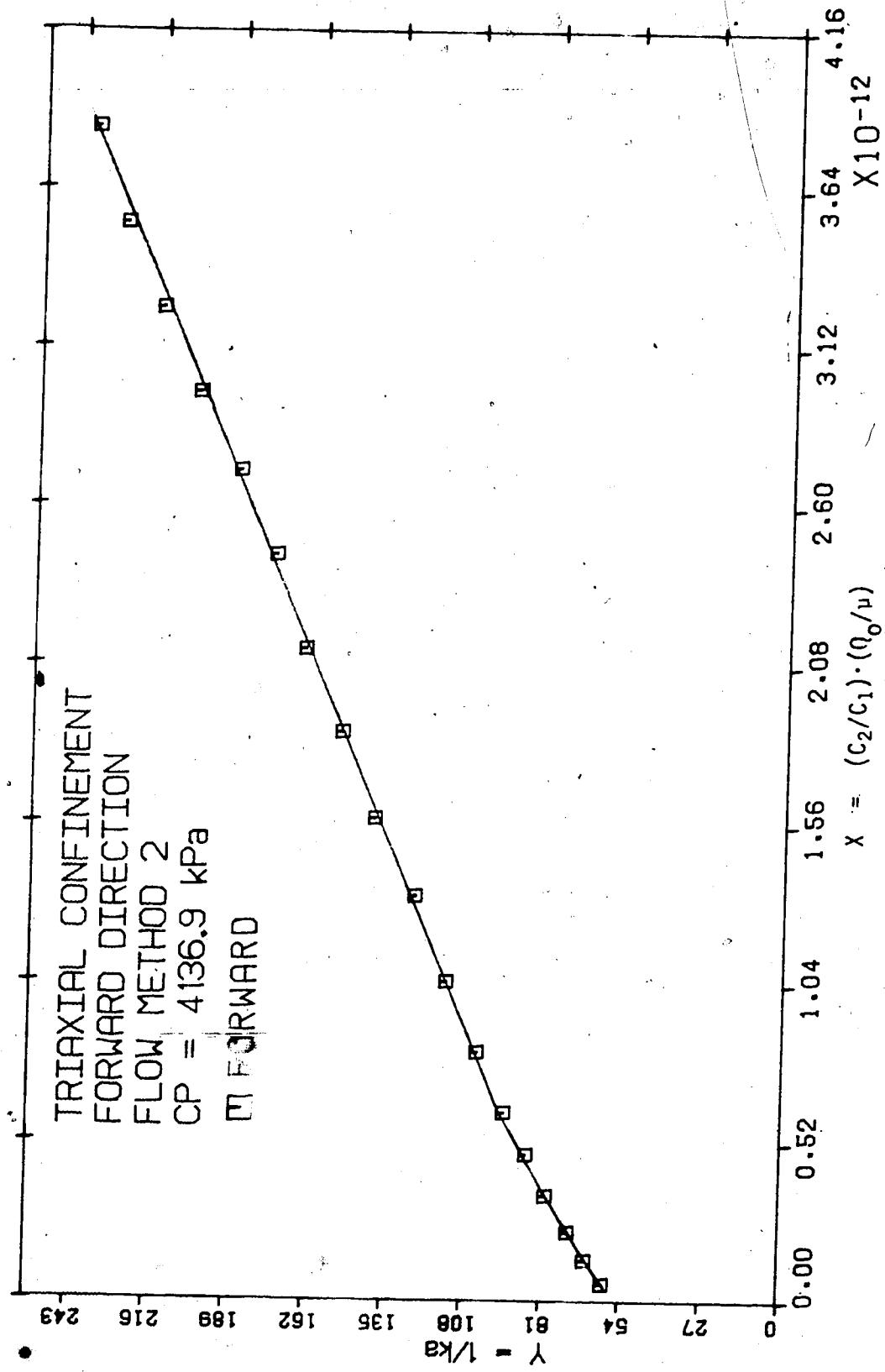


Figure 8-17: CORE 2 VISCO-INERTIAL PLOT RUN 3

TABLE 8-1: GRAPHICAL PARAMETER ESTIMATION FOR CORE 1C

## FORWARD RUNS

| Run<br>NO | CP<br>kPa | K<br>$\mu\text{m}^2$ | b<br>kPa | $k_{abs}$<br>$\mu\text{m}^2$ | $F_b$<br>$\text{m}^{-1}$<br>$\times 10^{12}$ |
|-----------|-----------|----------------------|----------|------------------------------|--|
| 2         | NILL      | 0.1200               | 164.06   | 0.2983                       | 0.7390                                       |
| 3         | 2759.9    | 0.2205               | 82.831   | 0.4444                       | 0.4688                                       |
| 4         | 3447.4    | 0.1761               | 116.93   | 0.4460                       | 0.8391                                       |
| 5         | 4136.9    | 0.2000               | 100.00   | 0.3729                       | 0.3089                                       |
| 6         | 4826.3    | 0.1469               | 140.15   | 0.2798                       | 0.4806                                       |
| 7         | 5515.8    | 0.1455               | 175.78   | 0.4522                       | 0.7252                                       |

## REVERSE RUNS

|   |        |        |        |        |        |
|---|--------|--------|--------|--------|--------|
| 3 | 2759.9 | 0.1546 | 87.500 | 0.2764 | 0.3791 |
| 4 | 3447.4 | 0.1172 | 216.18 | 0.3148 | 0.9349 |
| 5 | 4136.9 | 0.1846 | 141.24 | 0.5152 | 0.4001 |
| 6 | 4826.3 | 0.1739 | 141.23 | 0.4928 | 0.5751 |
| 7 | 5515.8 | 0.1247 | 117.93 | 0.2698 | 0.8991 |

TABLE 8-2: GRAPHICAL PARAMETER ESTIMATION FOR CORE 4A

## FORWARD RUNS

| Run<br>No | CP<br>kPa | $k$<br>$\mu\text{m}^2$ | $b$<br>kPa | $k_{abs}$<br>$\mu\text{m}^2$ | $R_b$<br>$\text{m}^{-1}$<br>$\times 10^{12}$ |
|-----------|-----------|------------------------|------------|------------------------------|--|
| 1         | NILL      | 0.1031                 | 298.30     | 0.6267                       | 2.1693                                       |
| 2         | 2759.9    | 0.2455                 | 81.250     | 0.4134                       | 0.4222                                       |
| 3         | 3447.4    | 0.1525                 | 235.36     | 0.5812                       | 0.9806                                       |
| 4         | 4136.9    | 0.1478                 | 79.942     | 0.2576                       | 1.0294                                       |
| 5         | 4826.3    | 0.1334                 | 180.69     | 0.4231                       | 1.2672                                       |
| 6         | 5515.8    | 0.1841                 | 161.29     | 0.5000                       | 1.0181                                       |

## REVERSE RUNS

|      |        |        |        |        |         |
|------|--------|--------|--------|--------|---------|
| 1.   | NILL   | 0.1250 | 212.50 | 0.3626 | 0.9160  |
| 2    | 2759.9 | 0.1388 | 119.48 | 0.2471 | 0.8118  |
| 3    | 3447.4 | 0.1279 | 112.07 | 0.2491 | 1.2426  |
| 4    | 4136.9 | 0.1200 | 102.43 | 0.2394 | 1.4216  |
| 5    | 4826.3 | 0.1338 | 169.23 | 0.4681 | 1.7179  |
| 6    | 5515.8 | 0.1090 | 164.21 | 0.2787 | 2.3077  |
| 7    | 5515.8 | 0.0002 | 233.71 | 0.0003 | 11117.5 |
| 1SEN | 4826.3 | 0.0086 | 24.713 | 0.0079 | 28.581  |

TABLE 8-3: PARAMETER ESTIMATION BY LINEAR FIT FOR CORE 1C  
( Using Truncated Plot Points)

FORWARD RUNS

| Run No | CP kPa | K $\mu\text{m}^2$ | b kPa  | Kcor $\mu\text{m}^2$ | Kabs $\mu\text{m}^2$ | Fb $\text{m}^{-1}$<br>$\times 10^{12}$ |
|--------|--------|-------------------|--------|----------------------|----------------------|--|
| 2      | NILL   | 0.1027            | 217.28 | 0.1036               | 0.3797               | 1.0991                                 |
| 3      | 2759.9 | 0.2257            | 79.622 | 0.2256               | 0.4502               | 0.5202                                 |
| 4      | 3447.4 | 0.1717            | 123.44 | 0.1726               | 0.4532               | 0.8627                                 |
| 5      | 4136.9 | 0.1926            | 110.77 | 0.1929               | 0.4040               | 0.3787                                 |
| 6      | 4826.3 | 0.1422            | 149.67 | 0.1425               | 0.3302               | 0.6233                                 |
| 7      | 5515.8 | 0.1454            | 176.94 | 0.1453               | 0.4537               | 0.7262                                 |

REVERSE RUNS

|   |        |        |        |        |        |        |
|---|--------|--------|--------|--------|--------|--------|
| 2 | NILL   | 0.0253 | 704.06 | 0.0261 | 0.0505 | 1.8252 |
| 3 | 2759.9 | 0.1547 | 88.223 | 0.1548 | 0.2751 | 0.3680 |
| 4 | 3447.4 | 0.1150 | 224.85 | 0.1167 | 0.4790 | 1.2504 |
| 5 | 4136.9 | 0.1872 | 154.62 | 0.1871 | 0.5139 | 0.3880 |
| 6 | 4826.3 | 0.1726 | 150.45 | 0.1725 | 0.4841 | 0.5690 |
| 7 | 5515.8 | 0.1171 | 143.20 | 0.1190 | 0.2777 | 0.9337 |

TABLE 8-4: PARAMETER ESTIMATION BY LINEAR FIT FOR CORE 4A  
( Using Truncated Plot Points)

| FORWARD RUNS |        |                   |        |                      |                      |                                   |
|--------------|--------|-------------------|--------|----------------------|----------------------|-----------------------------------|
| Run No       | CP kPa | K $\mu\text{m}^2$ | b kPa  | Kcor $\mu\text{m}^2$ | Kabs $\mu\text{m}^2$ | Fb $\text{m}^{-1} \times 10^{12}$ |
| 1            | NILL   | 0.1052            | 288.59 | 0.1052               | 0.6302               | 2.1869                            |
| 2            | 2759.9 | 0.2290            | 97.792 | 0.2288               | 0.4550               | 0.5452                            |
| 3            | 3447.4 | 0.1506            | 240.44 | 0.1518               | 0.6269               | 1.0731                            |
| 4            | 4136.9 | 0.1457            | 81.112 | 0.1457               | 0.2610               | 1.0245                            |
| 5            | 4826.3 | 0.1357            | 175.59 | 0.1356               | 0.4151               | 1.2334                            |
| 6            | 5515.8 | 0.1513            | 232.52 | 0.1504               | 0.7720               | 1.5253                            |
| REVERSE RUNS |        |                   |        |                      |                      |                                   |
| 1            | NILL   | 0.1141            | 245.84 | 0.1120               | 0.5286               | 1.7325                            |
| 2            | 2759.9 | 0.1338            | 131.23 | 0.1134               | 0.3232               | 1.4047                            |
| 3            | 3447.4 | 0.1234            | 123.06 | 0.1235               | 0.3020               | 1.8584                            |
| 4            | 4136.9 | 0.1191            | 102.87 | 0.1191               | 0.2411               | 1.4206                            |
| 5            | 4826.3 | 0.1288            | 180.93 | 0.1297               | 0.5011               | 1.7708                            |
| 6            | 5515.8 | 0.1096            | 161.29 | 0.1100               | 0.3562               | 2.0734                            |
| 7            | 5515.8 | 0.0002            | 244.88 | 0.0002               | 0.0004               | 16593.                            |
| 1SEN         | 4826.3 | 0.0055            | 102.28 | 0.0058               | 0.0098               | 36.042                            |

inspection may not always be quite accurate. By linear fitting those points suspected to be in the viscous region, and checking for the value of the linear correlation coefficient,  $R_{aa}$ , the best estimates of  $K$  and  $b$ , based on the method of least squares, are obtained. Similar treatment by linear fitting is carried out for the visco-inertial case.

The choice of  $b$  for the modified visco-inertial plot influences not only the value of the corrected  $K$ ,  $K_{cor}$ , but also the value of the corrected  $F_b$ . Poor values of  $b$ , reflected in the  $R_{aa}$  value being far from unity, gives poor values of the corrected parameters.

#### 8.6.2 Numerical Methods

Simultaneous parameter estimation has been undertaken by using:

- (a) the cubic equation (Equation 4.15),
- (b) the slippage corrected versions of the cubic and quadratic equations (Equations 4.16 and 4.18),
- (c) Case(b) when their combined variables are neglected.

They are summarized in Tables 8-5, 8-6, 8-7, 8-8.

Application of these cases have been made on both complete and truncated plot points. Complete plot points represent all the plot points; they span both higher and lower flow rates and pressures. Truncated plot points are those plot points which were used for the linear fit parameter estimation. They represent plot points with lower flow rates

TABLE 8-5: NUMERICAL PARAMETER ESTIMATION FOR CORE 1C  
(Using Complete Plot Points)

FORWARD RUNS

| Run No | CP kPa | ka $\mu\text{m}^2$ | Fb $\text{m}^{-1} \times 10^{12}$ | Gamma msec/kg $\times 10^{16}$ |
|--------|--------|--------------------|-----------------------------------|--------------------------------|
| 2      | NILL   | 0.3070             | 6.9138                            | 0.7701                         |
| 3      | 2759.9 | 0.2055             | 2.8954                            | 0.5431                         |
| 4      | 3447.4 | 0.2082             | 5.8621                            | 0.9165                         |
| 5      | 4136.9 | 0.1268             | 6.6411                            | 1.1546                         |
| 6      | 4826.3 | 0.0615             | 9.9375                            | 2.1428                         |
| 7      | 5515.8 | 0.0734             | 10.609                            | 2.1489                         |

REVERSE RUNS

|   |        |        |        |        |
|---|--------|--------|--------|--------|
| 2 | NILL   | 0.0600 | 3.1942 | 0.6476 |
| 3 | 2759.9 | 0.0100 | 4.8932 | 0.7332 |
| 4 | 3447.4 | 0.1086 | 5.5596 | 1.0283 |
| 5 | 4136.9 | 0.0954 | 7.4485 | 1.3169 |
| 6 | 4826.3 | 0.0643 | 11.038 | 2.2152 |
| 7 | 5515.8 | 0.0626 | 11.984 | 2.5433 |

TABLE 8-6: NUMERICAL PARAMETER ESTIMATION FOR CORE 4A  
(Using Complete Plot Points)

FORWARD RUNS

| Run No | CP KPa | ka $\mu\text{m}^2$ | Fb $\text{m}^{-1} \times 10^{12}$ | Gamma $\text{msec/kg} \times 10^{16}$ |
|--------|--------|--------------------|-----------------------------------|---------------------------------------|
| 1      | NILL   | 0.1046             | 17,890                            | 6.5656                                |
| 2      | 2759.9 | 0.1002             | 12.162                            | 6.5956                                |
| 3      | 3447.4 | 0.0702             | 23.051                            | 13.151                                |
| 4      | 4136.9 | 0.0705             | 22.392                            | 12.030                                |
| 5      | 4826.3 | 0.0550             | 23.510                            | 11.878                                |
| 6      | 5515.8 | 0.0501             | 32.302                            | 14.168                                |

REVERSE RUNS

|      |        |         |        |        |
|------|--------|---------|--------|--------|
| 1    | NILL   | 0.1116  | 15.605 | 7.3886 |
| 2    | 2759.9 | 0.0944  | 11.700 | 5.8221 |
| 3    | 3447.4 | 0.0632  | 21.893 | 11.673 |
| 4    | 4136.9 | 0.0565  | 20.636 | 13.171 |
| 5    | 4826.3 | 0.0950  | 18.744 | 9.7133 |
| 6    | 5515.8 | 0.0658  | 21.609 | 10.651 |
| 7    | 5515.8 | 0.00044 | 33240. | 76140. |
| 1SEN | 4826.3 | 0.0101  | 52.290 | 4.3530 |

TABLE 8-7: NUMERICAL PARAMETER ESTIMATION FOR CORE 1C  
(Using Truncated Plot Points)

FORWARD RUNS

| Run No. | CP kPa | K $\mu\text{m}^2$ | b kPa  | Kabs $\mu\text{m}^2$ | Fb $\text{m}^{-1} \times 10^{12}$ | Gamma $\text{msec/kg} \times 10^{16}$ |
|---------|--------|-------------------|--------|----------------------|-----------------------------------|---------------------------------------|
| 2       | NILL   | 0.1106            | 140.51 | 0.2341               | 0.8016                            | 0.3673                                |
| 3       | 2759.9 | 0.1956            | 104.67 | 0.3853               | 0.3519                            | 0.2091                                |
| 4       | 3447.4 | 0.1725            | 112.75 | 0.4529               | 0.4256                            | 0.1782                                |
| 5       | 4136.9 | 0.1673            | 114.43 | 0.3741               | 0.1881                            | 0.0438                                |
| 6       | 4826.3 | 0.1563            | 126.39 | 0.3984               | 0.1014                            | 0.1168                                |
| 7       | 5515.8 | 0.1420            | 128.33 | 0.4298               | 0.1929                            | 0.3864                                |

REVERSE RUNS

|   |        |        |        |        |        |        |
|---|--------|--------|--------|--------|--------|--------|
| 2 | NILL   | 0.0356 | 307.95 | 0.0600 | 1.0936 | 0.6476 |
| 3 | 2759.9 | 0.0497 | 515.45 | 0.2478 | 0.1109 | 0.0502 |
| 4 | 3447.4 | 0.1382 | 176.49 | 0.9372 | 0.2850 | 0.0932 |
| 5 | 4136.9 | 0.1744 | 159.08 | 0.4273 | 0.0471 | 0.0802 |
| 6 | 4826.3 | 0.1954 | 123.03 | 0.4664 | 0.1023 | 0.0191 |
| 7 | 5515.8 | 0.1025 | 148.61 | 0.2444 | 0.4687 | 0.1212 |

TABLE 8-8: NUMERICAL PARAMETER ESTIMATION FOR CORE 4A  
(Using Truncated Plot Points)

| FORWARD RUNS |        |                   |        |                      |                                   |                                |
|--------------|--------|-------------------|--------|----------------------|-----------------------------------|--------------------------------|
| Run No       | CP KPa | K $\mu\text{m}^2$ | b KPa  | Kabs $\mu\text{m}^2$ | Fb $\text{m}^{-1} \times 10^{12}$ | Gamma msec/kg $\times 10^{16}$ |
| 1            | NILL   | 0.1520            | 181.88 | 0.5950               | 0.8058                            | 0.1127                         |
| 2            | 2759.9 | 0.2014            | 134.03 | 0.4280               | 0.3183                            | 0.1965                         |
| 3            | 3447.4 | 0.1894            | 169.98 | 0.4334               | 0.3674                            | 0.8987                         |
| 4            | 4136.9 | 0.1469            | 106.58 | 0.2602               | 1.0016                            | 1.8775                         |
| 5            | 4826.3 | 0.1156            | 214.78 | 0.3723               | 0.6997                            | 0.3803                         |
| 6            | 5515.8 | 0.1123            | 273.19 | 0.3729               | 0.7998                            | 1.3308                         |
| REVERSE RUNS |        |                   |        |                      |                                   |                                |
| 1            | NILL   | 0.0839            | 359.55 | 0.2400               | 3.0924                            | 2.4012                         |
| 2            | 2759.9 | 0.0994            | 183.96 | 0.2969               | 0.9571                            | 0.8435                         |
| 3            | 3447.4 | 0.1022            | 168.73 | 0.2335               | 0.6719                            | 1.3693                         |
| 4            | 4136.9 | 0.1184            | 104.29 | 0.2309               | 1.0616                            | 0.2516                         |
| 5            | 4826.3 | 0.1490            | 101.96 | 0.4098               | 0.6830                            | 0.3795                         |
| 6            | 5515.8 | 0.1120            | 156.41 | 0.1849               | 1.0176                            | 3.1086                         |
| 7            | 5515.8 | 0.0002            | 232.01 | 0.0004               | 15080.                            | 76140.                         |
| 1SEN         | 4826.3 | 0.0080            | 33.505 | 0.0101               | 35.673                            | 4.3530                         |

and pressures.

Case (c) has generally more stable matrices than case (b), and parameters estimated with it are closer to those obtained by linear fit. It has slightly larger values of the error sum of squares than Case (b). However, Case (b) shows greater signs of ill-conditioning than Case (c), and therefore produces less reliable results. Non-linear effect in the combined variables, inspite of the linearization of the coefficients, could probably be the cause. The quadratic case (Case (c)) is the same as Equation 2.32 due to Piplapure.

The numerical method, which depends mainly on the model equation, and the number of data points used, appear to be superior to the combined graphical and linear fit methods because :

- (a) parameters can be estimated simultaneously,
- (b) interactions between parameters can be monitored properly.

However, the graphical/linear fit method gives fundamental parameter estimations on which the validity of the numerical methods can be checked.

Table 8-9 gives a summary of the parameter estimation for Core 4A, Runs 8 and 9, and for Core 2, Runs 1 to 4. It is observed that values of the inertial resistance coefficient,  $F_b$ , from both methods of parameter estimation for Runs 8 and 9, are both large and negative. Possible causes for this could be attributed to similar causes of the

TABLE 8-9: PARAMETER ESTIMATION  
For Core 4a Runs 8 and 9, Core 2 Runs 1 to 4  
BY GRAPHICAL/LINEAR FIT METHOD

| Flow Method | Run No | CP KPa | k $\mu\text{m}^2$ | b kPa  | k <sub>cor</sub> $\mu\text{m}^2$ | k <sub>abs</sub> $\mu\text{m}^2$ | F <sub>b</sub> m <sup>-1</sup><br>x 10 <sup>12</sup> |
|-------------|--------|--------|-------------------|--------|----------------------------------|----------------------------------|--|
| 2           | 8#     | 5515.8 | 0.00023           | 242.28 | 0.00019                          | 0.00032                          | -7614.7  |
| 2           | 9#     | 4826.3 | 0.00200           | 218.25 | 0.00140                          | 0.00206                          | -111.80  |
| 1           | 1*     | 4136.9 | 0.01680           | 269.86 | 0.01814                          | 0.03536                          | 10.237   |
| 1           | 2*     | 4136.9 | 0.01470           | 100.58 | 0.01480                          | 0.01910                          | 4.2042   |
| 2           | 3*     | 4136.9 | 0.00347           | 567.41 | 0.00390                          | 0.01600                          | 46.969   |
| 2           | 4*     | 4136.9 | 0.00276           | 819.92 | 0.00280                          | 0.01320                          | 37.914   |

BY NUMERICAL METHOD

| Flow Method | Run No | CP KPa | k $\mu\text{m}^2$ | b kPa  | k <sub>abs</sub> $\mu\text{m}^2$ | F <sub>b</sub> m <sup>-1</sup><br>x 10 <sup>12</sup> |
|-------------|--------|--------|-------------------|--------|----------------------------------|--|
| 2           | 8#     | 5515.8 | 0.00016           | 357.66 | 0.00038                          | -12470.  |
| 2           | 9#     | 4826.3 | 0.00078           | 801.79 | 0.00265                          | -299.97  |
| 1           | 1*     | 4136.9 | 0.01983           | 216.25 | 0.15130                          | 4.4176   |
| 1           | 2*     | 4136.9 | 0.01500           | 94.666 | 0.02716                          | 0.7191   |
| 2           | 3*     | 4136.9 | 0.01220           | 55.407 | 0.01669                          | 44.367   |
| 2           | 4*     | 4136.9 | 0.00304           | 733.16 | 0.01906                          | 53.958   |

where : # = core 4a - under uniaxial confinement

\* = core 2 - under triaxial confinement

abnormal Klinkenberg and visco-inertial plot profiles obtained for these runs. Generally, there are noticeable differences between parameters obtained by the numerical method, and those obtained by the graphical/linear fit method. Specifically for core 2, parameter values obtained in the forward direction differ from those obtained in the reverse direction.

### 8.7 Analysis of the Estimated Parameters

Parameter values,  $k_a$ ,  $b$ ,  $F_b$ , and  $\gamma$ , have been estimated numerically from the truncated plot points, under uniaxial confinement pressure. The values for  $k$  and  $b$ , which are in the viscous region, are close to those obtained by both graphical and linear fit methods. The estimated  $F_b$  and  $\gamma$  values obtained with complete plot points are higher than those obtained with truncated plot points. This suggests the possibility of  $\gamma$  becoming significant in the lower region of flow as soon as the flow rate is large enough to induce fluctuations. Hence, the magnitude of the inertial term increases with increasing flow rates and pressures, and both  $\gamma$  and  $F_b$  values begin to increase.

#### 8.7.1 Gas Permeability in the Higher Flow Regime

It is observed from Tables 8-1 to 8-8, that the values of the apparent gas permeability,  $k_a$ , obtained with the truncated plot points, are greater than those obtained with the complete plot points. In many cases,  $k_a$  values obtained

by using the complete plot points, are smaller than the absolute gas permeability,  $K$ , obtained from both graphical and linear fit methods with truncated plot points. It can be inferred from these observations that the apparent gas permeability,  $k_a$ , is higher at lower flow rates and pressures, (lower flow regime), and lower at higher flow rates and pressures (higher flow regime). In other words, since the flow regime depends on the magnitude of the flow rate, the apparent gas permeability,  $k_a$ , has an inverse relationship with  $Q_o$ .

According to Equation 2.7,  $k_a$  is inversely proportional to the mean pressure,  $P_m$ , and it decreases for increasing pressure towards its limiting value,  $k$ , at infinite mean pressure. This also corresponds to an inverse relationship of  $k_a$  with  $Q_o$ . Gas slippage decreases as  $k_a$  decreases, and with increasing pressures. From theory, the gas properties at higher pressures, approximate those of a liquid, and, therefore, the effect of gas slippage becomes minimal under this flow condition.

#### 8.7.2 Observations on Inertial Resistance Coefficient and Gamma

From Tables 8-1 to 8-8, the values of both gamma and the inertial resistance coefficient,  $F_b$ , obtained with the complete plot points under uniaxial confinement are higher than those obtained with the truncated plot points. This suggests that these parameters have higher values when

higher flow regimes are traversed, and lower values when lower flow regimes are traversed. This may seem anomalous at first since  $F_b$ , defined according to Equation A-22b, has a dimension of  $L^{-1}$ , and therefore is characteristic of the length scale in the porous media. Gamma, on the other hand, is defined according to Equation A-22c, and has a dimension of  $LTM^{-1}$ , which is the same as the reciprocal of viscosity.

When Darcy's law is obeyed at low flow rates, the inertial term is negligible. With increasing flow rate, deviation from Darcy's law is obtained, and a quadratic term is added to account for the increasing inertial term. As the flow rate further increases, with corresponding increases in the fluid pressure, the need for the cubic term becomes significant. It can then be inferred that generating both inertial and cubic terms beyond the flow regime in which Darcy's law is satisfied, is dependent on the flow regimes traversed as flow rates increase.

Similar observations have been recorded by several investigators. Irmay(75) remarked that the values of  $a$  and  $b$  in his Equation 2.22 differed as  $Re$  increased for higher flow rates. Using different gases with different viscosity values, Senturk(76) concluded that  $F_b$  could be a function of viscosity. Crafton's observations(77) are condensed in Equation 2.31:

### 8.7.3 Friction Factor - Reynolds Number Evaluation

From Equation A-12a, the Reynolds number,  $Re$ , is given as the ratio of the inertial to the viscous forces, and is given dimensionally as

$$Re = \frac{Lpq}{\mu} \quad (8.1)$$

Friction factor,  $f$ , is given as the ratio of the dissipative forces to the inertial forces(78). If Equations A-23 and A-23a are each divided by the inertial term, the following expressions are obtained:

$$\frac{-\frac{dp}{dx}}{F_b \rho q^2} = \frac{(\mu/k)q}{F_b \rho q^2} + 1 + \frac{\gamma \rho^2 q^3}{F_b \rho q^2} \quad (8.2)$$

$$\frac{-\frac{dp}{dx}}{F_b \rho q^2} = \frac{(\mu/k)q}{F_b \rho q^2} + 1 \quad (8.3)$$

Multiplying Equations 8.2 and 8.3 by 64 and simplifying, the following equations are obtained:

$$\frac{64(-\frac{dp}{dx})}{F_b \rho q^2} = \frac{64}{F_b \rho q k} + 64 + \frac{64 \gamma \rho q}{F_b} \quad (8.4)$$

$$\frac{64(-\frac{dp}{dx})}{F_b \rho q^2} = \frac{64}{F_b \rho q k} + 64 \quad (8.5)$$

For simplification, let

$$f = \frac{64(-\frac{dp}{dx})}{F_b \rho q^2} = \frac{\text{Dissipative Forces}}{\text{Inertial Forces}} \quad (8.6a)$$

$$R_e = \frac{F_b \rho q k}{\mu} = \frac{\text{Inertial Forces}}{\text{Viscous Forces}} \quad (8.6b)$$

$$T_f = 64 \left( \frac{\rho q}{F_b} \right) \quad (8.6c)$$

Also, let  $T_f$  be regarded as the turbulence factor which measures the increasing significance of the cubic term beyond the inertial term. Therefore substituting Equations 8-6a, 8.6b, and 8.6c into Equations 8.4 and 8.5, the following expressions are obtained respectively as:

$$f = \frac{64}{R_e} + 64 + T_f \quad (8.7)$$

$$f = \frac{64}{R_e} + 64 \quad (8.8)$$

Equation 8.8 is similar to the friction factor expression used by Mackett(79) for Forchheimer's quadratic equation. It differs from Equation 8.7 by the term  $T_f$  which is zero or negligible for the quadratic case, and a positive quantity

for the Forchheimer's cubic case.

A log-log plot of friction factor,  $f$ , against the Reynolds number,  $Re$ , should give, from Equations 8.7 and 8.8:

a) a linear viscous plot at low  $Re$ , represented by

$$f = 64/Re,$$

b) a deviation from linear viscous region due to the increasing inertial effects which corresponds to increases in  $Re$ . This is represented by  $f = 64$ .

c) an indication of any possible flow effects due higher  $Re$  beyond the inertial region. This is represented by  $f = Tf$ .

Similarly, Equation 4.12 is divided by its inertial term, (the quadratic term), and then multiplied through by 64 to get the following expression:

$$\frac{64(P_e^2 - P_w^2)}{C_2 \bar{z} T Q_0 F_b} = \frac{64}{F_b Q_0 k_a \cdot \frac{C_2}{C_1}} + 64 + \frac{64 C_3}{C_2} \cdot \frac{\gamma Q_0}{F_b} \quad (8.9)$$

The product of the absolute gas permeability,  $k$ , and the inertial resistance coefficient,  $F_b$ , gives the length scale for the Reynolds number. Using the apparent gas permeability,  $k_a$ , instead of the absolute gas permeability,  $k$ , gives a larger value of the length scale due to the gas slippage effect. For the purposes of the Reynolds number evaluation, the absolute gas permeability,  $k$ , is used.

Therefore, Equation 8.9 is reduced to Equation 8.7, where

$$f = \frac{64(P_e^2 - P_w^2)}{C_2 \bar{Z} \bar{T} Q_0^2 F_b} \quad (8.9a)$$

$$R_e = \frac{F_b Q_0 k_a}{\mu} \cdot \frac{C_2}{C_1} \quad (8.9b)$$

$$T_f = \frac{64 C_3}{C_2} \cdot \frac{\gamma Q_0}{F_b} \quad (8.9c)$$

Figure 8-18 shows the log-log plot of friction factor,  $f$ , against Reynolds number,  $R_e$ , for Core Sample 1C, Run 3, in which  $f$  decreases initially with increasing  $R_e$ , and later increases with increasing  $R_e$ . This is expected from Equation 8.7 and therefore the plot of  $f$  against  $R_e$  suggests an existence of some term beyond the visco-inertial flow regime.

### 8.8 Effect of Confining Pressure and Hysteresis

The object of running the experiments in both the forward and the reverse directions is to check for hysteresis. The general plot profiles are the same for both directions, but the estimated parameter values, as well as their trend with confining pressures, are significantly different in many cases.

The first runs of Core samples 4A and 1C in both directions, did not have any predefined CP. Their estimated

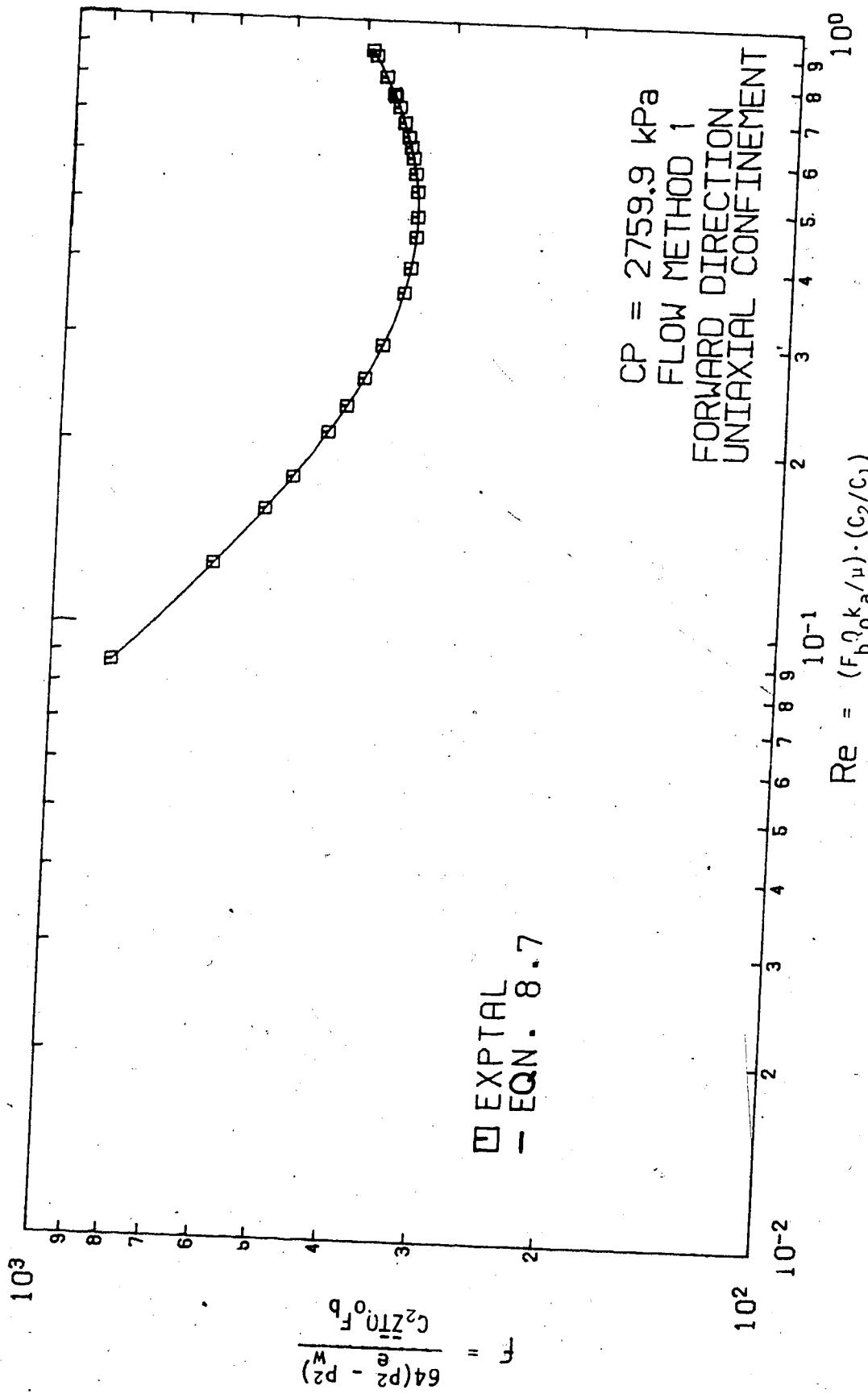


Figure 8-18 CORE, 1C RUN 3 FRICTION FACTOR-RE PLOT

parameters are observed, from Tables 8-1 to 8-8, to be seemingly out of phase with those of the other runs. Though the parameter values are expected to show a continuous trend with increasing predefined CP, the parameter values of the first runs, in many cases, are larger than those of the second runs. From this observation, confinement pressure appears to exhibit a substantial effect on the flow behavior.

The following points are gathered from Figures 8-19 to 8-22, and from Tables 8-1 to 8-8, for Core 1C and Core 4A, using Flow Method 1:

1. Permeability,  $k$ , decreases in forward direction, but increases in reverse direction, with increasing confining pressure, CP. The values in the forward direction are higher than those obtained in the reverse direction.
2. Gas slippage coefficient,  $b$ , increases in the forward direction, but decreases in the reverse direction, with increasing CP.
3. Inertial coefficient,  $F_b$ , increases in both forward and reverse directions with CP.
4. Gamma increases in both forward and reverse directions with CP.

From Table 8-9, parameter values obtained in the forward direction for Core 2, under triaxial confinement, are significantly different from those obtained in the reverse direction.

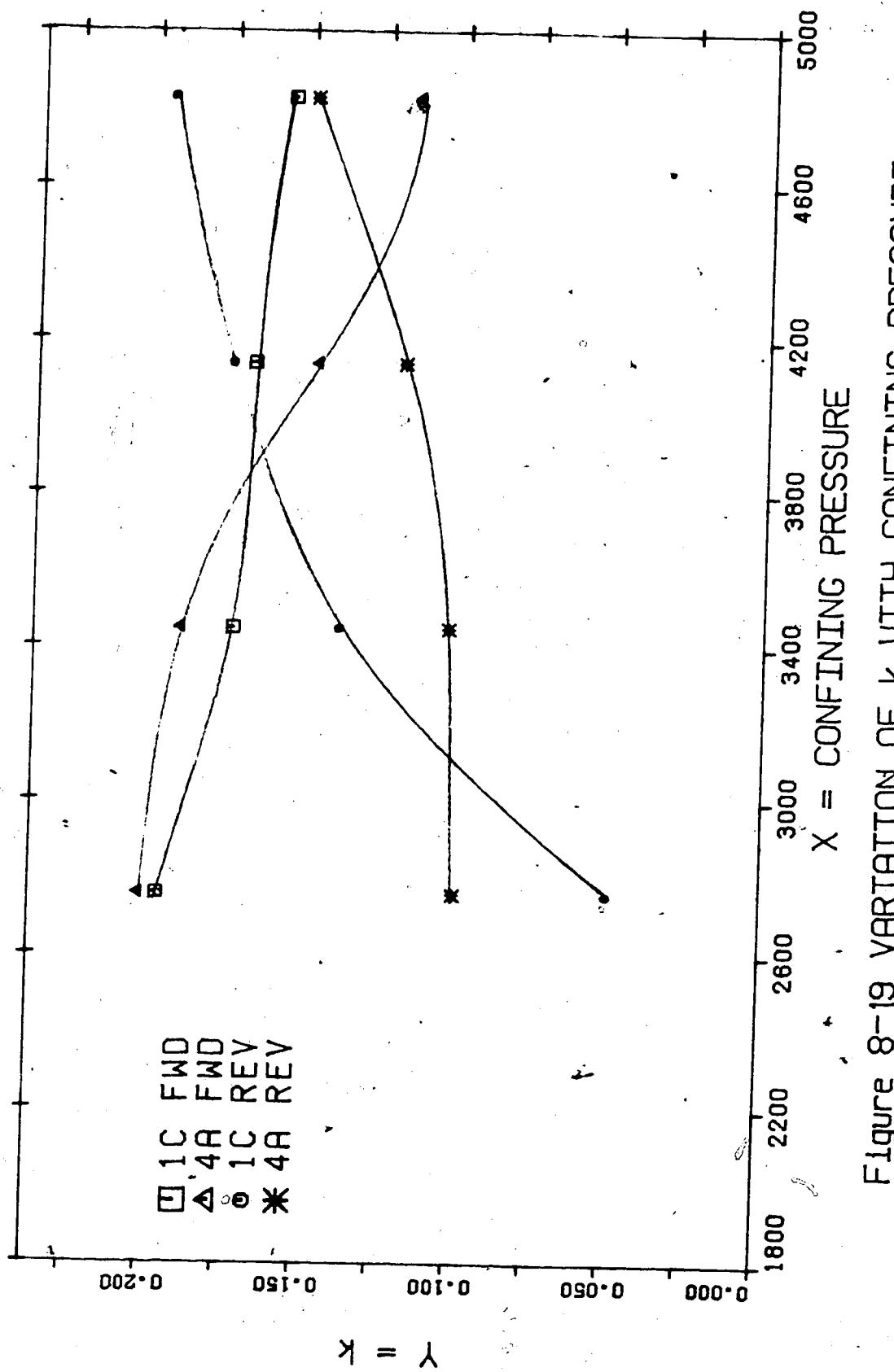


Figure 8-19 VARIATION OF  $k$  WITH CONFINING PRESSURE

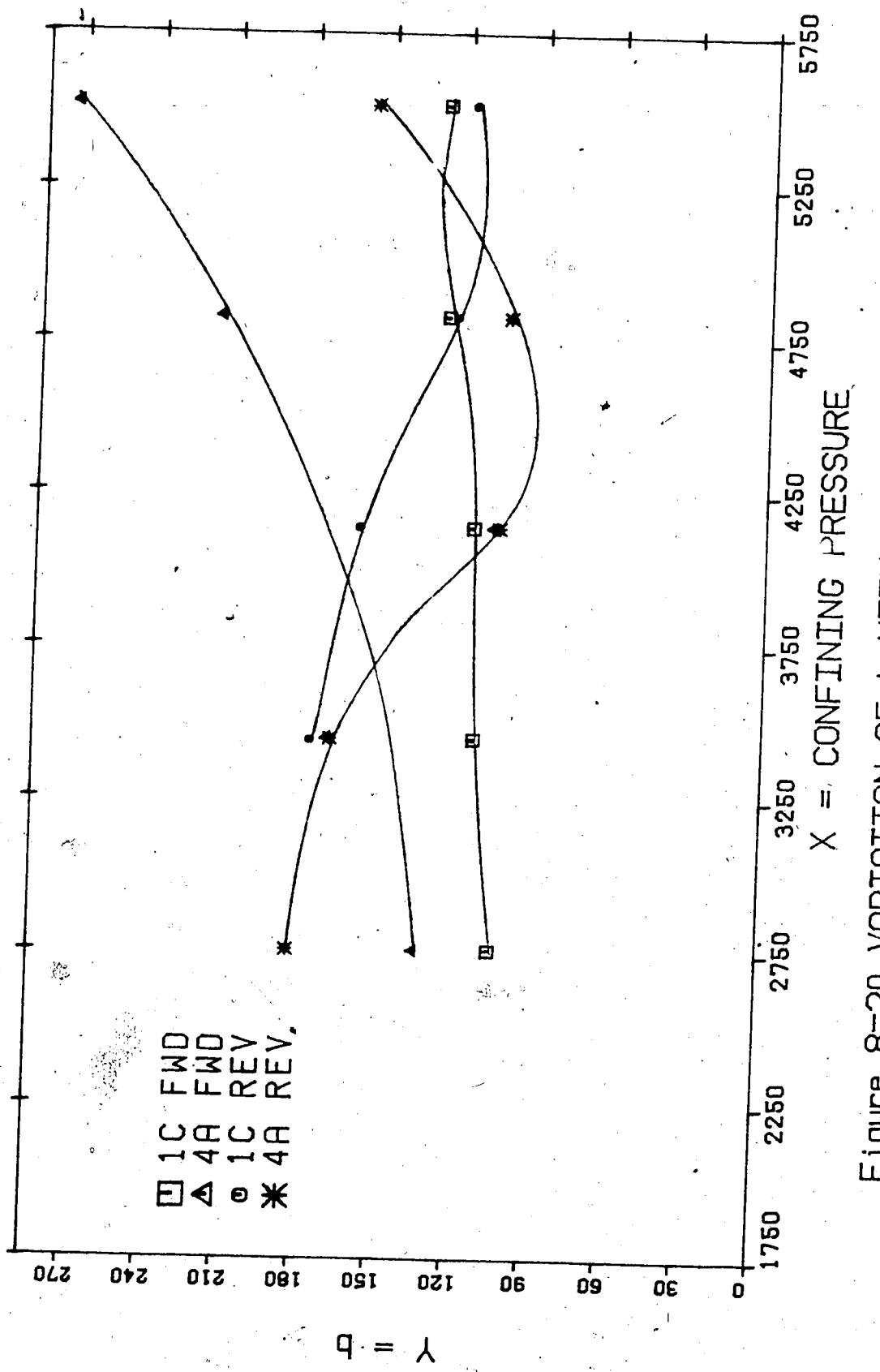


Figure 8-20 VARIATION OF  $b$  WITH CONFINING PRESSURE

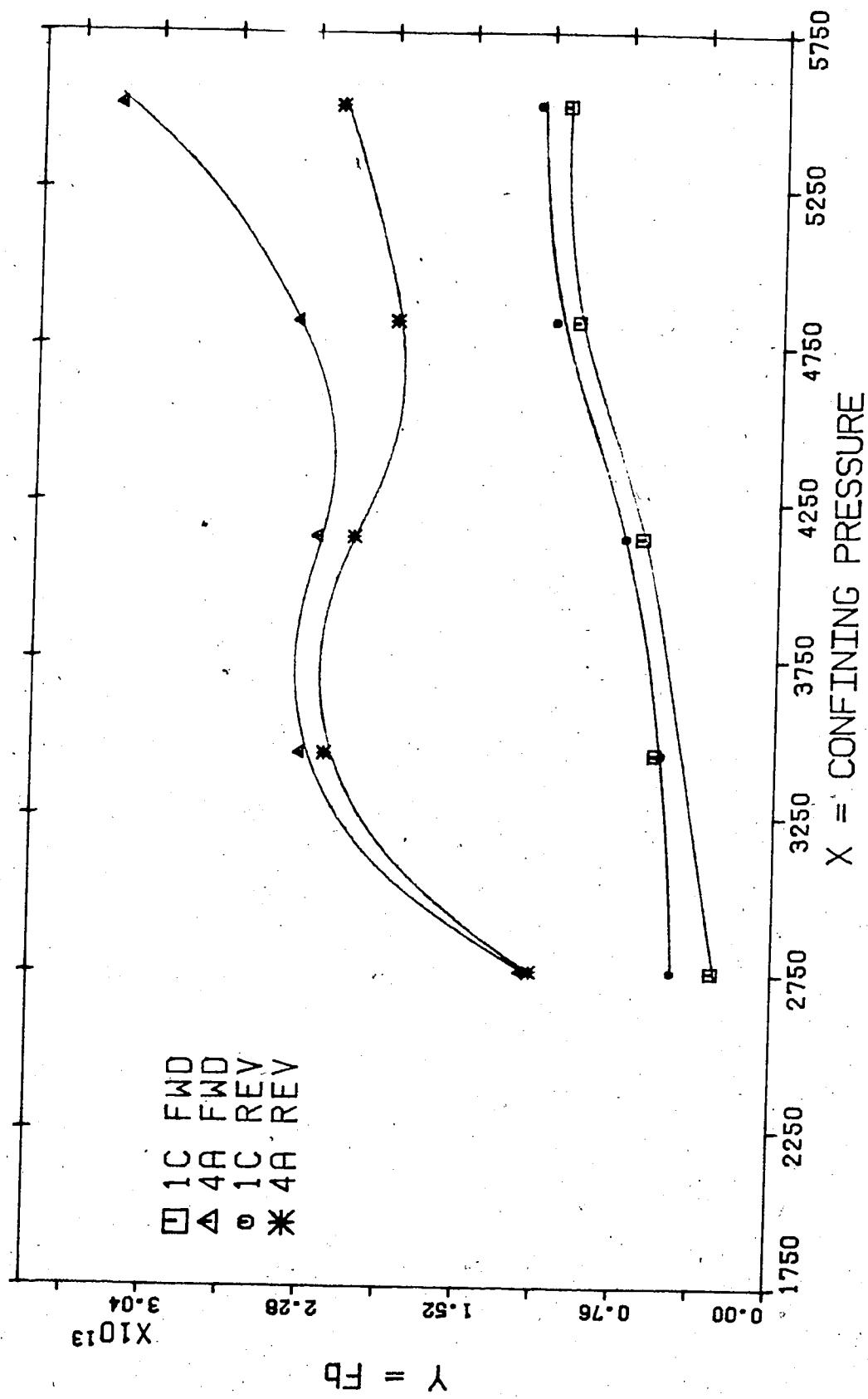


Figure 8-21 VARIATION OF  $F_b$  WITH CONFINING PRESSURE

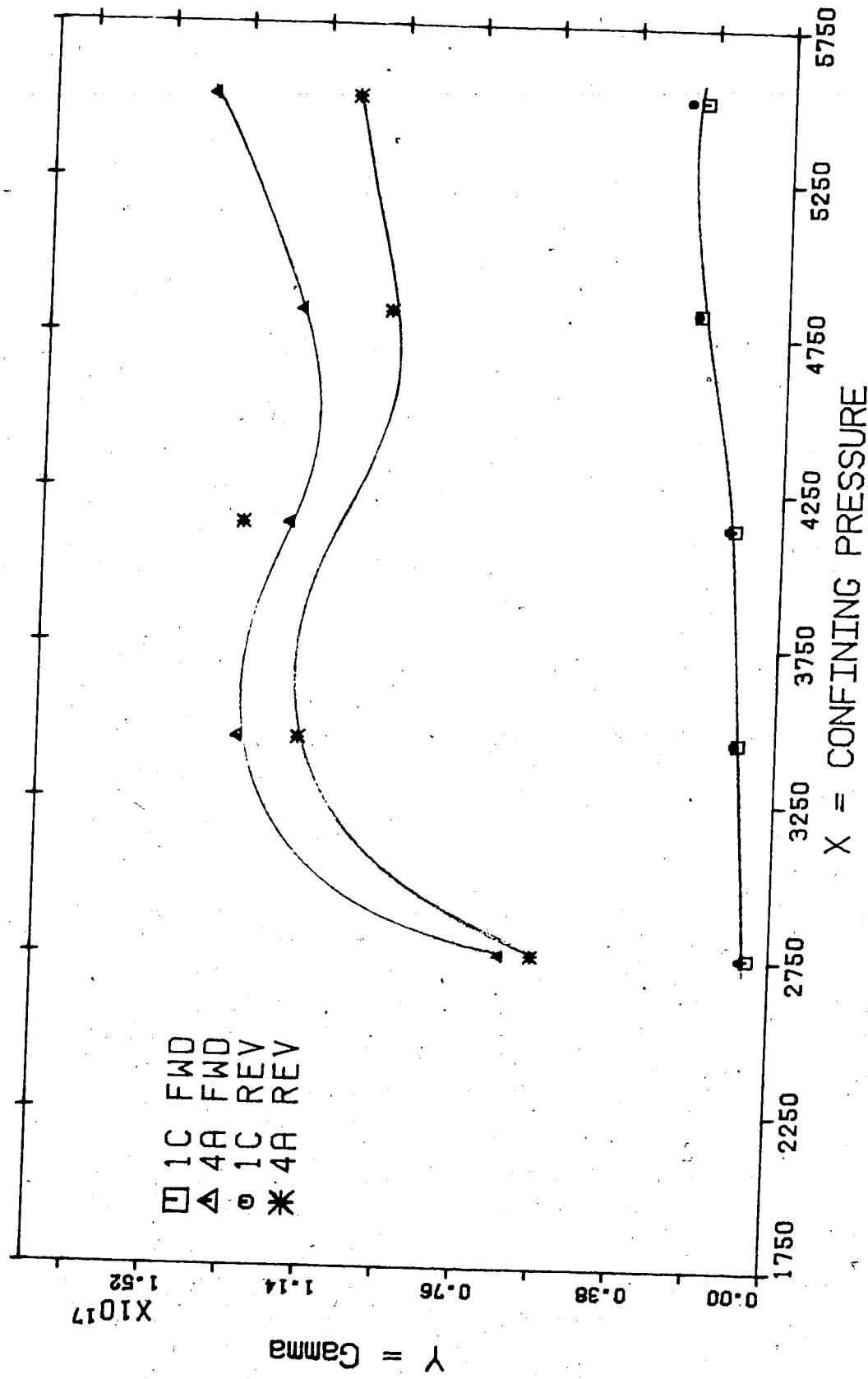


Figure 8-22 VARIATION OF GAMMA WITH CONFINING PRESSURE

The foregoing observations reflect a significant effect of hysteresis on the flow behavior especially on both permeability and gas slippage coefficient. Since confinement pressure affects compaction of the core sample, reorientation of the grains in the porous medium may take place, and could cause changes in permeability and porosity. Hence, conducting the flowing experiment in both forward and reverse directions could cause this change and, by deduction for gas flow, the gas slippage coefficient could also be affected.

#### 8.9 Gamma Relationship with Other Parameters

From theory in Appendix A, and the parametric analysis in Section 8.7, gamma appears to have some relationship with the other parameters, and with viscosity. Hence, by dimensional analysis, and from Equations A-22c and A-22d, gamma can be related by:

$$\gamma = f_n (\mu_{\phi}^c e_k^d F_b^f g) \quad (8.10)$$

which reduces to:

$$\gamma = f_n (\mu_{\phi}^c e_k^d F_b^{2d}) \quad (8.11)$$

It is noticed that gas slippage coefficient, b, does not appear in Equation 8.11. This is not surprising because gamma manifests itself at higher flow rates and pressures,

under which condition gas properties approximate to that of liquid.

Equation 8.11 can be represented in different forms, and used to fit the data from Tables 8-5 and 8-6 for a complete plot profile. The following forms are possible:

$$\gamma = a \mu^b (k F_b^2)^c \quad (8.12)$$

$$\gamma = a \mu^b k c_{F_b}^d \quad (8.13)$$

$$\gamma = a \mu^b k c_{F_b}^d \phi^e \quad (8.14)$$

$$\gamma = a \mu^b (k F_b^2)^c \phi^e \quad (8.15)$$

Table (8-10) gives a summary of the values of the coefficients, their accompanying error sum of squares and their standard deviations. The best fit is chosen therefrom.

Viscosity has been calculated at standard condition of pressure and temperature in order to obtain a reference viscosity value. Only nitrogen has been used.

The following observations are gathered from Table (8-10):

1. Fit 3 gives the best relationship of gamma with other parameters.

TABLE 8-10: VARIATION OF GAMMA WITH OTHER PARAMETERS

| Fit | a                       | b       | c       | d      | e      | Xsqr   | Stddev |
|-----|-------------------------|---------|---------|--------|--------|--------|--------|
| 1   | $1.5089 \times 10^{-5}$ | -0.8961 | 0.8961  |        |        | 0.3052 | 0.5403 |
| 2   | $1.8057 \times 10^{-3}$ | -1.6705 | -0.4622 | 1.6705 |        | 0.2625 | 0.5006 |
| 3   | $4.7353 \times 10^8$    | -0.6980 | -0.0527 | 0.6980 | 1.8014 | 0.0628 | 0.2446 |
| 4   | $6.5743 \times 10^7$    | -0.4257 | 0.4257  |        | 1.7594 | 0.1260 | 0.3468 |

2. The coefficient of the viscosity term, which is negative, bears a consistent relationship with that of the Fb term. This could depict a possible relationship between Fb and viscosity.
3. Porosity term exists and its coefficient is appreciable.
4. Gamma increases as Fb increases.
5. Both Fb and a constant quantity play a dominant role in the magnitude of gamma. This constant could be similar, or identical, to the dimensionless shape factor, C6, which was obtained in the derivation of gamma in Appendix A.
6. Both apparent gas permeability and viscosity have an inverse dependence on gamma.

Hence gamma is given from Fit 3 as

$$\gamma = 4.7353 \times 10^9 \left( \frac{F_b}{\mu} \right)^{0.6980} \frac{\phi^{1.8014}}{k_a^{0.0527}} \quad (8.16)$$

At the flow condition under which gamma manifests itself,  $k_a$  approximates to  $K$ , for decreasing gas slip. Hence, by replacing  $k_a$  with  $K$ , Equation 8.16 becomes

$$\gamma = 4.7353 \times 10^9 \left( \frac{F_b}{\mu} \right)^{0.6980} \frac{\phi^{1.8014}}{K^{0.0527}} \quad (8.17)$$

The variation of gamma, and the fitted gamma, with apparent gas permeability,  $k_a$ , is shown in Figure 8-23 on a log-log plot. The profile shows an inverse relationship of gamma with  $k_a$ .

This analysis, together with the relationship obtained for gamma is limited in the sense that data for them came from very limited number of cores and for nitrogen only. The data used are comprised of both forward and reverse runs using Flow Method 1, and for different confining pressures. While this may not be conclusive, this analysis

1. shows an inverse relationship of gamma with permeability.
2. shows less dependence of gamma on gas slippage coefficient.
3. supplies a possible estimate of gamma in terms of the other parameters and viscosity.
4. shows some correspondence in the properties of gamma

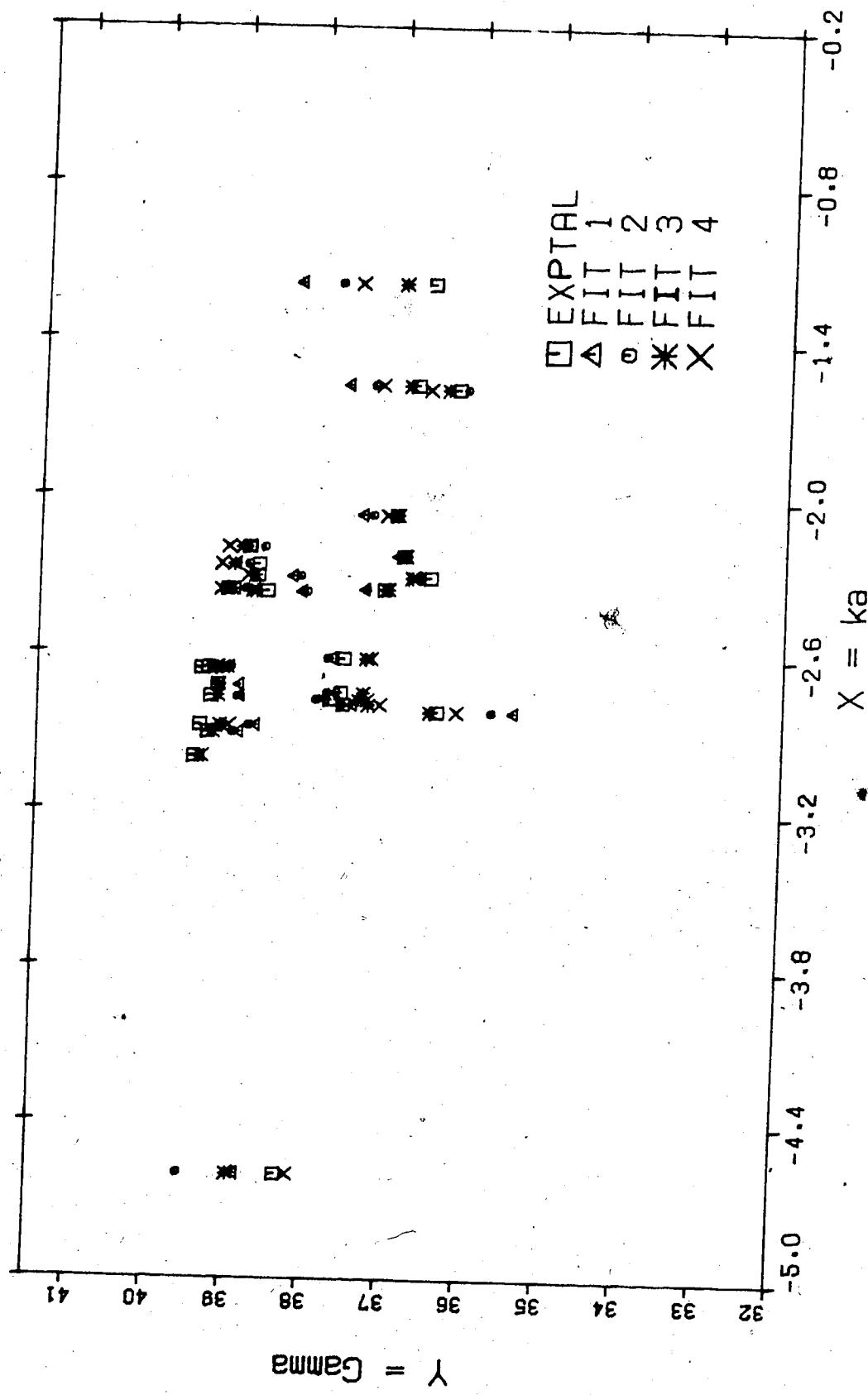


Figure 8-23 Variation of Gamma and Fitted Gamma with ka

between theory and experimental results.

Further investigation is needed to explain these findings.

#### 8.10 Nomenclature for Gamma

The foregoing treatment has shown the possible existence of gamma. This parameter has been identified as being within the cubic term, which is beyond the visco-inertial region. The cubic term, as treated in this work, is synonymous with the turbulent term. Hence gamma should have the properties associated with the turbulent flow. Since gamma should be known with a name that reflects its characteristics, it is suggested it be known as the 'turbulence resistance coefficient'.

## 9. CONCLUSION and RECOMMENDATIONS

### 9.1 CONCLUSIONS

There has been, generally, a good correspondence between theory and experimental results, especially in the properties of gamma - the flow parameter in the cubic term of the general cubic flow equation. Anomalous flow behavior was obtained when Flow Method 2 was used at lower pressures and on uniaxial radial confinement. Similar behavior was observed at considerably higher pressures when Flow Method 1 was used on triaxial radial confinement. It is believed that this behavior is dependent on the type of confinement and on the flow method used.

The following are the conclusions:

#### 9.1.1 Flow Phenomena

1. A general cubic flow equation has been developed from consideration of kinetic energy equation of mean flow and dimensional relations. The physical basis of the cubic term in the model flow equation has been established by using the boundary layer theory to explain the flow phenomenon at higher flow rates and pressures through a porous path.
2. Gamma, the main parameter in this cubic term, is
  - a. related to a characteristic dimensionless shape factor which is significant at higher flow rates.
  - b. is inversely related to viscosity.

3. The flow model equation
  - a. spans a wide range of flow rates,
  - b. is regarded as a modification to Forchheimer's quadratic equation,
  - c. has been modified, and corrected, for gas slippage effect.

#### 9.1.2 Experimental Evidence

1. In comparison with the quadratic equation, and under uniaxial confinement, the cubic equation gave a better fit on the experimental data.
2. Both  $F_b$  and gamma have been observed to depend on the flow rate and the flow regime.
3. Gamma has less dependence on the slippage coefficient in the higher flow regime.
4. Gamma has an inverse relationship with permeability and viscosity.
5. The log-log plot of the friction factor against the Reynolds number suggests an existence of some term beyond the visco-inertial flow regime.
6. Relationship between gamma and other parameters has been obtained. Both  $F_b$  and a constant dimensionless quantity have a considerable effect on the magnitude of gamma.

#### 9.1.3 The Flow Method

1. The flow method clearly has a very significant effect in the type of results obtained. The flow regime reached is

- linked to the flow method used.
- 2. Flow Method 1 appears to be better suited for higher flow regimes, and Flow Method 2 seems more appropriate for lower regimes, especially for viscous and visco-inertial flow regimes.
  - 3. Some relationship has been observed to exist between the horizontal stresses prevailing on the core material, and the flow method used. More pressure drawdown could be generated by using Flow Method 2 than by using Flow Method 1.

#### 9.1.4 Confining Pressure and Hysteresis

- 1. The stress-strain analysis for uniaxial confining pressure has been used to describe the state of stress and strain on the core during the flowing process with success.
- 2. There is a noticeable difference between the parameter values obtained by imposing a predefined confining pressure, CP, and those obtained without any predefined CP.
- 3. Both the apparent and the absolute permeability,  $k_a$  and  $k$ , are found to decrease with CP in the forward direction.
- 4. There is a noticeable increase of gamma and  $F_b$  with CP.
- 5. Gas slippage coefficient increases in the forward direction, but decreases in the reverse direction; with increasing CP. This may be due to compaction effect

which involves grain reorientation within the core material.

6. Hysteresis has been observed on the flowing experiments.

#### 9.1.5 Plot Profiles and Parameter Estimation

1. The plot profiles seem to be comprised of points in the lower flow rates and pressures, and those in the higher flow rates and pressures.
2. A quadratic fit appears suitable for lower flow rates and pressures, while a cubic fit appears suitable for all the plot points, especially those which are obtained at higher flow rates and pressures.
3. The plot profiles obtained by using Flow Method 2 showed some anomalous behavior beyond the viscous region. This is believed to be due to the combined effect of confinement and flow method.
4. The estimated parameter values from all the estimation methods are generally close for the truncated plot points. The numerical method seems superior to the graphical and linear fit methods.
5. The choice of gas slippage coefficient,  $b$ , affects not only the corrected gas permeability value, on the graphical method of parameter estimation, but also the value of the corrected  $F_b$  so obtained. The use of the linear correlation coefficient has increased the accuracy of the estimated value of  $b$ .
6. Under triaxial confinement, anomalous plot profiles were

noticed at considerably higher pressures with Flow Method 1. Normal behavior, for lower flow rates and pressures, was obtained with Flow Method 2. It is believed that the loose rubber overburden sleeve could not effect complete confinement at higher pressures.

### 9.2 RECOMMENDATIONS

The recommendations and suggestions for further research are presented below:

1. More investigation is needed on the flow phenomena involving the higher flow rates and pressures so as to
  - a. check the concept of a third order term in the flow equation,
  - b. discretize the flow regimes encountered,
  - c. examine the properties of gamma, and its relationship with other parameters.
  - d. check the friction factor-Reynolds number criterion.
2. There is need for more investigation on the effect of hysteresis on flow behavior.
3. The flow methods have to be applied to more radial flow experiments to ascertain their particular roles in
  - a. the flow regimes,
  - b. confining pressure effects,
  - c. possible hysteresis effect.Excessive drawdown pressures should not be used while using Flow Method 2.
4. The stress-strain analysis should be made prior to using

- any type of radial cell.
5. Numerical methods involving iterative procedures should be explored in addition to those with direct procedures.
  6. The loose rubber overburden sleeve, on the radial triaxial overburden cell, should be replaced by one which fits more closely.
  7. Further investigation should be made with triaxial overburden radial cell in order to study, and compare, the observations obtained with uniaxial confinement radial cell.
  8. The radial cell should be designed so that runs can be made using both uniaxial and triaxial confinement on the same core. This is expected to aid comparison between the two methods of confinement.
  9. The use of the average pressures in radial systems should be explored in preference to the use of the mean pressure.

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APPENDIX A  
DERIVATION OF SECOND AND THIRD ORDER EQUATIONS OF  
FORCHHEIMER

## A. APPENDIX A - DERIVATION OF SECOND AND THIRD ORDER EQUATIONS OF FORCHHEIMER

The general Navier-Stokes equations of motion of viscous fluid for incompressible fluid is given in cartesian tensor notation(79) as

$$\rho \frac{\partial u_j}{\partial t} + \partial u_i \frac{\partial u_j}{\partial x_i} = - \frac{\partial P}{\partial x_j} + \mu \frac{\partial^2 u_j}{\partial x_i \partial x_i} + \rho X_j \quad (A-1)$$

where Einstein summation convention is used for  $i, j = 1, 3$ , in horizontal and vertical directions respectively. The term,  $X$ , represents body or gravitational forces. Both density and viscosity are assumed to be constant.

For steady-state, horizontal flow Equations A-1 become

$$\rho \dot{u}_i \frac{\partial u_j}{\partial x_i} = - \frac{\partial P}{\partial x_j} + \mu \frac{\partial^2 u_j}{\partial x_i \partial x_i} \quad (A-2)$$

These equations comprise of inertial term on the left, and differential pressure gradient and viscous terms on the right. For flow along a capillary tube, and assuming a no-slip condition at the fluid-solid interface, Newton's law of viscosity is given by

$$\tau = - \mu \frac{\partial u_j}{\partial x_i} \quad (A-3)$$

where the viscous momentum is in the direction of the negative velocity gradient(80).

Equations A-2 can be rearranged in the form of

$$-\frac{\partial P}{\partial x_j} = \rho U_i \frac{\partial u_j}{\partial x_i} + \mu \frac{\partial^2 u_j}{\partial x_i \partial x_j} \quad (A-4)$$

Since the continuity equation for steady state, incompressible flow is given as

$$\frac{\partial u_i}{\partial x_i} = 0 \quad (A-4a)$$

the following expansion can be made

$$\frac{\partial u_i u_j}{\partial x_j} = u_i \frac{\partial u_j}{\partial x_i} + u_j \frac{\partial u_i}{\partial x_i} \quad (A-4b)$$

Subsequently Equations A-4 become

$$-\frac{\partial P}{\partial x_j} = \rho \frac{\partial u_i u_j}{\partial x_i} + u_j \frac{\partial^2 u_j}{\partial x_i \partial x_i} \quad (A-5)$$

#### A.1 Reynolds Equations

As velocity increases, local irregularities in flow streamlines begin to form and manifest themselves as fluctuations. Within a critical velocity range, which is characteristic of the medium through which the fluid is flowing, the produced fluctuations may be negligible. With higher velocities, these may no longer be negligible and Navier-Stokes equations are rewritten to reflect this new condition.

Fluctuation may be defined(81,82) as the difference between the measured field values of the measurable quantity and their time-averaged values. Its presence manifests itself in an apparent increase in the viscosity of the fundamental flow. It can be related as:

$$\text{Observed Value} = \text{Mean Value} + \text{Fluctuation}$$

Specifically, for velocity components and pressure, the relations are:

$$u_i = \bar{u}_i + u'_i \quad P = \bar{P} + p' \quad (A-6)$$

Time-averaged, or mean, values are taken at a fixed point in space over an interval of time such that steady state condition is achieved. By definition, this is represented as

$$\bar{u}_i = \lim_{T \rightarrow \infty} \frac{1}{T} \int_{t_0}^{t_0+T} u_i dt \quad (A-6a)$$

and by deduction, the mean of the fluctuations is zero:

$$\bar{u}'_i = \lim_{T \rightarrow \infty} \frac{1}{T} \int_{t_0}^{t_0+T} (u_i - \bar{u}_i) dt = 0 \quad (A-6b)$$

From the law of averages, the product of both the mean value and its generated fluctuations is zero. Consequently, the process of averaging products is related by

$$\overline{u_i u_j} = \overline{(\bar{u}_i + u'_i)(\bar{u}_j + u'_j)} = \bar{u}_i \bar{u}_j + \overline{u'_i u'_j} \quad (A-6c)$$

If  $\overline{u'_i u'_j} = 0$ , velocity fluctuations are not correlated and destructive combination takes place. This eventually brings fluctuations down to a minimum. If this term is non-zero, they are said to be correlated and will combine constructively to form an additional stress which acts against the smooth laminar flow of the mainstream. It is negative in sign.

Applying the averaging operation on Navier-Stokes Equations A-1, yields the Reynolds equations:

$$\rho \frac{\partial \bar{u}_j}{\partial x_i} + \rho \bar{u}_i \frac{\partial \bar{u}_j}{\partial x_i} = \frac{\partial \bar{p}}{\partial x_i} + \mu \frac{\partial^2 \bar{u}_j}{\partial x_i \partial x_i} + \rho x_j + \frac{\partial}{\partial x_i} \left( -\rho \overline{u'_i u'_j} \right) \quad (A-7)$$

where  $\tau_{ij} = -\overline{u'_i u'_j}$

= Reynolds stress tensor due to increased fluctuations. It contains nine components in 3-dimensional cartesian coordinates as shown

$$\begin{vmatrix} \sigma_{xx} & \tau_{xy} & \tau_{xz} \\ \tau_{xy} & \sigma_{yy} & \tau_{yz} \\ \tau_{xz} & \tau_{yz} & \sigma_{zz} \end{vmatrix} = \begin{vmatrix} \overline{\rho u'^2}_{xx} & \overline{\rho u' u'}_{xy} & \overline{\rho u' u'}_{xz} \\ \overline{\rho u' u'}_{xy} & \overline{\rho u'^2}_{yy} & \overline{\rho u' u'}_{yz} \\ \overline{\rho u' u'}_{xz} & \overline{\rho u' u'}_{yz} & \overline{\rho u'^2}_{zz} \end{vmatrix} \quad (A-7a)$$

The diagonal components are the normal stresses while the off-diagonal components are the shear stresses.

The addition of the Reynolds stresses on Reynolds equations differentiates them from the Navier-Stokes equations. For steady & the horizontal flow they become

$$-\frac{\partial \bar{P}}{\partial x_j} = \mu \frac{\partial^2 \bar{u}_j}{\partial x_i \partial x_i} + \rho \bar{u}_i \frac{\partial \bar{u}_j}{\partial x_i} + \frac{\partial}{\partial x_i} \left( \rho \bar{u}_i' \bar{u}_j' \right) \quad (A-8)$$

### A.2 Kinetic Energy of Mean Flow

Since the mean momentum of the fluctuations, by Equations A-6b is zero, the effect on mean flow of increasing velocity may best be treated as its effects on kinetic and turbulent energy equations (83).

Cross-multiplying the Reynolds equations with the mean velocity yields

$$\begin{aligned} \frac{\partial}{\partial t} \left( \frac{\rho \bar{u}_j \bar{u}_j}{2} \right) + \bar{u}_i \frac{\partial}{\partial x_i} \left( \frac{\rho \bar{u}_j \bar{u}_j}{2} \right) &= \rho \bar{u}_j \bar{u}_j - \frac{\partial}{\partial x_j} (\bar{P} \bar{u}_j) \\ + \mu \bar{u}_j \frac{\partial^2 \bar{u}_j}{\partial x_i \partial x_i} + \bar{u}_j \frac{\partial}{\partial x_i} \left( -\rho \bar{u}_j' \bar{u}_j' \right) \end{aligned} \quad (A-9)$$

where, from continuity equations (A-4a), the following relationships have been obtained:

$$\frac{\partial}{\partial x_j} \left( \frac{\rho \bar{u}_i \bar{u}_i \bar{u}_j}{2} \right) = \bar{u}_i \frac{\partial}{\partial x_i} \left( \frac{\rho \bar{u}_i \bar{u}_i}{2} \right) + \left( \frac{\rho \bar{u}_i \bar{u}_i}{2} \right) \frac{\partial \bar{u}_i}{\partial x_i} = \bar{u}_i \frac{\partial}{\partial x_i} \left( \frac{\rho \bar{u}_i \bar{u}_i}{2} \right) \quad (A-9a)$$

$$\frac{\partial}{\partial x_j} \left( \bar{P} \bar{u}_j \right) = \bar{u}_j \frac{\partial P}{\partial x_j} + \bar{P} \frac{\partial \bar{u}_j}{\partial x_j} = \bar{u}_j \frac{\partial \bar{P}}{\partial x_j} \quad (A-9b)$$

For steady state, horizontal flow, Equations A-9 become

$$\frac{\partial}{\partial x_i} \left( \frac{\rho \bar{u}_i \bar{u}_j \bar{u}_j}{2} \right) = - \bar{u}_j \frac{\partial \bar{P}}{\partial x_j} + \mu \bar{u}_j \frac{\partial^2 \bar{u}_j}{\partial x_i \partial x_i} + \bar{u}_j \frac{\partial}{\partial x_i} \left( - \rho \bar{u}_i' \bar{u}_j' \right) \quad (A-10)$$

(i)

(i)

(ii)

(iii)

where

(i) = Rate of change of Kinetic energy, + Rate at which work is done by the mean pressure;

(ii) = Rate at which work is done by viscous forces;  
= Work done on mean flow by mean viscous stress;

(iii) = Rate at which work is done on mean flow by Reynolds stresses.

If Equation A-7 is cross-multiplied by Equation A-6, an equation for the total kinetic energy is obtained. The difference between this equation and Equation A-9 gives the turbulent kinetic energy equation:

$$\begin{aligned} \frac{\partial}{\partial t} \left( \frac{\rho \bar{u}_i \bar{u}_j}{2} \right) + \bar{u}_i \frac{\partial}{\partial x_i} \left( \frac{\rho \bar{u}_i \bar{u}_j}{2} \right) + \rho \bar{u}_i' \bar{u}_j' \frac{\partial \bar{u}_j}{\partial x_i} + \frac{\partial}{\partial x_i} \left( \bar{u}_i \frac{\bar{u}_i' \bar{u}_j'}{2} \right) \\ = - \frac{\partial}{\partial x_j} (\bar{P}' \bar{u}_j') + \mu \bar{u}_j' \frac{\partial^2 \bar{u}_j}{\partial x_i \partial x_i} \end{aligned} \quad (A-11)$$

It is assumed that, for the case of the porous medium, flow may not be high enough to necessitate the use of this

equation. Rather an increase in velocity increases the chances of fluctuations becoming significant and resulting in large kinetic energy losses. Since measurable flow properties are usually taken at their mean values, the equation for the kinetic energy of mean flow may logically be very appropriate to use.

### A.3 Dimensional Similarity Criteria

- As defined by Schlichting(85), two flows about geometrically similar bodies are similar if the forces acting on a fluid particle bear a fixed ratio each time and at all geometrically similar points. Specifically for a steady, incompressible, horizontal flow, the condition of similar flow is satisfied only if at all corresponding points, a constant ratio of inertial to viscous forces exists. This criterion has been used by Hubbert(86) in his definition of Reynolds number for porous media.

By Monin and Yaglom(87), flows are geometrically similar if the properties of their boundaries are determined uniquely by the same length scale, and by some typical mean velocity. Let  $L$  and  $U$  be some typical mean length and velocity scales which respectively characterize such geometrically similar flows. Further let the length scale be taken generally as the distance along which the velocity scale  $U$  of the flow undergoes a perceptible change. Therefore the Reynolds number is given as

$$Re = UL\rho/\mu = \text{Inertial term/viscous term}$$

= dimensionless

(A-12a)

Combining dimensionally both the viscosity coefficient  $\mu$  and the density  $\rho$  with  $U$  and  $L$ , therefore, the following relations are obtained:

$$\rho \frac{\partial \bar{u}_i \bar{u}_j}{\partial x_i} \sim \frac{\rho U^2}{L} = \text{Const} \cdot \frac{\rho U^2}{L} \quad (A-12)$$

$$\mu \frac{\partial^2 \bar{u}_j}{\partial x_i \partial x_i} \sim \frac{\mu U}{L^2} = \text{Const} \cdot \frac{\mu U}{L^2} \quad (A-13)$$

$$\frac{\partial}{\partial x_i} (-\rho \bar{u}_i' \bar{u}_j') \sim -\frac{\rho U'^2}{L} = -\text{Const} \cdot \frac{\rho U'^2}{L} \quad (A-14)$$

$$\frac{\partial}{\partial x_i} \left( \frac{\rho \bar{u}_i \bar{u}_i \bar{u}_j}{2} \right) \sim \frac{\rho U^3}{2L} = \text{Const} \cdot \frac{\rho U^3}{2L} \quad (A-15)$$

$$\mu \bar{u}_j \frac{\partial^2 \bar{u}_i}{\partial x_i \partial x_i} \sim \frac{\mu U^2}{L^2} = \text{Const} \cdot \frac{\mu U^2}{L^2} \quad (A-16)$$

$$\bar{u}_j \frac{\partial}{\partial x_i} \left( -\rho \bar{u}_i' \bar{u}_j' \right) \sim -\frac{\rho U'^2 U}{L} = -\text{Const} \cdot \frac{\rho U'^2 U}{L} \quad (A-17)$$

where  $U'$  is due to fluctuation, for  $i = j$ .

Using these relations, Equations A-5, A-8, A-10 become for

steady state, 1-dimensional, horizontal flow:

$$-\frac{dp}{dx} = \frac{C_1 \mu U}{L^2} + \frac{C_2 \rho U^2}{L} \quad (A-18)$$

$$-\frac{dp}{dx} = \frac{C_1 \mu U}{L^2} + \frac{C_2 \rho U^2}{L} + \frac{C_3 \rho U'^2}{L} \quad (A-19)$$

$$-U \frac{dp}{dx} = \frac{C_4 \mu U^2}{L^2} + \frac{C_5 \rho U^3}{L} + \frac{C_6 \rho U'^2 U}{LU} \quad (A-20)$$

Equation A-19 is similar to Equation 2.30 which Crafton obtained by using the force momentum balance. But Equation A-20 underlines the use of kinetic energy considerations in explaining the flow conditions. Both equations contain the Reynolds stresses and have the same closure problem.

Explanation of flow phenomena as velocity increases may not be very effective if the force momentum balance for mean steady flow is used(87). By Equation A-6b, the mean of the produced fluctuation is zero, therefore, its mean force momentum is zero. This poses a serious setback for Equation A-19.

Two possible expressions may be obtained when Equation A-20 is divided by U. The first expression reverts to the Reynolds equation and is identical to Equation A-19. The second is:

$$-\frac{dp}{dx} = \frac{C_4 \mu U}{L^2} + \frac{C_5 \rho U^2}{L} + \frac{C_6 \rho U^3 U}{UL} \quad (A-21)$$

Where the first two terms on the right are identical to the first two terms on the right of Equations A-18 and A-19. Therefore their constant coefficients C1 and C4, and C2 and C5, are respectively equal. The constant coefficient, C6, of Equation A-21, is different from C3 of Equation A-19. The coefficients of the terms on the right hand side of Equations A-18, A-19, and A-21 are constants since both viscosity, density and length are assumed constant.

Close resemblances are noticed between Equations A-18 and 2.17, and between Equations A-21 and 2.24. This suggests that Equations A-18 and A-21 are similar, respectively, to the second and third order equations of Forchheimer. The velocity scales in the third term on the right of Equation A-21 have been obtained as a combination of velocity fluctuation and mean velocity. Equation A-21 is made up of the differential pressure gradient on the left, and the viscous, the inertial, and the cubic term on the right.

If M, L, T, are denoted as dimensions of mass, length and time, both the differential pressure gradient and viscosity are represented dimensionally as  $[ML^{-2}T^{-2}]$  and  $[ML^{-1}T^{-1}]$  respectively. The inertial resistance coefficient in the inertial term has a dimension of  $[L^{-1}]$ . The permeability expression in the viscous term, has a dimension  $[L^2]$ , and can

be expressed as

$$k = N d^2 \quad (A-21a)$$

where  $N$  is a shape factor (88)

If these dimensional relations are applied on Equation A-21, an equation of the form of:

$$-\frac{dP}{dx} = \frac{\mu}{k} U + \beta \rho U^2 + \gamma \rho^2 U^2 U \quad (A-22)$$

is obtained where

$$k = L^2/C4 = [L^2] \text{ by dimension} \quad (A-22a)$$

$$\beta = C5/L = Fb = [L^{-1}] = \text{by dimension} \quad (A-22b)$$

$$\gamma = C6/(UL\rho) = \text{Gamma} = [LTM^{-1}] \text{ by dimension} \quad (A-22c)$$

$$UL\rho = [ML^{-1}T^{-1}] = \text{dimension of viscosity}$$

This implies that:

$$C6 = \text{dimensionless}$$

$$\gamma\mu = (C6*\mu)/(UL\rho) = C6/Re = \text{dimensionless} \quad (A-22d)$$

Comparing Equations A-21a and A-22a, it follows that

$$1/C4 = N = \text{a shape factor for } k$$

Similarly coefficients  $C5$  and  $C6$  could be shape factors for  $Fb$ , and  $\gamma$  respectively.

The foregoing relations imply the following properties

of gamma:

1. It is related to turbulent viscosity or shear resistance.
2. It acts in opposite sense to flow due to Reynolds stresses.
3. It is related to a dimensionless shape factor  $C_6$  which is significant at higher flow rates.
4. It is a flow functional parameter.

By inspection of Equation A-22 and from Equation A-6, both  $U'$  and  $U$  have the same order of magnitude at higher flow rates corresponding to higher Reynolds numbers. Let them be represented by the same velocity scale  $q$  such that  $q$  is free to assume values of velocity for increasing flow rates, and for different flow situations. The scale  $q$ , therefore, becomes a measurable quantity, and Equation A-22 can be written as:

$$-\frac{dp}{dx} = \frac{\mu}{k} q + F_b \rho q^2 + \gamma \rho^2 q^3 \quad (A-23)$$

while quadratic Equation A-18 can similarly be written as

$$-\frac{dp}{dx} = \frac{\mu}{k} q + F_b \rho q^2 \quad (A-23a)$$

Equation A-23 is identical to the relationship given as Equation 2.25 by Firoozabadi and Katz.

#### A.4 Significance of the Cubic Term

The foregoing equations depict the dominance of the cubic term at higher flow rates when fluctuations  $U'$  combine with the mean velocity  $U$  in the ratio of 2:1. Gamma has the dimensions of  $LTM^{-1}$  which is also the dimensions of the reciprocal of viscosity. From these dimensions, the inverse dependence of gamma on the shear resistance to flow, (viscosity), is suggested. Hence from Equation A-23, the differential pressure gradient increases, for increasing flow rates, and as shear resistance decreases for increasing values of gamma. These can be refuted for each increasing value of the flow rate as:

$$\frac{dP}{dx} \propto \frac{1}{\mu} \propto \gamma \quad (A-24)$$

Muskat(89) described this condition as being physically unreasonable.

From boundary layer theory(90), and by the condition of no slip at the boundary, fluid flow is characterized by very high resistance to flow at the boundary. (no slip condition), and a much lesser resistance at the centre (potential flow condition). The boundary layer is closest to the fluid-solid interface, and at the boundary, the flow rate is zero by no-slip condition. The nominal boundary layer thickness,  $d$ , is the normal distance from the fluid-solid interface at which the velocity is 99% of the free stream velocity in the approximately potential flow region.

For a small cross-sectional area available to flow, as in Figure A-1, the effective available cross-sectional area, situated at the centre, is quite small. Resistance to flow induces a creeping flow condition in the centre. With higher flow rates, corresponding to a large differential pressure gradient, the flow encounters more resistance as the boundary layer is permeated. Subsequently, by overcoming this resistance, the boundary layer thickness decreases and the cross-sectional area open to the flow increases. The viscosity value within the increased cross-sectional area, open to flow, is lower than its original value before the boundary layer was decreased.

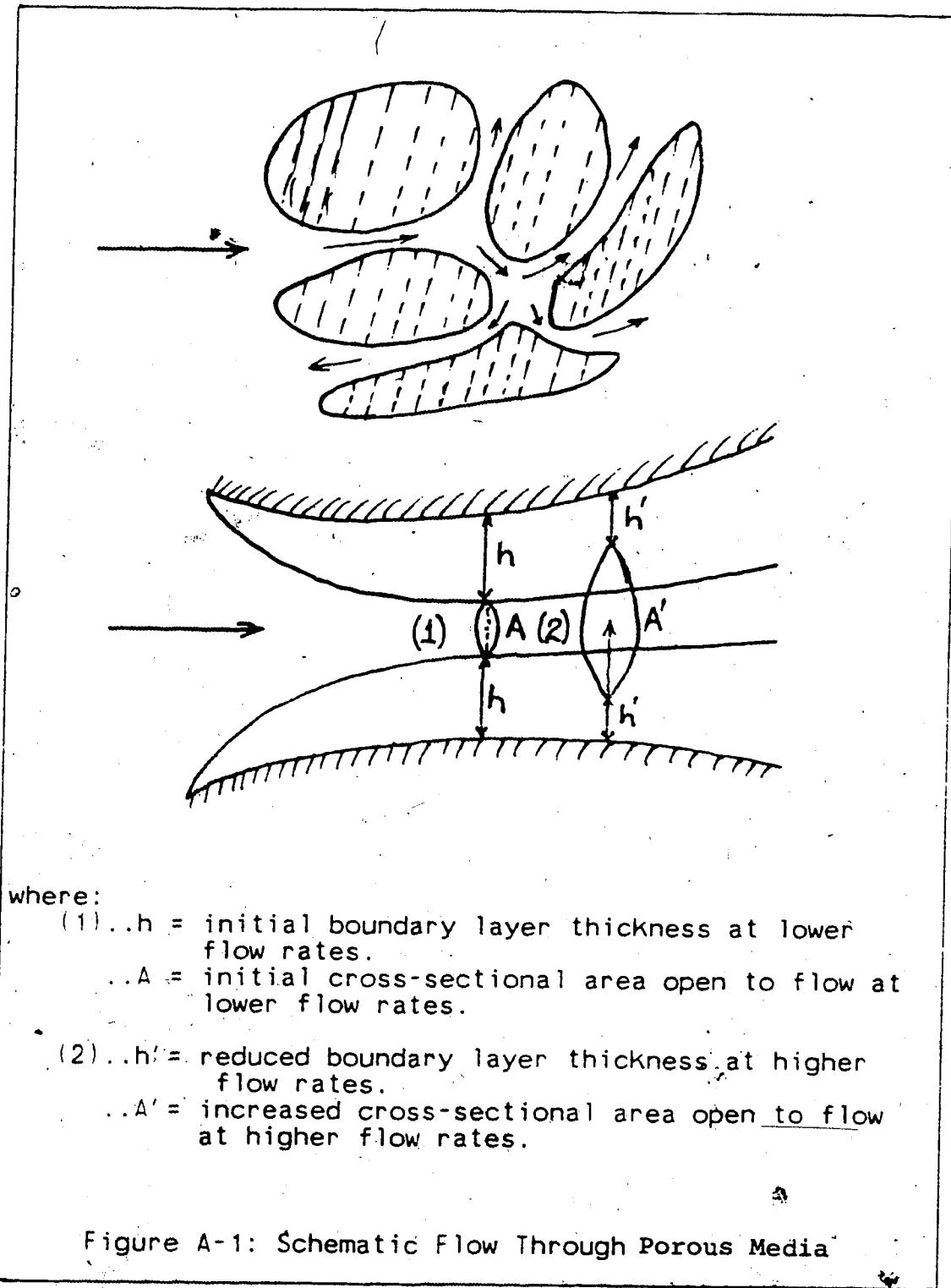
Therefore, with increasing flow rates,

1. differential pressure gradient increases,
2. boundary layer thickness decreases,
3. cross-sectional area open to flow increases;
4. resistance to flow in the increased cross-sectional area decreases as the boundary layer growth is reduced in the flow direction.

If  $A'$  is this new cross-sectional area, these can jointly be related as

$$\frac{dp}{dx} \propto \frac{1}{\mu} \propto A' \quad (A-25)$$

Equations A-24 and A-25, which are similar in expression, suggest that  $\gamma$  is related to  $A'$  and that as  $\gamma$  increases for increasing kinetic energy of flow, both



the boundary layer thickness, and the resistance to flow closest to the centre, decrease. For lower flow rates with negligible kinetic energy of flow, the cubic term is not necessary.

By Schlichting, the ratio of the boundary layer thickness to the length scale of the path varies inversely with the square root of the Reynolds number. Similarly, the total drag coefficient, D, or the skin frictional coefficient(91), varies inversely with the square root of the Reynolds number. Combining these relations, results in:

$$\frac{d}{L} \sim \frac{1}{\sqrt{R_e}} \sim D = \frac{\tau_0}{\rho U^2} = \sqrt{\frac{u}{\rho UL}} \quad (A-26)$$

which in rewritten form can be expressed as:

$$D^2 = \frac{\mu}{LU\rho} = \frac{C}{R_e} = \text{dimensionless} \quad (A-27)$$

Equation A-27 is similar to Equation A-22c and combining both equations, gives

$$\gamma u = \frac{\mu}{LU\rho} = \frac{C_6}{R_e} = D^2 = \text{dimensionless} \quad (A-28)$$

From this expression, gamma depends directly on the square of the skin frictional coefficient, and inversely on both viscosity and the Reynolds number. The dimensionless shape factor C6 is related to the square of the skin frictional coefficient,D.

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These factors, including gamma, operate at higher flow rates, and with higher Reynolds number, within the boundary layer.

APPENDIX B  
ADAPTATION OF MODEL EQUATIONS TO COMPUTER APPLICATIONS.

## B. APPENDIX B - ADAPTATION OF MODEL EQUATIONS TO COMPUTER APPLICATION

### B.1 Equations for Parameter Estimation

Equation 4.15 gives a general cubic flow equation in terms of  $k_a$ ,  $F_b$ , and gamma. In its rearranged form, as in Equation 4.23, it is:

$$\frac{P_e^2 - P_w^2}{CC1} = \frac{1}{k_a} + \frac{CC2}{CC1} \cdot F_b + \frac{CC3}{CC1} \cdot Y \quad (B-1)$$

This equation is linear in its coefficients. Designating the desired parameters as coefficients, a multiple linear regression technique for estimating the parameters can be developed. Subsequently, Equation B-1 is put in the following form:

$$Y = W_0 + W_1 X_1 + W_2 X_2 \quad (B-2)$$

where

$$Y = (P_e^2 - P_w^2)/CC1 \quad W_0 = 1/k_a$$

$$X_1 = CC2/CC1 \quad W_1 = F_b$$

$$X_2 = CC3/CC1 \quad W_2 = \text{gamma}$$

Equation 4.16, rearranged below,

$$\begin{aligned} \frac{P_e^2 - P_w^2}{CC1} &= \frac{1}{k} - \frac{PEW}{CC1} \cdot b + \frac{CC2}{CC1} \cdot F_b + \frac{CC2}{CC1 \times P_m} \cdot (b \cdot F_b) \\ &\quad + \frac{CC3}{CC1} \cdot Y + \frac{CC3}{CC1 \times P_m} \cdot (Y \cdot b) \end{aligned} \quad (B-3)$$

can similarly be treated if its combined parameters are

represented by single variables. In this condition, Equation B-3 reduces to

$$Y = W_0 + W_1 X_1 + W_2 X_2 + W_3 X_3 + W_4 X_4 + W_5 X_5 \quad (B-4)$$

where

$$Y = (P_e^2 - P_w^2)/CC1 \quad W_0 = 1/k$$

$$X_1 = -PEW/CC1 \quad W_1 = b$$

$$X_2 = CC2/CC1 \quad W_2 = F_b$$

$$X_3 = CC2/(CC1 \cdot P_m) \quad W_3 = b \cdot F_b$$

$$X_4 = CC3/CC1 \quad W_4 = \gamma$$

$$X_5 = CC3/(CC1 \cdot P_m) \quad W_5 = \gamma \cdot b$$

For the quadratic case, Equations B-1 and B-3 reduce to

$$\frac{P_e^2 - P_w^2}{CC1} = \frac{1}{k_a} + \frac{CC2}{CC1} \cdot F_b \quad (B-5)$$

$$\frac{P_e^2 - P_w^2}{CC1} = \frac{1}{k} - \frac{PEW}{CC1} \cdot b + \frac{CC2}{CC1} \cdot F_b + \frac{CC2}{CC1 \cdot P_m} \cdot (b \cdot F_b) \quad (B-6)$$

which are equivalent to Equations 4.21 and 4.22. By similarly linearizing their coefficients, the following forms are obtained:

$$Y = W_0 + W_1 X_1 \quad (B-7)$$

$$Y = W_0 + W_1 X_1 + W_2 X_2 + W_3 X_3 \quad (B-8)$$

They follow directly the form of Equations B-2 and B-4.

### B.2 Equations for Model Fitting The Data

Equation (4.22), reproduced below,

$$(P_e^2 - P_w^2) = \left( \frac{C_1 \mu T Z}{k_a} \right) \cdot Q_0 + (C_2 \bar{Z} \bar{T} F_b) \cdot Q_0^2 + (C_3 \bar{Z} \bar{T} Y) \cdot Q_0^3 \quad (B-9)$$

is a polynomial in  $Q_0$  which can be expressed as

$$Y = W_1 X_1 + W_2 X_2 + W_3 X_3 \quad (B-10)$$

where

$$Y = P_e^2 - P_w^2$$

$$W_1 = (C_1 \mu T Z) / k_a \quad X_1 = Q_0$$

$$W_2 = C_2 \bar{Z} \bar{T} F_b \quad X_2 = Q_0^2$$

$$W_3 = C_3 \bar{Z} \bar{T} Y \quad X_3 = Q_0^3$$

For the quadratic case, Equations B-9 and B-10,

respectively, reduce to

$$(P_e^2 - P_w^2) = \left( \frac{C_1 \mu T Z}{k_a} \right) \cdot Q_0 + (C_2 \bar{Z} \bar{T} F_b) \cdot Q_0^2 \quad (B-11)$$

$$Y = W_1 X_1 + W_2 X_2 \quad (B-12)$$

### B.3 Multiple Linear Regression

The foregoing equations have the following general form, which is linear in the coefficients.

$$Y_i = W_0 + W_1 X_{i1} + W_2 X_{i2} + \dots + W_n X_{in} + e_i \quad (B-13)$$

or, in condensed form:

$$Y_i = W_0 + \sum_{j=1}^n W_j X_{ij} + e_i \quad (B-13a)$$

for  $i = 1, N$ ,  $j = 1, n$ , and  $e$  is the error.

By method of least squares (92, 93), the sum of error squares with respect to each of the parameters should be a minimum such that

$$\frac{\partial (\sum_{i=0}^n e_i^2)}{\partial W_j} = 0 \quad (B-14)$$

Solution of this equation for the best estimates of  $W_j$ , and rearranging, results in normal equations of order  $n$ :

$$\sum_{i=1}^n (x_{ik} - \bar{x}_k)(y_i - \bar{y}) = \sum_{i=1}^n \left[ \left\{ \sum_{j=1}^n (x_{ik} - \bar{x}_k)(x_{ij} - \bar{x}_j) \right\} \cdot w_j \right] \quad (B-15)$$

where  $k = 1, n$ , and  $y, x$ , are respectively arithmetic means of  $Y$  and  $X$ . The value of  $w_0$  is determined as

$$w_0 = \bar{y} - \sum_{j=1}^n w_j \bar{x}_j \quad (B-16)$$

If in matrix form:

$$E(I) = Y - Y_{\text{mean}}$$

$$D(I, J) = X(I, J) - X_{\text{mean}}(J)$$

$$DT(J, I) = D(I, J) = X(I, K) - X_{\text{mean}}(K) = D_{\text{transpose}}$$

Equation B-15 becomes

$$DT(J, I) * E(I) = DT(J, I) * D(I, J) * W(J) \quad (B-18)$$

Furthermore, if

$$B(J) = DT(J, I) * E(I)$$

$$A(J, J) = DT(J, I) * D(I, J)$$

then

$$B(J) = A(J, J) * W(J) \quad (B-19)$$

and

$$W(J) = B(J) / A(J, J) \quad (B-20)$$

The value of  $w_0$  is calculated from Equation B-16.

The foregoing procedure is general for both parameter estimation and model fitting by polynomial fit.

### B.3.1 The Problem of Ill-Conditioned Matrices

The solution of Equation B-19 is possible only if the generated symmetrical matrix  $A(J;J)$  is non-singular and has an inverse. Ill-conditioning is a common problem encountered in this type of generated matrix. The determinant is very small and, depending on the number of parameters to be estimated, can be almost zero. This is a characteristic of Hilbert matrices found generally in physical systems(95).

Some suggested techniques aimed at minimizing such ill-conditioning are:

1. Run the program in double precision.
2. Ensure that the equations of D, and its transpose, are correct.
3. As much as possible, perform divisions last. Minimize truncations of significant decimal figures.
4. In matrix inversion, apply pivotal row interchange.
5. If possible, restrict the number of parameters to a minimum(96).

It was noticed that neglecting the combined variables enhanced the stability of the matrix system in terms of ill-conditioning.

#### B.4 Linear Fit For Graphical Parameter Estimation

A linear fit is performed for the plot points in the linear sections of both the Klinkenberg and the visco-inertial plots, by the method of least squares. The degree of linear fit is measured by the linear correlation coefficient,  $R_{aa}$ :

$$R_{aa} = \frac{N \sum X_i Y_i - \sum X_i \sum Y_i}{[\sum (X_i^2) - (\sum X_i)^2]^{\frac{1}{2}} [\sum (Y_i^2) - (\sum Y_i)^2]^{\frac{1}{2}}} \quad (B-21)$$

This linear fit does not require matrix formation.

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APPENDIX C  
GRAPHICAL RESULTS

170

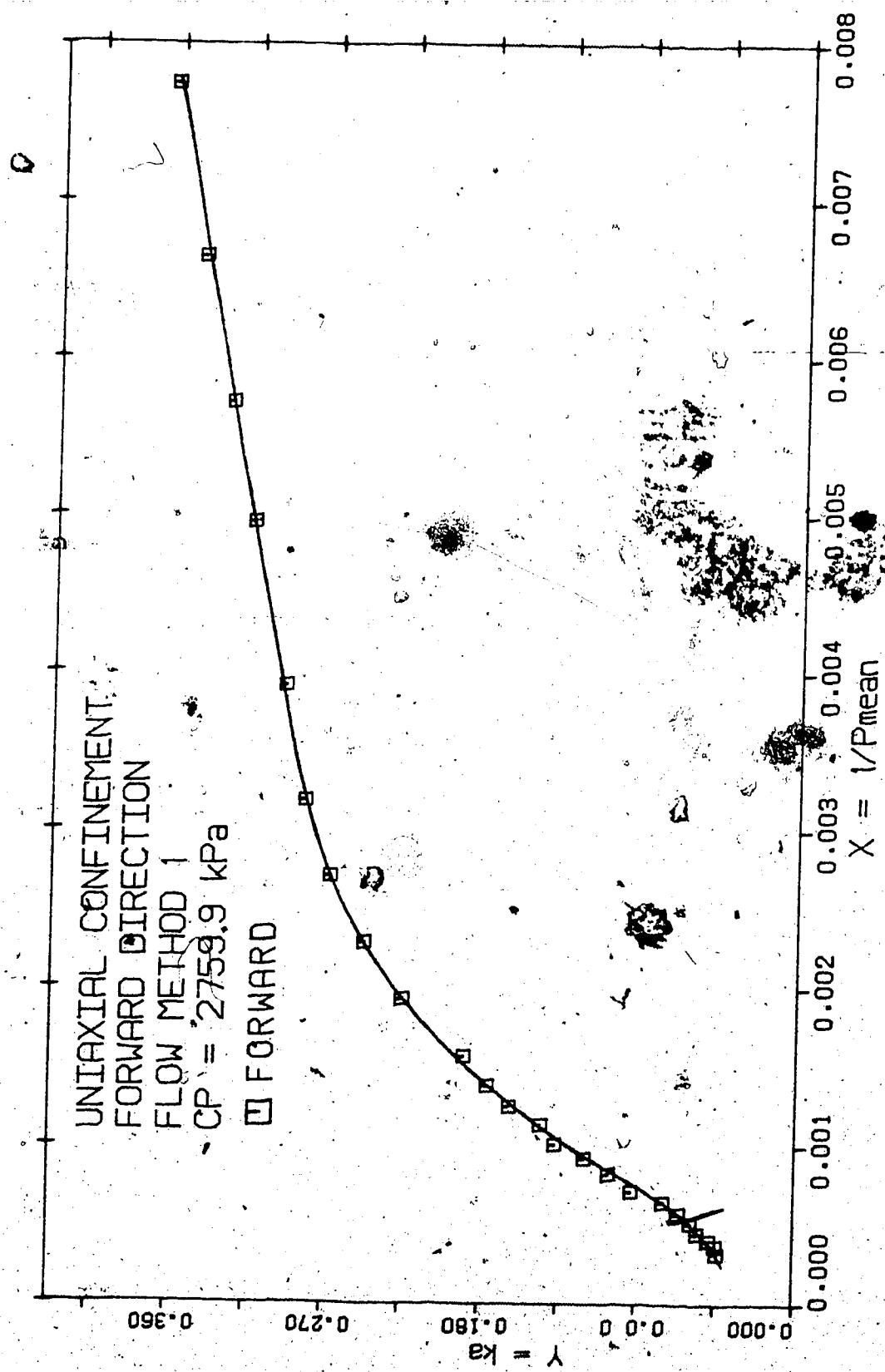


FIGURE C-1: CORE 1C KLINKENBERG PLOT RUN 3

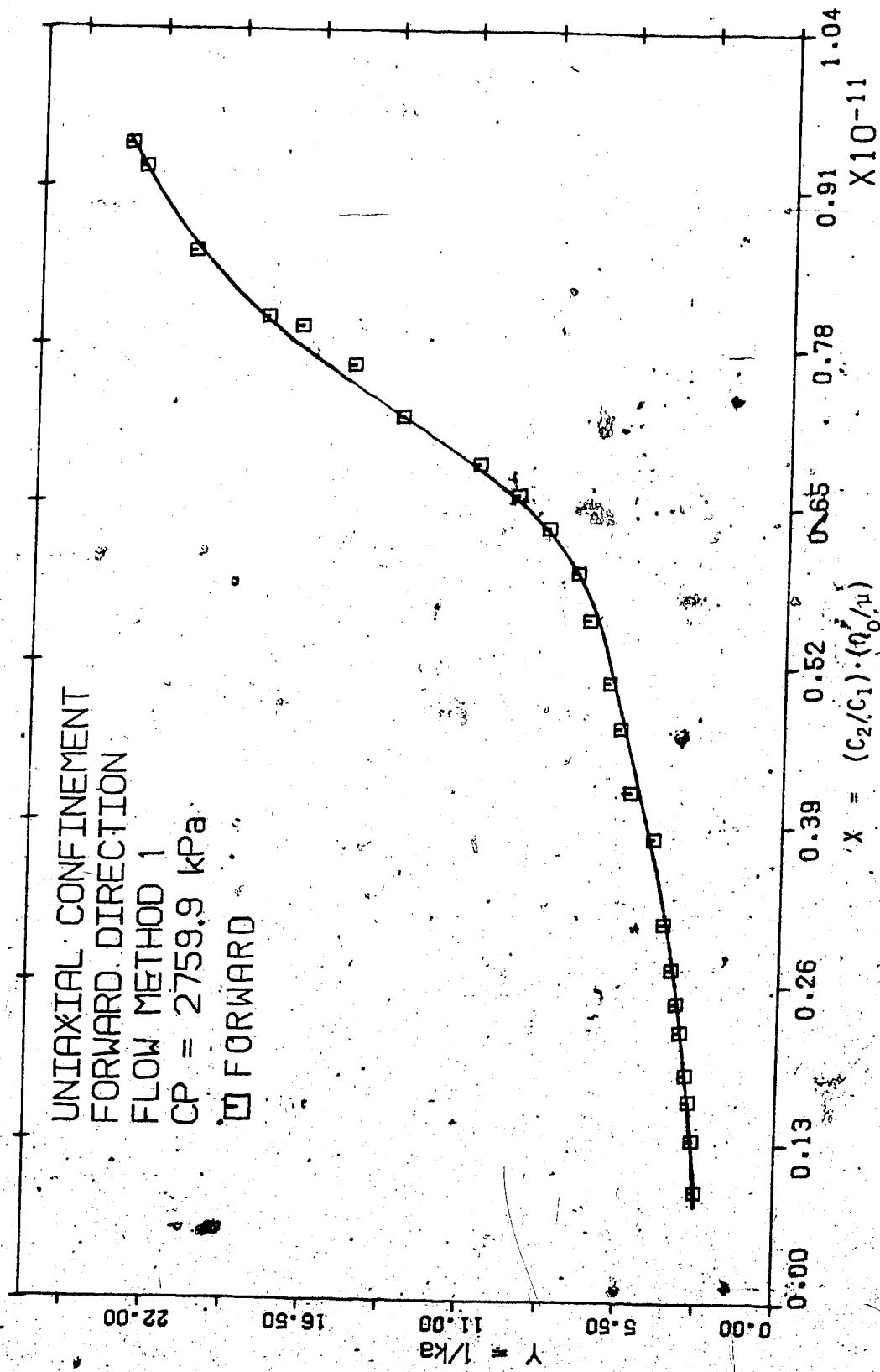


Figure C-2: CORE 1C VISCO-INERTIAL PLOT RUN 3

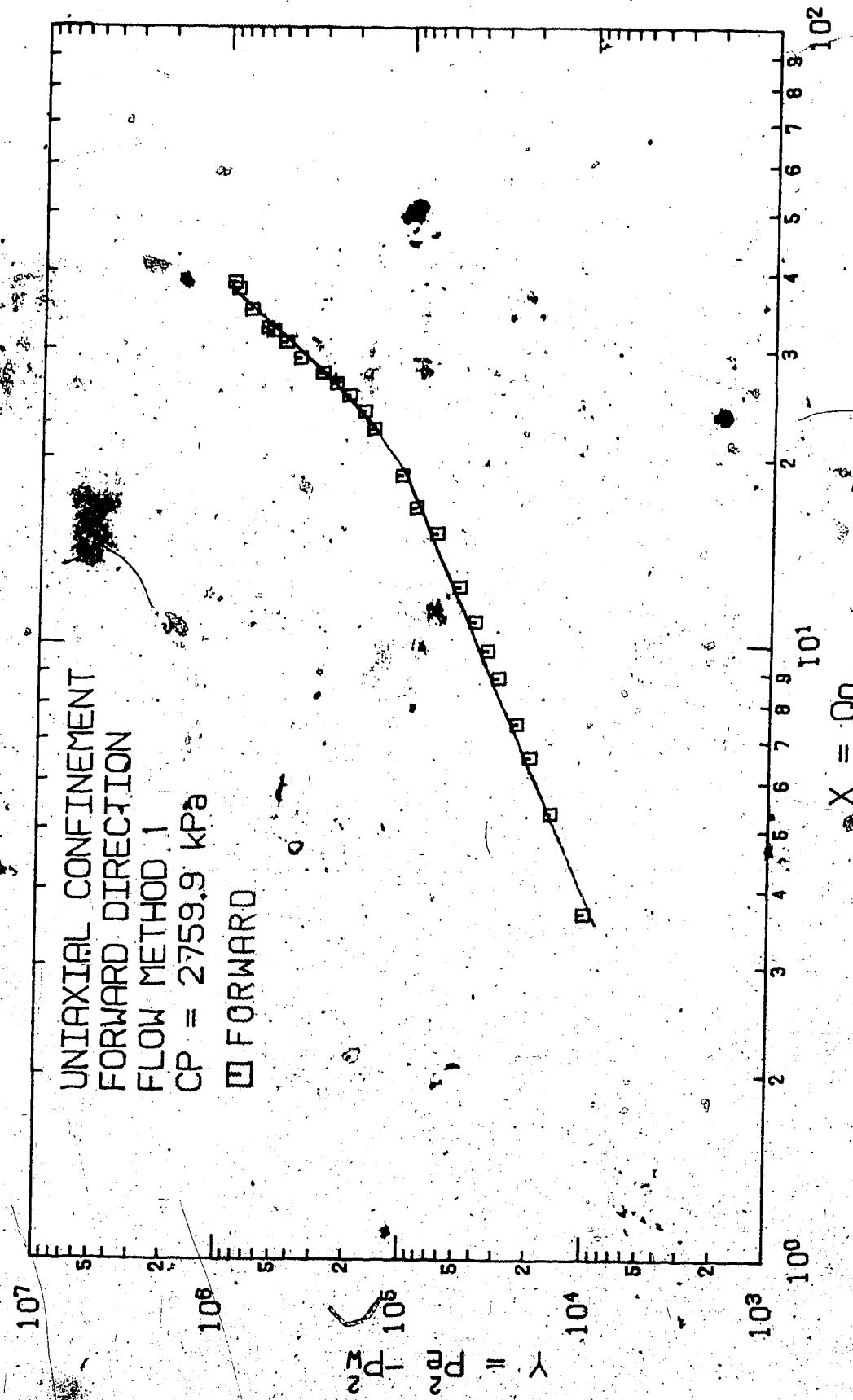


Figure C-3: CORE 1C BACK PRESSURE PLOT RUN 3

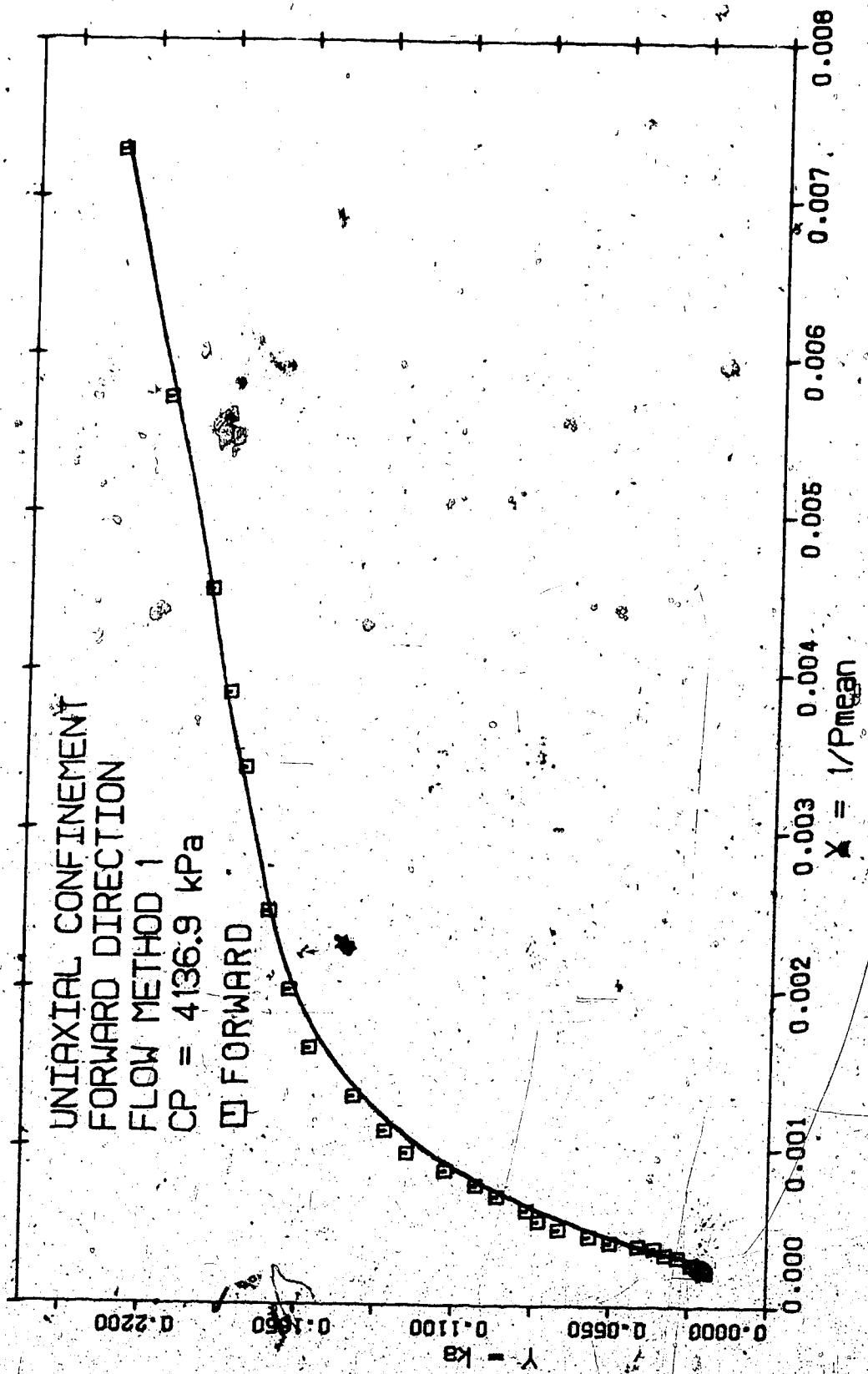


FIGURE C-4 CORE 4A KLINKENBERG PLOT RUN 4

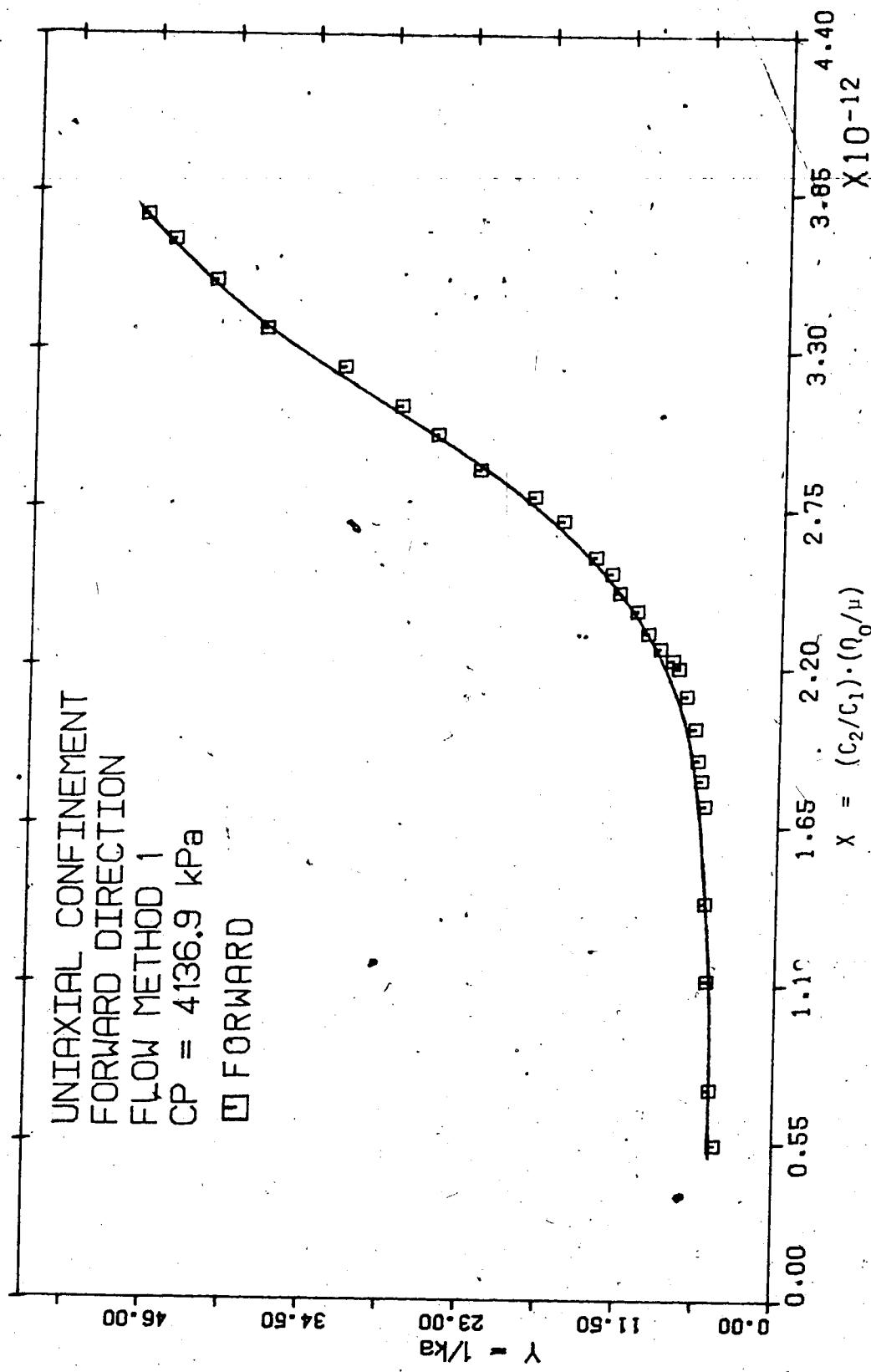


Figure C-5 CORE 4A VISCO-INERTIAL PLOT RUN 4

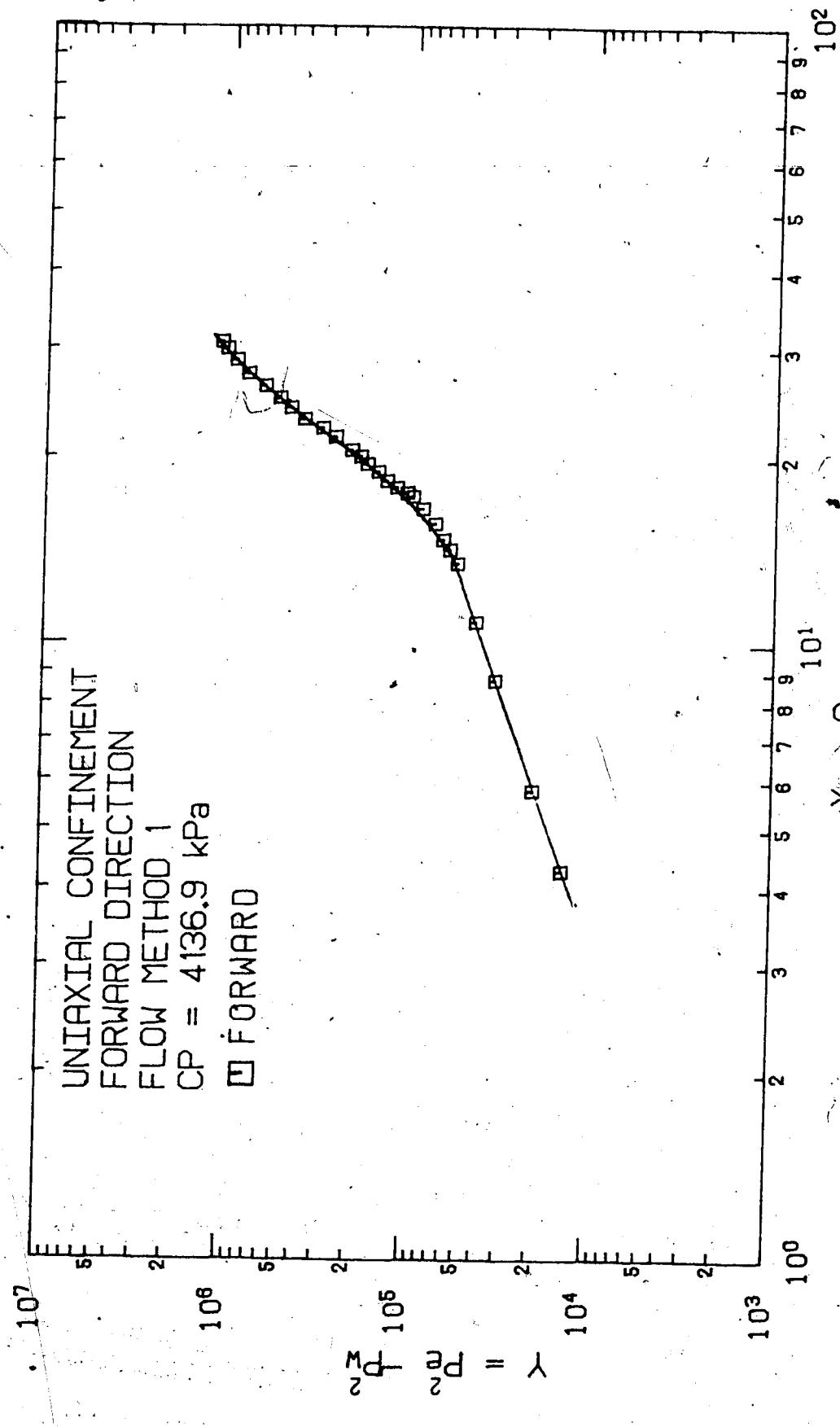


Figure C-6 CORE 4A BACK PRESSURE PLOT RUN 4

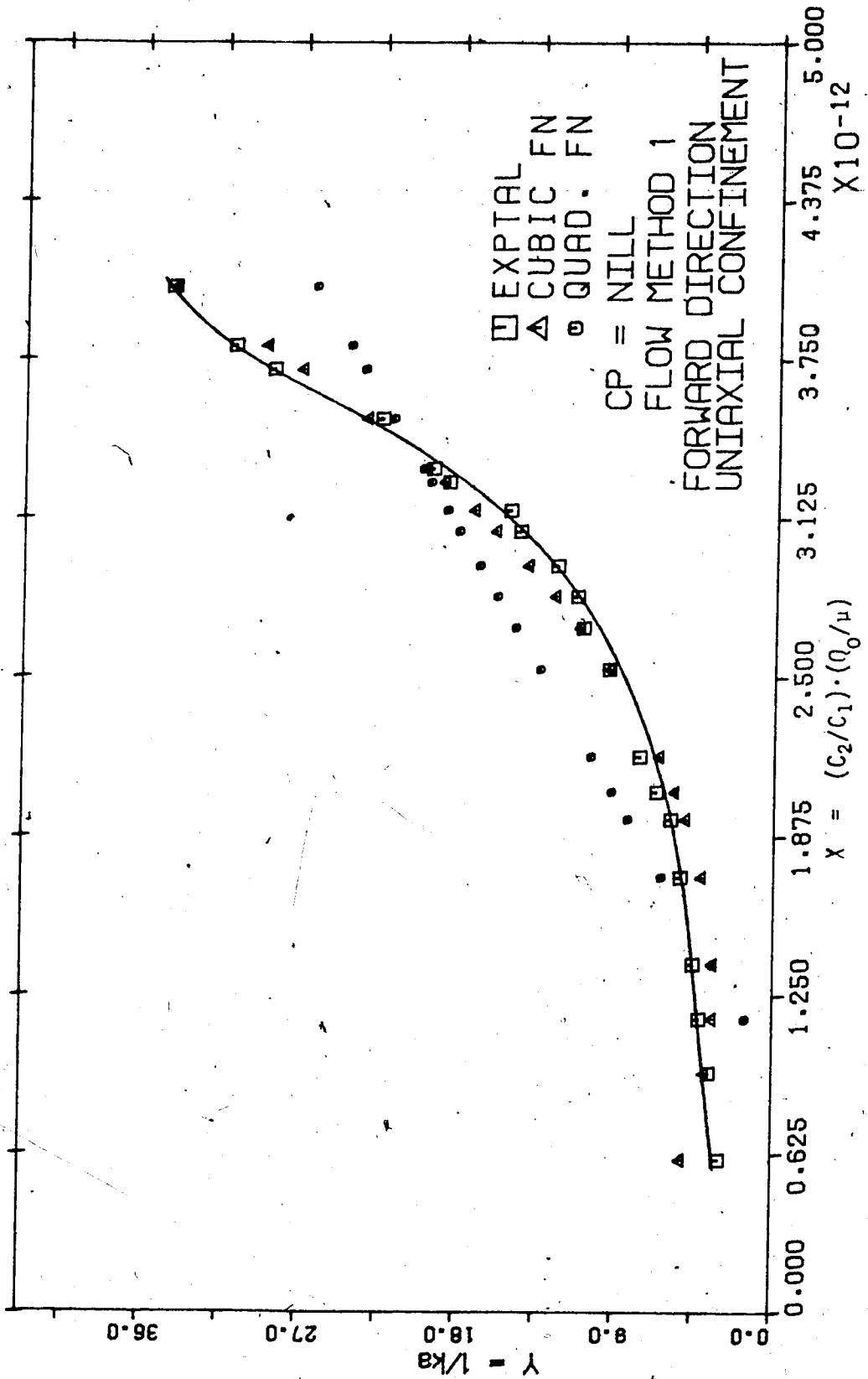


Figure C-7 CORE 4A RUN 1 VISCO-INERTIAL FIT PLOT

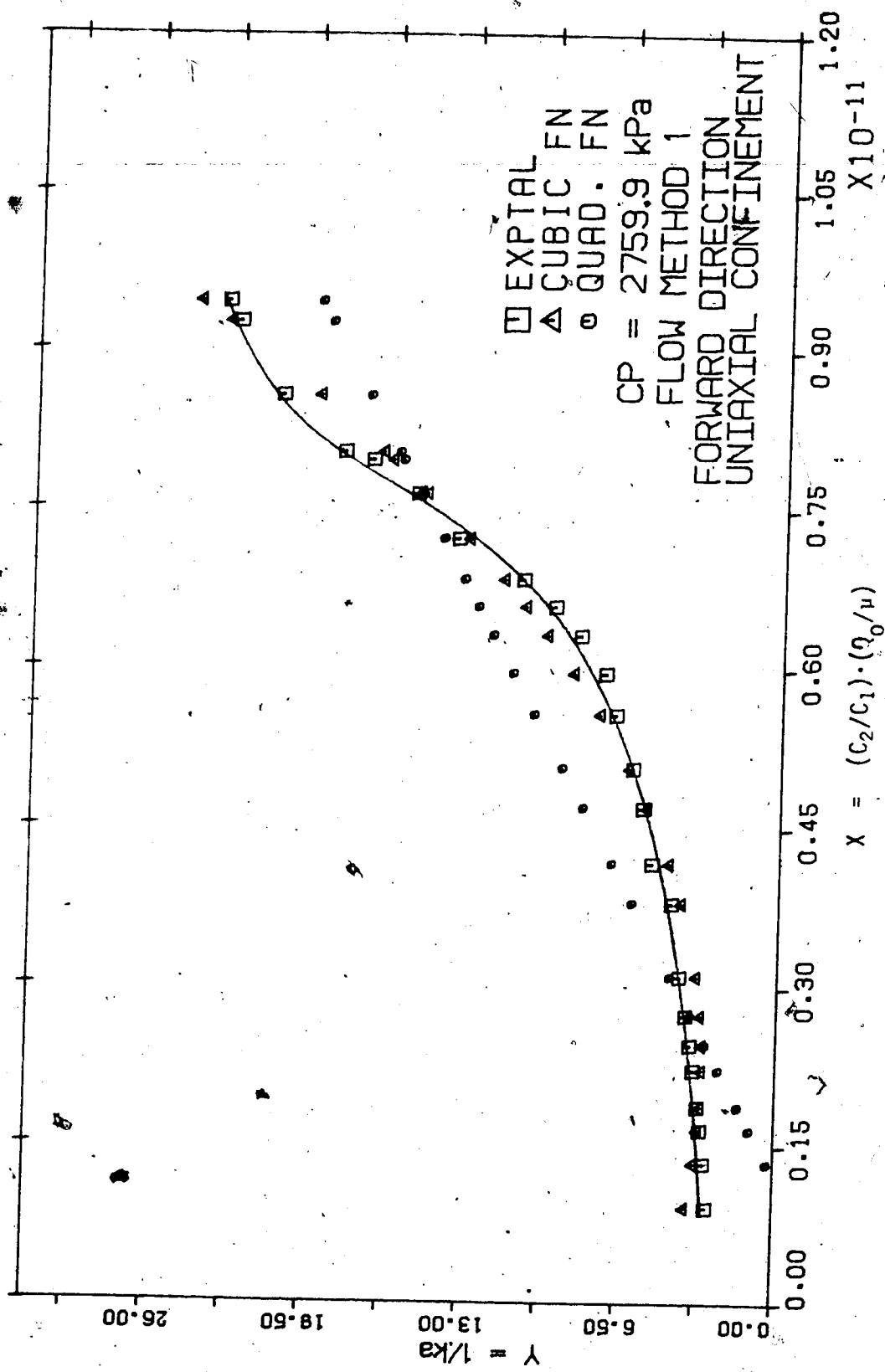


Figure C-8 CORE 1C RUN 3 VISCO-INERTIAL FIT PLOT

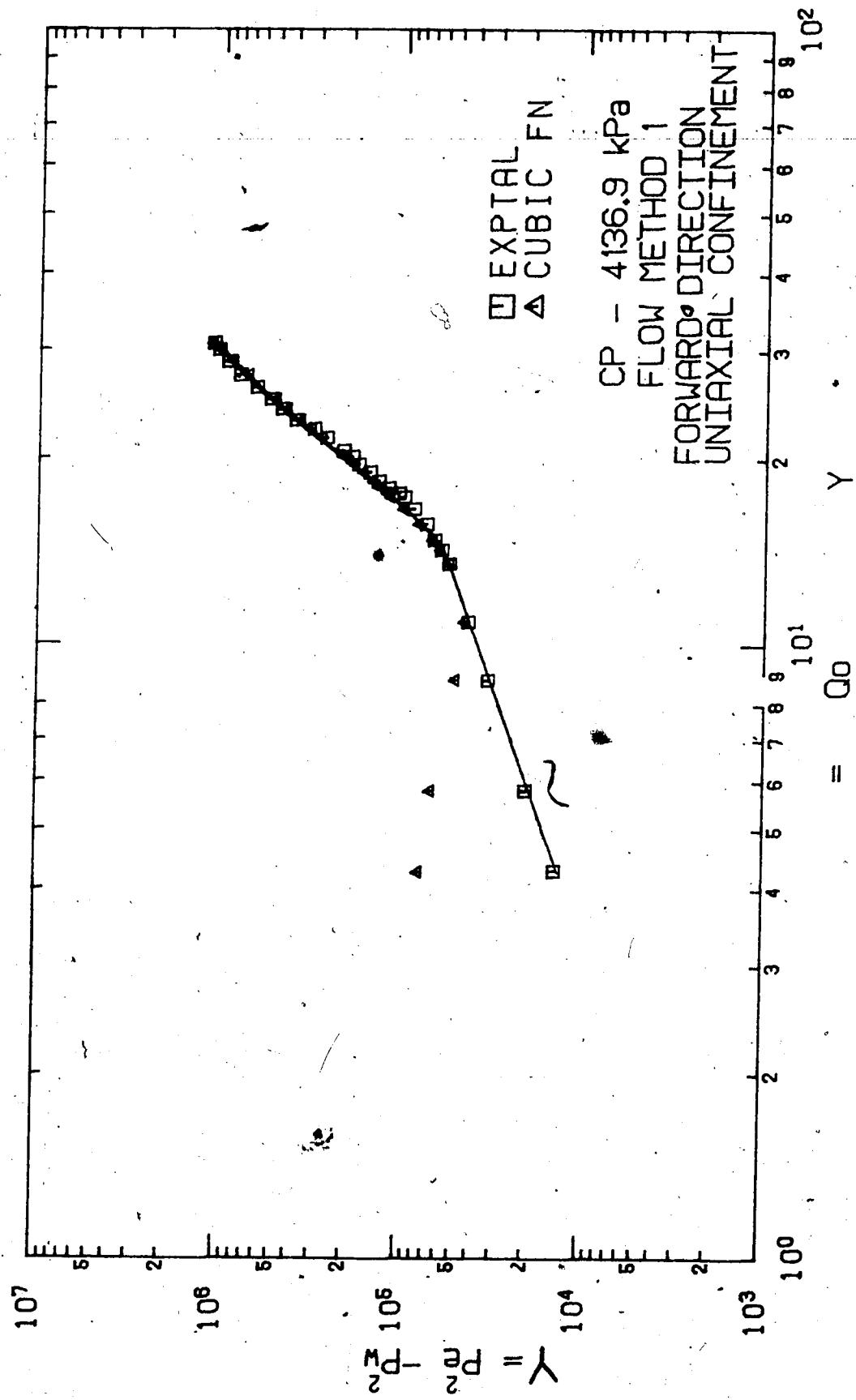


Figure C-9 CORE 4A RUN 4 BACK PRESSURE FIT PLOT X

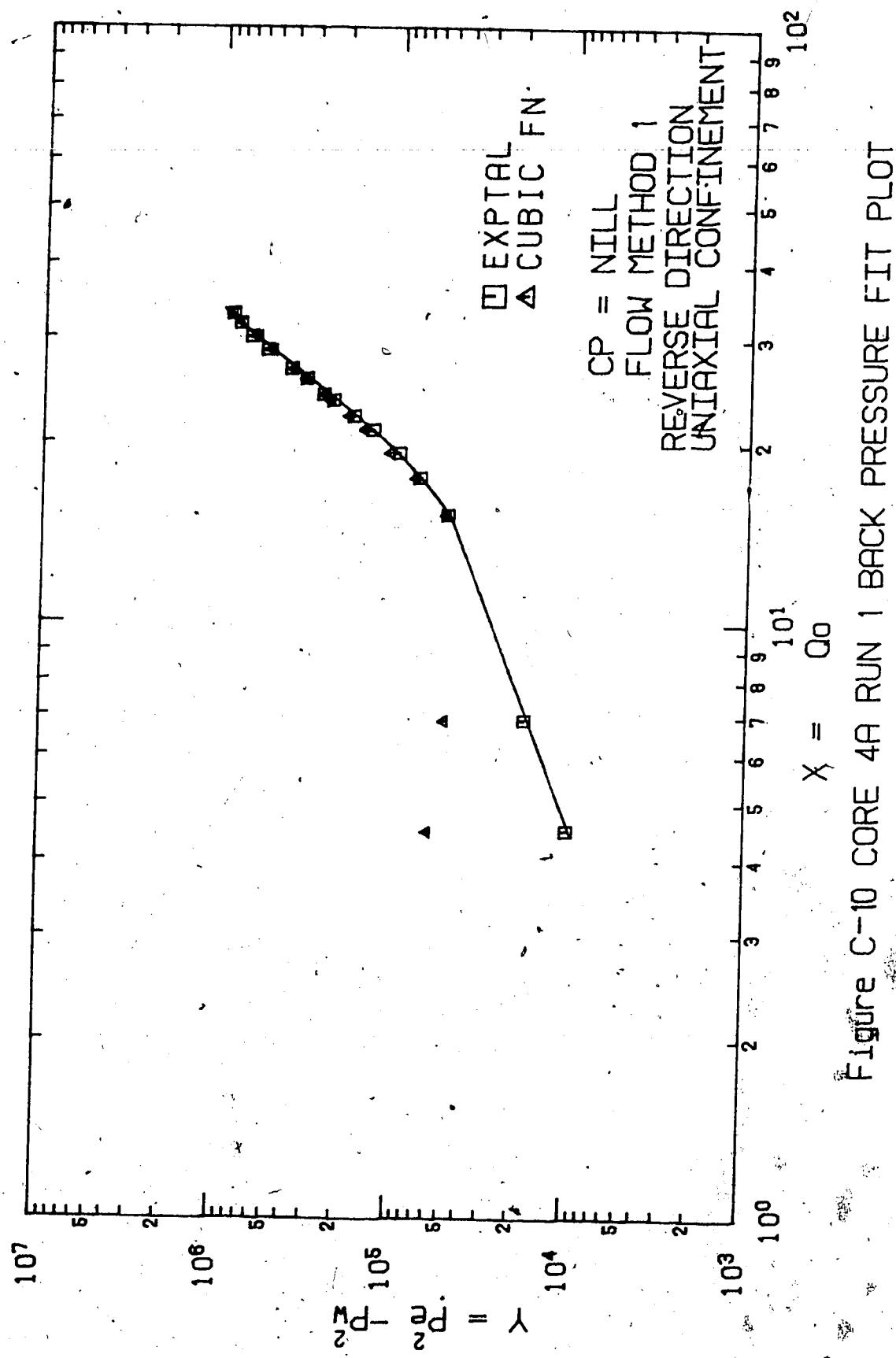


Figure C-10 CORE 4A RUN 1 BACK PRESSURE FIT PLOT

APPENDIX D  
COMPUTER PROGRAM AND DATA LISTINGS

## D. APPENDIX D - COMPUTER PROGRAM AND DATA LISTINGS

### D.1 PROGRAM LISTINGS

```
C          LIN
C
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
C      THIS PROGRAM PLOTS GRAPHS WITH A LINEAR LEAST SQUARE
C      FIT
C
C      Uses Applot++plotlib on Object File
C
C      Subroutines: LAGINT, LINXY, GRAPH, BOX
C
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
C      EM=MOLECULAR WEIGHT
C      TR, PR, DR=REDUCED TEMPERATURE, PRESSURE, AND DENSITY
C      RESPECTIVELY
C      VIS=VISCOSITY IN MICRO-PA.S
C      GG=GAS GRAVITY
C      IMPLICIT REAL*8(A-H,O-Z)
C      REAL*4 DIRECT
C      REAL*8 DLOG
C      DIMENSION PE(60), PW(60), TE(60), TW(60), PM(60), W(60),
C      $VIS(60), ZA(60), PB(60), CP(60), TS(60), QO(60), PEW(60),
C      $CC1(60), CC2(60), CC3(60), PD(60), X(60), RD(60), AX(60),
C      $T(60), TCW(60), E(60,100), ER(60), Y(60), AY(60),
C      $BX(60), BY(60), CX(60), CY(60), DX(60), DY(60), EX(60),
C      $EY(60), FX(60), FY(60), V(60), ZX(60), ZY(60), DIRECT(5),
C      $EW(60), DMP(60)
C      DATA TC, EM, PC, SIG/1.262D+02, 2.8013D+01, 3.35D+01,
C      $3.798D+00/
C      GG=EM/2.896D+01
C      PCA=PC*(1.01325D+02)
C      READ(5,1) NUM, NO, DIRECT, IFLAG
C      READ(5,90) RE, RW, H, POR
C      READ(5,3) NDJ
C      READ(5,4) R, BZ
C
C      IF(IFLAG.NE.(100)) GO TO 65
C      RE=RE*3.048D-01
C      RW=RW*3.048D-01
C      H=H*3.048D-01
C      GO TO 65
65  WRITE(6,2) NUM, NO, DIRECT
      WRITE(6,89) RE, RW, H, POR
      C1=((1.2954909D-03)*DLOG(RE/RW))/H
      C2=((2.9228842D-15)*GG*(1/RW-1/RE))/H*H
```

```

1 FORMAT(A4,A4,5A4,I3)
90 FORMAT(4F10.7)
3 FORMAT(2I3)
2 FORMAT(/4X,'CORE SAMPLE NO ',A4,2X,'RUN ',A4,2X,5A4)
89 FORMAT(4X,'RE=',F10.7,2X,'RW=',F10.7,2X,'H=',F10.7,
        12X,'POROSITY=',F10.7)
4 FORMAT(2F10.6)
IF(IFLAG.EQ.(100)) GO TO 67
C
      READ(5,6)(PE(I),PW(I),CP(I),PB(I),TCE(I),TCW(I),
$ TS(I),I=1,NDJ)
6 FORMAT(7F11.4)
      WRITE(6,7)
7 FORMAT(/14X,'PE',12X,'PW',12X,'CP',12X,'PB',11X,
$ 'TCE',12X,'TCW',12X,'TS',/13X,'(PSI)',9X,'(PSI)',9X,
$ '(PSI)',9X,'(CMHG)',7X,'(MVOLT)',8X,'(MVOLT)',8X,
$ '(SEC)')/
      WRITE(6,8)(PE(I),PW(I),CP(I),PB(I),TCE(I),TCW(I),
$ TS(I),I=1,NDJ)
8 FORMAT(6X,7E14.5)
C
      DO 9 I=1,NDJ
      PB(I)=PB(I)*1.33322D+00
      PE(I)=PB(I)+PE(I)*6.894757D+00
      PW(I)=PB(I)+PW(I)*6.894757D+00
      CP(I)=PB(I)+CP(I)*6.894757D+00
      TE(I)=((TCE(I)+2.145588250D-02)/4.046470588D-02)
$ +273.15
      TW(I)=((TCW(I)+2.145588250D-02)/4.046470588D-02)
$ +273.15
      W(I)=(TE(I)+TW(I))/2.
      QO(I)=(6.116438868D+02)/TS(I)
9 CONTINUE
      GO TO 71
C
67 READ(5,69)(PE(I),PW(I),W(I),QO(I),CP(I),I=1,NDJ)
69 FORMAT(5F12.8)
      DO 68 I=1,NDJ
      PE(I)=PE(I)*6.894757D+00
      PW(I)=PW(I)*6.894757D+00
      W(I)=((W(I)-3.2D+01)/1:8D+00)+2.7315D+02
      QO(I)=QO(I)*2.86364D+01
      CP(I)=CP(I)*6.894757D+00
68 CONTINUE
C
71 DO 70 I=1,NDJ
      PM(I)=(PE(I)+PW(I))/2.
      P=PM(I)
      PR=P/PCA
      T=W(I)
      TR=T/TC
C
      CALL LAGINT(P,T,TR,PR,Z,DR,VS)
      ZA(I)=Z

```

```

RD(I)=DR
VIS(I)=VS
C
PEW(I)=2.*(PE(I)-PW(I))
CC1(I)=C1*ZA(I)*VIS(I)*T*QO(I)
CC2(I)=C2*ZA(I)*T*QO(I)**2
CC3(I)=CC2(I)/PM(I)
70 CONTINUE
READ(5,18) EK1,BET, BETA,T,GAMA1
READ(5,19) EKC,BETAC,GAMAC
18 FORMAT(4E20.5)
19 FORMAT(3E20.5)
WRITE(6,10)
10 FORMAT(/11X,'PEW',11X,'CC1',9X,'CC2',9X,'CC3')
WRITE(6,11) (PEW(I),CC1(I),CC2(I),CC3(I),I=1,NDJ)
11 FORMAT(4X,4E13.5)
C
DO 77 I=1,NDJ
CY(I)=(PM(I)*PEW(I))
CX(I)=(QO(I))
EW(I)=(PE(I)-PW(I))
DMP(I)=EW(I)/PM(I)
77 CONTINUE
WRITE(6,78)
78 FORMAT(/8X,'EVALUATING BACK PRESSURE PLOT VALUES',
//13X,'Y=(PE2-PW2)',8X,'X=(QO)',14X,'PM',14X,'PE-PW',
$11X,'(PE-PW)/PM')
WRITE(6,51) (CY(I),CX(I),PM(I),EW(I),DMP(I),I=1,NDJ)
51 FORMAT(6X,5G18.5)
C
ICOUNT=1
DO 53 I=1,NDJ
Y(I)=(PE(I)*PE(I)-PW(I)*PW(I))/CC1(I)
X(I)=(CC2(I))/CC1(I)
53 CONTINUE
C
57 CALL LINXY(X,Y,NDJ,A,BETA,SIGMAA,SIGBET,RAA,VAR)
C
IF(ICOUNT-2) 58,59,60
58 EK=1.0/A
WRITE(6,54)
54 FORMAT(/8X,'EVALUATING VISCO-INERTIAL PLOT VALUES',
//11X,'Y=(PE2-PW2)/CC1',6X,'X=CC2/CC1')
WRITE(6,55) (Y(I),X(I),I=1,NDJ)
WRITE(6,52) EK,BETA,SIGMAA,SIGBET,VAR,RAA
52 FORMAT(/6X,'KABS=',G12.5,5X,'BETA=',G12.5,/6X,
$'SIGMAK=',G12.5,2X,'SIGMABETA=',E12.5,/6X,'VARIANCE='
$,G12.5,1X,'LIN.CORR.=',G12.5)
55 FORMAT(6X,2G18.5)
DO 101 I=1,NDJ
AX(I)=X(I)
AY(I)=Y(I)
101 CONTINUE
C

```

```

DO 56 I=1,NDJ
Y(I)=CC1(I)/(PE(I)*PE(I)-PW(I)*PW(I))
X(I)=1.0/PM(I)
56 CONTINUE
74 ICOUNT=ICOUNT+1
GO TO 57
C
59 EK=A
BK=BETA
B=BETA/A
WRITE(6,62)
62 FORMAT(//8X,'EVALUATING KLINKENBERG PLOT VALUES',
//11X,'Y=CC1/(PE2-PW2)',7X,'X=1/PM')
WRITE(6,63) (Y(I),X(I),I=1,NDJ)
WRITE(6,61) EK,B,SIGMAA,SIGBET,VAR,RAA
61 FORMAT(/6X,'K=',G12.5,10X,'B=',G12.5,/6X,'SIGMAK=',,
$G12.5,4X,'SIGMAB*K=',G12.5,/6X,'VARIANCE=',G12.5,
$1X,'LIN.CORR.=',G12.5)
63 FORMAT(6X,2G18.5)
DO 100 I=1,NDJ
BX(I)=X(I)
BY(I)=Y(I)
100 CONTINUE
C
DO 73 I=1,NDJ
Y(I)=(PEW(I)/CC1(I))*(PM(I)+B)
X(I)=(CC2(I)/CC1(I))*(1.0+B/PM(I))
73 CONTINUE
GO TO 74
C
60 EK=1.0/A
WRITE(6,12)
12 FORMAT(//8X,'EVALUATING MODIFIED VISCO-INERTIAL PLOT
VALUES',//11X,'Y=(PEW/CC1)*(PM+B)',6X,'X=(CC2/CC1)*
(1+B/PM)')
WRITE(6,13) (Y(I),X(I),I=1,NDJ)
WRITE(6,14) EK,BETA,SIGMAA,SIGBET,VAR,RAA,B
14 FORMAT(/6X,'KCOR=',G12.5,5X,'BETA=',G12.5,/6X,
$'SIGMAK=',G12.5,2X,'SIGMABETA=',E12.5,/6X,
$'VARIANCE=',G12.5,1X,'LIN.CORR.=',G12.5,/6X,'B=',,
$G12.5)
13 FORMAT(6X,2G18.5)
C DO 102 I=1,NDJ
C DX(I)=X(I)
C DY(I)=Y(I)
C 102 CONTINUE
C
DO 114 I=1,NDJ
EY(I)=CP(I)
EX(I)=(1.70)*(PW(I)*RW*RW-PE(I)*RE*RE)/(RE*RE-RW*RW)
V(I)=EY(I)/EX(I)
114 CONTINUE
WRITE(6,115)
115 FORMAT(/8X,'EFFECT OF CP ON HORIZONTAL STRESSES',

```

```

$//11X,'Y=CP APPLIED',6X,'X=HOR. STRESSES',6X,'V=RATIO')
      WRITE(6,33) (EY(I),EX(I),V(I),I=1,NDJ)
  33 FORMAT(6X,3G18.5)

C
      DO 202 I=1,NDJ
      FX(I)=0.25*EX(I)
      FY(I)=BY(I)
  202 CONTINUE
      WRITE(6,113)
  113 FORMAT(/8X,'EVALUATING CP AND Kabs PLOT VALUES',
$//14X,'Y=Kabs',8X,'X=GENERATED CP')
      WRITE(6,116) (FY(I),FX(I),I=1,NDJ)
  116 FORMAT(6X,2G18.5)

C
C      WRITE(6,38)
C      WRITE(6,36)
  36 FORMAT(/14X,'PE',12X,'PW',12X,'RD',10X,'VIS',12X,
$'T',13X,'ZA',13X,'QO',/13X,'(KPA)',9X,'(KPA)',19X,
$'(MICRO-PA.S)',5X,'(DEG K)',20X,'(CU M/D)')/
      WRITE(6,37)(PE(I),PW(I),RD(I),VIS(I),W(I),ZA(I),QO(I),
$,I=1,NDJ)
C      WRITE(6,38)
  37 FORMAT(6X,7E14.5)

C
      CALL GRAPH(AX,AY,BX,BY,CX,CY,DX,DY,EX,EY,FX,FY,NDJ,
$IFLAG,DIRECT)
C
  38 FORMAT(///)
      STOP
      END

C
C
C
C
      SUBROUTINE GRAPH(AX,AY,BX,BY,CX,CY,DX,DY,EX,EY,FX,FY,
$NDJ,IFLAG,DIRECT)
      IMPLICIT REAL*8(A-H,O-Z)
      REAL*4 X,Y,ALPHA,XLEG,YLEG,CPHOR1,CPHOR2,DIRECT,CPTYPE
      DIMENSION X(60),Y(60),ALPHA(20),AX(60),AY(60),BX(60),
$BY(60),CX(60),CY(60),DX(60),DY(60),EX(60),EY(60),
$FX(60),FY(60),ZX(60),ZY(60),CPTIT(2),DIRECT(5),
$XLEG(7),YLEG(7),CPHOR1(12),CPHOR2(12),TYPE(2),
$CPTYPE(5)
      XSIZE=8.0
      YSIZE=4.75

C
      READ(5,4) CPTIT
  91 READ(5,4) TYPE
      READ(5,31) CPTYPE
      WRITE(6,4) TYPE
      WRITE(6,31) CPTYPE
  31 FORMAT(5A4)

C
      X0=-1.5

```

```

Y0=-2.25
X1=X0+11.0
Y1=Y0+8.5
C
  ICOUNT=1
  KA=1
  KB=1
  KC=1
  ND=NDJ
  DO 333 I=1,NDJ
    X(I)=AX(I)
    Y(I)=AY(I)
  333 CONTINUE
C
  40 READ(7,50) HA,HB,VA,VB,PAGE
  50 FORMAT(4E20.4,A4)
    READ(7,1) ALPHA
    1 FORMAT(20A4)
    IF(ICOUNT-4) 21,81,21
  81 READ(7,2) CPHOR1
    READ(7,2) CPHOR2
    READ(7,3) XLEG
    READ(7,3) YLEG
    2 FORMAT(12A4)
    3 FORMAT(7A4)
    4 FORMAT(2A8)
C
  21 NF=1
    CALL CGPEP1(7.0,7.0)
    CALL CGPL(X,Y,X,ND,NF,KA,KB,KC,0,HA,HB,8.0,VA,VB,
$4.75,ALPHA,+6)
    CALL CGPEP5(-7.0,-0.35,DIRECT,8.0,0.15,0.0)
C
    CALL PLOT(8.0,4.75,3)
    CALL PLOT(8.0,4.5,2)
    CALL PLOT(0.0,0.0,3)
C
    CALL PSYM(1.1,3.65,0.15,TYPE,0.0,16)
    CALL PSYM(1.1,3.90,0.15,CPT1T,0.0,16)
    CALL PSYM(1.1,4.15,0.15,DIRECT,0.0,16)
    CALL PSYM(1.1,4.40,0.15,CPTYPE,0.0,16)
    IF(ICOUNT-4) 82,83,82
  83 CALL PSYM(2.75,-0.5,0.15,XLEG,0.0,28)
    CALL PSYM(-0.5,0.5,0.15,YLEG,90.0,28)
    CALL PSYM(2.,-0.9,0.15,CPHOR1,0.0,48)
    CALL PSYM(0.5,-1.2,0.15,CPHOR2,0.0,48)
  82 CALL BX(X0,Y0,X1,Y1,PAGE)
    IF(ICOUNT-4) 58,60,57
  58 IF(ICOUNT-3) 58,59,60
  56 IF(ICOUNT-2) 53,54,59
C
  53 DO 103 I=1,NDJ
    X(I)=BX(I)
    Y(I)=BY(I)

```

```

103 CONTINUE
 41 ICOUNT=ICOUNT+1
    GO TO 40
C
 54 DO 104 I=1,NDJ
    X(I)=CX(I)
    Y(I)=CY(I)
104 CONTINUE
  KA=2
  KB=2
    GO TO 41
C
C 55 DO 105 I=1,NDJ
C    X(I)=DX(I)
C    Y(I)=DY(I)
C 105 CONTINUE
C    KA=1
C    KB=1
C    GO TO 41
C
 59 DO 115 I=1,NDJ
    X(I)=EX(I)
    Y(I)=EY(I)/EX(I)
115 CONTINUE
  KA=1
  KB=1
    GO TO 41
C
 60 DO 116 I=1,NDJ
    X(I)=FX(I)
    Y(I)=FY(I)
116 CONTINUE
  KA=1
  KB=1
    GO TO 41
C
 57 NF=0
    CALL CGPL(X,Y,X,ND,NF,KA,KB,KC,0,HA,HB,8.0,VA,VB,
    &4.75,ALPHA,+6)
    RETURN
    END
.C
C
C
    SUBROUTINE BOX(X0,Y0,X1,Y1,PAGE)
    CALL PLOT(X0,Y0,3)
    CALL PLOT(X0,Y1,2)
    CALL PLOT(X1,Y1,2)
    CALL PLOT(X1,Y0,2)
    CALL PLOT(X0,Y0,2)
    CALL PSYM(8.9,-0.85,0.15,PAGE,270.0,4)
20 RETURN
    END
C

```

C  
C

SUBROUTINE LINXY(X,Y,NDJ,A,BETA,SIGMAA,SIGBET,RAA,VAR)  
IMPLICIT REAL\*8(A-H,O-Z)

DIMENSION X(160),Y(160)

SUM=0.0

SUMX=0.0

SUMY=0.0

SUMX2=0.0

SUMXY=0.0

SUMY2=0.0

C

DO 50 I=1,NDJ

WET=1.0

SUM=SUM+WET

SUMX=SUMX+WET\*X(I)

SUMY=SUMY+WET\*Y(I)

SUMX2=SUMX2+WET\*X(I)\*X(I)

SUMXY=SUMXY+WET\*X(I)\*Y(I)

SUMY2=SUMY2+WET\*Y(I)\*Y(I)

50 CONTINUE

C

DELTA=SUM\*SUMX2-SUMX\*SUMX

A=(SUMX2\*SUMY-SUMX\*SUMXY)/DELTA

BETA=(SUMXY\*SUM-SUMX\*SUMY)/DELTA

CJ=NDJ-2

S1=SUMY2+A\*A\*SUM+BETA\*BETA\*SUMX2

S2=A\*SUMY+BETA\*SUMXY-A\*BETA\*SUMX

VAR=(S1-2\*S2)/CJ

SIGMAA=DSQRT(VAR\*SUMX2/DELTA)

SIGBET=DSQRT(VAR\*SUM/DELTA)

DEN=DSQRT(DELTA\*(SUM\*SUMY2-SUMY\*SUMY))

RAA=(SUM\*SUMXY-SUMX\*SUMY)/DEN

RETURN

END

C

C

C

SUBROUTINE LAGINT(P,T,TR,PR,Z,DR,VS)

C

This subroutine performs Lagrangian Interpolation  
within a set of (TF,ZF) pairs to give the Z value  
corresponding to any TF. The degree of the  
interpolating polynomial is one less than the number  
of points supplied.

TF = Temperatures in deg. Kelvin

T = Desired Temperature in deg. Kelvin

ZF = Compressibility factor, dimensionless

Z = Desired Compressibility factor.

PA = Desired Pressure in Atmospheric Pressure

P = Desired Pressure in kPa

DR = Reduced density

VS = Viscosity in micro-Pascal.seconds

C

```
IMPLICIT REAL*8(A-H,O-Z)
DIMENSION TF(3),ZF(3)
DATA TF/2.90D+02,3.00D+02,3.10D+02/
ZC=0.290
N=3
PA=P/1.01325D+02
ZF(1)=0.99999064D+00-0.26875701D-03*PA+0.20388746D-05
C*PA*PA+0.50842660D-08*PA*PA*PA
ZF(2)=1.00000000D+00-0.18372581D-03*PA+0.21148108D-05
C*PA*PA+0.26213210D-08*PA*PA*PA
ZF(3)=0.99999132D+00-0.10218783D-03*PA+0.18772473D-05
C*PA*PA+0.25452826D-08*PA*PA*PA
Z=0.0
DO 20 I=1,N
TERM=ZF(I)
DO 10 J=1,N
IF(I.EQ.J) GO TO 10
ZF(I)=ZF(I)*(T-TF(J))/(TF(I)-TF(J))
10 CONTINUE
Z=Z+ZF(I)
20 CONTINUE
VS=1.0D+05*(0.1778D-03*(1.+0.8958D-03*(PA-1.))
C+0.612D-06*(PA-1.)*(PA-1.) +0.3997D-07*(PA-1.)*(PA-1.)
C*(PA-1.))+0.455D-06*(T-(273.16+25)))
DR=ZC*PR/(Z*TR)
RETURN
END
```



```

89 FORMAT(4X,'RE=',F10.7,2X,'RW=',F10.7,2X,'H=',F10.7,
        &2X,'POROSITY=',F10.7)
4 FORMAT(F10.6)
5 FORMAT(4X,'R=',F10.6)
   IF(IFLAG.EQ.(100)) GO TO 67
C
      READ(5,6)(PE(I),PW(I),CP(I),PB(I),TCE(I),TCW(I),TS(I),
        &I=1,NDJ)
6 FORMAT(7F11.4)
C
      WRITE(6,7)
7 FORMAT('1',14X,'PE',12X,'PW',12X,'CP',12X,'PB',11X,
        &'TCE',12X,'TCW',12X,'TS',/13X,'(PSI)',9X,'(PSI)',9X,
        &'(PSI)',9X,'(CMHG)',7X,'(MVOLT)',8X,'(MVOLT)',8X,
        &'(SEC)''')
C
      WRITE(6,8)(PE(I),PW(I),CP(I),PB(I),TCE(I),TCW(I),TS(I),
        &I=1,NDJ)
8 FORMAT(6X,7E14.4)
C
      DO 9 I=1,NDJ
      PB(I)=PB(I)*1.33322D+00
      PE(I)=PB(I)+PE(I)*6.894757D+00
      PW(I)=PB(I)+PW(I)*6.894757D+00
      CP(I)=PB(I)+CP(I)*6.894757D+00
      TE(I)=((TCE(I)+2.145588250D-02)/4.046470588D-02)
      &+273.15
      TW(I)=((TCW(I)+2.145588250D-02)/4.046470588D-02)
      &+273.15
      W(I)=(TE(I)+TW(I))/2.
      QO(I)=(6.116438868D+02)/TS(I)
9 CONTINUE
      GO TO 82
C
67 READ(5,86) (PE(I),PW(I),W(I),QO(I),CP(I),I=1,NDJ)
86 FORMAT(5F12.8)
      DO 68 I=1,NDJ
      PE(I)=PE(I)*6.894757D+00
      PW(I)=PW(I)*6.894757D+00
      W(I)=((W(I)-3.2D+01)/1.8D+00)+2.7315D+02
      QO(I)=QO(I)*2.86364D+01
      CP(I)=CP(I)*6.894757D+00
68 CONTINUE
C
82 DO 84 I=1,NDJ
      PM(I)=(PE(I)+PW(I))/2.
      P=PM(I)
      PR=P/PCA
      T=W(I)
      TR=T/TC
C
      CALL LAGINT(P,T,TR,PR,Z,DR,VS)
      ZA(I)=Z
      RD(I)=DR
      VIS(I)=VS
C

```

```

PEW(I)=2.*(PE(I)-PW(I))
CC1(I)=C1*ZA(I)*VIS(I)*T*QO(I)
CC2(I)=C2*ZA(I)*T*QO(I)*QO(I)
CC3(I)=CC2(I)/PM(I)
CC4(I)=C3*ZA(I)*T*QO(I)*QO(I)*QO(I)
CC5(I)=CC4(I)/PM(I)
84 CONTINUE
C   WRITE(6,38)
M=1
ICOUNT=1
C   WRITE(6,43)
43 FORMAT(6X,'Y',13X,'X1',14X,'X2',11X,'X3')
DO 52 I=1,NDJ
Y(I)=(PE(I)*PE(I)-PW(I)*PW(I))/(CC1(I))
X(I,1)=(-PEW(I)/CC1(I))
X(I,2)=CC2(I)/(CC1(I))
52 CONTINUE
C
READ(5,18) EK1,BE1,BETA1,GAMA1
READ(5,19) EKC,BETAC,GAMAC
18 FORMAT(4E20.5)
19 FORMAT(3E20.5)
C
N=2
63 NU=NDJ-N
FNU=NU
C
CALL MFIT(X,Y,NDJ,N,C0,C,DET,YF,XSQR)
C
WRITE(6,151) DET,(C(J),J=1,N)
151 FORMAT(6E20.5)
C
IF(ICOUNT-3) 73,95,95
73 IF(ICOUNT-2) 65,66,95
65 EK=1.0/C0
BE=C(1)
BETA=C(2)
WRITE(6,59) EK,BE,BETA,XSQR
59 FORMAT(/8X,'K=',G12.5,9X,'BE=',G13.5,8X,'BETA=',,
$G12.5,10X,'XSQR=',G12.5)
GO TO 70
C
66 EKA=1.0/C0
BETA=C(1)
GAMA=C(2)
WRITE(6,71) EKA,BETA,GAMA,XSQR
71 FORMAT(/9X,'KABS=',G12.5,6X,'BETA=',G13.5,8X,'GAMA=',,
$G12.5,/6X,'XSQR=',G12.5)
GO TO 70
C
95 EK=1/C0
BE=C(1)
BETA=C(2)
GAMA=C(3)

```

```

      WRITE(6,96) EK,BE,BETA,GAMA,XSQR
96 FORMAT(/11X,'K=',G12.5,18X,'BE=',G12.5,12X,'BETA=',
          G12.5,/8X,'GAMA=',G12.5,14X,'XSQR=',G12.5)
C   70 WRITE(6,54)
54 FORMAT(/16X,'LIN.CORR.COEF')
    DO 58 J=1,N
58 WRITE(6,55) (A(J,JJ),JJ=1,N)
55 FORMAT(4X,5E14.5)
C
C   POR=((1.0665790D+06)*(DABS(EK)**(-1.12))/BETA)**
C   (1./1.67)
      WRITE(6,40) DET,POR
40 FORMAT(12X,'DETERMINANT=',E14.5/4X,'FRACTIONAL
          &POROSITY=',E14.5)
      SSQ=0.0
      DO 69 I=1,NDJ
      ER(I)=(Y(I)-YF(I))
      ERR(I)=(ER(I)/Y(I))
      SSQ=SSQ+(ER(I)*ER(I))
69 CONTINUE
      WRITE(6,41)
41 FORMAT(/9X,'Y OBSERVED',6X,'Y CALC',8X,'REL.ERR')
      WRITE(6,42) (Y(I),YF(I),ERR(I),I=1,NDJ)
42 FORMAT(6X,3E14.5)
C
C   S=(SSQ/(NDJ-N))**0.5
      WRITE(6,85) SSQ,S
85 FORMAT(/4X,'SSQ=',E11.4,9X,'STD.DEV.=',E11.4/)
C
C   IF(ICOUNT-2).LE.60,94,64
C
60 N=2
      DO 62 I=1,NDJ
      Y(I)=(PE(I)*PE(I)-PW(I)*PW(I))/CC1(I)
      X(I,1)=CC2(I)/CC1(I)
      X(I,2)=CC4(I)/CC1(I)
62 CONTINUE
      ICOUNT=ICOUNT+1
      GO TO 63
C
94 N=3
      DO 99 I=1,NDJ
      Y(I)=(PE(I)*PE(I)-PW(I)*PW(I))/CC1(I)
      X(I,1)=-PEW(I)/CC1(I)
      X(I,2)=CC2(I)/CC1(I)
      X(I,3)=CC4(I)/CC1(I)
99 CONTINUE
      ICOUNT=ICOUNT+1
      GO TO 63
C
C   WRITE(6,38)
64 WRITE(6,36)
36 FORMAT(/14X,'PE',12X,'PW',12X,'RD',10X,'VIS',12X,

```

```

C' T', 13X, 'ZA', 13X, 'QO', /13X, '(KPA)', 9X, '(KPA)', 19X,
C' (MICRO-PA.S)', 5X, '(DEG K)', 20X, '(CU M/D)')/
      WRITE(6,37)(PE(I),PW(I),RD(I),VIS(I),W(I),ZA(I),QO(I)
C     ,I=1,NDJ
C     WRITE(6,38)
37 FORMAT(6X,7E14.4)
C     38 FORMAT(///)
      STOP
      END
C
C
C
SUBROUTINE MFIT(X,Y,NDJ,N,C0,C,A,DET,YF,XSQR)
IMPLICIT REAL*8(A-H,O-Z)
REAL*8 DLOG,DABS
DIMENSION A(5,5),E(60),D(60,5),DT(5,60),
C X(60,5),Y(60),YF(60),XMEAN(60),C(5),B(5)
C
      FNDJ=NDJ
      SUM=FNDJ
      YMEAN=0.0
      DO 28 J=1,N
      XMEAN(J)=0.0
28 CONTINUE
C
C ACCUMULATE SUMS
C
      DO 50 I=1,NDJ
      YMEAN=YMEAN+Y(I)
      DO 44 J=1,N
44 XMEAN(J)=XMEAN(J)+X(I,J)
50 CONTINUE
      YMEAN=YMEAN/SUM
      DO 53 J=1,N
53 XMEAN(J)=XMEAN(J)/SUM
C
C ACCUMULATE MATRICES B AND A
C
      DO 67 I=1,NDJ
      E(I)=Y(I)-YMEAN
      DO 67 J=1,N
      D(I,J)=X(I,J)-XMEAN(J)
67 CONTINUE
C
      DO 20 I=1,NDJ
      DO 20 J=1,N
      DT(J,I)=D(I,J)
20 CONTINUE
      DO 22 J=1,N
      DO 22 K=1,N
      A(J,K)=0.0
      B(J)=0.0
      DO 22 I=1,NDJ
      A(J,K)=A(J,K)+DT(J,I)*D(I,K)

```

```

      B(J)=B(J)+DT(J,I)*E(I)
22 CONTINUE
C
C      INVERT SYMMETRIC MATRIX
C
C      CALL MTINV(A,N,DET)
C
C      IF(DET) 101,91,101
C 91 C0=0.0
C      XSQR=0.0
C      GO TO 150
C
C      CALCULATE COEFFICIENTS
C
101 C0=YMEAN
XSQ=0.0
DO 17 I=1,NDJ
17 YF(I)=0.0
DO 108 J=1,N
C(J)=0.0
DO 104 K=1,N
104 C(J)=C(J)+A(J,K)*B(K)
C0=C0-C(J)*XMEAN(J)
DO 108 I=1,NDJ
108 YF(I)=YF(I)+X(I,J)*C(J)
FREE=N-NDJ-N-1
DO 113 I=1,NDJ
YF(I)=YF(I)+C0
113 XSQ=XSQ+(Y(I)-YF(I))*(Y(I)-YF(I))
XSQR=XSQ/FREEN
C 150 RETURN
RETURN
END
C
C
C
C      SUBROUTINE MTINV(A,N,DET)
C      IMPLICIT REAL*8(A-H,O-Z)
C      DIMENSION A(5,5),IK(5),JK(5)
C
C      This subroutine inverts asymmetric matrix and
C      calculates its determinant.
C
C      DET=1.0
DO 91 K=1,N
C
T=0.0
21 DO 30 I=K,N
DO 30 J=K,N
IF(DABS(T)-DABS(A(I,J))) 24,24,30
24 T=A(I,J)
IK(K)=I
JK(K)=J

```

30 CONTINUE

```

C      IF(T)41,32,41
32 DET=0.0
      GO TO 98
41 I=IK(K)
      IF(I-K)21,51,43
43 DO 50 J=1,N
      SAVE=A(K,J)
      A(K,J)=A(I,J)
50 A(I,J)=-SAVE
51 J=JK(K)
      IF(J-K) 21,61,53
53 DO 73 I=1,N
      SAVE=A(I,K)
      A(I,K)=A(I,J)
73 A(I,J)=-SAVE

```

```

C      61 DO 70 I=1,N
      IF(I-K) 63,70,63
63 A(I,K)=-A(I,K)/T
70 CONTINUE
      DO 80 I=1,N
      DO 80 J=1,N
      IF(I-K) 74,80,74
74 IF(J-K) 75,80,75
75 A(I,J)=A(I,J)+A(I,K)*A(K,J)
80 CONTINUE
      DO 90 J=1,N
      IF(J-K) 83,90,83
83 A(K,J)=A(K,J)/T
90 CONTINUE
      A(K,K)=1.0/T
91 DET=DET*T

```

```

C      DO 97 L=1,N
      K=N-L+1
      J=IK(K)
      IF(J-K) 94,94,92
92 DO 93 I=1,N
      SAVE=A(I,K)
      A(I,K)=-A(I,J)
93 A(I,J)=SAVE
94 I=JK(K)
      IF(I-K) 97,97,95
95 DO 96 J=1,N
      SAVE=A(K,J)
      A(K,J)=-A(K,J)
96 A(I,J)=SAVE
97 CONTINUE
98 RETURN
END

```

```

C          POLY
C
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
C      THIS PROGRAM PERFORMS A MULTIPLE LINEAR REGRESSION
C      CURVE FIT USING LEAST SQUARES, TO DETERMINE THE BEST
C      FIT.
C
C      Uses : Appplot++Plotlib on object file
C
C      Subroutines: LAGINT, MFIT, XPLOT, BOX, MTINV
C
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
C      EM=MOLECULAR WEIGHT
C      TR,PR,DR=REDUCED TEMPERATURE,PRESSURE, AND DENSITY
C      RESPECTIVELY
C      VIS=VISCOSITY IN MICRO-PA.S
C      GG=GAS GRAVITY
C      IMPLICIT REAL*8(A-H,O-Z)
REAL *4 DIRECT
REAL *8 DLOG,DABS
DIMENSION PE(60),PW(60),TE(60),TW(60),PM(60),W(60),
$VIS(60),ZA(60),PB(60),CP(60),TS(60),QO(60),PEW(60),
$YF(60),CC1(60),CC2(60),CC3(60),XMEAN(60),X(60,5),
$TC(60),TCW(60),E(60),A(5,5),B(5),ER(60),HY(60),
$ERR(60),DP(60),CC4(60),CC5(60),AX(60),AY(60),BY(60),
$CX(60),DY(60),EY(60),GY(60),FRC(60),RET(60),FRCM(60),
$FRM(60),SFT(60),SFC(60),DIRECT(5),RD(60),Y(60),C(5)
DATA TC,EM,PC,SIG/1.262D+02,2.8013D+01,3.35D+01,
$3.798D+00/
GG=EM/2.896D+01
PCA=PC*(1.01325D+02)
READ(5,1) NUM,NO,DIRECT,IFLAG
READ(5,90) RE,RW,H,POR
READ(5,3) NDJ
READ(5,4) R
C
IF(IFLAG.NE.(100)) GO TO 81
RE=RE*3.048D-01
RW=RW*3.048D-01
H=H*3.048D-01
GO TO 81
81 WRITE(6,2) NUM,NO,DIRECT
WRITE(6,89) RE,RW,H,POR
C
WRITE(6,5) R
C1=((1.2954909D-03)*DLOG(RE/RW))/H
C2=((2.9228842D-15)*GG*(1/RW-1/RE))/H*H
C3=(6.5946061D-21)*GG*GG*(1/(RW*RW)-1/(RE*RE))/H*H*H
1 FORMAT(A4,A4,5A4,I3)
90 FORMAT(4F10.7)
3 FORMAT(I3)
2 FORMAT(/4X,'CORE SAMPLE NO ',A4,2X,'RUN ',A4,2X,5A4)
89 FORMAT(4X,'RE=',F10.7,2X,'RW=',F10.7,2X,'H=',F10.7,

```

```

        F2X,'POROSITY=',F10.7)
4 FORMAT(F10.6)
5 FORMAT(4X,'R=',F10.6)
IF(IFLAG.EQ.(100)) GO TO 67
C
      READ(5,6)(PE(I),PW(I),CP(I),PB(I),TCE(I),TCW(I),TS(I)
      ,I=1,NDJ)
6 FORMAT(7F11.4)
C
      WRITE(6,7)
7 FORMAT('1',14X,'PE',12X,'PW',12X,'CP',12X,'PB',11X,
      'TCE',12X,'TCW',12X,'TS',/13X,'(PSI)',9X,'(PSI)',9X,
      '(PSI)',9X,(CMHG)',7X,'(MVOLT)',8X,'(MVOLT)',8X,
      '(SEC)''')
C
      WRITE(6,8)(PE(I),PW(I),CP(I),PB(I),TCE(I),TCW(I),TS(I)
      ,I=1,NDJ)
8 FORMAT(6X,7E14.4)
C
      DO 9 I=1,NDJ
      PB(I)=PB(I)*1.3332D+00
      PE(I)=PB(I)+PE(I)*6.894757D+00
      PW(I)=PB(I)+PW(I)*6.894757D+00
      CP(I)=PB(I)+CP(I)*6.894757D+00
      TE(I)=((TCE(I)+2.145588250D-02)/4.046470588D-02)
      +273.15
      TW(I)=((TCW(I)+2.145588250D-02)/4.046470588D-02)
      +273.15
      W(I)=(TE(I)+TW(I))/2.
      QO(I)=(6.116438868D+02)/TS(I)
9 CONTINUE
GO TO 82
C
67 READ(5,86) (PE(I),PW(I),W(I),QO(I),CP(I),I=1,NDJ)
86 FORMAT(5F12.8)
DO 68 I=1,NDJ
PE(I)=PE(I)*6.894757D+00
PW(I)=PW(I)*6.894757D+00
W(I)=((W(I)-3.2D+01)/1.8D+00)+2.7315D+02
QO(I)=QO(I)*2.86364D+01
CP(I)=CP(I)*6.894757D+00
68 CONTINUE
C
82,DO 84 I=1,NDJ
PM(I)=(PE(I)+PW(I))/2.
P=PM(I)
PR=P/PCA
T=W(I)
TR=T/TC
C
CALL LAGINT(P,T,TR,PR,Z,DR,VS)
ZA(I)=Z
RD(I)=DR
VIS(I)=VS
C
PEW(I)=2.*(PE(I)-PW(I))

```

```

CC1(I)=C1*ZA(I)*VIS(I)*T*QO(I)
CC2(I)=C2*ZA(I)*T*QO(I)*QO(I)
CC4(I)=CC2(I)/PM(I)
CC3(I)=C3*ZA(I)*T*QO(I)*QO(I)*QO(I)
CC5(I)=CC3(I)/PM(I)
84 CONTINUE
C   WRITE(6,38)
      READ(5,18) EK1,BE1,BETA1,GAMA1
      READ(5,19) EKC,BETAC,GAMAC
18 FORMAT(4E20.5)
19 FORMAT(3E20.5)
ICOUNT=1
C
N=3
CO=0.0
DO 52 I=1,NDJ
Y(I)=(PE(I)*PE(I)-PW(I)*PW(I))
X(I,1)=QO(I)
X(I,2)=QO(I)*QO(I)
X(I,3)=QO(I)*QO(I)*QO(I)
12 FORMAT(4G14.5)
52 CONTINUE
63 NU=NDJ-N
FNU=NU
C
CALL MFIT(X,Y,NDJ,N,CO,C,A,DET,YF,XSQR)
C
C   WRITE(6,151) DET,(C(J),J=1,N)
C 151 FORMAT(6E20.5)
C
IF(ICOUNT-2) 66,97,105
C
66 DO 15 I=1,NDJ
CX(I)=QO(I)
DY(I)=Y(I)
EY(I)=YF(I)
15 CONTINUE
WRITE(6,111)
111 FORMAT(/4X,'Back Pressure Cubic Fit')
WRITE(6,59) C(1),C(2),C(3),XSQR,DET
59 FORMAT(/8X,'C(1)=' ,G12.5,9X,'C(2)=' ,G13.5,8X,'C(3)=' ,
     ,G12.5,10X,'XSQR=' ,G12.5,4X,'DETERMINANT=' ,G12.5)
GO TO 70
C
97 EK=1.0/CO
BETA=C(1)
GAMA=C(2)
*   DO 10 I=1,NDJ
AX(I)=X(I,1)
AY(I)=Y(I)
BY(I)=YF(I)
10 CONTINUE
WRITE(6,60)
60 FORMAT(/4X,'Visco-Inertial Cubic Fit')

```

```

C      WRITE(6,71) EK,BETA,GAMA,XSQR,DET
71 FORMAT(/9X,'KABS=',G12.5,6X,'BETA=',G13.5,8X,'GAMA=',
  $ G12.5,/6X,'XSQR=',G12.5,4X,'DETERMINANT=',G12.5)
C      GO TO 70
C
105 EK=1.0/C0
      BETA=C(1)
      DO 106 I=1,NDJ
      HY(I)=YF(I)
106 CONTINUE
      WRITE(6,104)
104 FORMAT(/4X,'Visco-Inertial Quadratic Fit')
      WRITE(6,107) EK,BETA,XSQR,DET
107 FORMAT(/9X,'KABS=',G12.5,6X,'BETA=',G13.5,8X,'XSQR=',
  $ G12.5,4X,'DETERMINANT=',G12.5)
C
70 WRITE(6,54)
54 FORMAT(/16X,'LIN.CORR.COEF')
      DO 58 J=1,N
58 WRITE(6,55) (A(J,JJ),JJ=1,N)
55 FORMAT(4X,5E14.5)
C
      SSQ=0.0
      DO 69 I=1,NDJ
      ER(I)=(Y(I)-YF(I))
      ERR(I)=(ER(I)/Y(I))
      SSQ=SSQ+(ER(I)*ER(I))
C
C     EVALUATION OF REYNOLDS NUMBERS AND FRICTION FACTORS
C
C     EK1, BETA1, GAMA1 are parameters obtained with the
C     truncated plot points.
C     EKC, BETAC, GAMAC are parameters obtained with the
C     complete plot points.
C
      REC(I)=(CC2(I)*EKC*BETAC)/CC1(I)
      RET(I)=(CC2(I)*EK1*BETA1)/CC1(I)
      SFC(I)=(CC3(I)*GAMAC)/(CC2(I)*BETAC)
      SFT(I)=(CC3(I)*GAMA1)/(CC2(I)*BETA1)
      FRC(I)=64.0*(PE(I)*PE(I)-PW(I)*PW(I))/(CC2(I)*BETAC)
      FRT(I)=64.0*(PE(I)*PE(I)-PW(I)*PW(I))/(CC2(I)*BETA1)
      FRCM(I)=64.0(1.0/REC(I)+1.0*SFC(I))
      FRTM(I)=64.0(1.0/RET(I)+1.0+SFT(I))
69 CONTINUE
C      WRITE(6,46)
46 FORMAT(/6X,'PARAMETERS FOR REYNOLDS NO AND FRICTION
  $ FACTOR')
C      WRITE(6,44) EK1,BE1,BETA1
44 FORMAT(4X,'K=',E11.4,5X,'B=',E11.4,5X,'BETA=',E11.4)
C      WRITE(6,41)
41 FORMAT(/9X,'X(I,1)',6X,'Y OBSERVED',6X,'Y CALC',8X,
  $ 'REL.ERR',9X,'REY',6X,'FR.FACTOR')
C      WRITE(6,42) (X(I,1),Y(I),YF(I),ERR(I),REB(I),FRB(I),
  $ ,I=1,NDJ)

```

```

C      42 FORMAT(6X,6E14.5)
C      S=(SSQ/(NDJ-N))**0.5
C      WRITE(6,85) SSQ,S
C      85 FORMAT(/4X,'SSQ=',E11.4,9X,'STD.DEV.=',E11.4/)
C      IF(ICOUNT-2) 64,102,23
C
C      64 N=2
C      DO 62 I=1,NDJ
C      Y(I)=(PE(I)*PE(I)-PW(I)*PW(I))/CC1(I)
C      X(I,1)=CC2(I)/CC1(I)
C      X(I,2)=CC3(I)/CC1(I)
C      62 CONTINUE
C      ICOUNT=ICOUNT+1
C      GO TO 63
C
C      102 N=1
C      DO 103 I=1,NDJ
C      Y(I)=(PE(I)*PE(I)-PW(I)*PW(I))/CC1(I)
C      X(I,1)=CC2(I)/CC1(I)
C      103 CONTINUE
C      ICOUNT=ICOUNT+1
C      GO TO 63
C
C      23 WRITE(6,414)
C      414 FORMAT(/2X,'REC',3X,'FRT',5X,'RET',7X,'FRT',6X,
C      & 'FRCM',6X,'FRTM',4X,'SFC',4X,'SFT')
C      WRITE(6,424)(REC(I),FRC(I),RET(I),FRT(I),FRCM(I),
C      & FRTM(I),SFC(I),SFT(I),I=1,NDJ)
C      424 FORMAT(F6.3,F7.2,F8.3,3F10.2,2F7.3)
C
C      ' WRITE(6,38)
C      WRITE(6,36)
C      36 FORMAT(/14X,'PE',12X,'PW',12X,'RD',10X,'VIS',12X,
C      & 'T',13X,'ZA',13X,'QO',/13X,'(KPA)',9X,'(KPA)',19X,
C      & '(MICRO-PA.S)',5X,'(DEG K)',20X,'(CU M/D)''')
C      WRITE(6,37)(PE(I),PW(I),RD(I),VIS(I),W(I),ZA(I),QO(I)
C      &,I=1,NDJ)
C      WRITE(6,38)
C      37 FORMAT(6X,7E14.4)
C      38 FORMAT(///)
C
C      CALL XPLOT(AX,AY,BY,CX,DY,EY,GY,HY,REC,FRC,RET,FRT,
C      & NDJ,IFLAG,DIRECT,FRCM,FRTM)
C
C      STOP
C      END
C
C
C      SUBROUTINE XPLOT(AX,AY,BY,CX,DY,EY,GY,HY,REC,FRC,RET,
C      & FRT,NDJ,IFLAG,DIRECT,FRCM,FRTM)
C      IMPLICIT REAL*8(A-H,O-Z)

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REAL*4 X,Y,ALPHA,DIRECT,CPTYPE
DIMENSION AX(60),BY(60),AY(60),ALPHA(20),X(60),Y(60),
CX(60),DY(60),EY(60),GY(60),REB(60),FRB(60),HY(60),
CRET(60),FRT(60),REG(60),FRG(60),CPTIT(2),DIRECT(5),
CTYPE(2),CPTYPE(5)
REAL *8 TITLE(9) // 'EXPTAL ', 'CUBIC FN', 'QUAD. FN',
'EXPTAL ', 'CUBIC FN', 'EXPTAL ', 'EQN. 8.7',
'EXPTAL ', 'EQN. 8.7'
XSIZEx=8.0
YSIZE=4.75
ND=NDJ
KA=1
KB=1
KC=1
READ(5,4) CPTIT
WRITE(6,4) CPTIT
READ(5,4) TYPE
WRITE(6,4) TYPE
READ(5,31) CPTYPE
WRITE(6,31) CPTYPE
4 FORMAT(2A8)
31 FORMAT(5A4)
555 FORMAT(I3)
C
80 X0=-1.5
Y0=-2.25
X1=X0+11.0
Y1=Y0+8.5
C
DO 19 I=1,NDJ
X(I)=AX(I)
19 Y(I)=AY(I)
N=1
67 READ(7,21) HA,HB,VA,VB,PAGE
21 FORMAT(4E20.4,A4)
READ(7,20) ALPHA
20 FORMAT(20A4)
NF=1
CALL CGPEP1(7.0,7.0)
17 CALL CGPL(X,Y,X,ND,NF,KA,KB,KC,0,HA,HB,8.0,VA,VB,
4.75,ALPHA,+6)
IF(N.EQ.6.OR.N.EQ.7.OR.N.EQ.8.OR.N.EQ.9) GO TO 121
GO TO 122
122 CALL CGPEP5(-2.00,-2.00,TITLE(N),8,0.15,0.0)
GO TO 177
121 CALL CGPEP5(-7.50,-2.00,TITLE(N),8,0.15,0.0)
C
177 CALL PLOT(8.0,4.75,3)
CALL PLOT(8.0,4.5,2)
CALL PLOT(0.0,0.0,3)
C
IF(N.EQ.3.OR.N.EQ.5.OR.N.EQ.7.OR.N.EQ.9) GO TO 98
GO TO 97
98 CALL PSYM(5.70,0.95,0.15,CPTIT,0.0,16)

```

```

CALL PSYM(5.70,0.70,0.15,TYPE,0.0,16)
CALL PSYM(5.20,0.45,0.15,DIRECT,0.0,20)
CALL PSYM(5.20,0.25,0.15,CPTYPE,0.0,20)
CALL BOX(X0,Y0,X1,Y1,PAGE)

C
97 IF(N-8) 45,139,6
45 IF(N-7) 42,115,139
42 IF(N-6) 41,8,115
41 IF(N-5) 411,5,8
411 IF(N-4) 23,9,5
23 IF(N-3) 22,10,9
22 IF(N-2) 12,11,10

C
12 NF=3
KC=1
DO 18 I=1,NDJ
X(I)=AX(I)
18 Y(I)=BY(I)
15 N=N+1
IF(N.EQ.4) GO TO 67
IF(N.EQ.6) GO TO 67
IF(N.EQ.8) GO TO 67
GO TO 17

C
11 NF=130
KC=1
DO 30 I=1,NDJ
X(I)=AX(I)
30 Y(I)=HY(I)
GO TO 15

C
10 KA=2
KB=2
DO 16 I=1,NDJ
X(I)=CX(I)
16 Y(I)=DY(I)
GO TO 15

C
9 DO 14 I=1,NDJ
X(I)=CX(I)
14 Y(I)=DABS(EY(I))
NF=3
KC=1
GO TO 15

C
5 DO 44 I=1,NDJ
X(I)=RET(I)
44 Y(I)=FRT(I)
GO TO 15

C
8 DO 13 I=1,NDJ
X(I)=RET(I)
13 Y(I)=FRM(I)
NF=15

```

```
KC=5
GO TO 15
C
115 DO 144 1,1,NDJ
      X(I)=REC(I)
144 Y(I)=FRC(I)
      GO TO 15
C
139 DO 138 I=1,NDJ
      X(I)=REC(I)
138 Y(I)=FRCM(I)
      NF=3
      KC=5
      GO TO 15
C
6 NF=0
      CALL CGPL(X,Y,X,ND,NF,KA,KB,KC,0,HA,HB,8.0,VA,VB,
      $4.75,ALPHA,+6)
      RETURN
      END
```

D.2 EXPERIMENTAL DATA LISTINGS

CORE SAMPLE NO 1C RUN 2 FORWARD DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 12.000      | 3.2000      | 3.00        | 69.350       | 0.8650      | 0.8660      | 139.82      |
| 16.000      | 5.3000      | 5.00        | 69.350       | 0.8690      | 0.8700      | 107.90      |
| 20.000      | 9.1000      | 9.00        | 69.350       | 0.8760      | 0.8760      | 98.050      |
| 26.000      | 12.400      | 12.50       | 69.350       | 0.8870      | 0.8850      | 72.150      |
| 40.400      | 25.000      | 25.00       | 69.350       | 0.8900      | 0.8910      | 51.400      |
| 60.400      | 45.100      | 45.00       | 69.350       | 0.9000      | 0.8970      | 40.470      |
| 80.000      | 64.700      | 64.00       | 69.350       | 0.8990      | 0.9010      | 36.130      |
| 98.000      | 83.000      | 83.00       | 69.350       | 0.9000      | 0.8980      | 34.800      |
| 132.40      | 115.20      | 115.00      | 69.330       | 0.9060      | 0.9040      | 30.000      |
| 160.00      | 143.20      | 143.00      | 69.320       | 0.9090      | 0.9070      | 29.600      |
| 194.40      | 176.80      | 176.00      | 69.320       | 0.9140      | 0.9120      | 28.300      |
| 240.00      | 222.40      | 222.00      | 69.320       | 0.9150      | 0.9110      | 27.560      |
| 280.40      | 262.80      | 263.00      | 69.290       | 0.9090      | 0.9100      | 26.210      |
| 310.40      | 293.20      | 293.00      | 69.290       | 0.9200      | 0.9200      | 25.780      |
| 344.00      | 325.20      | 325.00      | 69.290       | 0.9230      | 0.9220      | 24.470      |
| 376.40      | 356.80      | 356.00      | 69.290       | 0.9250      | 0.9200      | 23.140      |
| 404.00      | 380.00      | 380.00      | 69.290       | 0.9250      | 0.9230      | 20.890      |
| 490.00      | 462.00      | 462.00      | 69.290       | 0.9290      | 0.9300      | 18.000      |
| 572.00      | 544.00      | 544.00      | 69.290       | 0.9230      | 0.9200      | 16.560      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 175.20      | 114.52      | 0.005 | 17.646         | 295.07       | 0.9997 | 4.375        |
| 202.77      | 129.00      | 0.006 | 17.654         | 295.17       | 0.9996 | 5.669        |
| 230.35      | 155.20      | 0.007 | 17.666         | 295.33       | 0.9996 | 6.238        |
| 271.72      | 177.95      | 0.008 | 17.682         | 295.58       | 0.9995 | 8.477        |
| 371.01      | 264.83      | 0.012 | 17.702         | 295.69       | 0.9993 | 11.90        |
| 508.90      | 403.41      | 0.017 | 17.732         | 295.88       | 0.9991 | 15.11        |
| 644.04      | 538.55      | 0.022 | 17.756         | 295.92       | 0.9988 | 16.93        |
| 768.14      | 664.72      | 0.026 | 17.774         | 295.90       | 0.9986 | 17.58        |
| 1005.3      | 886.71      | 0.035 | 17.818         | 296.05       | 0.9982 | 20.39        |
| 1195.6      | 1079.7      | 0.042 | 17.852         | 296.12       | 0.9978 | 20.66        |
| 1432.8      | 1311.4      | 0.050 | 17.896         | 296.24       | 0.9975 | 21.61        |
| 1747.2      | 1625.8      | 0.062 | 17.947         | 296.24       | 0.9970 | 22.19        |
| 2025.7      | 1904.3      | 0.072 | 17.990         | 296.16       | 0.9966 | 23.34        |
| 2232.5      | 2113.9      | 0.079 | 18.037         | 296.42       | 0.9964 | 23.73        |
| 2464.2      | 2334.6      | 0.088 | 18.079         | 296.48       | 0.9962 | 25.00        |
| 2687.6      | 2552.4      | 0.096 | 18.117         | 296.48       | 0.9960 | 26.43        |
| 2877.9      | 2712.4      | 0.102 | 18.150         | 296.51       | 0.9958 | 29.28        |
| 3470.8      | 3277.8      | 0.123 | 18.261         | 296.65       | 0.9954 | 33.98        |
| 4036.2      | 3843.1      | 0.144 | 18.360         | 296.45       | 0.9951 | 36.94        |

CORE SAMPLE NO 1C RUN 2 REVERSE DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.01620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 525.00      | 500.60      | 500.00      | 69.300       | 0.9210      | 0.9250      | 18.110      |
| 470.00      | 448.50      | 446.00      | 69.300       | 0.9230      | 0.9230      | 21.450      |
| 406.00      | 386.10      | 383.00      | 69.300       | 0.9170      | 0.9200      | 25.730      |
| 312.00      | 291.40      | 289.00      | 69.300       | 0.9190      | 0.9240      | 30.000      |
| 228.00      | 208.60      | 204.00      | 69.300       | 0.9190      | 0.9220      | 40.420      |
| 139.20      | 122.80      | 115.00      | 69.300       | 0.9200      | 0.9240      | 65.100      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 3712.1      | 3543.9      | 0.133 | 18.302         | 296.49       | 0.9952 | 33.77        |
| 3332.9      | 3184.7      | 0.119 | 18.232         | 296.49       | 0.9954 | 28.51        |
| 2891.7      | 2754.5      | 0.103 | 18.148         | 296.38       | 0.9958 | 23.77        |
| 2243.6      | 2101.5      | 0.079 | 18.039         | 296.45       | 0.9964 | 20.39        |
| 1664.4      | 1530.6      | 0.058 | 17.941         | 296.43       | 0.9972 | 15.13        |
| 1052.1      | 939.07      | 0.036 | 17.845         | 296.47       | 0.9981 | 9.395        |

CORE SAMPLE NO 1C RUN 3 FORWARD DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 8.0000      | 2.3000      | 400.00      | 70.280       | 0.8540      | 0.8540      | 167.70      |
| 12.000      | 4.5000      | 400.00      | 70.280       | 0.8550      | 0.8600      | 115.10      |
| 16.000      | 7.6000      | 400.00      | 70.280       | 0.8570      | 0.8600      | 93.150      |
| 20.000      | 11.400      | 400.00      | 70.280       | 0.8570      | 0.8650      | 82.300      |
| 27.800      | 19.200      | 400.00      | 70.280       | 0.8560      | 0.8640      | 69.300      |
| 36.000      | 27.900      | 400.00      | 70.280       | 0.8550      | 0.8640      | 62.600      |
| 44.000      | 35.900      | 400.00      | 70.280       | 0.8540      | 0.8630      | 56.200      |
| 54.000      | 45.600      | 400.00      | 70.280       | 0.8510      | 0.8640      | 49.500      |
| 66.400      | 56.900      | 400.00      | 70.400       | 0.8410      | 0.8540      | 40.500      |
| 84.000      | 73.900      | 400.00      | 70.400       | 0.8400      | 0.8510      | 36.810      |
| 96.800      | 86.000      | 400.00      | 70.400       | 0.8400      | 0.8560      | 32.700      |
| 108.00      | 96.600      | 400.00      | 70.400       | 0.8400      | 0.8590      | 30.280      |
| 120.40      | 107.60      | 400.00      | 70.400       | 0.8380      | 0.8570      | 27.500      |
| 136.00      | 123.00      | 400.00      | 70.400       | 0.8350      | 0.8550      | 25.700      |
| 150.40      | 136.00      | 400.00      | 70.400       | 0.8370      | 0.8550      | 24.220      |
| 169.40      | 154.20      | 400.00      | 70.400       | 0.8380      | 0.8550      | 23.200      |
| 196.40      | 180.70      | 400.00      | 70.400       | 0.8350      | 0.8560      | 22.300      |
| 219.20      | 200.40      | 400.00      | 70.400       | 0.8350      | 0.8550      | 21.100      |
| 248.00      | 228.00      | 400.00      | 70.400       | 0.8350      | 0.8550      | 19.900      |
| 278.00      | 257.00      | 400.00      | 70.400       | 0.8350      | 0.8550      | 19.080      |
| 316.80      | 296.80      | 400.00      | 70.540       | 0.8450      | 0.8550      | 18.850      |
| 353.20      | 331.20      | 400.00      | 70.540       | 0.8400      | 0.8570      | 17.640      |
| 382.80      | 358.80      | 400.00      | 70.540       | 0.8400      | 0.8580      | 16.290      |
| 438.00      | 416.00      | 416.00      | 70.540       | 0.8360      | 0.8560      | 15.900      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 148.86      | 109.56      | 0.005 | 17.631         | 294.78       | 0.9997 | 3.647        |
| 176.44      | 124.73      | 0.006 | 17.638         | 294.87       | 0.9997 | 5.314        |
| 204.01      | 146.10      | 0.006 | 17.643         | 294.90       | 0.9996 | 6.566        |
| 231.59      | 172.30      | 0.007 | 17.650         | 294.96       | 0.9996 | 7.432        |
| 285.37      | 226.08      | 0.009 | 17.657         | 294.93       | 0.9994 | 8.826        |
| 341.91      | 286.06      | 0.011 | 17.666         | 294.92       | 0.9993 | 9.771        |
| 397.07      | 341.22      | 0.014 | 17.674         | 294.90       | 0.9992 | 10.88        |
| 466.02      | 408.10      | 0.016 | 17.683         | 294.87       | 0.9991 | 12.36        |
| 551.67      | 486.17      | 0.019 | 17.685         | 294.62       | 0.9989 | 15.10        |
| 673.02      | 603.38      | 0.023 | 17.702         | 294.57       | 0.9986 | 16.62        |
| 761.27      | 686.81      | 0.027 | 17.718         | 294.64       | 0.9985 | 18.70        |
| 838.49      | 759.89      | 0.029 | 17.732         | 294.67       | 0.9983 | 20.20        |
| 923.99      | 835.73      | 0.032 | 17.742         | 294.62       | 0.9982 | 22.24        |
| 1031.5      | 941.91      | 0.036 | 17.757         | 294.56       | 0.9980 | 23.80        |
| 1130.8      | 1031.5      | 0.040 | 17.773         | 294.59       | 0.9978 | 25.25        |
| 1261.8      | 1157.0      | 0.044 | 17.794         | 294.60       | 0.9976 | 26.36        |
| 1448.0      | 1339.7      | 0.051 | 17.823         | 294.57       | 0.9973 | 27.43        |
| 1605.2      | 1475.6      | 0.057 | 17.847         | 294.56       | 0.9970 | 28.99        |
| 1803.8      | 1665.9      | 0.064 | 17.879         | 294.56       | 0.9967 | 30.74        |
| 2010.6      | 1865.8      | 0.071 | 17.913         | 294.56       | 0.9964 | 32.06        |
| 2278.3      | 2140.4      | 0.081 | 17.964         | 294.69       | 0.9961 | 32.45        |
| 2529.3      | 2377.6      | 0.090 | 18.005         | 294.65       | 0.9957 | 34.67        |
| 2733.4      | 2567.9      | 0.097 | 18.040         | 294.66       | 0.9955 | 37.55        |
| 3113.9      | 2962.3      | 0.112 | 18.106         | 294.59       | 0.9951 | 38.47        |

CORE SAMPLE NO 1C RUN 3 REVERSE DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 438.00      | 416.00      | 416.00      | 70.540       | 0.8360      | 0.8560      | 15.900      |
| 412.00      | 391.70      | 400.00      | 70.540       | 0.8340      | 0.8550      | 16.700      |
| 382.00      | 362.50      | 400.00      | 70.720       | 0.7860      | 0.7970      | 17.400      |
| 350.00      | 332.00      | 400.00      | 70.720       | 0.7880      | 0.8010      | 18.300      |
| 280.40      | 263.10      | 400.00      | 70.580       | 0.8330      | 0.8440      | 19.400      |
| 260.00      | 245.10      | 400.00      | 70.540       | 0.8320      | 0.8450      | 20.600      |
| 220.00      | 205.20      | 400.00      | 70.540       | 0.8330      | 0.8450      | 21.200      |
| 180.00      | 166.40      | 400.00      | 70.540       | 0.8300      | 0.8450      | 22.470      |
| 140.00      | 126.60      | 400.00      | 70.560       | 0.8310      | 0.8410      | 25.300      |
| 104.00      | 89.800      | 400.00      | 70.560       | 0.8300      | 0.8450      | 26.200      |
| 76.400      | 61.000      | 400.00      | 70.560       | 0.8340      | 0.8450      | 29.100      |
| 58.000      | 41.000      | 400.00      | 70.560       | 0.8300      | 0.8430      | 32.300      |
| 38.000      | 20.000      | 400.00      | 70.560       | 0.8250      | 0.8400      | 42.000      |
| 20.000      | 6.4000      | 400.00      | 70.560       | 0.8350      | 0.8410      | 78.200      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QC<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 3113.9      | 2962.3      | 0.112 | 18.106         | 294.59       | 0.9951 | 38.47        |
| 2934.7      | 2794.7      | 0.105 | 18.073         | 294.55       | 0.9953 | 36.63        |
| 2728.1      | 2593.6      | 0.098 | 17.977         | 293.24       | 0.9952 | 35.15        |
| 2507.5      | 2383.3      | 0.090 | 17.943         | 293.31       | 0.9955 | 33.42        |
| 2027.4      | 1908.1      | 0.072 | 17.911         | 294.40       | 0.9963 | 31.53        |
| 1886.7      | 1784.0      | 0.067 | 17.888         | 294.40       | 0.9965 | 29.69        |
| 1610.9      | 1508.8      | 0.057 | 17.843         | 294.41       | 0.9970 | 28.85        |
| 1335.1      | 1241.3      | 0.047 | 17.797         | 294.38       | 0.9974 | 27.22        |
| 1059.3      | 966.95      | 0.037 | 17.751         | 294.34       | 0.9979 | 24.18        |
| 811.13      | 713.22      | 0.028 | 17.710         | 294.33       | 0.9984 | 23.35        |
| 620.83      | 514.65      | 0.021 | 17.684         | 294.43       | 0.9988 | 21.02        |
| 493.97      | 376.76      | 0.016 | 17.659         | 294.35       | 0.9990 | 18.94        |
| 356.07      | 231.97      | 0.011 | 17.633         | 294.25       | 0.9993 | 14.56        |
| 231.97      | 138.20      | 0.007 | 17.622         | 294.39       | 0.9996 | 7.822        |

CORE SAMPLE NO 1C RUN 4 FORWARD DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 8.0000      | 2.1000      | 500.00      | 69.680       | 0.9100      | 0.9100      | 179.40      |
| 12.000      | 4.6000      | 500.00      | 69.680       | 0.9100      | 0.9130      | 129.80      |
| 18.000      | 9.4000      | 500.00      | 69.680       | 0.9100      | 0.9150      | 98.200      |
| 24.000      | 15.400      | 500.00      | 69.680       | 0.9120      | 0.9160      | 87.300      |
| 30.000      | 21.400      | 500.00      | 69.680       | 0.9100      | 0.9150      | 77.450      |
| 40.000      | 30.600      | 500.00      | 69.680       | 0.9100      | 0.9150      | 62.000      |
| 60.000      | 51.000      | 500.00      | 69.680       | 0.9100      | 0.9160      | 52.000      |
| 78.000      | 68.900      | 500.00      | 69.680       | 0.9100      | 0.9160      | 46.200      |
| 99.600      | 90.500      | 500.00      | 69.680       | 0.9080      | 0.9180      | 41.250      |
| 128.00      | 118.20      | 500.00      | 69.680       | 0.9080      | 0.9200      | 36.300      |
| 160.00      | 148.50      | 500.00      | 69.680       | 0.9050      | 0.9200      | 31.450      |
| 190.00      | 177.60      | 500.00      | 69.600       | 0.9080      | 0.9170      | 29.300      |
| 229.40      | 217.10      | 500.00      | 69.600       | 0.9080      | 0.9150      | 28.600      |
| 260.40      | 247.40      | 500.00      | 69.600       | 0.9080      | 0.9180      | 27.530      |
| 296.00      | 280.40      | 500.00      | 69.600       | 0.9070      | 0.9200      | 25.100      |
| 334.40      | 319.00      | 500.00      | 69.600       | 0.9070      | 0.9210      | 24.100      |
| 360.20      | 343.20      | 500.00      | 69.600       | 0.9050      | 0.9200      | 22.900      |
| 397.00      | 378.00      | 500.00      | 69.600       | 0.9050      | 0.9200      | 21.500      |
| 445.00      | 423.00      | 500.00      | 69.600       | 0.9040      | 0.9200      | 19.900      |
| 456.00      | 431.00      | 500.00      | 69.600       | 0.9030      | 0.9200      | 18.700      |
| 462.00      | 436.00      | 500.00      | 69.860       | 0.8610      | 0.8670      | 18.100      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 148.06      | 107.38      | 0.005 | 17.694         | 296.17       | 0.9997 | 3.409        |
| 175.64      | 124.61      | 0.005 | 17.699         | 296.21       | 0.9997 | 4.712        |
| 217.00      | 157.71      | 0.007 | 17.706         | 296.23       | 0.9996 | 6.229        |
| 258.37      | 199.08      | 0.008 | 17.714         | 296.27       | 0.9995 | 7.006        |
| 299.74      | 240.45      | 0.010 | 17.719         | 296.23       | 0.9994 | 7.897        |
| 368.69      | 303.88      | 0.012 | 17.729         | 296.23       | 0.9993 | 9.865        |
| 506.58      | 444.53      | 0.017 | 17.752         | 296.24       | 0.9990 | 11.76        |
| 630.69      | 567.95      | 0.022 | 17.772         | 296.26       | 0.9988 | 13.24        |
| 779.62      | 716.87      | 0.027 | 17.795         | 296.24       | 0.9985 | 14.83        |
| 975.43      | 907.86      | 0.034 | 17.827         | 296.27       | 0.9982 | 16.85        |
| 1196.1      | 1116.8      | 0.042 | 17.860         | 296.23       | 0.9978 | 19.45        |
| 1402.8      | 1317.3      | 0.050 | 17.893         | 296.23       | 0.9975 | 20.88        |
| 1674.4      | 1589.6      | 0.060 | 17.937         | 296.21       | 0.9971 | 21.39        |
| 1888.2      | 1798.6      | 0.067 | 17.973         | 296.24       | 0.9968 | 22.22        |
| 2133.6      | 2026.1      | 0.076 | 18.014         | 296.26       | 0.9965 | 24.37        |
| 2398.4      | 2292.2      | 0.086 | 18.060         | 296.27       | 0.9962 | 25.38        |
| 2576.3      | 2459.1      | 0.092 | 18.088         | 296.23       | 0.9960 | 26.71        |
| 2830.0      | 2699.0      | 0.101 | 18.131         | 296.23       | 0.9958 | 28.45        |
| 3161.0      | 3009.3      | 0.113 | 18.188         | 296.22       | 0.9955 | 30.74        |
| 3236.8      | 3064.4      | 0.115 | 18.200         | 296.21       | 0.9954 | 32.71        |
| 3278.5      | 3099.3      | 0.117 | 18.153         | 295.03       | 0.9951 | 33.79        |

CORE SAMPLE NO 1C RUN 4 REVERSE DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.0

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 536.00      | 511.00      | 513.00      | 69.840       | 0.8680      | 0.8750      | 17.000      |
| 498.00      | 477.00      | 500.00      | 69.840       | 0.8660      | 0.8760      | 18.400      |
| 469.00      | 451.00      | 500.00      | 69.840       | 0.8690      | 0.8780      | 19.700      |
| 446.00      | 433.60      | 500.00      | 69.840       | 0.8730      | 0.8820      | 21.300      |
| 408.00      | 397.30      | 500.00      | 69.840       | 0.8710      | 0.8780      | 22.400      |
| 307.00      | 296.00      | 500.00      | 69.840       | 0.8670      | 0.8790      | 23.500      |
| 290.00      | 281.00      | 500.00      | 69.850       | 0.8660      | 0.8810      | 25.600      |
| 259.00      | 250.50      | 500.00      | 69.850       | 0.8650      | 0.8800      | 27.350      |
| 218.00      | 209.60      | 500.00      | 69.860       | 0.8640      | 0.8750      | 29.700      |
| 168.00      | 158.80      | 500.00      | 69.870       | 0.8660      | 0.8750      | 31.900      |
| 130.00      | 120.60      | 500.00      | 69.870       | 0.8630      | 0.8760      | 36.100      |
| 98.000      | 88.300      | 500.00      | 69.870       | 0.8650      | 0.8760      | 42.300      |
| 76.200      | 65.900      | 500.00      | 69.870       | 0.8640      | 0.8740      | 46.900      |
| 55.000      | 44.500      | 500.00      | 69.920       | 0.8660      | 0.8750      | 55.600      |
| 40.000      | 30.000      | 500.00      | 69.920       | 0.8650      | 0.8750      | 70.400      |
| 25.000      | 14.500      | 500.00      | 69.920       | 0.8650      | 0.8760      | 82.300      |
| 16.000      | 6.3000      | 500.00      | 69.920       | 0.8670      | 0.8780      | 102.50      |
| 12.000      | 3.9000      | 500.00      | 69.920       | 0.8700      | 0.8760      | 129.80      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QD<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 3788.7      | 3616.3      | 0.136 | 18.258         | 295.22       | 0.9948 | 35.98        |
| 3526.7      | 3381.9      | 0.127 | 18.210         | 295.21       | 0.9950 | 33.24        |
| 3326.8      | 3202.6      | 0.120 | 18.178         | 295.27       | 0.9951 | 31.05        |
| 3168.2      | 3082.7      | 0.115 | 18.157         | 295.37       | 0.9952 | 28.72        |
| 2906.2      | 2832.4      | 0.105 | 18.107         | 295.29       | 0.9955 | 27.31        |
| 2209.8      | 2134.0      | 0.080 | 17.984         | 295.25       | 0.9962 | 26.03        |
| 2092.6      | 2030.6      | 0.076 | 17.966         | 295.27       | 0.9964 | 23.89        |
| 1878.9      | 1820.3      | 0.068 | 17.929         | 295.24       | 0.9966 | 22.36        |
| 1596.2      | 1538.3      | 0.057 | 17.879         | 295.17       | 0.9970 | 20.59        |
| 1251.5      | 1188.0      | 0.045 | 17.823         | 295.19       | 0.9976 | 19.17        |
| 989.47      | 924.66      | 0.035 | 17.780         | 295.17       | 0.9981 | 16.94        |
| 768.84      | 701.96      | 0.027 | 17.745         | 295.19       | 0.9985 | 14.46        |
| 618.53      | 547.52      | 0.021 | 17.719         | 295.16       | 0.9988 | 13.04        |
| 472.43      | 400.04      | 0.016 | 17.698         | 295.19       | 0.9991 | 11.00        |
| 369.01      | 300.06      | 0.012 | 17.681         | 295.18       | 0.9993 | 8.688        |
| 265.59      | 193.19      | 0.008 | 17.665         | 295.19       | 0.9995 | 7.432        |
| 203.53      | 136.66      | 0.006 | 17.658         | 295.24       | 0.9996 | 5.967        |
| 175.96      | 120.11      | 0.005 | 17.655         | 295.25       | 0.9997 | 4.712        |

CORE SAMPLE NO 1C RUN 5 FORWARD DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 7.0000      | 2.0000      | 600.00      | 70.330       | 0.8840      | 0.8840      | 199.70      |
| 10.000      | 3.3000      | 600.00      | 70.330       | 0.8880      | 0.8860      | 140.80      |
| 16.000      | 7.6000      | 600.00      | 70.330       | 0.8880      | 0.8880      | 97.400      |
| 21.900      | 10.500      | 600.00      | 70.330       | 0.8800      | 0.8930      | 64.700      |
| 29.600      | 16.000      | 600.00      | 70.330       | 0.8930      | 0.8950      | 47.400      |
| 40.400      | 27.500      | 600.00      | 70.330       | 0.8950      | 0.8990      | 40.950      |
| 60.000      | 47.800      | 600.00      | 70.330       | 0.8950      | 0.9000      | 33.000      |
| 80.000      | 69.000      | 600.00      | 70.330       | 0.9000      | 0.9030      | 31.000      |
| 99.800      | 90.100      | 600.00      | 70.330       | 0.9030      | 0.9030      | 30.500      |
| 132.00      | 123.60      | 600.00      | 70.330       | 0.9040      | 0.9040      | 29.900      |
| 160.00      | 151.80      | 600.00      | 70.280       | 0.9050      | 0.9090      | 28.980      |
| 187.80      | 179.80      | 600.00      | 70.280       | 0.9070      | 0.9100      | 27.600      |
| 210.40      | 201.60      | 600.00      | 70.280       | 0.9070      | 0.9080      | 26.300      |
| 240.00      | 231.00      | 600.00      | 70.280       | 0.9100      | 0.9100      | 25.140      |
| 270.00      | 260.40      | 600.00      | 70.280       | 0.9100      | 0.9140      | 24.100      |
| 299.60      | 288.20      | 600.00      | 70.280       | 0.9080      | 0.9130      | 23.150      |
| 350.00      | 338.30      | 600.00      | 70.280       | 0.8950      | 0.9080      | 22.200      |
| 402.00      | 390.00      | 600.00      | 70.280       | 0.8960      | 0.9100      | 21.300      |
| 455.00      | 441.00      | 600.00      | 70.280       | 0.8980      | 0.9100      | 20.400      |
| 508.00      | 492.00      | 600.00      | 70.280       | 0.9000      | 0.9100      | 19.400      |
| 542.00      | 524.00      | 600.00      | 70.280       | 0.8950      | 0.9100      | 18.760      |
| 584.00      | 563.00      | 600.00      | 70.280       | 0.8960      | 0.9100      | 17.320      |
| 600.00      | 578.00      | 600.00      | 70.280       | 0.8940      | 0.9070      | 16.800      |
| 632.00      | 609.00      | 610.00      | 70.280       | 0.9000      | 0.9080      | 16.250      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | OO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 142.03      | 107.55      | 0.005 | 17.664         | 295.53       | 0.9997 | 3.063        |
| 162.71      | 116.52      | 0.005 | 17.670         | 295.60       | 0.9997 | 4.344        |
| 204.08      | 146.17      | 0.006 | 17.676         | 295.63       | 0.9996 | 6.280        |
| 244.76      | 166.16      | 0.007 | 17.679         | 295.59       | 0.9996 | 9.454        |
| 297.85      | 204.08      | 0.009 | 17.695         | 295.77       | 0.9995 | 12.90        |
| 372.31      | 283.37      | 0.012 | 17.710         | 295.85       | 0.9993 | 14.94        |
| 507.45      | 423.33      | 0.017 | 17.733         | 295.86       | 0.9990 | 18.53        |
| 645.35      | 569.50      | 0.022 | 17.760         | 295.96       | 0.9988 | 19.73        |
| 781.86      | 714.98      | 0.027 | 17.784         | 296.00       | 0.9985 | 20.05        |
| 1003.9      | 945.96      | 0.036 | 17.821         | 296.02       | 0.9981 | 20.46        |
| 1196.9      | 1140.3      | 0.043 | 17.856         | 296.09       | 0.9978 | 21.11        |
| 1388.5      | 1333.4      | 0.050 | 17.889         | 296.13       | 0.9975 | 22.16        |
| 1544.4      | 1483.7      | 0.055 | 17.913         | 296.11       | 0.9972 | 23.26        |
| 1748.4      | 1686.4      | 0.063 | 17.949         | 296.17       | 0.9970 | 24.33        |
| 1955.3      | 1889.1      | 0.070 | 17.986         | 296.22       | 0.9967 | 25.38        |
| 2159.4      | 2080.8      | 0.077 | 18.017         | 296.18       | 0.9964 | 26.42        |
| 2506.9      | 2426.2      | 0.090 | 18.067         | 295.96       | 0.9960 | 27.55        |
| 2865.4      | 2782.7      | 0.103 | 18.131         | 296.00       | 0.9957 | 28.72        |
| 3230.8      | 3134.3      | 0.116 | 18.197         | 296.02       | 0.9954 | 29.98        |
| 3596.2      | 3485.9      | 0.130 | 18.265         | 296.05       | 0.9952 | 31.53        |
| 3830.7      | 3706.6      | 0.138 | 18.305         | 295.98       | 0.9950 | 32.60        |
| 4120.2      | 3975.4      | 0.148 | 18.360         | 296.00       | 0.9949 | 35.31        |
| 4230.6      | 4078.9      | 0.152 | 18.379         | 295.93       | 0.9949 | 36.41        |
| 4451.2      | 4292.6      | 0.160 | 18.426         | 296.02       | 0.9948 | 37.64        |

CORE SAMPLE NO 1C RUN 5 REVERSE DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 632.00      | 609.00      | 610.00      | 70.280       | 0.9000      | 0.9080      | 16.250      |
| 617.00      | 597.00      | 600.00      | 70.630       | 0.8920      | 0.8830      | 17.200      |
| 594.00      | 575.00      | 600.00      | 70.630       | 0.8700      | 0.8810      | 17.700      |
| 547.00      | 531.00      | 600.00      | 70.650       | 0.8630      | 0.8800      | 18.600      |
| 487.00      | 474.00      | 600.00      | 70.650       | 0.8600      | 0.8760      | 19.800      |
| 408.00      | 396.00      | 600.00      | 70.650       | 0.8600      | 0.8790      | 20.500      |
| 327.00      | 316.60      | 600.00      | 70.650       | 0.8600      | 0.8750      | 21.700      |
| 219.00      | 210.00      | 600.00      | 70.670       | 0.8620      | 0.8760      | 23.500      |
| 144.00      | 135.60      | 600.00      | 70.670       | 0.8610      | 0.8750      | 25.600      |
| 58.000      | 45.200      | 600.00      | 70.670       | 0.8600      | 0.8730      | 31.000      |
| 43.200      | 29.600      | 600.00      | 70.670       | 0.8650      | 0.8760      | 35.780      |
| 28.800      | 13.700      | 600.00      | 70.670       | 0.8680      | 0.8750      | 40.200      |
| 20.000      | 7.2000      | 600.00      | 70.670       | 0.8660      | 0.8750      | 55.000      |
| 12.000      | 3.7000      | 600.00      | 70.670       | 0.8690      | 0.8730      | 95.200      |
| 8.0000      | 2.0000      | 600.00      | 70.670       | 0.8670      | 0.8720      | 142.10      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 4451.2      | 4292.6      | 0.160 | 18.426         | 296.02       | 0.9948 | 37.64        |
| 4348.2      | 4210.3      | 0.157 | 18.389         | 295.61       | 0.9947 | 35.56        |
| 4189.7      | 4058.7      | 0.151 | 18.345         | 295.32       | 0.9947 | 34.56        |
| 3865.6      | 3755.3      | 0.140 | 18.279         | 295.22       | 0.9948 | 32.88        |
| 3451.9      | 3362.3      | 0.125 | 18.198         | 295.13       | 0.9950 | 30.89        |
| 2907.3      | 2824.5      | 0.105 | 18.101         | 295.17       | 0.9954 | 29.84        |
| 2348.8      | 2277.1      | 0.085 | 18.002         | 295.12       | 0.9960 | 28.19        |
| 1604.2      | 1542.1      | 0.058 | 17.879         | 295.16       | 0.9970 | 26.03        |
| 1087.1      | 1029.1      | 0.039 | 17.794         | 295.13       | 0.9979 | 23.89        |
| 494.11      | 405.86      | 0.016 | 17.695         | 295.09       | 0.9990 | 19.73        |
| 392.07      | 298.30      | 0.013 | 17.683         | 295.19       | 0.9993 | 17.09        |
| 292.79      | 188.68      | 0.009 | 17.668         | 295.22       | 0.9995 | 15.22        |
| 232.11      | 143.86      | 0.007 | 17.659         | 295.19       | 0.9996 | 11.12        |
| 176.96      | 119.73      | 0.005 | 17.653         | 295.21       | 0.9997 | 6.425        |
| 149.38      | 108.01      | 0.005 | 17.648         | 295.17       | 0.9997 | 4.304        |

CORE SAMPLE NO 1C RUN 6 FORWARD DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 8.0000      | 2.2000      | 700.00      | 70.380       | 0.8130      | 0.8070      | 193.80      |
| 14.000      | 6.3000      | 700.00      | 70.380       | 0.8210      | 0.8100      | 130.60      |
| 20.000      | 10.600      | 700.00      | 70.380       | 0.8250      | 0.8160      | 96.300      |
| 25.000      | 15.300      | 700.00      | 70.380       | 0.8270      | 0.8200      | 85.200      |
| 32.200      | 22.100      | 700.00      | 70.380       | 0.8310      | 0.8230      | 72.900      |
| 40.200      | 28.900      | 700.00      | 70.380       | 0.8330      | 0.8250      | 57.400      |
| 48.600      | 36.900      | 700.00      | 70.380       | 0.8350      | 0.8300      | 49.500      |
| 60.000      | 48.000      | 700.00      | 70.380       | 0.8360      | 0.8310      | 42.700      |
| 72.000      | 60.600      | 700.00      | 70.380       | 0.8390      | 0.8350      | 39.150      |
| 84.000      | 72.000      | 700.00      | 70.380       | 0.8400      | 0.8350      | 33.730      |
| 96.000      | 84.500      | 700.00      | 70.380       | 0.8390      | 0.8350      | 32.500      |
| 110.40      | 99.550      | 700.00      | 70.380       | 0.8350      | 0.8350      | 31.200      |
| 124.00      | 113.60      | 700.00      | 70.380       | 0.8380      | 0.8350      | 30.700      |
| 140.00      | 129.60      | 700.00      | 70.380       | 0.8350      | 0.8350      | 28.030      |
| 160.00      | 150.00      | 700.00      | 70.380       | 0.8340      | 0.8360      | 27.500      |
| 180.00      | 169.60      | 700.00      | 70.380       | 0.8340      | 0.8370      | 26.550      |
| 198.40      | 188.20      | 700.00      | 70.380       | 0.8320      | 0.8360      | 26.200      |
| 220.40      | 210.00      | 700.00      | 70.380       | 0.8340      | 0.8390      | 25.600      |
| 244.00      | 234.10      | 700.00      | 70.380       | 0.8330      | 0.8390      | 25.050      |
| 266.00      | 254.40      | 700.00      | 70.380       | 0.8330      | 0.8380      | 23.700      |
| 290.00      | 277.50      | 700.00      | 70.380       | 0.8340      | 0.8380      | 23.350      |
| 320.40      | 307.00      | 700.00      | 70.380       | 0.8300      | 0.8400      | 23.100      |
| 360.00      | 347.00      | 700.00      | 70.380       | 0.8300      | 0.8400      | 22.600      |
| 390.00      | 375.00      | 700.00      | 70.380       | 0.8330      | 0.8400      | 21.900      |
| 440.00      | 423.00      | 700.00      | 70.480       | 0.8490      | 0.8550      | 21.300      |
| 469.00      | 450.00      | 700.00      | 70.480       | 0.8460      | 0.8530      | 20.900      |
| 524.00      | 504.00      | 700.00      | 70.480       | 0.8480      | 0.8540      | 20.380      |
| 573.00      | 552.10      | 700.00      | 70.480       | 0.8450      | 0.8550      | 19.900      |
| 636.00      | 616.20      | 700.00      | 70.480       | 0.8450      | 0.8550      | 19.550      |
| 680.00      | 659.60      | 700.00      | 70.580       | 0.8450      | 0.8580      | 19.100      |
| 711.00      | 688.00      | 700.00      | 70.580       | 0.8450      | 0.8580      | 18.150      |
| 756.00      | 732.00      | 733.00      | 70.580       | 0.8500      | 0.8560      | 17.400      |

| PE<br>(KPA) | PW<br>(KPa) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 148.99      | 109.00      | 0.005 | 17.581         | 293.70       | 0.9997 | 3.156        |
| 190.36      | 137.27      | 0.006 | 17.593         | 293.83       | 0.9996 | 4.683        |
| 231.73      | 166.92      | 0.007 | 17.604         | 293.96       | 0.9995 | 6.351        |
| 266.20      | 199.32      | 0.009 | 17.613         | 294.03       | 0.9995 | 7.179        |
| 315.84      | 246.21      | 0.010 | 17.624         | 294.12       | 0.9994 | 8.390        |
| 371.00      | 293.09      | 0.012 | 17.635         | 294.17       | 0.9993 | 10.66        |
| 428.92      | 348.25      | 0.014 | 17.648         | 294.25       | 0.9991 | 12.36        |
| 507.52      | 424.78      | 0.017 | 17.661         | 294.28       | 0.9990 | 14.32        |
| 590.25      | 511.65      | 0.020 | 17.678         | 294.36       | 0.9988 | 15.62        |
| 672.99      | 590.25      | 0.023 | 17.692         | 294.38       | 0.9986 | 18.13        |
| 755.73      | 676.44      | 0.026 | 17.705         | 294.36       | 0.9985 | 18.82        |
| 855.01      | 780.21      | 0.030 | 17.718         | 294.32       | 0.9983 | 19.60        |
| 948.78      | 877.08      | 0.034 | 17.735         | 294.35       | 0.9981 | 19.92        |
| 1059.1      | 987.39      | 0.038 | 17.751         | 294.32       | 0.9979 | 21.82        |
| 1197.0      | 1128.0      | 0.043 | 17.774         | 294.32       | 0.9976 | 22.24        |
| 1334.9      | 1263.2      | 0.048 | 17.797         | 294.33       | 0.9974 | 23.04        |
| 1461.8      | 1391.4      | 0.052 | 17.816         | 294.29       | 0.9972 | 23.35        |
| 1613.4      | 1541.7      | 0.058 | 17.843         | 294.35       | 0.9969 | 23.89        |
| 1776.2      | 1707.9      | 0.064 | 17.870         | 294.34       | 0.9967 | 24.42        |
| 1927.8      | 1847.9      | 0.069 | 17.894         | 294.33       | 0.9964 | 25.81        |
| 2093.3      | 2007.1      | 0.075 | 17.922         | 294.34       | 0.9962 | 26.19        |
| 2302.9      | 2210.5      | 0.083 | 17.956         | 294.32       | 0.9959 | 26.48        |
| 2575.9      | 2486.3      | 0.093 | 18.003         | 294.32       | 0.9956 | 27.06        |
| 2782.8      | 2679.4      | 0.101 | 18.040         | 294.35       | 0.9954 | 27.93        |
| 3127.7      | 3010.4      | 0.113 | 18.118         | 294.74       | 0.9951 | 28.72        |
| 3327.6      | 3196.6      | 0.120 | 18.150         | 294.67       | 0.9950 | 29.27        |
| 3706.8      | 3568.9      | 0.134 | 18.222         | 294.71       | 0.9947 | 30.01        |
| 4044.7      | 3900.6      | 0.146 | 18.286         | 294.69       | 0.9945 | 30.74        |
| 4479.0      | 4342.5      | 0.162 | 18.374         | 294.69       | 0.9944 | 31.29        |
| 4782.5      | 4641.9      | 0.173 | 18.438         | 294.72       | 0.9943 | 32.02        |
| 4996.3      | 4837.7      | 0.181 | 18.481         | 294.72       | 0.9943 | 33.70        |
| 5306.5      | 5141.1      | 0.192 | 18.550         | 294.76       | 0.9944 | 35.15        |

CORE SAMPLE NO 1C RUN 6 REVERSE DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 756.00      | 732.00      | 733.00      | 70.580       | 0.8500      | 0.8560      | 17.400      |
| 707.00      | 684.00      | 700.00      | 70.580       | 0.8430      | 0.8540      | 18.250      |
| 665.00      | 643.00      | 700.00      | 70.580       | 0.8390      | 0.8520      | 18.850      |
| 633.00      | 613.00      | 700.00      | 70.580       | 0.8380      | 0.8490      | 19.400      |
| 587.00      | 570.00      | 700.00      | 70.580       | 0.8390      | 0.8500      | 20.000      |
| 492.00      | 477.00      | 700.00      | 70.000       | 0.8490      | 0.8510      | 21.250      |
| 455.00      | 441.00      | 700.00      | 70.000       | 0.8490      | 0.8510      | 21.800      |
| 396.00      | 382.00      | 700.00      | 69.940       | 0.8430      | 0.8500      | 22.550      |
| 354.00      | 341.00      | 700.00      | 69.940       | 0.8440      | 0.8460      | 23.100      |
| 275.00      | 261.40      | 700.00      | 69.940       | 0.8400      | 0.8450      | 23.800      |
| 257.00      | 245.00      | 700.00      | 69.940       | 0.8400      | 0.8410      | 24.400      |
| 232.00      | 221.00      | 700.00      | 69.940       | 0.8380      | 0.8380      | 25.300      |
| 201.00      | 192.00      | 700.00      | 69.940       | 0.8340      | 0.8380      | 26.500      |
| 159.00      | 150.00      | 700.00      | 69.940       | 0.8340      | 0.8370      | 27.300      |
| 111.00      | 101.00      | 700.00      | 69.940       | 0.8350      | 0.8390      | 29.100      |
| 80.400      | 70.200      | 700.00      | 69.940       | 0.8310      | 0.8370      | 33.650      |
| 60.400      | 49.790      | 700.00      | 69.940       | 0.8290      | 0.8360      | 39.320      |
| 48.200      | 36.400      | 700.00      | 69.870       | 0.8260      | 0.8360      | 40.700      |
| 20.400      | 9.5000      | 700.00      | 69.870       | 0.8280      | 0.8320      | 69.400      |
| 8.0000      | 2.0000      | 700.00      | 69.870       | 0.8310      | 0.8350      | 155.10      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 5306.5      | 5141.1      | 0.192 | 18.550         | 294.76       | 0.9944 | 35.15        |
| 4968.7      | 4810.1      | 0.180 | 18.472         | 294.65       | 0.9943 | 33.51        |
| 4679.1      | 4527.4      | 0.169 | 18.408         | 294.57       | 0.9943 | 32.45        |
| 4458.5      | 4320.6      | 0.162 | 18.362         | 294.53       | 0.9943 | 31.53        |
| 4141.3      | 4024.1      | 0.150 | 18.302         | 294.55       | 0.9944 | 30.58        |
| 3485.5      | 3382.1      | 0.126 | 18.183         | 294.69       | 0.9948 | 28.78        |
| 3230.4      | 3133.9      | 0.117 | 18.136         | 294.69       | 0.9950 | 28.06        |
| 2823.6      | 2727.0      | 0.102 | 18.059         | 294.60       | 0.9954 | 27.12        |
| 2534.0      | 2444.4      | 0.092 | 18.007         | 294.56       | 0.9957 | 26.48        |
| 1989.3      | 1895.5      | 0.071 | 17.911         | 294.50       | 0.9964 | 25.70        |
| 1865.2      | 1782.5      | 0.067 | 17.889         | 294.45       | 0.9966 | 25.07        |
| 1692.8      | 1617.0      | 0.061 | 17.858         | 294.39       | 0.9968 | 24.18        |
| 1479.1      | 1417.0      | 0.053 | 17.821         | 294.34       | 0.9971 | 23.08        |
| 1189.5      | 1127.5      | 0.043 | 17.774         | 294.33       | 0.9976 | 22.40        |
| 858.56      | 789.62      | 0.030 | 17.722         | 294.36       | 0.9983 | 21.02        |
| 647.58      | 577.26      | 0.022 | 17.685         | 294.29       | 0.9987 | 18.18        |
| 509.69      | 436.54      | 0.017 | 17.661         | 294.25       | 0.9990 | 15.56        |
| 425.48      | 344.12      | 0.014 | 17.645         | 294.22       | 0.9991 | 15.03        |
| 233.81      | 158.65      | 0.007 | 17.614         | 294.19       | 0.9996 | 8.813        |
| 148.31      | 106.94      | 0.005 | 17.607         | 294.27       | 0.9997 | 3.944        |

CORE SAMPLE NO 1C RUN 7 FORWARD DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 8.0000      | 2.0000      | 800.00      | 71.280       | 0.7710      | 0.7740      | 165.30      |
| 14.000      | 5.0000      | 800.00      | 71.280       | 0.7780      | 0.7810      | 101.00      |
| 22.000      | 11.800      | 800.00      | 71.280       | 0.7800      | 0.7840      | 77.600      |
| 30.000      | 17.300      | 800.00      | 71.280       | 0.7830      | 0.7850      | 55.850      |
| 40.400      | 28.000      | 800.00      | 71.280       | 0.7860      | 0.7880      | 48.550      |
| 60.200      | 49.000      | 800.00      | 71.280       | 0.7870      | 0.7870      | 41.900      |
| 80.000      | 69.200      | 800.00      | 71.280       | 0.7900      | 0.7870      | 36.500      |
| 100.00      | 90.000      | 800.00      | 71.280       | 0.7900      | 0.7900      | 33.370      |
| 132.00      | 122.50      | 800.00      | 71.280       | 0.7950      | 0.7950      | 30.680      |
| 159.80      | 150.80      | 800.00      | 71.280       | 0.7980      | 0.7950      | 29.630      |
| 200.00      | 190.80      | 800.00      | 71.280       | 0.7980      | 0.7950      | 28.250      |
| 240.40      | 229.80      | 800.00      | 71.280       | 0.8000      | 0.7960      | 27.300      |
| 280.40      | 267.80      | 800.00      | 71.280       | 0.8000      | 0.7950      | 25.750      |
| 300.40      | 286.20      | 800.00      | 71.280       | 0.8000      | 0.7950      | 24.700      |
| 348.00      | 333.00      | 800.00      | 69.850       | 0.7800      | 0.7800      | 23.800      |
| 420.00      | 403.00      | 800.00      | 69.800       | 0.7850      | 0.7850      | 22.750      |
| 474.00      | 455.00      | 800.00      | 69.800       | 0.7890      | 0.7860      | 21.650      |
| 495.00      | 475.00      | 800.00      | 69.800       | 0.7930      | 0.7900      | 21.100      |
| 557.00      | 535.00      | 800.00      | 69.800       | 0.7950      | 0.7940      | 20.250      |
| 612.00      | 590.00      | 800.00      | 69.800       | 0.8000      | 0.7960      | 19.750      |
| 675.00      | 652.00      | 800.00      | 69.800       | 0.8050      | 0.7990      | 18.870      |
| 711.00      | 687.00      | 800.00      | 69.800       | 0.8050      | 0.7990      | 18.300      |
| 754.00      | 728.00      | 800.00      | 69.800       | 0.8090      | 0.8010      | 17.600      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QD<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 150.19      | 108.82      | 0.005 | 17.539         | 292.77       | 0.9997 | 3.700        |
| 191.56      | 129.51      | 0.006 | 17.552         | 292.94       | 0.9996 | 6.056        |
| 246.72      | 176.39      | 0.008 | 17.563         | 293.01       | 0.9995 | 7.882        |
| 301.87      | 214.31      | 0.010 | 17.572         | 293.06       | 0.9994 | 10.95        |
| 373.58      | 288.09      | 0.012 | 17.587         | 293.13       | 0.9992 | 12.60        |
| 510.10      | 432.88      | 0.017 | 17.609         | 293.13       | 0.9989 | 14.60        |
| 646.61      | 572.15      | 0.022 | 17.633         | 293.17       | 0.9986 | 16.76        |
| 784.51      | 715.56      | 0.028 | 17.657         | 293.20       | 0.9983 | 18.33        |
| 1005.1      | 939.64      | 0.036 | 17.698         | 293.33       | 0.9979 | 19.94        |
| 1196.8      | 1134.8      | 0.043 | 17.731         | 293.36       | 0.9975 | 20.64        |
| 1474.0      | 1410.6      | 0.053 | 17.776         | 293.36       | 0.9970 | 21.65        |
| 1752.5      | 1679.4      | 0.063 | 17.823         | 293.40       | 0.9966 | 22.40        |
| 2028.3      | 1941.4      | 0.073 | 17.867         | 293.39       | 0.9961 | 23.75        |
| 2166.2      | 2068.3      | 0.078 | 17.890         | 293.39       | 0.9959 | 24.76        |
| 2492.5      | 2389.1      | 0.090 | 17.926         | 292.96       | 0.9954 | 25.70        |
| 2988.9      | 2871.6      | 0.108 | 18.017         | 293.08       | 0.9948 | 26.89        |
| 3361.2      | 3230.2      | 0.122 | 18.087         | 293.14       | 0.9945 | 28.25        |
| 3506.0      | 3368.1      | 0.127 | 18.118         | 293.24       | 0.9944 | 28.99        |
| 3933.4      | 3781.8      | 0.143 | 18.201         | 293.31       | 0.9941 | 30.20        |
| 4312.7      | 4161.0      | 0.157 | 18.280         | 293.40       | 0.9939 | 30.97        |
| 4747.0      | 4588.4      | 0.173 | 18.373         | 293.50       | 0.9938 | 32.41        |
| 4995.2      | 4829.8      | 0.182 | 18.425         | 293.50       | 0.9938 | 33.42        |
| 5291.7      | 5112.4      | 0.192 | 18.491         | 293.57       | 0.9939 | 34.75        |

CORE SAMPLE NO 1C RUN 7 REVERSE DIRECTION  
 RE=0.15239 RW=0.00319 H=0.02568 POROSITY=0.10620

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 799.00      | 772.50      | 800.00      | 69.980       | 0.8310      | 0.8270      | 17.000      |
| 776.00      | 752.50      | 800.00      | 69.980       | 0.8350      | 0.8270      | 18.000      |
| 745.00      | 727.00      | 800.00      | 69.980       | 0.8380      | 0.8320      | 19.450      |
| 694.00      | 678.50      | 800.00      | 69.980       | 0.8400      | 0.8350      | 20.400      |
| 604.00      | 590.00      | 800.00      | 70.000       | 0.8420      | 0.8350      | 21.400      |
| 526.00      | 514.00      | 800.00      | 70.000       | 0.8420      | 0.8350      | 22.500      |
| 464.00      | 453.00      | 800.00      | 70.000       | 0.84        | 0.8350      | 23.300      |
| 396.00      | 386.00      | 800.00      | 69.910       | 0.8200      | 0.8200      | 24.100      |
| 339.00      | 330.00      | 800.00      | 69.910       | 0.8220      | 0.8200      | 25.100      |
| 245.00      | 235.00      | 800.00      | 69.910       | 0.8250      | 0.8240      | 26.400      |
| 228.00      | 218.00      | 800.00      | 69.910       | 0.8270      | 0.8240      | 27.700      |
| 206.00      | 196.00      | 800.00      | 69.910       | 0.8300      | 0.8250      | 28.700      |
| 165.00      | 154.50      | 800.00      | 69.930       | 0.8330      | 0.8250      | 30.700      |
| 140.00      | 130.00      | 800.00      | 69.930       | 0.8330      | 0.8230      | 34.800      |
| 94.800      | 83.800      | 800.00      | 69.930       | 0.8350      | 0.8210      | 40.600      |
| 54.400      | 42.000      | 800.00      | 69.930       | 0.8320      | 0.8210      | 52.000      |
| 32.400      | 20.400      | 800.00      | 69.930       | 0.8350      | 0.8230      | 75.700      |
| 19.000      | 9.7000      | 800.00      | 69.930       | 0.8350      | 0.8250      | 123.00      |
| 14.000      | 5.2000      | 800.00      | 69.930       | 0.8340      | 0.8260      | 145.20      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG.K) | ZA     | QD<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 5602.2      | 5419.5      | 0.203 | 18.588         | 294.17       | 0.9942 | 35.98        |
| 5443.6      | 5281.6      | 0.198 | 18.556         | 294.22       | 0.9942 | 33.98        |
| 5229.9      | 5105.8      | 0.190 | 18.518         | 294.32       | 0.9942 | 31.45        |
| 4878.3      | 4771.4      | 0.178 | 18.446         | 294.38       | 0.9942 | 29.98        |
| 4257.8      | 4161.2      | 0.155 | 18.320         | 294.40       | 0.9943 | 28.58        |
| 3720.0      | 3637.2      | 0.135 | 18.217         | 294.41       | 0.9946 | 27.18        |
| 3292.5      | 3216.7      | 0.120 | 18.138         | 294.43       | 0.9949 | 26.25        |
| 2823.5      | 2754.6      | 0.103 | 18.032         | 293.94       | 0.9952 | 25.38        |
| 2430.5      | 2368.5      | 0.088 | 17.964         | 293.97       | 0.9957 | 24.37        |
| 1782.4      | 1713.5      | 0.064 | 17.858         | 294.06       | 0.9966 | 23.17        |
| 1665.2      | 1596.3      | 0.060 | 17.840         | 294.08       | 0.9968 | 22.08        |
| 1513.5      | 1444.6      | 0.054 | 17.817         | 294.13       | 0.9971 | 21.31        |
| 1230.9      | 1158.5      | 0.044 | 17.772         | 294.17       | 0.9976 | 19.92        |
| 1058.5      | 989.55      | 0.038 | 17.744         | 294.14       | 0.9979 | 17.58        |
| 746.86      | 671.01      | 0.026 | 17.693         | 294.14       | 0.9985 | 15.07        |
| 468.31      | 382.81      | 0.016 | 17.647         | 294.11       | 0.9991 | 11.76        |
| 316.62      | 233.89      | 0.010 | 17.626         | 294.17       | 0.9994 | 8.080        |
| 224.23      | 160.11      | 0.007 | 17.614         | 294.19       | 0.9996 | 4.973        |
| 189.76      | 129.08      | 0.006 | 17.609         | 294.19       | 0.9996 | 4.212        |

CORE SAMPLE NO 4A RUN 1 FORWARD DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.22714

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 8.0000      | 2.0000      | 4.00        | 70.170       | 0.7460      | 0.7450      | 129.60      |
| 16.000      | 7.8000      | 8.00        | 70.170       | 0.7510      | 0.7500      | 82.700      |
| 24.000      | 15.000      | 15.00       | 70.170       | 0.7550      | 0.7510      | 67.450      |
| 30.000      | 20.000      | 21.00       | 70.170       | 0.7570      | 0.7550      | 56.830      |
| 44.200      | 33.500      | 33.00       | 70.170       | 0.7580      | 0.7550      | 45.450      |
| 60.000      | 49.700      | 50.00       | 70.170       | 0.7600      | 0.7590      | 40.100      |
| 80.000      | 70.400      | 71.00       | 70.170       | 0.7630      | 0.7600      | 37.930      |
| 100.00      | 90.400      | 90.00       | 70.170       | 0.7650      | 0.7610      | 35.480      |
| 150.20      | 140.80      | 139.00      | 70.160       | 0.7700      | 0.7660      | 30.600      |
| 166.00      | 155.40      | 155.00      | 70.160       | 0.7700      | 0.7680      | 28.700      |
| 192.00      | 182.00      | 182.00      | 70.160       | 0.7700      | 0.7670      | 27.400      |
| 232.20      | 222.60      | 223.00      | 70.160       | 0.7710      | 0.7700      | 26.200      |
| 260.00      | 249.40      | 250.00      | 70.160       | 0.7720      | 0.7720      | 25.000      |
| 296.00      | 286.00      | 287.00      | 70.160       | 0.7730      | 0.7700      | 24.300      |
| 346.00      | 335.00      | 335.00      | 70.160       | 0.7610      | 0.7550      | 23.430      |
| 410.00      | 400.00      | 400.00      | 70.160       | 0.7660      | 0.7590      | 22.950      |
| 458.00      | 447.00      | 446.00      | 70.160       | 0.7670      | 0.7620      | 21.600      |
| 508.00      | 494.50      | 495.00      | 70.160       | 0.7690      | 0.7650      | 20.400      |
| 562.00      | 548.40      | 547.00      | 70.160       | 0.7700      | 0.7650      | 19.820      |
| 609.00      | 594.00      | 593.00      | 70.160       | 0.7730      | 0.7680      | 18.600      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 148.71      | 107.34      | 0.005 | 17.509         | 292.10       | 0.9997 | 4.719        |
| 203.87      | 147.33      | 0.006 | 17.522         | 292.23       | 0.9996 | 7.396        |
| 259.03      | 196.97      | 0.008 | 17.533         | 292.29       | 0.9994 | 9.068        |
| 300.39      | 231.45      | 0.010 | 17.542         | 292.36       | 0.9994 | 10.76        |
| 398.30      | 324.53      | 0.013 | 17.558         | 292.38       | 0.9991 | 13.46        |
| 507.24      | 436.22      | 0.017 | 17.579         | 292.45       | 0.9989 | 15.25        |
| 645.13      | 578.94      | 0.023 | 17.603         | 292.50       | 0.9986 | 16.13        |
| 783.03      | 716.84      | 0.028 | 17.627         | 292.54       | 0.9983 | 17.24        |
| 1129.1      | 1064.3      | 0.041 | 17.688         | 292.66       | 0.9976 | 19.99        |
| 1238.1      | 1165.0      | 0.044 | 17.706         | 292.68       | 0.9974 | 21.31        |
| 1417.3      | 1348.4      | 0.051 | 17.735         | 292.67       | 0.9970 | 22.32        |
| 1694.5      | 1628.3      | 0.061 | 17.783         | 292.72       | 0.9966 | 23.35        |
| 1886.2      | 1813.1      | 0.068 | 17.816         | 292.76       | 0.9962 | 24.47        |
| 2134.4      | 2065.4      | 0.078 | 17.858         | 292.75       | 0.9959 | 25.17        |
| 2479.1      | 2403.3      | 0.090 | 17.901         | 292.41       | 0.9953 | 26.11        |
| 2920.4      | 2851.4      | 0.107 | 17.984         | 292.52       | 0.9948 | 26.65        |
| 3251.3      | 3175.5      | 0.119 | 18.046         | 292.57       | 0.9944 | 28.32        |
| 3596.1      | 3503.0      | 0.132 | 18.111         | 292.63       | 0.9941 | 29.98        |
| 3968.4      | 3874.6      | 0.145 | 18.183         | 292.65       | 0.9939 | 30.86        |
| 4292.4      | 4189.0      | 0.157 | 18.250         | 292.72       | 0.9937 | 32.88        |

CORE SAMPLE NO 4A RUN 1 REVERSE DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.0

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 609.00      | 594.00      | 593.00      | 70.160       | 0.7730      | 0.7680      | 18.600      |
| 552.00      | 537.00      | 535.00      | 70.160       | 0.7700      | 0.7670      | 19.300      |
| 496.00      | 481.70      | 479.00      | 70.160       | 0.7700      | 0.7660      | 20.340      |
| 442.00      | 429.00      | 423.00      | 69.750       | 0.7710      | 0.7710      | 21.400      |
| 380.00      | 369.00      | 364.00      | 69.750       | 0.7680      | 0.7700      | 23.000      |
| 352.00      | 342.30      | 336.00      | 69.750       | 0.7670      | 0.7700      | 23.900      |
| 312.00      | 303.30      | 296.00      | 69.750       | 0.7670      | 0.7700      | 25.400      |
| 264.00      | 255.00      | 249.00      | 69.720       | 0.7700      | 0.7720      | 26.000      |
| 200.00      | 191.00      | 184.00      | 69.720       | 0.7650      | 0.7680      | 27.600      |
| 152.00      | 143.00      | 130.00      | 69.720       | 0.7660      | 0.7640      | 29.100      |
| 98.000      | 88.400      | 88.00       | 69.720       | 0.7670      | 0.7680      | 31.800      |
| 58.200      | 46.400      | 43.00       | 69.700       | 0.7660      | 0.7700      | 35.020      |
| 32.600      | 19.000      | 17.00       | 69.700       | 0.7660      | 0.7700      | 40.400      |
| 14.200      | 6.2000      | 9.00        | 69.700       | 0.7650      | 0.7700      | 89.100      |
| 8.0000      | 2.3000      | 7.00        | 69.700       | 0.7650      | 0.7700      | 136.60      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 4292.4      | 4189.0      | 0.157 | 18.250         | 292.72       | 0.9937 | 32.88        |
| 3899.4      | 3796.0      | 0.143 | 18.170         | 292.67       | 0.9939 | 31.69        |
| 3513.3      | 3414.7      | 0.128 | 18.096         | 292.66       | 0.9942 | 30.07        |
| 3140.5      | 3050.8      | 0.115 | 18.032         | 292.73       | 0.9946 | 28.58        |
| 2713.0      | 2637.2      | 0.099 | 17.954         | 292.68       | 0.9950 | 26.59        |
| 2519.9      | 2453.1      | 0.092 | 17.921         | 292.67       | 0.9953 | 25.59        |
| 2244.2      | 2184.2      | 0.082 | 17.874         | 292.67       | 0.9957 | 24.08        |
| 1913.2      | 1851.1      | 0.070 | 17.820         | 292.73       | 0.9962 | 23.52        |
| 1471.9      | 1409.9      | 0.053 | 17.742         | 292.62       | 0.9969 | 22.16        |
| 1141.0      | 1078.9      | 0.041 | 17.687         | 292.59       | 0.9976 | 21.02        |
| 768.64      | 702.45      | 0.027 | 17.629         | 292.65       | 0.9983 | 19.23        |
| 494.20      | 412.84      | 0.017 | 17.585         | 292.66       | 0.9989 | 17.47        |
| 317.69      | 223.93      | 0.010 | 17.556         | 292.66       | 0.9994 | 15.14        |
| 190.83      | 135.67      | 0.006 | 17.539         | 292.65       | 0.9996 | 6.865        |
| 148.08      | 108.78      | 0.005 | 17.533         | 292.65       | 0.9997 | 4.478        |

CORE SAMPLE NO 4A RUN 2 FORWARD DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.22600

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 8.0000      | 2.0000      | 400.00      | 70.270       | 0.7950      | 0.7920      | 131.45      |
| 12.000      | 4.7000      | 400.00      | 70.280       | 0.7950      | 0.7950      | 96.750      |
| 18.000      | 8.7000      | 400.00      | 70.280       | 0.7960      | 0.7990      | 66.330      |
| 24.000      | 13.100      | 400.00      | 70.280       | 0.7960      | 0.8010      | 50.120      |
| 30.000      | 17.800      | 400.00      | 70.280       | 0.7960      | 0.8020      | 39.870      |
| 40.200      | 28.400      | 400.00      | 70.280       | 0.7960      | 0.8050      | 34.500      |
| 54.200      | 43.800      | 400.00      | 70.280       | 0.7940      | 0.8050      | 32.000      |
| 70.000      | 60.400      | 400.00      | 70.280       | 0.7940      | 0.8060      | 30.640      |
| 90.000      | 81.000      | 400.00      | 70.280       | 0.7940      | 0.8060      | 29.700      |
| 108.20      | 99.300      | 400.00      | 70.280       | 0.7930      | 0.8050      | 28.350      |
| 138.20      | 129.40      | 400.00      | 70.260       | 0.7940      | 0.8050      | 27.200      |
| 160.00      | 151.00      | 400.00      | 70.280       | 0.7910      | 0.8040      | 26.260      |
| 186.60      | 176.40      | 400.00      | 70.280       | 0.7920      | 0.8040      | 25.100      |
| 210.80      | 199.40      | 400.00      | 70.280       | 0.7900      | 0.8040      | 24.300      |
| 238.00      | 226.70      | 400.00      | 70.280       | 0.7900      | 0.8050      | 23.700      |
| 265.00      | 251.00      | 400.00      | 70.470       | 0.7520      | 0.7520      | 22.300      |
| 298.00      | 282.00      | 400.00      | 70.470       | 0.7570      | 0.7570      | 21.200      |
| 325.00      | 307.00      | 400.00      | 70.470       | 0.7600      | 0.7600      | 20.130      |
| 372.00      | 353.00      | 400.00      | 70.470       | 0.7600      | 0.7630      | 19.00       |
| 406.00      | 385.40      | 400.00      | 70.470       | 0.7630      | 0.7650      | 18.110      |
| 435.00      | 413.00      | 413.00      | 70.470       | 0.7660      | 0.7660      | 17.130      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 148.84      | 107.47      | 0.005 | 17.563         | 293.29       | 0.9997 | 4.653        |
| 176.44      | 126.10      | 0.006 | 17.568         | 293.33       | 0.9996 | 6.322        |
| 217.80      | 153.68      | 0.007 | 17.576         | 293.39       | 0.9996 | 9.221        |
| 259.17      | 184.02      | 0.008 | 17.583         | 293.41       | 0.9995 | 12.20        |
| 300.54      | 216.43      | 0.010 | 17.589         | 293.43       | 0.9994 | 15.34        |
| 370.87      | 289.51      | 0.012 | 17.602         | 293.46       | 0.9992 | 17.73        |
| 467.39      | 395.69      | 0.016 | 17.617         | 293.44       | 0.9990 | 19.11        |
| 576.33      | 510.14      | 0.020 | 17.635         | 293.45       | 0.9988 | 19.96        |
| 714.23      | 652.17      | 0.025 | 17.658         | 293.45       | 0.9985 | 20.59        |
| 839.71      | 778.35      | 0.030 | 17.677         | 293.43       | 0.9982 | 21.57        |
| 1046.5      | 885.85      | 0.037 | 17.710         | 293.44       | 0.9978 | 22.40        |
| 1196.9      | 1134.8      | 0.043 | 17.732         | 293.39       | 0.9975 | 23.29        |
| 1380.3      | 1309.9      | 0.050 | 17.762         | 293.40       | 0.9972 | 24.37        |
| 1547.1      | 1468.5      | 0.056 | 17.787         | 293.38       | 0.9969 | 25.17        |
| 1734.7      | 1656.7      | 0.063 | 17.819         | 293.39       | 0.9966 | 25.81        |
| 1921.1      | 1824.5      | 0.069 | 17.797         | 292.26       | 0.9961 | 27.43        |
| 2148.6      | 2038.3      | 0.078 | 17.840         | 292.39       | 0.9958 | 28.85        |
| 2334.7      | 2210.6      | 0.084 | 17.874         | 292.46       | 0.9955 | 30.38        |
| 2658.8      | 2527.8      | 0.096 | 17.931         | 292.50       | 0.9951 | 31.86        |
| 2893.2      | 2751.2      | 0.105 | 17.975         | 292.56       | 0.9948 | 33.77        |
| 3093.2      | 2941.5      | 0.112 | 18.012         | 292.61       | 0.9946 | 35.71        |

CORE SAMPLE NO 4A RUN 2 REVERSE DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.0

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 435.00      | 413.00      | 413.00      | 70.470       | 0.7660      | 0.7660      | 17.130      |
| 400.00      | 379.30      | 400.00      | 70.470       | 0.7650      | 0.7670      | 18.200      |
| 386.00      | 367.40      | 400.00      | 70.470       | 0.7650      | 0.7700      | 19.000      |
| 350.00      | 333.80      | 400.00      | 70.470       | 0.7650      | 0.7710      | 20.200      |
| 320.00      | 308.40      | 400.00      | 70.540       | 0.7660      | 0.7710      | 21.750      |
| 291.00      | 281.60      | 400.00      | 70.540       | 0.7670      | 0.7720      | 23.330      |
| 260.00      | 251.60      | 400.00      | 70.540       | 0.7680      | 0.7730      | 24.700      |
| 237.00      | 229.00      | 400.00      | 70.540       | 0.7700      | 0.7730      | 25.300      |
| 180.00      | 171.40      | 400.00      | 70.520       | 0.7690      | 0.7750      | 27.100      |
| 168.00      | 159.50      | 400.00      | 70.520       | 0.7700      | 0.7750      | 28.350      |
| 146.00      | 137.90      | 400.00      | 70.520       | 0.7720      | 0.7760      | 30.400      |
| 125.00      | 117.00      | 400.00      | 70.520       | 0.7720      | 0.7760      | 32.250      |
| 105.60      | 97.300      | 400.00      | 70.770       | 0.7400      | 0.7400      | 33.320      |
| 96.000      | 87.500      | 400.00      | 70.770       | 0.7530      | 0.7560      | 34.400      |
| 70.800      | 60.800      | 400.00      | 70.770       | 0.7560      | 0.7560      | 35.700      |
| 46.000      | 34.400      | 400.00      | 70.770       | 0.7600      | 0.7610      | 41.700      |
| 30.200      | 21.100      | 400.00      | 70.770       | 0.7610      | 0.7640      | 67.200      |
| 10.800      | 3.2000      | 400.00      | 70.770       | 0.7680      | 0.7670      | 119.80      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 3093.2      | 2941.5      | 0.112 | 18.012         | 292.61       | 0.9946 | 35.71        |
| 2851.9      | 2709.1      | 0.03  | 17.969         | 292.61       | 0.9949 | 33.61        |
| 2755.3      | 2627.1      | 0.100 | 17.955         | 292.65       | 0.9950 | 32.19        |
| 2507.1      | 2395.4      | 0.091 | 17.914         | 292.66       | 0.9953 | 30.28        |
| 2300.4      | 2220.4      | 0.084 | 17.881         | 292.67       | 0.9956 | 28.12        |
| 2100.4      | 2035.6      | 0.076 | 17.850         | 292.70       | 0.9959 | 26.22        |
| 1886.7      | 1828.8      | 0.069 | 17.816         | 292.72       | 0.9962 | 24.76        |
| 1728.1      | 1672.9      | 0.063 | 17.791         | 292.75       | 0.9965 | 24.18        |
| 1335.1      | 1275.8      | 0.048 | 17.726         | 292.76       | 0.9972 | 22.57        |
| 1252.3      | 1193.7      | 0.045 | 17.713         | 292.77       | 0.9974 | 21.57        |
| 1100.7      | 1044.8      | 0.040 | 17.691         | 292.81       | 0.9976 | 20.12        |
| 955.86      | 900.71      | 0.034 | 17.668         | 292.81       | 0.9979 | 18.97        |
| 822.44      | 765.21      | 0.029 | 17.608         | 291.97       | 0.9982 | 18.36        |
| 756.25      | 697.64      | 0.027 | 17.613         | 292.33       | 0.9983 | 17.78        |
| 582.50      | 513.55      | 0.020 | 17.587         | 292.36       | 0.9987 | 17.13        |
| 411.51      | 331.53      | 0.014 | 17.564         | 292.47       | 0.9991 | 14.67        |
| 302.57      | 239.83      | 0.010 | 17.550         | 292.52       | 0.9993 | 9.102        |
| 168.82      | 116.42      | 0.005 | 17.536         | 292.65       | 0.9997 | 5.106        |

CORE SAMPLE NO 4A RUN 3 FORWARD DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.0

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 8.0000      | 2.0000      | 500.00      | 70.270       | 0.7950      | 0.7920      | 155.60      |
| 14.000      | 4.6000      | 500.00      | 70.270       | 0.8040      | 0.8000      | 88.500      |
| 22.000      | 9.5000      | 500.00      | 70.270       | 0.8050      | 0.8050      | 57.500      |
| 28.000      | 14.400      | 500.00      | 70.270       | 0.8100      | 0.8070      | 48.300      |
| 38.200      | 24.800      | 500.00      | 70.270       | 0.8100      | 0.8100      | 41.300      |
| 50.000      | 38.400      | 500.00      | 70.270       | 0.8110      | 0.8110      | 40.300      |
| 64.000      | 53.500      | 500.00      | 70.270       | 0.8140      | 0.8140      | 39.270      |
| 80.000      | 70.500      | 500.00      | 70.270       | 0.8140      | 0.8160      | 38.000      |
| 98.000      | 89.000      | 500.00      | 70.270       | 0.8130      | 0.8160      | 37.350      |
| 120.00      | 111.60      | 500.00      | 70.270       | 0.8100      | 0.8140      | 36.800      |
| 140.00      | 130.90      | 500.00      | 70.270       | 0.8080      | 0.8100      | 35.000      |
| 169.20      | 160.00      | 500.00      | 70.340       | 0.8050      | 0.8100      | 34.000      |
| 194.20      | 185.40      | 500.00      | 70.340       | 0.8010      | 0.8100      | 33.530      |
| 224.40      | 215.40      | 500.00      | 70.340       | 0.8000      | 0.8070      | 32.650      |
| 257.60      | 248.70      | 500.00      | 70.340       | 0.7980      | 0.8070      | 31.500      |
| 282.40      | 273.00      | 500.00      | 70.340       | 0.7950      | 0.8060      | 30.800      |
| 316.00      | 306.00      | 500.00      | 70.340       | 0.7950      | 0.8060      | 29.530      |
| 360.40      | 350.00      | 500.00      | 70.340       | 0.7980      | 0.8050      | 28.600      |
| 412.00      | 400.00      | 500.00      | 70.380       | 0.8100      | 0.8070      | 27.230      |
| 455.00      | 441.00      | 500.00      | 70.380       | 0.8110      | 0.8060      | 25.700      |
| 504.00      | 489.50      | 500.00      | 70.380       | 0.8150      | 0.8060      | 24.450      |
| 534.00      | 519.00      | 517.00      | 70.380       | 0.8110      | 0.8060      | 23.550      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QD<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 148.84      | 107.47      | 0.005 | 17.563         | 293.29       | 0.9997 | 3.931        |
| 190.21      | 125.40      | 0.006 | 17.577         | 293.50       | 0.9996 | 6.911        |
| 245.37      | 159.19      | 0.007 | 17.587         | 293.57       | 0.9995 | 10.64        |
| 286.74      | 192.97      | 0.009 | 17.597         | 293.66       | 0.9994 | 12.66        |
| 357.07      | 264.68      | 0.011 | 17.610         | 293.70       | 0.9993 | 14.81        |
| 438.42      | 358.44      | 0.015 | 17.625         | 293.72       | 0.9991 | 15.18        |
| 534.95      | 462.55      | 0.018 | 17.644         | 293.80       | 0.9989 | 15.58        |
| 645.27      | 579.77      | 0.023 | 17.663         | 293.82       | 0.9986 | 16.10        |
| 769.37      | 707.32      | 0.027 | 17.683         | 293.81       | 0.9984 | 16.38        |
| 921.06      | 863.14      | 0.033 | 17.705         | 293.75       | 0.9981 | 16.62        |
| 1059.0      | 996.21      | 0.038 | 17.723         | 293.67       | 0.9978 | 17.48        |
| 1260.4      | 1196.9      | 0.045 | 17.754         | 293.64       | 0.9974 | 17.99        |
| 1432.7      | 1372.1      | 0.052 | 17.780         | 293.59       | 0.9971 | 18.24        |
| 1641.0      | 1578.9      | 0.059 | 17.812         | 293.54       | 0.9968 | 18.73        |
| 1869.9      | 1808.5      | 0.068 | 17.848         | 293.51       | 0.9964 | 19.42        |
| 2040.9      | 1976.0      | 0.074 | 17.875         | 293.46       | 0.9961 | 19.86        |
| 2272.5      | 2203.6      | 0.083 | 17.914         | 293.46       | 0.9958 | 20.71        |
| 2578.6      | 2506.9      | 0.094 | 17.967         | 293.49       | 0.9954 | 21.39        |
| 2934.5      | 2851.7      | 0.107 | 18.037         | 293.66       | 0.9950 | 22.46        |
| 3230.9      | 3134.4      | 0.117 | 18.090         | 293.66       | 0.9947 | 23.80        |
| 3568.8      | 3468.8      | 0.130 | 18.154         | 293.71       | 0.9945 | 25.02        |
| 3775.6      | 3672.2      | 0.138 | 18.191         | 293.66       | 0.9943 | 25.97        |

CORE SAMPLE NO 4A RUN 3 REVERSE DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.0

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 534.00      | 519.00      | 517.00      | 70.380       | 0.8110      | 0.8060      | 23.550      |
| 488.00      | 472.90      | 500.00      | 69.950       | 0.8440      | 0.8330      | 24.400      |
| 427.00      | 412.50      | 500.00      | 69.950       | 0.8530      | 0.8400      | 25.600      |
| 372.00      | 357.40      | 500.00      | 69.950       | 0.8600      | 0.8550      | 26.400      |
| 315.00      | 299.90      | 500.00      | 69.950       | 0.8570      | 0.8570      | 27.300      |
| 263.00      | 249.10      | 500.00      | 69.950       | 0.8570      | 0.8600      | 28.200      |
| 236.00      | 222.30      | 500.00      | 69.800       | 0.8590      | 0.8600      | 29.000      |
| 188.00      | 176.80      | 500.00      | 69.800       | 0.8600      | 0.8600      | 30.500      |
| 138.00      | 129.00      | 500.00      | 69.800       | 0.8600      | 0.8600      | 32.900      |
| 104.00      | 95.800      | 500.00      | 69.800       | 0.8600      | 0.8620      | 39.100      |
| 61.600      | 52.000      | 500.00      | 69.800       | 0.8600      | 0.8630      | 44.730      |
| 40.000      | 28.700      | 500.00      | 69.800       | 0.8600      | 0.8640      | 52.200      |
| 24.200      | 14.200      | 500.00      | 69.800       | 0.8610      | 0.8610      | 78.650      |
| 10.000      | 3.2000      | 500.00      | 69.800       | 0.8630      | 0.8630      | 154.05      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QD<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 3775.6      | 3672.2      | 0.138 | 18.191         | 293.66       | 0.9943 | 25.97        |
| 3457.9      | 3353.8      | 0.125 | 18.165         | 294.40       | 0.9948 | 25.07        |
| 3037.3      | 2937.3      | 0.110 | 18.097         | 294.60       | 0.9952 | 23.89        |
| 2658.1      | 2557.4      | 0.096 | 18.042         | 294.87       | 0.9956 | 23.17        |
| 2265.1      | 2161.0      | 0.081 | 17.973         | 294.86       | 0.9961 | 22.40        |
| 1906.6      | 1810.7      | 0.068 | 17.915         | 294.90       | 0.9966 | 21.69        |
| 1720.2      | 1625.8      | 0.061 | 17.885         | 294.92       | 0.9969 | 21.09        |
| 1389.3      | 1312.1      | 0.050 | 17.833         | 294.93       | 0.9974 | 20.05        |
| 1044.5      | 982.48      | 0.037 | 17.778         | 294.93       | 0.9979 | 18.59        |
| 810.11      | 753.58      | 0.029 | 17.742         | 294.96       | 0.9984 | 15.64        |
| 517.78      | 451.59      | 0.018 | 17.695         | 294.97       | 0.9990 | 13.67        |
| 368.85      | 290.94      | 0.012 | 17.671         | 294.98       | 0.9993 | 11.72        |
| 259.91      | 190.96      | 0.008 | 17.654         | 294.96       | 0.9995 | 7.777        |
| 162.01      | 115.12      | 0.005 | 17.642         | 295.01       | 0.9997 | 3.970        |

CORE SAMPLE NO 4A RUN 4 FORWARD DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.0

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 10.000      | 2.7000      | 600.00      | 70.000       | 0.7770      | 0.7770      | 143.00      |
| 16.000      | 7.6000      | 600.00      | 70.000       | 0.7780      | 0.7780      | 105.75      |
| 24.000      | 13.200      | 600.00      | 70.000       | 0.7780      | 0.7790      | 69.980      |
| 30.000      | 18.100      | 600.00      | 70.000       | 0.7750      | 0.7760      | 56.300      |
| 36.000      | 22.600      | 600.00      | 70.000       | 0.7740      | 0.7760      | 45.270      |
| 50.200      | 39.300      | 600.00      | 70.000       | 0.7740      | 0.7760      | 43.000      |
| 64.000      | 54.500      | 600.00      | 69.940       | 0.7740      | 0.7760      | 41.380      |
| 80.000      | 71.400      | 600.00      | 69.940       | 0.7750      | 0.7760      | 39.020      |
| 100.20      | 92.000      | 600.00      | 69.940       | 0.7740      | 0.7760      | 36.900      |
| 121.40      | 113.60      | 600.00      | 69.940       | 0.7740      | 0.7760      | 35.250      |
| 140.60      | 133.30      | 600.00      | 70.140       | 0.7600      | 0.7670      | 34.800      |
| 160.00      | 152.60      | 600.00      | 70.170       | 0.7630      | 0.7700      | 34.100      |
| 180.20      | 172.70      | 600.00      | 70.170       | 0.7640      | 0.7730      | 33.300      |
| 200.00      | 192.40      | 600.00      | 70.200       | 0.7640      | 0.7750      | 32.190      |
| 230.00      | 222.30      | 600.00      | 70.200       | 0.7660      | 0.7750      | 31.300      |
| 260.40      | 253.00      | 600.00      | 70.250       | 0.7650      | 0.7800      | 30.400      |
| 290.00      | 282.50      | 600.00      | 70.250       | 0.7650      | 0.7750      | 29.700      |
| 320.60      | 312.20      | 600.00      | 70.300       | 0.7550      | 0.7560      | 28.300      |
| 352.00      | 343.00      | 600.00      | 70.300       | 0.7550      | 0.7610      | 27.400      |
| 367.60      | 356.70      | 600.00      | 70.300       | 0.7520      | 0.7570      | 26.500      |
| 383.80      | 371.30      | 600.00      | 70.300       | 0.7570      | 0.7630      | 25.400      |
| 432.00      | 419.20      | 600.00      | 70.300       | 0.7600      | 0.7650      | 24.500      |
| 457.00      | 442.40      | 600.00      | 70.300       | 0.7570      | 0.7650      | 23.450      |
| 528.00      | 512.30      | 600.00      | 70.300       | 0.7600      | 0.7650      | 22.400      |
| 562.00      | 544.90      | 600.00      | 70.300       | 0.7600      | 0.7660      | 21.300      |
| 602.00      | 584.00      | 600.00      | 70.300       | 0.7700      | 0.7740      | 20.400      |
| 645.00      | 627.00      | 627.00      | 71.100       | 0.7500      | 0.7560      | 19.900      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 162.27      | 111.94      | 0.005 | 17.545         | 292.88       | 0.9997 | 4.277        |
| 203.64      | 145.73      | 0.006 | 17.553         | 292.91       | 0.9996 | 5.784        |
| 258.80      | 184.34      | 0.008 | 17.560         | 292.92       | 0.9995 | 8.740        |
| 300.17      | 218.12      | 0.010 | 17.563         | 292.84       | 0.9994 | 10.86        |
| 341.54      | 249.15      | 0.011 | 17.568         | 292.83       | 0.9993 | 13.51        |
| 439.44      | 364.29      | 0.015 | 17.585         | 292.83       | 0.9991 | 14.22        |
| 534.51      | 469.01      | 0.018 | 17.601         | 292.83       | 0.9988 | 14.78        |
| 644.83      | 585.53      | 0.023 | 17.619         | 292.84       | 0.9986 | 15.68        |
| 784.10      | 727.56      | 0.028 | 17.641         | 292.83       | 0.9983 | 16.58        |
| 930.27      | 876.49      | 0.033 | 17.665         | 292.83       | 0.9980 | 17.35        |
| 1062.9      | 1012.6      | 0.038 | 17.673         | 292.55       | 0.9977 | 17.58        |
| 1196.7      | 1145.7      | 0.043 | 17.698         | 292.62       | 0.9974 | 17.94        |
| 1336.0      | 1284.3      | 0.048 | 17.723         | 292.67       | 0.9972 | 18.37        |
| 1472.5      | 1420.1      | 0.053 | 17.746         | 292.70       | 0.9969 | 19.00        |
| 1679.4      | 1626.3      | 0.061 | 17.782         | 292.72       | 0.9966 | 19.54        |
| 1889.1      | 1838.0      | 0.069 | 17.819         | 292.77       | 0.9962 | 20.12        |
| 2093.1      | 2041.4      | 0.076 | 17.850         | 292.71       | 0.9959 | 20.59        |
| 2304.2      | 2246.3      | 0.084 | 17.869         | 292.35       | 0.9955 | 21.61        |
| 2520.7      | 2458.6      | 0.092 | 17.909         | 292.41       | 0.9952 | 22.32        |
| 2628.2      | 2553.1      | 0.096 | 17.923         | 292.33       | 0.9951 | 23.08        |
| 2739.9      | 2653.7      | 0.100 | 17.948         | 292.46       | 0.9950 | 24.08        |
| 3072.3      | 2984.0      | 0.112 | 18.010         | 292.52       | 0.9946 | 24.97        |
| 3244.6      | 3144.0      | 0.118 | 18.038         | 292.49       | 0.9944 | 26.08        |
| 3734.2      | 3625.9      | 0.136 | 18.131         | 292.52       | 0.9940 | 27.31        |
| 3968.6      | 3850.7      | 0.145 | 18.176         | 292.54       | 0.9938 | 28.72        |
| 4244.4      | 4120.3      | 0.155 | 18.240         | 292.76       | 0.9937 | 29.98        |
| 4541.9      | 4417.8      | 0.166 | 18.279         | 292.29       | 0.9934 | 30.74        |

CORE SAMPLE NO 4A RUN 4 REVERSE DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.0

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 645.00      | 627.00      | 627.00      | 71.100       | 0.7500      | 0.7560      | 19.900      |
| 597.00      | 579.20      | 600.00      | 71.100       | 0.7570      | 0.7600      | 20.600      |
| 557.00      | 540.00      | 600.00      | 71.130       | 0.7590      | 0.7600      | 21.400      |
| 519.00      | 505.10      | 600.00      | 71.220       | 0.7720      | 0.7770      | 22.700      |
| 453.00      | 440.00      | 600.00      | 71.220       | 0.7750      | 0.7800      | 23.920      |
| 414.00      | 403.00      | 600.00      | 71.270       | 0.7860      | 0.7900      | 24.980      |
| 338.00      | 326.30      | 600.00      | 71.270       | 0.7860      | 0.7900      | 25.800      |
| 297.00      | 287.00      | 600.00      | 71.270       | 0.7850      | 0.7900      | 26.900      |
| 256.80      | 247.00      | 600.00      | 71.270       | 0.7850      | 0.7900      | 27.900      |
| 218.00      | 208.00      | 600.00      | 71.270       | 0.7860      | 0.7900      | 28.450      |
| 170.00      | 160.20      | 600.00      | 71.270       | 0.7880      | 0.7880      | 29.780      |
| 140.00      | 130.22      | 600.00      | 71.270       | 0.7880      | 0.7880      | 31.400      |
| 110.00      | 100.20      | 600.00      | 71.270       | 0.7880      | 0.7920      | 33.930      |
| 79.800      | 68.400      | 600.00      | 71.270       | 0.7880      | 0.7920      | 36.150      |
| 59.600      | 47.300      | 600.00      | 71.270       | 0.7880      | 0.7930      | 40.280      |
| 39.800      | 27.000      | 600.00      | 71.270       | 0.7890      | 0.7920      | 50.900      |
| 20.000      | 8.0000      | 600.00      | 71.270       | 0.7890      | 0.7890      | 78.900      |
| 12.000      | 4.1000      | 600.00      | 71.270       | 0.7890      | 0.7890      | 139.70      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZÄ     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 4541.9      | 4417.8      | 0.166 | 18.279         | 292.29       | 0.9934 | 30.74        |
| 4211.0      | 4088.2      | 0.154 | 18.218         | 292.42       | 0.9936 | 29.69        |
| 3935.2      | 3818.0      | 0.144 | 18.165         | 292.45       | 0.9938 | 28.58        |
| 3673.3      | 3577.5      | 0.134 | 18.134         | 292.82       | 0.9941 | 26.94        |
| 3218.3      | 3128.6      | 0.117 | 18.053         | 292.89       | 0.9945 | 25.57        |
| 2949.4      | 2873.6      | 0.108 | 18.047         | 293.15       | 0.9949 | 24.49        |
| 2425.4      | 2344.8      | 0.088 | 17.925         | 293.15       | 0.9955 | 28.71        |
| 2142.8      | 2073.8      | 0.078 | 17.877         | 293.14       | 0.9959 | 22.74        |
| 1865.6      | 1798.0      | 0.068 | 17.830         | 293.14       | 0.9963 | 21.92        |
| 1598.1      | 1529.1      | 0.058 | 17.786         | 293.15       | 0.9968 | 21.50        |
| 1267.1      | 1199.6      | 0.045 | 17.733         | 293.15       | 0.9974 | 20.54        |
| 1060.3      | 992.85      | 0.038 | 17.699         | 293.15       | 0.9978 | 19.48        |
| 853.44      | 785.87      | 0.030 | 17.668         | 293.20       | 0.9982 | 18.03        |
| 645.22      | 566.62      | 0.022 | 17.634         | 293.20       | 0.9986 | 16.92        |
| 505.95      | 421.14      | 0.017 | 17.612         | 293.22       | 0.9989 | 15.18        |
| 369.43      | 281.18      | 0.012 | 17.590         | 293.22       | 0.9992 | 12.02        |
| 232.91      | 150.18      | 0.007 | 17.568         | 293.18       | 0.9995 | 7.752        |
| 177.76      | 123.29      | 0.006 | 17.561         | 293.18       | 0.9996 | 4.378        |

CORE SAMPLE NO 4A RUN 5 FORWARD DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.22714

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 8.0000      | 2.2000      | 700.00      | 70.800       | 0.7700      | 0.7680      | 138.20      |
| 12.000      | 4.2000      | 700.00      | 70.800       | 0.7800      | 0.7770      | 96.330      |
| 19.400      | 9.6000      | 700.00      | 70.800       | 0.7890      | 0.7890      | 67.330      |
| 26.000      | 14.400      | 700.00      | 70.840       | 0.7840      | 0.7840      | 51.500      |
| 32.400      | 20.200      | 700.00      | 70.840       | 0.7830      | 0.7870      | 44.230      |
| 40.000      | 28.320      | 700.00      | 70.840       | 0.7850      | 0.7850      | 41.900      |
| 54.000      | 43.500      | 700.00      | 70.840       | 0.7850      | 0.7860      | 38.150      |
| 70.000      | 60.000      | 700.00      | 70.840       | 0.7850      | 0.7890      | 34.180      |
| 90.000      | 81.000      | 700.00      | 70.840       | 0.7850      | 0.7880      | 31.900      |
| 108.00      | 99.000      | 700.00      | 70.880       | 0.7850      | 0.7890      | 29.700      |
| 126.40      | 117.80      | 700.00      | 70.910       | 0.7840      | 0.7880      | 28.000      |
| 142.00      | 133.40      | 700.00      | 70.910       | 0.7820      | 0.7850      | 27.200      |
| 160.00      | 150.60      | 700.00      | 70.910       | 0.7830      | 0.7890      | 26.430      |
| 180.00      | 170.60      | 700.00      | 70.910       | 0.7840      | 0.7900      | 26.000      |
| 199.60      | 189.20      | 700.00      | 70.910       | 0.7740      | 0.7800      | 25.400      |
| 224.40      | 214.20      | 700.00      | 70.950       | 0.7800      | 0.7850      | 24.900      |
| 250.00      | 236.20      | 700.00      | 70.950       | 0.7780      | 0.7850      | 24.050      |
| 272.00      | 257.80      | 700.00      | 70.950       | 0.7780      | 0.7850      | 23.700      |
| 300.00      | 284.00      | 700.00      | 70.950       | 0.7760      | 0.7850      | 22.900      |
| 334.60      | 319.00      | 700.00      | 69.640       | 0.7720      | 0.7800      | 22.620      |
| 386.20      | 371.50      | 700.00      | 69.640       | 0.7720      | 0.7800      | 22.200      |
| 433.00      | 418.40      | 700.00      | 69.640       | 0.7750      | 0.7800      | 21.700      |
| 488.00      | 473.30      | 700.00      | 69.640       | 0.7700      | 0.7760      | 21.230      |
| 522.00      | 507.30      | 700.00      | 69.590       | 0.7690      | 0.7750      | 20.880      |
| 571.00      | 556.50      | 700.00      | 69.590       | 0.7690      | 0.7750      | 20.500      |
| 604.00      | 589.00      | 700.00      | 69.590       | 0.7660      | 0.7770      | 20.100      |
| 662.00      | 647.00      | 700.00      | 69.590       | 0.7750      | 0.7850      | 19.500      |
| 710.00      | 694.10      | 700.00      | 69.570       | 0.7740      | 0.7850      | 18.800      |
| 745.00      | 728.00      | 728.00      | 69.600       | 0.7780      | 0.7870      | 18.270      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 149.55      | 109.56      | 0.005 | 17.535         | 292.68       | 0.9997 | 4.426        |
| 177.13      | 123.35      | 0.006 | 17.549         | 292.92       | 0.9996 | 6.349        |
| 228.15      | 160.58      | 0.007 | 17.568         | 293.18       | 0.9995 | 9.084        |
| 273.71      | 193.73      | 0.009 | 17.569         | 293.06       | 0.9994 | 11.88        |
| 317.84      | 233.72      | 0.010 | 17.576         | 293.08       | 0.9994 | 13.83        |
| 370.24      | 289.70      | 0.012 | 17.585         | 293.08       | 0.9992 | 14.60        |
| 466.76      | 394.37      | 0.016 | 17.601         | 293.09       | 0.9990 | 16.03        |
| 577.08      | 508.13      | 0.020 | 17.621         | 293.13       | 0.9988 | 17.89        |
| 714.97      | 652.92      | 0.025 | 17.643         | 293.12       | 0.9985 | 19.17        |
| 839.13      | 777.08      | 0.030 | 17.663         | 293.13       | 0.9982 | 20.59        |
| 966.04      | 906.74      | 0.035 | 17.682         | 293.10       | 0.9979 | 21.84        |
| 1073.6      | 1014.3      | 0.038 | 17.697         | 293.04       | 0.9977 | 22.49        |
| 1197.7      | 1132.9      | 0.043 | 17.719         | 293.10       | 0.9975 | 23.14        |
| 1335.6      | 1270.8      | 0.048 | 17.743         | 293.13       | 0.9972 | 23.52        |
| 1470.7      | 1399.0      | 0.053 | 17.753         | 292.88       | 0.9970 | 24.08        |
| 1641.8      | 1571.4      | 0.059 | 17.787         | 293.02       | 0.9967 | 24.56        |
| 1818.3      | 1723.1      | 0.065 | 17.813         | 292.99       | 0.9964 | 25.43        |
| 1970.0      | 1872.1      | 0.071 | 17.839         | 292.99       | 0.9962 | 25.81        |
| 2163.0      | 2052.7      | 0.078 | 17.869         | 292.97       | 0.9959 | 26.71        |
| 2399.8      | 2292.3      | 0.087 | 17.905         | 292.86       | 0.9955 | 27.04        |
| 2755.6      | 2654.2      | 0.100 | 17.967         | 292.86       | 0.9950 | 27.55        |
| 3078.3      | 2977.6      | 0.112 | 18.027         | 292.89       | 0.9947 | 28.19        |
| 3457.5      | 3356.1      | 0.126 | 18.091         | 292.78       | 0.9943 | 28.81        |
| 3691.8      | 3590.5      | 0.135 | 18.134         | 292.76       | 0.9941 | 29.29        |
| 4029.7      | 3929.7      | 0.147 | 18.200         | 292.76       | 0.9939 | 29.84        |
| 4257.2      | 4153.8      | 0.156 | 18.244         | 292.75       | 0.9937 | 30.43        |
| 4657.1      | 4553.7      | 0.171 | 18.335         | 292.96       | 0.9937 | 31.37        |
| 4988.0      | 4878.4      | 0.183 | 18.404         | 292.94       | 0.9936 | 32.53        |
| 5229.4      | 5112.2      | 0.191 | 18.459         | 293.02       | 0.9936 | 33.48        |

CORE SAMPLE NO 4A RUN 5 REVERSE DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.22714

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 745.00      | 728.00      | 728.00      | 69.600       | 0.7780      | 0.7870      | 18.270      |
| 703.00      | 687.40      | 700.00      | 69.500       | 0.7800      | 0.7850      | 19.500      |
| 667.00      | 652.00      | 700.00      | 69.500       | 0.7790      | 0.7850      | 20.200      |
| 632.00      | 618.00      | 700.00      | 69.500       | 0.7750      | 0.7850      | 21.000      |
| 612.00      | 598.70      | 700.00      | 69.500       | 0.7750      | 0.7850      | 21.600      |
| 551.00      | 537.00      | 700.00      | 69.500       | 0.7720      | 0.7810      | 22.000      |
| 506.00      | 492.90      | 700.00      | 69.500       | 0.7720      | 0.7820      | 22.950      |
| 464.00      | 451.60      | 700.00      | 69.730       | 0.7780      | 0.7850      | 23.800      |
| 423.00      | 410.60      | 700.00      | 69.730       | 0.7790      | 0.7860      | 24.300      |
| 368.00      | 355.00      | 700.00      | 69.760       | 0.7810      | 0.7870      | 25.100      |
| 342.00      | 329.80      | 700.00      | 69.800       | 0.7820      | 0.7900      | 25.950      |
| 289.00      | 276.60      | 700.00      | 69.800       | 0.7850      | 0.7900      | 26.500      |
| 250.00      | 238.80      | 700.00      | 69.850       | 0.7870      | 0.7940      | 27.370      |
| 228.00      | 218.30      | 700.00      | 69.850       | 0.7860      | 0.7930      | 28.500      |
| 180.00      | 170.20      | 700.00      | 69.850       | 0.7850      | 0.7920      | 29.250      |
| 152.00      | 143.40      | 700.00      | 69.910       | 0.7860      | 0.7950      | 30.600      |
| 120.00      | 111.40      | 700.00      | 69.910       | 0.7870      | 0.7940      | 32.000      |
| 91.800      | 83.000      | 700.00      | 69.910       | 0.7890      | 0.7950      | 33.900      |
| 67.600      | 57.400      | 700.00      | 69.910       | 0.7900      | 0.7950      | 36.150      |
| 49.800      | 38.300      | 700.00      | 70.000       | 0.7900      | 0.7950      | 38.930      |
| 36.000      | 25.500      | 700.00      | 70.000       | 0.7900      | 0.7950      | 49.900      |
| 26.200      | 15.500      | 700.00      | 70.000       | 0.7900      | 0.7950      | 57.700      |
| 16.000      | 5.9000      | 700.00      | 70.000       | 0.7920      | 0.7950      | 73.630      |
| 10.000      | 2.6000      | 700.00      | 70.000       | 0.7930      | 0.7950      | 109.85      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 5229.4      | 5112.2      | 0.191 | 18.459         | 293.02       | 0.9936 | 33.48        |
| 4939.7      | 4832.1      | 0.181 | 18.397         | 293.02       | 0.9936 | 31.37        |
| 4691.5      | 4588.0      | 0.172 | 18.345         | 293.01       | 0.9937 | 30.28        |
| 4450.1      | 4353.6      | 0.163 | 18.293         | 292.96       | 0.9937 | 29.13        |
| 4312.3      | 4220.5      | 0.158 | 18.266         | 292.96       | 0.9938 | 28.32        |
| 3891.7      | 3795.1      | 0.142 | 18.178         | 292.87       | 0.9940 | 27.80        |
| 3581.4      | 3491.1      | 0.131 | 18.120         | 292.88       | 0.9942 | 26.65        |
| 3292.1      | 3206.6      | 0.120 | 18.072         | 292.99       | 0.9945 | 25.70        |
| 3009.4      | 2924.0      | 0.110 | 18.021         | 293.02       | 0.9948 | 25.17        |
| 2630.3      | 2540.6      | 0.096 | 17.955         | 293.06       | 0.9952 | 24.37        |
| 2451.1      | 2366.9      | 0.089 | 17.927         | 293.10       | 0.9955 | 23.57        |
| 2085.6      | 2000.1      | 0.075 | 17.866         | 293.14       | 0.9960 | 23.08        |
| 1816.8      | 1739.6      | 0.066 | 17.825         | 293.22       | 0.9964 | 22.35        |
| 1665.1      | 1598.3      | 0.060 | 17.799         | 293.19       | 0.9967 | 21.46        |
| 1334.2      | 1266.6      | 0.048 | 17.744         | 293.17       | 0.9973 | 20.91        |
| 1141.2      | 1081.9      | 0.041 | 17.716         | 293.22       | 0.9976 | 19.99        |
| 920.58      | 861.28      | 0.033 | 17.680         | 293.22       | 0.9980 | 19.11        |
| 726.14      | 665.47      | 0.026 | 17.651         | 293.25       | 0.9984 | 18.04        |
| 559.29      | 488.96      | 0.019 | 17.624         | 293.27       | 0.9988 | 16.92        |
| 436.68      | 357.39      | 0.015 | 17.604         | 293.27       | 0.9991 | 15.71        |
| 341.54      | 269.14      | 0.011 | 17.589         | 293.27       | 0.9993 | 12.26        |
| 273.97      | 200.19      | 0.009 | 17.579         | 293.27       | 0.9994 | 10.60        |
| 203.64      | 134.00      | 0.006 | 17.569         | 293.29       | 0.9996 | 8.307        |
| 162.27      | 111.25      | 0.005 | 17.565         | 293.30       | 0.9997 | 5.568        |

CORE SAMPLE NO 4A RUN 6 FORWARD DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.22714

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 8.5000      | 2.0000      | 800.00      | 71.050       | 0.7770      | 0.7730      | 150.15      |
| 15.000      | 5.4000      | 800.00      | 71.050       | 0.7800      | 0.7770      | 90.400      |
| 20.000      | 9.2000      | 800.00      | 71.050       | 0.7850      | 0.7800      | 70.100      |
| 26.000      | 14.800      | 800.00      | 71.050       | 0.7860      | 0.7840      | 59.300      |
| 33.000      | 21.400      | 800.00      | 71.050       | 0.7860      | 0.7840      | 50.200      |
| 43.000      | 30.800      | 800.00      | 71.050       | 0.7870      | 0.7840      | 41.700      |
| 52.000      | 40.500      | 800.00      | 71.050       | 0.7870      | 0.7890      | 38.600      |
| 62.000      | 50.800      | 800.00      | 71.050       | 0.7900      | 0.7900      | 35.900      |
| 74.200      | 63.600      | 800.00      | 71.050       | 0.7950      | 0.7930      | 34.800      |
| 90.000      | 80.000      | 800.00      | 71.050       | 0.7940      | 0.7940      | 33.400      |
| 100.00      | 90.000      | 800.00      | 71.050       | 0.7950      | 0.7950      | 32.150      |
| 120.40      | 111.00      | 800.00      | 71.050       | 0.7960      | 0.7960      | 31.000      |
| 140.00      | 130.00      | 800.00      | 71.050       | 0.7950      | 0.7990      | 29.200      |
| 170.00      | 160.70      | 800.00      | 71.050       | 0.7950      | 0.7990      | 28.650      |
| 200.00      | 190.50      | 800.00      | 71.050       | 0.7930      | 0.7990      | 27.770      |
| 230.00      | 220.30      | 800.00      | 70.980       | 0.8000      | 0.8000      | 26.900      |
| 260.00      | 250.00      | 800.00      | 70.940       | 0.8000      | 0.8000      | 26.200      |
| 289.60      | 278.90      | 800.00      | 70.940       | 0.8010      | 0.8050      | 25.600      |
| 329.60      | 318.00      | 800.00      | 70.940       | 0.8010      | 0.8030      | 25.000      |
| 364.80      | 352.10      | 800.00      | 70.940       | 0.8000      | 0.8050      | 24.400      |
| 416.00      | 402.80      | 800.00      | 70.940       | 0.7990      | 0.8050      | 23.700      |
| 464.00      | 450.80      | 800.00      | 70.940       | 0.8000      | 0.8050      | 23.300      |
| 520.00      | 507.50      | 800.00      | 70.940       | 0.8000      | 0.8030      | 22.900      |
| 569.00      | 555.20      | 800.00      | 70.480       | 0.8150      | 0.8200      | 22.200      |
| 615.00      | 601.40      | 800.00      | 70.480       | 0.8150      | 0.8200      | 21.750      |
| 669.00      | 654.70      | 800.00      | 70.440       | 0.8150      | 0.8190      | 21.200      |
| 725.00      | 711.10      | 800.00      | 70.430       | 0.8100      | 0.8200      | 20.800      |
| 767.00      | 752.00      | 800.00      | 70.430       | 0.8100      | 0.8200      | 20.100      |
| 807.00      | 792.00      | 800.00      | 70.430       | 0.8100      | 0.8200      | 19.750      |
| 836.00      | 819.00      | 820.00      | 70.430       | 0.8140      | 0.8210      | 18.800      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 153.33      | 108.51      | 0.005 | 17.542         | 292.83       | 0.9997 | 4.074        |
| 198.15      | 131.96      | 0.006 | 17.552         | 292.92       | 0.9996 | 6.766        |
| 232.62      | 158.16      | 0.007 | 17.561         | 293.02       | 0.9995 | 8.725        |
| 273.99      | 196.77      | 0.009 | 17.570         | 293.08       | 0.9994 | 10.31        |
| 322.25      | 242.27      | 0.010 | 17.577         | 293.08       | 0.9993 | 12.18        |
| 391.20      | 307.08      | 0.013 | 17.588         | 293.09       | 0.9992 | 14.67        |
| 453.25      | 373.96      | 0.015 | 17.601         | 293.15       | 0.9990 | 15.85        |
| 522.20      | 444.98      | 0.018 | 17.615         | 293.20       | 0.9989 | 17.04        |
| 606.32      | 533.23      | 0.021 | 17.633         | 293.30       | 0.9987 | 17.58        |
| 715.25      | 646.31      | 0.025 | 17.651         | 293.30       | 0.9985 | 18.31        |
| 784.20      | 715.25      | 0.028 | 17.663         | 293.33       | 0.9983 | 19.02        |
| 924.85      | 860.04      | 0.033 | 17.687         | 293.35       | 0.9980 | 19.73        |
| 1060.0      | 991.04      | 0.038 | 17.709         | 293.38       | 0.9978 | 20.95        |
| 1266.8      | 1202.7      | 0.045 | 17.743         | 293.38       | 0.9974 | 21.35        |
| 1473.7      | 1408.2      | 0.053 | 17.775         | 293.35       | 0.9970 | 22.03        |
| 1680.4      | 1613.5      | 0.061 | 17.814         | 293.45       | 0.9967 | 22.74        |
| 1887.2      | 1818.3      | 0.068 | 17.848         | 293.45       | 0.9964 | 23.35        |
| 2091.3      | 2017.5      | 0.076 | 17.885         | 293.52       | 0.9961 | 23.89        |
| 2367.1      | 2287.1      | 0.086 | 17.931         | 293.50       | 0.9957 | 24.47        |
| 2609.8      | 2522.2      | 0.095 | 17.973         | 293.51       | 0.9954 | 25.07        |
| 2962.8      | 2871.8      | 0.108 | 18.034         | 293.50       | 0.9950 | 25.81        |
| 3293.7      | 3202.7      | 0.120 | 18.095         | 293.51       | 0.9946 | 26.25        |
| 3679.9      | 3593.7      | 0.134 | 18.167         | 293.49       | 0.9943 | 26.71        |
| 4017.1      | 3921.9      | 0.146 | 18.249         | 293.88       | 0.9942 | 27.55        |
| 4334.2      | 4240.5      | 0.158 | 18.312         | 293.88       | 0.9941 | 28.12        |
| 4706.5      | 4607.9      | 0.172 | 18.388         | 293.87       | 0.9940 | 28.85        |
| 5092.6      | 4996.8      | 0.186 | 18.468         | 293.82       | 0.9939 | 29.41        |
| 5382.2      | 5278.8      | 0.197 | 18.531         | 293.82       | 0.9940 | 30.43        |
| 5658.0      | 5554.5      | 0.207 | 18.594         | 293.82       | 0.9940 | 30.97        |
| 5857.9      | 5740.7      | 0.214 | 18.642         | 293.88       | 0.9941 | 32.53        |

CORE SAMPLE NO 4A RUN 6 REVERSE DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.22714

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 836.00      | 819.00      | 820.00      | 70.430       | 0.8140      | 0.8210      | 18.800      |
| 814.00      | 798.20      | 800.00      | 70.440       | 0.8110      | 0.8190      | 19.400      |
| 786.00      | 773.00      | 800.00      | 70.440       | 0.8090      | 0.8190      | 20.500      |
| 747.00      | 734.30      | 800.00      | 70.440       | 0.8050      | 0.8150      | 20.900      |
| 694.00      | 682.00      | 800.00      | 70.440       | 0.8090      | 0.8150      | 21.500      |
| 652.00      | 640.50      | 800.00      | 70.440       | 0.8070      | 0.8140      | 22.000      |
| 606.00      | 595.20      | 800.00      | 70.440       | 0.8060      | 0.8140      | 22.680      |
| 557.00      | 546.10      | 800.00      | 70.440       | 0.8100      | 0.8150      | 23.000      |
| 514.00      | 504.00      | 800.00      | 70.440       | 0.8080      | 0.8140      | 23.850      |
| 454.00      | 444.40      | 800.00      | 69.700       | 0.8020      | 0.8130      | 25.000      |
| 402.00      | 394.00      | 800.00      | 69.700       | 0.8020      | 0.8100      | 26.800      |
| 346.00      | 338.50      | 800.00      | 69.700       | 0.8040      | 0.8100      | 28.000      |
| 313.40      | 306.30      | 800.00      | 69.670       | 0.8040      | 0.8100      | 29.200      |
| 273.60      | 266.50      | 800.00      | 69.620       | 0.8040      | 0.8070      | 30.700      |
| 222.60      | 215.20      | 800.00      | 69.620       | 0.8040      | 0.8080      | 31.900      |
| 174.40      | 167.30      | 800.00      | 69.620       | 0.8050      | 0.8100      | 34.450      |
| 139.30      | 132.00      | 800.00      | 70.360       | 0.7730      | 0.7760      | 36.800      |
| 100.00      | 91.400      | 800.00      | 70.360       | 0.7710      | 0.7760      | 38.600      |
| 60.000      | 48.100      | 800.00      | 70.400       | 0.7700      | 0.7760      | 40.600      |
| 32.000      | 17.000      | 800.00      | 70.460       | 0.7720      | 0.7760      | 47.400      |
| 20.000      | 7.1000      | 800.00      | 70.460       | 0.7750      | 0.7800      | 68.000      |
| 12.000      | 4.4000      | 800.00      | 70.460       | 0.7770      | 0.7790      | 127.20      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 5857.9      | 5740.7      | 0.214 | 18.642         | 293.88       | 0.9941 | 32.53        |
| 5706.2      | 5597.3      | 0.209 | 18.605         | 293.82       | 0.9941 | 31.53        |
| 5513.2      | 5423.6      | 0.202 | 18.561         | 293.80       | 0.9940 | 29.84        |
| 5244.3      | 5156.7      | 0.192 | 18.497         | 293.70       | 0.9939 | 29.27        |
| 4878.9      | 4796.1      | 0.179 | 18.420         | 293.75       | 0.9939 | 28.45        |
| 4589.3      | 4510.0      | 0.168 | 8.358          | 293.71       | 0.9940 | 27.80        |
| 4272.1      | 4197.7      | 0.156 | 18.293         | 293.70       | 0.9941 | 26.97        |
| 3934.3      | 3859.1      | 0.144 | 18.229         | 293.76       | 0.9942 | 26.59        |
| 3637.8      | 3568.9      | 0.133 | 18.171         | 293.72       | 0.9944 | 25.65        |
| 3223.1      | 3157.0      | 0.118 | 18.090         | 293.64       | 0.9947 | 24.47        |
| 2864.6      | 2809.5      | 0.105 | 18.024         | 293.60       | 0.9951 | 22.82        |
| 2478.5      | 2426.8      | 0.090 | 17.958         | 293.62       | 0.9955 | 21.84        |
| 2253.7      | 2204.7      | 0.082 | 17.919         | 293.62       | 0.9958 | 20.95        |
| 1979.2      | 1930.3      | 0.072 | 17.871         | 293.59       | 0.9962 | 19.92        |
| 1627.6      | 1576.6      | 0.059 | 17.813         | 293.60       | 0.9968 | 19.17        |
| 1295.3      | 1246.3      | 0.047 | 17.761         | 293.64       | 0.9974 | 17.75        |
| 1054.2      | 1003.9      | 0.038 | 17.684         | 292.82       | 0.9977 | 16.62        |
| 783.28      | 723.99      | 0.028 | 17.639         | 292.80       | 0.9983 | 15.85        |
| 507.54      | 425.50      | 0.017 | 17.593         | 292.78       | 0.9989 | 15.07        |
| 314.57      | 211.15      | 0.010 | 17.562         | 292.81       | 0.9994 | 12.90        |
| 231.83      | 142.89      | 0.007 | 17.554         | 292.89       | 0.9996 | 8.995        |
| 176.68      | 124.28      | 0.006 | 17.549         | 292.91       | 0.9996 | 4.809        |

CORE SAMPLE NO 4A RUN 7 FORWARD DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.22714

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 30.000      | .25500E-01  | 800.00      | 69.400       | 0.7870      | 0.7840      | 13002.      |
| 37.000      | .31600E-01  | 800.00      | 69.400       | 0.7860      | 0.7850      | 9860.0      |
| 41.000      | .47900E-01  | 800.00      | 69.400       | 0.7890      | 0.7870      | 8588.8      |
| 46.000      | .51900E-01  | 800.00      | 69.450       | 0.7895      | 0.7850      | 7398.5      |
| 49.700      | .59100E-01  | 800.00      | 69.400       | 0.7790      | 0.7860      | 6678.0      |
| 56.000      | .77200E-01  | 800.00      | 69.400       | 0.7890      | 0.7850      | 5665.0      |
| 65.000      | .97400E-01  | 800.00      | 69.440       | 0.7860      | 0.7860      | 4602.5      |
| 78.000      | .14680      | 800.00      | 69.500       | 0.7890      | 0.7860      | 3585.0      |
| 89.000      | .16640      | 800.00      | 69.500       | 0.7890      | 0.7860      | 2958.0      |
| 100.30      | .17280      | 800.00      | 69.500       | 0.7900      | 0.7860      | 2465.8      |
| 108.50      | .22000      | 800.00      | 69.730       | 0.7460      | 0.7500      | 2166.3      |
| 124.00      | .26000      | 800.00      | 69.730       | 0.7410      | 0.7460      | 1748.0      |
| 140.00      | .30350      | 800.00      | 69.730       | 0.7440      | 0.7450      | 1446.3      |
| 159.00      | .40110      | 800.00      | 69.730       | 0.7410      | 0.7450      | 1188.0      |
| 179.00      | .51670      | 800.00      | 69.730       | 0.7410      | 0.7450      | 988.92      |
| 201.00      | .68890      | 800.00      | 69.730       | 0.7410      | 0.7440      | 834.00      |
| 225.00      | .85780      | 800.00      | 69.730       | 0.7450      | 0.7410      | 698.50      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 299.37      | 92.701      | 0.007 | 17.564         | 293.09       | 0.9995 | 0.4704E-01   |
| 347.63      | 92.743      | 0.008 | 17.568         | 293.09       | 0.9995 | 0.6203E-01   |
| 375.21      | 92.856      | 0.009 | 17.573         | 293.15       | 0.9994 | 0.7121E-01   |
| 409.75      | 92.950      | 0.009 | 17.575         | 293.14       | 0.9994 | 0.8267E-01   |
| 435.19      | 92.933      | 0.010 | 17.572         | 293.02       | 0.9994 | 0.9159E-01   |
| 478.63      | 93.058      | 0.011 | 17.580         | 293.13       | 0.9993 | 0.1080       |
| 540.74      | 93.250      | 0.012 | 17.584         | 293.10       | 0.9993 | 0.1329       |
| 630.45      | 93.671      | 0.013 | 17.593         | 293.14       | 0.9992 | 0.1706       |
| 706.29      | 93.806      | 0.015 | 17.599         | 293.14       | 0.9991 | 0.2068       |
| 784.20      | 93.850      | 0.016 | 17.605         | 293.15       | 0.9990 | 0.2481       |
| 841.05      | 94.482      | 0.017 | 17.565         | 292.17       | 0.9989 | 0.2823       |
| 947.92      | 94.758      | 0.019 | 17.568         | 292.05       | 0.9988 | 0.3499       |
| 1058.2      | 95.058      | 0.021 | 17.578         | 292.08       | 0.9986 | 0.4229       |
| 1189.2      | 95.731      | 0.024 | 17.587         | 292.04       | 0.9985 | 0.5149       |
| 1327.1      | 96.528      | 0.026 | 17.598         | 292.04       | 0.9983 | 0.6185       |
| 1478.8      | 97.715      | 0.029 | 17.610         | 292.03       | 0.9982 | 0.7334       |
| 1644.3      | 98.880      | 0.032 | 17.624         | 292.04       | 0.9980 | 0.8757       |

CORE SAMPLE NO 4A RUN 8 FORWARD DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.22714

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 20.000      | .17500E-01  | 800.00      | 69.400       | 0.7880      | 0.7840      | 15258.      |
| 26.000      | .21600E-01  | 800.00      | 69.400       | 0.7860      | 0.7840      | 11590.      |
| 34.000      | .20160      | 800.00      | 69.400       | 0.7860      | 0.7850      | 8868.0      |
| 42.000      | .11000E-01  | 800.00      | 69.400       | 0.7890      | 0.7860      | 7133.0      |
| 48.000      | .18000E-01  | 800.00      | 69.400       | 0.7890      | 0.7850      | 6144.0      |
| 56.000      | .77400E-01  | 800.00      | 69.440       | 0.7860      | 0.7860      | 5333.5      |
| 66.000      | .16200E-01  | 800.00      | 69.440       | 0.7900      | 0.7900      | 4498.0      |
| 76.000      | .50400E-01  | 800.00      | 69.500       | 0.7910      | 0.7860      | 3994.0      |
| 84.000      | .10080      | 800.00      | 69.500       | 0.7890      | 0.7860      | 3584.0      |
| 92.000      | .10440      | 800.00      | 69.500       | 0.7890      | 0.7860      | 3188.0      |
| 100.00      | .82800E-01  | 800.00      | 69.500       | 0.7900      | 0.7860      | 2898.8      |
| 110.00      | .20600      | 800.00      | 69.730       | 0.7460      | 0.7500      | 2597.3      |
| 124.00      | .20600      | 800.00      | 69.730       | 0.7410      | 0.7460      | 2230.0      |
| 140.00      | .30350      | 800.00      | 69.730       | 0.7440      | 0.7450      | 1795.3      |
| 160.00      | .40110      | 800.00      | 69.730       | 0.7410      | 0.7450      | 1375.0      |
| 180.00      | .51670      | 800.00      | 69.730       | 0.7410      | 0.7450      | 967.92      |
| 200.00      | .75890      | 800.00      | 69.730       | 0.7410      | 0.7440      | 654.00      |
| 226.00      | .95760      | 800.00      | 69.730       | 0.7450      | 0.7410      | 363.50      |
| 230.00      | .98770      | 800.00      | 69.730       | 0.7400      | 0.7410      | 218.50      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 230.42      | 92.646      | 0.006 | 17.559         | 293.10       | 0.9996 | 0.4009E-01   |
| 271.79      | 92.674      | 0.007 | 17.562         | 293.08       | 0.9996 | 0.5277E-01   |
| 326.95      | 93.915      | 0.008 | 17.567         | 293.09       | 0.9995 | 0.6897E-01   |
| 382.11      | 92.601      | 0.009 | 17.573         | 293.14       | 0.9994 | 0.8575E-01   |
| 423.47      | 92.650      | 0.009 | 17.576         | 293.13       | 0.9994 | 0.9955E-01   |
| 478.69      | 93.112      | 0.011 | 17.579         | 293.10       | 0.9993 | 0.1147       |
| 547.63      | 92.690      | 0.012 | 17.589         | 293.20       | 0.9993 | 0.1360       |
| 616.66      | 93.006      | 0.013 | 17.593         | 293.17       | 0.9992 | 0.1531       |
| 671.82      | 93.354      | 0.014 | 17.596         | 293.14       | 0.9991 | 0.1707       |
| 726.98      | 93.379      | 0.015 | 17.600         | 293.14       | 0.9991 | 0.1919       |
| 782.13      | 93.230      | 0.016 | 17.605         | 293.15       | 0.9990 | 0.2110       |
| 851.39      | 94.386      | 0.017 | 17.566         | 292.17       | 0.9989 | 0.2355       |
| 947.92      | 94.386      | 0.019 | 17.568         | 292.05       | 0.9988 | 0.2743       |
| 1058.2      | 95.058      | 0.021 | 17.578         | 292.08       | 0.9986 | 0.3407       |
| 1196.1      | 95.731      | 0.024 | 17.588         | 292.04       | 0.9985 | 0.4448       |
| 1334.0      | 96.528      | 0.026 | 17.599         | 292.04       | 0.9983 | 0.6319       |
| 1471.9      | 98.198      | 0.029 | 17.609         | 292.03       | 0.9982 | 0.9352       |
| 1651.2      | 99.568      | 0.032 | 17.624         | 292.04       | 0.9980 | 1.683        |
| 1678.8      | 99.775      | 0.033 | 17.624         | 291.98       | 0.9979 | 2.799        |

CORE SAMPLE NO 4A RUN 9 REVERSE DIRECTION  
 RE=0.15256 RW=0.00820 H=0.02600 POROSITY=0.22714

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 292.00      | .49100E-01  | 700.00      | 69.780       | 0.8630      | 0.8600      | 23.500      |
| 286.00      | .49100E-01  | 700.00      | 69.780       | 0.8630      | 0.8600      | 25.400      |
| 280.00      | .49100E-01  | 700.00      | 69.780       | 0.8630      | 0.8600      | 28.100      |
| 274.00      | .98200E-01  | 700.00      | 69.780       | 0.8630      | 0.8580      | 33.500      |
| 264.00      | .49100E-01  | 700.00      | 69.780       | 0.8620      | 0.8580      | 41.400      |
| 252.00      | .49100E-01  | 700.00      | 69.780       | 0.8600      | 0.8580      | 61.100      |
| 240.00      | .49100E-01  | 700.00      | 69.780       | 0.8600      | 0.8580      | 80.600      |
| 224.00      | .98200E-01  | 700.00      | 69.780       | 0.8600      | 0.8550      | 107.18      |
| 212.00      | .49100E-01  | 700.00      | 69.780       | 0.8590      | 0.8550      | 122.35      |
| 200.00      | .49100E-01  | 700.00      | 70.120       | 0.8150      | 0.8140      | 137.15      |
| 180.00      | .49100E-01  | 700.00      | 70.140       | 0.8340      | 0.8340      | 159.50      |
| 160.00      | .98200E-01  | 700.00      | 70.140       | 0.8400      | 0.8430      | 189.20      |
| 140.00      | .49100E-01  | 700.00      | 70.140       | 0.8470      | 0.8450      | 221.30      |
| 120.00      | .98200E-01  | 700.00      | 70.140       | 0.8530      | 0.8530      | 268.10      |
| 100.00      | .73700E-01  | 700.00      | 70.140       | 0.8530      | 0.8500      | 330.10      |
| 80.000      | .49100E-01  | 700.00      | 70.140       | 0.8560      | 0.8530      | 424.60      |
| 60.000      | .98200E-01  | 700.00      | 70.140       | 0.8590      | 0.8590      | 576.90      |
| 40.000      | .49100E-01  | 700.00      | 70.140       | 0.8600      | 0.8570      | 884.00      |
| 20.000      | .49100E-01  | 700.00      | 70.140       | 0.8600      | 0.8560      | 1830.5      |

| PE<br>(KPA) | PW<br>(KPA) | RD <sub>c</sub> | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-----------------|----------------|--------------|--------|--------------|
| 2106.3      | 93.371      | 0.040           | 17.794         | 294.97       | 0.9978 | 26.03        |
| 2064.9      | 93.371      | 0.040           | 17.790         | 294.97       | 0.9978 | 24.08        |
| 2023.6      | 93.371      | 0.039           | 17.787         | 294.97       | 0.9979 | 21.77        |
| 1982.2      | 93.709      | 0.038           | 17.782         | 294.95       | 0.9979 | 18.26        |
| 1913.2      | 93.371      | 0.037           | 17.776         | 294.93       | 0.9980 | 14.77        |
| 1830.5      | 94.217      | 0.035           | 17.769         | 294.91       | 0.9980 | 10.01        |
| 1747.8      | 93.709      | 0.034           | 17.762         | 294.91       | 0.9981 | 7.589        |
| 1637.5      | 93.709      | 0.032           | 17.751         | 294.87       | 0.9982 | 5.707        |
| 1554.7      | 93.371      | 0.030           | 17.744         | 294.86       | 0.9983 | 4.999        |
| 1472.4      | 95.179      | 0.029           | 17.690         | 293.81       | 0.9983 | 4.460        |
| 1334.6      | 94.528      | 0.026           | 17.701         | 294.29       | 0.9985 | 3.835        |
| 1196.7      | 94.189      | 0.024           | 17.698         | 294.48       | 0.9986 | 3.233        |
| 1058.8      | 93.851      | 0.021           | 17.692         | 294.59       | 0.9988 | 2.764        |
| 920.88      | 94.189      | 0.019           | 17.689         | 294.76       | 0.9989 | 2.281        |
| 782.99      | 94.020      | 0.016           | 17.677         | 294.72       | 0.9990 | 1.853        |
| 645.09      | 93.851      | 0.014           | 17.669         | 294.80       | 0.9992 | 1.441        |
| 507.20      | 94.189      | 0.011           | 17.663         | 294.91       | 0.9993 | 1.060        |
| 369.30      | 93.851      | 0.008           | 17.652         | 294.90       | 0.9995 | 0.6919       |
| 231.41      | 93.851      | 0.006           | 17.641         | 294.88       | 0.9996 | 0.3341       |

CORE SAMPLE NO 2 RUN 1 FORWARD DIRECTION  
 RE=0.15225 RW=0.00815 H=0.02803 POROSITY=0.23432

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 10.000      | .30000      | 600.00      | 69.930       | 0.8150      | 0.8390      | 539.00      |
| 20.000      | 5.4000      | 600.00      | 69.930       | 0.8140      | 0.8450      | 285.10      |
| 30.000      | 6.2000      | 600.00      | 69.930       | 0.8180      | 0.8450      | 154.50      |
| 40.000      | 13.000      | 600.00      | 69.930       | 0.8190      | 0.8500      | 114.37      |
| 60.000      | 38.900      | 600.00      | 69.930       | 0.8220      | 0.8500      | 100.65      |
| 80.000      | 60.800      | 600.00      | 69.930       | 0.8250      | 0.8520      | 90.000      |
| 110.00      | 91.800      | 600.00      | 69.930       | 0.8260      | 0.8520      | 77.530      |
| 145.00      | 127.60      | 600.00      | 69.900       | 0.8310      | 0.8540      | 67.600      |
| 179.00      | 159.00      | 600.00      | 69.900       | 0.8350      | 0.8540      | 56.300      |
| 208.00      | 186.80      | 600.00      | 69.900       | 0.8370      | 0.8550      | 50.400      |
| 252.00      | 231.20      | 600.00      | 69.900       | 0.8380      | 0.8550      | 46.500      |
| 297.00      | 276.30      | 600.00      | 69.900       | 0.8440      | 0.8530      | 43.000      |
| 348.00      | 327.50      | 600.00      | 69.900       | 0.8440      | 0.8530      | 39.800      |
| 368.00      | 342.50      | 600.00      | 69.500       | 0.8230      | 0.8410      | 34.200      |
| 393.00      | 367.00      | 600.00      | 69.880       | 0.8460      | 0.8550      | 32.450      |
| 413.00      | 386.00      | 600.00      | 69.500       | 0.8260      | 0.8480      | 29.730      |
| 456.00      | 430.00      | 600.00      | 69.500       | 0.8260      | 0.8550      | 26.750      |
| 511.00      | 487.00      | 600.00      | 69.500       | 0.8300      | 0.8550      | 23.660      |
| 551.00      | 528.00      | 600.00      | 69.500       | 0.8330      | 0.8530      | 22.230      |
| 592.00      | 570.00      | 600.00      | 69.500       | 0.8360      | 0.8500      | 20.000      |
| 617.00      | 596.00      | 604.00      | 69.500       | 0.8360      | 0.8480      | 17.730      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 162.18      | 95.301      | 0.005 | 17.600         | 294.12       | 0.9997 | 1.135        |
| 231.13      | 130.46      | 0.007 | 17.611         | 294.18       | 0.9996 | 2.145        |
| 300.07      | 135.98      | 0.008 | 17.619         | 294.23       | 0.9995 | 3.959        |
| 369.02      | 182.86      | 0.010 | 17.632         | 294.30       | 0.9994 | 5.348        |
| 506.92      | 361.44      | 0.016 | 17.659         | 294.34       | 0.9990 | 6.077        |
| 644.81      | 512.43      | 0.021 | 17.684         | 294.40       | 0.9987 | 6.796        |
| 851.66      | 726.17      | 0.029 | 17.718         | 294.41       | 0.9983 | 7.889        |
| 1092.9      | 972.96      | 0.038 | 17.761         | 294.50       | 0.9979 | 9.048        |
| 1327.4      | 1189.5      | 0.046 | 17.800         | 294.55       | 0.9975 | 10.86        |
| 1527.3      | 1381.1      | 0.053 | 17.834         | 294.59       | 0.9972 | 12.14        |
| 1830.7      | 1687.3      | 0.065 | 17.885         | 294.60       | 0.9967 | 13.15        |
| 2140.9      | 1998.2      | 0.076 | 17.939         | 294.65       | 0.9962 | 14.22        |
| 2492.6      | 2351.2      | 0.089 | 17.989         | 294.65       | 0.9958 | 15.37        |
| 2629.9      | 2454.1      | 0.094 | 18.002         | 294.24       | 0.9956 | 17.88        |
| 2802.8      | 2623.5      | 0.100 | 18.052         | 294.70       | 0.9955 | 18.85        |
| 2940.2      | 2754.0      | 0.105 | 18.061         | 294.36       | 0.9952 | 20.57        |
| 3236.7      | 3057.4      | 0.116 | 18.119         | 294.45       | 0.9950 | 22.87        |
| 3615.9      | 3450.4      | 0.130 | 18.193         | 294.50       | 0.9947 | 25.85        |
| 3891.7      | 3733.1      | 0.140 | 18.247         | 294.51       | 0.9945 | 27.51        |
| 4174.4      | 4022.7      | 0.151 | 18.303         | 294.51       | 0.9944 | 30.58        |
| 4346.7      | 4201.9      | 0.157 | 18.337         | 294.49       | 0.9943 | 34.50        |

CORE SAMPLE NO 2 RUN 2 REVERSE DIRECTION  
 RE=0.15225 RW=0.00815 H=0.02803 POROSITY=0.23432

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 631.00      | 608.00      | 605.00      | 69.840       | 0.7860      | 0.7650      | 16.820      |
| 609.00      | 583.00      | 600.00      | 69.840       | 0.7870      | 0.7600      | 18.200      |
| 557.00      | 528.00      | 600.00      | 69.840       | 0.7850      | 0.7680      | 20.240      |
| 516.00      | 486.00      | 600.00      | 69.840       | 0.7890      | 0.7650      | 21.780      |
| 477.00      | 448.00      | 600.00      | 69.840       | 0.7890      | 0.7690      | 24.360      |
| 435.00      | 405.00      | 600.00      | 69.840       | 0.7900      | 0.7700      | 25.600      |
| 393.00      | 361.60      | 600.00      | 70.000       | 0.7900      | 0.7700      | 26.950      |
| 340.00      | 306.00      | 600.00      | 70.000       | 0.7900      | 0.7700      | 28.280      |
| 300.00      | 263.70      | 600.00      | 70.000       | 0.7900      | 0.7730      | 29.680      |
| 270.00      | 234.40      | 600.00      | 70.080       | 0.7910      | 0.7670      | 32.600      |
| 235.00      | 201.00      | 600.00      | 70.080       | 0.7900      | 0.7640      | 37.300      |
| 200.00      | 171.00      | 600.00      | 70.140       | 0.7910      | 0.7630      | 48.030      |
| 176.50      | 147.00      | 600.00      | 70.140       | 0.7910      | 0.7560      | 52.400      |
| 147.00      | 115.00      | 600.00      | 70.150       | 0.7910      | 0.7550      | 56.500      |
| 113.00      | 80.200      | 600.00      | 70.160       | 0.7910      | 0.7530      | 69.220      |
| 90.400      | 56.000      | 600.00      | 70.160       | 0.7900      | 0.7530      | 81.150      |
| 63.600      | 32.000      | 600.00      | 70.230       | 0.7910      | 0.7540      | 118.88      |
| 45.800      | 5.4800      | 600.00      | 70.340       | 0.7870      | 0.7510      | 141.80      |
| 25.800      | 1.3000      | 600.00      | 70.340       | 0.7860      | 0.7550      | 313.60      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QD<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 4443.7      | 4285.1      | 0.162 | 18.280         | 292.84       | 0.9937 | 36.36        |
| 4292.0      | 4112.8      | 0.156 | 18.246         | 292.80       | 0.9937 | 33.61        |
| 3933.5      | 3733.5      | 0.142 | 18.176         | 292.87       | 0.9940 | 30.22        |
| 3650.8      | 3444.0      | 0.131 | 18.122         | 292.88       | 0.9942 | 28.08        |
| 3381.9      | 3182.0      | 0.121 | 18.075         | 292.93       | 0.9944 | 25.11        |
| 3092.3      | 2885.5      | 0.111 | 18.022         | 292.96       | 0.9947 | 23.89        |
| 2803.0      | 2586.5      | 0.100 | 17.970         | 292.96       | 0.9951 | 22.70        |
| 2437.5      | 2203.1      | 0.086 | 17.905         | 292.96       | 0.9956 | 21.63        |
| 2161.8      | 1911.5      | 0.075 | 17.858         | 292.99       | 0.9960 | 20.61        |
| 1955.0      | 1709.6      | 0.068 | 17.821         | 292.93       | 0.9963 | 18.76        |
| 1713.7      | 1479.3      | 0.059 | 17.780         | 292.88       | 0.9967 | 16.40        |
| 1472.5      | 1272.5      | 0.051 | 17.743         | 292.88       | 0.9971 | 12.73        |
| 1310.4      | 1107.0      | 0.045 | 17.712         | 292.80       | 0.9974 | 11.67        |
| 1107.1      | 886.42      | 0.037 | 17.677         | 292.78       | 0.9978 | 10.83        |
| 872.65      | 646.50      | 0.028 | 17.638         | 292.76       | 0.9983 | 8.836        |
| 716.82      | 479.65      | 0.022 | 17.612         | 292.75       | 0.9986 | 7.537        |
| 532.14      | 314.26      | 0.016 | 17.586         | 292.77       | 0.9990 | 5.145        |
| 409.56      | 131.56      | 0.010 | 17.557         | 292.68       | 0.9994 | 4.313        |
| 271.66      | 102.74      | 0.007 | 17.546         | 292.72       | 0.9995 | 1.950        |

CORE SAMPLE NO 2 RUN 3 FORWARD DIRECTION  
 RE=0.15225 RW=0.00815 H=0.02803 POROSITY=0.23432

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 10.000      | .25000E-01  | 600.00      | 71.040       | 0.7630      | 0.7710      | 1419.1      |
| 20.000      | .25000E-01  | 600.00      | 71.040       | 0.7650      | 0.7710      | 617.90      |
| 30.000      | .44000E-01  | 600.00      | 71.040       | 0.7650      | 0.7750      | 370.40      |
| 42.000      | .12300      | 600.00      | 71.040       | 0.7650      | 0.7710      | 243.10      |
| 54.000      | .37000E-01  | 600.00      | 71.040       | 0.7670      | 0.7710      | 175.20      |
| 66.000      | .61000E-01  | 600.00      | 71.040       | 0.7670      | 0.7690      | 136.25      |
| 82.000      | .10000      | 600.00      | 71.040       | 0.7660      | 0.7600      | 103.30      |
| 100.00      | .74000E-01  | 600.00      | 71.040       | 0.7660      | 0.7700      | 80.550      |
| 120.00      | .12000      | 600.00      | 71.040       | 0.7660      | 0.7660      | 63.600      |
| 140.00      | .49000E-01  | 600.00      | 71.040       | 0.7660      | 0.7630      | 53.400      |
| 160.00      | .25000E-01  | 600.00      | 71.040       | 0.7660      | 0.7660      | 45.300      |
| 180.00      | .25000E-01  | 600.00      | 71.040       | 0.7660      | 0.7670      | 39.530      |
| 200.00      | .25000E-01  | 600.00      | 71.040       | 0.7660      | 0.7590      | 34.530      |
| 220.00      | .25000E-01  | 600.00      | 71.040       | 0.7660      | 0.7590      | 31.000      |
| 240.00      | .25000E-01  | 600.00      | 71.040       | 0.7650      | 0.7640      | 28.300      |
| 260.00      | .25000E-01  | 600.00      | 71.040       | 0.7650      | 0.7570      | 25.880      |
| 280.00      | .25000      | 600.00      | 71.040       | 0.7650      | 0.7560      | 23.820      |
| 300.00      | .54000      | 600.00      | 71.040       | 0.7640      | 0.7560      | 21.860      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VIS<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|----------------|--------------|--------|--------------|
| 163.66      | 94.884      | 0.005 | 17.533         | 292.63       | 0.9997 | 0.4310       |
| 232.61      | 94.884      | 0.006 | 17.540         | 292.66       | 0.9996 | 0.9899       |
| 301.55      | 95.015      | 0.007 | 17.547         | 292.71       | 0.9995 | 1.651        |
| 384.29      | 95.560      | 0.009 | 17.552         | 292.66       | 0.9994 | 2.516        |
| 467.03      | 94.967      | 0.010 | 17.559         | 292.68       | 0.9993 | 3.491        |
| 549.77      | 95.133      | 0.012 | 17.565         | 292.66       | 0.9992 | 4.489        |
| 660.08      | 95.401      | 0.014 | 17.568         | 292.54       | 0.9991 | 5.921        |
| 784.19      | 95.222      | 0.016 | 17.583         | 292.66       | 0.9990 | 7.593        |
| 922.08      | 95.539      | 0.019 | 17.592         | 292.61       | 0.9988 | 9.617        |
| 1060.0      | 95.050      | 0.021 | 17.601         | 292.57       | 0.9987 | 11.45        |
| 1197.9      | 94.884      | 0.024 | 17.614         | 292.61       | 0.9985 | 13.50        |
| 1335.8      | 94.884      | 0.026 | 17.625         | 292.62       | 0.9984 | 15.47        |
| 1473.7      | 94.884      | 0.029 | 17.632         | 292.52       | 0.9982 | 17.71        |
| 1611.6      | 94.884      | 0.032 | 17.643         | 292.52       | 0.9981 | 19.73        |
| 1749.5      | 94.884      | 0.034 | 17.656         | 292.57       | 0.9979 | 21.61        |
| 1887.3      | 94.884      | 0.037 | 17.663         | 292.49       | 0.9978 | 23.63        |
| 2025.2      | 96.436      | 0.039 | 17.674         | 292.47       | 0.9976 | 25.68        |
| 2163.1      | 98.435      | 0.042 | 17.684         | 292.46       | 0.9975 | 27.98        |

CORE SAMPLE NO 2 RUN 4 REVERSE DIRECTION  
 RE=0.15225 RW=0.00815 H=0.02803 POROSITY=0.23432

| PE<br>(PSI) | PW<br>(PSI) | CP<br>(PSI) | PB<br>(CMHG) | TCE<br>(MV) | TCW<br>(MV) | TS<br>(SEC) |
|-------------|-------------|-------------|--------------|-------------|-------------|-------------|
| 300.00      | .22000      | 600.00      | 70.820       | 0.7530      | 0.7610      | 19.500      |
| 280.00      | .12000      | 600.00      | 70.820       | 0.7540      | 0.7610      | 21.530      |
| 260.00      | .74000E-01  | 600.00      | 70.820       | 0.7550      | 0.7660      | 24.350      |
| 240.00      | .74000E-01  | 600.00      | 70.820       | 0.7560      | 0.7660      | 27.250      |
| 220.00      | .61000E-01  | 600.00      | 70.820       | 0.7600      | 0.7630      | 30.600      |
| 200.00      | .49000E-01  | 600.00      | 70.820       | 0.7600      | 0.7610      | 35.340      |
| 180.00      | .49000E-01  | 600.00      | 70.810       | 0.7710      | 0.7730      | 41.170      |
| 160.00      | .27000      | 600.00      | 70.810       | 0.7740      | 0.7760      | 48.830      |
| 140.00      | .49000E-01  | 600.00      | 70.810       | 0.7740      | 0.7770      | 58.370      |
| 120.00      | .49000E-01  | 600.00      | 70.810       | 0.7770      | 0.7750      | 71.930      |
| 100.00      | .19600      | 600.00      | 70.810       | 0.7770      | 0.7750      | 88.670      |
| 80.000      | .49000E-01  | 600.00      | 70.810       | 0.7800      | 0.7790      | 113.30      |
| 60.000      | .25000E-01  | 600.00      | 70.810       | 0.7800      | 0.7800      | 16.80       |
| 40.000      | .25000E-01  | 600.00      | 70.810       | 0.7840      | 0.7760      | 255.20      |
| 30.000      | .25000E-01  | 600.00      | 70.810       | 0.7840      | 0.7790      | 360.28      |
| 20.000      | .25000E-01  | 600.00      | 70.810       | 0.7870      | 0.7840      | 555.20      |
| 10.000      | .98000E-01  | 600.00      | 70.810       | 0.7890      | 0.7860      | 1240.8      |

| PE<br>(KPA) | PW<br>(KPA) | RD    | VISA<br>(U-PAS) | T<br>(DEG K) | ZA     | QO<br>(M3/D) |
|-------------|-------------|-------|-----------------|--------------|--------|--------------|
| 2162.8      | 95.935      | 0.042 | 17.681          | 292.39       | 0.9975 | 31.37        |
| 2025.0      | 95.246      | 0.039 | 17.670          | 292.40       | 0.9976 | 28.41        |
| 1887.1      | 94.929      | 0.037 | 17.662          | 292.47       | 0.9978 | 25.12        |
| 1749.2      | 94.929      | 0.034 | 17.652          | 292.49       | 0.9979 | 22.45        |
| 1611.3      | 94.839      | 0.032 | 17.641          | 292.50       | 0.9981 | 19.99        |
| 1473.4      | 94.756      | 0.029 | 17.629          | 292.47       | 0.9982 | 17.31        |
| 1335.5      | 94.743      | 0.026 | 17.631          | 292.76       | 0.9984 | 14.86        |
| 1197.6      | 96.267      | 0.024 | 17.624          | 292.83       | 0.9985 | 12.53        |
| 1059.7      | 94.743      | 0.021 | 17.613          | 292.84       | 0.9987 | 10.48        |
| 921.78      | 94.743      | 0.019 | 17.603          | 292.86       | 0.9988 | 8.503        |
| 783.88      | 95.757      | 0.016 | 17.592          | 292.86       | 0.9990 | 6.898        |
| 645.99      | 94.743      | 0.014 | 17.585          | 292.94       | 0.9991 | 5.398        |
| 508.09      | 94.578      | 0.011 | 17.575          | 292.96       | 0.9993 | 3.780        |
| 370.20      | 94.578      | 0.009 | 17.564          | 292.96       | 0.9994 | 2.397        |
| 301.25      | 94.578      | 0.007 | 17.560          | 292.99       | 0.9995 | 1.698        |
| 232.30      | 94.578      | 0.006 | 17.559          | 293.09       | 0.9996 | 1.102        |
| 163.35      | 95.081      | 0.005 | 17.556          | 293.14       | 0.9997 | 0.4929       |

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