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Behaviour of Steel Columns Reinforced with Welded Steel Plates

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Abstract

Current design criteria for steel columns reinforced with welded steel plates, usually based on the SSRC column curve 2, have not been verified. Considering the complexity of different influencing factors of the reinforced columns, the use of current design criteria may not be appropriate. A better understanding of the parameters associated with reinforced columns is therefore required.

A parametric study, using 317 finite element models of reinforced steel columns with varying parameters, was conducted. The study showed that column slenderness and initial out-of-straightness remain the important factors for reinforced columns. The interactions of the orientation of the reinforcing plates and the buckling direction were observed to affect the strength of reinforced columns. These observations require further experimental confirmation. A detailed statistical analysis was then conducted to determine the factored resistance of reinforced columns and to evaluate their performances. The results showed that the current design approach of using the SSRC column curve 2 is appropriate for use with a resistance factor of 0.9.

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List of Symbols

Note: Symbols appearing in the text with and without a bar, e.g., A and A, denote the mean and nominal values, respectively.

- A Gross area of the cross section
- A_C Area of the rolled section
- A_P Area of the cover plates
- B Buckling axis of the reinforced column
- b_f Flange width of the rolled section
- b_p Plate width
- CF_v Yield strength of the rolled section columns
- C_r Factored compressive resistance of steel columns
- D Direction of reinforcing plates
- d Depth of the rolled section
- E Modulus of elasticity of steel
- E_t Tangent modulus of steel
- F The stress defined by an expression of F_v and λ
- F_e Elastic buckling stress
- F_v Yield strength of steel
- $f(\lambda)$ A function of the non-dimension slenderness ratio, λ
- G Geometric properties
- g Side of the fillet welds connecting the plates to the rolled section
- IRS Initial residual stresses in the rolled section and the cover plates before reinforcing
- I_x , I_y Moment of inertia of cross section about principal x and y axes
- K Effective length factor
- k Side of the fillet connecting the flanges to the web in the rolled section
- L The column length
- MF Maximum magnitude of the residual stress at the flange tips
- MP Maximum magnitude of the residual stress at the reinforced plate edges
- n Coefficient in the equation for C_r in the Clause 13.3.1 of CSA/CAN-S16.1-94
- P External axial load of the column

- P₀ Preload before reinforcing
- P_i Functions of the non-dimension slenderness ratio, λ (i = 1, 2, 3)
- P_{cr} Column strength
- P_{exp} Load carrying capacity of the reinforced column obtained from experiment
- P_{fea} Load carrying capacity of the reinforced column obtained from finite element analysis
- PF_v Yield strength of the cover plates
- P_{r1} Load carrying capacity of the reinforced column predicted using SSRC curve 1
- P_{r2} Load carrying capacity of the reinforced column predicted using SSRC curve 2
- P_{rc1} Load carrying capacity of the reinforced column predicted using CSA curve 1
- P_{rc2} Load carrying capacity of the reinforced column predicted using CSA curve 2
- P_{rv} Yield strength of the reinforced column
- PS Designation of the initial residual stress pattern before reinforcing
- P_{u2} Load carrying capacity of the unreinforced column predicted using the SSRC column curve 2
- P_{uy} Yield strength of the unreinforced rolled section column
- R The resistance
- r Radius of gyration of the cross section
- r_x , r_y Radius of gyration of the cross section about principal x and y axes
- S The load effect
- t_f Thickness of the flanges in the rolled section
- t_p Thickness of the cover plates
- V_A Coefficient of variation of the area
- V_{CFv} Coefficient of variation of the yield strength of the rolled section columns
- V_{Cr} Coefficient of variation of the factored compressive resistance of steel columns
- V_E Coefficient of variation of the modulus of elasticity
- V_F Coefficient of variation of the stress defined as a function of F_y and λ
- V_{Fv} Coefficient of variation of the yield stress
- V_G Coefficient of variation of the relevant geometric property
- V_I Coefficient of variation of the moment of inertia
- V_M Coefficient of variation of the relevant material property

 V_{P} Coefficient of variation of the ratio of the strength of a column obtained from test to the strength predicted using the design equation V_{PFv} Coefficient of variation of the yield strength of the cover plates V_R Coefficient of variation of the resistance V_r Coefficient of variation of the radius of gyration V_{ς} Coefficient of variation of the load effect Thickness of the web w Separation factor α α' The load factor Separation factor for the resistance $\alpha_{\mathbf{R}}$ Separation factor for the load effect α_{s} Coefficient of thermal expansion in the longitudinal direction of the steel column α_{x} β Safety index Δτ Temperature change in the steel column δ_0 Initial imperfection of the unreinforced column δ_{i} Initial out-of-straightness of the reinforced column λ Non-dimensional slenderness parameter Measured-to-nominal ratio of the area ρ_{A} Measured-to-nominal ratio of the yield strength of rolled section columns ρ_{CFv} ρ_{Cr} Measured-to-nominal ratio of the factored compressive resistance of steel columns Measured-to-nominal ratio of the modulus of elasticity ρ_{E} Measured-to-nominal experimental ratio ρ_{ex} Measured-to-nominal ratio of the stress defined as a function of F_v and λ ρ_{F} Measured-to-nominal ratio of the yield strength ρ_{Fv} Measured-to-nominal ratio of $f(\lambda)$ $\rho_{f(\lambda)}$ Measure-to-nominal ratio of the relevant geometric property ρ_{G} Measured-to-nominal ratio of the moment of inertia $\rho_{\rm I}$ Measure-to-nominal ratio of the relevant material property ρм Normalized professional ratio ρ_n

- ρ_p Ratio of the strength of a column obtained from test to the strength predicted by the design equation
- ρ_{PFy} Measured-to-nominal ratio of the yield strength of cover plates
- ρ_R Measured-to-nominal ratio of the resistance
- ρ_r Measured-to-nominal ratio of the radius of gyration
- ρ_S Measured-to-nominal ratio of the load effect
- ρ_s Simulated professional ratio
- ρ_{seq} Simulated professional ratio predicted by the best-fit equation
- ρ_{λ} Measured-to-nominal ratio of the non-dimensional slenderness ratio
- σ_{b_r} Standard deviation of the width of the flange of the rolled section
- σ_{b_n} Standard deviation of the width of the cover plate
- σ_d Standard deviation of the depth of the rolled section
- σ_E Standard deviation of the modulus of elasticity
- σ_F Standard deviation of the stress defined by an expression of F_v and λ
- σ_{Fv} Standard deviation of the yield strength
- σ_g Standard deviation of side of fillet welds connecting the plates to the rolled section
- σ_k Standard deviation of the side of the fillet connecting the flanges to the web of the rolled section
- σ_R Standard deviation of the resistance
- $\sigma_{\rm r}$ Standard deviation of the radius of gyration
- $\sigma_{\rm S}$ Standard deviation of the load effect
- σ_{t_f} Standard deviation of the thickness of the flange of the rolled section
- σ_{t_p} Standard deviation of the thickness of the cover plate
- $\sigma_{\rm w}$ Standard deviation of the thickness of the web of the rolled section
- σ_{wr} Residual stress resulting from welding
- σ_x Longitudinal thermal stress in the steel column
- Resistance factor

Chapter 1

Introduction

1.1 General Background

It might be necessary to strengthen steel columns many years after construction. Most columns in existing structures carry some load at the time of reinforcing. Common reinforcement of a steel column is welding or bolting cover plates on the column. In bridges, for example, the cover plates would preferably be bolted since welding would introduce potential fatigue problems, while columns in other structures have been reinforced by welding cover plates.

There are no specific design criteria for reinforced columns in Canada up to now. In practice, many engineers would use the same design criteria for reinforced columns as those for rolled W section columns.

Since the limit states design method was employed in Canada in 1974 (Canadian Standards Association, 1974), the design criteria for steel columns have been used based on the SSRC multiple curves. CSA standard CAN3-S16.1-M84 – "Steel Structures for Building – Limit States Design" (Canadian Standards Association, 1984) adopted the SSRC curve 1 and the SSRC curve 2 for the design of steel columns. Each curve presents the behaviour and strength of different kinds of steel columns. The lower curve of clause 13.3.1 (based on the SSRC curve 2) is used for hot-rolled W section columns. Kennedy and Gad Aly (1980) suggested that the higher curve of clause 13.3.2 (based on the SSRC curve 1) was appropriate for class H hollow structural sections. Based on the study of Chernenko and Kennedy (1989), clause 13.3.2 is also used for Canadian WWF columns.

In the current Canadian standard the five-part equation developed by the SSRC was replaced by a double exponential representation with a single parameter, which was proposed by Loov (1996). The column curve described by the expression corresponding to the SSRC curve 1 is CSA curve 1, and the column curve described by the expression corresponding to the SSRC curve 2 is CSA curve 2. The CSA curves were demonstrated to accurately approximate the corresponding SSRC curves.

1.2 Statement of the Problem

In the columns reinforced with welding cover plates, welding can introduce tensile residual stresses at the flange tips of the rolled section and the edges of the reinforcing plates. Since yielding begins at the tips and progresses inwardly, these tensile residual stresses may be beneficial to the column strength by delaying the deterioration in minor axis stiffness.

Out-of-straightness, more specifically called camber or sweep depending about which axis the out-of-straightness occurs, is generally understood to be an important influencing factor for any column. The current S16.1 column curve (CSA curve 2) for rolled W sections is based on a maximum allowable out-of-straightness of L/1000 for both axes (Bjorhovde, 1972). It is more acceptable for S16.1 column curve to be based on statistical quantities, that is, on the mean values and associated coefficients of variations.

Furthermore, more influencing factors exist in reinforced columns, such as orientation of reinforcing plates, welding residual stress, geometric and material properties of the rolled section and plates, comparing to rolled W sections. The addition of reinforcing plates may affect the behaviour and strength of reinforced columns a lot. These parameters may have individual and combined effects on the prediction of reinforced column strength.

The differences between reinforced columns with welded cover plates and rolled W sections affect column strengths over the full range of column lengths, and suggest that reinforced columns with welded cover plates may be unnecessarily penalized with the CSA curve 2 (or SSRC curve 2) along with rolled W sections. Different column curves should be used for the two types of sections.

1.3 Objectives and Scope

In order to understand the uncertainty problems in reinforced columns with welded cover plates, the research was designed with the following objectives:

- 1. To review the existing literature on reinforced columns.
- 2. To develop a finite element model for reinforced columns under load.
- 3. To select the parameters influencing the behaviour and the strength of reinforced columns and to study the effects of parameters on reinforced columns.
- 4. To investigate statistically the resistance of reinforced columns produced in Canada by evaluating resistance factors appropriate for use with existing column curves for reinforced columns with welded cover plates under load.
- 5. To assess existing design criterion for reinforced columns with welded cover plates under load.

In the research, the analyzed rolled W section columns were only reinforced with welding cover plates because there is a potential for the welding residual stresses to improve the strength of welded reinforced columns. The beneficial effect of the welding residual stresses is not present when the reinforcing plates are bolted on the column.

A finite element program, ABAQUS (Hibbitt et al, 1997), was used to assess the effects of variations in parameters on the behaviour and the strength of reinforced columns. Out-of-straightness was restricted to a superposition of four buckling modes of the column. The study was limited to centrally loaded, pin-ended columns, buckling about the major or minor centroidal axis, and laterally supported about the other axis when required. Local buckling, buckling about both axes simultaneously, and lateral torsional buckling were not considered. Resistance factors were evaluated for values of the slenderness parameter, λ, of 0.4, 1.1 and 1.5 in two categories, in respect of the design criteria of CAN3-S16.1-M84 and CAN/CSA-S16.1-94, respectively. The first category includes columns reinforced with plates parallel to the flanges with buckling about the strong axis of the W shape and columns reinforced with plates parallel to the web with buckling about the weak axis of the W shape. The second category includes columns reinforced with plates parallel to the flanges with buckling about the weak axis of the W shape, and columns reinforced with plates parallel to the web with buckling about the strong axis of the W shape.

1.4 Organization of the Thesis

A literature review is presented in Chapter 2. This outlines the research done for parameters influencing the behaviour and strength of reinforced columns. A design method for reinforced columns is also discussed. Chapter 3 presents a description of a finite element model setup and analytical procedure for reinforced columns with welding cover plates. Parametric studies to assess the effect of the parameters on the reinforced columns are presented in Chapter 4. Based on a review of the principle of limit states philosophy associated with the column design process, the statistical analysis to give the resistance factor for the design of a reinforced column and to verify which design curve given in the code is appropriate to the reinforced column design is treated in Chapter 5. Finally, a summary, conclusion and recommendations for further research are presented in Chapter 6.

Initial geometrical, material and load conditions of all the finite element analytical models are tabulated in Appendix A. Appendix B presents the results of the analytical models. The statistical analysis data for the columns in category 2 (columns reinforced with plates parallel to the flanges and buckling about the weak axis of the rolled section and columns reinforced with plates parallel to the web and buckling about the strong axis of the rolled section) are presented in Appendix C.

Chapter 2

Literature Review

2.1 Factors Influencing Column Strength

2.1.1 Introduction

A seemingly simple structural column in fact functions as a complex individual structural member because of the effects of various parameters such as the interaction between the responses and characteristics of the material, the cross-section, the method of fabrication, the imperfections and other geometric factors, and the end conditions. Geschwindner *et al.* (1994) suggested that the following parameters affect column strength:

- 1. Material properties
 - (a) Stress-strain relationship
 - (b) Yield strength
- 2. Shape of cross-section
 - (a) Area of steel
 - (b) Shape of the cross-section (W, C, WT, etc.)
 - (c) Buckling axis
- 3. Length
- 4. End support conditions
 - (a) Without sway, pinned or otherwise
 - (b) With sway, pinned or otherwise
- 5. Residual stress magnitude and distribution
- 6. Initial imperfections
 - (a) Magnitude
 - (b) Distribution along column length

It is generally accepted that the yield strength and the modulus of elasticity are the most important material properties. For very short columns, the load carrying capacity may reflect a strength increase due to strain-hardening, but for hot-rolled structural

shapes, other factors such as local buckling may limit this strength increase. Therefore, the yield strength represents the practical limit of capacity of a very short column. For long columns, the capacity is influenced more by stiffness, which is a function of the magnitude of the tangent modulus and the cross-section moment of inertia (Galambos, 1998; Geschwindner *et al.*, 1994).

The shape of a cross-section is obviously important. For a given stress level, the load-carrying capacity will be larger for a column of larger area. The distribution of the area in the cross-section is expressed as the moment of inertia, which affects the capacity of columns that fail by buckling. The buckling axis is another factor influencing the behaviour of columns. The buckling capacity of columns about different axes is governed by different moments of inertia and corresponding slenderness ratios, which are defined as ratio of the effective column length to the radius of gyration. The geometry of the cross-section also influences the residual stress distribution.

The effective length concept has been introduced to account for the effect of column length and boundary conditions on the capacity of columns (Galambos, 1968). Euler first developed an analytical model based on the assumption that both ends of the column were completely free to rotate as the column reached its buckling strength. This situation will sometimes arise. For other end restraint conditions, the actual length of the column, L, is replaced by its effective length, KL, that is, the length of a pin-ended column of the same capacity as the column with other end restraint conditions. This effective length corresponds to the distance between points of inflection (points of zero bending moment) on the buckled shape. As a parameter of the effective length, the end support condition also has been generally understood as a column strength parameter.

The influence of column effective length is explicit in design calculations, whereas the influence of other factors such as initial imperfections and residual stresses may be hidden in the design approach, although they may also be important. In design practice, a non-dimensional slenderness parameter, λ , has been found to be the most important factor (Chen and Lui, 1987). This parameter is taken as the square root of the ratio of the yield stress to the elastic buckling stress, which is expressed as:

$$\lambda = \sqrt{\frac{F_y}{F_e}} = \frac{KL}{r} \sqrt{\frac{F_y}{\pi^2 E}}$$
 [2.1]

where F_e is the elastic buckling stress, E is the modulus of elasticity, r is the radius of gyration of the cross-section, and F_y is the yield strength of the column. With respect to their slenderness, columns are generally referred to as short columns, intermediate columns or slender columns. Slender columns buckle when the cross-section is still elastic. Intermediate columns buckle when part of the cross-section has yielded under the combined action of the applied axial load and the residual stresses. Short columns usually fail by local buckling after yielding of the full cross-section.

Residual stresses, formed during the cooling process after hot rolling, are influenced by the distribution of material in the cross-section. It is commonly accepted that the flange tips of a rolled I-shape section are subjected to residual compressive stresses because these areas, which possess less material and more exposed surface area, cool down faster than the flange to web junctions, which possess a larger volume of material to surface area ratio. This differential cooling gives rise to compressive residual stresses at the flange tips and tensile residual stresses at the flange to web junctions. Residual stresses have a major impact on the load-carrying capacity of a steel column. The investigation conducted by Huber (1956) demonstrated that the strength of a column could be reduced by the presence of residual stresses. Residual stresses also cause nonlinearity of the stress versus strain relationship as soon as any part of the cross-section starts to yield due to the combination of the applied stresses and the residual stresses. The tangent and reduced modulus theories developed by Engesser (Galambos, 1968) are commonly applied for these inelastic columns. The critical loads computed by these theories correspond closely to experimental results.

Initial out-of-straightness has long been recognized as a significant column strength parameter. It has been found that the effect of initial out-of-straightness is different for columns of different lengths (Galambos, 1998). Short and slender columns are not affected much by initial out-of-straightness, but intermediate columns are significantly affected (Galambos, 1998). The magnitude of initial out-of-straightness in any member is limited by manufacturing tolerance limits set by CSA standard G40.20-92 (Canadian

Standards Association, 1992), which reflects standard mill practice. Thus, the maximum value of out-of-straightness for rolled wide-flange shapes is set at L/1000, where L is the member length, with some minor modifications for longer sections and certain geometries (Canadian Standards Association, 1992).

The studies show that the influence of above parameters is more severe for intermediate length columns than for short or slender columns (Kulak and Gilmor, 1998; Galambos, 1998; Kennedy et al., 1976). The effect on the reinforced column also has to be studied to verify their importance.

Tall (1989) suggested that other factors influence the strength of reinforced steel columns as follows:

- 1. The orientation of the cover plates welded to the column;
- 2. Different grades of steel for cover plates and rolled section.

Reinforcement is usually understood to be the welding or bolting of cover plates to the flanges or the web of the cross section (Tall, 1989). The orientation of cover plates welded to the flanges can be either parallel to the flanges or to the web, as shown in Figure 2.1. Different reinforcing plate orientations for the same size of cross-section will introduce different moments of inertia and slenderness ratios for the reinforced column, both of which can affect the strength of the reinforced column significantly.

2.1.2 Initial Out-of-straightness

Although not perfectly straight, structural shapes are expected to satisfy the straightness requirements of the applicable materials delivery standard, CSA standard G40.20-92. Before the limit states design philosophy was developed, the effects of any initial imperfections were covered through the factor of safety (Geschwindner *et al.*, 1994). The strength of the straight member was used as the actual criterion. With the limit states design philosophy, all of the major parameters have to be accounted for. Therefore, the influence of initial out-of-straightness must be reflected in the strength equations.

Initial imperfections are the result of the cooling process for the shape once the column has been rolled to its final dimensions. When the column is left on the cooling bed to cool in air for a period of time, other members are also placed on the bed, and the heat dissipation from the member is not uniform neither throughout the cross section nor along the length or around its sides. As a result, heat is usually retained longer in the midlength portion of the column and in the parts of the cross section exposed most directly to the heat of the adjacent members on the cooling bed.

The resulting non-uniform cooling leads not only to residual stresses but also a steel member in a curved configuration along its length. The amount of curvature or initial out-of-straightness is difficult to predict because of a number of uncontrollable factors. The maximum out-of-straightness is limited typically based on the length of the member by CAN/CSA-G40.20-92. The maximum values are set as $\delta_i = L/1000$ for column length less than 14 m and $\delta_i = 10 + (L-14000)/1000$ for column lengths larger than 14 m. The value is different for different cross sections. A column that does not meet this straightness requirement is rotorized or gag-straightened to bring it into compliance with the code. Bjorhovde (1972) found that the mean value of initial out-of-straightness of rolled W sections, is L/1500, which is less than the code limit.

It was found that the shape of the initial imperfections, i.e., their variation along the length of the column, differs from the commonly assumed half sine wave (Bjorhovde, 1972). However, if it is assumed that the initial shape of the axis of the pinned-ended column is sinusoidal, the resulting average stress at which the maximum stress in the column equals the yield stress can be obtained by the Perry-Robertson equation (Johnston, 1976). This average stress gives a conservative estimate of the strength of an initially curved column.

Geschwindner et al. (1994) suggested that the combined effect of residual stresses and initial out-of-straightness could not be obtained merely by combining the two terms. In some cases, and for certain slenderness ratio ranges, the strength of a column with residual stresses and initial out-of-straightness is less than what would be found if the

effects of both were added. In other cases, it is not as critical as the sum would seem to indicate.

2.1.3 Residual Stresses

The manufacturing method is one of the primary factors influencing the distribution and magnitude of initial residual stresses in the cross-section. The research in this paper is limited to hot-rolled sections with welded reinforcement. Therefore, the review of the literature was limited to factors related to these specific manufacturing processes.

Under the combined action of residual stresses and applied axial loads, yielding of a column will start when the sum of the applied stress and the maximum compressive residual stress reaches the yield strength of the material. Beyond this point, the column becomes inelastic. Considering the inelastic behaviour of columns, Engesser presented the tangent modulus theory (Galambos, 1998). According to this theory, once the column has become inelastic, its behaviour is dictated by the tangent modulus. The tangent modulus column strength equation can be expressed as:

$$P_{cr} = \frac{\pi^2 E_t}{(KL/r)^2} A$$
 [2.2]

where P_{cr} is the column strength, E_t is the tangent modulus, and A is the gross area of the cross section.

Huber (1956) measured residual stresses in a series of hot-rolled W-shapes ranging from a light section: 4WF13 (W100x19) to a heavy section: 36WF150 (W920x223). The results showed that there are significantly different patterns for different sections. In rolled sections the residual stresses at the flange tips and the middle of the web are compressive stresses, while the residual stresses in the flange to web junctions are tensile. It was also observed that in most hot-rolled W sections the maximum compressive residual stress is approximately 30% of the yield strength (Chen and Atsuta, 1976).

Residual stress patterns and magnitude also vary widely in steel plates. According to CSA standard G40.20-92, plates are classified as universal mill (UM) plates, sheared plates, or flame cut plates. The initial residual stresses are significantly different in the

three types of plates. The edges of a UM plate or a sheared plate are in compression and the edges of a flame cut plate are in tension (Geschwindner et al., 1994). Tall (1961) and Nagaraja Rao and Tall (1963) investigated residual stress patterns for UM plates ranging from 150 x 6 mm to 300 x 18 mm. All the plates investigated had almost the same initial residual stress pattern with different maximum magnitudes at the edges of the plates. The measured maximum compressive residual stresses were about 30% of the yield strength.

The welding of built-up shapes is an even greater contributor to residual stresses than the differential cooling of hot-rolled shapes (Nagaraja Rao et al., 1964). Nonuniform cooling and restrained shrinkage of welds cause high residual stresses. Masubuchi (1980) observed that the maximum magnitude of tensile residual stress at the weld center is as high as the yield strength of the weld metal. Tall (1961) presented residual stress patterns measured in plates after welding of the plates to wide flange sections. Plate sizes ranged from 150x6 mm to 300x18 mm. The zone of tensile residual stresses resulting from welding along the plate edges was observed to extend from 18 mm to 37.5 mm. The thicker plates have wider zones of tensile residual stresses. The welding residual stresses were observed to reach the yield point value only at and in the vicinity of the weld.

2.2 Research on Reinforced Columns

Very little research has been conducted on reinforced steel columns, although many structures have been strengthened. The processes used to reinforce columns were simply presumed to be safe. Nagaraja Rao and Tall (1962) reported on the work of Wilson and Brown who conducted tests on the strengthening of columns of a viaduct in 1935. Cover plates were welded to the existing sections. It was observed that in some cases the residual stresses in reinforced columns after welding might reach the yield point.

Sparagen and Grapnel (1946) presented an early review of all the literature on structures reinforced under load. They concluded that residual stresses, although high, did not seriously affect the ultimate strength of a column.

2.2.1 Work of Tall and Co-workers

Nagaraja Rao and Tall (1963) conducted an experimental investigation of the effect of welding cover plates to a wide flange column under load. The program consisted of residual stress distribution determination, tension coupon testing, stub column tests, and column tests.

Three pin-ended columns were tested: an unreinforced column, a column reinforced under load, and a column reinforced under no load. Ancillary tests were conducted to investigate the residual stress distribution and the yield strength of the columns.

In the investigation, a 8WF31 (W200x46) shape was selected as the core column and 180x9.5 mm plates were used for the cover plates. The rolled shape and the cover plates were of ASTM A7 steel. Because the selected rolled shape has one of the lowest shape factors (the area, the moment of inertia and so on) and b/t ratio in available rolled sections, the results of tests would be conservative for other sections. The preload at the time of welding was fixed to 405 kN (the stress on the section was 69 MPa).

For convenience in testing and in comparing results, all the three pin-ended columns were tested with boundary conditions that allowed buckling about the weak axis. The column specimens were 2440 mm long, resulting in a slenderness ratio of about 48. The reinforced columns had an out-of-straightness of 0.51 mm (L/4880) for the column reinforced under load and 0.77 mm (L/3190) for the column reinforced under no load, whereas the unreinforced column had an out-of-straightness of 4.335 mm (L/560).

Residual stresses were measured both in the unreinforced and in the reinforced column using the method of sectioning. It was observed that the compressive residual stresses at the flange tips of the unreinforced section were changed to high tensile residual stresses as a result of welding of the reinforcing plates.

The effect of welding sequence on residual stresses in a reinforced column was also investigated. In the experiments, two welding sequences were selected. The welding methods conformed to ASCE-AWS standards then. The two welding sequences were:

- 1) Welding each flange one after another, stage by stage, as shown in Figure 2.2 (a).
- 2) Welding two diagonally opposite flanges simultaneously, as shown in Figure 2.2 (b).

The residual stress distributions with different welding sequences were also measured by the method of sectioning. Welding residual stresses in the flanges and plates were almost the same under different welding sequences, whereas welding residual stresses in the web were found to be different. The maximum magnitudes of the residual stresses in the flanges and plates were close to each other under different welding sequences. Masubuchi (1980) suggested that as far as residual stresses along the weld are concerned, the effect of welding sequence is minor.

The test results on the reinforced columns indicated that the stress level in the reinforced columns at buckling is greater than the buckling stress for the unreinforced section. A theoretical tangent modulus column curve for the reinforced section was developed from the stub-column test results (Nagaraja Rao and Tall, 1963). The resulting column curve for a reinforced W200x46 fell above the Column Research Council (CRC) curve, indicating a higher strength from the reinforced section. The CRC curve represents an average curve for bending about either the strong or the weak axis (Huber, 1958).

The following observations were made from the test results:

- The pin-ended column showed that the reinforced sections had a higher capacity.
 One of the pin-ended columns tested reached 98% of its yield strength. The stub
 column tests and the pin-ended column tests showed that welding for shorter
 reinforced columns did not reduce the buckling stresses.
- 2. The influence of welding is confined to a very small area in the vicinity of the weld. The properties of the material in the major portion of the section are not affected enough to change the strength of the reinforced section.

A later paper presented by Tall (1989) extended the discussion of the welded reinforcement based on the previous investigation. Further discussion of the residual stress magnitude and distribution was presented.

The initial compressive residual stresses in W-shapes would contribute to a reduction of compressive strength. To minimize the loss of compressive strength, it is desirable to change the compressive residual stresses to tensile stresses at the critical portion of the rolled section, i.e. the tips of the flanges, to delay yielding of this portion under a compressive load. Welding of cover plates to the flanges or simply laying a weld bead on the flange tips are two ways to achieve this. The change of residual stress distribution resulting from welding alone was demonstrated to result in a marked improvement in column strength (Fujita, 1960). Tall (1989) proposed that the increase in strength resulting from welded cover plates on a rolled section is substantially greater than for welding alone because of the combined effect of the additional material and welding residual stresses.

The test results presented by Nagaraja Rao and Tall (1963) showed a 10% increase in strength after reinforcement. Because the non-dimensionalized strength is defined with respect to the yield strength of the total cross section, which differs before and after reinforcement, the actual absolute increase in strength would be considerably higher. Therefore, the reinforcement of a column may result in the column being assigned a higher column curve if the concept of multiple column curves is considered.

It has been shown (Alpsten and Tall, 1970; Brozzetti et al., 1970; Bjorhovde et al., 1972; Kishima et al., 1969) that welding has a greater influence on the overall distribution of residual stresses in small and medium-size shapes, than in the case of heavy shapes. Tall (1989) therefore proposed that welding alone on the flange tips would improve the strength of rolled sections of light and medium size more than for heavy rolled shapes.

Tall (1989) suggested that the width of the cover plate should not be smaller than the width of the flange less the size of two fillet welds — this ensures that the weld is as close as possible to the flange tips so as to be effective in changing the residual stresses at the flange tips from compression to tension. The maximum effect of reinforcement is obtained when the reinforcing weld is as close as possible to the edge of the flange of the rolled section.

Comparing test results for columns reinforced under load and reinforced under no load, Tall (1989) suggested that preload would not affect the strength of short reinforced columns.

2.2.2 Work of Brown (1988)

Brown (1988) proposed a simplified model of a reinforced steel column to evaluate its strength when reinforced under load. The model consists of two flexible columns, one representing the unreinforced column (or column core) and the other representing the reinforcing plates, tied together with rigid links to enforce compatibility of displacements between the column core and the reinforcing plates. Depending on the slenderness of the column, three ranges of column response were identified: 1) the core column forms a plastic hinge and the reinforcement provides the additional capacity until it reaches the maximum load capacity (i.e. the capacity of the column reinforced under load is the same as a column reinforced under no load); 2) after the core column fails, the reinforcing plates provide additional capacity, but the reinforced column does not reach the full capacity of a column reinforced under no load; 3) failure of the reinforced column takes place when the stress in both the column core and the reinforcing plates is below the buckling stress of the same column reinforced under no load. The model proposed by Brown does not account for any of the residual stresses and depends on SSRC column design curves to account for the presence of the residual stresses.

2.3 Summary

Residual stresses, initial geometric imperfections, and material and geometric properties were identified as the most influential factors for the capacity of unreinforced columns. A review of the literature has indicated that additional factors affect the strength and behaviour of welded reinforced columns. These factors, however, have received little attention. The direction of cover plates, the grade of steel used for the columns and the cover plates, the magnitude of the load carried by the unreinforced column when the reinforcing plates are welded to the column, and the buckling axes were identified as

potentially important parameters for reinforced columns. The effect of these parameters on the strength and behaviour of reinforced steel columns needs to be investigated.

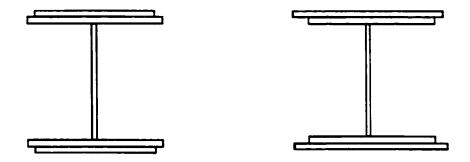
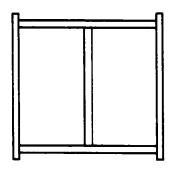


Plate narrower than flange

Plate wider than flange

(a) Reinforcing Plates Parallel to Flange



(b) Reinforcing Plates Parallel to Web

Figure 2.1 Formation of the Analysis Models

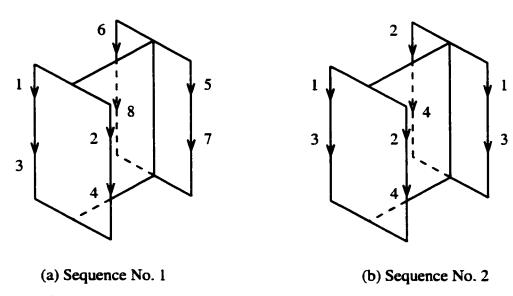


Figure 2.2 Welding Sequences Investigated by Nagaraja Rao and Tall (1963)

Chapter 3

Finite Element Modelling of Reinforced Steel Column

3.1 General

A review of the literature indicated that the number of tests conducted on steel columns reinforced under load with welded steel plates is very limited. To fully understand the behaviour of reinforced steel columns, a large number of tests are required in order to incorporate a wide range of parameters that affect the behaviour of reinforced steel columns. However, it is uneconomical to conduct a large experimentally based investigation. In order to extend the database of test results, a numerical model was used to investigate the full range of parameters not covered by the test. The performance of this model was first verified by comparing the predicted strength and behaviour with test results.

The objective of this chapter is to develop and validate a finite element model of a reinforced steel column. The finite element model was developed using the commercial software ABAQUS, version 5.7 (Hibbitt et al., 1997). ABAQUS was used because of its ability to perform non-linear large displacement and finite strain post-buckling analysis.

This chapter consists of two parts. In the first part, the geometry and the boundary conditions of the finite element model are described. The initial conditions and load process of the numerical analysis are also discussed. In the second part, the results of the numerical analysis are compared with the results of physical tests to validate the numerical models.

3.2 Description of the Model

To investigate the full range of possible cross sections for reinforced columns, two groups of finite element models were studied in the following investigation:

1) Columns reinforced with plates parallel to the flanges as shown in Figure 2.1(a) and,

2) Columns reinforced with cover plates parallel to the web as shown in Figure 2.1(b).

The finite element model of a reinforced column is composed of three parts: the rolled section, the reinforcing plates, and the welds connecting the reinforcing plates to the rolled section. The rolled section and reinforcing plates were discretized using element S4R from the ABAQUS finite element library. Element S4R is a four-node, doubly curved, general purpose shell element with finite strain capability and six degrees of freedom per node (Hibbitt *et al.*, 1997). The welded joint between the reinforcing plates and the rolled section was modelled using the two-node linear beam element B31. Element B31 has six degrees of freedom per node and transverse shear deformation capability.

When the beam elements were introduced into the model to simulate the welds, the minimum stiffness required to prevent relative displacement between two corresponding nodes on the rolled section and the reinforcing plate was determined by gradually increasing the beam stiffness until the relative displacements were considered negligible. This process was necessary to avoid potential convergence problems with excessively stiff beam elements.

The beam elements were added when the finite element model was built. In order to prevent interaction between the rolled shape and the reinforcing plates when residual stresses and column preload were added, the beam elements had to be deactivated in the first load step. To prevent rigid body motions of the reinforcing plates before their reattachment to the rolled section, the plates were connected to the rolled section using a tie connection at the column mid-height (see Figure 3.1). The single tie connection prevented rigid body motion of the reinforcing plates while allowing independent straining of the plates and rolled section during the application of the residual stresses.

3.2.1 Finite Element Mesh

The mesh size used for modelling the reinforced columns was based on two principal considerations: 1) a sufficient number of elements had to be used to form the cross-

section so that the residual stress pattern could be modelled accurately, and 2) the aspect ratio of the elements was below 3.0 to avoid potential numerical problems (Gaylord *et al.*, 1997).

Figures 3.2 and 3.3 show the finite element mesh of the column reinforced with plates parallel to the flanges and that of the column reinforced with plates parallel to the web, respectively. To allow for welding of the cover plates, the cover plates were narrower and shorter than the W-shape column by about 20 mm.

The width of the elements adjacent to the tips of the flanges and webs were determined to accurately simulate the welding residual stresses in the cross-section. Welding residual stress distributions in steel plates were investigated both experimentally and analytically by Tall (1961). The thickness of the plates ranged from 1/4 inch (6 mm) to 3/4 inch (19 mm). The distance from the welded edge of the plate to the point where a residual stress reversal occurs was found to vary from 18 mm to 38 mm. Nagaraja Rao and Tall (1963) also investigated the welding residual stress distribution in a 8WF31 (W200x46) section with 180x9.5 mm plates welded parallel to the flanges. The residual stresses changed from positive to negative about 25 mm from the welded edge in the flanges and the cover plates. Therefore, the width of the elements adjacent to the edges in the flanges and the cover plates was chosen as 25 mm. This was found to provide welding residual stress patterns similar to the residual stress patterns obtained experimentally. This will be discussed further in the following.

3.2.2 Material Properties

An isotropic, elastic-perfectly plastic material model was used for all the elements of the reinforced columns. The elastic range for all the elements was defined with an elastic modulus of 200 000 MPa and Poisson's ratio of 0.3. The yield stress level was varied for the rolled section and the reinforcing plates as described in the following chapter. The material properties for the beam elements, used to model the welds, were the same as for the cover plates.

The residual stresses in the plates and rolled section were introduced by imposing a temperature gradient in the cross-section. In order to introduce longitudinal residual stresses only, orthotropic thermal expansion properties were used. The value of the coefficient of thermal expansion was taken as 1.17×10^{-6} /°C in the longitudinal direction and zero in the transverse and through-thickness directions. The stress caused by constrained thermal expansion is expressed as

$$\sigma_{x} = -E \alpha_{x} \Delta \tau \tag{3.1}$$

where σ_x is the longitudinal stress, E is the modulus of elasticity, α_x is the coefficient of thermal expansion in the longitudinal direction and $\Delta \tau$ is the temperature change.

3.2.3 Boundary Conditions

The parametric study presented in the following chapter was limited to centrally loaded, pin-ended columns. The pinned ends were modelled using constraint equations between the centroid of the end cross-sections and each node on the cross-sections to force the ends of the column to remain plane and create a hinge about the centroid at each end of the column.

A restraint about the weak axis or the strong axis was added at the column end crosssection to make the corresponding slenderness ratio less than the slenderness ratio about the other axis. These end restraints were added either to promote buckling about the strong axis in some of the specimens included in the parametric study, or to prevent simultaneous buckling about two axes in very few of the specimens investigated with nearly equal stiffness about the strong and the weak axes.

In some of the reinforced column models with reinforcing plates parallel to the web and $\lambda = 0.4$, unexpected local buckling occurred at the ends of the column. This failure mode will be discussed further in section 4.2. However, because an investigation of local buckling of reinforced steel columns is beyond the scope of this research project, the thickness of the first three rows of elements near the ends was increased by a factor of three to prevent local buckling.

3.2.4 Initial Conditions

The response of the column in the post-buckling range is of interest to assess the stability, or lack thereof, of the column after the peak load has been reached. To analyse the post-buckling behaviour, the bifurcation problem that exists when no initial imperfections are present in the model must be transformed into a problem with continuous response. This can be accomplished by introducing a geometric imperfection. Since the exact shape of the initial imperfections was not known a priori, initial imperfections, consisting of a superposition of multiple buckling modes, were introduced in the unreinforced column model. In ABAQUS this is accomplished in two analysis runs: 1) in the first run an eigenvalue buckling analysis is performed on the "perfect" geometry to determine the possible buckling modes; 2) in the second analysis run the initial imperfections are introduced by adding the buckling modes to the "perfect" geometry. The first four buckling modes were selected in the second analysis run. Research demonstrated that the first eigen mode provides the most critical imperfections (Galambos, 1968; Chen and Atsuta, 1976). In this research project the first four modes were used with different scaling factors to form the initial imperfections. The largest scaling factor was used for the first mode and the scaling factors of the 2nd mode, 3rd mode, and 4th mode were taken as 1/5, 1/10 and 1/20 of the first mode's scaling factor, respectively. Since the position of the maximum perturbation differs for all four modes, a trial and error method was adopted in the analysis to make the maximum magnitude of the superimposed eigenmodes equal to the desired magnitude of initial imperfection.

The second, third, and fourth buckling modes are not necessarily in the same plane as the first, and predominant, mode. Therefore, the superposition of the first four buckling modes introduces initial imperfections in the weak and in the strong axis directions. The magnitude of the initial imperfections in the strong direction was approximately 20%, or less, of the magnitude in the weak direction.

Initial temperatures at all the nodes in the model were defined as zero when the initial condition was defined. Based on this initial condition, temperature changes could be

introduced to simulate the initial residual strains, and stresses, and welding residual stresses in the load step described in the following section.

3.2.5 Loading Process

Because reinforcement of steel columns is usually performed while the column is carrying some load, loading of the reinforced steel column models had to be performed in several steps. The following steps were adopted in the analysis:

- 1) The finite element model initially consisted of the rolled section and the reinforcing plates attached to the rolled section with beam elements. It was therefore necessary in the first load step to de-activate the beam elements and perform an equilibrium iteration on the structure.
- Initial residual stresses were introduced in the rolled section and in the reinforcing plates in accordance to the temperature versus stress relationship presented in Equation [3.1].
- 3) An axial load representing the dead load and partial live load on the unreinforced column was introduced. The preload was varied from 40% to 60% of the unreinforced column strength.
- 4) The beam elements used to simulate the weld attachment between the reinforcing plates and the rolled section were re-activated. This was performed within a load step to ensure that equilibrium is maintained in the process of attaching the reinforcing plates to the wide flange section.
- 5) The welding residual stresses were introduced by increasing the temperature at the flange tips to create a strain at the flange tips of 70% or 100% of the yield strength of the wide flange section. It is generally accepted that the residual stress due to welding (at the tips of the flange and plates) can reach the yield stress (Tall, 1961).
- 6) The next step consisted of removing the initial preload applied in load step 3. Both the residual stresses and initial imperfections in the reinforced column were determined at the end of this step.
- 7) Riks' method was used to load the reinforced steel column into the pre- and post-buckling ranges.

Nonlinear static stress analysis was used for stable problem analyses such as removing and adding elements, imposing initial residual stress, pre-loading, welding effects, and removing pre-loading. As a consequence, large-displacement effects were included in all the steps of the loading process. The first five steps were performed using a load control Newton-Raphson procedure. In order to trace the post-buckling response of the reinforced columns the modified Riks method (Riks,1979) was used in the last load step.

3.3 Validation of the Finite Element Model

3.3.1 General

Because of the limited number of test results on reinforced steel columns, it is difficult to collect enough test data to fully validate the numerical model. Nagaraja Rao and Tall (1963) provided a set of test results for columns reinforced under load and under no load. To validate the numerical model as much as possible, both cases were compared with the analysis results.

3.3.2 Description of the Tests

The experimental investigation presented by Nagaraja Rao and Tall (1962) used W200x47 (8WF31) columns with 178x9.5 mm reinforcing plates, both of ASTM A7 structural steel. The reinforcing plates were placed parallel to the flanges. The weighted mean yield stress, determined from a stub-column test, was 256.5 MPa. The length of the column was 2440 mm, giving a non-dimensional slenderness ratio, λ , of about 0.5.

The column reinforced under load had an out-of-straightness of 0.5 mm (L/4900) after reinforcing. The out-of-straightness of the column reinforced under no load was 0.762 mm (L/3200). It should be noted that the reported initial out-of-straightness for both unreinforced columns was L/565, which is significantly greater than the maximum allowable initial out-of-straightness for wide flange sections.

An axial load was applied through end fixtures that allowed the columns to buckle freely about their weak axes. For the column reinforced under load, the pre-load applied before reinforcing was 30 percent of the capacity of the rolled section, namely 405 kN (Nagaraja Rao and Tall, 1963).

3.3.3 Initial Conditions of the Numerical Analyses

Based on the investigation by Nagaraja Rao and Tall (1963), two numerical models were developed to model column reinforcement under load and reinforcement under no load. The geometrical details and the numerical analysis results of the two models are summarized in Table 3.1. The first model consists of the column reinforced under load and the second model represents the column reinforced under no load.

The same initial residual stresses were used for the two numerical models as illustrated in Figure 3.4. Columns (8) and (9) of Table 3.1 present the initial residual stress distributions in the sections. Column (8) shows the maximum magnitude at the tips of the flanges and Column (9) shows the maximum magnitude at the reinforcing plate edges.

Figure 3.5 shows a comparison between the initial residual stress distributions in the cross-section of the numerical models and the test specimen. The figure presents both the input values of residual stresses in the finite element model and the output value, obtained at the end of an equilibrium step in the loading process described above. The numerical model replicates successfully the measured residual stresses.

The magnitude of the initial imperfections reported by Nagaraja Rao and Tall (1962) was also replicated in the finite element models. Since the residual stresses introduce deformations in the model, the initial geometry had to be adjusted so that the magnitude of initial imperfections at the end of the residual stress load step was equal to the measured value. A trial and error procedure was used for this purpose. Columns (11) and (12) of Table 3.1 present the magnitude of the initial out-of-straightness before reinforcing and the ratio of this initial out-of-straightness to the column length, respectively. Columns (13) and (14) present the value of the out-of-straightness after

reinforcing and the ratio of this out-of-straightness to the column length, respectively. Column (17) presents the magnitude of the preload applied on the rolled section before welding of the reinforcing plates. A comparison of the initial out-of-straightness of the unreinforced columns in the numerical models with the measured initial out-of-straightness (L/565) indicates that the model cannot predict accurately the effect of the reinforcing plates addition on the initial imperfections. The assumed initial imperfections in the two models before reinforcement of the columns were significantly smaller than the measured values, although the final initial imperfection magnitude in the numerical model is almost identical to the measured value.

Nagaraja Rao and Tall (1963) suggested that average stress at the flange tips after reinforcing was 70% of the measured yield strength. This magnitude was used also for the finite element models.

3.3.4 Behaviour of the Column Reinforced under Load

The welding residual stress patterns in the test specimens were investigated for different welding sequences (Nagaraja Rao and Tall, 1963). Figure 3.6 shows the residual stresses at mid-thickness of the plates after welding. The figure shows residual stresses obtained from the finite element analysis and the residual stresses measured on test specimens fabricated using two different welding sequences as described in Section 2.3.1. Since the experimental data represent surface residual stresses, interpolation between the two surfaces was used to obtain the mid-thickness residual stresses for the flanges and the reinforcing plates. Measured pattern I was obtained for the first welding sequence and measured pattern II was obtained for the second welding sequence. Figure 3.6 shows that the predicted residual stress patterns in the flange and cover plates are similar to the measured patterns. Although the measured residual stresses showed a significant gradient through the thickness near the flange tips, no attempt was made to incorporate this phenomenon in the numerical model. The model therefore used an average stress through the thickness.

The next step in the validation process is to compare the predicted load response of the reinforced column with the reported test results. Figure 3.7 compares the axial load ratio, P/P_{ry}, versus the mid-height lateral deflection response for the numerical model with the test result. The out-of-straightness value after welding for both cases was 0.5 mm. From the figure, we can make the following observations:

- 1) The shapes of the curves are similar.
- 2) The post-buckling range is accurately predicted by the finite element method.
- 3) The predicted and measured peak strengths are almost the same. Column (19) from Table 3.1 presents the ratio of the predicted to measured peak load. The difference between the predicted capacity and measured capacity is only 0.1%.
- 4) The slopes of the elastic portion of the response curves are identical.

It can therefore be concluded that the strength and behaviour of steel columns reinforced under load can be predicted very well with the proposed finite element model for the slenderness tested. It should be noted that the capacity of the column was very close to its yield strength, indicating that the reinforced column fell into the short column range. The model still remains to be validated in the intermediate length range.

3.3.5 Behaviour of the Column Reinforced under no Load

To further validate the numerical model, the column reinforced under no load was also modelled and analysed. The geometrical and material properties of the experimental model were the same as those of the first numerical model except for the magnitude of the pre-load, as presented in Table 3.1. The initial conditions used in the numerical model were discussed in section 3.3.3.

A comparison between measured and predicted residual stresses in the cross-section for the column reinforced under no load is presented in Figure 3.8. Again, a good agreement between the measured and predicted residual stresses is observed. Although the discrepancy between measured predicted values is more significant in the web the residual stresses in the web are not as influential on the column behaviour and capacity as those encountered at the flange tips.

The measured and predicted axial load versus mid-height lateral deflection response for the column reinforced under no load are shown in Figure 3.9. It can be observed that the curves are similar. The predicted peak strength from the numerical model is 97.7% of the yield strength. This is only slightly higher than the measured strength of 96% of the yield strength. It can therefore be concluded once more that the finite element model predicts the test results accurately.

3.3.6 Validation of the Finite Element Models in the Intermediate Length Range

Because of insufficient experimental data to validate the finite element models over the full range of material response, experimental data for unreinforced columns of intermediate length were used to validate the finite element models in the elastic-toplastic range.

Huber and Beedle (1954) presented the results of a series of tests on 8WF31 (W200x46) steel columns of different lengths. The steel was ASTM A7 structural steel, with a weighted average yield strength of 260 MPa. A residual stress pattern similar to the pattern illustrated in Figure 3.4 was used in the finite element models. The value of the peak residual stress was measured using the sectioning method (Huber and Beedle, 1954) and was reported to be 84 MPa. The details for two of the test specimens from Huber and Beedle (1954) are presented in Table 3.2. Other test specimens used to validate the finite element models were obtained from Beedle and Tall (1960) who reported tests on W-shape columns performed by other investigators. The material was also reported to be ASTM A7 structural steel. Since material properties were not specifically reported for these specimens, the same value as reported by Huber and Beedle (1954), namely, 260 MPa, was used for these test specimens. Table 3.2 also presents a summary of the properties used for these columns. It should be noted that since the magnitude of initial imperfections was not reported for these columns, values were assumed. In order to attempt to bracket the actual magnitude of initial imperfections two values were assumed, namely, L/1500 and L/10 000. The larger of the two values is significantly larger than those reported for the columns tested by Huber and Beedle

(1954) whereas the smaller of the two values is significantly smaller than those reported by Huber and Beedle.

Finite element analyses for the test specimens presented in Table 3.2 were conducted and the results are reported in column (8). The test results are reported in column (9) and the predicted-to-test ratios are reported in column (10). As can be seen, the test to predicted ratio for the first two test specimens is very close to 1.0, indicating an excellent correlation between the finite element models and the test results. The other predicted column capacities are not in such good agreement with the test results, however. The lack of agreement is attributed to the uncertainty in some of the important parameters of the finite element model that had to be assumed. It can be seen from Table 3.2 that a reduction of initial imperfection improves considerably the prediction of the test results. It is also expected that the assumption made about the actual yield strength of the test specimens would have an effect on the test-to-predicted ratio. Considering these later uncertainties, it is considered that the finite element models are able to predict accurately the strength of columns in the intermediate length range.

Table 3.1 Models Used for the Validation of the Finite Element Method for Reinforced Columns

FEA							RS	RSBR ^e W	Welding					Yield Strength	rength		Preload	
model	model I-section Plate $D^a B^b L^c \lambda^d$	Plate	<u>"</u>	æ	Ľ	ہر	MF	MP ^g	Residual	ශ්	_	κο	_	I-section plate	plate	-0 -0	P ₀ /P _{u2} ^m	P ₀ P ₀ /P _{u2} P _{fea} /P _{ry}
Z _o	No.				(mm)				Stress	M¹(mm)	ratio	M'(mm)	ratio	(MPa)	(MPa)	(K N		•
$\widehat{\Xi}$	(2)	(2) (3) (4) (5) (6) (7)	€	3	9	()		(8) (9)		(11)	(12)	(11) (12) (13) (14)	(14)	(15) (16) (17)	(91)	(17)	(17) (18)	(61)
_	W200x46 180x9.52 F W 2440 0.5	180x9.52	Ľ,	≩	2440	0.5		0.15F,	0.3F, 0.15F, 0.7F,	0.47	L/5200	0.47 L/5200 0.50 L/4900	L/4900	260	260	405	0.3	0.97
7	W200x46 180x9,52 F W 2440 0,5	180x9.52	F	≩	2440	0.5		0.15F	0.3F, 0.15F, 0.7F,	0.49	0.49 L/5000	0.50	L/4900	260	260	0	0.0	0.98
a) D -	a) D - Direction of reinforcing plates	f reinfor	cing	pla	tes					F - Parall	- Parallel to the flanges	flanges						
b) B -	b) B - Buckling axis of the reinforced column	xis of the	reir	Jorc	o pa	mnfa	=			W - Weal	s axis of	W - Weak axis of the I-section	ion					
c) [c) L Column length	oth																

 $\frac{1}{2}$ d) λ - Non-dimensional slenderness parameter of the reinforced column

e) RSBR - Residual stress before reinforcing

f) MF -Maximum magnitude of the residual stress at the flange tips

Fy - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress at the reinforcing plate edges h) δ_0 - Initial imperfection before reinforcing.

i) δ₁ - Out-of-straightness after reinforcing, no load.

j) M - Out-of-straightness in the weak direction.

k) ratio - The ratio of the out-of-straightness to the column length, L.

1) P₀ - Preload before reinforcing

m) P_{u2} - Load carrying capacity of the unreinforced column predicted using SSRC curve 2

n) P_{fea} - Load carrying capacity obtained from the finite element analysis

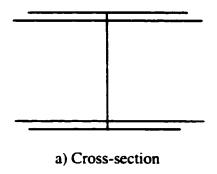
P_{1y} - Yield strength of the reinforced column

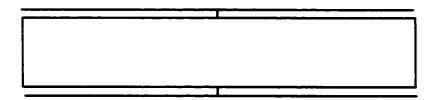
Table 3.2 Finite Element Models for the Unreinforced Columns

	Column		RSBR ^b	Initial Im	perfection				
I-section	Length	λ^{a}	MFc	before re	inforcing	F_y	P _{fea} /P _{uy} ^f	P_{exp}/P_{uy}^{g}	P _{fea} /P _{exp}
	L (mm)			ratio ^d	Me(mm)	(MPa)			
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)	(10)
W200x46*	4166	0.9	0.3F _y	L/6300	0.66	260	0.73	0.75	0.97
W200x46*	2946	1.2	$0.3F_{v}$	L/2400	1.22	260	0.83	0.82	1.01
W310x74**	4034	0.9	0.4F _v	L/1500	2.69	260	0.67	0.76	0.88
W310x74**	4034	0.9	$0.4F_{v}$	L/10000	0.40	260	0.75	0.76	0.98
W200x36**	3436	1.0	0.25F _v	L/1500	2.29	260	0.65	0.73	0.89
W200x36**	3436	1.0	0.25F _v	L/10000	0.34	260	0.72	0.73	0.98
W150x22**	3210	1.0	0.25F _v	L/1500	2.14	260	0.70	0.73	0.96
W150x22**	3210			L/10000	0.32	260	0.73	0.73	1.00

Note:

- * Test results reported by Huber and Beedle (1954)
- ** Test results reported by Beedle and Tall (1960)
- a) λ Slenderness parameter of the reinforced column
- b) RSBR Residual stress before reinforcing
- c) MF -Maximum magnitude of the residual stress in the flange.
 - F_y Yield stress of the unreinforced column
- d) ratio The ratio of the out-of-straightness to the column length, L.
- e) M Out-of-straightness in the weak direction.
- f) P_{fea} Load carrying capacity obtained from the finite element analysis
 - P_{uv} Yield strength of the rolled section column
- g) P_{exp} Experimetal strength of the rolled section column





b) The View from the Direction Parallel to the Web

Figure 3.1 Position of the Tie Connection between Column and Cover Plates

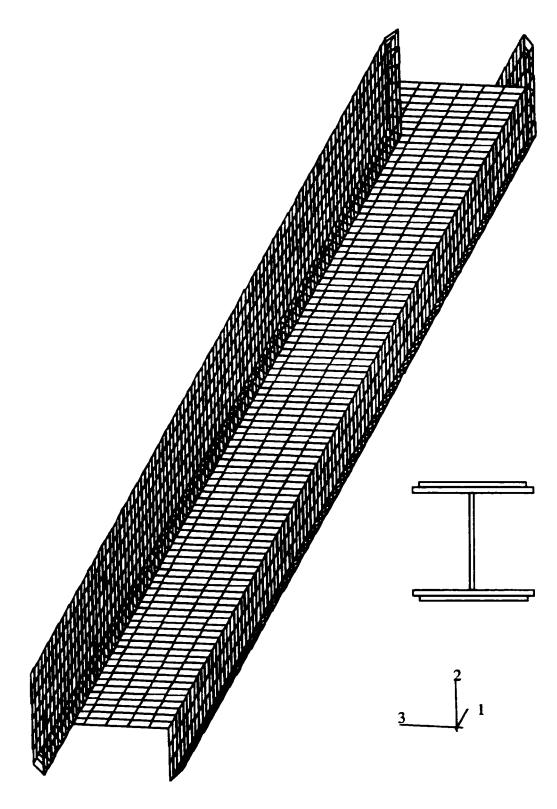


Figure 3.2 Finite Element Mesh of Column Reinforced with the Plates Parallel to the Flanges

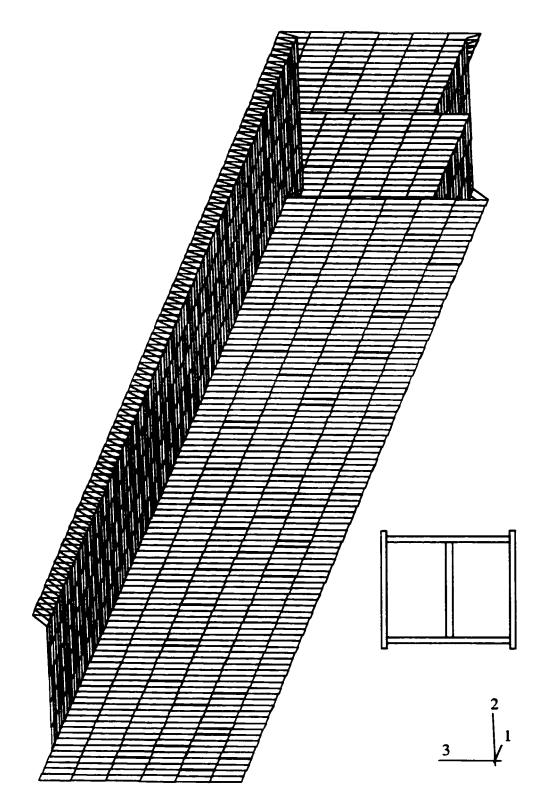
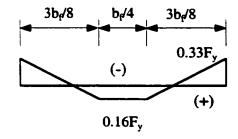
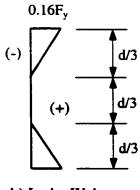


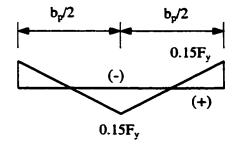
Figure 3.3 Finite Element Mesh of Column Reinforced with the Plates Parallel to the Web



a) In the Flanges



b) In the Web



a) In the Plates

 b_f - the width of the flanges w - the width of the web

 b_p - the width of the plates

Figure 3.4 Initial Residual Stress Pattern Used in the Numerical Models

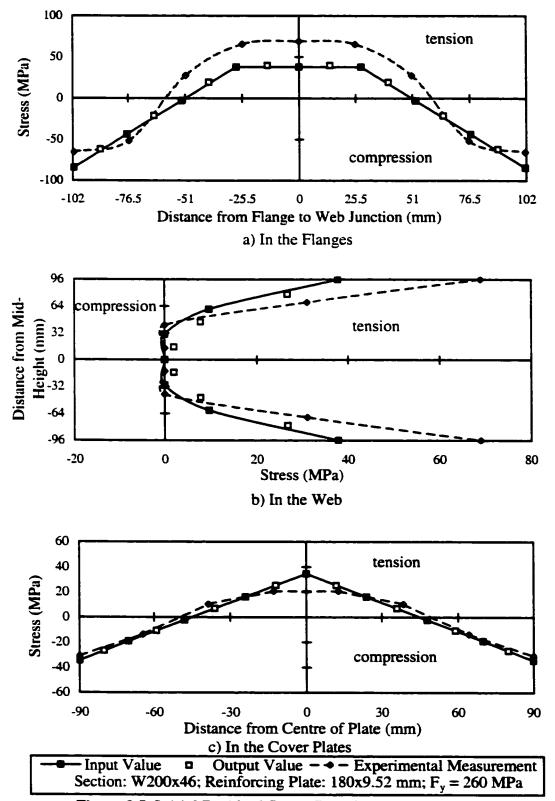


Figure 3.5 Initial Residual Stress Distribution in I-Section and Cover Plates of Finite Element Models

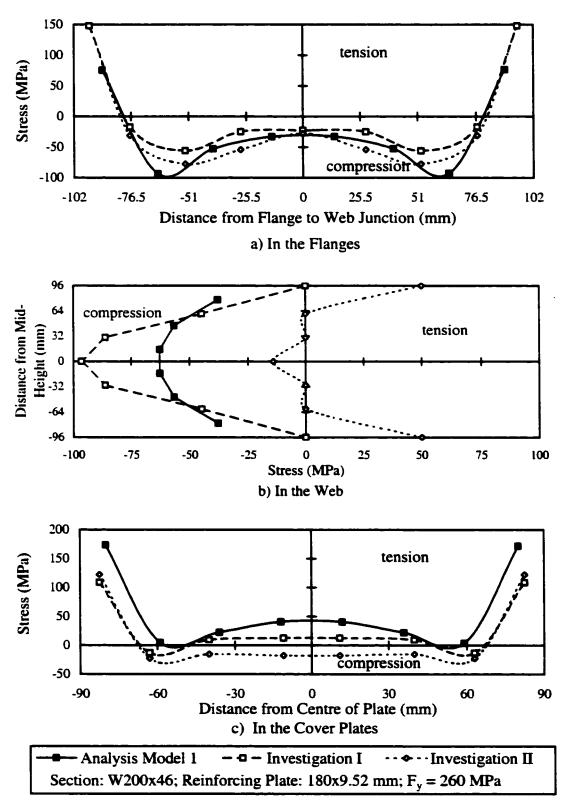


Figure 3.6 Welding Residual Stresses in the Column Reinforced under Load

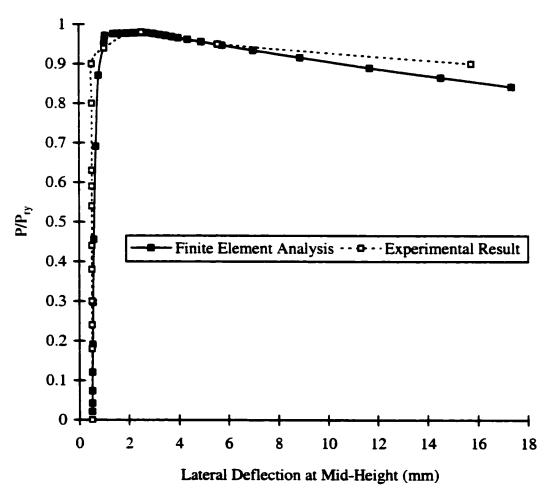


Figure 3.7 Axial Load versus Lateral Deflection Curves for the Column Reinforced under Load

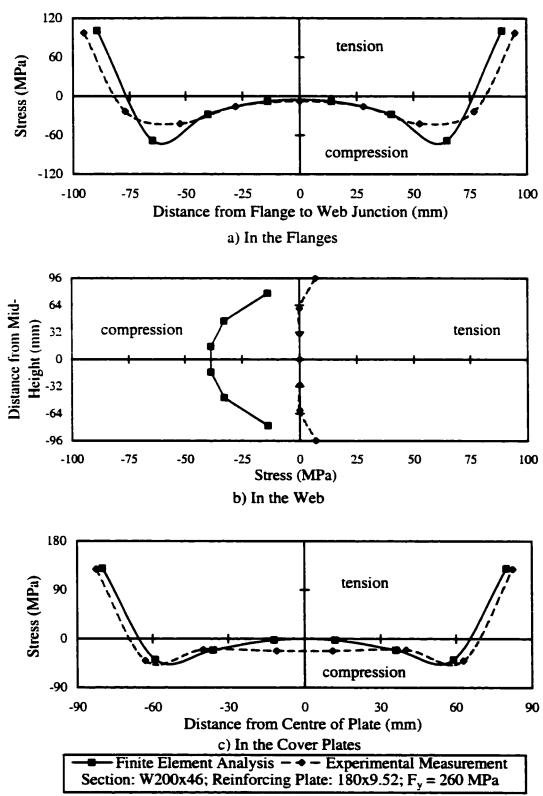


Figure 3.8 Welding Residual Stresses in the Column Reinforced under no Load

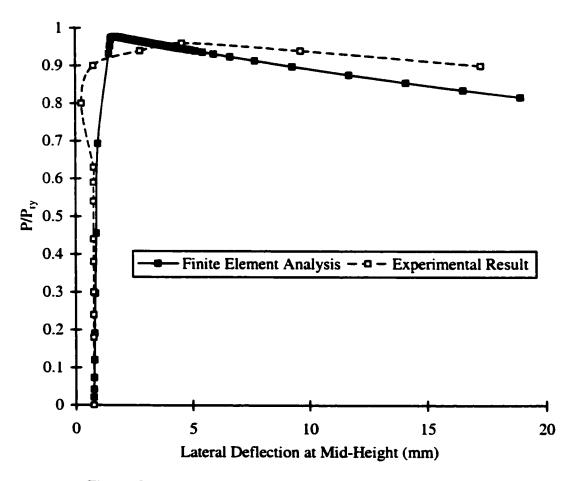


Figure 3.9 Axial Load versus Lateral Deflection Curves for the Column Reinforceed under no Load

Chapter 4

Parametric Study

4.1 General

Testing of full-scale columns is the most direct and reliable approach to examine the strength and behaviour of reinforced steel columns. However, because of the lack of previous test results and the impossibility of testing a large number of specimens to examine all the parameters that may affect the strength and behaviour of reinforced steel columns an alternative approach is desirable. A practical and expedient approach is to use the finite element analysis model presented and validated in Chapter 3 to expand the limited database of test results.

The first stage of this investigation will consist of identifying the parameters that affect the strength and behaviour of reinforced columns. A parametric study using a range of values for each parameter was then conducted. The database obtained from this parametric study provided sufficient information to perform a statistical analysis, which will be discussed in the next chapter. This chapter presents a description of buckling behaviour, the selection of the parameters affecting the strength and behaviour of reinforced columns, and the results of the parametric study.

4.2 Buckling Behaviour

A study of buckling modes was conducted using the finite element models presented in Chapter 3. Figure 4.1 shows the buckled shape of columns reinforced with plates parallel to the flanges. The figure illustrates buckling about the weak axis (Figure 4.1 (a)) and buckling about the strong axis (Figure 4.1 (b)). Figure 4.2 shows the buckled shape of columns reinforced with plates parallel to the web. Buckling about the weak axis is illustrated in Figure 4.2 (a) and buckling about the strong axis is illustrated in Figure 4.2 (b). For the specimens under investigation, the expected buckled shape always occurs as overall buckling about the weak axis of the reinforced column. For columns reinforced with plates parallel to the flanges the weak axis of the reinforced section always coincides with the weak axis of the rolled shape. When a column is reinforced

with plates parallel to the web, however, the weak axis of the reinforced section may coincide with the strong axis of the rolled shape. In order to trigger buckling about the strong axis of the reinforced section, some column models were restrained at both ends to provide rotational fixity in the weak direction of the reinforced section.

Unexpected local buckling of the reinforcing plates near the supports was found to govern the capacity of some columns reinforced with plates parallel to the web. Figure 4.3 shows the buckled shape of a W310x179 column reinforced with 350x25 mm plates parallel to the web and $\lambda = 0.4$. In order to prevent local buckling of the reinforcing plates, the thickness was increased three times near the ends. Figure 4.4 shows the resulting load versus mid-height lateral deflection curves for the above local buckling model and the corresponding overall buckling model. It can be observed that, for the example shown in Figure 4.4, the overall buckling capacity is about 10 percent higher than the local buckling capacity. Although local buckling failure is beyond the scope of this investigation, it is a possible failure mode that should be investigated experimentally.

4.3 Selection of Parameters

Many parameters were found to influence the strength and behaviour of steel columns, with or without reinforcement (Geschwindner *et al.*, 1994; Feder and Lee, 1959; Tall, 1961; Nagaraja Rao and Tall, 1963). The following parameters were selected for this investigation:

- 1. Reinforced column slenderness.
- 2. Residual stress pattern and magnitude before welding.
- 3. Magnitude of residual stresses after welding of reinforcing plates.
- 4. Initial out-of-straightness of the reinforced columns.
- 5. Magnitude of the load applied on the column during the reinforcing process.
- 6. Steel grades of the I-section and cover plates.
- 7. Orientation of the reinforcing plates.
- 8. Direction of the buckling of the reinforced columns.
- 9. I-section to cover plates area ratio.

In order to investigate the effect of these parameters on reinforced columns, short columns, intermediate, and long columns were analysed. The slenderness parameter, λ , of the reinforced columns was taken as 0.4, 1.0, 1.1, and 1.5 for the parametric study.

The effect of the magnitude and pattern of residual stresses in the rolled section and the reinforcing plates before welding was investigated for 12 cases: six residual stress patterns with two different magnitudes for each pattern. The magnitude of the peak residual stresses was taken as 0.3 and 0.1 times the yield strength of the material for the rolled section, and 0.3 and 0.15 times the yield strength of the material for the reinforcing plates. All 12 cases were expected to cover the full range of initial residual stresses before welding (Huber, 1956; Tall, 1961; Nagaraja Rao and Tall, 1963). In addition, two cases were used to investigate the effect of welding residual stresses. The peak residual stresses at the flange tip were taken as 70% of the yield strength as suggested by Nagaraja Rao and Tall [1963], or 100% of the yield strength as suggested by the work of Tall [1961] and Huber [1956].

After the cover plates are added on to the rolled section under preload, it is difficult to control the out-of-straightness of the reinforced columns. In the following work three values for the initial imperfections before reinforcing were mainly used to investigate the effect of the initial out-of-straightness of reinforced columns, namely, L/8000, L/2000, and L/1000. L/8000 was used to simulate very small initial imperfections. L/2000 is close to the mean value reported for columns of hot rolled wide flange shapes (Bjorhovde, 1988), and L/1000 is the maximum initial imperfection allowed by CSA/CAN-G40.20-92. For columns longer than 10 m, CSA standard G40.20-92 suggests a different allowable initial imperfection as follows:

- 1. The allowable initial imperfection is 10 mm when 10 m < L \le 14 m;
- 2. The allowable initial imperfection is [10+(L-14000)/1000] mm, when L > 14 m.

The pre-load on columns before reinforcing consists of dead loads and a portion of the design live loads. The magnitudes of the dead and live loads vary depending on the type of structure. In the parametric study, two pre-load magnitudes were selected, namely, 40% of the load-carrying capacity of the unreinforced column predicted using the SSRC column curve 2 and 60% of load carrying capacity of the unreinforced column.

Three combinations of steel grades of I-sections and cover plates were investigated: both the rolled section and the reinforcing plates have a yield strength of 260 MPa or 300 MPa, and the rolled shape has a yield strength of 230 MPa and the reinforcing plates have a yield strength of 350 MPa. The third combination is believed to represent a condition where the difference in grades is maximised.

The effect of reinforcing plate orientation was investigated for two cases: reinforcing plates parallel to the flanges and reinforcing plates parallel to the web. The effect of the buckling direction for reinforced columns was also investigated for two cases: column buckling about the weak axis of the rolled section and column buckling about the strong axis of the rolled section.

Nine different combinations of rolled section sizes and reinforcing plate sizes were selected in the analysis: W200x46 with 180x9.52 mm cover plates, similar to the test specimens used by Nagaraja and Tall (1962); W310x179 with four different cover plate sizes, namely 290x25 mm, 290x16 mm, 350x25 mm and 350x16 mm; and W150x30 with four different cover plate sizes, namely 130x5 mm, 130x8 mm, 175x5 mm and 175x8 mm. All the different cases investigated cover a range of W-shape to reinforcing plate area ratio from 2.71 to 5.83.

4.4 Analysis Results

With a total of 39 variables selected for nine parameters, the corresponding number of combinations for a full factorial design would equal 4x12x2x3x2x3x2x2x9 = 62208, which is too many. In order to reduce the total number of samples considerably, a fractional factorial design (Hines and Montgomery, 1972) was adopted for the parametric study. Table 4.1 summarizes the various combinations of parameters investigated in the parametric study. A total of 317 numerical models with different variables were analysed. Table A.1 of Appendix A presents the details concerning the geometric and material properties for each model investigated. Table B.1 of Appendix B summarises the

magnitude of the pre-loads used in each case and the analysis results compared with the load carrying capacity predicted using different SSRC column curves and CSA column curves for each case. The load carrying capacity is presented as a ratio of the peak load determined from the finite element analysis to the yield strength.

The following sections present a detailed discussion of the parametric study. The effects of the various parameters investigated on the strength and behaviour of reinforced steel columns are discussed.

4.5 Effect of Column Slenderness

It is commonly understood that the slenderness parameter is the most important factor affecting the strength of columns. Based on the definition of the slenderness parameter given in Section 2.1, different geometric and material properties result in different non-dimensional slenderness ratios for columns. The effect of slenderness parameters on the strength of 315 reinforced columns is illustrated in Figure 4.5 where the load carrying capacity is plotted against the slenderness parameter. Although the results show a fairly large scatter, the relationship between the slenderness parameter and the strength of the columns is obvious. The large scatter results from the large range of parameters investigated in the parametric study.

4.6 Effect of Residual Stresses

The residual stresses in reinforced steel columns were modelled in two different stages, namely, the residual stresses in the rolled section and reinforcing plates before welding and the residual stresses after welding the reinforcing plates to the rolled section. Both stages must be considered to obtain a representative residual stress distribution in the reinforced column.

4.6.1 Residual Stresses before Welding

The residual stresses in the rolled section and in the reinforcing plates before welding were investigated separately. Their effect on the strength and behaviour of the reinforced column are presented in the following.

Residual Stresses in the Rolled Section

The patterns and magnitudes of the initial residual stresses in the rolled section depend on factors related to the manufacturing process, as discussed in Chapter 2. The initial residual stress distributions vary from section to section. Based on an investigation presented by Huber [1956], four different initial residual stress patterns were selected in this research for the wide flange section. The peak compressive residual stresses selected for this investigation were taken as 30% and 10% of the yield strength of the rolled section. The 30% level is representative of rolled sections (Chen and Atsuta, 1976) and the 10% level represents a lower bound value.

Six different patterns of residual stresses in the rolled section and in the reinforcing plates were considered and illustrated in Figure 4.6. For each residual stress pattern, two magnitudes of the peak residual stresses in the wide flange sections were investigated. The first four and the sixth residual stress patterns presented in Figure 4.6 are studied in this section. The fifth pattern is discussed in the next section.

Two columns configurations were used to investigate the effect of initial residual stresses. The first configuration consisted of a W200x46 section reinforced with 180x9.5 mm plates parallel to the flanges and buckling about the weak axis of the rolled section. The second configuration consisted of a W310x179 section reinforced with 350x25 mm plates parallel to the web and buckling about the strong axis of the rolled section. Table 4.2 summarises the finite element analysis models used in this part of the investigation. A description of the initial residual stress pattern and magnitudes is presented in columns (2) to (4). The other parameters are kept constant and are summarised in Table A.1.

Figures 4.7 and 4.8 depict the residual stress distributions after welding in the cross-section of a W200x46 column reinforced with 180x9.5 mm plates. The maximum magnitude of residual stresses resulting from welding was taken as $1.0 \, \text{F}_y$ at the flange tips for each model. Figures 4.7 and 4.8 show that the residual stresses after welding in the reinforced section are very similar despite the significant difference in initial residual stress patterns and magnitudes.

Figures 4.9 and 4.10 show the axial load response for the eight reinforced columns described in Table 4.2. Except for residual stress pattern 4-1, all other initial residual stress patterns investigated resulted in the same behaviour and strength of the reinforced columns. Residual stress pattern 4-1 resulted in about a 7% reduction in strength compared to the other specimens investigated. A summary of the peak to yield strength ratio for each case investigated is presented in column (5) of Table 4.2. An examination of the analysis results for two W310x179 columns reinforced with 350x25 mm plates parallel to the web and buckling about the strong axis of the rolled section, as shown in Table 4.2, also indicates that initial residual stresses have little effect on the strength of reinforced steel columns.

Residual Stresses in the Cover Plates

In order to investigate the effect of initial residual stresses in the reinforcing plates, two peak magnitudes $(0.3F_y)$ and $(0.15F_y)$ were chosen for the initial residual stresses in the reinforcing plates based on investigations by Tall [1961] and Nagaraja Rao and Tall [1963]. Four W200x46 columns reinforced with 180x9.5 mm plates parallel to the flanges and buckling about the weak axis of the rolled section were used to investigate the effect of this parameter on the strength and behaviour of reinforced steel columns. The models used for this study are described in detail in Table 4.3.

Despite differences in initial residual stresses, the residual stress patterns and magnitudes after welding the reinforcing plates were essentially all the same. All the welding residual stress patterns for these four models are similar to the pattern shown in

Figure 4.11 (a). The axial load response of columns with different residual stress magnitude is presented in Figure 4.12. As expected from an examination of the residual stresses after welding, all the specimens display the same strength and behaviour. The ratios of peak load to yield load for these cases are summarized in Column (5) of Table 4.3. Since this parameter was found to have little effect, a typical initial stress pattern can therefore be used for the remaining part of this study. Pattern 1-3, illustrated in Figure 4.6, was selected for all the following numerical models.

4.6.2 Effect of the Magnitude of the Welding Residual Stresses

Nagaraja Rao and Tall (1963) have shown that high tensile residual stresses are developed at the flange tips as a result of welding reinforcing plates to a rolled W-shape. The distributions of the welding residual stresses were found to be very similar in the research, as shown in Figure 4.11. The magnitudes of these welding residual stresses were reported to be in the order of 70 percent of the yield strength of the material. Welding residual stresses equal to the yield strength of the material have also been reported elsewhere (Masubuchi, 1980). In order to cover the full range of possible welding residual stresses, a residual stress pattern was investigated with two residual stress magnitudes, namely 70 percent and 100 percent of the yield strength at the flange tips, as illustrated in Figure 4.11. The control parameter in the analysis is the magnitude of the residual stresses at the flange tips. The residual stresses in the remaining portions of the cross-section are governed by the size of the reinforced cross-section and the initial residual stresses in the reinforcing plates and the wide flange section.

This section presents the procedure used for four W310x179 columns reinforced with 290x16 mm plates parallel to the flanges and buckling about the weak axis of the rolled section (all samples used the same procedure). Two different values of the slenderness parameter, λ , were investigated, namely 1.1 and 1.5. The residual stress distribution in the reinforcing plates and the wide flange section before welding is pattern 1-3 of Figure 4-6. A description of the models used for this part of the investigation is given in Table 4.4 where column (4) lists the magnitude of the peak welding residual stress. A comparison of the load carrying capacity listed in column (5) shows that, for a given

slenderness parameter, the difference in the strength of columns with different welding residual stresses is negligibly small.

The effect of welding residual stress magnitude on the strength and the behaviour of reinforced columns is illustrated in Figure 4.13. As for the peak strength, the effect of welding residual stress magnitude on the strength and the behaviour of the reinforced steel columns is negligible. The following investigation therefore uses a representative peak welding residual stress of $1.0 \, F_y$ at the flange tips.

4.7 Effect of the Initial Out-of-straightness

The shape and magnitude of initial out-of-straightness in a reinforced column result from a combination of deformations. These deformations are the initial imperfection of the unreinforced rolled section resulting from the rolling process, the deformation resulting from the preload on the unreinforced column, and the deformation resulting from the welding process during reinforcement of the columns. Although the magnitude of initial out-of-straightness must be controlled in rolled shapes and other fabricated columns, current Canadian standards do not provide any specific requirement for the initial out-of-straightness in a reinforced column. However, CAN/CSA G40.20-92 specifies some limitations for the initial imperfection in unreinforced rolled sections, as described in Section 4.3. The effect of initial out-of-straightness in reinforced columns was therefore investigated in light of the limitations set for unreinforced columns.

The effect of initial out-of-straightness on the strength and behaviour of reinforced steel columns is illustrated using nine W310x179 columns reinforced with 290x25 mm plates parallel to the flanges. All the reinforced columns have their weak axes in the same direction as the weak axes of the rolled sections. Table 4.5 presents a description of the models used for this investigation. Three different values of slenderness were used in the columns, as shown in Column (3) of Table 4.5. In order to obtain different magnitudes of initial out-of-straightness in the reinforced columns, the magnitudes of the initial imperfections in the unreinforced columns were varied as shown in Column (4). Column (5) of Table 4.5 presents the initial out-of-straightness of the reinforced column for each

model. This out-of-straightness value was obtained following the removal of the axial load on the column after welding the plates to the column. It can be observed that the magnitude of the out-of-straightness increases after strengthening the column and this effect is more significant with slender columns than with short ones.

The predicted load carrying capacity of the reinforced columns is presented in column (6) of Table 4.5. It can be seen that the initial out-of-straightness for intermediate and long columns significantly affects column strength. The column strength decreases with increasing initial out-of-straightness magnitude. For example, for $\lambda = 1.1$, a change in the initial out-of-straightness from L/1790 to L/820 results in a decrease in load carrying capacity of 8.5%, as shown in the table. An increase in the initial out-of-straightness from L/7190 to L/820 results in a reduction in strength of 15%. A similar trend is observed in columns with $\lambda = 1.5$, but the strength of short columns ($\lambda = 0.4$) is not significantly affected by the magnitude of the initial out-of-straightness.

Figure 4.14 shows the axial load versus lateral deflection at mid-height for columns with different initial out-of-straightness after reinforcing for $\lambda = 1.1$. It can be observed that with increasing initial out-of-straightness of the reinforced columns, the lateral deflections at the peak load increase, and the load carrying capacity decreases.

4.8 Effect of the Pre-load

Six W310x179 columns reinforced with 290x25 mm plates parallel to the flanges were used to present the effect of pre-load on the strength and behaviour of reinforced columns. Their buckling axis was the weak axis of the rolled section. A description of the reinforced columns is presented in Table 4.6. Columns with two different preloads, namely 0.4 and 0.6 times the load carrying capacity of the unreinforced column predicted using the SSRC column curve 2, and three slenderness values ($\lambda = 0.4$, 1.1, and 1.5) were investigated. Columns (3) and (4) present the pre-load magnitude and the ratio of the pre-load to the load carrying capacity of the unreinforced column predicted using SSRC column curve 2, respectively. An examination of the predicted capacity presented in

column (5) of Table 4.6 indicates that the magnitude of the preload does not significantly affect the strength of reinforced columns.

Plots of the axial load versus axial deformation for different pre-load magnitudes are presented in Figure 4.15 for columns with λ of 1.1. It can be observed that the shapes of the curves are identical. The same observation can also be made for columns with λ of 0.4 and 1.5. The pre-load magnitude does not significantly affect the pre- and post-buckling behaviour of reinforced columns within the range of preload investigated. The following investigation was therefore carried out with a preload of 0.6 times the load carrying capacity of the unreinforced column predicted using the SSRC column curve 2.

4.9 Effect of Steel Grade

Columns in many older structures are either of grade A9 or A36 steel, a relatively low nominal yield strength compared to more modern structural steels that would typically be used for reinforcing plates. Reinforced steel columns may therefore be composite columns with different steel grades. In order to cover a broad range of these composite columns, columns with two different combinations of steel grades were investigated: 1) columns with the same steel grade for the plate and rolled section ($F_y = 300 \text{ MPa}$), which will serve as a reference, and; 2) reinforced columns with $F_y = 230 \text{ MPa}$ for the rolled section and $F_y = 350 \text{ MPa}$ for the plates.

Table 4.7 gives a description of the numerical models used to illustrate this investigation. The effect of material yield strength was studied for two different reinforcing plate orientations, buckling about the weak axis and buckling about the strong axis of the rolled section, and three different values for the slenderness parameter, λ . A comparison of the predicted load carrying capacities presented in column (5) of Table 4.7 indicates that varying the steel grades of the reinforced column does not significantly affect the strength of reinforced columns when the capacity is expressed as a ratio of the yield capacity of the cross-section.

4.10 Effect of Reinforcing Plate Orientation

Steel columns can be reinforced with steel plates either welded parallel to the flanges or parallel to the web in Figure 2.1. In order to investigate the effect of plate orientation on the strength and behaviour of reinforced steel columns, columns of different slenderness and different slenderness were modelled for buckling about either the strong or the weak axis. A summary of these models is presented in Table 4.8.

An examination of Table 4.8 reveals that short columns ($\lambda = 0.4$) are not affected by the orientation of the reinforcing plates. This is expected since short columns fail by yielding, rather than by buckling. For buckling about the weak axis of the W-shape section and $\lambda = 1.1$ and 1.5, it seems that columns are weaker when the reinforcing plates are parallel to the flanges. A reduction of strength of 7.5% to 10% is observed in the sample columns presented in Table 4.8. When the buckling axis is the strong axis of the strong W-shape section, columns with reinforcing plates parallel to the web are weaker than the columns with plates parallel to the flanges. A reduction in strength-to-yield ratio of about 7% is observed for the selected sample columns.

4.11 Effect of Buckling Axis

The investigation has so far focused on columns buckling about the weak axis of the reinforced column. When wide flange sections are reinforced with plates parallel to the flanges, the weak axis of the reinforced section coincides with the weak axis of the wide flange section. However, when the reinforcing plates are parallel to the web of the wide flange section, the weak axis of the reinforced section may be at right angle to the weak axis of the unreinforced section. On the other hand, buckling of a column may take place about the strong axis of the cross-section if the braced length in the weak axis direction is shorter than the braced length in the strong direction. In some cases presented in this section, additional bracing in the weak axis direction was provided to force the column to buckle about its strong axis.

Table 4.9 presents a summary of the columns used to illustrate the effect of the buckling axis. Columns reinforced with plates parallel to the flanges and with plates

parallel to the web were investigated. The direction of the buckling axis orientation relative to the unreinforced and reinforced sections are presented in columns (2) and (3), respectively. It is seen that the effect of plate orientation on the buckling capacity of columns varies according to column slenderness and the buckling direction relative to the unreinforced section major axis. For intermediate and long columns ($\lambda = 1.1$, 1.5) with reinforcing plates parallel to the flanges, buckling about the strong axis of the rolled section was observed to result in a larger strength-to-yield ratio than in the case of buckling about the weak axis. For intermediate and long columns with reinforcing plates parallel to the web, buckling about the strong axis of the rolled section was observed to result in a lower strength-to-yield ratio than buckling about the weak axis of the rolled section. This observation holds whether or not the strong axis of the reinforced section is in the same direction as the strong axis of the unreinforced section.

4.12 Effect of W-Shape to Plate Area Ratio

The effect of the ratio of the wide flange section area to the reinforcing plate area on the strength and behaviour of the reinforced columns was investigated for three non-dimensional slenderness ratios. For each slenderness ratio, the area ratio was varied either by: 1) changing the plate area while keeping the wide flange section constant; and, 2) by changing both the I-section and the reinforcing plate dimensions. The results of this investigation are presented in Table 4.10 and Table 4.11.

The results presented in Table 4.10 show that, despite a variation in the area ratio from 1.57 to 2.46, the predicted strength-to-yield ratio remained essentially the same for all three non-dimensional slenderness ratios. Table 4.11 shows a variation in area ratio from 1.57 to 2.92 obtained by changing both the size of the rolled section and the size of the reinforcing plates. Except for the columns with a slenderness ratio, λ , of 1.1, the change in capacity is insignificant. Although a significant change in capacity is observed for the column with $\lambda = 1.1$, considering the large change in area ratio used for these analyses, the strength of the columns is considered to be insensitive to the ratio of the rolled section area to the cover plate area.

4.13 Summary

This chapter presents some of the results of a parametric study that includes a total of 315 reinforced steel columns. The parameters investigated were column slenderness, residual stresses, initial out-of-straightness, preload magnitude, yield strength of the I-section and reinforcing plates material, plate orientation, buckling axis, and the I-section area to the cover plate area ratio. The following conclusions were drawn from this parametric study.

- The non-dimensional column slenderness, expressed as the slenderness parameter,
 λ, is the most important parameter affecting column strength.
- 2. While the initial residual stress is an important factor for the load carrying capacity of an unreinforced column, the investigation demonstrated that variations in the initial residual stresses, before welding the reinforcing plates, do not affect significantly the load carrying capacity of a reinforced column. The investigation also demonstrated that varying the maximum welding residual stress from 70% to 100% of the yield strength of the materials does not affect significantly the predicted strength of reinforced columns.
- Initial out-of-straightness affects the behaviour and strength of intermediate and long reinforced columns significantly. The strength of reinforced columns decreases as the initial out-of-straightness increases.
- 4. A change in the preload from 40% to 60% of the load carrying capacity of the unreinforced column does not affect the behaviour and the predicted strength-to-yield strength ratio of reinforced columns significantly.
- 5. The use of different grades in reinforced columns was found to have a negligible effect on the strength-to-yield ratio of reinforced columns.
- 6. In the numerical model results, it was observed that the interaction of the plate orientation and the buckling axis affects the behaviour and the strength of

intermediate and long columns significantly. For columns of same slenderness and reinforced with plates parallel to the web the capacity of the column is larger when buckling occurs about the weak axis of the unreinforced section. The converse was observed when columns are reinforced with plates parallel to the flanges. Intermediate and long columns buckling about the strong axis of the rolled section, have a higher strength-to-yield ratio when the reinforcing plates are parallel to the flanges compared to columns reinforced with plates parallel to the web. The converse was observed for intermediate and long columns with reinforcing plates parallel to the web.

7. The effect of I-section to reinforcing plate area ratio on the predicted strength-to-yield ratio was found to be insignificant.

Table 4.1 Fractional Factorial Design and Designations of the Numerical Models."

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Table 4.1 (cont'd)

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a) The number in the table refers to the finite element analysis model number described in detail in Appendix A and Appendix B.

b) λ - Slenderness parameter of the reinforced column

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c) δ_0 - Initial imperfection of the unreinforced column presented by a ratio of the imperfection to the column length, L.

d) The allowable maximum initial imperfection varies for the column longer than 10 m. $\dot{}$ c) IRS - Initial residual stress before welding

f) WRS - Residual stress after welding g) P_0 - Preload.

 $P_{u2}\,$ - Load carrying capacity of the unreinforced column predicted using the SSRC column curve 2

h) $F_y\,$ - Yield strength of the reinforced column

A - Yield strength of the rolled section and the cover plates = 260 MPa.

B - Yield strength of the rolled section and the cover plates = 300 MPa.

C - Yield strength of I-section = 230 MPa and yield strength of the cover plates = 350 MPa.

i) W - The weak axis of the rolled section

S - The strong axis of the rolled section

Table 4.2 Models Used to Study the Effect of Initial Residual Stresses in the I-Section before Welding

FEA model	Initial Res	idual Stresses befo	ore Welding	5 m c
Number ^a	PS ^b	MF ^c	MP^d	P _{fea} /P _{ry} e
(1)	(2)	(3)	(4)	(5)
W200x46 column	with 180x9.52	mm plates parallel	to the flanges	
Buckling about th	e weak axis of t	he rolled section		
3	1-1	$0.3F_{y}$	$0.15F_y$	0.65
5	1-2	$0.1F_{y}$	$0.15F_{v}$	0.65
7	2-1	$0.3F_{y}$	$0.15F_{v}$	0.63
8	2-2	0.1F _v	$0.15F_{v}$	0.66
9	3-1	$0.3F_{v}$	0.15F _v	0.65
10	3-2	$0.1F_{v}$	$0.15F_{v}$	0.66
11	4-1	$0.3F_{v}$	$0.15F_{v}$	0.60
12	4-2	$0.1F_{y}$	$0.15F_{y}$	0.65
W310x179 colum	nn with 350x25 r	nm plates parallel	to the web	· · · · · · · · · · · · · · · · · · ·
Buckling about th	ne strong axis of	the rolled section		
82	3-3	0.3F _y	0.3F _y	0.54
83	3-4	0.1F _y	0.3F _v	0.54

a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B.

b) PS - Designation of the residual stress pattern

c) MF - Magnitude of the initial residual stresses before welding at the flange tips

d) MP - Magnitude of the initial residual stresses before welding at the plate edges F_{ν} - Yield stress of the rolled section column

e) P_{fea} - load carrying capacity of the reinforced column obtained from the finite element analysis

P_{ry} - Yield strength of the reinforced column

Table 4.3 Models Used to Study the Effect of Initial Residual Stresses in the Cover Plates before Welding

FEA model	Initial Resi	dual Stresses before	ore Welding	- D /D C
Number ^a	PS ^b	MF ^c	MP^d	P_{fea}/P_{ry}^e
(1)	(2)	_ (3)	(4)	(5)
W200x46 column	with 180x9.52	mm plates parallel	to the flanges	
Buckling about the	e weak axis of the	he rolled section	_	
3	1-1	$0.3F_{y}$	0.15F _y	0.65
4	1-3	$0.3F_{v}$	$0.3F_{v}$	0.65
5	1-2	$0.1F_{v}$	$0.15F_{y}$	0.65
6	1-4	$0.1F_{v}$	$0.3F_{v}$	0.65

- a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B.
- b) PS Designation of the initial residual stress pattern
- c) MF Magnitude of the initial residual stresses before welding at the flange tips
- d) MP Magnitude of the initial residual stresses before welding at the plate edges F_{ν} Yield stress of the unreinforced column
- e) P_{fea} load carrying capacity of the reinforced column obtained from the finite element analysis
 - P_{ry} Yield strength of the reinforced column

Table 4.4 Models Used to Study the Effect of Varying Welding Residual Stress Magnitude

FEA	Column	Slenderness	Welding	
model	Length	Parameter	Residual	P_{fea}/P_{ry}^{c}
Number ^a	L (mm)	λ	Stress ^b	
(1)	(2)	(3)	(4)	(5)
W310x179 colum	n with reinforcing	plates 290x16 paralle	el to the flanges	
Buckling about th	e weak axis of the	rolled section	_	
90	7197	1.1	F _y	0.56
		1.1 1.1	F _y 0.7F _y	0.56 0.57
90	7197	1.1 1.1 1.5	,	

- a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B.
- b) F_y Yield stress of the steel of the rolled section
- c) P_{fea} load carrying capacity of the reinforced column obtained from the finite element analysis
 - P_{ry} Yield strength of the reinforced column

Table 4.5 Models Used to Study the Effect of the Initial Out-of-straightness

		_			
FEA	Column	Slenderness		- •	
model	Length	Parameter	$\delta_0^{\ b}$	$\delta_{\mathfrak{i}}^{\ \mathbf{c}}$	P_{fex}/P_{ry}^{d}
Number ^a	L (mm)	λ			
(1)	(2)	(3)	(4)	(5)	(6)
W310x179 colu	mn with 290x2	5 mm plates paral	lel to the flangs		-
Buckling about	the weak axis o	of the rolled section	n		
17	2631	0.4	L/8000	L/7850	1.00
18	2631	0.4	L/2000	L/1930	0.99
19	2631	0.4	L/1000	L/970	0.97
23	7235	1.1	L/8000	L/7190	0.64
24	7235	1.1	L/2000	L/1790	0.60
25	7235	1.1	L/1000	L/820	0.56
30	9866	1.5	L/8000	L/6270	0.42
31	9866	1.5	L/2000	L/1570	0.38
32	9866	1.5	L/1000	L/790	0.35

a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B.

b) δ_0 - Initial imperfection of the unreinforced rolled section column

L - The column length

c) δ_i - Initial out-of-straightness of the reinforced column

d) P_{fea} - load carrying capacity of the reinforced column obtained from the finite element analysis

P_{rv} - Yield strength of the reinforced column

Table 4.6 Models Used to Study the Effect of the Preload

FEA	Slenderness	Preid	oad, Po	
model	Parameter	P ₀	P ₀ /P _{u2} ^b	P_{fea}/P_{ry}^{c}
Number ^a	λ	(kN)	10/1 u2	
(1)	(2)	(3)	(4)	(5)
W310x179 colum	in with 290x25 mm pl	lates parallel to th	e flanges	
Buckling about th	e weak axis of the rol	led section		
19	0.4	3760	0.6	0.97
20	0.4	2507	0.4	0.98
25	1.1	2152	0.6	0.56
27	1.1	1435	0.4	0.57
32	1.5	1401	0.6	0.35
33	1.5	934	0.4	0.36

a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B. b) P_{u2} - Load carrying capacity of the rolled section (predicted using SSRC curve 2)

c) P_{fea} - load carrying capacity of the reinforced column obtained from the finite element analysis

P_{rv} - Yield strength of the reinforced column

Table 4.7 Models Used to Study the Effect of the Steel Grades

FEA	Slenderness	Yield Strength	, F _y (MPa)	
model	Parameter	Rolled Section	Distan	P_{fea}/P_{rv}^{b}
Number ^a	λ	Rolled Section	Plates	•
(1)	(2)	(3)	(4)	(5)
W310x179 colur	nn with 290x25 mm	plates parallel to the	flanges	
Buckling about the	he weak axis of the	rolled section		
19	0.4	300	300	0.97
21	0.4	230	350	0.96
25	1.1	300	300	0.56
28	1.1	230	350	0.51
32	1.5	300	300	0.35
34	1.5	230	350	0.34
W310x179 colur	nn with 290x25 mm	plates parallel to the	flanges	
Buckling about t	he strong axis of the	rolled section		
36	0.4	300	300	0.93
38	0.4	230	350	0.94
42	1.1	300	300	0.61
45	1.1	230	350	0.57
47	1.5	300	300	0.40
49	1.5	230	350	0.37
W310x179 colur	nn with 350x16 mm	n plates parallel to the	web	
Buckling about t	he weak axis of the	rolled section		
145	0.4	300	300	0.95
147	0.4	230	350	0.97
151	1.1	300	300	0.60
154	1.1	230	350	0.62
156	1.5	300	300	0.40
158	1.5	230	350	0.40
W310x179 colu	mn with 350x16 mn	n plates parallel to the	web	
Buckling about t	the strong axis of the	e rolled section		
160	0.4	300	300	0.93
162	0.4	230	350	0.89
166	1.1	300	300	0.56
169	1.1	230	350	0.51
171	1.5	300	300	0.38
173	1.5	230	350	0.36

<sup>a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B.
b) P_{fea} - load carrying capacity of the reinforced column obtained</sup>

b) P_{fea} - load carrying capacity of the reinforced column obtained from the finite element analysis

P_{ry} - Yield strength of the reinforced column

Table 4.8 Models Used to Study the Effect of Reinforced Plate Orientations

FEA	Cover	Column	Slenderness		
model	Plates	Length	Parameter	$\delta_i^{\ c}$	P_{fea}/P_{ry}^{d}
Number ^a	Orientation ^b	L (mm)	λ	•	-
(1)	(2)	(3)	(4)	(5)	(6)
W310x179 col	umn with 350x16	mm plates and	l buckling about th	ne weak axis of	the I-section
115	F	2827	0.4	L/974	0.98
307	W	3720	0.4	L/979	0.97
121	F	7772	1.1	L/786	0.56
308	W	10229	1.1	L/789	0.62
126	F	10598	1.5	L/788	0.37
156	W	13948	1.5	L/791	0.40
W310x179 col	umn with 350x16	mm plates and	l buckling about th	he strong axis o	f the I-section
130	F	4928	0.4	L/974	0.94
160	W	4155	0.4	L/995	0.93
135	F	13551	1.1	L/1582	0.63
309	W	11425	1.1	L/1574	0.59
310	F	18479	1.5	L/1249	0.41
171	W	15579	1.5	L/1245	0.38

a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B.

b) F - Cover plates parallel to the flanges

W - Cover plates parallel to the web

c) δ_i - Initial out-of-Straightness of the reinforced column

L - The column length

d) P_{fea} - load carrying capacity of the reinforced column obtained from the finite element analysis

P_{ry} - Yield strength of the reinforced column

Table 4.9 Models Used to Study the Effect of the Buckling Axis

FEA	Bucklin	g Axis	Column	Slenderness	· ··· <u>-</u> -	
model	Unreinforced	Reinforced	Length	Parameter	δ_i^{b}	P_{fea}/P_{ry}^{c}
Number	Section	Section	(mm)	λ		
(1)	(2)	(3)	(4)	(5)	(6)	(7)
W150x30 co	lumn with 130:	x5 mm plates	parallel to the	e flanges		
175	Weak axis	Weak axis	1236	0.4	L/973	0.97
192	Strong axis	Strong axis	2300	0.4	L/964	0.95
181	Weak axis	Weak axis	3399	1.1	L/885	0.60
311	Strong axis	Strong axis	6326	1.1	L/858	0.63
186	Weak axis	Weak axis	4635	1.5	L/848	0.37
312	Strong axis	Strong axis	8626	1.5	L/808	0.44
W310x179 c	column with 35	0x25 mm plate	es parallel to	the web		
313	Weak axis	Weak axis	4103	0.4	L/983	0.96
66	Strong axis	Strong axis	4030	0.4	L/984	0.93
314	Weak axis	Weak axis	11281	1.1	L/985	0.63
72	Strong axis	Strong axis	11083	1.1	L/1000	0.54
315	Weak axis	Weak axis	15383	1.5	L/1194	0.41
77	Strong axis	Strong axis	15113	1.5	L/1200	0.37
W310x179	column with 35	0x16 mm plate	es parallel to	the web		
307	Weak axis	Weak axis	3720	0.4	L/974	0.97
160	Strong axis	Strong axis	4155	0.4	L/995	0.93
316	Weak axis	Weak axis	10229	1.1	L/1862	0.68
165	Strong axis	Strong axis	11425	1.1	L/1870	0.60
317	Weak axis	Weak axis	13948	1.5	L/1227	0.42
171	Strong axis	Strong axis	15579	1.5	L/1235	0.38

a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B.

b) δ_i - Initial out-of-Straightness of the reinforced column

L - The column length

c) P_{fea} - load carrying capacity of the reinforced column obtained from the finite element analysis

P_{rv} - Yield strength of the reinforced column

Table 4.10 Models Used to Study the Effect of Cover Plate Size

FEA		- · · · ·		Column	Slenderness	
model	I-Section	Plate	I-section Area	Length	parameter	P _{fea} /P _{ry} b
numbera			Plate Area	(mm)	λ	
(1)	(2)	(3)	(4)	(5)	(6)	(7)
19	W310x179	290x25	1.57	2631	0.4	0.97
84	W310x179	290x16	2.46	2617	0.4	0.98
115	W310x179	350x16	2.04	2827	0.4	0.98
25	W310x179	290x25	1.57	7235	1.1	0.56
90	W310x179	290x16	2.46	7197	1.1	0.56
121	W310x179	350x16	2.04	7772	1.1	0.56
32	W310x179	290x25	1.57	9866	1.5	0.35
95	W310x179	290x16	2.46	9813	1.5	0.36
126	W310x179	350x16	2.04	10598	1.5	0.36

a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B.

Table 4.11 Models Used to Study the Effect of the Size of the I-section

FEA model	I-Section	Plate	I-section Area	Column Length	Slenderness parameter	P _{fea} /P _{rv} ^b
Number ^a			Plate Area	L (mm)	λ	ica ty
(1)	(2)	(3)	(4)	(5)	(6)	(7)
19	W310x179	290x25	1.57	2631	0.4	0.97
175	W150x30	130x5	2.92	1236	0.4	0.97
25	W310x179	290x25	1.57	7235	1.1	0.56
181	W150x30	130x5	2.92	3399	1.1	0.60
32	W310x179	290x25	1.57	9866	1.5	0.35
186	W150x30	130x5	2.92	4635	1.5	0.37

a) The FEA model number refers to the finite element analysis model number described in detail in Appendix A and Appendix B.

P_{ry} - Yield strength of the reinforced column

b) P_{fea} - load carrying capacity of the reinforced column obtained from the finite element analysis

P_{ry} - Yield strength of the reinforced column

b) P_{fea} - load carrying capacity of the reinforced column obtained from the finite element analysis

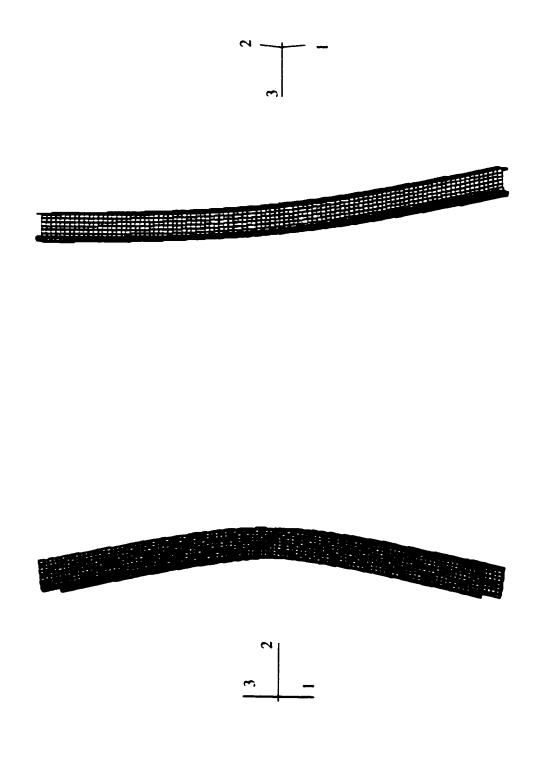


Figure 4.1 Deformed Shape of Columns Reinforced with Plates Parallel to the Flanges

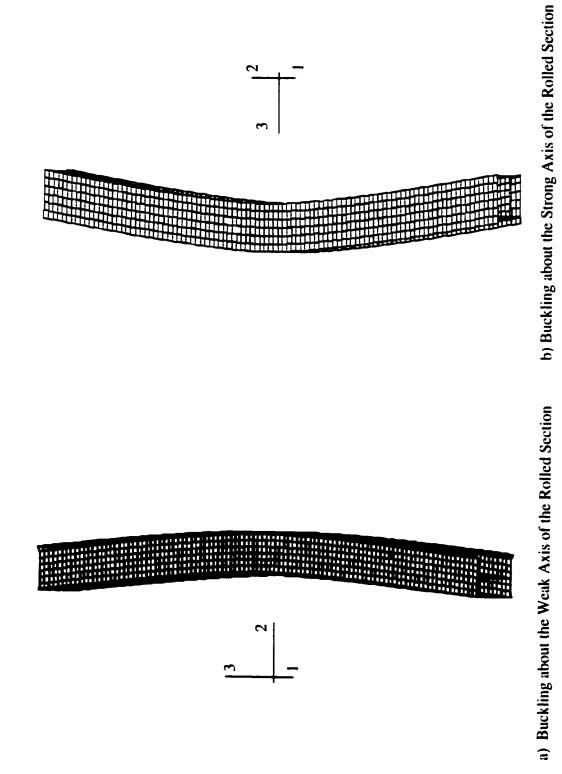
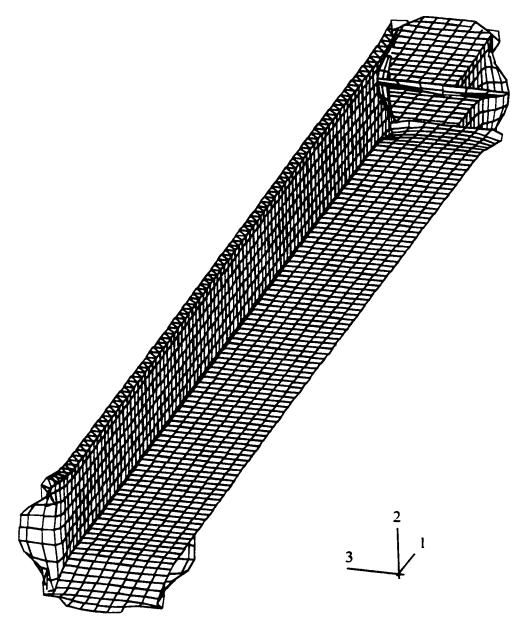


Figure 4.2 Deformed Shape of Columns Reinforced with Plates Parallel to the Web



Section: W310x179; Reinforced Plate: 350x25 mm;

 $F_y = 300 \text{ MPa}; \qquad \lambda = 0.4$

Figure 4.3 Local Buckled Shape of the Column Reinforced with Plates Parall to the Web

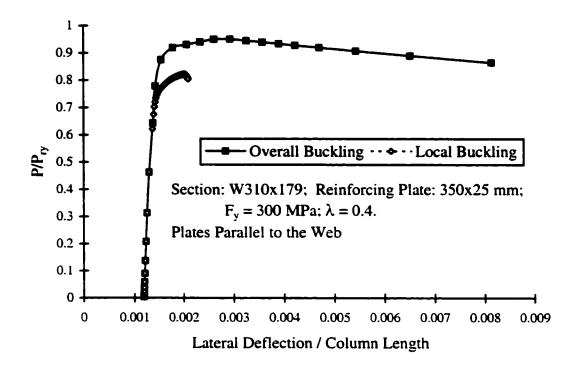


Figure 4.4 Load versus Lateral Deflection Curves for the Columns with Overall Buckling and Local Buckling

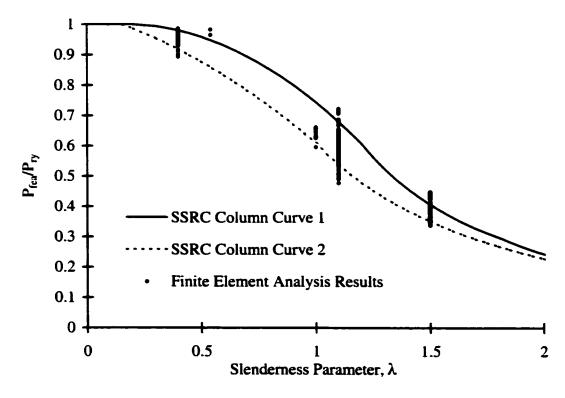


Figure 4.5 Strength of All Reinforced Column Samples

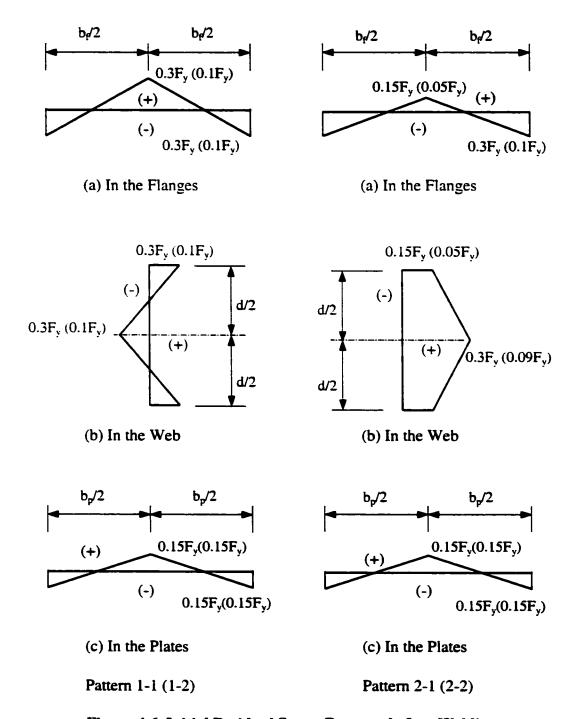


Figure 4.6 Initial Residual Stress Patterns before Welding

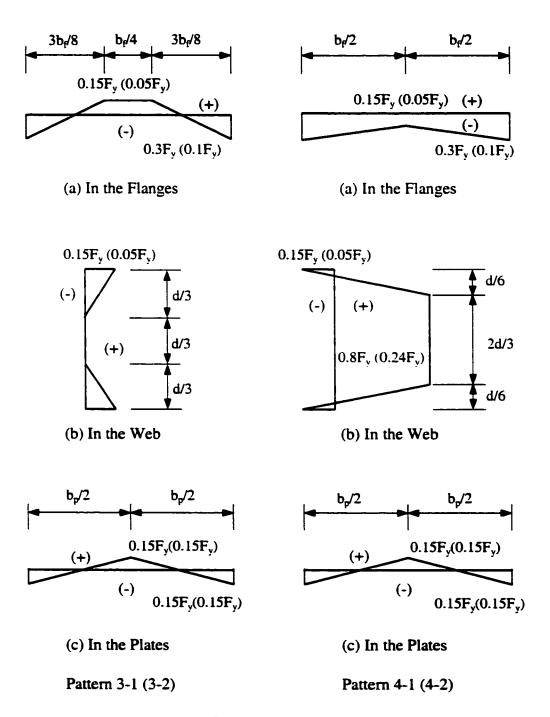


Figure 4.6 (Cont'd)

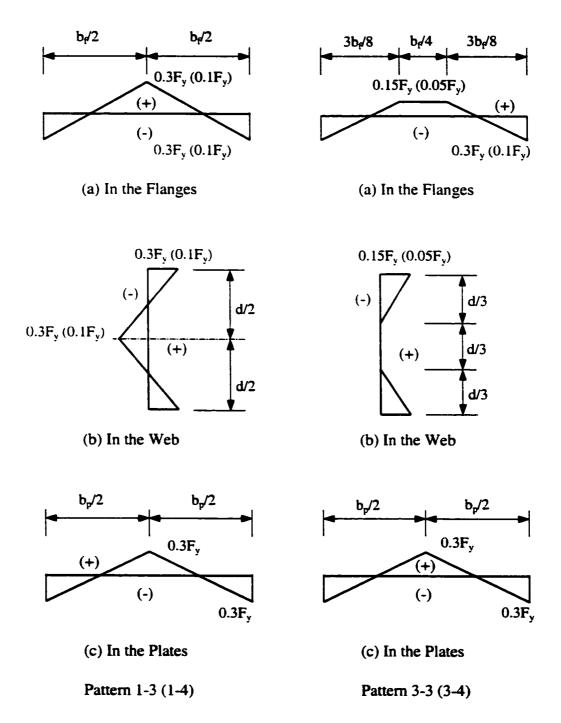


Figure 4.6 (Cont'd)

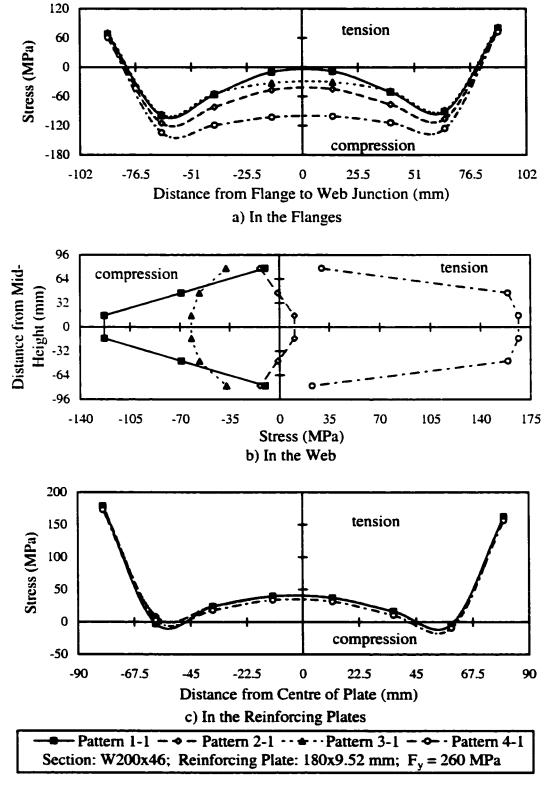


Figure 4.7 Residual Stress Distributions after Welding for Maximum Initial Residual Stress of 0.3 F_y

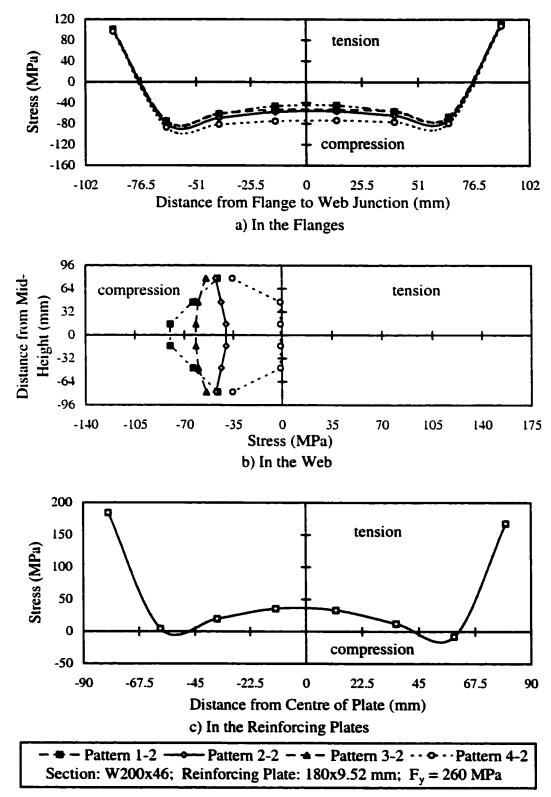


Figure 4.8 Residual Stress Distributions after Welding for Maximum Initial Residual Stress of $0.1F_y$

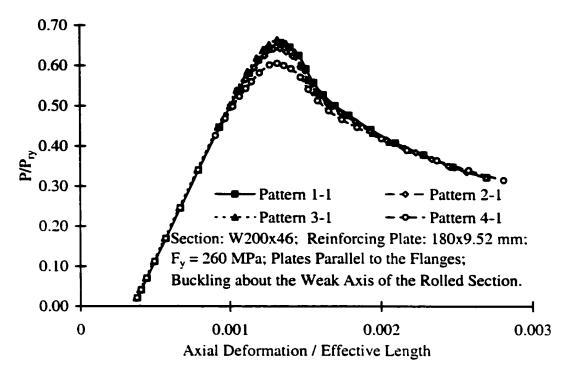


Figure 4.9 Effect of Initial Residual Stress Patterns for Maximum Magnitude of 0.3F_v

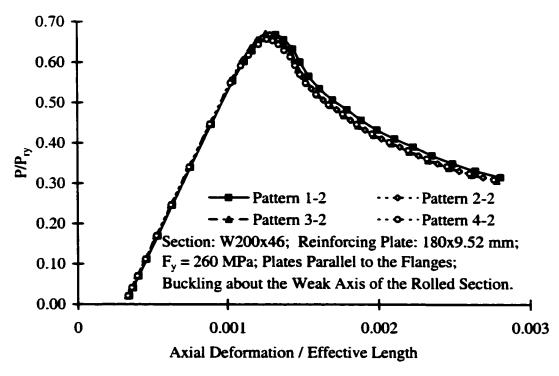


Figure 4.10 Effect of Initial Residual Stress Patterns for Maximum Magnitude of $0.1F_y$

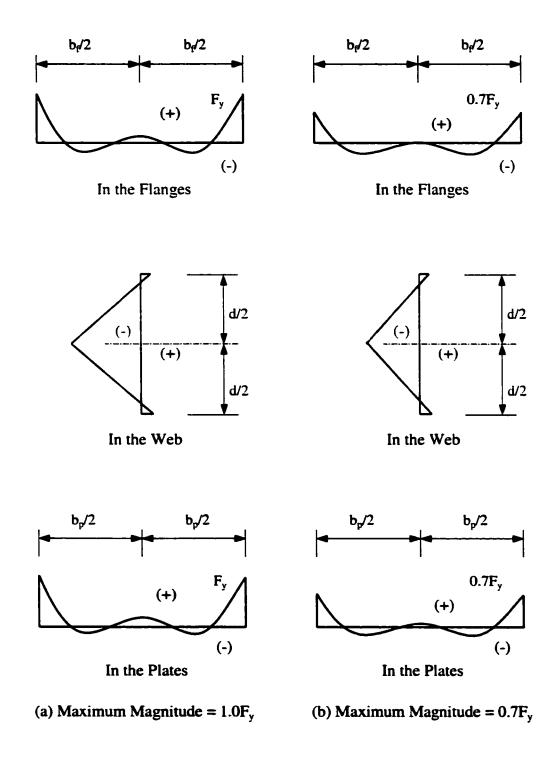


Figure 4.11 Residual Stress Patterns after Welding

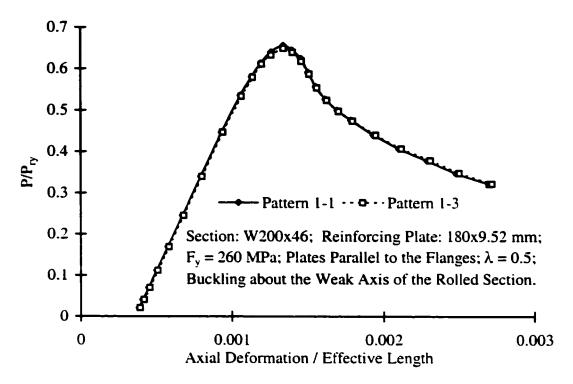


Figure 4.12 Effect of Varying Initial Residual Stress Patterns before Welding in the Cover Plates

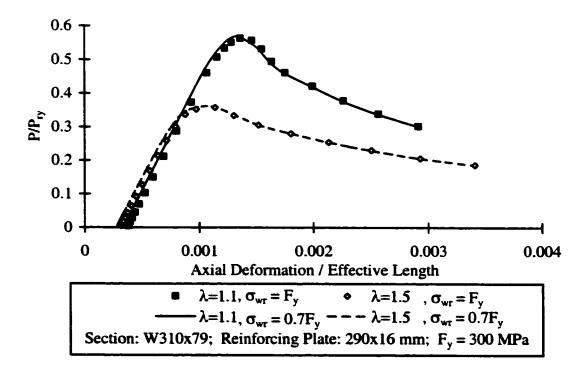


Figure 4.13 Effect of Varying Welding Residual Stresses with Different Slenderness Ratios

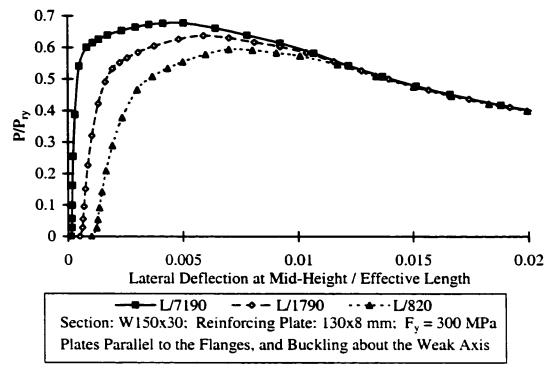


Figure 4.14 Effect of the Initial Out-of-straightness ($\lambda = 1.1$)

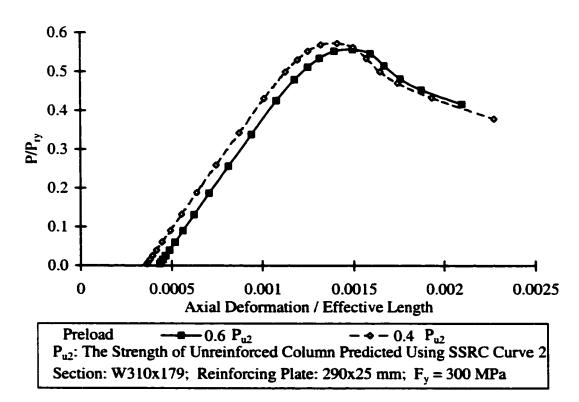
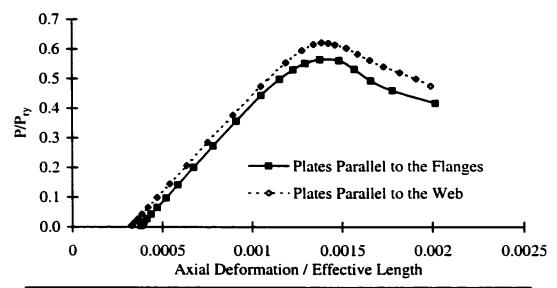


Figure 4.15 Effect of the Preload Magnitudes (λ =1.1)



Section: W310x179; Reinforcing Plate: 350x16 mm; $F_y = 300$ MPa Columns Buckle about the Weak Axis of the Rolled Section.

Figure 4.16 Effect of the Reinforcing Plate Orientation ($\lambda = 1.1$)

Chapter 5

Limit States Design

5.1 Background

A statistical-based design philosophy, which provides a uniform level of safety for various structural components, is used for the design of steel structures in Canada. Considering the variation in the resistance and load effects, corresponding resistance and load factors are determined using statistical analysis. The resistance factor for steel columns reinforced with welded steel plates is the prime concern in this research.

5.1.1 Column Resistance Based on CAN/CSA-S16.1-94

Based on the investigation conducted by Bjorhovde (1972), three strength curves have been developed to predict the strength of steel columns of different shapes and types (Johnston, 1976). The Structural Stability Research Council proposed the equations used to describe these column curves, and these three curves are therefore called SSRC column curves. CSA standard CAN3-S16.1-M84 - "Steel Structures for Building – Limit States Design" (Canadian Standards Association, 1984) adopted the first two SSRC curves for the design of steel columns. The equations for the first two SSRC curves are in five parts and are expressed as follows:

SSRC column curve 1

(1) For
$$0 \le \lambda \le 0.15$$
 $C_r = \phi A F_y$ (stub column)

(2) For
$$0.15 \le \lambda \le 1.2$$
 $C_r = \phi A F_y (0.990 + 0.122\lambda - 0.367\lambda^2)$

(3) For
$$1.2 \le \lambda \le 1.8$$
 $C_r = \phi A F_v (0.051 + 0.801 \lambda^{-2})$ [5.1]

(4) For
$$1.8 \le \lambda \le 2.8$$
 $C_r = \phi A F_y (0.008 + 0.942 \lambda^{-2})$

(5) For
$$2.8 \le \lambda$$
 $C_r = \phi A F_v \lambda^{-2}$

SSRC column curve 2

(1) For
$$0 \le \lambda \le 0.15$$
 $C_r = \phi A F_v$ (stub column)

(2) For
$$0.15 \le \lambda \le 1.0$$
 $C_r = \phi A F_v (1.035 - 0.202\lambda - 0.222\lambda^2)$

(3) For
$$1.0 \le \lambda \le 2.0$$
 $C_r = \phi A F_y (-0.111 + 0.636 \lambda^{-1} + 0.087 \lambda^{-2})$ [5.2]

(4) For
$$2.0 \le \lambda \le 3.6$$
 $C_r = \phi A F_y (0.009 + 0.877 \lambda^{-2})$

(5) For
$$3.6 \le \lambda$$
 $C_r = \phi A F_v \lambda^{-2}$

where Part (5) of equations [5.1] and [5.2] corresponds to the Euler's elastic buckling resistance. SSRC column curve 1 represents a higher strength than SSRC column curve 2. In the Canadian standard, SSRC columns curve 1 is adopted for the hollow structural shape of Class H, which are sections that are hot formed or cold formed followed by stress relieving, and for welded wide flange sections with flanges made of flame cut plates. These particular sections possess higher strength in compression because of the more favourable residual stress pattern present in these sections. All other sections are designed based on SSRC column curve 2. It should be noted that Equations [5.1] and [5.2] provide a factored resistance, with the resistance factor, ϕ , taken as 0.9 for columns.

In 1995, Loov proposed a double exponential equation using a single parameter, n, to replace the five-part equations proposed by SSRC. This expression, which was adopted by the CSA standard CAN/CSA-S16.1-94 (Canadian Standards Association, 1994), takes the following form:

$$C_r = \phi A F_v (1 + \lambda^{2n})^{-1/n}$$
 [5.3]

where n = 2.24 for CSA column curve 1, corresponding to SSRC curve 1, and n = 1.34 for CSA column curve 2, corresponding to SSRC curve 2. It was demonstrated that this expression never deviates by more than approximately 3% from the corresponding values given by Equations [5.1] and [5.2] (Loov, 1996).

5.1.2 Principles of Limit States Design

Limit states design is a design method that requires the structure not to exceed the limit states that govern its strength and behaviour for any realistic load or load combinations. There are basically two categories of limit states that are pertinent to the structural design process: ultimate limit states (ULS) and serviceability limit states (SLS). Ultimate limit states deal with strength conditions for the structure. Exceeding an ULS

implies a local or overall structural failure. On the other hand, exceeding a serviceability limit state means that a structure is not behaving or serving in the way it was intended to. The SLS therefore considers the performance under normal operating conditions. In the design of columns, serviceability limit states are seldom a concern. The following, therefore, focuses on ultimate limit states.

The general criterion for an ultimate limit state can be expressed in the following form:

$$\phi R \ge \alpha S$$
 [5.4]

where R is the nominal resistance, ϕ is the resistance factor, S is the nominal value of the load effect, and α ' is the load factor.

It is clearly understood that there is always a possibility that failure will occur. In order to ensure that the probability of failure is acceptably small, the factors α ' and ϕ have to be set at a suitable value by applying the principles of probability theory to the statistical analysis of the load effects and the resistance.

Figure 5.1 shows possible distribution curves for the load effect, S, and the resistance, R. The variables are assumed to be statistically independent. Galambos and Ravindra (1973a) combined the two curves to produce a risk frequency distribution curve, as illustrated in Figure 5.2. The probability of failure is equivalent to the probability of the ratio R/S being less than 1.0 (the load effect exceeding the resistance), or the natural log of (R/S) being less than 0. This probability of failure is therefore a function of the distance $\beta\sigma_{ln(R/S)}$ shown in Figure 5.2, which provides the margin of safety. The factor β is called the safety index and $\sigma_{ln(R/S)}$ is the standard deviation of the natural log of (R/S). The probability of failure can be set at any desired level by selecting an appropriate value of β . The safety index is, therefore, a measure of the safety or reliability of the structure. Galambos and Ravindra (1973a) proposed a first order simplification method to express the safety index, β , in algebraic form. In Figure 5.2,

$$\sigma_{\ln(R/S)}^2 = \left(\frac{\partial \overline{\ln(R/S)}}{\partial \overline{R}}\right)^2 \cdot \sigma_R^2 + \left(\frac{\partial \overline{\ln(R/S)}}{\partial \overline{S}}\right)^2 \cdot \sigma_S^2 = \frac{\sigma_R^2}{\overline{R}^2} + \frac{\sigma_S^2}{\overline{S}^2} = V_R^2 + V_S^2$$
 [5.5]

Thus,

$$\beta (V_R^2 + V_S^2)^{1/2} = \ln \overline{R/S}$$
 [5.6]

from which,

$$\overline{R/S} = e^{\beta(V_R^2 + V_S^2)^{\frac{1}{2}}}$$
 [5.7]

Then,

$$\beta = \frac{\ln \frac{\overline{R}}{\overline{S}}}{(V_R^2 + V_S^2)^{1/2}}$$
 [5.8]

Allen (1975) proposed a more accurate expression for the safety index as follows:

$$\beta = \frac{\ln \left[\frac{\overline{R}}{\overline{S}} \left(\frac{1 + V_S^2}{1 + V_R^2} \right)^{\frac{1}{2}} \right]}{\ln \left[\left(1 + V_S^2 \right) \left(1 + V_R^2 \right) \right]^{\frac{1}{2}}}$$
 [5.9]

Based on the investigation of Allen (1975) and the work done by Galambos and Ravindra (1973b), a value of 3.0 was adopted for β for most members of building structures in Canada.

Lind (1971) proposed an approximate equation as follows:

$$(V_R^2 + V_S^2)^{1/2} = \alpha (V_R + V_S)$$
 [5.10]

where α is called a separation variable. Galambos and Ravindra (1973b) extended this concept further by introducing two separation variables, α_R and α_S , such that:

$$(V_R^2 + V_S^2)^{1/2} = \alpha_R V_R + \alpha_S V_S$$
 [5.11]

Galambos and Ravindra (1973b, 1977) also used an error minimization process to demonstrate that a single value of $\alpha = 0.55$ could be used for the conservative approximate equation [5,10], leading to an acceptably small error. Substituting Equation [5.10] into Equation [5.8] results in the following expression for the safety index:

$$\beta = \frac{\ln \frac{\overline{R}}{\overline{S}}}{\alpha(V_R + V_S)}$$
 [5.12]

Solving for the mean value of resistance, \overline{R} , we obtain

$$\overline{R} = \overline{S} e^{\beta \alpha (V_R + V_S)}$$
 [5.13]

which can be rewritten as follows

$$\overline{R} e^{-\beta \alpha V_R} = \overline{S} e^{\beta \alpha V_S}$$
 [5.14]

Equation [5.14] relates to the mean values of the resistance and load effect. If the ratio of the mean to nominal value of the resistance (also called the bias coefficient for the resistance) is expressed as

$$\rho_{R} = \frac{\overline{R}}{R}$$
 [5.15]

and the ratio of the mean to nominal value of the load effect (the bias coefficient of the load effect) is expressed as

$$\rho_{\rm S} = \frac{\bar{\rm S}}{\rm S} \tag{5.16}$$

Equation [5.14] can be rewritten as follows

$$\rho_{R} \cdot e^{(-\beta \alpha V_{R})} \cdot R = \rho_{S} \cdot e^{(\beta \alpha V_{S})} \cdot S$$
 [5.17]

from which, compared with [5.4], the resistance factor, ϕ , can be defined as

$$\phi = \rho_R \cdot e^{(-\beta \alpha V_R)}$$
 [5.18]

and the load effect factor, α' , can be defined as

$$\alpha' = \rho_S \cdot e^{(\beta \alpha V_S)}$$
 [5.19]

The ratio of the mean to the nominal resistance, ρ_R , of a member consists of three parts: the ratio of the mean to the nominal cross-sectional properties, ρ_G ; the ratio of the mean to the nominal material properties, ρ_M ; and the professional ratio, ρ_P (i.e., the ratio of the actual load carrying capacity of a column to that predicted by the design equation). The professional ratio indicates how well the design equation fits the test results. Then,

$$\rho_{R} = \rho_{G} \cdot \rho_{M} \cdot \rho_{P} \tag{5.20}$$

The above ratios are assumed to be independent random variables. Therefore, the coefficient of variation for the resistance, V_R , of a member is given by:

$$V_R^2 = V_G^2 + V_M^2 + V_P^2$$
 [5.21]

where V_G , V_M , and V_P are the coefficient of variation associated with ρ_G , ρ_M , and ρ_P respectively.

5.1.3 Determination of the Resistance of a Steel Column

For steel columns, CSA standards CAN3-S16.1-M84 and CAN/CSA-S16.1-94 use different equations based on different approximations, i.e., the SSRC column curves and the CSA curves as shown in Equation [5.1] through [5.2] and [5.3] respectively. In general, the factored resistance of a steel column can be expressed as:

$$C_r = \phi \cdot A \cdot F_v \cdot f(\lambda)$$
 [5.22]

where $f(\lambda)$ is a function of the non-dimensional slenderness ratio, λ , defined in Equation [4.1]. Therefore, the mean-to-nominal ratio of the resistance, ρ_{R} , for intermediate columns becomes

$$\rho_{R} = \rho_{A} \cdot \rho_{F_{V}} \cdot \rho_{f(\lambda)} \cdot \rho_{p}$$
 [5.23]

Because the slenderness parameter, λ , is a function of the yield strength, F_y , the two terms F_y and $f(\lambda)$ can be grouped as

$$F = F_{y} \cdot f(\lambda) \tag{5.24}$$

where F is a function of the terms F_y and $f(\lambda)$, and

$$\overline{F} = \overline{F}_{y} \cdot f(\overline{\lambda})$$
 [5.25]

Thus,

$$\rho_{F} = \rho_{F_{v}} \cdot \rho_{f(\lambda)} \tag{5.26}$$

As discussed above, the properties governing the strength of intermediate columns are the yield strength, F_y , the slenderness parameter, λ . As a result, Equation [5.23] becomes

$$\rho_{R} = \rho_{A} \cdot \rho_{F} \cdot \rho_{P} \tag{5.27}$$

and the coefficient of variation, V_R, for intermediate columns follows

$$V_{R} = (V_{A}^{2} + V_{F}^{2} + V_{P}^{2})^{1/2}$$
 [5.28]

However, the resistance of stub columns depends on the area of the column, A, and yield strength, F_y , as shown in the first part of equations [5.1] and [5.2]. Therefore, the mean-to-nominal ratio of the resistance, ρ_R for short columns becomes

$$\rho_{R} = \rho_{A} \cdot \rho_{F_{V}} \cdot \rho_{P} \tag{5.29}$$

and the coefficient of variation, V_R, for short columns follows

$$V_{R} = \left(V_{A}^{2} + V_{F_{y}}^{2} + V_{P}^{2}\right)^{1/2}$$
 [5.30]

Furthermore, the resistance of slender columns depends on the moment of inertia of the cross-section, I, and elastic modulus of the column, E, as seen by comparing Equation [4.1] with the fifth part of equations [5.1] and [5.2]. Therefore, the mean-to-nominal ratio of the resistance, ρ_R , for slender columns becomes

$$\rho_{R} = \rho_{1} \cdot \rho_{F} \cdot \rho_{P} \tag{5.31}$$

and the coefficient of variation, V_R, for slender columns follows

$$V_{R} = (V_{I}^{2} + V_{E}^{2} + V_{P}^{2})^{1/2}$$
 [5.32]

The values for the mean-to-nominal ratios and the coefficients of variation for the cross-sectional properties are investigated by using simple statistical analyses. The method of probability study for the professional factor is presented in section 5.2.3. The following derivation of ρ_F and V_F follow the work of Kennedy and Gad Aly (1980).

5.1.4 The Material Factor

The bias coefficient and the coefficients of variation for the material factor vary with the different design criteria for the resistance of a steel column. The column equations of clause 13.3.1 from CAN3-S16.1-M84 (identical to SSRC column curve 2) are taken in this sub-section to illustrate the procedure for determining the bias coefficients and the coefficients of variation for the material factor in the research. From a comparison of Equation [5.22] with Equation [5.2], the expressions for $f(\lambda)$ are obtained as follows:

(1)
$$f(\lambda) = 1.0$$
 , for $0 \le \lambda \le 0.15$

(2)
$$f(\lambda) = 1.035 - 0.202\lambda - 0.222\lambda^2$$
, for $0.15 \le \lambda \le 1.0$

(3)
$$f(\lambda) = -0.111 + 0.636\lambda^{-1} + 0.087\lambda^{-2}$$
, for $1.0 \le \lambda \le 2.0$ [5.33]

(4)
$$f(\lambda) = 0.009 + 0.877\lambda^{-2}$$
, for $2.0 < \lambda < 3.6$

(5)
$$f(\lambda) = \lambda^{-2}$$
, for $3.6 < \lambda$

Thus, the mean of $f(\lambda)$ is:

(1)
$$f(\overline{\lambda}) = 1.0$$
 , for $0 \le \overline{\lambda} \le 0.15$

(2)
$$f(\bar{\lambda}) = 1.035 - 0.202\bar{\lambda} - 0.222\bar{\lambda}^2$$
, for $0.15 < \bar{\lambda} < 1.0$

(3)
$$f(\overline{\lambda}) = -0.111 + 0.636\overline{\lambda}^{-1} + 0.087\overline{\lambda}^{-2}$$
, for $1.0 \le \overline{\lambda} \le 2.0$ [5.34]

(4)
$$f(\bar{\lambda}) = 0.009 + 0.877\bar{\lambda}^{-2}$$
, for $2.0 \le \bar{\lambda} \le 3.6$

(5)
$$f(\overline{\lambda}) = \overline{\lambda}^{-2}$$
 , for $3.6 \le \overline{\lambda}$

The mean-to-nominal ratio of λ is defined as

$$\rho_{\lambda} = \frac{\overline{\lambda}}{\lambda}$$
 [5.35]

Based on the definition of the slenderness parameter, λ , given by Equation [4.1], the mean-to-nominal ratio of the slenderness parameter can be obtained as

$$\rho_{\lambda} = \left(\frac{\rho_{F_{y}}}{\rho_{r}^{2} \cdot \rho_{E}}\right)^{1/2}$$
 [5.36]

Combining [5.25], [5.34], and [5.35] gives

(1)
$$\bar{F} = \bar{F}_y$$
 , for $0 \le \lambda \le 0.15$

(2)
$$\bar{F} = \bar{F}_y (1.035 - 0.202 \lambda \rho_{\lambda} - 0.222 \lambda^2 \rho_{\lambda}^2)$$
 , for $0.15 \le \lambda \le 1.0$

(3)
$$\overline{F} = \overline{F}_y(-0.111 + 0.636\lambda^{-1}\rho_{\lambda}^{-1} + 0.087\lambda^{-2}\rho_{\lambda}^{-2})$$
, for $1.0 \le \lambda \le 2.0$ [5.37]

(4)
$$\overline{F} = \overline{F}_y (0.009 + 0.877 \lambda^{-2} \rho_{\lambda}^{-2})$$
 , for $2.0 \le \lambda \le 3.6$

(5)
$$\vec{F} = \vec{F}_v(\lambda^{-2}\rho_\lambda^{-2})$$
, for $3.6 \le \lambda$

Thus, the mean-to-nominal ratio for the material factor with SSRC column curve 2 can be deduced as follows:

(1)
$$\rho_F = \rho_{F_v}$$
, for $0 \le \lambda \le 0.15$

(2)
$$\rho_{F} = \rho_{F_{y}} \frac{(1.035 - 0.202\lambda\rho_{\lambda} - 0.222\lambda^{2}\rho_{\lambda}^{2})}{(1.035 - 0.202\lambda - 0.222\lambda^{2})}, \text{ for } 0.15 \le \lambda \le 1.0$$

(3)
$$\rho_{F} = \rho_{F_{y}} \frac{(-0.111 + 0.636\lambda^{-1}\rho_{\lambda}^{-1} + 0.087\lambda^{-2}\rho_{\lambda}^{-2})}{(-0.111 + 0.636\lambda^{-1} + 0.087\lambda^{-2})} , \text{ for } 1.0 \le \lambda \le 2.0 \quad [5.38]$$

(4)
$$\rho_F = \rho_{F_y} \frac{(0.009 + 0.877\lambda^{-2}\rho_{\lambda}^{-2})}{(0.009 + 0.877\lambda^{-2})}$$
, for $2.0 \le \lambda \le 3.6$

$$(5) \quad \rho_{\rm F} = \rho_{\rm F} \rho_{\rm r}^2 \qquad , \text{ for } 3.6 \le \lambda$$

The mean-to-nominal ratios for the material factors derived from the other criteria can be obtained using the same procedure.

Applying the definition of the associated coefficient of variation, V_F, gives

$$V_{F} = \frac{\sigma_{F}}{\overline{F}}$$
 [5.39]

Having assumed that the variables affecting F, that is, F_y , r, and E, are independent, fundamental statistical equations for the standard deviation (Kennedy and Neville, 1976) are adopted to calculate the value of V_F as follows:

$$\sigma_{F} = \left[\left(\frac{\partial \overline{F}}{\partial \overline{F}_{y}} \right)^{2} \cdot \sigma_{F_{y}}^{2} + \left(\frac{\partial \overline{F}}{\partial \overline{r}} \right)^{2} \cdot \sigma_{r}^{2} + \left(\frac{\partial \overline{F}}{\partial \overline{E}} \right)^{2} \cdot \sigma_{E}^{2} \right]^{1/2}$$
 [5.40]

The components in Equation [5.40] can be obtained respectively. Using the second part of SSRC curve 2 as an example, the terms in [5.40] can be obtained as follows:

$$(1) \quad \left(\frac{\partial \overline{F}}{\partial \overline{F}_{y}}\right)^{2} \cdot \sigma_{F_{y}}^{2} = (1.035 - 0.303\lambda - 0.444\lambda^{2})^{2} \cdot \frac{\overline{F}_{y}^{2}}{\overline{F}_{y}^{2}} \cdot \sigma_{F_{y}}^{2}$$

$$= (1.035 - 0.303\lambda - 0.444\lambda^{2})^{2} \cdot \overline{F}_{y}^{2} \cdot V_{F_{y}}^{2}$$

$$= P_{1}^{2} \cdot \overline{F}_{y}^{2} \cdot V_{F_{y}}^{2}$$

$$(2) \quad \left(\frac{\partial \overline{F}}{\partial \overline{r}}\right)^{2} \cdot \sigma_{r}^{2} = (0.202\overline{\lambda} + 0.444\overline{\lambda}^{2})^{2} \cdot \frac{\overline{F}_{y}^{2}}{\overline{r}^{2}} \cdot \sigma_{r}^{2}$$

$$= (0.202\lambda + 0.444\lambda^{2})^{2} \cdot \overline{F}_{y}^{2} \cdot V_{r}^{2}$$

$$= P_{2}^{2} \cdot \overline{F}_{y}^{2} \cdot V_{r}^{2}$$

$$(3) \quad \left(\frac{\partial \overline{F}}{\partial \overline{E}}\right)^{2} \cdot \sigma_{E}^{2} = (0.101\lambda + 0.222\lambda^{2})^{2} \cdot \frac{\overline{F}_{y}^{2}}{\overline{E}^{2}} \cdot \sigma_{E}^{2}$$

$$= (0.101\lambda + 0.222\lambda^{2})^{2} \cdot \overline{F}_{y}^{2} \cdot V_{E}^{2}$$

where P_1 , P_2 , and P_3 present the portions of Eq. [5.41] that are functions of λ . Therefore, Eq. [5.40] can be expressed as

 $= P_2^2 \cdot \overline{F}_v^2 \cdot V_E^2$

$$\sigma_{F} = \overline{F}_{y} \cdot (P_{1}^{2} \cdot V_{F_{y}}^{2} + P_{2}^{2} \cdot V_{r}^{2} + P_{3}^{2} \cdot V_{E}^{2})^{1/2}$$
 [5.42]

and thus [5.39], in general form, becomes

$$V_{F} = \frac{\sigma_{F}}{\overline{F}} = \frac{\overline{F}_{y} \cdot (P_{1}^{2} \cdot V_{F_{y}}^{2} + P_{2}^{2} \cdot V_{r}^{2} + P_{3}^{2} \cdot V_{E}^{2})^{1/2}}{\overline{F}_{y} \cdot f(\overline{\lambda})}$$

$$= \frac{(P_{1}^{2} \cdot V_{F_{y}}^{2} + P_{2}^{2} \cdot V_{r}^{2} + P_{3}^{2} \cdot V_{E}^{2})^{1/2}}{f(\overline{\lambda})}$$
[5.43]

5.1.5 Summary

In general, the statistical quantities ρ_R and V_R for the SSRC curves are:

In accordance to SSRC column curve2:

[5.44]

I. For short columns: $0 \le \lambda \le 0.15$

$$\rho_R = \rho_A \cdot \rho_{F_y} \cdot \rho_P$$

$$V_R = (V_A^2 + V_{F_y}^2 + V_P^2)^{1/2}$$

II. For intermediate columns: $0.15 \le \lambda \le 3.6$

$$\rho_{\rm p} = \rho_{\rm A} \cdot \rho_{\rm E} \cdot \rho_{\rm p}$$

where
$$\rho_F = \rho_{F_y} \, \frac{(1.035 - 0.202 \, \lambda \, \rho_{\lambda} - 0.222 \, \lambda^2 \, \rho_{\lambda}^2)}{(1.035 - 0.202 \, \lambda - 0.222 \, \lambda^2)} \qquad \text{, for } 0.15 \leq \lambda \leq 1.0$$

$$\rho_F = \rho_{F_y} \, \frac{(-0.111 + 0.636 \, \lambda^{-1} \, \rho_{\lambda}^{-1} + 0.087 \, \lambda^{-2} \, \rho_{\lambda}^{-2})}{(-0.111 + 0.636 \, \lambda^{-1} + 0.087 \, \lambda^{-2})} \, , \, \text{for } 1.0 \leq \lambda \leq 2.0$$

$$\rho_{\rm F} = \rho_{\rm F_y} \frac{(0.009 + 0.877 \,\lambda^{-2} \,\rho_{\lambda}^{-2})}{(0.009 + 0.877 \,\lambda^{-2})} \qquad , \, {\rm for} \, \, 2.0 \le \lambda \le 3.6$$

$$V_R = (V_A^2 + V_F^2 + V_P^2)^{1/2}$$

where
$$V_F = \frac{(P_1^2 \cdot V_{F_y}^2 + P_2^2 \cdot V_r^2 + P_3^2 \cdot V_E^2)^{1/2}}{f(\overline{\lambda})}$$

where
$$f(\bar{\lambda}) = 1.035 - 0.202 \bar{\lambda} - 0.222 \bar{\lambda}^2$$

, for $0.15 \le \lambda \le 1.0$

and
$$P_1 = 1.035 - 0.303 \,\overline{\lambda} - 0.444 \,\overline{\lambda}^2$$

$$P_2 = 0.202 \,\overline{\lambda} + 0.444 \,\overline{\lambda}^2$$

$$P_3 = 0.101\,\overline{\lambda} + 0.222\,\overline{\lambda}^2$$

where
$$f(\bar{\lambda}) = -0.111 + 0.636 \bar{\lambda}^{-1} + 0.087 \bar{\lambda}^{-2}$$

 $for 1.0 \le \lambda \le 2.0$

and
$$P_1 = -0.111 + 0.318 \,\overline{\lambda}^{-1}$$

$$P_2 = 0.636 \overline{\lambda}^{-1} + 0.174 \overline{\lambda}^{-2}$$

$$P_3 = 0.318 \,\overline{\lambda}^{-1} + 0.087 \,\overline{\lambda}^{-2}$$

where
$$f(\bar{\lambda}) = 0.009 + 0.877 \,\bar{\lambda}^{-2}$$

, for $2.0 \le \lambda \le 3.6$

and
$$P_1 = 0.009$$

$$P_2 = 1.754 \,\overline{\lambda}^{-2}$$

$$P_3 = 0.877 \overline{\lambda}^2$$

III. For long columns: $\lambda \le 3.6$

$$\rho_R = \rho_I \cdot \rho_E \cdot \rho_P$$

$$V_R = (V_I^2 + V_E^2 + V_P^2)^{1/2}$$

In accordance to SSRC column curve 1:

[5.45]

I. For short columns: $0 \le \lambda \le 0.15$

$$\rho_R = \rho_A \cdot \rho_{F_y} \cdot \rho_P$$

$$V_R = (V_A^2 + V_{F_v}^2 + V_P^2)^{1/2}$$

II. For intermediate columns: $0.15 \le \lambda \le 3.6$

$$\rho_R = \rho_A \cdot \rho_F \cdot \rho_P$$

where
$$\rho_F = \rho_{F_y} \frac{(0.99 + 0.122\lambda\rho_{\lambda} - 0.367\lambda^2\rho_{\lambda}^2)}{(0.99 + 0.122\lambda - 0.367\lambda^2)}$$

, for
$$0.15 \le \lambda \le 1.2$$

$$\rho_{\rm F} = \rho_{\rm Fy} \frac{(0.051 + 0.801\lambda^{-2}\rho_{\lambda}^{-2})}{(0.051 + 0.801\lambda^{-2})}$$

, for
$$1.2 \le \lambda \le 1.8$$

$$\rho_{F} = \rho_{F_{y}} \frac{(0.008 + 0.942\lambda^{-2}\rho_{\lambda}^{-2})}{(0.008 + 0.942\lambda^{-2})}$$

, for
$$1.8 \le \lambda \le 2.8$$

$$V_R = (V_A^2 + V_F^2 + V_P^2)^{1/2}$$

where
$$V_F = \frac{(P_1^2 \cdot V_{F_y}^2 + P_2^2 \cdot V_r^2 + P_3^2 \cdot V_E^2)^{1/2}}{f(\overline{\lambda})}$$

where
$$f(\bar{\lambda}) = 0.990 + 0.122\bar{\lambda} - 0.367\bar{\lambda}^2$$

, for
$$0.15 \le \lambda \le 1.2$$

and
$$P_1 = 0.990 + 0.183\overline{\lambda} - 0.734 \overline{\lambda}^2$$

$$P_2 = -0.122\overline{\lambda} + 0.734 \overline{\lambda}^2$$

$$P_3 = -0.061\overline{\lambda} + 0.367\overline{\lambda}^2$$

where
$$f(\bar{\lambda}) = 0.051 + 0.801 \bar{\lambda}^{-2}$$

, for
$$1.2 \le \lambda \le 1.8$$

and
$$P_1 = 0.051$$

$$P_2 = 1.602 \overline{\lambda}^{-2}$$

$$P_3 = 0.801\overline{\lambda}^{-2}$$

where
$$f(\bar{\lambda}) = 0.008 + 0.942 \bar{\lambda}^{-2}$$

, for
$$1.8 \le \lambda \le 2.8$$

and
$$P_1 = 0.008$$

$$P_2 = 1.884\overline{\lambda}^{-2}$$

 $P_3 = 0.942\overline{\lambda}^{-2}$

III. For long columns: $\lambda \ge 3.6$

$$\rho_R = \rho_1 \cdot \rho_E \cdot \rho_P$$

$$V_R = (V_I^2 + V_E^2 + V_P^2)^{1/2}$$

For the column curve from Clause 13.3.1 of CAN/CSA-S16.1-94, the statistical quantities ρ_R and V_R are:

$$\begin{split} & \rho_R = \rho_A \cdot \rho_F \cdot \rho_P \\ & \text{where } \rho_F = \rho_{F_y} \, \frac{(1 + \lambda^{2n} \rho_\lambda^{2n})^{-1/n}}{(1 + \lambda^{2n})^{-1/n}} \\ & V_R = (V_A^2 + V_F^2 + V_P^2)^{1/2} \\ & \text{where } V_F = \frac{(P_1^2 \cdot V_{F_y}^2 + P_2^2 \cdot V_r^2 + V_E^2)^{1/2}}{f(\overline{\lambda})} \\ & \text{where } f(\overline{\lambda}) = (1 + \overline{\lambda}^{2n})^{-1/n} \\ & \text{and } P_1 = (1 + \overline{\lambda}^{2n})^{\frac{(n+1)}{n}} \\ & P_2 = 2\overline{\lambda}^{2n} \, (1 + \overline{\lambda}^{2n})^{\frac{(n+1)}{n}} \\ & P_3 = \overline{\lambda}^{2n} \, (1 + \overline{\lambda}^{2n})^{\frac{(n+1)}{n}} \end{split}$$

5.2 Statistical Parameters

The parametric study presented in Chapter 4 indicated that the orientation of the plates and buckling direction may affect the behaviour and strength of reinforced steel columns. Therefore, different performance factors may be applicable to columns with different reinforcing plate orientations and different buckling directions. It was therefore decided to conduct a statistical analysis on four separate groups, namely, two different reinforcing plate orientations and two different buckling directions. However, a statistical

analysis of these four groups of columns indicated that the above four groups could be merged into two groups since their performance factors were very similar. The first group includes columns reinforced with plates parallel to the flanges and buckling about the strong axis of the rolled section and columns reinforced with plates parallel to the web and buckling about the weak axis of the rolled section. The second group includes columns reinforced with plates parallel to the flanges and buckling about the weak axis of the rolled section, and columns reinforced with plates parallel to the web and buckling about the strong axis of the rolled section. The following sections present the statistical analysis results for these two groups.

The basic data related to cross-sectional and material properties required for the following analysis were obtained from Kennedy and Gad Aly (1980) and Chernenko and Kennedy (1988). The ratios of the mean value to the nominal value and the coefficients of variation for the geometrical and material properties are tabulated in Table 5.1. Since no statistical data were available for the sizes of the welds and fillets between the flanges and web of the rolled section, these values were taken as nominal values.

5.2.1 Geometrical Variations

The cross-sectional properties used to determine the capacity of a column are the area, A, the moment of inertia about principal centroidal axes, I_x and I_y , and the corresponding radii of gyration, r_x and r_y . Variations in these geometrical properties can be derived from statistical data concerning the flange and web thickness, t_f and w, the flange width, b_f , and the depth, d, of the rolled section, the reinforcing plate width and thickness, b_p and t_p the size of the fillet at the junction of the web and the flanges, k, and the size of the fillet weld joining the reinforcing plates to the flanges, g. These cross-sectional dimensions are summarized in Figure 5.3.

For any geometrical property, the ratio of the mean to the nominal value is defined as

$$\rho_{\rm G} = \frac{\overline{\rm G}}{\rm G} \tag{5.47}$$

The mean value for the area of a given reinforced column section is given as

$$\overline{A} = 2\overline{b}_f \overline{t}_f + (\overline{d} - 2\overline{t}_f) \overline{w} + 2\overline{b}_p \overline{t}_p + 0.858 \overline{k}^2 + 2\overline{g}^2$$
 [5.48]

Since the actual shape of the fillet between the web and the flanges of the rolled shape is not perfectly circular, the actual horizontal and vertical sides of the fillet are not the same. In order to simplify the calculations without sacrificing accuracy significantly, the fillet shape in the numerical model was assumed to be the complement of a quarter circle with identical side lengths, k. The size of the fillet, k, was taken as the average of the two sides given in the section properties tables (CISC Handbook, 1995). In addition, the weld face was assumed to be a straight line with equal leg size, g.

The mean values of the moments of inertia about the major and minor centroidal axes, \bar{I}_x and \bar{I}_y , for a column reinforced with plates parallel to the flanges are

$$\bar{I}_{x} = \frac{1}{6} \bar{b}_{f} \bar{t}_{f}^{3} + \frac{1}{2} \bar{b}_{f} \bar{t}_{f} (\bar{d} - \bar{t}_{f})^{2} + \frac{1}{12} \bar{w} (\bar{d} - 2\bar{t}_{f})^{3} + \frac{1}{6} \bar{b}_{p} \bar{t}_{p}^{3} + \frac{1}{2} \bar{b}_{p} \bar{t}_{p} (\bar{d} + \bar{t}_{p})^{2}
+ 0.03 \bar{k}^{4} + 0.858 \bar{k}^{2} (\frac{1}{2} \bar{d} - \bar{t}_{f} - 0.223 \bar{k})^{2} + \frac{1}{9} \bar{g}^{4} + 2 \bar{g}^{2} (\frac{1}{2} \bar{d} + \frac{1}{3} \bar{g})^{2}$$
[5.49]

and

$$\bar{I}_{y} = \frac{1}{6}\bar{t}_{f}\bar{b}_{f}^{3} + \frac{1}{12}(\bar{d} - 2\bar{t}_{f})\bar{w}^{3} + \frac{1}{6}\bar{t}_{p}\bar{b}_{p}^{3} + \frac{1}{9}\bar{g}^{4} + 2\bar{g}^{2}(\frac{1}{2}\bar{b}_{p} + \frac{1}{3}\bar{g})^{2} + 0.03\bar{k}^{4} + 0.858\bar{k}^{2}(\frac{1}{2}\bar{w} + 0.223\bar{k})^{2}$$
[5.50]

The mean values for the moment of inertia about the major and minor centroidal axes, \bar{I}_x and \bar{I}_y , for a column reinforced with plates parallel to the web are

$$\bar{I}_{x} = \frac{1}{6}\bar{b}_{f}\bar{t}_{f}^{3} + \frac{1}{2}\bar{b}_{f}\bar{t}_{f}(\bar{d} - \bar{t}_{f})^{2} + \frac{1}{12}\bar{w}(\bar{d} - 2\bar{t}_{f})^{3} + \frac{1}{6}\bar{b}_{p}^{3}\bar{t}_{p} + 0.03\bar{k}^{4}
+ 0.858\bar{k}^{2}(\frac{1}{2}\bar{d} - \bar{t}_{f} - 0.223\bar{k})^{2} + \frac{1}{9}\bar{g}^{4} + 2\bar{g}^{2}(\frac{1}{2}\bar{d} + \frac{1}{3}\bar{g})^{2}$$
[5.51]

and

$$\bar{I}_{y} = \frac{1}{6} \bar{t}_{f} \bar{b}_{f}^{3} + \frac{1}{12} (\bar{d} - 2\bar{t}_{f}) \bar{w}^{3} + \frac{1}{6} \bar{b}_{p} \bar{t}_{p}^{3} + \frac{1}{2} \bar{b}_{p} \bar{t}_{p} (\bar{b}_{f} + \bar{t}_{p})^{2} + 0.03 \bar{k}^{4}
+ 0.858 \bar{k}^{2} (\frac{1}{2} \bar{w} + 0.223 \bar{k})^{2} + \frac{1}{9} \bar{g}^{4} + 2 \bar{g}^{2} (\frac{1}{2} \bar{b}_{f} - \frac{1}{3} \bar{g})^{2}$$
[5.52]

The associated radii of gyration for reinforced columns are

$$\bar{r}_x = \sqrt{\frac{\bar{I}_x}{\bar{A}}}$$
 [5.53]

and

$$\bar{r}_y = \sqrt{\frac{\bar{I}_y}{\bar{A}}}$$
 [5.54]

The coefficient of variation for the cross-sectional properties can be obtained from the assumption that the dimensions are independent variables. It is given as

$$V_{G} = \frac{1}{\overline{G}} \left[\left(\frac{\partial \overline{G}}{\partial \overline{b}_{f}} \right)^{2} \sigma_{b_{f}}^{2} + \left(\frac{\partial \overline{G}}{\partial \overline{t}_{f}} \right)^{2} \sigma_{t_{f}}^{2} + \left(\frac{\partial \overline{G}}{\partial \overline{w}} \right)^{2} \sigma_{w}^{2} + \left(\frac{\partial \overline{G}}{\partial \overline{b}_{p}} \right)^{2} \sigma_{b_{p}}^{2} + \left(\frac{\partial \overline{G}}{\partial \overline{t}_{p}} \right)^{2} \sigma_{t_{p}}^{2} + \left(\frac{\partial \overline{G}}{\partial \overline{d}} \right)^{2} \sigma_{d}^{2} + \left(\frac{\partial \overline{G}}{\partial \overline{b}} \right)^{2} \sigma_{w}^{2} + \left(\frac{\partial \overline{G}}{\partial \overline{b}} \right)^{2} \sigma_{g}^{2} \right]^{\frac{1}{2}}$$

$$(5.55)$$

where the partial derivatives are evaluated at the mean. The variation in weld and fillet sizes was not considered in this work because actual measurements were not available. This assumption is justified given their small area compared to the total area of the reinforced cross-section.

Table 5.2 presents a summary of the mean, the measured-to-nominal ratio, and the associated coefficient of variation for the relevant geometric properties of reinforced columns consisting of the most widely used wide flange shapes reinforced with various plate thicknesses. The nominal values for the cross-sectional dimensions were obtained from the handbook of steel construction (CISC Handbook, 1995). With known statistical quantities, ρ_G and V_G , the mean values of the geometric properties t_f , w, v, v, and v are shown in Table 5.2. The statistical quantities, v and v and v are shown in Table 5.2. The statistical quantities, v and v are shown in Table 5.2. The statistical quantities, v and v are shown in the table, using equations [5.47] to [5.55].

In order to consider all the sections as a whole, the mean value of the mean-tonominal ratios is calculated as follows

$$\bar{\rho}_{G} = \frac{1}{n} \sum_{i=1}^{i=n} \rho_{G_{i}} = \frac{1}{n} \left(\rho_{G_{1}} + \rho_{G_{2}} + \dots + \rho_{G_{n}} \right)$$
 [5.56]

and the associated coefficient of variation can be obtained using the moment algebra for the distribution of different parameter variables as given by Benjamin and Cornell (1970) as follows:

$$V_{G} = \frac{1}{\overline{\rho}_{G}} \cdot \frac{1}{(n-1)^{1/2}} \left[\sum_{i=1}^{n} (V_{G_{i}} \cdot \rho_{G_{i}})^{2} + \sum_{i=1}^{n} (\rho_{G} - \rho_{G_{i}})^{2} \right]^{1/2}$$
 [5.57]

The mean-to-nominal ratios and the coefficient of variation for the geometrical properties are summarized in Table 5.3. The statistical quantities, ρ_G and V_G , for the geometric variations A, I_x , I_y , r_x and r_y , are obtained using equations [5.56] and [5.57] based on the statistical data shown in Table 5.2. The statistical quantities, ρ_G and V_G , for the radius of gyration, r, and the moment of inertia, I, were obtained by pooling the data for the x-axis and y-axis properties together in equations [5.56] and [5.57]. These quantities are presented in Table 5.3.

5.2.2 Material Variations

The important material properties for the calculation of column capacity are the yield strength, F_y, and the modulus of elasticity, E. The statistical parameters for the material properties for the rolled section were obtained from Kennedy and Gad Aly (1980). The statistical parameters for the reinforcing plates were obtained from Chernenko and Kennedy (1988). These statistical parameters are summarized in Tables 5.1 (a) and 5.1 (b), respectively.

Possible rolled sections for columns reinforced with corresponding reinforcing plates were chosen from the CISC handbook (1995) to take the ratios of the components' area to the total area into account. The descriptions of the cross-sections are given in Table A.1. With known geometric properties for a section, the mean value of the rolled section area, A_C , and the plate area, A_P , can be obtained for each sample. Therefore, the ratio of the rolled section area to the total area, $\frac{A_C}{A_C + A_P}$, can be obtained for each sample. An analysis of the numerical models presented in Table A.1 gives a mean ratio of the rolled

section area to the total area, $\frac{A_C}{A_C + A_P}$, of 0.645. Because the sum of the ratio of the reinforcing plate area (A_P) to total area and the ratio of the W shape area (A_C) to total area should be 1.0, the mean ratio of the plates area to the total area, $\frac{A_P}{A_C + A_P}$, for the numerical models is 0.355.

With a known mean ratio of the component area to the total area, the weighted yield strength of the cross-section can be obtained as:

$$F_{y} = CF_{y} \cdot \frac{A_{C}}{A_{C} + A_{P}} + PF_{y} \cdot \frac{A_{p}}{A_{C} + A_{p}}$$
 [5.58]

where CF_y is the mean yield strength of the rolled section and PF_y is the mean yield strength of the reinforcing plates. Furthermore, the mean value of the weighted mean-to-nominal ratios for the yield strength was approximated as follows:

$$\rho_{F_y} = \rho_{CF_y} \cdot \frac{A_C}{A_C + A_P} + \rho_{PF_y} \cdot \frac{A_P}{A_C + A_P}$$
 [5.59]

The yield strengths of the rolled columns and reinforcing plates are considered as independent random variables. The coefficient of variation for the yield strength thus becomes (Kennedy and Neville 1976),

$$V_{F_{v}} = \left[V_{CF_{v}}^{2} + V_{PF_{v}}^{2}\right]^{1/2}$$
 [5.60]

Following the procedure presented in Section 5.1.4, the statistical parameters for the material properties can be obtained. Table 5.4 presents the statistical parameters for the material properties with different SSRC and CSA curves. The mean-to-nominal ratio of the radius of gyration, ρ_r , and its associated coefficient of variation, V_r , are obtained from Table 5.1 for the samples. The mean-to-nominal ratio of the modulus of elasticity, ρ_E , and its associated coefficient of variation, V_E , were obtained from the investigation presented by Chernenko and Kennedy (1988) as 1.013 and 0.015, respectively.

5.2.3 Professional Factors

As a factor indicating how well the design equation fits the experimental results, the professional factor accounts for variations in column capacity other than those considered as cross-sectional and material properties. The professional factor shows the relationship between the measured strength (or strength predicted using the finite element model) of a reinforced column and that predicted by the design equation. The column design equations evaluated in this work are those in CAN3-S16.1-M84 (equations [5.1] and [5.2]) and CAN/CSA-S16.1-94 (equation [5.3]). These equations were derived based on test results and analysis results that included the effect of initial out-of-straightness and residual stresses. The equations show that the slenderness parameter, λ , is the prime factor determining the load carrying capacity of a column. Furthermore, effects of cross-sectional property, orientation of reinforcing plates, axes of buckling, preload magnitude, and non-linear interaction of the above parameters on the strength of the reinforced column have to be considered.

Chernenko and Kennedy (1988) proposed that the effect of the statistical variation of out-of-straightness and residual stresses on column strength could be assessed independently. The parametric study presented in Chapter 4 indicated that the effect of initial residual stresses on reinforced column strength is negligibly small. Since the residual stresses in the reinforced section do not vary much, only the effect of the statistical variations in out-of-straightness will be assessed independently in the following. The effects of variation in residual stresses, steel grade, buckling axis, preload magnitude, and the geometrical properties of the rolled section and reinforcing plates will be considered as a whole. The professional ratio can be therefore expressed as

$$\rho_{\rm P} = \rho_{\rm s} \cdot \rho_{\rm n} \cdot \rho_{\rm ex} \tag{5.61}$$

where ρ_s is the simulated professional ratio, that is, the ratio of the strength determined by the computer simulation to that predicted by the design equations for the mean value of out-of-straightness for a given value of λ . ρ_n is the normalized simulated professional ratio, which accounts for the other parameters. The third term, ρ_{ex} , is the mean value of

the ratio of the experimental strength to the strength predicted by computer simulations. Consequently, the professional ratio, ρ_p , is the experimental to the predicted ratio.

5.2.3.1 The Effect of Out-of-straightness

As discussed in Chapter 3, the effect of initial out-of-straightness of the rolled section before reinforcement and the effect of the initial out-of-straightness of the reinforced column can contribute to the scatter in the test results. The following therefore looks at the effect of initial out-of-straightness in the unreinforced wide flange section and the initial out-of-straightness in the reinforced column.

Figures 5.4, 5.5 and 5.6 present analysis data for the columns from Group 1 (columns reinforced with plates parallel to the flanges and buckling about the strong axis of the rolled section and columns reinforced with plates parallel to the web and buckling about the weak axis of the rolled section). Plots of the simulated professional ratio, ρ_s , versus out-of-straightness for non-dimensional slenderness ratio are presented for values of the slenderness ratio, λ , of 0.4, 1.1, and 1.5. The magnitude of the initial out-of-straightness was varied from 0 to L/1000 (the maximum initial imperfection permissible by CSA standard G40.20). The simulated professional ratio, ρ_s , is taken as the ratio of the strength obtained from the finite element analysis to the strength predicted using SSRC curve 2. All the data used to plot figures 5.4, 5.5 and 5.6 are presented in Column (7) of Tables 5.5, 5.6 and 5.7, respectively. Columns (6), (8) and (9) from these tables present the simulated professional ratios based on SSRC curve 1, CSA curve 1 and CSA curve 2, respectively. For the second group of columns (columns reinforced with plates parallel to the web and buckling about the weak axis of the rolled section and columns reinforced with plates parallel to the web and buckling about the strong axis of the rolled section) the same procedure was used to obtain the professional ratios. This data is presented in Appendix C.

Figures 5.4, 5.5, and 5.6 show an average line obtained using the method of least squares (Kennedy and Neville, 1976). The equation of the regression line for each value of the slenderness parameter is presented in Table 5.8. Figures 5.4 to 5.6 show that the

ratio of the professional factor decreases with increasing initial imperfection for a given value of λ . The slopes of the average lines (i.e., the rate of decrease of strength with out-of-straightness) for $\lambda = 1.1$ and 1.5 are much greater than those for $\lambda = 0.4$. As expected, initial imperfections have a greater effect on reducing the column strength for $\lambda = 1.1$ and 1.5 than for $\lambda = 0.4$. In fact, the slope of the regression line is close to zero for $\lambda = 0.4$.

To investigate the effect of the variation in initial imperfection on reinforced column strength, the mean value and coefficient of variation are used as a basis of comparison. In this research, a mean initial imperfection of L/1500, a standard deviation of L/15000, and a coefficient of variation of 0.1 were used based on the work of Bjorhovde (1988) on rolled, unreinforced columns. The limited work on reinforced columns presented by Nagaraja Rao and Tall (1963) indicates that initial out-of-straightness in reinforced columns can be smaller than for unreinforced columns.

A vertical line, representing a mean value of out-of-straightness of $\delta_0/L = 1/1500 = 0.000667$ is shown in Figures 5.4, 5.5, and 5.6. Two additional vertical lines are drawn at one standard deviation from the mean value in each figure. The intersection point of the mean line with the regression line gives the simulated professional ratio for the mean out-of-straightness, ρ_s . The vertical distance between the intercepts of the right and left standard deviation lines gives the values of two standard deviations of ρ_s associated with out-of-straightness. Therefore, the standard deviation for the simulated professional ratio is calculated by multiplying the slope of the equation by one standard deviation for out-of-straightness, that is, slope x 0.0000667. From this, the coefficient of variation is calculated directly. The mean values of the professional ratios and the corresponding coefficients of variation for the mean out-of-straightness are also given in Table 5.8.

5.2.3.2 Miscellaneous Factors

Figures 5.4, 5.5 and 5.6 show significant scatter of the simulated professional ratios about the regression lines. This scatter is attributed to influencing factors such as initial residual stresses, welding residual stresses, buckling axis, preload magnitude, cross-sectional geometry, and the fundamental non-linearity of the relation between the

strength of the columns and the governing parameters. The effect of these miscellaneous factors can be assessed by normalizing the plotted professional ratio for any slenderness parameter by dividing by the value obtained from the linear expression given in Table 5.8 for that specific slenderness parameter. The normalized professional ratios, ρ_n , are presented in Tables 5.9, 5.10, and 5.11 for the λ values of 0.4, 1.1, and 1.5, respectively.

The average value of the normalized professional ratio, ρ_n , should equal 1.00 for any slenderness parameter for which a best-fit straight line has been used. The normalized professional ratio for $\lambda = 0.4$, 1.1, and 1.5 are plotted in Figures 5.7, 5.8, and 5.9 respectively. The mean values in each case are nearly equal to 1.00. The scatter of the distribution is large and, as a result, the coefficient of variation is large. The mean values and coefficients of variations for all different slenderness parameters are given in Table 5.12 as ρ_n and V_n , respectively.

5.2.3.3 Experimental Factor

The experimental ratio, ρ_{ex} , is defined as

$$\rho_{\rm ex} = \frac{P_{\rm ex}}{P_{\rm fea}} \tag{5.62}$$

where, P_{ex} is the strength obtained from the experiment and P_{fea} is the strength obtained from finite element analysis. Because of the very small number of test results, the statistical value of the experimental ratio cannot be evaluated with any degree of confidence. In order to provide more support for the statistical analysis for the experimental factor for reinforced columns, the experimental ratios for unreinforced columns were used in the research. The unreinforced columns used for the partial validation of the finite element models were used for this purpose. A description of these columns is presented in Table 3.1. The geometric properties and the initial imperfections of the rolled section, as well as the initial residual stresses are the same for both unreinforced and reinforced columns. The loading procedures for both reinforced and unreinforced columns are similar, except for the introduction of the reinforcing plates and the welds in the reinforced columns. Therefore, the experimental ratios for unreinforced columns are expected to be similar to those for reinforced columns.

Table 5.13 presents the results of the analysis of the unreinforced columns described in Table 3.2. Columns (4) and (5) present the predicted and measured capacities, respectively. The results are normalized in terms of the yield strength. Column (6) presents the experimental ratio calculated using Equation [5.62]. The first two test specimens were obtained from the work of Huber and Beedle (1954). The experimental ratio for these two specimens is close to 1.0 The lack of information about initial imperfections for the other three test specimens makes it difficult to obtain an accurate prediction of the test results. In order to verify the effect of initial imperfections on the finite element analysis results, two values of initial imperfection were assumed for each test specimen, as shown in Column (3) of Table 5.13. The results show that the initial imperfection has a significant effect on strength. Within the range of initial imperfections presented in Table 5.13, the experimental results can be accurately simulated.

In general, it can be expected that ρ_{ex} would be closer to 1.0 with a smaller coefficient of variation as long as the geometric and material properties of the reinforced columns are accurately determined. Therefore, the mean value of ρ_{ex} was taken as 1.0 and the coefficient of variance was taken as 0.0 to reflect the high accuracy of the finite element analysis when the physical parameters of the column are accurately defined. Further testing of reinforced steel column should be conducted to verify these values.

5.2.3.4 **Summary**

The professional factors are given in Table 4.12 for the two groups of columns defined in section 5.2, for the three values of slenderness parameter used in this study, and for the reference strength calculated using four different column curves from two standards. These four column curves are SSRC column curve 1, SSRC column curve 2, and the equivalent curves adopted by CAN/CSA-S16.1-94. The strength predicted using SSRC column curve 2 is lower than that predicted using SSRC column curve 1, which leads to higher values of ρ_s for SSRC column curve 2 than for SSRC column curve 1.

Since the column curves used in CAN/CSA-S16.1-94 are close approximations of the original SSRC column curves (Loov, 1996), the column strength predicted using

SSRC curves are in very good agreement with that predicted using the CSA curves. The corresponding values of ρ_s are also similar.

5.3 Evaluation of the Performance Factors

Resistance factors for reinforced columns were calculated using Equation [5.18]. A coefficient of separation, α , of 0.55, and a reliability index, β , of 3.0, consistent with the limit state design for building in Canada, were used. Table 5.14 presents the assembled data and calculated resistance factors for three values of the slenderness parameter, the two groups of columns, and four columns curves. Examination of the resistance factors for reinforced columns within one group indicates that the CSA curves can approximate the SSRC curves very well. The maximum difference in the values of the performance factors between SSRC column curve 2 and the corresponding CSA column curve is about 2.8% for $\lambda = 1.1$.

As expected, SSRC column curve 1 yields lower resistance factors than SSRC column curve 2. Within the range studied, the strength of reinforced steel columns in Group 1 (columns reinforced with plates parallel to the flanges and buckling about the strong axis of the rolled section, and columns with reinforcing plates parallel to the web and buckling about the weak axis of the rolled section) can be predicted conservatively with SSRC curve 1 or CSA column curve with n=2.24.

On the other hand, resistance factors calculated for the columns from Group 2 for SSRC curve 1 vary from 0.82 to 0.99. It is therefore unconservative to use SSRC column curve 1 to predict the capacity of the columns from Group 2. The resistance factors obtained for SSRC curve 2 vary from 1.02 to 1.06. They are from 1.13 to 1.18 times the current value of ϕ . SSRC column curve 2 and the CSA column curve with n=1.34 can therefore be used conservatively to predict the capacity of reinforced steel columns from Group 2 (columns reinforced with plates parallel to the web and buckling about the strong axis of the rolled section, and columns with reinforcing plates parallel to the flanges and buckling about the weak axis of the rolled section).

Table 5.1.a Statistical Parameters for Rolled W Sections (from Kennedy and Gad Aly, 1979)

December	Desirentian	Statist	tical Parameters
Property	Designation	Mean/Nominal, ρ	Coefficient of Variation, V
b _f	The width of the flanges	1.005	0.014
t_f	The thickness of the flanges	0.976	0.042
w	The thickness of the web	1.017	0.038
d	The depth of the section	1.000	0.00195*
k	The side of the fillet	1.000	0.000
\mathbf{F}_{y}	The yield strength	1.070	0.065
E	The modulus of elasticity	1.000	0.019

Note: The average side of the fillet was assummed without measurement available.

Table 5.1.b Statistical Parameters for Cover Plates (from Chernenko and Kennedy, 1988)

December	Designation	Statist	tical Parameters
Property	Designation	Mean/Nominal, ρ	Coefficient of Variation, V
b _p	The width of the plates	0.999	0.003
t _p	The thickness of the plates	1.010	0.008
g	The side of the weld	1.000	0.000
$\mathbf{F}_{\mathbf{y}}$	The yield strength	1.133	0.059
E	The modulus of elasticity	1.038	0.026

Note: The average side of the weld was assummed without measurement available.

^{*} Mean value of the distribution, which depends on range in depths of rolled sections.

Table 5.2.a Statistical Parameters for the Geometric Properties of the Reinforced Columns

		<u> </u>	Flora		10h					11					
:			٩							MCII		i			
Š.	W-Section	ρ	ت	ד	>	~	Plate	مْ	- -	50	∢	4	Α	× >	<u> </u>
		(mm)	(mm)	(mm)	(mm)	(mm)		(mm)	(mm)	(mm)	(mm ²)	(mm ²)			(10°mm ⁴)
-	W310x179	313	28.1	333	18.0	16.3	290x25	290	25	22	38268	38174	0.998	0.0003	940.5
7	W310x179	313	28.1	333	18.0	16.3	290x20	290	20	<u>8</u>	35048	34927	0.997	0.0004	827.0
က	W310x179	313	28.1	333	18.0	16.3	290x16	290	91	4	32472	32330	0.996	0.0004	740.2
4	W310x158	310	25.1	327	15.5	17.0	290x25	290	25	22	35568	35496	0.998	0.0004	865.5
S	W310x158	310	25.1	327	15.5	17.0	290x20	290	20	<u>8</u>	32348	32249	0.997	0.0004	755.5
9	W310x158	310	25.1	327	15.5	17.0	290x16	290	91	14	29772	29651	966.0	0.0004	671.5
7	W310x158	310	25.1	327	15.5	17.0	290x12	290	12	9	27260	27118	0.995	0.0005	592.9
œ	W310x143	309	22.9	323	14.0	13.9	290x20	290	20	81	30448	30366	0.997	0.0004	6'90'
6	W310x143	309	22.9	323	14.0	13.9	290x16	290	91	4	27872	27768	0.996	0.0005	624.8
9	W310x143	309	22.9	323	14.0	13.9	290x12	290	12	0	25360	25235	0.995	0.0005	548.0
=	W310x143	309	22.9	323	14.0	13.9	290x10	290	01	œ	24128	23992	0.994	0.0005	511.5
2	W310x129	308	20.6	318	13.1	14.7	290x20	290	20	<u>&</u>	28748	28687	0.998	0.0004	6969
13	W310x129	308	20.6	318	13.1	14.7	290x16	290	91	4	26172	26089	0.997	0.0005	577.1
7	W310x129	308	20.6	318	13.1	14.7	290x12	290	12	9	23660	23556	966.0	0.0005	502.4
15	W310x129	308	20.6	318	13.1	14.7	290x10	290	01	œ	22428	22313	0.995	9000'0	466.9
9	W310x118	307	18.7	314	6.11	16.2	290x16	290	91	14	24672	24605	0.997	0.0005	538.8
17	W310x118	307	18.7	314	6.11	16.2	290x12	290	12	<u> </u>	22160	22071	966.0	9000.0	465.9
<u>∞</u>	W310x118	307	18.7	314	6.11	16.2	290x10	290	01	œ	20928	20828	0.995	9000.0	431.3
61	W310x118	307	18.7	314	6.11	16.2	290x8	290	∞	9	19712	10961	0.994	0.0007	397.9
70	W310x107	306	17.0	311	6.01	14.3	290x16	290	91	7	23272	23219	0.998	9000'0	505.9
21	W310x107	306	17.0	311	6.01	14.3	290x12	290	12	0	20760	20685	966'0	9000.0	434.3
22	W310x107	306	17.0	311	6.01	14.3	290x10	290	01	œ	19528	19442	966.0	0.0007	400.3
23	W310x107	306	17.0	311	6.01	14.3	290x8	290	œ	9	18312	18216	0.995	0.0007	367.5
5 4	W310x86	254	16.3	310	9.1	15.1	230x16	230	91	4	18752	18711	0.998	9000.0	404.2
25	W310x86	254	16.3	310	9.1	15.1	230x12	230	12	9	16720	16662	0.997	0.0007	346.7

Table 5.2.a (Cont'd)

		Fla	Flange		do/	Fillet		Plate	ıte	Weld					
Š.	W-Section	کٍ	<u>.</u> .	p	>	. ¥	Plate	ث	ے۔	s	4	4	ρ	>	Ľ
		(mm)	(mm)	(mm)	(mm)	(mm)		(mm)	(mm)	(mm)	(mm ²)	(mm ²)			(10°mm ⁴)
5 0	W310x86	254	16.3	310	9.1	15.1	230×10	230	9	œ	15728	15662	966.0	0.0008	319.5
27	W310x86	254	16.3	310	9.1	15.1	230x8	230	∞	9	14752	14677	0.995	0.0008	293.3
58	W310x79	254	14.6	306	8 .8	17.0	230x12	230	13	9	15820	15776	0.997	0.0008	322.1
29	W310x79	254	14.6	306	8 .8	17.0	230x10	230	9	œ	14828	14776	966.0	0.0008	295.6
30	W310x79	254	14.6	306	æ. æ.	17.0	230x8	230	œ	9	13852	13791	966.0	0.0009	270.1
31	W310x67	204	14.6	306	8.5	15.3	180×12	180	12	01	13030	13002	0.998	0.0009	259.4
32	W310x67	204	14.6	306	8.5	15.3	180×10	180	9	œ	12238	12203	0.997	0.0009	238.2
33	W310x67	204	14.6	306	8.5	15.3	180x8	180	∞	9	11462	11420	0.996	0.0010	218.0
34	W310x60	203	13.1	303	7.5	15.1	180x12	180	12	01	12110	12088	0.998	0.0010	240.7
35	W310x60	203	13.1	303	7.5	15.1	180x10	180	01	œ	11318	11289	0.997	0.0010	219.9
36	W310x60	203	13.1	303	7.5	15.1	180x8	180	œ	9	10542	10507	0.997	0.0011	200.0
37	W250x73	254	14.2	253	9.8	12.5	230x12	230	12	9	15000	14952	0.997	0.0007	213.0
38	W250x73	254	14.2	253	9.8	12.5	230x10	230	0	∞	14008	13951	966.0	0.0007	194.4
39	W250x73	254	14.2	253	9.8	12.5	230x8	230	œ	9	13032	12966	0.995	0.0008	176.6
40	W250x73	254	14.2	253	9.8	12.5	230x6	230	9	9	12112	12038	0.994	0.0009	160.2
4	W250x58	203	13.5	252	8.0	12.7	180x 12	180	12	01	11940	11611	0.998	0.0008	166.0
42	W250x58	203	13.5	252	8.0	12.7	180×10	180	01	∞	11148	11112	0.997	0.0000	151.2
43	W250x58	203	13.5	252	8.0	12.7	180x8	180	∞	9	10372	10329	966.0	0.000	137.2
4	W250x58	203	13.5	252	8.0	12.7	180x6	180	9	9	9652	9603	0.995	0.0010	124.4
45	W250x49	202	0.11	247	7.4	12.8	180×10	180	<u>e</u>	∞	8/66	8566	0.998	0.0010	132.2
46	W250x49	202	0.11	247	7.4	12.8	180x8	180	œ	9	9202	9116	0.997	0.0011	9.811
47	W250x49	202	11.0	247	7.4	12.8	180x6	180	9	9	8482	8449	0.996	0.0012	106.3
48	W200x59	205	14.2	210	9.1	8.0	180x12	180	12	01	12050	12013	0.997	0.0007	116.5
49	W200x59	205	14.2	210	9.1	8.0	180×10	180	01	∞	11258	11214	966'0	0.0007	0.901
80	W200x59	205	14.2	210	9.1	8.0	180x8	180	&	9	10482	10431	0.995	0.0008	95.9

Table 5.2.a (Cont'd)

No. W-Section b ₁ t ₁ d w k Plate b ₁ t ₁ g A A D ₁ D		i	Fla	Flange	≯	qə,	Fillet		Pla	ıle	Weld					
(mm) (mm) <th< th=""><th></th><th>W-Section</th><th>ρ^ζ</th><th>7</th><th>Р</th><th>*</th><th>*</th><th>Plate</th><th>þ</th><th>f.</th><th>8</th><th>⋖</th><th>4</th><th>ΡΑ</th><th>></th><th>-*</th></th<>		W-Section	ρ ^ζ	7	Р	*	*	Plate	þ	f.	8	⋖	4	ΡΑ	>	- *
W200x59 205 14.2 210 9.1 8.0 180x6 180 6 9762 9705 0.994 0.0009 W200x52 204 12.6 206 7.9 7.7 180x10 180 1 111 0.997 0.0009 W200x52 204 12.6 206 7.9 7.7 180x10 180 8 6 9572 9529 0.997 0.0009 W200x52 204 12.6 206 7.9 7.7 180x10 180 8 6 9572 9597 0.0009 W200x46 203 11.0 203 7.2 7.7 180x10 180 8 6 8722 880 0.996 0.0009 W200x46 203 11.0 203 7.2 7.7 180x10 180 8 6 8728 880 0.996 0.0009 W200x47 166 11.8 180 8 6 8722 8929	- 1		(mm)	(mm)	(mm)	(mm)	(mm)		(mm)	(mm)	(mm)	(mm ²)	(mm ²)			(10°mm ⁴)
W200x52 204 12.6 206 7.9 7.7 180x10 180 10 11140 11111 0.997 0.0008 W200x52 204 12.6 206 7.9 7.7 180x10 180 10 8 10348 10312 0.997 0.0008 W200x52 204 12.6 206 7.9 7.7 180x6 180 8 6 8572 9590 0.0009 W200x46 203 11.0 203 7.2 7.7 180x6 180 8 6 8772 9590 0.0009 W200x46 203 11.0 203 7.2 7.7 180x6 180 8 6 8772 9590 0.0009 W200x42 166 11.8 205 7.2 8.1 140x10 140 10 8 8 8 9590 0.0010 W200x42 166 11.8 20 7.2 140x10 140 8 <		W200x59	205	14.2	210	9.1	8.0	9x081	180	9	9	9762	9705	0.994	0.0009	86.9
W200x52 204 12.6 206 7.9 7.7 180x10 180 8 6 9572 9529 0.996 0.0009 W200x52 204 12.6 206 7.9 7.7 180x8 180 6 9572 9529 0.996 0.0009 W200x52 204 12.6 206 7.9 7.7 180x8 180 6 8852 8802 0.994 0.0009 W200x46 203 11.0 203 7.2 7.7 180x8 180 6 8826 9929 0.0009 W200x42 166 11.8 203 7.2 7.7 180x8 180 6 6 8095 0.099 0.0010 W200x42 166 11.8 205 7.2 7.7 180x8 140 6 6 8095 0.099 0.0010 W200x42 166 11.8 10.2 7.2 140x10 140 6 6 8095		W200x52	204	12.6	206	6.7	7.7	180x12	180	12	9	11140	==	0.997	0.0007	106.1
204 12.6 206 7.9 7.7 180x8 180 8 6 9572 9529 0.996 0.0009 204 12.6 206 7.9 7.7 180x6 180 6 8852 8802 0.994 0.0009 203 11.0 203 7.2 7.7 180x10 180 6 8528 8802 0.995 0.0009 203 11.0 203 7.2 7.7 180x10 180 6 8052 8012 0.995 0.0010 203 11.0 203 7.2 7.7 180x6 180 6 8052 8092 0.995 0.0010 166 11.8 203 7.2 1.7 180x6 140 6 6 8052 8099 0.0010 166 11.8 205 7.2 8.1 140x6 140 6 6 8052 8099 0.0010 165 11.2 8.1		W200x52	204	12.6	206	7.9	7.7	180×10	180	2	œ	10348	10312	0.997	0.0008	95.9
W200x52 204 12.6 206 7.9 7.7 180x6 180 6 6 8852 8802 0.994 0.0009 W200x46 203 11.0 203 7.2 7.7 180x10 180 6 8548 9522 0.997 0.0009 W200x46 203 11.0 203 7.2 7.7 180x8 180 6 8 9548 9522 0.997 0.0009 W200x46 203 11.0 203 7.2 7.7 180x8 180 6 8 9548 9522 0.997 0.0010 W200x42 166 11.8 205 7.2 8.1 140x10 140 6 6 7592 754 0.995 0.0010 W200x42 166 11.8 205 7.2 8.1 140x10 140 6 6 7592 754 0.995 0.0010 W200x42 165 10.2 201 6.2 8.0 140x10	_	W200x52	204	12.6	206	7.9	7.7	180x8	180	œ	9	9572	9529	966'0	0.0009	86.2
W200x46 203 11.0 203 7.2 7.7 180x10 180 8 9548 9522 0.997 0.0009 W200x46 203 11.0 203 7.2 7.7 180x8 180 8 6 8772 8739 0.996 0.0010 W200x46 203 11.0 203 7.2 7.7 180x6 180 6 8052 8012 0.995 0.0010 W200x42 166 11.8 205 7.2 8.1 140x10 140 10 8 8208 8185 0.995 0.0010 W200x42 166 11.8 205 7.2 8.1 140x10 140 6 6 7592 754 0.995 0.0010 W200x42 166 11.8 205 7.2 8.1 140x6 140 6 6 7592 754 0.995 0.0011 W200x31 134 10.2 201 6.2 8.0		W200x52	204	12.6	206	7.9	7.7	180x6	180	9	9	8852	8802	0.994	0.0009	77.5
W200x46 203 11.0 203 7.2 7.7 180x8 180 8 6 8772 8739 0.996 0.0010 W200x46 203 11.0 203 7.2 7.7 180x6 180 6 6 8052 8012 0.995 0.0010 W200x42 166 11.8 205 7.2 8.1 140x10 140 8 6 7592 7564 0.995 0.0010 W200x42 166 11.8 205 7.2 8.1 140x6 140 6 6 792 759 0.0010 W200x42 166 11.8 205 7.2 8.1 140x6 140 6 6 792 750 0.0010 W200x36 165 10.2 201 6.2 8.0 140x6 140 6 6 6592 6990 0.999 0.0011 W200x37 13 10.2 201 6.2 8.0 140x		W200x46	203	0.11	203	7.2	7.7	180x10	180	9	∞	9548	9522	0.997	0.0009	87.4
W200x46 203 11.0 203 7.2 17.1 180x6 180 6 6 8052 8012 0.995 0.0010 W200x42 166 11.8 205 7.2 8.1 140x10 140 8 8208 8185 0.997 0.0010 W200x42 166 11.8 205 7.2 8.1 140x6 140 6 6 7592 7564 0.995 0.0010 W200x42 166 11.8 205 7.2 8.1 140x6 140 6 6 7592 7564 0.995 0.0010 W200x36 165 10.2 201 6.2 8.0 140x10 140 6 6 7592 7594 0.0011 W200x36 165 10.2 201 6.2 8.0 140x10 140 6 6 7592 7594 0.0011 W200x37 134 10.2 210 6.4 7.9 110x10	_	W200x46	203	0.11	203	7.2	7.7	180x8	180	œ	9	8772	8739	0.996	0.0010	78.0
W200x42 166 11.8 205 7.2 8.1 140x10 140 8 8208 8185 0.997 0.0010 W200x42 166 11.8 205 7.2 8.1 140x8 140 6 6 7592 7564 0.996 0.0010 W200x42 166 11.8 205 7.2 8.1 140x6 140 6 6 7592 7564 0.996 0.0010 W200x36 165 10.2 201 6.2 8.0 140x6 140 6 6 7582 758 0.0011 W200x36 165 10.2 201 6.2 8.0 140x6 140 6 6 6292 6265 0.996 0.0011 W200x37 134 10.2 210 6.4 7.9 110x8 110 8 6 5822 6850 0.996 0.0013 W200x37 133 8.4 207 5.8 7.8 <	~	W200x46	203	0.11	203	7.2	7.7	180x6	180	9	9	8052	8012	0.995	0.0010	69.5
W200X42 166 II.8 205 7.2 8.1 140x8 140 8 6 7592 7564 0.996 0.0010 W200X42 166 II.8 205 7.2 8.1 140x6 140 6 6 7032 6999 0.995 0.0011 W200X36 165 10.2 201 6.2 8.0 140x8 140 6 6 7032 6999 0.995 0.0011 W200X36 165 10.2 201 6.2 8.0 140x8 140 6 6 6292 6285 0.996 0.0011 W200X36 165 10.2 201 6.2 8.0 140x8 110 8 6 6292 6285 0.996 0.0011 W200X31 134 10.2 210 6.4 7.9 110x8 110 8 6 5822 5820 0.999 0.0013 W200X31 133 8.4 207 <		W200x42	991	8.1.	205	7.2	8.1	140x10	140	0	œ	8208	8185	0.997	0.0010	74.4
W200X42 166 11.8 205 7.2 8.1 140x6 140 6 6 7032 6999 0.995 0.0011 W200X36 165 10.2 201 6.2 8.0 140x10 140 10 8 7468 7451 0.998 0.0011 W200X36 165 10.2 201 6.2 8.0 140x6 140 6 6 6292 6830 0.997 0.0011 W200X31 134 10.2 210 6.4 7.9 110x10 110 8 6.328 6320 0.999 0.0013 W200X31 134 10.2 210 6.4 7.9 110x8 110 8 6 5822 6265 0.996 0.0013 W200X31 134 10.2 210 6.4 7.9 110x8 110 8 6 5822 5820 0.999 0.0013 W200X31 133 8.4 207 5.8		W200x42	991	8. 	205	7.2	8 .	140x8	140	∞	9	7592	7564	0.996	0.0010	8.99
W200x36 165 10.2 201 6.2 8.0 140x10 140 10 8 7468 7451 0.998 0.0011 W200x36 165 10.2 201 6.2 8.0 140x6 140 6 6 6292 6285 0.996 0.0013 W200x36 165 10.2 201 6.2 8.0 140x6 140 6 6 6292 6265 0.996 0.0013 W200x31 134 10.2 210 6.4 7.9 110x10 110 8 6 5332 5820 0.998 0.0013 W200x31 134 10.2 210 6.4 7.9 110x8 110 8 6 5322 5820 0.998 0.0013 W200x27 133 8.4 207 5.8 7.8 110x6 110 6 5022 5924 6994 0.998 0.0016 W200x27 133 8.4 207		W200x42	991	8.1	205	7.2	8.1	140x6	140	9	9	7032	6669	0.995	0.0011	09.1
165 10.2 201 6.2 8.0 140x8 140 8 6 6852 6830 0.997 0.0011 165 10.2 201 6.2 8.0 140x6 140 6 6 6292 6265 0.996 0.0013 134 10.2 210 6.4 7.9 110x10 110 8 6 5832 5820 0.996 0.0013 134 10.2 210 6.4 7.9 110x8 110 8 6 5832 5820 0.996 0.0013 133 8.4 207 5.8 7.8 110x8 110 7 6 5002 4994 0.999 0.0015 133 8.4 207 5.8 7.8 110x7 110 7 6 5002 4994 0.999 0.0015 153 8.4 207 5.8 7.8 110x6 110 6 6 5022 5042 5095		W200x36	165	10.2	201	6.2	8.0	140×10	140	0	∞	7468	7451	0.998	0.0011	2.99
165 10.2 201 6.2 8.0 140x6 140 6 6 6292 6265 0.996 0.0013 134 10.2 210 6.4 7.9 110x10 110 8 6 328 6320 0.999 0.0013 134 10.2 210 6.4 7.9 110x6 110 6 5 332 5820 0.999 0.0013 134 10.2 210 6.4 7.9 110x6 110 6 6 5392 5375 0.999 0.0015 133 8.4 207 5.8 7.8 110x7 110 7 6 5002 4994 0.999 0.0015 133 8.4 207 5.8 7.8 110x6 110 6 6 4772 9.99 0.0016 153 9.3 157 6.6 6.0 130x8 130 8 6 5925 5925 0.997 0.0011 153		W200x36	165	10.2	201	6.2	8.0	140x8	140	œ	9	6852	6830	0.997	0.0011	59.4
134 10.2 210 6.4 7.9 110x10 110 8 6328 6320 0.999 0.0012 134 10.2 210 6.4 7.9 110x8 110 8 6 5832 5820 0.998 0.0013 134 10.2 210 6.4 7.9 110x6 110 6 6 5392 5375 0.998 0.0015 133 8.4 207 5.8 7.8 110x7 110 7 6 5002 4994 0.998 0.0016 133 8.4 207 5.8 7.8 110x6 110 6 6 4782 4772 0.998 0.0016 153 9.3 157 6.6 6.0 130x8 130 8 6 5642 5625 0.997 0.0011 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.996 0.		W200x36	165	10.2	201	6.2	8.0	140x6	140	9	9	6292	6265	966'0	0.0013	52.9
134 10.2 210 6.4 7.9 110x8 110 8 6 5832 5820 0.998 0.0013 134 10.2 210 6.4 7.9 110x6 110 6 6 5392 5375 0.997 0.0015 133 8.4 207 5.8 7.8 110x7 110 7 6 5002 4994 0.999 0.0015 133 8.4 207 5.8 7.8 110x7 110 7 6 5002 4994 0.999 0.0016 153 8.4 207 5.8 7.8 110x7 110 6 6 4782 4772 0.998 0.0016 153 9.3 157 6.6 6.0 130x8 130 8 6 5642 5643 0.997 0.0011 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.996 <th></th> <th>W200x31</th> <th>134</th> <th>10.2</th> <th>210</th> <th>6.4</th> <th>7.9</th> <th>110x10</th> <th>01</th> <th>0</th> <th>œ</th> <th>6328</th> <th>6320</th> <th>0.999</th> <th>0.0012</th> <th>59.5</th>		W200x31	134	10.2	210	6.4	7.9	110x10	01	0	œ	6328	6320	0.999	0.0012	59.5
134 10.2 210 6.4 7.9 110x6 110 6 6 5392 5375 0.997 0.0015 133 8.4 207 5.8 7.8 110x8 110 7 6 5002 4994 0.999 0.0016 133 8.4 207 5.8 7.8 110x7 110 6 6 4782 4772 0.998 0.0016 153 9.3 157 6.6 6.0 130x8 130 8 6 5942 5925 0.997 0.0011 153 9.3 157 6.6 6.0 130x7 130 7 6 5682 5663 0.997 0.0011 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.996 0.0012 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.996 0.0012 313 28.1 333 18.0 16.3 350x25 <		W200x31	134	10.2	210	6.4	7.9	110x8	011	œ	9	5832	5820	0.998	0.0013	53.1
133 8.4 207 5.8 7.8 110x8 110 8 6 5222 5216 0.999 0.0015 133 8.4 207 5.8 7.8 110x7 110 7 6 5002 4994 0.998 0.0016 153 8.4 207 5.8 7.8 110x6 110 6 6 4782 4772 0.998 0.0016 153 9.3 157 6.6 6.0 130x8 130 8 6 5682 5663 0.997 0.0011 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.997 0.0012 153 9.3 180 16.3 350x25 350 25 22 41268 41202 0.998 0.0003 313 28.1 333 18.0 16.3 350x20 350 20 18 37448 37350 0.997 0.0003		W200x31	134	10.2	210	6.4	7.9	110x6	011	9	9	5392	5375	0.997	0.0015	47.6
133 8.4 207 5.8 7.8 110x7 110 7 6 5002 4994 0.998 0.0016 133 8.4 207 5.8 7.8 110x6 110 6 6 4782 4772 0.998 0.0016 153 9.3 157 6.6 6.0 130x8 130 7 6 5682 5663 0.997 0.0011 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.997 0.0012 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.996 0.0012 313 28.1 333 18.0 16.3 350x25 350 25 22 41268 41202 0.998 0.0003 313 28.1 333 18.0 16.3 350x20 350 20 18 37448 37350 0.997 0.0003		W200x27	133	8.4	207	5.8	7.8	110x8	0	œ	9	5222	5216	0.999	0.0015	47.0
133 8.4 207 5.8 7.8 110x6 110 6 6 4782 4772 0.998 0.0016 153 9.3 157 6.6 6.0 130x8 130 8 6 5942 5925 0.997 0.0011 153 9.3 157 6.6 6.0 130x7 130 7 6 5682 5663 0.997 0.0011 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.996 0.0012 313 28.1 333 18.0 16.3 350x25 350 25 22 41268 41202 0.998 0.0003 313 28.1 333 18.0 16.3 350x20 350 20 18 37448 37350 0.997 0.0003		W200x27	133	8.4	207	5.8	7.8	110x7	011	7	9	5005	4994	0.998	0.0016	44.3
153 9.3 157 6.6 6.0 130x8 130 8 6 5942 5925 0.997 0.0011 153 9.3 157 6.6 6.0 130x7 130 7 6 5682 5663 0.997 0.0011 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.996 0.0012 313 28.1 333 18.0 16.3 350x25 350 25 22 41268 41202 0.998 0.0003 313 28.1 333 18.0 16.3 350x20 350 20 18 37448 37350 0.997 0.0003		W200x27	133	8.4	207	5.8	7.8	110x6	011	9	9	4782	4772	0.998	0.0016	41.6
153 9.3 157 6.6 6.0 130x7 130 7 6 5682 5663 0.997 0.0011 153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.996 0.0012 313 28.1 333 18.0 16.3 350x25 350 25 22 41268 41202 0.998 0.0003 313 28.1 333 18.0 16.3 350x20 350 20 18 37448 37350 0.997 0.0003		W150x30	153	9.3	157	9.9	0.9	130x8	130	∞	9	5942	5925	0.997	0.0011	31.8
153 9.3 157 6.6 6.0 130x6 130 6 6 5422 5400 0.996 0.0012 313 28.1 333 18.0 16.3 350x25 350 25 22 41268 41202 0.998 0.0003 313 28.1 333 18.0 16.3 350x20 350 20 18 37448 37350 0.997 0.0003		W150x30	153	9.3	157	9.9	0.9	130x7	130	7	9	5682	2 663	0.997	0.0011	29.9
313 28.1 333 18.0 16.3 350x25 350 25 22 41268 41202 0.998 0.0003 313 28.1 333 18.0 16.3 350x20 350 20 18 37448 37350 0.997 0.0003		W150x30	153	9.3	157	9.9	0.9	130x6	130	9	9	5422	5400	966.0	0.0012	28.0
313 28.1 333 18.0 16.3 350x20 350 20 18 37448 37350 0.997 0.0003		W310x179	313	28.1	333	18.0	16.3	350x25	350	25	22	41268	41202	0.998	0.0003	653.8
		W310x179	313	28.1	333	0.81	16.3	350x20	350	20	<u>&</u>	37448	37350	0.997	0.0003	608.1

Table 5.2.a (Cont'd)

		Fla	Flange		qp/	Fillet		PĮį	Plate	Weld					
No.	. W-Section	þ	<u>.</u> -	p	3	*	Plate	ď	P _G	50	4	«	ρ	>	L
		(mm)	(mm)	(mm)	(mm)	(mm)		(mm)	(mm)	(mm)	(mm ²)	(mm ²)			(10°mm ⁴)
92	W310x179	313	28.1	333	0.81	16.3	350x16	350	91	14	34392	34268	966.0	0.0004	571.7
11	W310x158	310	25.1	327	15.5	17.0	350x25	350	25	23	38658	38614	0.999	0.0003	597.0
78	W310x158	310	25.1	327	15.5	17.0	350x20	350	20	<u>8</u>	34748	34671	0.998	0.0004	548.8
79	W310x158	310	25.1	327	15.5	17.0	350x16	350	91	4	31692	31589	0.997	0.0004	512.7
80	W310x158	310	25.1	327	15.5	17.0	350x12	350	12	0	28700	28571	0.996	0.0005	478.6
∞	W310x143	309	22.9	323	14.0	13.9	350x20	350	20	<u>«</u>	32848	32788	0.998	0.0004	508.3
82	W310x143	306	22.9	323	14.0	13.9	350x16	350	91	7	29792	29706	0.997	0.0004	472.3
83	W310x143	309	22.9	323	14.0	13.9	350x12	350	12	2	26800	26688	0.996	0.0005	438.4
%	W310x143	309	22.9	323	14.0	13.9	350x10	350	01	∞	25328	25203	0.995	0.0005	422.1
82	W310x129	308	20.6	318	13.1	14.7	350x20	350	20	<u>&</u>	31148	31109	0.999	0.0004	468.1
	W310x129	308	20.6	318	13.1	14.7	350x16	350	91	1	28092	28027	0.998	0.0005	432.4
	W310x129	308	20.6	318	13.1	14.7	350x12	350	12	01	25100	25009	966.0	0.0005	398.6
	W310x129	308	20.6	318	13.1	14.7	350x10	350	01	∞	23628	23524	966.0	9000'0	382.4
	W310x118	307	18.7	314	6.11	16.2	350x16	350	91	4	26592	26542	0.998	0.0005	400.3
	W310x118	307	18.7	314	6.11	16.2	350x12	350	12	0	23600	23524	0.997	9000.0	366.6
	W310x118	307	18.7	314	6.11	16.2	350x10	350	0	œ	22128	22039	966.0	9000'0	350.5
	W310x118	307	18.7	314	6.11	16.2	350x8	350	∞	9	20672	20570	0.995	9000.0	334.7
	W310x107	306	17.0	311	6.01	14.3	350x16	350	91	4	25192	25157	0.999	0.0005	372.0
	W310x107	306	17.0	311	6.01	14.3	350x12	350	12	0	22200	22139	0.997	9000.0	338.4
95	W310x107	306	17.0	311	6.01	14.3	350x10	350	01	∞	20728	20654	966.0	9000.0	322.3
	W310x107	306	17.0	311	6.01	14.3	350x8	350	∞	9	19272	19185	0.995	0.0007	306.6
6	W310x86	254	16.3	310	9.1	15.1	330x16	330	91	4	21952	21941	000.1	9000.0	304.3
86	W310x86	254	16.3	310	9.1	15.1	330x12	330	12	01	19120	19085	0.998	0.0007	275.4
8	W310x86	254	16.3	310	9.1	15.1	330x10	330	0	∞	17728	17680	0.997	0.0007	261.6
2	W310x86	254	16.3	310	9.1	15.1	330x8	330	8	9	16352	16292	966.0	0.0008	248.2

Table 5.2.a (Cont'd)

		먑	Flange	M	့	Fillet		Pl	Plate	Weld					
No.	W-Section	þ	ت	D	3	-*	Plate	þ	ء۔ ا	8	V	«	β	>	-×
		(mm)	(mm)	(mm)	(mm)	(mm)		(mm)	(mm)	(mm)	(mm ²)	(mm ²)			(10°mm ⁴)
<u></u>	W310x79	254	14.6	306	8.8	17.0	330x12	330	12	10	18220	18199	0.999	0.0007	254.4
102	W310x79	254	14.6	306	8.8	17.0	330x10	330	01	×	16828	16794	0.998	0.0008	240.6
103	W310x79	254	14.6	306	8.8	17.0	330x8	330	∞	9	15452	15406	0.997	0.0008	227.3
<u>8</u>	W310x67	204	14.6	306	8.5	15.3	330x12	330	12	9	16630	16635	000.1	0.0007	222.0
105	W310x67	204	14.6	306	8.5	15.3	330x10	330	01	∞	15238	15231	000.	0.0008	208.2
901	W310x67	204	14.6	306	8.5	15.3	330x8	330	∞	9	13862	13843	0.999	0.0009	194.9
107	W310x60	203	13.1	303	7.5	15.1	330x12	330	12	0	15710	15721	1.00.1	0.0008	205.3
108	W310x60	203	13.1	303	7.5	15.1	330×10	330	01	œ	14318	14317	000.	0.0009	9.161
8	W310x60	203	13.1	303	7.5	15.1	330x8	330	œ	9	12942	12929	0.999	0.0000	178.3
0 - -	W250x73	254	14.2	253	9.8	12.5	280x12	280	12	01	16200	16163	0.998	0.0007	0.091
Ξ	W250x73	254	14.2	253	9.8	12.5	280x10	280	0	∞	15008	14960	0.997	0.0007	151.4
112	W250x73	254	14.2	253	9.8	12.5	280x8	280	œ	9	13832	13774	966.0	0.0008	143.2
113	W250x73	254	14.2	253	9.8	12.5	280x6	280	9	9	12712	12644	0.995	0.0008	135.8
-	W250x58	203	13.5	252	8.0	12.7	280x12	280	12	0	14340	14333	000.	0.0007	134.5
115	W250x58	203	13.5	252	8.0	12.7	280×10	280	01	∞	13148	13131	0.999	0.0008	126.0
911	W250x58	203	13.5	252	8.0	12.7	280x8	280	∞	9	11972	11944	0.998	0.000	117.7
117	W250x58	203	13.5	252	8.0	12.7	280x6	280	9	9	10852	10814	966.0	0.0010	110.4
8	W250x49	202	0.11	247	7.4	12.8	280×10	280	01	∞	11978	11977	000.	0.0009	109.3
611	W250x49	202	0.11	247	7.4	12.8	280x8	280	∞	9	10802	10790	0.999	0.0010	0.101
120	W250x49	202	0.11	247	7.4	12.8	280x6	280	9	9	9682	0996	0.998	0.0011	93.7
121	W200x59	202	14.2	210	1.6	8.0	230x12	230	12	0	13250	13224	0.998	0.0007	9.78
122	W200x59	205	14.2	210	1.6	8.0	230x10	230	0	∞	12258	12223	0.997	0.0007	82.6
123	W200x59	205	14.2	210	1.6	8.0	230x8	230	œ	9	11282	11239	966.0	0.0008	6.77
124	W200x59	205	14.2	210	1.6	8.0	230x6	230	9	9	10362	10310	0.995	0.0008	73.9
125	W200x52	204	12.6	206	7.9	7.7	230x12	230	12	2	12340	12322	0.999	0.0007	0.67

Table 5.2.a (Cont'd)

No. W-Section b ₁ t ₁ d w k Plate b ₁ t ₁ d mm) (mm) (mm) (mm ¹) (mm ¹			Fla	Flange	≱	qə	Fillet		Pla	jį.	Weld					
126 W200x52 204 126 206 7.9 7.7 230x10 230 10 8 11348 11321 0.998 0.0008 128 W200x52 204 126 206 7.9 7.7 230x8 230 8 6 10372 1037 0.997 0.0008 128 W200x52 204 126 206 7.9 7.7 230x8 230 8 6 10372 1037 0.997 0.0008 128 W200x46 203 11.0 203 7.2 7.7 230x8 230 8 6 9452 9408 0.995 0.0009 130 W200x46 203 11.0 203 7.2 7.7 230x8 230 8 6 9452 9408 0.995 0.0009 131 W200x46 203 11.0 203 7.2 7.7 230x8 230 8 6 9452 9408 0.995 0.0009 131 W200x42 166 11.8 205 7.2 8.1 230x8 230 8 6 9052 9546 0.995 0.0009 132 W200x42 166 11.8 205 7.2 8.1 230x8 230 8 6 9052 9209 0.0009 133 W200x42 166 11.8 205 7.2 8.1 230x8 230 8 6 9328 9208 0.0010 135 W200x42 166 11.8 205 7.2 8.1 230x8 230 8 6 8252 8618 0.995 0.0010 135 W200x42 166 11.8 205 7.2 8.1 230x8 230 8 6 8252 8244 0.999 0.0010 135 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 8325 8244 0.999 0.0010 135 W200x36 154 10.2 201 6.4 7.9 230x10 230 6 6 8325 8324 0.999 0.0010 135 W200x31 134 10.2 210 6.4 7.9 230x10 230 6 6 8325 6829 1.001 0.0011 140 W200x27 133 8.4 207 5.8 7.8 230x8 230 6 6 6262 6500 0.0010 140 W200x30 153 8.4 207 5.8 7.8 230x8 230 6 6 6262 6500 0.0010 0.0011 140 W150x30 153 9.3 157 6.6 6.0 180x7 180 7 6 6382 6399 0.0010 0.0011 140 W150x30 153 9.3 157 6.6 6.0 180x7 180 7 6 6382 6390 0.0010 0.0011 140 W150x30 153 9.3 157 6.6 6.0 180x7 180 7 6 6025 6020 0.999 0.0010 140 W150x30 153 9.3 157 6.6 6.0 180x7 180 7 6 6028 6020 6020 0.991 0.0011 140 W150x30 153 9.3 157 6.6 6.0 180x7 180	Š.		þ	ح	P	3	~	Plate	p	t _p	ಎ	<	∢	ρ	>	Ľ
126 W200x52 204 12.6 206 7.9 7.7 230x10 230 10 8 11348 11321 0.998 0.0008 127 W200x52 204 12.6 206 7.9 7.7 230x8 230 8 6 10372 10377 0.997 0.0008 128 W200x52 204 12.6 206 7.9 7.7 230x6 230 6 6 9452 9408 0.997 0.0009 129 W200x46 203 11.0 203 7.2 7.7 230x6 230 6 6 9452 9408 0.0997 0.0009 130 W200x42 166 11.8 205 7.2 8.1 230x10 230 6 6 8452 8418 0.995 0.0009 131 W200x42 166 11.8 205 7.2 8.1 230x10 230 6 6 8452 8618 0.995	ļ		(mm)	(mm)	(mm)	(mm)	(mm)		(mm)	(mm)	(mm)	(mm ²)	(mm ²)			(10°mm ⁴)
127 W200x52 204 12.6 206 7.9 7.7 230x8 230 8 6 10372 10337 0.997 0.0008 128 W200x52 204 12.6 206 7.9 7.7 230x6 230 6 6 9452 9408 0.995 0.0009 129 W200x54 203 11.0 203 7.2 7.7 230x10 230 6 6 9452 9408 0.995 0.0009 130 W200x46 203 11.0 203 7.2 7.7 230x10 8 6 9572 9546 0.997 0.0009 131 W200x46 203 11.0 203 7.2 7.7 230x6 230 6 6 9452 9408 0.999 0.0009 131 W200x42 166 11.8 205 7.2 8.1 230x6 230 6 6 9452 8618 0.996 0.0010 <th>126</th> <th>W200x52</th> <th>204</th> <th>12.6</th> <th>206</th> <th>7.9</th> <th>7.7</th> <th>230x10</th> <th>230</th> <th>9</th> <th>∞</th> <th>11348</th> <th>11321</th> <th>0.998</th> <th>0.0008</th> <th>74.1</th>	126	W200x52	204	12.6	206	7.9	7.7	230x10	230	9	∞	11348	11321	0.998	0.0008	74.1
128 W200X52 204 12.6 206 7.9 7.7 230x6 230 6 9452 9408 0.995 0.0009 129 W200X46 203 11.0 203 7.2 7.7 230x10 230 6 6 9452 9408 0.995 0.0009 130 W200X46 203 11.0 203 7.2 7.7 230x10 230 6 6 8652 8618 0.995 0.0009 131 W200X46 203 11.0 203 7.2 7.7 230x10 230 6 6 8652 8618 0.995 0.0009 131 W200X46 206 11.8 205 7.2 8.1 230x10 230 6 6 8652 8618 0.995 0.0009 134 W200X42 166 11.8 205 7.2 8.1 230x10 230 6 6 8112 8095 0.997 0.0010	127	W200x52	5 8	12.6	206	7.9	7.7	230x8	230	œ	9	10372	10337	0.997	0.0008	69.4
129 W200x46 203 11.0 203 7.2 7.7 230x10 230 10 8 10548 10531 0.998 0.0008 130 W200x46 203 11.0 203 7.2 7.7 230x8 230 6 6 8652 8618 0.997 0.0009 131 W200x46 203 11.0 203 7.2 7.7 230x6 230 6 6 8652 8618 0.999 0.0010 132 W200x42 166 11.8 205 7.2 8.1 230x10 8 6 6 8652 8618 0.999 0.0010 133 W200x42 166 11.8 205 7.2 8.1 230x8 230 6 6 8112 8090 0.0010 134 W200x42 166 11.8 205 7.2 8.1 230x8 230 6 6 8112 8099 0.0010 13	128	W200x52	504	12.6	206	7.9	7.7	230x6	230	9	9	9452	9408	0.995	0.000	65.4
130 W200x46 203 11.0 203 7.2 7.7 230x8 230 8 6 9572 9546 0.997 0.0009 131 W200x46 203 11.0 203 7.2 7.7 230x6 230 6 6 8652 8618 0.999 0.0010 132 W200x42 166 11.8 205 7.2 8.1 230x10 230 6 6 8652 8618 0.999 0.0010 133 W200x42 166 11.8 205 7.2 8.1 230x10 230 6 6 8112 8089 0.997 0.0010 134 W200x42 166 11.8 205 7.2 8.1 230x10 230 6 6 8112 8090 0.0010 135 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 8110 8000 9 6 8100 </th <th>129</th> <th>W200x46</th> <th>203</th> <th>0.11</th> <th>203</th> <th>7.2</th> <th>7.7</th> <th>230x10</th> <th>230</th> <th>0</th> <th>∞</th> <th>10548</th> <th>10531</th> <th>0.998</th> <th>0.0008</th> <th>8.99</th>	129	W200x46	203	0.11	203	7.2	7.7	230x10	230	0	∞	10548	10531	0.998	0.0008	8.99
131 W200x46 203 11.0 203 7.2 7.7 230x6 230 6 6 8652 8618 0.996 0.0010 132 W200x42 166 11.8 205 7.2 8.1 230x10 230 6 6 8652 8618 0.999 0.0010 133 W200x42 166 11.8 205 7.2 8.1 230x8 230 6 6 8112 8089 0.997 0.0010 134 W200x42 166 11.8 205 7.2 8.1 230x8 230 6 6 8112 8089 0.997 0.0010 135 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 8292 8284 0.999 0.0010 136 W200x36 165 10.2 201 6.2 8.0 230x8 230 6 6 8208 1.000 0.0010	130	W200x46	203	0.11	203	7.2	7.7	230x8	230	œ	9	9572	9546	0.997	0.000	62.2
132 W200x42 166 11.8 205 7.2 8.1 230x10 230 10 8 10008 10002 0.999 0.0008 133 W200x42 166 11.8 205 7.2 8.1 230x8 230 6 6 8112 8089 0.997 0.0010 134 W200x42 166 11.8 205 7.2 8.1 230x6 230 6 6 8112 8089 0.997 0.0010 135 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 8228 9.268 1.000 0.0009 136 W200x36 165 10.2 201 6.2 8.0 230x10 23 6 6 8228 8.08 0.099 0.0010 137 W200x36 165 10.2 201 6.4 7.9 230x8 230 6 6 8228 1.000 0.0011 </th <th>131</th> <th>W200x46</th> <th>203</th> <th>0.11</th> <th>203</th> <th>7.2</th> <th>7.7</th> <th>230x6</th> <th>230</th> <th>9</th> <th>9</th> <th>8652</th> <th>8198</th> <th>966.0</th> <th>0.0010</th> <th>58.1</th>	131	W200x46	203	0.11	203	7.2	7.7	230x6	230	9	9	8652	8198	966.0	0.0010	58.1
133 W200x42 166 11.8 205 7.2 8.1 230x8 230 8 6 9032 9017 0.998 0.0009 134 W200x42 166 11.8 205 7.2 8.1 230x6 230 6 6 8112 8089 0.997 0.0010 135 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 8112 8089 0.997 0.0010 135 W200x36 165 10.2 201 6.2 8.0 230x10 8 6 8292 8284 0.999 0.0010 137 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 7752 7757 1.002 0.0010 138 W200x31 134 10.2 210 6.4 7.9 230x8 230 6 6 8725 879 0.001 14	132	W200x42	991	8. =	205	7.2		230x10	230	9	œ	10008	10002	0.999	0.0008	62.3
134 W200x42 166 11.8 205 7.2 8.1 230x6 230 6 6 8112 8089 0.997 0.0010 135 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 8128 9268 1.000 0.0009 136 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 7372 7355 0.998 0.0010 137 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 7372 7355 0.998 0.0010 138 W200x31 134 10.2 210 6.4 7.9 230x8 230 6 6 6832 829 0.0010 139 W200x31 134 10.2 210 6.4 7.9 230x8 230 6 6 6832 6829 1.000 0.0011	133	W200x42	991	8.	205	7.2		230x8	230	∞	9	9032	9017	0.998	0.0000	57.6
135 W200x36 165 10.2 201 6.2 8.0 230x10 230 10 8 9268 9268 1.000 0.0009 136 W200x36 165 10.2 201 6.2 8.0 230x8 230 8 6 8292 8284 0.999 0.0010 137 W200x36 165 10.2 201 6.2 8.0 230x10 230 6 6 7752 7757 1.001 0.0010 139 W200x31 134 10.2 210 6.4 7.9 230x6 230 6 6 7752 7757 1.001 0.0010 139 W200x31 134 10.2 210 6.4 7.9 230x6 230 6 6 6832 6829 1.000 0.0010 140 W200x31 134 10.2 210 6.4 7.9 230x6 230 6 6 6832 6829 1.000 <td< th=""><th>10 2</th><td>W200x42</td><td>991</td><td>8.</td><td>205</td><td>7.2</td><td></td><td>230x6</td><td>230</td><td>9</td><td>9</td><td>8112</td><td>808</td><td>0.997</td><td>0.0000</td><td>53.6</td></td<>	10 2	W200x42	991	8.	205	7.2		230x6	230	9	9	8112	808	0.997	0.0000	53.6
W200x36 165 10.2 201 6.2 8.0 230x8 230 8 6 8292 8284 0.999 0.0010 W200x36 165 10.2 201 6.2 8.0 230x6 230 6 6 7372 735 0.998 0.0011 W200x31 134 10.2 210 6.4 7.9 230x10 230 8 6 7752 7757 1.002 0.0011 W200x31 134 10.2 210 6.4 7.9 230x6 230 8 6 7752 7757 1.002 0.0012 W200x31 134 10.2 210 6.4 7.9 230x6 230 8 6 7752 7757 1.002 0.0011 W200x31 134 10.2 210 6.4 7.9 230x8 230 8 6 7154 1.050 0.0012 W200x27 133 8.4 207 5.8 <t< th=""><th>135</th><td>W200x36</td><td>165</td><td>10.2</td><td>201</td><td>6.2</td><td></td><td>230x10</td><td>230</td><td>01</td><td>œ</td><td>9268</td><td>9268</td><td>000.</td><td>0.000</td><td>55.8</td></t<>	135	W200x36	165	10.2	201	6.2		230x10	230	01	œ	9268	9268	000.	0.000	55.8
W200x36 165 10.2 201 6.2 8.0 230x6 230 6 6 7372 7355 0.998 0.0011 W200x31 134 10.2 210 6.4 7.9 230x10 230 6 6 7752 7757 1.001 0.0011 W200x31 134 10.2 210 6.4 7.9 230x6 230 6 6 6832 6829 1.001 0.0011 W200x37 133 8.4 207 5.8 7.8 230x7 230 7 6 6682 6690 1.001 0.0013 W200x27 133 8.4 207 5.8 7.8 230x7 230 6 6 6682 6690 1.001 0.0013 W200x27 133 8.4 207 5.8 7.8 230x6 230 6 6 6682 6690 1.001 0.0013 W150x30 153 9.3 157 <th< th=""><th>136</th><td>W200x36</td><td>165</td><td>10.2</td><td>201</td><td>6.2</td><td></td><td>230x8</td><td>230</td><td>∞</td><td>9</td><td>8292</td><td>8284</td><td>0.999</td><td>0.000</td><td>51.1</td></th<>	136	W200x36	165	10.2	201	6.2		230x8	230	∞	9	8292	8284	0.999	0.000	51.1
W200x31 134 10.2 210 6.4 7.9 230x10 230 8 6 7752 7757 1.002 0.0010 W200x31 134 10.2 210 6.4 7.9 230x6 230 8 6 7752 7757 1.001 0.0011 W200x31 134 10.2 210 6.4 7.9 230x6 230 6 6 6832 6829 1.000 0.0012 W200x27 133 8.4 207 5.8 7.8 230x7 230 7 6 6682 6690 1.001 0.0013 W200x27 133 8.4 207 5.8 7.8 230x7 230 6 6 6622 6690 1.001 0.0013 W200x27 133 8.4 207 5.8 7.8 230x6 230 6 6 6222 6226 1.001 0.001 W150x30 153 9.3 157	137	W200x36	165	10.2	201	6.2		230x6	230	9	9	7372	7355	0.998	0.0011	47.1
W200x31 134 10.2 210 6.4 7.9 230x8 230 8 6 7752 7757 1.001 0.0011 W200x31 134 10.2 210 6.4 7.9 230x6 230 6 6 6832 6829 1.000 0.0012 W200x27 133 8.4 207 5.8 7.8 230x7 230 7 6 6682 6690 1.001 0.0013 W200x27 133 8.4 207 5.8 7.8 230x6 230 6 6 6682 6690 1.001 0.0013 W200x27 133 8.4 207 5.8 7.8 230x6 230 6 6 6222 6226 1.001 0.0013 W150x30 153 9.3 157 6.6 6.0 180x7 180 7 6 6382 6369 0.999 0.0010 W150x30 153 9.3 157 6	138	W200x31	134	10.2	210	6.4		230x10	230	01	∞	8728	8742	1.002	0.0010	53.2
W200x31 134 10.2 210 6.4 7.9 230x6 230 6 6 6832 6829 1.000 0.0012 W200x27 133 8.4 207 5.8 7.8 230x7 230 7 6 6682 6690 1.001 0.0013 W200x27 133 8.4 207 5.8 7.8 230x6 230 6 6 6222 6226 1.001 0.0013 W150x30 153 9.3 157 6.6 6.0 180x8 180 8 6 6742 6733 0.999 0.0010 W150x30 153 9.3 157 6.6 6.0 180x6 180 6 6 6022 6066 0.998 0.0010 W150x30 153 9.3 157 6.6 6.0 180x6 180 6 6 6022 6006 0.997 0.0011	139	W200x31	134	10.2	210	6.4		230x8	230	œ	9	7752	7757	<u>.00.</u>	0.0011	48.4
W200x27 133 8.4 207 5.8 7.8 230x8 230 8 6 7142 7154 1.002 0.0012 W200x27 133 8.4 207 5.8 7.8 230x7 230 7 6 6682 6690 1.001 0.0013 W200x27 133 8.4 207 5.8 7.8 230x6 230 6 6 6222 6226 1.001 0.0014 W150x30 153 9.3 157 6.6 6.0 180x7 180 7 6 6382 6369 0.999 0.0010 W150x30 153 9.3 157 6.6 6.0 180x6 180 6 6 6022 6006 0.997 0.0011	140	W200x31	134	10.2	210	6.4		230x6	230	9	9	6832	6839	000.	0.0012	44.4
W200x27 133 8.4 207 5.8 7.8 230x7 230 7 6 6682 6690 1.001 0.0013 W200x27 133 8.4 207 5.8 7.8 230x6 230 6 6 6222 6226 1.001 0.0014 W150x30 153 9.3 157 6.6 6.0 180x7 180 7 6 6382 6369 0.999 0.0010 W150x30 153 9.3 157 6.6 6.0 180x6 180 6 6 6022 6369 0.998 0.0010 W150x30 153 9.3 157 6.6 6.0 180x6 180 6 6 6022 6006 0.997 0.0011	14	W200x27	133	8.4	207	5.8		230x8	230	œ	9	7142	7154	1.002	0.0012	42.9
W200x27 133 8.4 207 5.8 7.8 230x6 230 6 6 6222 6226 1.001 0.0014 W150x30 153 9.3 157 6.6 6.0 180x8 180 8 6 6742 6733 0.999 0.0010 W150x30 153 9.3 157 6.6 6.0 180x6 180 6 6 6022 6006 0.997 0.0011	142	W200x27	133	8.4	207	5.8		230x7	230	7	9	6682	0699	1.00.1	0.0013	40.8
W150x30 153 9.3 157 6.6 6.0 180x8 180 8 6 6742 6733 0.999 0.0010 W150x30 153 9.3 157 6.6 6.0 180x7 180 7 6 6382 6369 0.998 0.0010 W150x30 153 9.3 157 6.6 6.0 180x6 180 6 6 6022 6006 0.997 0.0011	143	W200x27	133	8.4	207	5.8		230x6	230	9	9	6222	6226	1.00.	0.0014	38.8
W150x30 153 9.3 157 6.6 6.0 180x7 180 7 6 6382 6369 0.998 0.0010 W150x30 153 9.3 157 6.6 6.0 180x6 180 6 6 6022 6006 0.997 0.0011	4	W150x30	153	9.3	157	9.9		180x8	180	œ	9	6742	6733	0.999	0.0010	25.4
W150x30 153 9.3 157 6.6 6.0 180x6 180 6 6 6022 6006 0.997 0.0011	145	W150x30	153	9.3	157	9.9		180x7	180	7	9	6382	6969	0.998	0.0010	24.4
	146	W150x30	153	9.3	157	9.9		180x6	180	9	9	6022	9009	0.997	0.0011	23.4

Table 5.2.b Statistical Parameters of the Geometric Properties of the Reinforced Columns

Vr		0.0004	0.0005	0.0005	0.0005	0.0005	9000.0	9000.0	0.0005	0.0006	0.0007	0.0007	9000.0	9000.0	0.0007	0.0008	0.0007	0.0008	0.0008	0.0009	0.0007	0.0008	0.0009	0.0000	0.0008	0.000
ριγ	,	000.	000.	000.1	000.1	000.1	000.1	000.	00.1	000.1	000.	1.00.1	000.1	000.1	000.1	000.	000.1	000.	000.	000.1	000.	000.	000.	000.	000.	1.000
ر ا	(mm)	83.7	82.7	81.9	83.7	82.7	81.7	80.8	82.8	81.8	80.9	80.5	82.7	91.8	9.08	80.1	9.18	80.5	80.0	79.5	81.7	80.5	79.9	79.4	66.4	65.5
ŗ	(mm)	83.7	82.7	81.9	83.6	82.6	81.7	80.8	82.8	8.18	80.9	80.4	82.7	91.8	9.08	80.1	81.5	80.4	79.9	79.4	9.18	80.5	79.9	79.4	66.4	65.5
Vr		0.0003	0.0004	0.0004	0.0003	0.0004	0.0004	0.0005	0.0004	0.0005	0.0005	0.0006	0.0004	0.0005	0.0006	0.0006	0.0005	90000	0.0007	0.0007	0.0006	0.0007	0.0007	0.0008	0.0006	0.0007
ρα		1.00	1.00.1	1.00.1	1.00.1	1.00.1	1.00.1	000.	1.00.1	1.00.1	000.	000.	1.00.1	1.00.	000.	000.	1.00.1	000.	000.1	000.1	1.00.1	000.	000.	000.1	000.1	000.
ے ا	(mm)	157.0	153.8	151.1	156.2	153.0	150.3	147.5	152.5	149.8	147.1	145.6	151.3	148.6	145.8	144.3	147.9	145.0	143.6	142.1	147.5	144.7	143.2	141.6	146.9	144.0
, ×	(mm)	156.8	153.6	151.0	156.0	152.8	150.2	147.5	152.4	149.7	147.0	145.6	151.2	148.5	145.7	144.3	147.8	145.0	143.6	142.1	147.4	144.6	143.2	141.7	146.8	144.0
> _y		0.0008	0.0009	0.0010	0.0000	0.0010	0.0011	0.0012	0.0010	0.0011	0.0013	0.0014	0.0011	0.0012	0.0014	0.0015	0.0013	0.0014	9100.0	0.0017	0.0013	0.0015	0.0016	0.0018	0.0014	0.0016
ρ _{ly}		0.998	0.997	0.996	0.998	0.998	0.997	966.0	0.998	0.997	966.0	0.995	0.998	0.998	966.0	0.996	0.998	0.997	0.996	0.995	0.998	0.997	0.997	966.0	0.998	0.997
- >	(10°mm ⁴)	267.3	239.2	216.8	248.5	220.3	6761	177.2	208.3	0.981	165.2	155.4	196.2	173.8	153.0	143.2	163.7	143.0	133.2	123.7	154.8	134.0	124.2	114.8	82.4	71.5
- ^	(10°mm ⁴)	267.9	239.9	217.6	248.9	220.8	9.861	177.9	208.8	186.5	165.9	156.1	196.5	174.2	153.6	143.8	164.1	143.4	133.7	124.3	155.0	134.4	124.7	115.3	82.6	71.7
>		0.0006	90000	0.0007	0.0006	0.0007	0.0008	0.000	0.0007	0.0008	0.0009	0.0010	0.0008	0.000	0.0010	0.0011	0.000	0.0011	0.0012	0.0013	0.0010	0.0012	0.0013	0.0014	0.0000	0.0012
ρ		1.000	0.998	0.997	000.1	0.999	0.997	0.996	0.999	0.998	966'0	0.995	000.	0.998	966'0	0.995	0.998	0.997	0.995	0.994	0.999	0.997	966'0	0.994	0.999	0.997
-*	10°mm ⁴)	940.5	825.7	738.0	865.7	754.6	2.699	590.3	706.3	623.3	545.7	8.809	9.959	575.9	500.5	464.7	538.0	464.3	429.3	395.6	505.3	432.9	398.6	365.4	403.7	345.6
No.		_	7	က	4	\$	9	7	œ	6	<u> </u>	=	12	13	14	15	91	17	<u>∞</u>	61	70	21	22	23	24	25

Table 5.2.b (Cont'd)

Vr,	0.0010			_							_	_	_	_	_	_	_	_	_	_	_	_	_	_	_
ρη	000.	000.1	000.1	000.	000.1	000.	000.	000.	000.	000.	000.	000.	000.	000.	1.00.1	000.	000.	000.	000.1	000.1	000.1	000.1	000.	000.	000.
	65.1	64.7	65.1	64.6	64.2	51.1	50.7	50.3	51.1	50.7	50.3	66.3	62.9	65.6	65.5	52.0	51.6	51.2	51.2	51.0	50.5	50.4	53.0	52.7	52.4
ry (mm)	65.1	64.7	65.1	64.6	64.2	51.2	50.7	50.4	51.2	50.7	50.3	66.3	62.9	65.6	65.5	52,0	. 51.6	51.2	51.2	51.0	50.5	50.4	53.0	52.7	52.4
V _r	0.0008	0.0008	0.0007	0.0008	0.0009	0.0008	0.0008	0.000	0.0008	0.0009	0.0010	0.0007	0.0008	0.0000	0.0010	0.0008	0.0008	0.000	0.0010	0.0010	0.0011	0.0012	0.0007	0.0008	0.0000
ρα	000.	000.	000.	000.	000.	000.	000.1	0.999	000.1	000.	0.999	1.00.1	000.	000.	000.	000.	000.	000.	000.1	000.1	000.	0.999	1.00.1	1.00.1	000.
r _x	142.5	141.0	142.7	141.2	139.6	141.1	139.5	137.8	141.0	139.3	137.7	119.2	117.8	116.4	115.0	117.9	116.5	115.0	113.5	115.1	113.5	6:111	98.4	97.1	95.7
r _x (mm)	142.5	141.0	142.7	141.2	139.6	141.1	139.5	137.9	141.0	139.4	137.7	119.2	117.8	116.4	115.0	117.9	116.5	115.0	113.5	115.1	113.5	112.0	98.3	97.0	95.7
>,	0.0018	0.0019	0.0017	0.0019	0.0021	0.0018	0.0019	0.0021	0.0019	0.0021	0.0022	0.0018	0.0019	0.0021	0.0023	6100.0	0.0020	0.0022	0.0024	0.0023	0.0025	0.0027	0.0018	0.0020	0.0021
$\rho_{\rm ly}$	966.0	0.995	0.997	0.997	0.996	0.997	966'0	966'0	0.997	0.997	966.0	0.997	0.997	0.996	0.995	0.997	0.997	966.0	0.995	0.998	0.997	966.0	0.997	966.0	0.996
- - - - - - - - - - - - - - - - - - -	66.3	61.5	6.99	61.7	56.9	34.0	31.4	28.9	31.6	29.0	56.6	65.8	60.7	55.8	51.7	32.2	29.6	27.1	25.2	25.9	23.4	21.5	33.7	31.1	28.7
I, (10°mm [±])	9.99	8.19	0.79	62.0	57.1	34.1	31.5	29.1	31.7	29.1	26.7	62.9	6.09	96.0	52.0	32.2	29.7	27.2	25.3	25.9	23.5	21.6	33.8	31.2	28.8
>	0.0013	0.0014	0.0013	0.0014	0.0015	0.0013	0.0014	0.0015	0.0014	0.0015	0.0017	0.0013	0.0014	0.0016	0.0017	0.0013	0.0014	0.0016	0.0018	0.0016	0.0018	0.0020	0.0012	0.0014	0.0015
		0.994	0.997	966'0	0.995	0.998	966'0	0.995	0.998	0.997	966'0	0.998	0.997	0.995	0.994	0.998	0.997	966'0	0.994	0.998	0.997	0.995	866'0	0.997	966.0
	318.1	291.7	321.3	294.5	268.7	258.7	237.4	216.9	240.2	219.2	1.661	212.6	193.8	175.7	159.2	9'591	150.8	136.6	123.7	131.9	118.2	105.8	116,3	105.6	95.5
S. O.	26	27	28	29	30	31	32	33	34	35	36	37	38	36	40	4	45	43	44	45	46	47	48	49	20

Table 5.2.b (Cont'd)

V_{lx} I_y I_y I_y
26.8
0.0014 31.2 31.2
28.7
26.2
24.3
26.2
23.7
21.8
14.3
13.0
12.1
12.9
11.7
6.7
6,1
5.7
5.3
5.1
4.9
8 .8
8.4
8 .
547.1 5

Table 5.2.b (Cont'd)

	ρ_{ry} V_{r_y}		1.007 0.0004	1.008 0.0003	1.007 0.0003	1.007 0.0004	1.007 0.0004	1.008 0.0003	1.007 0.0004	1.007 0.0005	1.007 0.0005							1.007 0.0006		1.007 0.0004		1.007 0.0006	1.007 0.0007	_	1.007 0.0005	
	حي	(mm)		6 129.6	4 123.3	7 117.5	_	2 125.1	_	<u>~</u>	_	_											107.8			
	ŗ,	(mm)	5 115.2	_	5 122.	_	6 110.0	5 124.2	5 118.4	5 111.0	_	5 125.6	_	7 112.9	_	_	_	3 110.3	_	_		_	107.1			
	V _r		0.0005	0.0005	0.0005	0.0005	0.0006	0.000	0.0006	0.0006	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.0008	0.000	0.000	0.000	0.0008	0.000	0.0007	0.0008	(
	ρα		0.998	0.998	0.998	0.998	_	_	_	_	_	0.998	_		_	_	_	0.998	0.998	0.998	0.998	0.998	0.998	0.997	0.997	1
	Ž	(mm)	128.7	124.0	125.4	126.9	128.9	124.1	125.6	127.6	128.8	122.3	123.8	125.7	126.9	122.4	124.4	125.6	127.0	121.2	123.2	124.4	125.8	117.4	119.7	•
	٦,	(mm)	128.9	124.3	125.7	127.2	129.1	124.4	125.9	127.9	129.1	122.6	124.1	126.0	127.2	122.7	124.6	125.8	127.2	121.5	123.5	124.7	126.1	117.7	120.0	
	> ,		0.0006	0.0004	0.0005	0.0006	0.0007	0.0005	0.0006	0.0008	0.000	0.0005	0.0006	0.0008	0.0009	0.0007	0.0008	0.0009	0.0011	0.0007	0.0008	0.0010	0.0011	0.0006	0.0008	
	ριγ		_	1.014	1.013	1.0.1	-000	1.013	1.012	1.010	1.009	1.014	1.012	1.011	1.009	1.013	1.01	1.010	1.008	1.013	1.012	1.0.1	1.009	1.014	1.013	•
	- ^	(10^{6}mm^{4})	460.9	648.7	527.4	436.3	350.4	513.0	422.4	337.0	296.3	498.4	408.3	323.4	282.9	396.4	312.0	271.7	232.8	385.5	301.7	261.7	222.9	246.6	190.2	
	- ^	(10°mm ⁴)	456.1	9,669	520.8	431.4	347.2	506.3	417.5	333.7	293.8	491.6	403.3	320.1	280.3	391.3	308.6	269.1	230.8	380.4	298.2	258.9	220.9	243.1	187.8	/ . / .
	>		0.0000	0.0008	0.000	0.0010	0.000	0.0010	0.0011	0.0011	0.0012	0.0011	0.0011	0.0012	0.0013	0.0012	0.0013	0.0014	0.0015	0.0013	0.0015	0.0015	0.0016	0.0014	0.0015	71000
	ρ _{ix}	(0.993	0.995	0.994	0.993	0.992	0.994	0.993	0.992	0.991	0.994	0.993	0.992	0.991	0.994	0.992	0.992	0.991	0.994	0.993	0.992	0.991	0.994	0.993	
l	- ×	(10°mm ⁴	9.795	593.8	545.4	509.0	474.7	505.2	469.0	434.8	418.4	465.4	429.5	395.4	379.1	397.8	363.8	347.6	331.7	369.7	335.9	319.6	303.8	302.6	273.4	2020
	No.		9/	11	78	6/	80	8	82	83	%	82	98	87	88	86	8	6	95	93	94	95	96	67	86	5

Table 5.2.b (Cont'd)

P _{lx} V _{lx}	- ^	- -^.	ρ_{ly}	>	ŗ,	٦	ρ	Vrx	ŗ	ح ا	ρη	V _y
) (__ mm ₀))	(10°mm ⁴)				(mm)	(mm)			(mm)	(mm)		
0.0016 183.2	185.6		1.013	0.0008	118.2	117.8	0.997	0.0009	100.3	101.0	1.007	0.0005
0.0017 156.9	158.9		1.012	0.0010	9.611	119.3	0.997	0.0009	9.96	97.3	1.007	0.0006
0.0018 131.7	133.1		1.0.1	0.0011	121.3	121.0	0.997	0.0010	92.3	92.9	1.007	0.0007
0.0015 115.1	116.8		1.015	0.0008	115.5	115.2	0.997	0.0008	83.2	83.8	1.007	0.0005
9.76 9100.0	6.86		1.014	0.0009	116.9	9.911	0.997	0.0009	80.0	9.08	1.007	0.0006
0.0017 80.8	81.8		1.012	0.0011	118.6	118.2	0.997	0.000	76.3	76.8	1.007	0.0007
0.0016 111.8	113.5		1.015	0.0008	114.3	114.0	0.997	0.0009	84.4	85.0	1.007	0.0006
0.0017 94.5	95.8		1.014	0.000	115.7	115.3	0.997	0.0010	81.2	81.8	1.007	0.0006
0.0018 77.8	78.8		1.013	0.0011	117.4	117.0	0.997	0.0010	77.5	78.1	1.007	0.0007
0.0017 160.8	162.8		1.013	0.000	99.4	99.1	0.998	0.0009	9.66	100.4	1.007	0.0006
0,0018 138.4	140.0		1.012	0.0010	100.4	100.2	0.998	0.0010	96.0	7.96	1.007	0.0006
116.8	118.0		1.010	0.0012	101.7	101.5	0.998	0.0000	6.16	92.6	1.007	0.0007
0.0020 96.7	97.5		1.008	0.0014	103.4	103.2	0.998	0.0011	87.2	87.8	1.007	0.0008
0.0016 98.5	6.66		1.014	0.0008	6.96	9.96	0.997	0.0009	82.9	83.5	1.007	0.0005
0.0017 83.7	84.8		1.013	0.0010	6.76	9.76	0.997	0.0009	8.6/	80.3	1.007	0.0006
0.0018 69.4	70.3		1.012	0.0011	99.2	6.86	0.998	0.0010	76.2	7.92	1.007	0.0007
0.0020 56.3	8.99		010.1	0.0014	6.001	9.001	0.998	0.0011	72.0	72.5	1.007	0.0008
0.0020 79.3	80.5		1.014	0.0010	95.5	95.3	0.997	0.0011	81.4	82.0	1.007	0.0007
0.0021 65.2	99.1		1.013	0.0012	2.96	96.5	0.997	0.0012	77.7	78.3	1.007	0.0008
0.0023 52.2	52.8		1.0.1	0.0015	98.4	98.1	0.997	0.0013	73.4	73.9	1.007	0.0009
0.0016 87.4	88.6		1.013	0.0009	81.3	81.1	0.998	0.0009	81.2	8.18	1.007	0.0005
0.0017 74.9	75.8		1.012	0.0010	82.1	6.18	0.998	0.0009	78.2	78.7	1.007	9000.0
0.0018 62.9	63.5		010.1	0.0012	83.1	82.9	0.998	0.0000	74.7	75.2	1.007	0.0007
0.0019 51.9	52.3		1.008	0.0014	84.4	84.3	0.998	0.0000	70.7	71.2	1.007	0.0008
0.0018 84.2	1				000	0 06	9000		7 00	0	000	

Table 5.2.b (Cont'd)

٧,		0.0006	0.0007	0.000	0.0007	0.0008	0.0000	0.0006	0.0007	0.0000	0.0007	0.0008	0.0009	0.0007	0.0008	0.0009	0.0008	0.0009	0.0010	0.0008	0.0000	0.0010
ρι		1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007	1.007
٦٠	(mm)	80.1	9.9/	72.5	81.4	77.8	73.6	6.79	64.8	61.1	69.2	1.99	62.4	57.6	54.8	51.6	55.8	54.2	52.5	6.09	59.3	57.5
- >	(mm)	79.5	76.0	72.0	80.8	77.2	73.1	67.4	64.3	60.7	68.7	65.6	62.0	57.2	54.4	51.2	55.4	53.9	52.2	60.4	58.8	57.1
۷ ټ		0.0010	0.0011	0.0012	0.0011	0.0012	0.0013	0.000	0.0011	0.0012	0.0011	0.0012	0.0013	0.001	0.0012	0.0013	0.0013	0.0014	0.0015	0.0013	0.0014	0.0014
ρα		0.998	0.998	0.998	0.998	0.998	0.998	0.998	0.998	0.998	0.998	0.998	0.998	0.997	0.997	0.997	0.997	0.997	0.997	0.998	0.998	0.998
٦ -	(mm)	9.08	81.6	83.0	79.4	80.4	81.8	78.7	7.67	8I.I	77.4	78.3	7.67	77.8	78.8	80.4	77.2	6.77	78.7	61.2	61.7	62.2
r,	(mm)	80.8	81.8	83.2	9.62	90.8	81.9	78.9	6.62	81.3	77.6	78.5	79.9	78.0	79.1	9.08	77.5	78.2	79.0	61.4	8.19	62.4
>,		0.0010	0.0012	0.0015	0.0011	0.0013	0.0016	0.0010	0.0012	0.0014	0.0010	0.0012	0.0015	0.000	0.0012	0.0014	0.0012	0.0013	0.0015	0.0013	0.0014	0.0016
ρίγ			1.011	1.009	1.013	1.012	1.010	1.014	1.013	1.0.1	1.015	1.014	1.012	910:1	1.015	1.013	1.015	1.015	1.014	1.013	1.012	1.01
ا - ا	(10°mm ⁴)	72.7	9.09	49.4	69.7	57.7	46.7	46.1	37.8	30.2	44.4	36.1	28.6	29.0	23.3	18.2	22.2	19.7	17.2	25.0	22.4	8.61
-^ -^`	(10°mm ⁺)	71.8	59.9	49.0	8.89	57.0	46.2	45.5	37.3	29.9	43.7	35.7	28.3	28.5	23.0	17.9	21.9	19.4	6.91	24.6	22.1	9.61
>,		0.0019	0.0020	0.0021	0.0021	0.0023	0.0024	0.0018	0.0020	0.0021	0.0020	0.0022	0.0024	0.0019	0.0021	0.0023	0.0024	0.0025	0.0026	0.0024	0.0025	0.0026
ρ _{ix}		0.993	0.992	0.991	0.994	0.993	0.992	0.995	0.994	0.992	0.995	0.994	0.993	966'0	0.995	0.994	966'0	0.995	0.995	0.994	0.993	0.993
	10"mm")	73.6	6'89	64.8	66.4	61.7	97.6	62.0	57.3	53.2	55.5	8.09	46.7	53.0	48.2	44 . -1	42.7	40.6	38.6	25.2	24.3	23.3
No.	-	126	127	128	129	130	131	132		16	135	136	137	138	136	140	14	142	143	144	145	146

Table 5.3 Statistical Quantities, ρ_{G} and $V_{\text{G}},$ for Geometric Variations

Geometric		
Variation	$ ho_{G}$	V_{G}
G		
A	0.997	0.002
I_{κ}	0.995	0.003
$\mathbf{I}_{\mathbf{y}}$	1.004	0.008
r_x	0.999	0.002
r _y	1.004	0.004
I	1.000	0.008
r	1.001	0.004

Note: A - area of the column reinforced with welded steel plates

 I_x - moment of inertia of cross section about principal x axis

I_v - moment of inertia of cross section about principal y axis

 r_x - radius of gyration of the cross section about principal x axis

 r_y - radius of gyration of the cross section about principal y axis

I - moment of inertia of cross section

r - radius of gyration of the cross section

Table 5.4 Statistical Parameters for the Material Properties

Reference	Slenderness									
Criteria	parameter	ρ	>	$\rho_{\rm Fy}$	> ³	$\rho_{\rm E}$	∨ E	હ	ρF	>
	ィ									
SSRC	0.4	1.00.1	0.004	1.092	0.088	1.038	0.026	1.025	1.090	0.085
Curve	=:	<u>100:</u>	0.004	1.092	0.088	1.038	0.026	1.025	1.062	0.039
1	1.5	<u>-00.</u>	0.004	1.092	0.088	1.038	0.026	1.025	1.047	0.026
SSRC	9.4	<u>-00.</u>	0.004	1.092	0.088	1.038	0.026	1.025	1.088	0.080
Curve	- :	<u>.00.</u>	0.004	1.092	0.088	1.038	0.026	1.025	1.057	0.034
2	1.5	<u></u>	0.004	1.092	0.088	1.038	0.026	1.025	1.055	0.031
CSA	0.4	1.00.1	0.004	1.092	0.088	1.038	0.026	1.025	1.001	0.086
Curve	=	1.00	0.004	1.092	0.088	1.038	0.026	1.025	090:1	0.036
-	1.5	1.00	0.00	1.092	0.088	1.038	0.026	1.025	1.047	0.026
CSA	0.4	1.00.1	0.004	1.092	0.088	1.038	0.026	1.025	1.088	0.080
Curve	-	<u>1.00</u>	0.004	1.092	0.088	1.038	0.026	1.025	1.062	0.040
2	1.5	1.00	0.004	1.092	0.088	1.038	0.026	1.025	1.053	0.029

Table 5.5 Simulated Professional Factors for Columns from Group 1 ($\lambda = 0.4$)

FEA			Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0	<u> </u>	(P_{fea}/P_{rl})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	$(P_{\text{fea}}/P_{\text{rc}2})$
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
36	F	S	L/1000	0.933	0.952	1.016	0.940	0.992
37	F	S	L/1000	0.934	0.953	1.017	0.941	0.994
38	F	S	L/1000	0.936	0.955	1.019	0.943	0.995
39	F	S	L/1000	0.942	0.961	1.025	0.949	1.001
51	G	W	L/1000	0.951	0.970	1.035	0.958	1.011
52	G	W	L/1000	0.957	0.976	1.041	0.964	1.017
53	G	W	L/1000	0.958	0.977	1.043	0.965	1.018
54	G	W	L/1000	0.962	0.982	1.048	0.970	1.023
100	F	S	L/1000	0.939	0.958	1.022	0.946	0.999
101	F	S	L/1000	0.941	0.960	1.024	0.948	1.000
102	F	S	L/1000	0.933	0.952	1.015	0.940	0.992
103	F	S	L/1000	0.939	0.958	1.022	0.946	0.999
130	F	S	L/1000	0.938	0.957	1.021	0.944	0.997
131	F	S	L/1000	0.940	0.959	1.023	0.946	0.999
132	F	S	L/1000	0.934	0.953	1.017	0.941	0.994
133	F	S	L/1000	0.939	0.958	1.022	0.946	0.998
145	G	W	L/1000	0.952	0.971	1.036	0.959	1.012
146	G	W	L/1000	0.969	0.989	1.055	0.976	1.030
147	G	W	L/1000	0.965	0.985	1.051	0.972	1.026
148	G	W	L/1000	0.968	0.987	1.053	0.975	1.029
190	F	S	L/8000	0.977	0.997	1.063	0.984	1.039
191	F	S	L/2000	0.963	0.982	1.048	0.970	1.024
192	F	S	L/1000	0.947	0.966	1.031	0.954	1.007
193	F	S	L/1000	0.954	0.973	1.038	0.961	1.014
194	F	S	L/1000	0.940	0.959	1.023	0.947	1.000
195	F	S	L/1000	0.947	0.966	1.031	0.954	1.007
209	G	W	L/8000	1.000	1.021	1.089	1.008	1.064
210	G	W	L/2000	0.981	1.001	1.067	0.988	1.043
211	G	W	L/1000	0.963	0.982	1.048	0.970	1.024
212	G	W	L/1000	0.968	0.987	1.053	0.975	1.029
213	G	W	L/1000	0.963	0.982	1.048	0.970	1.024
214 Note:	G	W	L/1000	0.968	0.987	1.053	0.975	1.029

Prv - Yield strength of reinforced column

P_{fea} - Finite element analysis after reinforcing

P_{r1} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

Table 5.5 (Cont'd)

FEA			Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0		(P_{fea}/P_{rl})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
262	F	S	L/1000	0.944	0.963	1.027	0.951	1.004
263	F	S	L/1000	0.952	0.971	1.036	0.959	1.012
264	F	S	L/1000	0.946	0.966	1.030	0.953	1.006
265	F	S	L/1000	0.952	0.971	1.036	0.959	1.012
277	G	W	L/1000	0.956	0.975	1.041	0.963	1.017
278	G	W	L/1000	0.961	0.980	1.046	0.968	1.021
279	G	W	L/1000	0.961	0.980	1.046	0.968	1.022
280	G	W	L/1000	0.964	0.984	1.049	0.971	1.025

Prv - Yield strength of reinforced column

P_{fea} - Finite element analysis after reinforcing

P_{r1} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{re1} - Capacity after reinforcing (CSA1)

P_{re2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

S - Strong axis of the rolled section

Table 5.6 Simulated Professional Factors for Columns from Group 1 ($\lambda = 1.1$)

FEA			Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0	•	(P_{fea}/P_{r1})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
40	F	S	L/8000	0.682	1.003	1.265	1.033	1.266
41	F	S	L/2000	0.630	0.926	1.168	0.954	1.169
42	F	S	L/1400	0.609	0.895	1.129	0.922	1.130
43	F	S	L/1400	0.618	0.908	1.146	0.935	1.147
44	F	S	L/1400	0.626	0.920	1.161	0.948	1.162
45	F	S	L/1400	0.566	0.832	1.049	0.856	1.050
46	F	S	L/1400	0.585	0.860	1.086	0.886	1.086
55	G	W	L/8000	0.707	1.040	1.312	1.071	1.313

Note: δ_0 - Initial imperfection

P_{rv} - Yield strength of reinforced column

P_{fea} - Finite element analysis after reinforcing

 P_{r1} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{re1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

Table 5.6 (Cont'd)

FEA			Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0	·	(P_{fea}/P_{rl})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
56	G	W	L/2000	0.636	0.935	1.180	0.963	1.181
57	G	\mathbf{w}	L/1100	0.588	0.865	1.091	0.890	1.092
58	G	W	L/1100	0.599	0.880	1.111	0.907	1.112
59	G	W	L/1100	0.611	0.898	1.133	0.925	1.134
60	G	W	L/1150	0.609	0.896	1.130	0.922	1.131
61	G	W	L/1150	0.631	0.927	1.170	0.955	1.171
104	F	S	L/8000	0.687	1.011	1.275	1.041	1.276
105	F	S	L/2000	0.639	0.939	1.185	0.967	1.185
106	F	S	L/1350	0.614	0.902	1.138	0.929	1.139
107	F	S	L/1350	0.624	0.917	1.157	0.944	1.158
108	F	S	L/1350	0.630	0.926	1.169	0.954	1.170
109	F	S	L/1400	0.566	0.833	1.051	0.857	1.051
110	F	S	L/1400	0.589	0.865	1.092	0.891	1.093
134	F	S	L/8000	0.683	1.004	1.266	1.034	1.267
135	F	S	L/2000	0.633	0.931	1.174	0.958	1.175
136	F	S	L/1350	0.609	0.896	1.130	0.922	1.131
137	F	S	L/1350	0.619	0.910	1.148	0.937	1.149
138	F	S	L/1350	0.627	0.922	1.163	0.950	1.164
139	F	S	L/1400	0.559	0.822	1.037	0.847	1.038
140	F	S	L/1400	0.582	0.856	1.080	0.882	1.081
149	G	W	L/8000	0.723	1.062	1.340	1.094	1.341
150	G	W	L/2000	0.656	0.965	1.217	0.993	1.218
151	G	W	L/1000	0.601	0.884	1.115	0.910	1.116
152	G	W	L/1000	0.612	0.899	1.135	0.926	1.136
153	G	W	L/1000	0.621	0.913	1.152	0.940	1.152
154	G	W	L/1100	0.620	0.912	1.151	0.939	1.151
155	G	W	L/1100	0.641	0.942	1.188	0.970	1.189
196	F	S	L/8000	0.713	1.049	1.323	1.080	1.324
197	F	S	L/2000	0.670	0.985	1.243	1.014	1.244
198	F	S	L/1000	0.629	0.924	1.166	0.952	1.167
199	F	S	L/1000	0.646	0.950	1.199	0.979	1.200
200	F	S	L/1000	0.643	0.946	1.193	0.974	1.194

Pry - Yield strength of reinforced column

P_{fea} - Finite element analysis after reinforcing

P_{r1} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

Table 5.6 (Cont'd)

FEA			Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P _{fea} /P _{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0	•	(P_{fea}/P_{rl})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
201	F	S	L/1000	0.590	0.868	1.095	0.894	1.096
202	F	S	L/1000	0.611	0.899	1.134	0.926	1.135
215	G	W	L/8000	0.723	1.063	1.341	1.094	1.342
216	G	W	L/2000	0.656	0.965	1.218	0.994	1.219
217	G	W	L/1000	0.603	0.886	1.118	0.913	1.119
218	G	W	L/1000	0.615	0.904	1.141	0.931	1.142
219	G	W	L/1000	0.616	0.906	1.143	0.933	1.144
220	G	W	L/1000	0.627	0.922	1.164	0.950	1.165
221	G	W	L/1000	0.650	0.956	1.206	0.985	1.207
266	F	S	L/8000	0.718	1.055	1.332	1.087	1.333
267	F	S	L/2000	0.668	0.983	1.240	1.012	1.241
268	F	S	L/1000	0.627	0.922	1.163	0.949	1.164
269	F	S	L/1000	0.647	0.951	1.200	0.979	1.201
270	F	S	L/1000	0.648	0.953	1.202	0.981	1.203
271	F	S	L/1000	0.603	0.886	1.118	0.912	1.119
272	F	S	L/1000	0.618	0.909	1.147	0.936	1.148
281	G	W	L/8000	0.718	1.056	1.332	1.088	1.333
282	G	W	L/2000	0.646	0.951	1.199	0.979	1.200
283	G	W	L/1000	0.588	0.865	1.091	0.891	1.092
284	G	W	L/1000	0.601	0.884	1.116	0.911	1.117
285	G	W	L/1000	0.608	0.894	1.128	0.921	1.129
286	G	W	L/1000	0.619	0.911	1.149	0.938	1.150
287	G	W	L/1000	0.644	0.947	1.194	0.975	1.195

P_{ry} - Yield strength of reinforced column

P_{fea} - Finite element analysis after reinforcing

P_{rl} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

Table 5.7 Simulated Professional Factors for Columns from Group 1 ($\lambda = 1.5$)

FEA			Out-of-		SSRC 1	SSRC 2	CSA I	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0	•	(P_{fea}/P_{r1})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
47	F	S	L/1900	0.403	0.989	1.145	0.969	1.125
48	F	S	L/1900	0.411	1.011	1.170	0.990	1.150
49	F	S	L/1250	0.375	0.921	1.066	0.902	1.048
50	F	S	L/1250	0.385	0.946	1.095	0.927	1.076
62	G	W	L/1350	0.384	0.944	1.092	0.924	1.074
63	G	W	L/1350	0.396	0.973	1.126	0.953	1.107
64	G	W	L/1350	0.390	0.958	1.109	0.939	1.090
65	G	W	L/1350	0.404	0.992	1.149	0.972	1.129
111	F	S	L/1300	0.394	0.967	1.119	0.947	1.100
112	F	S	L/1300	0.400	0.983	1.138	0.963	1.118
113	F	S	L/1250	0.378	0.928	1.074	0.909	1.056
114	F	S	L/1250	0.388	0.955	1.105	0.935	1.086
141	F	S	L/1300	0.391	0.961	1.113	0.942	1.094
142	F	S	L/1300	0.398	0.977	1.131	0.957	1.112
143	F	S	L/1250	0.375	0.922	1.067	0.903	1.049
144	F	S	L/1250	0.386	0.949	1.098	0.930	1.080
156	G	W	L/1400	0.398	0.978	1.131	0.958	1.112
157	G	W	L/1400	0.412	1.012	1.172	0.991	1.152
158	G	W	L/1350	0.403	0.990	1.146	0.970	1.126
159	G	W	L/1350	0.411	1.011	1.170	0.990	1.150
203	F	S	L/8000	0.490	1.203	1.392	1.178	1.369
204	F	S	L/2000	0.459	1.127	1.304	1.104	1.282
205	F	S	L/1000	0.436	1.072	1.240	1.050	1.219
206	F	S	L/1000	0.441	1.083	1.253	1.061	1.232
207	F	S	L/1000	0.424	1.043	1.207	1.021	1.186
208	F	S	L/1000	0.432	1.061	1.228	1.039	1.207
222	G	W	L/8000	0.456	1.120	1.296	1.097	1.274
223	G	W	L/2000	0.414	1.017	1.177	0.996	1.157
224	G	\mathbf{W}	L/1000	0.380	0.933	1.080	0.914	1.061
225	G	W	L/1000	0.393	0.965	1.117	0.946	1.098
226	G	W	L/1000	0.396	0.973	1.127	0.953	1.107
_227	G	W	L/1000	0.409	1.005	1.163	0.984	1.143

Prv - Yield strength of reinforced column

P_{fea} - Finite element analysis after reinforcing

P_{rl} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

Table 5.7 (Cont'd)

FEA model	D	В	Out-of- Straightness	P _{fea} /P _{ry}	SSRC 1	SSRC 2	CSA 1 ρ _s	CSA 2
No.	•••••		δ_0		(P_{fea}/P_{rl})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
273	F	S	L/1000	0.444	1.091	1.263	1.069	1.241
274	F	S	L/1000	0.449	1.102	1.276	1.080	1.254
275	F	S	L/1000	0.427	1.049	1.214	1.028	1.193
276	F	S	L/1000	0.436	1.072	1.241	1.050	1.220
288	G	W	L/1000	0.378	0.930	1.076	0.911	1.058
289	G	W	L/1000	0.395	0.969	1.122	0.950	1.103
290	G	W	L/1000	0.395	0.971	1.124	0.951	1.105
291	G	W	L/1000	0.411	1.010	1.169	0.989	1.149

P_{ry} - Yield strength of reinforced column

P_{fea} - Finite element analysis after reinforcing

 P_{r1} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

Table 5.8.a Best Fit Lines for the Professional Factors for Columns from Group 1

Reference	Equation	λ	Equation for Best Fit Curve	ρ_{s}	$\overline{V_s}$
Criteria	No.	_	$\rho_{\rm s}={\rm m}\;\delta_0/{\rm L}+{\rm b}$	$\delta_0/L = 0.00067$	
SSRC	i	0.4	$\rho_{\rm s} = -44.341 \text{x} \delta_0 / \text{L} + 1.014$	0.984	0.003
curve	2	1.1	$\rho_s = -137.03x\delta_0/L + 1.0277$	0.936	0.010
1	3	1.5	$\rho_{\rm s} = -95.487 \text{x} \delta_0 / \text{L} + 1.0807$	1.017	0.006
SSRC	1	0.4	$\rho_s = -47.304 \times \delta_0 / L + 1.0818$	1.050	0.003
curve	2	1.1	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.2966$	1.181	0.010
2	3	1.5	$\rho_s = -110.51 \times \delta_0 / L + 1.2507$	1.177	0.006
CSA	1	0.4	$\rho_{\rm s} = -43.776 \text{x} \delta_0 / \text{L} + 1.0011$	0.972	0.003
curve	2	1.1	$\rho_s = -141.12x\delta_0/L + 1.0583$	0.964	0.010
1	3	1.5	$\rho_{\rm s} = -93.526 \text{x} \delta_0 / \text{L} + 1.0585$	0.996	0.006
CSA	1	0.4	$\rho_{\rm s} = -46.211 {\rm k} \delta_0 / {\rm L} + 1.0568$	1.026	0.003
curve	2	1.1	$\rho_s = -173.02x\delta_0/L + 1.2976$	1.182	0.010
2	3	1.5	$\rho_{\rm s} = -108.63 \text{x} \delta_0 / \text{L} + 1.2294$	1.157	0.006

Table 5.8.b Best Fit Lines for the Professional Factors for Columns from Group 2

Reference	Equation	λ	Equation for Best Fit Curve	$ ho_{s}$	$\overline{V_s}$
Criteria	Number		$\rho_{\rm s}={\rm m}\;\delta_0/{\rm L}+{\rm b}$	$\delta_0/L = 0.00067$	
SSRC	1	0.4	$\rho_s = -45.968 \times \delta_0 / L + 1.0217$	0.991	0.003
curve	2	1.1	$\rho_{\rm s} = -133.23 {\rm x} \delta_0 / {\rm L} + 0.9505$	0.861	0.010
1	3	1.5	$\rho_{\rm s} = -164.5 \times \delta_0 / L + 1.0558$	0.946	0.012
SSRC	1	0.4	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.057	0.003
curve	2	1.1	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.087	0.010
2	3	1.5	$\rho_s = -190.38 \times \delta_0 / L + 1.2219$	1.094	0.012
CSA	1	0.4	$\rho_{\rm s} = -45.383 \text{ x} \delta_0 / \text{L} + 1.0087$	0.978	0.003
curve	2	1.1	$\rho_s = -137.2x\delta_0/L + 0.9788$	0.887	0.010
1	3	1.5	$\rho_s = -161.12x\delta_0/L + 1.0341$	0.926	0.012
CSA	1	0.4	$\rho_{\rm s} = -47.907 \text{x} \delta_0 / \text{L} + 1.0648$	1.033	0.003
curve	2	1.1	$\rho_{\rm s} = -168.21 \text{x} \delta_0 / \text{L} + 1.2001$	1.087	0.010
2	3	1.5	$\rho_s = -187.13 \text{x} \delta_0 / \text{L} + 1.2011$	1.076	0.012

Table 5.9 Normalized Professional Factors for Columns from Group 1 (λ = 0.4)

FEA			Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_s = m \delta_0/L + b$	$ ho_{seq}$	ρ_{n}
No.			δ_0	(P_{fea}/P_{r2})	SSRC 2	·	$ ho_{s}/ ho_{seq}$
_ (1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
36	F	S	L/1000	1.016	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	0.982
37	F	S	L/1000	1.017	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	0.983
38	F	S	L/1000	1.019	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	0.985
39	F	S	L/1000	1.025	$\rho_s = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	0.991
51	G	W	L/1000	1.035	$\rho_{\rm s} = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	1.000
52	G	W	L/1000	1.041	$\rho_s = -47.3x\delta_0/L + 1.082$	1.034	1.006
53	G	W	L/1000	1.043	$\rho_{\rm s} = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	1.008
54	G	W	L/1000	1.048	$\rho_{\rm s} = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	1.013
100	F	S	L/1000	1.022	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	0.988
101	F	S	L/1000	1.024	$\rho_s = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	0.990
102	F	S	L/1000	1.015	$\rho_s = -47.3 \text{ x} \delta_0 / \text{L} + 1.082$	1.034	0.982
103	F	S	L/1000	1.022	$\rho_s = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	0.988
130	F	S	L/1000	1.021	$\rho_s = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	0.986
131	F	S	L/1000	1.023	$\rho_s = -47.3 \text{ x} \delta_0 / \text{L} + 1.082$	1.034	0.989
132	F	S	L/1000	1.017	$\rho_s = -47.3x\delta_0/L + 1.082$	1.034	0.983
133	F	S	L/1000	1.022	$\rho_s = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	0.988
145	G	W	L/1000	1.036	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	1.002
146	G	W	L/1000	1.055	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	1.020
147	G	W	L/1000	1.051	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	1.016
148	G	W	L/1000	1.053	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	1.018
190	F	S	L/8000	1.063	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.076	0.988
191	F	S	L/2000	1.048	$\rho_s = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.058	0.990
192	F	S	L/1000	1.031	$\rho_{\rm s} = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	0.996
193	F	S	L/1000	1.038	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	1.003
194	F	S	L/1000	1.023	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	0.989
195	F	S	L/1000	1.031	$\rho_{\rm s} = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	0.996
209	G	W	L/8000	1.089	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.076	1.012
210	G	W	L/2000	1.067	$\rho_s = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.058	1.009
211	G	W	L/1000	1.048	$\rho_{\rm s} = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	1.013
212	G	W	L/1000	1.053	$\rho_{\rm s} = -47.3 {\rm x} \delta_0 / {\rm L} + 1.082$	1.034	1.018
213	G	W	L/1000	1.048	$\rho_{\rm s} = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	1.013
214	G	W	L/1000	1.053	$\rho_{\rm s} = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	1.018
277 Note:	G	W	L/1000	1.041	$\rho_s = -47.3 \text{x} \delta_0 / \text{L} + 1.082$	1.034	1.006

Note: L - Column length

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges
G - Parallel to the web

S - Strong axis of the rolled section P_{r2} - Capacity after reinforcing (SSRC2)

P_{fea} - Finite element analysis after reinforcing

 ρ_{seq} - Professional ratio predicted by the best-fit equation

Table 5.9 (Cont'd)

FEA			Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_{\rm s} = {\rm m} \delta_0 / {\rm L} + {\rm b}$	$ ho_{seq}$	$\rho_{\rm n}$
No.			δ_0	(P_{fea}/P_{r2})	SSRC 2		ρ_s/ρ_{seq}
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
278	G	W	L/1000	1.046	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	1.011
279	G	W	L/1000	1.046	$\rho_s = -47.3 \text{ x} \delta_0 / \text{L} + 1.082$	1.034	1.011
280	G	W	L/1000	1.049	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	1.014
262	F	S	L/1000	1.027	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.035	0.993
26 3	F	S	L/1000	1.036	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.035	1.001
264	F	S	L/1000	1.030	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	0.996
265	F	S	L/1000	1.036	$\rho_s = -47.3 \times \delta_0 / L + 1.082$	1.034	1.002

Note: L - Column length

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section

G - Parallel to the web

P_{r2} - Capacity after reinforcing (SSRC2)

P_{fea} - Finite element analysis after reinforcing

 ρ_{seq} - Professional ratio predicted by the best-fit equation

Table 5.10 Normalized Professional Factors for Columns from Group 1 ($\lambda = 1.1$)

FEA			Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_s = m \delta_0/L + b$	$ ho_{seq}$	$\rho_{\mathfrak{n}}$
No.			δ_0	(P_{fea}/P_{r2})	SSRC 2		$\rho_{\rm s}/\rho_{\rm seq}$
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
40	F	S	L/8000	1.265	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.275	0.992
41	F	S	L/2000	1.168	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.210	0.965
42	F	S	L/1400	1.129	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.172	0.963
43	F	S	L/1400	1.146	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.172	0.977
44	F	S	L/1400	1.161	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.172	0.990
45	F	S	L/1450	1.049	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.177	0.891
46	F	S	L/1400	1.086	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.172	0.927
55	G	W	L/8000	1.312	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.275	1.029
56	G	W	L/2000	1.180	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.210	0.975
57	G	W	L/1100	1.091	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.143	0.954
58	G	W	L/1100	1.111	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.143	0.971
_ 59	G	W	L/1100	1.133	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.143	0.991

Note: L - Column length

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section

G - Parallel to the web

P_{r2} - Capacity after reinforcing (SSRC2)

P_{fea} - Finite element analysis after reinforcing

 ρ_{seq} - Professional ratio predicted by the best-fit equation

Table 5.10 (Cont'd)

FEA			Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_{\rm s} = {\rm m} \delta_0 / {\rm L} + {\rm b}$	$ ho_{seq}$	$\rho_{\mathtt{n}}$
No.			δ_0	(P_{fea}/P_{r2})	SSRC 2	·	ρ_s/ρ_{seq}
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
60	G	W	L/1150	1.130	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.148	0.984
61	G	W	L/1150	1.170	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.148	1.019
104	F	S	L/8000	1.275	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.275	1.000
105	F	S	L/2000	1.185	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.210	0.979
106	F	S	L/1350	1.138	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.168	0.975
107	F	S	L/1350	1.157	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.168	0.990
108	F	S	L/1350	1.169	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.168	1.001
109	F	S	L/1400	1.051	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.172	0.896
110	F	S	L/1400	1.092	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.172	0.932
134	F	S	L/8000	1.266	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.275	0.993
135	F	S	L/2000	1.174	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.210	0.970
136	F	S	L/1350	1.130	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.167	0.968
137	F	S	L/1350	1.148	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.169	0.982
138	F	S	L/1350	1.163	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.169	0.995
139	F	S	L/1400	1.037	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.172	0.885
140	F	S	L/1400	1.080	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.172	0.922
149	G	W	L/8000	1.340	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.275	1.051
150	G	W	L/2000	1.217	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.210	1.006
151	G	W	L/1000	1.115	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.128	0.989
152	G	W	L/1000	1.135	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.128	1.006
153	G	W	L/1000	1.152	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.128	1.021
154	G	W	L/1100	1.151	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.136	1.012
155	G	W	L/1100	1.188	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.136	1.046
196	F	S	L/8000	1.323	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.275	1.038
197	F	S	L/2000	1.243	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.210	1.027
198	F	S	L/1000	1.166	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.124	1.038
199	F	S	L/1000	1.199	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.124	1.067
200	F	S	L/1000	1.193	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.124	1.062
201	F	S	L/1000	1.095	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	0.974
202	F	S	L/1000	1.134	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.124	1.009
215	G	W	L/8000	1.341	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.275	1.051
216	G	W	L/2000	1.218	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.210	1.006
217 Notes	G	W	L/1000	1.118	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	0.995

B - Buckling axis

- D Orientation of reinforcing plate
 - F Parallel to the flanges
 - G Parallel to the web
- W Weak axis of the rolled section
- S Strong axis of the rolled section
- P_{r2} Capacity after reinforcing (SSRC2)
- P_{fea} Finite element analysis after reinforcing
- ρ_{seq} Professional ratio predicted by the best-fit equation

Table 5.10 (Cont'd)

FEA		-	Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_s = m \delta_0/L + b$	$ ho_{seq}$	$ ho_{ m n}$
No.			δ_0	(P_{fea}/P_{r2})	SSRC 2		ρ_{s}/ρ_{seq}
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
218	G	W	L/1000	1.141	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.124	1.015
219	G	W	L/1000	1.143	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.124	1.017
220	G	W	L/1000	1.164	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.124	1.036
221	G	W	L/1000	1.206	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.124	1.073
281	G	W	L/8000	1.332	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.275	1.045
282	G	W	L/2000	1.199	$\rho_{\rm s} = -172.88 \text{x} \delta_0 / \text{L} + 1.30$	1.210	0.991
283	G	W	L/1000	1.091	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	0.971
284	G	W	L/1000	1.116	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.124	0.993
285	G	W	L/1000	1.128	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	1.004
286	G	W	L/1000	1.149	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.124	1.023
287	G	W	L/1000	1.194	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	1.063
266	F	S	L/8000	1.332	$\rho_{\rm s} = -172.88 \times \delta_0 / L + 1.30$	1.276	1.044
267	F	S	L/2000	1.240	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.214	1.022
268	F	S	L/1000	1.163	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	1.035
269	F	S	L/1000	1.200	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	1.068
270	F	S	L/1000	1.202	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	1.070
271	F	S	L/1000	1.118	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	0.995
272	F	S	L/1000	1.147	$\rho_s = -172.88 \times \delta_0 / L + 1.30$	1.124	1.021

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section

G - Parallel to the web

P_{r2} - Capacity after reinforcing (SSRC2)

P_{fea} - Finite element analysis after reinforcing

 ρ_{seq} - Professional ratio predicted by the best-fit equation

Table 5.11 Normalized Professional Factors for Columns from Group 1 ($\lambda = 1.5$)

FEA			Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_{\rm s} = {\rm m} \delta_0 / {\rm L} + {\rm b}$	$ ho_{seq}$	$\rho_{\mathtt{n}}$
No.			δ_0	(P_{fea}/P_{r2})	SSRC 2	•	$\rho_{\rm s}/\rho_{\rm seq}$
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
47	F	S	L/1900	1.145	$\rho_{\rm s} = -110.51 {\rm k} \delta_0 / {\rm L} + 1.25$	1.193	0.960
48	F	S	L/1900	1.170	$\rho_s = -110.51x\delta_0/L + 1.25$	1.193	0.981
49	F	S	L/1250	1.066	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.163	0.917
50	F	S	L/1250	1.095	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.163	0.942
62	G	W	L/1350	1.092	$\rho_s = -110.51x\delta_0/L + 1.25$	1.169	0.934
63	G	W	L/1350	1.126	$\rho_{\rm s} = -110.51 {\rm k} \delta_0 / {\rm L} + 1.25$	1.169	0.964
64	G	W	L/1350	1.109	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.168	0.950
65	G	W	L/1350	1.149	$\rho_s = -110.51x\delta_0/L + 1.25$	1.168	0.983
111	F	S	L/1300	1.119	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.164	0.961
112	F	S	L/1300	1.138	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.164	0.977
113	F	S	L/1300	1.074	$\rho_{\rm s} = -110.51 \times \delta_0 / L + 1.25$	1.163	0.923
114	F	S	L/1300	1.105	$\rho_{\rm s} = -110.51 {\rm k} \delta_0 / {\rm L} + 1.25$	1.163	0.950
141	F	S	L/1300	1.113	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.164	0.956
142	F	S	L/1300	1.131	$\rho_s = -110.51x\delta_0/L + 1.25$	1.164	0.972
143	F	S	L/1300	1.067	$\rho_{\rm s} = -110.51 {\rm k} \delta_0 / {\rm L} + 1.25$	1.163	0.918
144	F	S	L/1300	1.098	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.163	0.945
156	G	W	L/1400	1.131	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.171	0.966
157	G	W	L/1400	1.172	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.171	1.000
158	G	W	L/1350	1.146	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.170	0.979
159	G	W	L/1350	1.170	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.170	1.000
203	F	S	L/8000	1.392	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.237	1.126
204	F	S	L/2000	1.304	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.195	1.091
205	F	S	L/1000	1.240	$\rho_{\rm s} = -110.51 {\rm k} \delta_0 / {\rm L} + 1.25$	1.140	1.088
206	F	S	L/1000	1.253	$\rho_{\rm s} = -110.51 {\rm k} \delta_0 / {\rm L} + 1.25$	1.140	1.099
207	F	S	L/1000	1.207	$\rho_{\rm s} = -110.51 {\rm k} \delta_0 / {\rm L} + 1.25$	1.140	1.058
208	F	S	L/1000	1.228	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.140	1.077
222	G	W	L/8000	1.296	$\rho_{\rm s} = -110.51 \times \delta_0 / L + 1.25$	1.237	1.048
223	G	W	L/2000	1.177	$\rho_{\rm s} = -110.51 \times \delta_0 / L + 1.25$	1.195	0.985
224	G	W	L/1000	1.080	$\rho_{\rm s} = -110.51 \times \delta_0 / L + 1.25$	1.140	0.947
225	G	W	L/1000	1.117	$\rho_{\rm s} = -110.51 {\rm k} \delta_0 / {\rm L} + 1.25$	1.140	0.980
226	G	W	L/1000	1.127	$\rho_{\rm s} = -110.51 \times \delta_0 / L + 1.25$	1.140	0.988
227	G	W	L/1000	1.163	$\rho_{\rm s} = -110.51 \text{x} \delta_0 / \text{L} + 1.25$	1.140	1.020
288	G	W	L/1000	1.076	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.140	0.944

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges
G - Parallel to the web

S - Strong axis of the rolled section P_{r2} - Capacity after reinforcing (SSRC2)

 ρ_{seq} - Professional ratio predicted by the best-fit equation

Pfea - Finite elemetn analysis after reinforcing

Table 5.11 (Cont'd)

FEA model No.	D	В	$\begin{array}{c} \text{Out-of-}\\ \text{Straightness}\\ \delta_0 \end{array}$	$\begin{array}{c} \text{SSRC 2} \\ \rho_{\text{s}} \\ (P_{\text{fea}}/P_{\text{r2}}) \end{array}$	$\rho_s = m \delta_0 / L + b$ SSRC 2	$ ho_{\text{seq}}$	SSRC 2 ρ_n ρ_s/ρ_{seq}
_(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
289	G	W	L/1000	1.122	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.140	0.984
290	G	W	L/1000	1.124	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.140	0.986
291	G	W	L/1000	1.169	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.140	1.025
273	F	S	L/1000	1.263	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.140	1.108
274	F	S	L/1000	1.276	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.140	1.119
275	F	S	L/1000	1.214	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.140	1.065
276	F	S	L/1000	1.241	$\rho_s = -110.51 \times \delta_0 / L + 1.25$	1.140	1.088

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section P_{r2} - Capacity after reinforcing (SSRC2)

G - Parallel to the web P_{r2} - Capa P_{fea} - Finite elemetr analysis after reinforcing

 ρ_{seq} - Professional ratio predicted by the best-fit equation

Table 5.12.a Statistical Parameters for the Professional Factors for Columns from Group 1

Reference									
стітегіа	λ	$ ho_{s}$	V_s	$\rho_{\mathfrak{a}}$	V_n	$ ho_{ex}$	V_{ex}	ρ_{p}	V_p
SSRC	0.4	0.984	0.003	1.000	0.012	1.000	0.000	0.984	0.013
curve	1.1	0.936	0.010	1.000	0.043	1.000	0.000	0.936	0.044
1	1.5	1.017	0.006	1.000	0.061	1.000	0.000	1.017	0.061
SSRC	0.4	1.050	0.003	1.000	0.012	1.000	0.000	1.050	0.013
curve	1.1	1.181	0.010	1.000	0.043	1.000	0.000	1.181	0.044
2	1.5	1.177	0.006	1.000	0.061	1.000	0.000	1.177	0.061
CSA	0.4	0.972	0.003	1.000	0.012	1.000	0.000	0.972	0.013
curve	1.1	0.964	0.010	1.000	0.043	1.000	0.000	0.964	0.044
1	1.5	0.996	0.006	1.000	0.061	1.000	0.000	0.996	0.061
CSA	0.4	1.026	0.003	1.000	0.012	1.000	0.000	1.026	0.013
curve	1.1	1.182	0.010	1.000	0.043	1.000	0.000	1.182	0.044
2	1.5	1.157	0.006	1.000	0.061	1.000	0.000	1.157	0.061

Table 5.12.b Statistical Parameters for the Professional Factors for Columns from Group 2

Reference									
criteria	λ	$\rho_{\rm s}$	V_s	$\rho_{\mathtt{n}}$	V _n	$ ho_{ex}$	V_{ex}	$ ho_{ m p}$	V_p
SSRC	0.4	0.991	0.003	1.000	0.023	1.000	0.000	0.991	0.023
curve	1.1	0.861	0.010	1.000	0.057	1.000	0.000	0.861	0.057
1	1.5	0.946	0.012	0.997	0.043	1.000	0.000	0.942	0.045
SSRC	0.4	1.057	0.003	1.000	0.023	1.000	0.000	1.057	0.023
curve	1.1	1.087	0.010	1.000	0.057	1.000	0.000	1.087	0.057
2	1.5	1.094	0.012	1.000	0.043	1.000	0.000	1.094	0.045
CSA	0.4	0.978	0.003	1.000	0.023	1.000	0.000	0.978	0.023
curve	1.1	0.887	0.010	1.000	0.057	1.000	0.000	0.887	0.057
1	1.5	0.926	0.012	1.000	0.043	1.000	0.000	0.926	0.045
CSA	0.4	1.033	0.003	1.000	0.023	1.000	0.000	1.033	0.023
curve	1.1	1.087	0.010	1.000	0.057	1.000	0.000	1.087	0.057
2	1.5	1.076	0.012	1.000	0.043	1.000	0.000	1.076	0.045

Table 5.13 Professional Factors for Unreinforced Columns

Section	Slenderness parameter λ	Initial Imperfection δ_0	P _{fea} /P _{uy}	P _{ex} /P _{uy}	$\rho_{\rm ex} = P_{\rm fea}/P_{\rm ex}$
(1)	(2)	(3)	(4)	(5)	(6)
W200x46*	0.92	L/6308	0.73	0.75	0.973
W200x46*	1.22	L/2417	0.83	0.82	1.011
W310x74**	0.925	L/1500	0.67	0.76	0.879
W310x74**	0.925	L/8000	0.75	0.76	0.980
W200x36**	0.955	L/1500	0.65	0.73	0.892
W200x36**	0.955	L/6000	0.72	0.73	0.979
W150x22**	0.992	L/1500	0.70	0.73	0.956
W150x22**	0.992	L/5000	0.73	0.73	1.000

Note:

^{*} Test results reported by Huber and Beedle (1954)

^{**} Test results reported by Beedle and Tall (1960)

 $[\]delta_0$ - Initial imperfection of the rolled section columns

L - Column length

P_{fea} - Finite element analyzed critical load capacity of the rolled section column

P_{uv} - Yielding strength of the rolled section column

Pex - Experimetal strength of the rolled section column

 $[\]rho_{ex}$ - Experimetal ratio

pcre-apver 0.90 0.93 0.95 90: <u>+</u> 0.92 0.93 0.93 $\rho_G \rho_F \rho_P = \{V_G^2 + V_F^2 + V_P^2\}^{1/2}$ 0.085 990'0 0.058 0.055 0.069 0.066 0.057 0.059 0.081 0.087 Table 5.14.a Resistance Factors for Reinforced Columns from Group 1 0.081 1.070 1.245 1.238 1.019 1.139 .040 .252 0.991 1.113 0.061 0.044 0.044 0.013 0.044 0.044 Coefficient of Variation 0.013 0.013 0.061 0.061 0.061 0.026 0.085 0.039 0.034 0.086 0.036 0.026 0.040 0.080 0.029 0.031 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 $\mathbf{V}_{\mathbf{G}}$ 0.984 0.936 1.017 1.050 0.964 1.026 1.177 0.972 0.996 1.182 1.181 .157 Mean/Nominal Ratio 1.062 1.047 060. .088 1.057 .055 1.060 .047 880. .062 .053 160 PG=PA 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.4 Reference criteria SSRC SSRC curve curve curve curve CSA CSA

ρ_{εν}ε-αβνετ 0.81 0.90 0.92 0.84 0.89 $[V_G^2 + V_F^2 + V_P^2]^{1/2}$ 0.088 0.069 Table 5.14.b Resistance Factors for Reinforced Columns from Group 2 0.052 0.084 0.067 0.055 0.0890.068 0.052 0.070 0.084 ραρισρ 0.912 1.146 0.984 1.147 1.065 0.937 0.967 1.120 1.152 .130 1.077 1.151 0.045 0.023 0.057 0.023 0.057 0.045 0.045 Coefficient of Variation 0.023 0.057 0.023 0.057 0.045 > 0.039 0.085 0.026 0.080 0.034 0.086 0.036 0.026 0.040 0.031 0.0800.029 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.002 0.942 0.978 1.057 1.087 1.094 0.887 0.926 1.033 1.087 1.076 0.991 0.861 Mean/Nominal Ratio 060 .062 .047 .088 1.057 .055 .086 1.088 1.062 780. .053 160: $\rho_G \!\!=\!\! \rho_A$ 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.997 0.4 0.4 Reference criteria SSRC SSRC curve curve curve curve CSA CSA

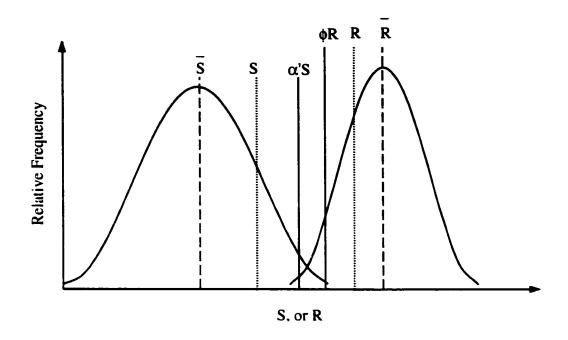


Figure 5.1 Frequency Distributions for Load Effect, S, and Resistance, R

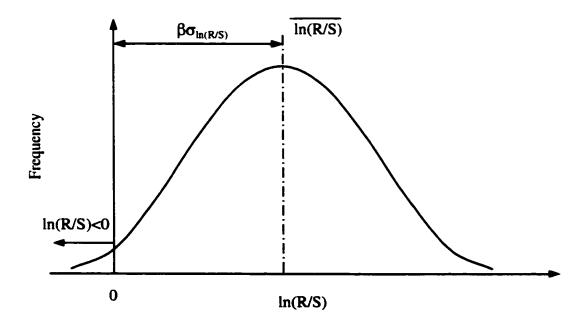
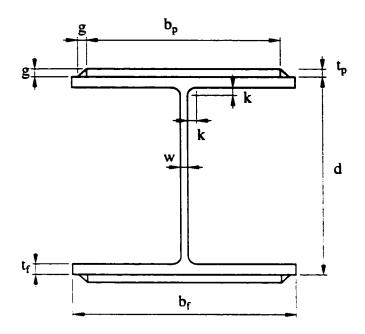
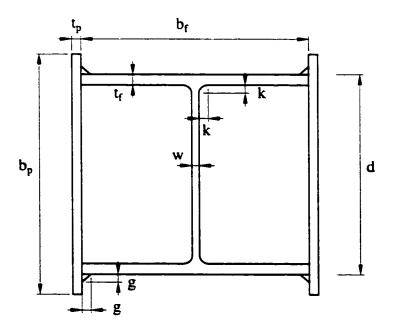


Figure 5.2 Frequency Distribution for ln(R/S)



a) Column Reinforced with Plates Parallel to the Flanges



b) Column Reinforced with Plates Parallel to the Web

Figure 5.3 Geometric Dimensions for Reinforced Columns

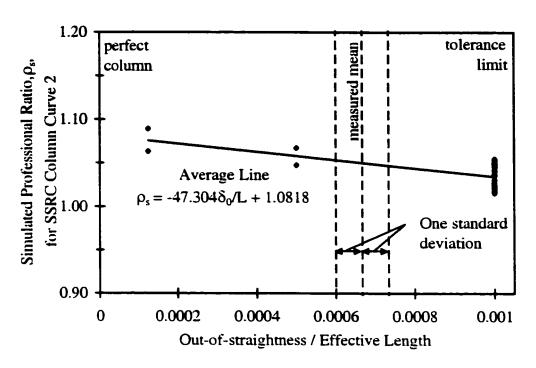


Figure 5.4 Simulated Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 1 ($\lambda = 0.4$)

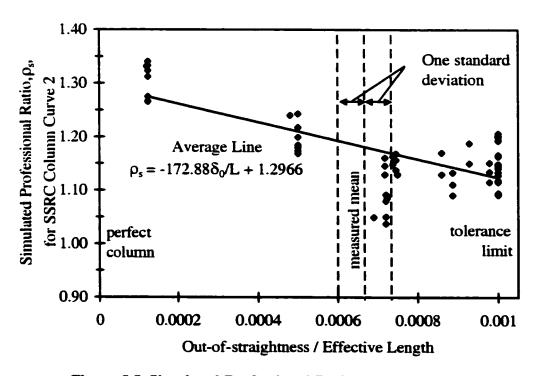


Figure 5.5 Simulated Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 1 $(\lambda = 1.1)$

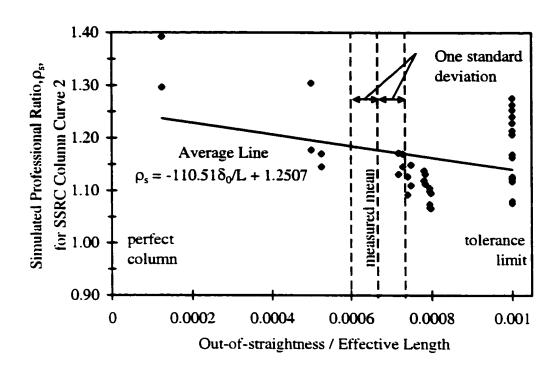


Figure 5.6 Simulated Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 1 ($\lambda = 1.5$)

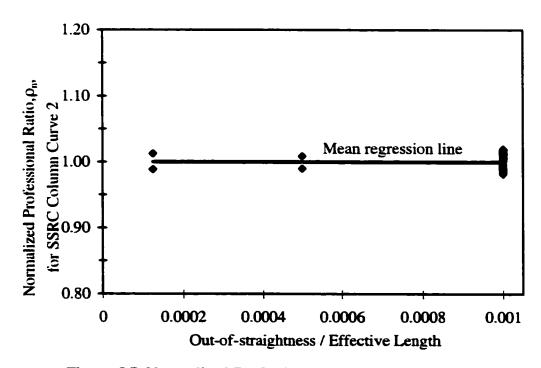


Figure 5.7 Normalized Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 1 (λ =0.4)

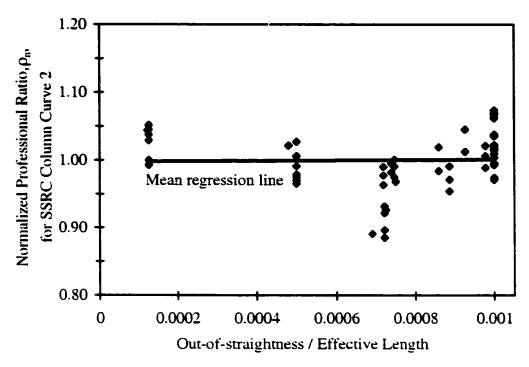


Figure 5.8 Normalized Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 1 (λ = 1.1)

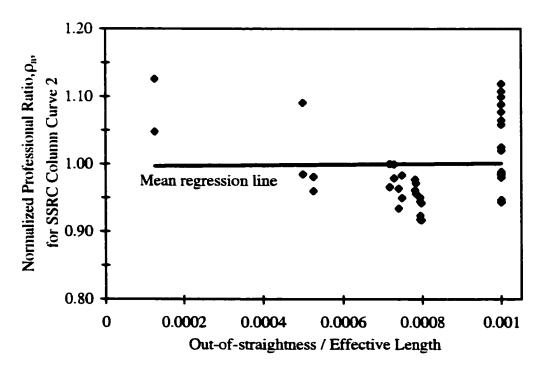


Figure 5.9 Normalized Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 1 ($\lambda = 1.5$)

Chapter 6

Summary, Conclusions and Recommendations

6.1 Summary

A study of loaded steel wide flange columns reinforced with welded steel plates has been presented in this thesis. A review of the literature has indicated that there is no specific guideline to assess the strength of steel columns reinforced with welded cover plates, although many steel columns have been reinforced in this way. With little knowledge about the effect of parameters that may affect the strength and behaviour of reinforced steel columns, reinforced columns are commonly designed using the lower column curve in \$16.1. Although this column curve gives the lowest predicted column capacity, the level of safety obtained from such a design procedure is not known. It is also possible that the residual stresses introduced by the welding process may improve the capacity of reinforced column beyond that predict by the commonly used column curve. Research on the effect of parameters and a design guideline for steel columns reinforced with welded cover plates is therefore necessary to obtain an appropriate column curve for steel columns reinforced under load with welded cover plates.

Numerous parameters may affect the strength of steel columns reinforced with welded cover plates that would not affect the strength of rolled W section columns. Welding of reinforcing plates to a wide flange section changes the compressive residual stresses normally present at the tip of the flanges to high tensile residual stresses, which may be beneficial to the strength of the reinforced column. Interaction of influencing factors, such as residual stresses, preload magnitude, orientation of reinforcing plates, buckling direction, steel grades and geometric properties, may also significantly affect the behaviour and strength of reinforced columns. To understand the behaviour and strength of the columns reinforced with welded cover plates, the effect of these influencing parameters was studied numerically.

A finite element model was developed to study the effect of varying parameters on reinforced columns. The model and analysis procedure were validated by comparing the strength and behaviour of reinforced and unreinforced columns with the results of tests on ten columns failing in various ranges of material response. A total of 317 finite element models of wide flange steel columns reinforced under load were developed to study specifically the effect of: 1) residual stress pattern and magnitude in the wide flange; 2) residual stress magnitude in the reinforcing plates; 3) the magnitude of the load on the unreinforced section at the time of the reinforcing procedure; 4) steel grade both in the wide flange section and the reinforcing plates; 5) the orientation of the reinforcing plates; 6) the buckling axis; and 7) the relative area of reinforcing plates and wide flange section. The finite element software ABAQUS was used to perform the analysis.

The load versus deflection response for the reinforced columns investigated was obtained for values of the non-dimensional slenderness parameters, λ , of 0.4, 1.1 and 1.5, which cover the range from short to close to the limit between intermediate and slender columns. Residual stresses before welding varied from 10 to 30 percent of the yield strength of the wide flange section and from 15 to 30 percent of the yield strength of the reinforcing plates. The peak welding residual stress was varied from 70 to 100 percent of the yield strength of the rolled section. The initial imperfections of the unreinforced column investigated ranged from near zero to the maximum limit of L/1000 permitted by industry standards. The preload magnitudes investigated were taken as 40 percent and 60 percent of the load carrying capacity of the reinforced columns predicted using SSRC curve 2. The orientation of reinforcing plates was parallel to the flanges or parallel to the web. Buckling about the weak axis of the wide flange section or its strong axis were both investigated. All most commonly used wide flange sections varying from W310x179 to W150x30 were investigated.

A statistical analysis of the analysis results was performed to evaluate the performance of reinforced steel columns based on the limit states philosophy. Statistical data on geometric properties, material properties, and initial imperfections for both rolled sections and plates were collected from the literature. A statistical analysis of the data was performed to obtain the appropriate magnitude of resistance factor to use with each of the two column curves used in the Canadian standard. The design criteria for steel

columns reinforced with welded cover plates were recommended based on this statistical analysis.

6.2 Conclusions

The following conclusions can be drawn based on the results of the work described above.

- 1. The slenderness parameter and out-of-straightness have the most significant effect on the strength and behaviour of reinforced steel columns.
- 2. Varying the initial residual stress pattern does not significantly affect the behaviour and strength of reinforced columns. The investigation also demonstrated that varying the maximum welding residual stress pattern from 70% to 100% of the yield strength of the materials does not significantly affect the predicted strength of reinforced columns either.
- 3. There is an interactive effect between the orientation of the reinforcing plates and the buckling direction on intermediate and slender reinforced columns. For intermediate and slender reinforced columns buckling about the weak axis of the rolled section, columns reinforced with plates parallel to the web show a higher strength than columns reinforced with plates parallel to the flanges. For intermediate reinforced columns buckling about the strong axis of the rolled section, columns reinforced with plates parallel to the flanges show a higher strength than columns reinforced with plates parallel to the web. On the other hand, for intermediate and long columns with reinforcing plates parallel to the flanges, buckling about the strong axis of the rolled section may introduce a higher strength-to-yield ratio of the reinforced column. For intermediate columns with reinforcing plates parallel to the web, buckling about the weak axis of the rolled section may introduce a higher strength-to-yield ratio of the reinforced column.

- 4. Difference in steel grades between the wide flange section and the reinforcing plates was found to have a negligible effect on the behaviour and strength of reinforced columns.
- The effect of the rolled section area to the reinforcing plate area ratio on the predicted strength-to-yield ratio of reinforced columns was found to be insignificant.
- 6. The effect of the preload magnitude varying from 40% to 60% of the load carrying capacity of the unreinforced column was found to be negligible.
- 7. SSRC curve 2 and corresponding CSA curve 2 used with a resistance factor of 0.9 are appropriate for reinforced steel columns.

6.3 Recommendations for Future Research

The work presented herein is only based on the numerical analyses. Since the lack of experimental data in the intermediate to slender range, the finite element model used for this study had only been partly validated by comparison with existing test results in the short range. Consequently, there is a need for more tests in the intermediate to slender range. The following issues should be investigated based on the test results.

- 1. Fully validating the finite element model used for this study by comparison with the test results in the intermediate to slender range.
- 2. The ratio of test strength to the computer simulation developed in this thesis.
- 3. The effect of reinforcing process on the out-of-straightness of the column, i.e., if the welding process increases the out-of-straightness of the intermediate or long column, as shown in the numerical analysis.
- 4. The local buckling shapes observed in the short columns described in Section 4.2.

5. The interactive effect between the buckling axis and the direction of the reinforced plates on the strength and behaviour of the columns reinforced with welded cover plates.

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Appendix A

Analysed Reinforced Columns Description

Analysed Reinforced Columns Description

All 317 model analyses were performed to verify the behaviour of the reinforced steel columns with welded cover plates in the analysis. The initial geometrical conditions and the material conditions of the analysed specimen are presented in Table A.1.

Column (1) of Table A.1 presents the number of the finite element analysed models. Columns (2) and (3) present the designations of rolled sections and reinforcing plates respectively.

Column (4) "D" is the direction of reinforcing plates. In this column, F represents that the cover plates are reinforced on the column along the flanges, as shown in Figure 2.1 (a). G represents that the cover plates are attached to the column at the flange tips of the column and parallel to the web, as shown in Figure 2.1 (b).

Column (5) "B" is the buckling axis of the reinforced column. W represents the same axis as the weak axis of the I-section before reinforcing, and S represents the same axis as the strong axis of the I-section before reinforcing. Column (6) presents the length of the reinforced column. This length is the same as that of the I-section. The length of the cover plates is 20 mm shorter than this length. Column (7) " λ " is the slenderness parameter of the reinforced column on the buckling direction.

Columns (8), (9) and (10) present the initial residual stress before welding in the cross section. Column (8) "PS" is the pattern of the initial residual stress before welding. The patterns are shown in the Figures 4.6. Column (9) "MF" is the maximum magnitude of the initial residual stress before welding at the flange tips in the rolled section. The values are presented by the ratio of the maximum residual stress to the yield strength of the rolled section. Column (10) "MP" is the maximum magnitude of the initial residual stress before welding at the edges of the cover plates. The values are presented by the ratio of the maximum residual stress to the yield strength of the rolled section. Column (11) presents the maximum magnitude residual stress after welding at the flange tips of the

rolled section. The values are presented by the ratio of the maximum residual stress to the yield strength of the rolled section.

Columns (12), (13) and (14) present the initial imperfection of the rolled section columns before reinforcing. The column (12) presents the ratio of the out-of-straightness to the column length, L. The out-of-straightness is on the expected buckling direction of the reinforced column. The maximum allowable initial out-of-straightness varies for columns longer than 10 m in accordance of CAN/CSA-S16.1-94. The column (13) "W" is the magnitude of the initial imperfection on the weak axis of the unreinforced column. The column (14) "S" is the initial imperfection on the strong axis of the unreinforced column.

Columns (15), (16) and (17) present the out-of-straightness of the reinforced columns after reinforcing the cover plates to the rolled section without any pre-load. The column (15) presents the ratio of the out-of-straightness to the column length, L. The out-of-straightness is on the expected buckling direction of the reinforced column. The column (16) "W" is the magnitude of the out-of-straightness on the weak axis of the unreinforced column. The column (14) "S" is the out-of-straightness on the strong axis of the unreinforced column.

Columns (18) and (19) present the yield strength of the I-section and the cover plates respectively.

Table A.1 Analysed Reinforced Column Description

FEA	-				Column			IRS		Welding		Initial Imperfection	ection	Ont-	Out-of-straightness	itness	Yield Strength	rength
mod	model I-section	Plate		æ	D* B Length λ'	~	PSe	MF	MP	Residual		before reinforcing	rcing	after re	after reinforcing, no load		=	plate
No.					L (mm)	_				Stress	ratio	IW ⁱ (mm	IW'(mm) IS ^J (mm)	ratio	IW'(mm)	IS'(mm)	(MPa)	(MPa)
\equiv	(2)	3	€	3	(4) (5) (6)	8	€	9	<u>@</u>	(E)	(12)	(13)	(14)	(15)	(16)	(17)	- 1	(61)
-	W200x46 180x9.52 F	180x9.52	2 F	≥	2440	0.5	2-1	0.3F _v	0.15F _v	0.7F _v	175200	0.47	0.00	L/4900	0.50		260	260
2	W200x46 180x9.52 F	180x9.52	고	≩	2440	0.5	2-1	$0.3F_{v}$	0.15F	$0.7F_{v}$	L/5000	0.49	0.00	L/4900	0.50		260	260
€.	W200x46 180x9.52 F	180x9.52	7 F	≥	4504	1.0	Ξ	$0.3F_{v}$	0.15F,		71000	4.51	0.00	17890	5.04		260	260
4	W200x46 180x9.52	180x9.52	<u>ح</u>	≥	4504	0:	1-3	$0.3F_{v}$	$0.3F_{v}$	ت	71000	4.51	0.00	L/890	5.04		260	260
~	W200x46 180x9.52	180x9.52	<u>ح</u>	≥	4504	0:1	1-2	0.1F,	0.15F _v	بر	2000	4.50	0.00	17890	5.04		260	260
9	W200x46 180x9.52	180x9.52	<u> </u>	≩	4504	1.0	4	0.1F,	$0.3F_{v}$	ت	2000	4.50	0.00	L/890	5.04		260	260
7	W200x46 180x9.52	180x9.52	<u>ج</u>	≥	4504	0:1	2-1	0.3F _v	0.15F,	بر	7000	4.50	0.00	L/890	5.04		260	260
∞	W200x46 180x9.52	180x9.52	<u>ر</u>	≥	4504	0:1	2-2	0.1F,	0.15F	تا	71000	4.50	0.00	L/890	5.04		260	260
6	W200x46 180x9.52	180x9.52	E.	≩	4504	1.0	3-1	0.3F _v	0.15F,	بر س	71000	4.50	0.00	L/890	5.04		260	260
으 153	W200x46 180x9.52	180x9.52	<u>.</u>	≥	4504	0:1	3-2	0.1F _v	0.15F,	, tr ₃	71000	4.50	0.00	L/890	5.04		260	260
; = 3	W200x46 180x9.52	180x9.52	E.	≩	4504	0:1	4-1	0.3F _v	0.15F	, tr ₂	71000	4.50	0.00	17890	5.04		260	260
12	W200x46 180x9.52	180x9.52	元	≩	4504	0:1	4-2	0.1F,	0.15F,	, tr ₃	71000	4.50	0.00	L/890	5.04		260	260
13	W200x46 180x9.52	180x9.52	<u>ب</u>	S	8316	1.0	1-1	$0.3F_{v}$	0.15F,	س	71000	99:1	8.32	L/850		9.83	260	260
14	W200x46 180x9.52	180x9.52	<u></u>	S	8316	1.0	1-2	0.1F _v	0.15F	, п,	1000	99:1	8.32	L/850		9.83	260	260
15	W200x46 180x9.52	180x9.52	다	S	8316	1.0	3-1	$0.3F_{v}$	0.15F,	. п _у	71000	99.1	8.32	L/850		9.83	260	260
91	W200x46 180x9.52 F	180x9.52	T.	S	8316	0.1	3-2	0.1F,	0.15F,	. ፔ,	71000	99:1	8.32	L/850		9.83	260	260
17	W310x179 290x25	290x25	Ľ	≩	2631	0.4	1-3	$0.3F_{y}$	$0.3F_{\chi}$	F,	L/8000	0.33	0.00	L/850	0.34		300	300
a D	a) D - Direction of reinforcing plates	of reinfore	cing	[클	ıtes			F - Par	allel to	F - Parallel to the flanges	es		G - Para	G - Parallel to the web	e web			
b) B	b) B - Buckling axis of the reinforced column	xis of the	rein	. <u>J</u>	ced colu	E		W - W	eak axi	W - Weak axis of the I-section	-section		S - Stroi	ng axis o	S - Strong axis of the I-section	tion		
1	D		;	1	:						1	•		•		•		

f) MF -Maximum magnitude of the residual stress in the flange. d) IRS - Initial residual stress before reinforcing

> g) MP -Maximum magnitude of the residual stress in the reinforcing plate. e) PS - Residual stress pattern, as illustrated in Figure 4-1.

c) λ - Slenderness parameter of the reinforced column

Fy - Yield stress of the unreinforced column

h) ratio - The ratio of the out-of-straightness to the column length, L.

i) W - Out-of-straightness in the weak direction.

FEA	A				Column			IRS		Welding	Initia	Initial Imperfection	ction	Out	Out-of-straightness	'	Yield Strength	ength
mod	model 1-section	Plate Da B Length	<u>_</u>	1 8	ength	۳	PSe	MF	MPg	Residual	befo	before reinforcing	cing	after rei	after reinforcing, no load		I-section plate	plate
Ž	ć			_1	L (mm)					Stress	ratio	W ^h (mm) S ⁱ (mm	S'(mm)	ratio	Wh(mm) Si(mm)	(III	(MPa)	(MPa)
=) (2)	(3)	(4) (5)	(5)	9	6	€	6	(OE)	(E)	(12)	(13)	(14)		(16) (17)	2	(18)	(61)
=	W3	290x25	ഥ	≥	2631	0.4	1-3	0.3F,	0.3Fy	π,	172000	1.32	0.00	1/1930	1.36		300	300
51	9 W310x179 290x25	290x25	Œ	≱	2631	0.4	<u>1-3</u>	0.3F,	$0.3F_{v}$. ፑ,	71000	2.64	0.00	L/970	2.72		300	300
30	W310x179 290x25	290x25	[工	≩	2631	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$. ፔ,	71000	2.63	0.00	L/980	2.69		300	300
21		290x25	Ľ,	≥	2740	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$	Г.	71000	2.75	0.00	02677	2.83		230	350
22		290x25	Œ	≥	2740	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$. ፑ,	71000	2.75	0.00	17980	2.80		230	350
23		290x25	12.	≩	7235	1.1	1-3	$0.3F_{v}$	$0.3F_{v}$. ፫	178000	0.83	0.00	17190	10:1		300	300
24		290x25	12	≱	7235	-	1-3	$0.3F_{v}$	$0.3F_{v}$. ፑ,	L/2000	3.34	0.00	1790	4.06		300	300
25		290x25	Ľ.	≥	7235	=	1-3		$0.3F_{y}$. ፑ,	71000	7.23	0.00	L/820	8.82		300	300
		290x25	1	≥	7235	=	1-3		$0.3F_{v}$	0.7F,	71000	7.25	0.00	L/820	8.86		300	300
52 154		290x25	Ľ,	≱	7235	Ξ:	1-3	_	$0.3\overline{F_{y}}$	Т,	71000	7.25	0.00	1 200	8.09		300	300
		290x25	Œ	≥	7535	Ξ	1-3	_	$0.3F_{y}$. π <u>,</u>	71000	7.54	0.00	L/830	9.03		230	320
50		290x25	Œ,	≥	7535	==	1-3		$0.3F_{y}$	Пу	71000	7.54	0.00	1	8.34		230	350
8		290x25	Œ,	≱	9986	1.5	1-3	0.3F,	$0.3F_{y}$	E,	178000	1.24	0.00	L/6270	1.57		300	300
31	W310x179 290x25	290x25	Ľ.	≩	9986	1.5	I-3	0.3F	$0.3F_{y}$	Œ,	L/2000	4.94	0.00	1/1570	6.28		300	300
32	2 W310x179 290x25	290x25	Œ	≥	9986	1.5	<u>1-3</u>	0.3F _y	$0.3F_{y}$	ፎ	21000	9.88	0.00	738	12.56		300	300
33	3 W310x179 290x25	290x25	<u> </u>	≩	9986	1.5	<u>-3</u>	0.3F _y	$0.3F_{y}$	ቪ	71000	9.88	0.00	7890	11.13		300	300
8	4 W310x179 290x25	290x25	<u> </u>	≯	10275	1.5	1-3	$0.3F_{y}$	$0.3F_{\chi}$	F,	171000	10.00	0.00	L/830	12.46		230	350
) (a	a) D - Direction of reinforcing plates	f reinforc	ing	plate	Se		_	F - Para	illel to	F - Parallel to the flanges	S		G - Para	G - Parallel to the web	web			
P) H	b) B - Buckling axis	cis	1					W - We	sak axi	W - Weak axis of the I-section	section		S - Stroi	ig axis of	S - Strong axis of the I-section			
	,		•			•					201.1		- then	Lafer.	S. mine Carrier			

d) IRS - Initial residual stress before reinforcing

c) λ - Slenderness parameter of the reinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate. e) PS - Residual stress pattern, as illustrated in Figure 4-1.

f) MF -Maximum magnitude of the residual stress in the flange. $F_y - Yield \ stress \ of \ the \ unreinforced \ column$

h) ratio - The ratio of the out-of-straightness to the column length, L. i) W - Out-of-straightness in the weak direction.

j) S - Out-of-straightness in the strong direction.

FEA		Column		ပ	Column			IRS		Welding		Initial Imperfection	ection) O	Out-of-straightness	itness	Yield Strength	rength
300	model I-section	Plate	<u>"</u>	3 P	D* B' Length	ر م	PS¢	MF	ΜPg	Residual	befo	before reinforcing	rcing	after rei	after reinforcing, no load	no load	I-section plate	plate
Š	o.				L (mm)		,			Stress	ratio	W ^h (mm)		ratio	W ^h (mm)	S ⁱ (mm)	(MPa)	(MPa)
=	(2)	3	(4) (5)	(5)	9	6	€	<u>6</u>	(<u>0</u>	Ξ	:	(13)	(14)	;	(16)	(11)	(18)	(19)
le.	5 W310x179 290x25	290x25	Ľ	_ ≥	10275	1.5	1-3	0.3F _v	0.3F,	F,	71000	10.00	0.00	17920	11.18		230	350
36	6 W310x179 290x25	290x25	Œ	S	5064	0.4	1-3	_	$0.3F_{y}$	Г,	71000	1.02	90.9	17950		5.33	300	300
3,	7 W310x179 290x25	290x25	Έ,	S	5064	0.4	1-3	_	0.3F,	π <u>,</u>	71000	1.02	90.9	7960		5.25	300	300
**	8 W310x179 290x25	290x25	Ľ.	S	5273	0.4	1-3	0.3F,	$0.3F_{v}$, п <u>т</u> ,	71000	1.08	5.27	096/1		5.52	230	350
39	9 W310x179 290x25	290x25	Œ	S	5273	0.4	1-3		$0.3F_{v}$. г,	71000	1.08	5.27	0/6/7		5.43	230	350
4	0 W310x179 290x25	290x25	<u> </u>	S	3923		1-3	_	$0.3F_{v}$. т ,	78000	0.35	1.74	L/6050		2.30	300	300
4	1 W310x179 290x25	290x25	12.	S	3923	=	1-3	$0.3F_{v}$	$0.3F_{v}$. г,	172000	1.39	96.9	L1510		9.22	300	300
42	2 W310x179 290x25	290x25	ഥ	S	3923	-:	1-3	_	$0.3F_{v}$. ኬ	71400	1.99	10.00	L/1050		13.25	300	300
4	3 W310x179 290x25	290x25	ĹŢ.	S	3923	=	1-3		0.3F,	$0.7F_{y}$	171400	1.99	10.00	171051		13.27	300	300
₹ 155	4 W310x179 290x25	290x25	ĬŽ.	S	13923	-:	1-3	_	0.3F,	Γ,	L/1400	1.99	10.00	71130		11.71	300	300
4.	5 W310x179 290x25	290x25	Ľ.	S	14500	=	1-3		0.3F,	. π _.	L/1450	1.96	10.00	171070		13.57	230	350
4	6 W310x179 290x25	290x25	<u>[]</u>	S	14500	=	1-3	0.3F,	$0.3F_{y}$	ቪ	71400	2.06	10.49	L/1200		12.15	230	350
74	7 W310x179 290x25	290x25	Ľ,	S	98681	1.5	1-3	_	0.3F,	. π _y	71900	1.99	9.99	1/1320		14.42	300	300
48	8 W310x179 290x25	290x25	IT.	S	98681	1.5	1-3	$0.3F_{y}$	$0.3F_{y}$	π'n	71900	1.99	6.66	L/1560		12.14	300	300
49	9 W310x179 290x25	290x25	Œ	S	19772	1.5	1-3	0.3F,	0.3F	Ē,	1/1250	3.06	15.77	1		22.09	230	350
2	0 W310x179 290x25	290x25	Ľ,	S	19772	1.5	1-3	$0.3F_{y}$	0.3F	ᄄ	L/1250	3.06	15.77	L/1050		18.92	230	350
51	1 W310x179 350x25	350x25	Ŋ	W	4103	0.4	1-3 ($0.3F_{y}$	F,	0001/1	4.11	0.00	L/940	4.89		300	300
a) D	a) D - Direction of reinforcing plates	of reinforc	ing	plate	S			- Para	le to	F - Parallel to the flanges	SS		G - Para	G - Parallel to the web	e web			
b) B	b) B - Buckling axis	xis		ı				¥ - We	ak axi	W - Weak axis of the I-section	section		S - Stron	ng axis of	S - Strong axis of the I-section	tion		
•			,		,													

d) IRS - Initial residual stress before reinforcing

f) MF -Maximum magnitude of the residual stress in the flange. Fy - Yield stress of the unreinforced column

> g) MP -Maximum magnitude of the residual stress in the reinforcing plate. h) ratio - The ratio of the out-of-straightness to the column length, L.

e) PS - Residual stress pattern, as illustrated in Figure 4-1. c) λ - Slenderness parameter of the reinforced column

i) W - Out-of-straightness in the weak direction.

FEA	V				Column			IRS		Welding	Initia	Initial Imperfection	ection	Ont	Out-of-straightness	ıtness	Yield Strength	rength
mod	model 1-section	Plate	<u>"</u>	8	D" B" Length	یر	PSc	ΜF	MPg	Residual	befo	before reinforcing	rcing	after rei	after reinforcing, no load	no load	I-section plate	plate
Ž	ند			_	L (mm)					Stress	ratio	W ^h (mm) S ⁱ (mm	S'(mm)		W ^h (mm)	S'(mm)	(MPa)	(MPa)
Ξ	(2)	3	(4) (5)	3	9	8	®	<u>6</u>	(000)	(E)	(12)	(13)	(14)	(15)	(16)	(11)	(18)	(61)
22	W310x179 350x25	350x25		≥	4103	0.4	1-3	0.3F _y	0.3F,	F,	71000	4.11	0.00	7900	4.58		300	300
53	3 W310x179 350x25	350x25	O	≥	4231	0.4	1-3	0.3F,	0.3F	Т _у	71000	4.23	0.00	L/870	4.89		230	320
54	W310x179 350x25	350x25	9	≱	4231	0.4	1-3	0.3F,	$0.3F_{y}$. Œ ₂	71000	4.23	0.00	016/1	4.64		230	320
55	W310x179	350x25	ŋ	≩	11281	Ξ	1-3	0.3F,	$0.3F_{y}$. Ε _χ	178000	1.41	0.00	L4530	2.49		300	300
26	5 W310x179	350x25	Ö	≩	11281	<u> </u>	1-3	$0.3F_{v}$	$0.3F_{v}$	ι τ ,	1/2000	5.64	0.00	L114	9.91		300	300
57	W310x179 350x25	350x25	ŋ	≥	11281	1.1	1-3	$0.3F_{v}$	$0.3F_{v}$. ሞ,	11150	10.00	0.00	1640	17.57		300	300
58	3 W310x179 350x25	350x25	Ö	≥	11281	1.1	I-3	$0.3F_{v}$	$0.3F_{v}$	0.7F _v	11150	10.00	0.00	1640	17.61		300	300
59		350x25	Ö	≥	11281	Ξ:	1-3	$0.3F_{v}$	$0.3F_{v}$		VIIS 0	10.00	0.00	17820	13.65		300	300
		350x25	ŋ	≥	11634	-	1-3	$0.3F_{v}$	$0.3F_{v}$, FT ₂	11150	10.00	0.00	2700	16.79		230	320
ತ 156		350x25	ŋ	≥	11634	1.1	1-3	$0.3F_{v}$	$0.3F_{v}$. ፔ	171150	10.00	0.00	1/880	13.36		230	320
	W310x179	350x25	ŋ	≥	15383	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$. Œ	1/1350	11.38	0.20	L720	21.36		300	300
63	3 W310x179	350x25	Ö	≥	15383	1.5	1-3	$0.3F_{y}$	$0.3F_{y}$. π _y	1/1350	11.38	0.20	17970	15.92		300	300
Z	W310x179	350x25	Ö	₹	15864	1.5	1-3	$0.3F_{y}$	$0.3F_{y}$. π _y	1/1350	11.86	0.21	L/720	22.03		230	320
65	W310x179	350x25	Ö	≥	15864	1.5	1-3	$0.3F_{y}$	$0.3F_{y}$	ᄄ	171350	11.86	0.21	1/970	16.53		230	320
3	5 W310x179 350x25	350x25	O	S	4030	0.4	1-3	$0.3F_{y}$	$0.3F_{y}$	Ē,	71000	0.00	4.03	L/980		4.09	300	300
29	W310x179 350x25	350x25	Ö	S	4030	0.4	1-3	$0.3F_{y}$	$0.3F_{y}$	Г.	71000	0.00	4.03	L/990		4.08	300	300
89	350x25 W310x179	350x25	g	S	4156	0.4	1-3	$0.3F_{y}$	$0.3F_{y}$	F,	71000	0.80	4.15	1/991		4.22	230	350
a D	a) D - Direction of reinforcing plates	f reinford	sing	plat	es			F - Par	allel to	F - Parallel to the flanges	S		G - Para	G - Parallel to the web	e web			
b) B	b) B - Buckling axis	cis		,				W - W	eak axi	W - Weak axis of the I-section	section		S - Stroi	ng axis o	S - Strong axis of the I-section	tion		
	, ;		•	٠		-						Indiana.	and land	The Party	To min Con			

d) IRS - Initial residual stress before reinforcing e) PS - Residual stress pattern, as illustrated in Figure 4-1. c) λ - Slenderness parameter of the reinforced column

f) MF -Maximum magnitude of the residual stress in the flange. $F_{\rm y}$ - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate.

h) ratio - The ratio of the out-of-straightness to the column length, L. i) W - Out-of-straightness in the weak direction.

FEA					Column			IRS		Welding	Initia	Initial Imperfection	ection	Ont	Out-of-straightness	itness	Yield Strength	ength
mode	model 1-section	Plate	<u>"</u>	2 2	Da Bb Length	~	PSc	MF	MP^g	Residual	pelo	before reinforcing	rcing	after re	after reinforcing, no load	no load	I-section plate	plate
So.				_	L(mm)					Stress	ratio	W ^h (mm) S'(mm	S'(mm)	ratio	W ^h (mm)	S'(mm)		(MPa)
Ξ	(2)	3	€	(4) (5)	9	0	€	6	9	(11)	(12)	(13)	(14)		(16) (17)	(17)	(18)	(61)
9	W310x179 350x25	350x25	O	S	4156	0.4	1-3	0.3F,	0.3F _y	ᄄ	71000	08.0	4.15	L/992		4.19	230	320
20	W310x179 350x25	350x25	g	S	11083	Ξ	1-3	0.3F,	$0.3F_{y}$	π,	178000	0.30	1.39	L/7240		1.53	300	300
71	W310x179 350x25	350x25	Ö	S	11083	Ξ	1-3	$0.3F_{v}$	$0.3F_{y}$. Γ.	L/2000	1.08	5.54	71810		6.12	300	300
72	W310x179 350x25	350x25	Ö	S	11083	==	1-3	$0.3F_{y}$	0.3F,	.π _{>}	71100	1.94	9.99	71000		1.04	300	300
73	W310x179	350x25	Ö	S	11083	Ξ	1-3	$0.3F_{v}$	$0.3F_{v}$	0.7F _v	71100	1.94	66.6	10017		11.06	300	300
74	W310x179	350x25	Ö	S	11083	=	1-3	$0.3F_{v}$	$0.3F_{v}$	E	71100	1.94	9.99	L/1050		10.55	300	300
75	W310x179	350x25	Ö	S	11429	Ξ	1-3	$0.3F_{v}$	$0.3F_{v}$	ت	171150	1.85	10.00	L1050		10.87	230	350
26	W310x179	350x25	Ö	S	11429	=	1-3	$0.3F_{v}$	$0.3F_{v}$	ب	171150	1.85	10.00	171050		10.47	230	350
11	W310x179	350x25	Ö	S	15113	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$. ፫	1/1350	2.20	<u> </u>	L/1200		12.64	300	300
æ 157			Ö	S	15113	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$	بت. ب	1/1350	2.20	11.11	L/1280		11.85	300	300
79	W310x179	350x25	Ö	S	15585	1.5	1-3	0.3F _v	$0.3F_{v}$	تح	1/1350	2.30	11.58	L/1200		12.99	230	350
80	W310x179	350x25	Ö	S	15585	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$	ت	1/1350	2.30	11.58	L/1270		12.27	230	350
8	W310x179	350x25	O	S	11083	Ξ	4	0.1F,	$0.3F_{v}$	гг _у	71100	1.94	9.98	71000		11.01	300	300
82	W310x179	350x25	g	S	11083	Ξ	3-3	$0.3F_{v}$	$0.3F_{v}$. π _.	21100	1.94	66.6	71000		1.04	300	300
83	W310x179 350x25	350x25	Ö	S	11083	Ξ	3-4	0.1F,	$0.3F_{y}$. π _y	21100	1.94	66.6	71000		1.04	300	300
84	W310x179 290x16	290x16	<u> </u>	≥	2617	0.4	1-3	$0.3F_{v}$	0.3F _v	Г.	71000	2.62	0.00	17980	5.66		300	300
85	W310x179 290x16	290x16	Ŧ	≽	2617	0.4	1-3	0.3F	0.3F	F,	21000	2.62	0.00	17981	2.68		300	300
a) D	a) D - Direction of reinforcing plates	reinfor	ing	plat	es			F - Par	allel to	F - Parallel to the flanges	s		G - Para	G - Parallel to the web	e web			
b) B	b) B - Buckling axis	is	ŀ	,				W - W	eak axi	W - Weak axis of the I-section	section		S - Stror	ng axis o	S - Strong axis of the I-section	tion		
c) y -	c) λ - Slendemess parameter of the reinforced column	paramet	er o	fthe	reinfor	ced	olun	E			d) IRS -	Initial re	sidual st	ress befo	d) IRS - Initial residual stress before reinforcing	cing		

j) S - Out-of-straightness in the strong direction.

f) MF -Maximum magnitude of the residual stress in the flange. $F_{\rm y}$ - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate.

c) λ - Slendemess parameter of the reinforced column e) PS - Residual stress pattern, as illustrated in Figure 4-1.

h) ratio - The ratio of the out-of-straightness to the column length, L.

i) W - Out-of-straightness in the weak direction.

FEA	A				Column			IRS		Welding	Imitig	Initial Imperfection	ection	Ont	Out-of-straightness	itness	Yield Strength	rength
mod	model 1-section	Plate	<u>_</u>	B, L	D" Bb Length	λ ^c P	PS _c	MF	MP ^g	Residual	befo	before reinforcing	rcing	after re	after reinforcing, no load	no load	I-section plate	plate
Š	<u>ن</u>				L (mm)					Stress		W ^h (mm)	S'(mm)	ratio	W ^h (mm)	S ⁱ (mm)	(MPa)	(MPa)
\equiv	(2)	3	(4) (5)	(S)	9	6	®	<u>6</u>	9	(E)	(12)	(13)	(14)	(15)		(11)	(18)	(19)
8	W310x179 290x16	290x16	ı	≥	2786	0.4	-3 0	0.3F _v (0.3F,	F,	71000	2.79	0.00	17982	2.86		230	350
87	7 W310x179 290x16	290x16	Ľ,	≩	2786	0.4	-3 0		$0.3\overline{\mathrm{F}_{\mathrm{y}}}$. ፍ,	71000	2.79	0.00	L/983	2.83		230	350
&	3 W310x179 290x16	290x16	ĬŢ,	>	7197	1.1	-3 0		0.3F,	. π _{>}	178000	0.00	0.00	0069/1	<u>3</u>		300	300
8	W310x179 290x16	290x16	Œ	≥	7197		-3 0		J.3F,	. Œ ₂	172000	3.60	0.00	L1130	4.17		300	300
S		290x16	ĹŢ.	3	7197	1.1	-3 0	_	J.3F,	. г.	71000	7.20	0.00	L/860	8.34		300	300
6	-	290x16	Ľ.	≩	7197	1.1	-3 0	_).3F,	0.7F _v	71000	7.20	0.00	17861	8.36		300	300
92	-	290x16	ĹŢ.	≥	7197	1.1	-3 0).3F,	ت	21000	7.20	0.00	L/920	7.80		300	300
93	-	290x16	1	≥	7662		-3 0	_).3F,	ب	71000	7.66	0.00	17870	8.78		230	350
		290x16	1	≩	7662		-3 0).3F,	ت	71000	99.2	0.00	L/930	8.25		230	350
ි 158		290x16	Ľ,	}	9813	1.5.1	-3 0	_).3F,	ت	71000	9.81	0.00	L/840	11.71		300	300
		290x16	Ľ,	≥	9813	1.5	-3 0	_).3F,	0.7F _v	71000	9.81	0.00	L/840	11.76		300	300
97		290x16	12.	≥	9813	1.5.1	-3 0	_).3F,	Г	71000	9.81	0.00	L/920	10.69		300	300
86		290x16	Ľ,	_ ≽	0447	1.5.1	-3 0	_).3F,	, г,	1/1050	10.00	0.00	L/890	11.77		230	320
8		290x16	Ľ.	_ ≽	10447	1.5.1	-3 0	_	0.3F,	. п.	L/1050	10.00	0.00	0/6/7	10.82		230	320
200	0 W310x179 290x16	290x16	Œ	S	4880	0.4	-3 0	_).3F,	. ኬ,	71000	0.97	4.88	1/970		5.04	300	300
10	I W310x179 290x16	290x16	Ľ.	S	4880	0.4	-3 0	_	0.3F,	Π,	71000	0.97	4.88	1 980		2.00	300	300
102	2 W310x179 290x16	290x16	H	S	9615	0.4	1-3 0	0.3F _y (0.3F,	F,	71000	1.06	5.20	1/970		5.37	230	320
a) D	a) D - Direction of reinforcing plates	reinforc	ing	plate	SS		ĭ	- Para	lel to	F - Parallel to the flanges	s		G - Parallel to the web	llet to th	e web			
b) B	b) B - Buckling axis	is					3	/ - We	ak axis	W - Weak axis of the I-section	section		S - Stron	g axis o	S - Strong axis of the I-section	tion		
٠.	,		•	٠		•								Lac.	- Trible			

d) IRS - Initial residual stress before reinforcing W - Weak axis of the 1-section

f) MF -Maximum magnitude of the residual stress in the flange. $F_{\rm y}$ - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate. h) ratio - The ratio of the out-of-straightness to the column length, L. e) PS - Residual stress pattern, as illustrated in Figure 4-1.

c) λ - Slenderness parameter of the reinforced column

i) W - Out-of-straightness in the weak direction.

Table A.1 (cont'd)

FEA	_				Column			IRS		Welding	Initia	Initial Imperfection	ection	Ont-c	Out-of-straightness	itness	Yield Strength	rength
mode	model 1-section	Plate	<u>"</u>	æ	D' B' Length λ	່.	PSc	MF	MPg	Residual	peto	before reinforcing	rcing	after rein	after reinforcing, no load	no load	I-section plate	plate
N _o				_	L (mm)					Stress	ratio	W ^h (mm)	W ^h (mm) S ⁱ (mm)	ratio	W ^h (mm)	S'(mm)	(MPa)	(MPa)
Ξ	(2)	3	£	(4) (5)	9	6	€	<u>6</u>	9	(=)	•	(13)	(14)	(15)	(16)	(11)	(18)	(61)
103	W310x179 290x16	290x16	뜨	S	9619	0.4	1-3	0.3F _y	0.3F,	ፍ	71000	1.06	5.20	17980		5.31	230	350
<u>इ</u>	W310x179 290x16	290x16	<u> </u>	S	13420	-:	1-3	$0.3F_{y}$	0.3F,	. π _y	178000	0.33	1.68	L/6500		2.07	300	300
105	W310x179 290x16	290x16	Ι,	S	13420	Ξ	1-3	$0.3F_{v}$	$0.3F_{y}$. π _y	1/2000	1.32	6.71	L/1620		8.27	300	300
106		290x16	Ι.	S	13420	-	1-3	$0.3F_{v}$	$0.3F_{y}$. ፑ,	171350	1.96	10.00	V1090		12.32	300	300
107	W310x179 290x16	290x16	Ľ.	S	13420	Ξ:	<u>-3</u>	0.3F,	0.3F	0.7F _y	11350	1.96	10.00	<u> </u>		12.34	300	300
108	-	290x16	ſŢ,	S	13420	Ξ	1-3	$0.3F_{v}$	$0.3F_{v}$	π,	1/1350	1.96	10.00	L/1200		11.23	300	300
80	W310x179 290x16	290x16	Ľ.	S	14287	1.1	<u>I-3</u>	$0.3F_{v}$	0.3F,	. ፑ,	L1400	2.02	10.29	L1150		12.49	230	350
011	W310x179 290x16	290x16	Ľ.	S	14287	=	1-3	$0.3F_{v}$	$0.3F_{v}$. 吓,	L/1400	2.02	10.29	1/1250		11.47	230	350
		290x16	Œ	S	18300	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$. ፑ,	L1300	2.78	14.29	17970		18.81	300	300
2 159		290x16	Ľ.	S	18300	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$		L/1300	2.78	14.29	71110		16.50	300	300
113	W310x179 290x16	290x16	Έ.	S	19483	1.5	1-3	$0.3F_{v}$	0.3F _v	ت	L/1250	3.01	15.48	17970		19.99	230	350
114	W310x179 290x16	290x16	ĬŢ,	S	19483	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$. ፑ,	11250	3.01	15.48	71100		17.72	230	350
115	W310x179 350x16	350x16	Ľ	≩	2827	0.4	I-3	$0.3F_{v}$	$0.3F_{y}$. ፑ,	71000	2.83	0.00	1/970	2.90		300	300
911	W310x179 350x16	350x16	Ľ,	≯	2827	0.4	<u>-3</u>	0.3F	$0.3F_{v}$. ፑ,	71000	2.83	0.00	0/6/7	2.92		300	300
117	W310x179 350x16	350x16	Ľ	≱	2982	0.4	I-3	$0.3F_{v}$	0.3F,	П,	1/1000	2.99	0.00	0/6/7	3.10		230	350
118	W310x179 350x16	350x16	Ľ,	≱	2982	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$, ιτ _,	71000	2.99	0.00	076/7	3.06		230	350
119		350x16	Ħ	*	2777	1.1	1-3	$0.3F_{y}$	$0.3F_{y}$	F,	1/8000	0.97	0.00	176300	1.23		300	300
a) D	a) D - Direction of reinforcing plates	reinfor	cing	pla	es			F - Para	lel to	F - Parallel to the flanges	S		G - Para	G - Parallel to the web	web			
b) B	b) B - Buckling axis	iis		,				W - W	ak ax	W - Weak axis of the I-section	section		S - Stro	S - Strong axis of the I-section	the I-sec	tion		

b) B - Buckling axis of the I-section c) λ - Meak axis of the I-section c) λ - Slenderness parameter of the reinforced column d) IRS

f) MF -Maximum magnitude of the residual stress in the flange. $F_{\rm y}$ - Yield stress of the unreinforced column

d) IRS - Initial residual stress before reinforcing

e) PS - Residual stress pattern, as illustrated in Figure 4-1. f) MF -Maxig) MP -Maximum magnitude of the residual stress in the reinforcing plate. F_y -

h) ratio - The ratio of the out-of-straightness to the column length, L. i) W - Out-of-straightness in the weak direction.

Table A.1 (cont'd)

匠	FEA		1	ľ	Column			IRS		Welding	Initia	Initial Imperfection	ection	O _{ut} -	Out-of-straightness	Iness	Yield Strength	rength
OIII	model 1-section	Plate	<u>"</u>	В	Da Bb Length	- ا	PSc	MF	MP^g	Residual	peto	before reinforcing	rcing	after rei	after reinforcing, no load	no load	I-section	plate
Z	ĵo.			_	L(mm)					Stress	ratio	W ^h (mm)	S'(mm)	ratio		S'(mm)	(MPa)	(MPa)
==	1) (2)	3	(4) (5)	(5)	9	6	®	<u>6</u>	(<u>0</u>	(11)	(12)	(13)	(14)		(16)	(11)	(18)	<u>(61)</u>
==	120 W310x179 350x16	9 350x16	江	≩	7772	=	1-3	0.3F,	0.3F,	Œ,	1/2000	3.88	_	1/1570	4.94		300	300
7	121 W310x179 350x16	9 350x16	ĹŢ.	≩	<i>2777</i>	=======================================	1-3	0.3F,	$0.3F_{y}$. Œ	71000	77.7	0.00	L790	9.88		300	300
7	122 W310x179 350x16	9 350×16	ĹŢ,	≥	2777	=======================================	-3	0.3F,	$0.3F_{y}$	$0.7F_{y}$	71000	1.17	0.00	L790	06.6		300	300
7	123 W310x179 350x16	9 350×16	ĹT.	≥	7772	==	<u>-3</u>	$0.3F_{v}$	$0.3F_{v}$	Γ,	71000	77.7	0.00	L/880	8.88		300	300
~		9 350x16	Ľ.	≩	8200		-3	0.3F _v	$0.3F_{v}$, দ _ু	71000	8.20	0.00	D/800	10.23		230	350
1		9 350x16	Ľ,	≩	8200	=======================================	-3).3F,	$0.3F_{v}$	بت. د	2000	8.20	0.00	L/890	9.28		230	350
=======================================	-	9 350x16	Ľ.	≥	10598	1.5	-3	0.3F _v	$0.3F_{v}$, بر	L1400	7.57	0.00	1730	13.44		300	300
1		9 350x16	Ľ	≥	10598	1.5	<u>-3</u>	0.3F _v	0.3F _v	, بر	L/1050	10.00	0.00	L/910	11.63		300	300
7		9 350x16	ĹŢ,	≥	11182	1.5	-3	$0.3F_{\odot}$	0.3F _v	. ⊏	2138	00.01	0.00	L/850	13.13		230	350
2 160		9 350x16	ĹŢ.	≥	11182	1.5	-3	$0.3F_{ m c}$	$0.3F_{v}$	تا	218	10.00	0.00	0/6/7	11.51		230	350
_		9 350x16	Ľ,	S	4928	0.4	-3	$0.3F_{c}$	0.3F _v	بر	21000	0.97	4.93	096/1		5.11	300	300
=		9 350x16	Ľ.	S	4928	0.4	-3),3F,	$0.3F_{v}$, 따	71000	0.97	4.93	0/6/7		5.07	300	300
=		9 350x16	Œ,	S	5199	0.4	-3	$0.3F_{v}$	0.3F,	, п _у	1000	<u>5</u>	5.20	096/1		5.40	230	320
=	133 W310x179 350x16	9 350x16	Œ,	S	5199	0.4	<u>-3</u>).3F,	$0.3F_{v}$. 따	71000	5 .	5.20	17980		5.33	230	320
13	134 W310x179 350x16	9 350x16	Œ,	S	13551	=======================================	-3).3F,	$0.3F_{y}$. ፑ,	L/8000	0.33	1.69	176330		2.14	300	300
13	135 W310x179 350x16	9 350x16	Ľ,	S	13551		-3	0.3F,	$0.3F_{y}$	π <u>,</u>	L/2000	1.33	6.77	171590		8.56	300	300
=======================================	136 W310x179 350x16	9 350x16	Œ	S	13551	1.1	1-3 (0.3F,	$0.3F_{y}$	F,	1/1350	1.96	10.00	1/1070		12.64	300	30
🚡	a) D - Direction of reinforcing plates	of reinfore	ing	plat	es		1	: - Para	llel to	F - Parallel to the flanges	S		G - Para	G - Parallel to the web	web			
(q	b) B - Buckling axis	axis						V - We	ak axi	W - Weak axis of the I-section	section		S - Stron	S - Strong axis of the I-section	the I-sect	tion		
•	,		•			•						• • • •	•		,			

d) IRS - Initial residual stress before reinforcing

f) MF -Maximum magnitude of the residual stress in the flange. F_{y} - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate. h) ratio - The ratio of the out-of-straightness to the column length, L.

e) PS - Residual stress pattern, as illustrated in Figure 4-1. c) λ - Slenderness parameter of the reinforced column

i) W - Out-of-straightness in the weak direction.

FEA	Column				Column			IRS		Welding		Initial Imperfection	ection	Ont	Out-of-straightness	htness	Yield Strength	rength
mode	model I-section	Plate	<u>"</u>	æ	D" Bb Length λ'	~	PSc	MF	MP	Residual	bef	before reinforcing	rcing	after re	inforcing	after reinforcing, no load	I-section plate	plate
No.				_	L(mm)	_				Stress	ratio	W ^h (mm) S'(mm)	S'(mm)		W ^h (mm)	S'(mm)	(MPa)	(MPa)
Ξ	(2)	3	₹	(4) (5)	9	6	€	6	(<u>0</u>	$\widehat{\Xi}$	(12)	(13)	(14)	(15)	(91)	(17)	(18)	(61)
137	W310x179 350x16	350x16	뜨	S	13551	=	1-3	0.3F,	0.3F,	0.7F _y	L/1350	1.96	10.00	1/1070		12.66	300	300
138	W310x179 350x16	350x16	Ι.	S	13551	<u>:</u>	1-3	$0.3F_{v}$	$0.3F_{v}$	تت	11350	1.96	10.00	7118		11.40	300	300
139	W310x179 350x16	350x16	Ľ,	S	14297	:	<u>1-3</u>	$0.3F_{v}$	$0.3F_{v}$	بت. د	171400	2.02	10.30	1/1120		12.79	230	350
140		350x16	Ľ.	S	14297	Ξ	1-3	$0.3F_{v}$	$0.3F_{v}$	در	L/1400		10.30	1/1230		<u>2</u> .	230	350
141		350x16	Ľ,	S	18479	1.5	1-3	0.3F _v	0.3F _v	, Œ	71300	2.81	14.48	L/940		19.68	300	300
142	W310x179 350x16	350x16	[]	S	18479	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$	س	L1300	2.81	14.48	L1090		17.03	300	300
143	W310x179 350x16	350x16	<u> </u>	S	19496	1.5	1-3	0.3F _v	$0.3F_{\bullet}$	ت	11250	3.01	15.49	L/940		20.59	230	350
44	W310x179 350x16	350x16	ſĽ.	S	19496	1.5	1-3	$0.3F_{\nu}$	0.3F _v	, IT,	1/1250		15.49	71090		18.03	230	350
145	W310x179 350x16	350x16	Ö	≩	3720 0.4	0.4	1-3	$0.3F_{c}$	0.3F _v	<u>ب</u>	71000		0.00	17890	4.20		300	300
161 5	W310x179 350x16	350x16	Ö	≥	3720	0.4	1-3	$0.3F_{v}$	0.3F _v	, L	71000		0.00	L/920	4.04 40.4		300	300
	W310x179 350x16	350x16	ŋ	≥	3925	0.4	1-3	$0.3F_{c}$	0.3F _v	, IT ,	71000		0.00	2900	4.39		230	350
148	W310x179 350x16	350x16	Ö	≩	3925	0.4	1-3	$0.3F_{c}$	$0.3F_{v}$	` ਸ਼੍ਰ	71000	3.93	0.00	17930	4.22		230	350
149	W310x179 350x16	350x16	ŋ	≱	10229	Ξ	1-3	$0.3F_{c}$	0.3F _v	`ਧ	78000		0.00	L/4990	2.05		300	300
150	W310x179 350x16	350x16	g	≱	10229	=	1-3	$0.3F_{c}$	0.3F _v	`ਧ	L/2000	5.11	0.00	L/1250	8.21		300	300
151	W310x179 350x16	350x16	Ö	≱	10229	=	1-3	0.3F _v	0.3F _v	`ਧ	71000	10.00	0.00	1/640	16.05		300	300
152		350x16	Ö	≥	10229		1-3	$0.3F_{c}$	0.3F _v	$0.7\dot{F}_{v}$	71000	10.00	0.00	L/640	16.08		300	300
153		350x16	Ö	≩	10229	Ξ	1-3	$0.3F_{y}$	$0.3F_{y}$	$\mathbf{F}_{\mathbf{y}}$	71000	10.00	0.00	1790	12.99		300	300
a) D	a) D - Direction of reinforcing plates	reinfor	cing	Pg	les			F - Par	allel to	F - Parallel to the flanges	es		G - Para	G - Parallel to the web	e web			
b) B	b) B - Buckling axis	is	ŀ					W - W	eak axi	W - Weak axis of the I-section	section		S - Stror	ng axis o	- Strong axis of the I-section	ction		
c) \(\cdot \)	c) λ - Slenderness parameter of the reinforced	parame	er o	fthe	reinfor		column	=			d) IRS - Initial	- Initial re	residual stress before reinforcing	ess befo	re reinfo	rcing		
		•									1		•	•	:			

c) λ - Slenderness parameter of the reinforced column
 e) PS - Residual stress pattern, as illustrated in Figure 4-1.
 p) MP -Maximum magnitude of the residual stress in the reit

f) MF -Maximum magnitude of the residual stress in the flange. F_{y} - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate.

i) W - Out-of-straightness in the weak direction.

h) ratio - The ratio of the out-of-straightness to the column length, L.

п—	FEA					Column			IRS		Welding	Initia	Initial Imperfection	ection	Out	Out-of-straightness	tness	Yield Strength	rength
=	node	model I-section Plate Da B Length	Plate	<u></u>	æ	Length	~	PS	MF	MP ⁸	Residual	pefo	before reinforcing	rcing	after rei	after reinforcing, no load	no load	I-section plate	plate
	Ö.				_	L (mm)					Stress	ratio	W ^h (mm)	S'(mm)		W ^h (mm)	S ⁱ (mm)	(MPa)	(MPa)
•	Ξ	(2)	(3)	4	3	(4) (5) (6)	6	8	ව	9	(E)	(12)	(13)	(14)	(15)		(11)	(18)	(61)
!	154	W310x179 350x16	350x16	O	≥	10792	=	1-3	0.3F _y	0.3F,	π,	71100	10.00	0.00	17700	15.54		230	320
	155	W310x179 350x16	350x16	ŋ	≩	10792	. :	1-3	0.3F,	$0.3F_{y}$. π _y	Z138	10.00	0.00	L/850	12.80		230	320
	156	W310x179 350x16	350x16	Ö	€	13948	1.5	1-3	0.3F,	$0.3F_{y}$	ቪ	L1400	10.00	0.18	L/800	17.63		300	300
	157	W310x179 350x16	350x16	Ö	≩	13948	1.5	<u>-3</u>	0.3F,	0.3F,	. π _y	L1400	10.00	0.18	L 1030	13.50		300	300
	158	W310x179 350x16	350x16	Ö	≩	14716	1.5	1-3	0.3F,	0.3F	π,	L/1350	10.71	0.19	738	18.77		230	320
	159	W310x179	350x16	Ö	≩	14716	1.5	1-3	0.3F,	$0.3F_{y}$. π _y	171350	10.71	0.19	171020	14.42		230	350
	3	W310x179 350x16	350x16	Ö	S	4155	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$. г.	71000	0.85	4.15	171000		4.17	300	300
	191	W310x179 350x16	350x16	ŋ	S	4155	0.4	1-3	0.3F _v	$0.3F_{v}$. г.	71000	0.85	4.15	71000		4.19	300	300
	162		350x16	ŋ	S	4383	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$, tr _y	71000	0.99	4.38	71000		4.43	230	350
162	163		350x16	Ŋ	S	4383	0.4	<u>1-3</u>	0.3F,	$0.3F_{y}$. ጥ	71000	0.99	4.38	71000		4.41	230	350
	<u> </u>		350x16	g	S	11425	Ξ	1-3	0.3F	0.3F,	. ጥ	178000	0.28	1.43	177480		1.53	300	300
	165	W310x179 350x16	350×16	Ö	S	11425	=	1-3	0.3F,	$0.3F_{y}$	Т _у	L/2000	1.11	5.71	171870		6.12	300	300
	99	W310x179 350x16	350x16	Ö	S	11425	-:	1-3	0.3F,	0.3F,	Г,	1/1150	1.94	10.00	1/1070		10.71	300	300
	167	W310x179 350x16	350x16	Ö	S	11425	=:	1-3	0.3F	0.3F	$0.7F_{y}$	L1150	1.94	10.00	171070		10.73	300	300
	891	W310x179 350x16	350x16	O	S	11425	Ξ	<u>-3</u>	$0.3F_{y}$	$0.3F_{y}$	π _y	L/1150	1.94	10.00	1/1120		10.37	300	300
	691	W310x179 350x16	350x16	Ö	S	12054	- :	1-3	0.3F,	0.3F,	ጢ	L/2750	0.99	4.38	L/2420		2.00	230	320
	170	W310x179 350x16	350x16	ŋ	S	12054	I :	1-3	0.3F	0.3F,	뜨	1/1200	1.96	10.00	171170		10.32	230	350
i ez	0	a) D - Direction of reinforcing plates	reinfor	cing	pla	Ses			F - Par	allel to	F - Parallel to the flanges	S		G - Para	G - Parallel to the web	; web			
4) B -	b) B - Buckling axis	.is						W - W	eak axi	W - Weak axis of the I-section	section		S - Stror	g axis of	S - Strong axis of the I-section	tion		

j) S - Out-of-straightness in the strong direction.

f) MF -Maximum magnitude of the residual stress in the flange. $F_{\rm y}$ - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate.

c) λ - Slenderness parameter of the reinforced column e) PS - Residual stress pattern, as illustrated in Figure 4-1.

h) ratio - The ratio of the out-of-straightness to the column length, L.

i) W - Out-of-straightness in the weak direction.

d) IRS - Initial residual stress before reinforcing

FEA					Column		i	IRS		Welding	Initia	Initial Imperfection	ection	-jnO	Out-of-straightness	tness	Yield Strength	ength
mod	model I-section	Plate	"	æ	B ^b Length	ټر	PS	ΜF	MP	Residual	pefor	before reinforcing	rcing	after re	after reinforcing, no load	no load	I-section plate	plate
No.	-				L (mm)	_				Stress	ratio	Wh(mm) Si(mm)	S ⁱ (mm)		W ^h (mm)	S'(mm)		(MPa)
Ξ	(2)	(3)	€	(4) (5)	9	ϵ	€	<u>6</u>	9	(11)	•	(13)	(14)	(15)	(16) (17)	(17)		61)
171	W310x179 350x16	350x16	O	S	15579	1.5	1-3	0.3F _v	0.3F _y	ተ	1/1350	2.27	11.58	L/1240		12.61	300	300
172	W310x179 350x16	350x16	O	S	15579	1.5	I-3		$0.3F_{v}$. 따,	1/1350	2.27	11.58	L1290		12.06	300	300
173		350x16	Ö	S	16437	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$	ت	171300	2.44	12.44	1/1220		13.44	230	350
174		350x16	Ö	S	16437	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$	· 吒	U1300	2.44	12.44	L/1270		12.90	230	350
175	W150x30	130x5	Ľ,	≩	1236	0.4	1-3	$0.3F_{v}$, гг _,	71000	1.24	0.00	7980	_		300	300
176	W150x30	130x5	Œ	€	1236	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$, ਯੂ	71000	1.24	0.00	D867	1.27		300	300
171	W150x30	130x5	ĬŢ.	≥	1327	0.4	1-3	$0.3F_{v}$	0.3F	تر	71000	1.32	0.00	17980	1.35		230	350
178	3 W150x30	130x5	ĬŢ.	≱	1327	0.4	1-3	0.3F _v	0.3F	بت	Z1000	1.32	0.00	7980	1.34		230	350
179	W150x30	130x5	Ľ,	≩	3399	\exists	1-3	$0.3F_{v}$	0.3F	بتا ب	78000	0.43	0.00	L7100	0.48		300	300
. <u>8</u> 163		130x5	<u> </u>	≩	3399	Ξ	<u>1-3</u>	$0.3F_{v}$	$0.3F_{v}$, بر	172000	1.70	0.00	L1940			300	300
		130x5	Ľ	≥	3399	=	1-3	$0.3F_{v}$	$0.3F_{v}$, IT,	71000	3.40	0.00	17890			300	300
182		130x5	Ľ.	≥	3399	-	1-3	0.3F	$0.3F_{v}$	0.7F _v	71000	3.40	0.00	7890			300	300
183		130x5	Ľ.	≩	3399	- :	1-3	$0.3F_{v}$	$0.3F_{v}$	Г,	71000	3.40	0.00	L/930	3.64		300	300
184	W150x30	130x5	Ľ.	≩	3647	Ι.	1-3	$0.3F_{v}$	0.3F _v	. г.	71000	3.65	0.00	7900	•		230	320
185	W150x30	130x5	Ľ,	₹	3647	=	1-3	$0.3F_{v}$	0.3F _v	. ፑ,	71000	3.65	0.00	L940	3.88		230	350
186	W150x30	130x5	Ľ.	≥	4635	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$. π _{>}	71000	4.64	0.00	L/850	5.47		300	300
187		130x5	1	≩	4635	1.5	1-3	$0.3F_{y}$	$0.3F_{y}$	F,	1/1000	4.64	0.00	17920	5.05		300	8
a) D	a) D - Direction of reinforcing plates	of reinfore	ing	pla	tes			F - Par	allel to	F - Parallel to the flanges	Š		G - Para	G - Parallel to the web	e web			
b) B	b) B - Buckling axis	xis		ı				W - W	eak axi	- Weak axis of the I-section	section		S - Stroi	ng axis o	S - Strong axis of the I-section	tion		
לנס	c) A - Slenderness parameter of the reinforced column	s paramet	er o	fthe	reinfor	ced	Inloc	uu			d) IRS - Initial	Initial re	sidual st	ress befo	residual stress before reinforcing	cing		
: ;) •														1. 0	

f) MF -Maximum magnitude of the residual stress in the flange. $F_{\rm y}$ - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate. e) PS - Residual stress pattern, as illustrated in Figure 4-1.

h) ratio - The ratio of the out-of-straightness to the column length, L. i) W - Out-of-straightness in the weak direction.

j) S - Out-of-straightness in the strong direction.

Æ					Column			IRS ^d		Welding	Initia	Initial Imperfection	ection	Out	Out-of-straightness	tness	Yield Strength	ength
mod	model 1-section	Plate	<u>م</u>	B _b I	D" B" Length	۳	PS^c	MF	MP ⁸	Residual	petor	before reinforcing	rcing	after rei	after reinforcing, no load	no load	I-section plate	plate
No				_	L (mm)					Stress	ratio	W ^h (mm)	S'(mm)		W ^h (mm)	S'(mm)	(MPa)	(MPa)
Ξ	(2)	3	4	(4) (5)	9	6	€	<u></u>	(<u>0</u>	(E)	•	(13)	(14)	(15)		(11)	(18)	(61)
188	×	130x5	ഥ	≥	4973	1.5	-3	0.3F _v	0.3F,	я, ,	71000	4.97	0.00	17860	5.78		230	350
189	W150x30	130x5	Έ,	≱	4973	1.5	<u>-3</u>	0.3F _y	$0.3F_{y}$	ιτ _ν	71000	4.97	0.00	17920	5.38		230	320
8	W150x30	130x5	1	S	2300	0.4	1-3	0.3F,	0.3F,	Б	178000	90.0	0.29	L7730		0.30	300	300
161	W150x30	130x5	<u>L</u>	S	2300	0.4	<u>-3</u>	$0.3F_{v}$	$0.3F_{v}$	따,	L/2000	0.23	1.15	L194 0		1.19	300	300
192	W150x30	130x5	Œ.	S	2300	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$. п.	Z1000	0.45	2.30	7960		2.38	300	300
193	W150x30	130x5	Ĩ.	S	2300	0.4	I-3	$0.3F_{v}$	$0.3F_{v}$	بت. د	71000	0.45	2.30	7980		2.36	300	300
194	W150x30	130x5	[_	S	2468	0.4	1-3	0.3F _v	$0.3F_{v}$	تع	71000	0.49	2.47	1/960		2.56	230	320
195		130x5	Ľ,	S	2468	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$	۲.,	71000	0.49	2.47	17980		2.52	230	320
	_	130x5	<u> </u>	S	6326	Ξ	I-3	$0.3F_{v}$	$0.3F_{v}$	بتر د	L/8000	0.16	0.79	7/0000		96.0	300	300
62 164		130x5	<u> </u>	S	6326	1:1	<u>-3</u>	$0.3F_{v}$	$0.3\overline{F_{v}}$	ιτ _{>}	L/2000	0.62	3.16	L/1650		3.85	300	300
		130x5	Ľ.	S	6326	<u>-:</u>	1-3	0.3F,	$0.3F_{v}$, п,	Z1000	1.24	6.33	L/820		7.71	300	300
661	W150x30	130x5	Ľ.	S	6326	-:	1-3	0.3F,	0.3F,	0.7F _y	71000	1.24	6.33	L/8 20		7.72	300	300
200	W150x30	130x5	Ľ	S	6326	=	1-3	0.3F _v	$0.3F_{v}$	Т	171000	1.24	6.33	17880		7.17	300	300
201	W150x30	130x5	<u>(*</u> ,	S	6787	1:1	<u>1-3</u>	0.3F,	0.3F,	. π _y	71000	1.33	6.79	1/810		8.40	230	350
202	W150x30	130x5	11,	S	6787	1:1	1-3	0.3F _v	$0.3F_{y}$	ιτ ₂	71000	1.33	6.79	L/890		7.66	230	350
203	W150x30	130x5	Ľ	S	8626	1.5	<u>-3</u>	$0.3F_{v}$	$0.3F_{v}$	Гг	L/8000	0.00	1.08	L/6050		1.43	300	300
202		130x5	Ľ	S	8626	1.5	1-3	$0.3F_{y}$	$0.3F_{y}$	F,	L/2000	0.00	4.31	171500		5.74	300	300
a) D	a) D - Direction of reinforcing plates	reinfor	cing	plate	es			F - Par	allel to	F - Parallel to the flanges	Š		G - Para	G - Parallel to the web	e web			
b) B	b) B - Buckling axis	iis	1	1				W - W	eak axi	W - Weak axis of the I-section	section		S - Stroi	ng axis of	S - Strong axis of the I-section	tion		

j) S - Out-of-straightness in the strong direction.

f) MF -Maximum magnitude of the residual stress in the flange. F_y - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate.

c) λ - Slenderness parameter of the reinforced column e) PS - Residual stress pattern, as illustrated in Figure 4-1.

h) ratio - The ratio of the out-of-straightness to the column length, L.

i) W - Out-of-straightness in the weak direction.

d) IRS - Initial residual stress before reinforcing

FEA				ర	Column	! :		IRS		Welding	Initia	Initial Imperfection	ction	Ŏ	Out-of-straightness	ntness	Yield Strength	rength
mode	model I-section	Plate	7	D" B" Length	ength	ヹ	PS	MF	MPg	Residual	pefor	before reinforcing	rcing	after re	after reinforcing, no load	, no load	I-section plate	plate
No.				ļ	L (mm)					Stress	ratio	W ^h (mm) S ⁱ (mm	S'(mm)		W ^h (mm)	S ⁱ (mm)	(MPa)	(MPa)
Ξ	(2)	(3)	(4) (5)	(5)	9	8	(8)	<u></u>	(<u>0</u>	(E)	•	(13)	(14)	(15)	(91)	(11)	(18)	(61)
205	W150x30	130x5	Ľ.	S	8626	1.5	1-3	0.3F, (J.3F,	ጙ	71000	0.00	8.63	17750		11.57	300	300
206	W150x30	130x5	Ľ	S	8626	1.5	1-3 0).3F, (0.3F,	πŻ	71000	0.00	8.63	17860		10.06	300	300
207	W150x30	130x5	Ľ	S	9254	1.5	1-3 0	_	0.3F,	П,	71000	1.82	9.25	17750		12.28	230	320
208	W150x30	130x5	Ľ,	S	9254	1.5	1-3 0		0.3F,	. ጥ	71000	1.82	9.25	L/860		10.76	230	320
209	W150x30	175x5	ŋ	- *	1770	0.4	1-3 0	_	0.3F,	. ሞ,	78000	0.22	0.00	L/7050			300	300
210	W150x30	175x5	Ö	- *	1770	0.4	1-3 ($0.3F_{v}$. 吓	L/2000	0.88	0.00	U1770			300	300
211	W150x30	175x5	Ö	- ≯	1770	0.4	1-3 ($0.3F_{v}$, гг _,	71000	1.77	0.00	17880			300	300
212		175x5	Ö	_ ≽	1770	0.4	1-3 0		0.3F,	, tr _,	71000	1.77	0.00	17920			300	300
		175x5	Ö	- ≽	1873	0.4	1-3 ().3F, ($0.3F_{v}$, п _у	71000	1.87	0.00	7900			230	320
165 27 27	W150x30	175x5	Ö	- ≯	1873	0.4	1-3 0		$0.3F_{v}$	ت	71000	1.87	0.00	L/930			230	320
	W150x30	175x5	Ö	≯	4867	-	1-3 ().3F, ($0.3F_{v}$	Г.	L/8000	0.61	0.00	L/4900			300	300
216		175x5	Ö	≯	4867	-	1-3 ($0.3F_{v}$	تر	L/2000	2.43	0.00	L/1220	3.99		300	300
217		175x5	g	≫	4867	=	1-3 ($3F_{v}$. ፑ,	71000	4.86	0.00	1/610			300	300
218		175x5	Ö	≫	4867	=	1-3 0		$0.3F_{v}$	0.7F _v	71000	4.86	0.00	17610			300	300
219		175x5	Ö	₩	4867	-:	1-3 ($0.3F_{v}$	E,	71000	4.86	0.00	7760			300	300
220		175x5	Ö	₩	5150	-:	1-3 ($0.3F_{v}$	· 따	71000	5.15	0.00	17630			230	320
221		175x5	Ŋ	W S	5150	1.1	1-3	_	0.3F,	F,	71000	5.15	0.00	1780	89.9		230	350
E D	a) D - Direction of reinforcing plates	f reinfor	cing	plate	sa.		H	- Para	llel to	F - Parallel to the flanges	. yı		G - Parallel to the web	llel to th	e web			
b) B.	b) B - Buckling axis	is)				>	V - We	ak axi	W - Weak axis of the I-section	section		S - Stroi	ng axis o	S - Strong axis of the I-section	ction		
1)				

d) IRS - Initial residual stress before reinforcing

f) MF -Maximum magnitude of the residual stress in the flange. $F_{\rm y}$ - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate. h) ratio - The ratio of the out-of-straightness to the column length, L.

e) PS - Residual stress pattern, as illustrated in Figure 4-1. c) λ - Slenderness parameter of the reinforced column

i) W - Out-of-straightness in the weak direction.

j) S - Out-of-straightness in the strong direction.

FEA				Column	E E		IRSª	8_	Welding	1	Initial Imperfection	ection	Ont-	Out-of-straightness	ıtness	Yield Strength	rength
mode	model I-section	Plate	D, H	B ^b Length	gth ?	re PSe	° MF	MP^{g}	Residual	befo	before reinforcing	rcing	after rei	after reinforcing, no load	no load	I-section plate	plate
S. O.				L (mm	(MI				Stress	ratio	W ^h (mm)	S'(mm)	ratio	W ^h (mm) S'(mm)	S'(mm)		(MPa)
Ξ	(2)	3	(4) (5)	• _) (3	(8)	<u>6</u>	9	(11)		(13)			(16)	(11)	(18)	(61)
222	W150x30	175x5	C	W 663	37 1	5 1-	3 0.3F	0.3F,		7,8000	0.83	0.00	174410	1.50		300	300
223	W150x30	175x5	O O	W 6637	37 1	.S 1-3	3 0.3F _v	$0.3F_{\rm v}$, FT ₂	L/2000	3.32	0.00	71100	6.05		300	300
224	-	175x5	C	W 6637	37 1	.S I-3	3 0.3F,	$0.3F_{v}$		71000	6.64	0.00	L/550	12.10		300	300
225		175x5	Ö	× 66	37 1	.5 I-3	3 0.3F _v	$0.3F_{\rm v}$		71000	6.64	0.00	L/30	9.16		300	300
226	W150x30	175x5	Ö	X 70;	23	.5 1-3	3 0.3F	$0.3F_{v}$		71000	7.02	0.00	17570	12.32		230	350
227		175x5	9	.07 ∨	23	.5 1-3) 0.3F,	$0.3F_{v}$		71000	7.02	0.00	1740	9.46		230	350
228		175x5	0	S 20.	22 0	6-1 4:	0.3F,	$0.3F_{v}$		178000	0.05	0.25	17860		0.26	300	300
229		175x5	Ö	S 20.	22 0	.4 1-3	0.3F	$0.3F_{v}$		L/2000	0.20	10.1	D1960		1.03	300	300
		175x5	٥	S 20.	22 0	5-1 4:	3 0.3F _v	$0.3F_{v}$		71000	0.40	2.02	L/980		2.06	300	300
2 166	W150x30	175x5	Ü	S 20.	22 0	£-1 ♣:	3 0.3F,	$0.3F_{v}$		71000	0.40	2.02	7990		2.05	300	300
		175x5	Ö	\$ 21.	40 0	.4 I-3	3 0.3F _v	$0.3F_{v}$		71000	0.42	2.14	7990		2.17	230	350
233		175x5	Ö	S 2140	40 0	£-1 4:) 0.3F	$0.3F_{v}$, IT,	Z1000	0.42	2.14	17990		2.16	230	350
234		175x5	Ö	S 55	1 1959	.1 -3) 0.3F _v	$0.3F_{v}$		L/8000	0.14	69.0	L7200		0.77	300	300
235	-	175x5	Ü	S 5561	1 19	.1 1-3	0.3F,	$0.3F_{v}$		L/2000	0.54	2.78	71800		3.10	300	300
236	-	175x5	Ü	S 5561	1 19	.1 1-3	3 0.3F,	0.3F	Œ,	71000	1.09	5.56	1 /900		6.22	300	300
237		175x5	Ö	S 5561	1 19	.1 1-3	3 0.3Fy	$0.3F_{y}$	$0.7F_{y}$	71000	1.09	5.56	Z/200		6.21	300	300
238	W150x30	175x5	Ü	S 556	1 19	.1 1-3	_		F,	L/1000	1.09	5.56	L/940		5.95	300	38
a) D	a) D - Direction of reinforcing plates	f reinfor	cing p	Mates			F - Pa	rallel to	F - Parallel to the flanges	es		G - Para	G - Parallel to the web	s web			
b) B.	b) B - Buckling axis	Kis)				W - W	Veak ax	- Weak axis of the I-section	section		S - Stron	ng axis of	S - Strong axis of the I-section	tion		
)		•		,					:	•		1.6	J			

W - Weak axis of the I-section c) λ - Slenderness parameter of the reinforced column

f) MF -Maximum magnitude of the residual stress in the flange. F_{y} - Yield stress of the unreinforced column d) IRS - Initial residual stress before reinforcing

e) PS - Residual stress pattern, as illustrated in Figure 4-1.

g) MP-Maximum magnitude of the residual stress in the reinforcing plate. h) ratio - The ratio of the out-of-straightness to the column length, L.

i) W - Out-of-straightness in the weak direction.

j) S - Out-of-straightness in the strong direction.

FEA	A			ರ	Column			IRS		Welding	Initia	Initial Imperfection	ection) j	Out-of-straightness	itness	Yield Strength	rength
mod	model I-section	Plate	<u>"</u>	B B	D* B' Length	بر ۳	PS¢	MF	ΜÞβ	Residual	befo	before reinforcing	rcing	after rei	after reinforcing, no load	no load	1-section plate	plate
S.	ند.			_	L(mm)					Stress	ratio	Wh(mm) Si(mm	S'(mm)	ratio	W ^h (mm)	S'(mm)	(MPa)	(MPa)
Ξ	(2)	3	(4) (5)		9	(C)	æ	<u>(S</u>	(<u>0</u>	(E)	(12)	(13)	(14)		(16)	(11)	(18)	(61)
239	9 W150x30	175x5	O	S	5885	=	1-3	0.3F,	0.3F,	ቪ	71000	1.15	5.88	L/900		6.51	230	320
240	0 W150x30	175x5	ŋ	S	5885	-	1-3 ($0.3F_{y}$. π _y	71000	1.15	5.88	L/940		6.26	230	350
24	1 W150x30	175x5	Ö	S	7583	1.5	1-3	0.3F _v	0.3F,	. ፔ,	0008/7	0.19	0.95	0689/7		1.10	300	300
24,	2 W150x30	175x5	Ö	S	7583	1.5	1-3	0.3F _v	$0.3F_{y}$. ፑ,	L/2000	0.74	3.79	L1700		4.44	300	300
243		175x5	ŋ	S	7583	1.5	1-3	$0.3F_{v}$	0.3F _v	. ፑ,	71000	1.49	7.58	L/850		8.94	300	300
244		175x5	ŋ	S	7583	1.5	1-3	0.3F _v	$0.3F_{v}$	ت	71000	1.49	7.58	016/1		8.35	300	300
24.	-	175x5	Ö	S	8024	1.5	1-3 ($0.3F_{v}$	$0.3F_{v}$, гг ,	71000	1.58	8.02	D98/7		9.34	230	320
24		175x5	Ö	S S	8024	1.5	1-3	0.3F _v	0.3F _v	بر س	71000	1.58	8.02	L910		8.78	230	350
		130x8	Ľ,	- ≯	1234	0.4	1-3 (0.3F _v	$0.3F_{v}$	ت	71000	1.23	0.00	17970	1.28		300	300
167		130x8	ï	 ≯	1234	0.4	1-3	0.3F _v	$0.3F_{v}$	بت. د	71000	1.23	0.00	7980	1.26		300	300
249		130x8	Ľ	- ≽	1295	0.4	1-3 ($0.3F_{v}$	0.3F _v	. بت _ع	71000	1.30	0.00	L970	1.33		230	350
250		130x8	H	- ≽	1295	0.4	1-3 ($0.3F_{v}$	$0.3F_{v}$	Г,	71000	1.30	0.00	17980	1.32		230	320
25		130x8	H	₩ 3	3392		1-3	0.3F _v	0.3F,	. π <u>,</u>	L/8000	0.42	0.00	176770	0.50		300	300
252		130x8	ī	¥ 3	3392	1.1	1-3 (0.3F,	0.3F,	. ፔ,	L/2000	1.70	0.00	1700	2.01		300	300
25.	3 W150x30	130x8	ī	₩	3392		1-3 ($0.3F_{y}$	0.3F,	ۍ د کې	71000	3.39	0.00	L/850	4.01		300	38
254	4 W150x30	130x8	ï	₩	3392	1.1	1-3	0.3F,	0.3F,	$0.7F_{y}$	71000	3.39	0.00	L/850	4.02		300	300
255	5 W150x30	130x8	F	W 3	3392	1.1	1-3 (0.3Fy	$0.3F_{y}$	ъ	L/1000	3.39	0.00	17910	3.72		300	300
E) D	a) D - Direction of reinforcing plates	f reinfor	cing	plate	S		<u> </u>	7 - Para	llel to	F - Parallel to the flanges	Š.		G - Para	G - Parallel to the web	; web			
b) B	b) B - Buckling axis	x is	1	,				₩ - We	sak axi	W - Weak axis of the 1-section	section		S - Stron	g axis of	S - Strong axis of the I-section	tion		
		,	٠	•		-	•				out w	Taite in 1	and land	and backer	- roinfor			

c) λ - Slenderness parameter of the reinforced column

d) IRS - Initial residual stress before reinforcing

f) MF -Maximum magnitude of the residual stress in the flange. $F_y - Yield \ stress \ of \ the \ unreinforced \ column$ e) PS - Residual stress pattern, as illustrated in Figure 4-1.

g) MP-Maximum magnitude of the residual stress in the reinforcing plate. h) ratio - The ratio of the out-of-straightness to the column length, L.

i) W - Out-of-straightness in the weak direction.

j) S - Out-of-straightness in the strong direction.

Table A.1 (cont'd)

ÆA	1				Column	_		IRS		Welding	Initia	Initial Imperfection	ction	Out-	Out-of-straightness	tness	Yield Strength	ength
mode	model 1-section	Plate	<u>"</u>	B,	B ^b Length	٠,۲	PSc	MF	MPg	Residual	pefo	before reinforcing	rcing	after rei	after reinforcing, no load	no load	I-section	plate
No.				_	L (mm)					Stress	ratio	W ^h (min) S'(min	S'(mm)	ratio	W ^h (mm) S'(mm)	S'(mm)		(MPa)
Ξ	(2)	3	€	(4) (5)	9	8	€	6	9	(E)		(13)	(14)	:	(16)	(17)	(18)	(19)
256	×	130x8	ഥ	≥	3560	=	1-3	0.3F _v	0.3F _v	F,	71000	3.56	0.00	7,860	4.13		230	350
257		130x8	Ľ,	≥	3560	=	1-3	0.3F,	0.3F,	т,	71000	3.56	0.00	L/920	3.86		230	350
258		130x8	Ľ,	≥	4626	1.5	1-3	$0.3F_{v}$	0.3F _v	ب	2000	4.63	0.00	17800	5.76		300	300
259		130x8	Δ.	≥	4626	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$	بت. د	71000	4.63	0.00	7890	5.18		300	300
260		130x8	Ľ.	≩	4854	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$. بت _ع	71000	4.85	0.00	L/820	5.91		230	350
261		130x8	Ľ.	≩	4854	1.5	1-3	0.3F _c	$0.3F_{v}$	ب	71000	4.85	0.00	200	5.37		230	350
262		130x8	Ľ,	S	2366	0.4	1-3	0.3F _v	0.3F _v	` لتــُ	71000	0.46	2.36	L/950		2.50	300	300
263	-	130x8	Ľ.	S	2366	0.4	1-3	0.3F _v	$0.3F_{v}$	`Ľ	71000	0.46	2.36	1/970		2.45	300	300
		130x8	Ľ,	S	2483	0.4	1-3	0.3F _v	$0.3F_{v}$	بت	71000	0.23	2.48	L/950		2.61	230	320
765 361 361		130x8	Ľ.	S	2483	0.4	1-3	0.3F _v	$0.3F_{v}$	`Œ	71000	0.23	2.48	17970		2.56	230	350
		130x8	Ι.,	S	6507	-:	1-3	$0.3F_{v}$	$0.3F_{v}$	ت	L/8200	0.00	0.79	L/6140		90.1	300	300
267		130x8	Ľ,	S	6507	Ι:	1-3	$0.3F_{v}$	$0.3F_{v}$. بر	1/2100	0.00	3.12	171560		4.19	300	300
268		130x8	Ľ.	S	6507	1.1	1-3	$0.3F_{v}$	$0.3F_{v}$, π _,	71000	0.00	6.51	L/740		8.75	300	300
269		130x8	ĬŢ,	S	6507	=	1-3	$0.3F_{v}$	$0.3F_{v}$	0.7F,	71000	0.00	6.51	L740		8.76	300	300
270		130x8	Ľ.	S	6507	Ξ	1-3	$0.3F_{v}$	$0.3F_{v}$		71000	0.00	6.51	L/850		1.7.1	300	300
271		130x8	Έ.	S	6828	=	1-3	$0.3F_{v}$	$0.3F_{v}$		71000	0.00	6.83	1760		9.00	230	350
272		130x8	F	S	6828	-:	1-3	$0.3F_{y}$	$0.3F_{\chi}$		71000	0.00	6.83	17850		8.02	230	350
Q (a	a) D - Direction of reinforcing plates	f reinfor	ring	plat	es			F - Par	allel to	F - Parallel to the flanges	Š		G - Para	G - Parallel to the web	; web			
b) B	b) B - Buckling axis	cis	1	ı				W - W.	eak axi	W - Weak axis of the I-section	section		S - Stron	g axis of	S - Strong axis of the I-section	tion		
c) \(\zeta \)	c) \(\cdot \). Slendemess parameter of the reinforced column	parame	er o	fthe	reinfor	o poo.	olum	Ħ			d) IRS - Initia		residual stress before reinforcing	ess befor	re reinfor	cing		
		L					i						٠		:	,	5	

j) S - Out-of-straightness in the strong direction. h) ratio - The ratio of the out-of-straightness to the column length, L.

f) MF -Maximum magnitude of the residual stress in the flange. $F_y - Yield \ stress \ of \ the \ unreinforced \ column$

g) MP -Maximum magnitude of the residual stress in the reinforcing plate.

i) W - Out-of-straightness in the weak direction.

e) PS - Residual stress pattern, as illustrated in Figure 4-1.

Table A.1 (cont'd)

FEA				Column			IRS	P. 6	Welding	Initia	Initial Imperfection	ection	Out	Out-of-straightness	ıtness	Yield Strength	rength
mode	model 1-section	Plate	D* B	Da Bb Length	gth A	C. PSe	MF	MP	Residual	befo	before reinforcing	rcing	after rei	after reinforcing, no load	no load	I-section	ı plate
Z O				L (mm)	(m				Stress	ratio	W ^h (mm)	S'(mm)		W ^h (mm)		(MPa)	(MPa)
Ξ	(2)	(3)	(4) (5)	9		(8)	<u>S</u>	(OT)	(11)	(12)	(13)	(14)		(91)		(18)	(19)
273	W150x30	130x8	F	\$ 8873	73 1	.5 1-3	3 0.3F	١.	R,	71000	-0.04	8.87	17680		13.00	300	300
274	W150x30	130x8	E.	\$ 8873	73	.5 1-3	3 0.3F,	_	ፎ	71000	-0.04	8.87	17810		10.93	300	300
275	W150x30	130x8	Ľ.	\$ 9310	1 01	.5 1-3	3 0.3F	_	π _y	71000	-0.04	9.31	D/00		13.34	230	350
276	W150x30	130x8	Z.	\$ 9310	1 01	.5 1-3	3 0.3F	_	ᅜ	71000	-0.04	9.31	17820		11.38	230	350
277	W150x30	175x8	S S	V 1947	47 0	-1 4:	3 0.3F		<u>π</u> ,	71000	1.95	0.00	L/850	2.31		300	300
278	W150x30	175x8	S	V 1947	47 0	.4 L-3	3 0.3F	_	. π _y	71000	1.95	0.00	7900	2.17		300	300
279	W150x30	175x8	Ö	V 2012	12 0	.4 I-3	1 0.3F		. ௩	21000	2.01	0.00	17870	2.31		230	350
280	W150x30	175x8	5	V 2012	12 0	.4 I-3	1 0.3F		. ፑ,	71000	2.01	0.00	L/920	2.20		230	350
	W150x30	175x8	S O	V 5353	53 1	.1 1-3	9 0.3F _v		. ፑ,	L/8000	0.67	0.00	L/4420	1.21		300	300
28 169	W150x30	175x8	S	V 5353	53 1.	.1 1-3	9 0.3F _v	, 0.3F,	, IT,	L/2000	2.68	0.00	71100	4.85		300	300
283	W150x30	175x8	S	V 5353	53 1	.1 1-3	0.3F,		. ፑ,	71000	5.35	0.00	L/550	9.71		300	300
284	W150x30	175x8	S	V 5353	53	.1 1-3	9 0.3F,	_	$0.7F_{v}$	71000	5.35	0.00	17550	9.72		300	300
285	W150x30	175x8	C	V 5353	53	.1 1-3	3 0.3F,	_	T,	71000	5.35	0.00	17720	7.46		300	300
286	W150x30	175x8	₹ U	V 5531	31	.1 1-3	3 0.3F,		. ፑ,	71000	5.53	0.00	17580	9.56		230	350
287	W150x30	175x8	S S	V 553	31	.1 1-3	3 0.3F		ቪ	71000	5.53	0.00	L/130	7.53		230	320
288	W150x30	175x8	S D	v 7300	98	.5 I-3	3 0.3F	_	ቪ	Z1000	7.30	0.00	L/530	13.92		300	300
289	W150x30	175x8	N D	V 73(7300	.5 I-3	3 0.3F		F,	L1000	7.30	0.00	1710	10.33		300	8
a) D.	a) D - Direction of reinforcing plates	of reinfor	cing p	lates			F - P.	arallel to	F - Parallel to the flanges	Se		G - Para	G - Parallel to the web	e web			
b) B -	b) B - Buckling axis	xis					→	Veak ax	W - Weak axis of the I-section	section		S - Stror	ng axis of	S - Strong axis of the I-section	tion		
	.			•	,	•							L.6	J			

e) PS - Residual stress pattern, as illustrated in Figure 4-1. c) λ - Slenderness parameter of the reinforced column

f) MF -Maximum magnitude of the residual stress in the flange. $F_y - Yield \ stress \ of \ the \ unreinforced \ column$ g) MP -Maximum magnitude of the residual stress in the reinforcing plate.

h) ratio - The ratio of the out-of-straightness to the column length, L. i) W - Out-of-straightness in the weak direction.

j) S - Out-of-straightness in the strong direction.

d) IRS - Initial residual stress before reinforcing

Table A.1 (cont'd)

FEA					Column			IRS		Welding	Initia	Initial Imperfection	ection	Out-	Out-of-straightness	tness	Yield Strength	rength
mode	model 1-section	Plate	<u>"</u>	B, I	D" B" Length	ئر	PSe	MF	ΜPg	Residual	peto	before reinforcing	rcing	after rei	after reinforcing, no load		1-section plate	plate
N.	_				L (mm)	_				Stress	ratio	W ^h (mm)	S'(mm)	ratio	W ^h (mm)	S'(mm)	(MPa)	(MPa)
Ξ	(2)	3	₹	(4) (5)	9	6	€	9	9	Ξ	•	(13)			(91)	(17)	(18)	(19)
290	≩	175x8	O	≥	7543	1.5	1-3	0.3F	0.3F _v	F,	71000	7.54	0.00	L/530	14.19		230	350
291		175x8	Ö	≱	7543	1.5	1-3	0.3F,	$0.3F_{v}$. п <u>,</u>	71000	7.54	0.00	L/10	10.60		230	320
292		175x8	Ö	S	9961	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$	다	71000	0.38	1.96	17970		2.02	300	300
293		175x8	g	S	9961	0.4	1-3	0.3F _v	$0.3F_{v}$	` Œ	71000	0.38	1.96	7980		2.00	300	300
294		175x8	Ö	S	2032	0.4	1-3	0.3F _v	$0.3F_{v}$	<u>ب</u> ت	21000	0.40	2.03	L/980		2.08	230	350
295		175x8	Ö	S	2032	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$	س	71000	0.40	2.03	L/980		2.07	230	350
296		175x8	Ö	S	5407	==	1-3	0.3F _v	$0.3F_{v}$	` Œ	L/8000	0.13	99.0	0169/7		0.78	300	300
297		175x8	g	S	5407	-:	1-3	$0.3F_{v}$	$0.3F_{v}$	· 따	L/2000	0.53	2.70	1/1/30		3.13	300	300
298		175x8	Ö	S	5407	Ξ:	<u>1-3</u>	$0.3F_{v}$	$0.3F_{v}$	ت	21000	90.1	5.41	098/7		6.28	300	300
ි 170		175x8	Ö	S	5407	=	1-3	$0.3F_{v}$	$0.3F_{c}$	$0.7F_{v}$	71000	90.1	5.41	098/7		6.27	300	300
		175x8	Ö	S	5407	-:	1-3	$0.3F_{v}$	0.3F _v	, L	71000	90.1	5.41	L/910		5.92	300	300
301		175x8	Ö	S	5587	-:	1-3	0.3F _v	$0.3F_{v}$	<u>.</u> ت	71000	69.	5.59	L/880		6.35	230	350
305		175x8	Ö	S	5587	-:	1-3	$0.3F_{v}$	$0.3F_{v}$	ت	71000	1.09	5.59	L/930		6.04	230	350
303		175x8	Ö	S	7373	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$, Œ,	71000	1.45	7.37	L/810		9.14	300	300
304		175x8	Ö	S	7373	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$. ቪ	171000	1.45	7.37	L/880		8.37	300	300
305		175x8	Ö	S	819/	1.5	1-3	$0.3F_{v}$	0.3F,	. ቪ,	71000	1.50	7.62	L/820		9.26	230	350
306		175x8	Ö	S	8192	1.5	1-3	$0.3F_{\chi}$	$0.3F_{y}$	F,	1/1000	1.50	7.62	17890		8.55	230	350
a D D	a) D - Direction of reinforcing plates	f reinford	sing	plate	SS			F - Par	allel to	F - Parallel to the flanges	S		G - Para	G - Parallel to the web	web			
b) B	b) B - Buckling axis	xis)	1			-	₩ - ₩	eak axi	- Weak axis of the I-section	section		S - Stron	ig axis of	S - Strong axis of the I-section	tion		
c) \	c) A - Slenderness parameter of the reinforced column	; paramel	ero	fthe	reinfor	o pao.	olum	s			d) IRS -	Initial re	d) IRS - Initial residual stress before reinforcing	ess befor	e reinfor	cing		
		•		:			į	•			1 411 0			As See the	Inchistor,		the flance	4

f) MF -Maximum magnitude of the residual stress in the flange.

Fy - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate. h) ratio - The ratio of the out-of-straightness to the column length, L.

e) PS - Residual stress pattern, as illustrated in Figure 4-1.

i) W - Out-of-straightness in the weak direction.

j) S - Out-of-straightness in the strong direction.

Table A.1 (cont'd)

FEA					Column			IRS		Welding		Initial Imperfection	ection	Out	Out-of-straightness	tness	Yield Strength	ength
mode	model I-section Plate Da Bb Length	Plate	Ď	æ	Length	ېر	PS	MF	ΜPg	Residual	pelo	before reinforcing	rcing	after rei	after reinforcing, no load		I-section plate	plate
S.					L(mm)	_				Stress	ratio	Wh(mm) Shm	S'(mm)	ratio	ratio Wh(mm) Si(mm)	S'(mm)	(MPa)	(MPa)
Ξ	(2)	3	₹	3	(3) (4) (5) (6) (7) (8)	6	€	6	<u>(e</u>	(E)	(12)	(13)	(14)	(15)	(16)	(11)	(18)	(19)
307	W310x179 350x16 G W 3720 0.4	350x16	O	≥	3720	0.4	1-3	0.3F _v	0.3F _v	F,	L/1105	3.37		62677	3.80		300	300
308	W310x179 350x16 G	350x16	Ö	≩	10229	=	<u>1-3</u>	$0.3F_{v}$	$0.3F_{v}$	Т,	1/1267	8.07		L/789	12.96		300	300
309		350x16	g	S	11425	=	1-3	$0.3F_{v}$	$0.3F_{v}$. L.	L/1697		6.73	L/1574		7.26	300	300
310		350x16	Ľ.	S	18479	1.5	1-3	$0.3F_{v}$	$0.3F_{v}$	د	L/2029		9.11	L/858		7.37	300	300
311	W150x30 130x5	130x5	Ľ	S	6326	=	1-3	$0.3F_{v}$	$0.3F_{v}$, 따 <u>,</u>	Z1081		5.85	F7808		10.67	300	300
312		130x5	ĹŢ.	S	8626	1.5	<u>.3</u>	$0.3F_{v}$	$0.3F_{v}$. ፔ.	LI133		7.61	7860		10.03	300	300
313		350x25	O	€	4103 0	0.4	1-3	$0.3F_{v}$	$0.3F_{v}$, п _,	171171	3.50		17983	4.17		300	300
314	W310x179 350x25	350x25	Ö	≥	11281	=	1-3	$0.3F_{v}$	$0.3F_{v}$, 吓,	1741	6.48		L/985	11.45		300	300
315	W310x179 350x25	350x25	Ö	≥	15383	1.5	<u>1-3</u>	$0.3F_{v}$	$0.3F_{v}$. ፫	L/2263	08.9		V1194	12.89		300	300
9E 171	316 W310x179 350x16 G W	350x16	Ö	≩	10229	Ξ	1-3	$0.3F_{v}$	$0.3F_{v}$, π ,	1/3010	3.40		171862	5.49		300	300
317	W310x179 350x16 G W 13948	350x16	Ö	≱	13948	1.5	1-3	$0.3F_{y}$	$0.3F_{\chi}$	F	L/2189	6.37		1/1227	11.36		300	38
a) D	a) D - Direction of reinforcing plates	Freinford	cing	Pla	tes			F - Par	allel to	F - Parallel to the flanges	sa		G - Para	G - Parallel to the web	web :			
b) B.	b) B - Buckling axis	iis						W - W	eak axi	W - Weak axis of the I-section	section		S - Stron	ig axis of	S - Strong axis of the I-section	tion		
c) y -	c) λ - Slenderness parameter of the reinforce	parame	er o	fthe	reinfor	rced	ed column	E			d) IRS -	. Initial re	d) IRS - Initial residual stress before reinforcing	ess befor	e reinfor	sing		

c) λ - Slenderness parameter of the reinforced column
 e) PS - Residual stress pattern, as illustrated in Figure 4-1.

f) MF -Maximum magnitude of the residual stress in the flange. $F_{\rm y}$ - Yield stress of the unreinforced column

g) MP -Maximum magnitude of the residual stress in the reinforcing plate.

h) ratio - The ratio of the out-of-straightness to the column length, L.

i) W - Out-of-straightness in the weak direction.

j) S - Out-of-straightness in the strong direction.

Appendix B

Analysis Results Description

Analysis Results Description

All 317 model analyses were performed to verify the behavior of the reinforced steel columns with welded cover plates in the analysis. The preload condition and analysis results of the analysed models are presented in Table B.1.

Column (1) of Table B.1 presents the number of the finite element analysed models. Columns (2) and (3) present the designations of I-sections and cover plates respectively.

Column (4) "D" is the direction of reinforcing plates. In this column, F represents that the cover plates are reinforced on the column along the flanges, as shown in Figure 2.1 (a). G represents that the cover plates are attached to the column at the flange tips of the column and parallel to the web, as shown in Figure 2.1 (b).

Column (5) "B" is the buckling axis of the reinforced column. W represents the same axis as the weak axis of the I-section before reinforcing, and S represents the same axis as the strong axis of the I-section before reinforcing. Column (6) " λ " is the slenderness parameter of the reinforced column on the buckling direction. Column (7) "A" is the area of the reinforced cross-section consisting of the rolled section and cover plates.

Columns (8) and (9) present the yield strength of the I-section and the cover plates respectively. Columns (10) and (11) present the pre-loads on the column before reinforcing. Column (10) " P_0 " is the magnitude of the pre-load on the column before reinforcing. Column (11) " P_0/P_{u2} " presents the ratio of the pre-load to the expected load carrying capacity of the unreinforced column predicted using SSRC curve 2.

Column (12) "P_{fex}/P_{ry}" presents the ratio of the load carrying capacity of reinforced column by mathematical model analyses to the yield strength of the reinforced column. Column (13) "P_{r1}/P_{ry}" presents the ratio of the load carrying capacity of the reinforced column predicted using SSRC curve 1 to the yield strength of the reinforced column. Column (14) "P_{r2}/P_{ry}" presents the ratio of the load carrying capacity of the reinforced column predicted using SSRC curve 2 to the yield strength of the reinforced column. Column (15) "P_{rc1}/P_{ry}" presents the ratio of the load carrying capacity of the reinforced

column predicted using CSA curve 1 to the yield strength of the reinforced column. Column (16) " P_{rc2}/P_{ry} " presents the ratio of the load carrying capacity of the reinforced column predicted using CSA curve 2 to the yield strength of the reinforced column.

Table B.1 Analysis Result Description

FEA							Yield Strength	trength	Prel	Preload					
model	1 1-section	Plate	<u>"</u>	æ	ېر	Area	I-section	plate	P_0^d	P_0/P_{u2}^c	P_{fca}/P_{ry}^{f}	$P_{rl}/P_{ry}^{\ \ \beta}$	P_{r2}/P_{ry}^{h}	P_{rcl}/P_{ry}^{l}	P_{rc2}/P_{ry}^{j}
Z O						(mm ²)	(MPa)	(MPa)	(KN)						
Ξ	(2)	(3)	<u>4</u>	(5)	9	(1)	(8)	(6)	(10)	(II)	(12)	(13)	(14)	(15)	(91)
-	W200x46	180x9.52	ഥ	≥	0.5	9287	260	260	405	0.3	0.97	0.95	0.86	0.97	0.88
2	W200x46	180x9.52	ഥ	≩	0.5	9287	700	260	0	0.0	0.98	0.95	0.86	0.97	0.88
£	W200x46	180x9.52	Œ	≱	0:1	9287	260	260	405	0.4	0.65	0.75	0.61	0.73	09'0
4	W200x47	180x9.53	Ľ.	≥	0:1	9287	260	260	405	0.4	0.65	0.75	0.61	0.73	0.60
5	W200x46	180x9.52	Ľ	≥	0:1	9287	260	260	405	0.4	0.65	0.75	0.61	0.73	0.60
9	W200x47	180x9.53	Ľ	≩	0:1	9287	260	260	405	0.4	0.65	0.75	0.61	0.73	0.60
7	W200x46	180x9.52	Ľ	≩	0.1	9287	260	260	405	0.4	0.63	0.75	0.61	0.73	09.0
- 00	W200x46	180x9.52	Œ	≥	0:1	9287	260	260	405	0.4	99.0	0.75	0.61	0.73	0.60
6	W200x46	180x9.52	Œ	≥	0.1	9287	260	260	405	0.4	0.65	0.75	0.61	0.73	0.60
2 175	W200x46	180x9.52	Œ	≩	0.1	9287	260	260	405	0.4	99.0	0.75	0.61	0.73	0.60
=	W200x46	180x9.52	Ľ,	≱	0.1	9287	260	260	405	0.4	0.60	0.75	0.61	0.73	0.60
12	W200x46	180x9.52	II.	≩	0:1	9287	260	260	405	0.4	0.65	0.75	0.61	0.73	0.60
13	W200x46	180x9.52	12	S	0.1	9287	260	260	405	0.5	0.63	0.75	0.61	0.73	0.60
7	W200x46	180x9.52	ĹŢ.	S	0:1	9287	260	260	405	0.5	0.63	0.75	0.61	0.73	0.60
15	W200x46	180x9.52	Ľ	S	0:1	9287	260	260	405	0.5	0.63	0.75	0.61	0.73	0.60
91	W200x46	180x9.52	ഥ	S	1.0	9287	260	260	405	0.5	0.63	0.75	0.61	0.73	0.60
a) D	a) D - Orientation of reinforcing plates	f reinforcing	plate	S		F - Parall	- Parallel to the flanges	anges			G - Paral	G - Parallel to the web	web		
b) B -	b) B - Buckling axis of the reinforced column	of the reinf	orced	colum	_	W - Weal	- Weak axis of the rolled		section		S - Strong	S - Strong axis of the rolled section	he rolled	section	
)														

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) W - Weak axis of the rolled section d) P₀ - Pre-load

 $P_{\rm ry}$ - Yield strength of the reinforced column f) P_{fea} - Load carrying capacity obtained from the finite element analysis c) λ - Slenderness parameter.

g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1) h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

FEA							Yield Strength	trength	Pre	Preload					
model	el I-section	Plate	D.	Bp	~	Area	I-section	plate	P ₀ d	P_0/P_{u2}^c	P _{fee} /P _{ry} f	$P_{rl}/\!\!/P_{ry}^{\ \ 8}$	$P_{r2}/P_{ry}^{\ \ h}$	Prel/Pry	$P_{rc2}/P_{ry}^{\ j}$
Š						(mm ²)	(MPa)	(MPa)	(kN)					0 0 0 0 0 0 0 0 0	
Ξ	(2)	(3)	€	(5)	9	(7)	(8)	(6)	(10)	(11)	(12)	(13)	(14)	(15)	(10)
12	W310x179	290x25	Œ.	≥	0.4	37300	300	300	3760	9.0	1.00	0.98	0.92	0.99	0.94
8	W310x179	290x25	Ľ.	≱	0.4	37300	300	300	3760	9.0	0.99	0.98	0.92	0.99	0.94
61	W310x179	290x25	Ľ.	≱	0.4	37300	300	300	3760	9.0	0.97	0.98	0.92	0.99	0.94
70	W310x179	290x25	Œ,	≱	0.4	37300	300	300	2507	0.4	0.98	0.98	0.92	0.99	0.94
21	W310x179	290x25	Ľ	≩	0.4	37300	230	350	2926	9.0	96.0	0.98	0.92	0.99	0.94
22	W310x179		Ľ,	≩	0.4	37300	230	350	1950	0.4	96.0	0.98	0.92	0.99	0.94
23	W310x179		Ľ.	≩	=	37300	300	300	2152	9.0	0.6	99.0	0.54	99.0	0.54
75	W310x179	290x25	Ľ	≩	-:	37300	300	300	2152	9.0	0.60	99.0	0.54	99.0	0.54
25	W310x179	290x25	Œ	≩	Ξ	37300	300	300	2152	9.0	0.56	89.0	0.54	99.0	0.54
२ १७६	W310x179	290x25	Œ,	≱	Ξ:	37300	300	300	2152	9.0	0.56	99.0	0.54	99.0	0.54
27	W310x179	290x25	Œ,	≱	=	37300	300	300	1435	0.4	0.57	99.0	0.54	99.0	0.54
28	W310x179	290x25	Ľ	≩	Ξ	37300	230	350	9981	9.0	0.51	99.0	0.54	99.0	0.54
29	W310x179	290x25	Œ,	≩	-	37300	230	350	1244	0.4	0.54	89.0	0.54	99.0	0.54
30	W310x179	290x25	Œ	≯	1.5	37300	300	300	1401	9.0	0.42	0.41	0.35	0.42	0.36
31	W310x179	290x25	Œ	≯	1.5	37300	300	300	1401	9.0	0.38	0.41	0.35	0.42	0.36
32	W310x179	290x25	Ľ	≱	1.5	37300	300	300	1401	9.0	0.35	0.41	0.35	0.42	0.36
a) D	a) D - Orientation of reinforcing plates	reinforcing	g plate	Š		F - Parall	- Parallel to the flanges	anges			G - Paral	G - Parallel to the web	web		
2	b) B - Buckling axis of the reinforced column	of the rein	forced	rolin	Ę	W - Wea	W - Weak axis of the rolled section	e rolled se	ection		S - Stron	S - Strong axis of the rolled	the rolled	1 section	
מ (כ	- Durning and		1		=	; ;	: .> !: :				1				į

W - Weak axis of the rolled section d) P₀ - Pre-load b) B - Buckling axis of the reinforced column

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2)

 $P_{\rm ry}$ - Yield strength of the reinforced column c) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load f) P_{rea} - Load carrying capacity obtained from the finite element analysis

g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) $P_{\rm r2}$ - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

									,						
FEA							Yield Strength	rength	Pre	Preload					
model	I-section	Plate	<u>"</u>	æ	ž	Arca	I-section	plate	$\mathbf{P}_0^{\mathbf{q}}$	P_0/P_{u2}^c	Pfca/Pry	$P_{rl}/P_{ry}^{\ \ B}$	$P_{r2}/P_{ry}^{\ \ h}$	P_{rel}/P_{ry}^{l}	$P_{rc2}/P_{ry}^{\ j}$
So.						(mm ₂)	(MPa)	(MPa)	(kN)						
Ξ	(2)	(3)	€	(5)	9	6	(8)	(6)	(10)	(11)	(12)	(13)	(14)	(15)	(16)
33	W310x179	290x25	ഥ	≱	1.5	37300		300	934	0.4	0.36	0.41	0.35	0.42	0.36
34	W310x179	290x25	<u></u>	≱	1.5	37300	230	350	1224	9.0	0.34	0.41	0.35	0.42	0.36
35	W310x179	290x25	Œ,	≥	1.5	37300	230	350	816	0.4	0.35	0.41	0.35	0.42	0.36
36	W310x179	290x25	Œ	S	0.4	37300	300	300	3697	9.0	0.93	0.98	0.92	0.09	0.94
37	W310x179	290x25	Ľ,	S	0.4	37300	300	300	2465	0.4	0.93	0.98	0.92	0.09	0.94
38	W310x179	290x25	ഥ	S	0.4	37300	230	350	2883	9.0	0.94	0.98	0.92	0.09	0.94
30	W310x179	290x25	Ľ,	S	0.4	37300	230	350	1922	0.4	0.94	0.98	0.92	0.09	0.94
9	W310x179	290x25	<u> </u>	S	Ξ	37300	300	300	1901	9.0	89.0	99.0	0.54	99.0	0.54
	W310x179	290x25	Ľ,	S	Ξ	37300	300	300	1901	9.0	0.63	99.0	0.54	99.0	0.54
5 771	W310x179	290x25	Ľ	S	=	37300	300	300	1901	9.0	0.61	99.0	0.54	99.0	0.54
43	W310x179	290x25	Œ	S	=	37300	300	300	1901	9.0	0.62	99.0	0.54	99.0	0.54
4	W310x179	290x25	Ľ	S	Ξ:	37300	300	300	1268	0.4	0.63	99.0	0.54	99.0	0.54
45	W310x179	290x25	I	S	=	37300	230	350	1651	9.0	0.57	99.0	0.54	99.0	0.54
46	W310x179	290x25	ᄄ	S	Ξ	37300	230	350	1011	0.4	0.59	99.0	0.54	99.0	0.54
47	W310x179	290x25	<u> </u>	S	1.5	37300	300	300	1227	9.0	0.40	0.41	0.35	0.42	0.36
48	W310x179	290x25	F	S	1.5	37300	300	300	818	0.4	0.41	0.41	0.35	0.42	0.36
a) D -	a) D - Orientation of reinforcing plates	reinforcing	g plate:	يم		F - Parall	F - Parallel to the flanges	ınges			G - Paral	G - Parallel to the web	web		
b) B -	b) B - Buckling axis of the reinforced colun	of the rein	forced	colum	9	W - Weal	- Weak axis of the rolled section	e rolled se	ction		S - Strony	S - Strong axis of the rolled section	he rolled	section	

b) B - Buckling axis of the reinforced column W - Weak axis of the Ic) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2)

 $P_{\rm ry}$ - Yield strength of the reinforced column g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1) f) P_{fea} - Load carrying capacity obtained from the finite element analysis

h) P_{12} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) Prel - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

j) P_{rc2} - Load carrying capacity of the reinforced column (predicted using the CSA curve 2)

Table B.1 (cont'd)

FEA							Yield Strength	rength	Pre	Preload					
model	I-section	Plate	Ω	B	۲ς	Area	I-section	plate	Pod	P_0/P_{u2}^c	Pfca/Pry	P_{ri}/P_{ry}^{8}	P_{r2}/P_{ry}^{h}	$P_{\rm rcl}/P_{\rm ry}^{\ i}$	$P_{rc2}/P_{ry}^{\ \ j}$
Š.						(mm ²)	(MPa)	(MPa)	(kN)						
Ξ	(2)	(3)	₹	3	9	6	(8)	6	(10)	(11)	(12)	(13)	(14)	(15)	(91)
69	W310x179	290x25	뜨	S	1.5	37300	230	350	1075	9.0	0.37	0.41	0.35	0.42	0.36
20	W310x179	290x25	Ţ,	S	1.5	37300	230	350	717	0.4	0.39	0.41	0.35	0.42	0.36
51	W310x179	350x25	Ö	≱	0.4	40300	300	300	3353	9.0	0.95	0.98	0.92	0.99	0.94
52	W310x179	350x25	Ö	≱	0.4	40300	300	300	2235	0.4	96.0	0.98	0.92	0.99	0.94
53	W310x179	350x25	ŋ	≱	0.4	40300	230	350	2662	9.0	96'0	0.98	0.92	0.99	0.94
54	W310x179	350x25	Ö	≱	0.4	40300	230	350	1775	0.4	96.0	0.98	0.92	0.09	0.94
55	W310x179	350x25	Ö	≩	1.1	40300	300	300	11511	9.0	0.71	99.0	0.54	99.0	0.54
26	W310x179	350x25	Ö	≱	1:1	40300	300	300	1151	9.0	0.64	99.0	0.54	99.0	0.54
	W310x179	350x25	ŋ	≩	Ξ:	40300	300	300	1151	9.0	0.59	0.68	0.54	99.0	0.54
چ 178	W310x179	350x25	Ö	≱	Ξ	40300	300	300	1151	9.0	0.60	0.68	0.54	99.0	0.54
	W310x179	350x25	ŋ	≱	Ξ	40300	300	300	992	0.4	0.61	0.68	0.54	99.0	0.54
3	W310x179	350x25	ŋ	≩	Ξ	40300	230	350	1026	9.0	0.61	99.0	0.54	99.0	0.54
19	W310x179	350x25	ŋ	≱	<u> </u>	40300	230	350	684	0.4	0.63	0.68	0.54	99.0	0.54
62	W310x179	350x25	ŋ	≩	1.5	40300	300	300	899	9.0	0.38	0.41	0.35	0.42	0.36
63	W310x179	350x25	ŋ	≩	1.5	40300	300	300	445	0.4	0.40	0.41	0.35	0.42	0.36
3	W310x179	350x25	Ö	≩	1.5	40300	230	350	622	9.0	0.39	0.41	0.35	0.42	0.36
a) D - (a	a) D - Orientation of reinforcing plates	reinforcing	g plate	Š		F - Paral	F - Parallel to the flanges	ınges			G - Paral	G - Parallel to the web	web		
F. B. E.	b) B - Buckling axis of the reinforced column	of the rein	forced	Colum	5	W - Wea	W - Weak axis of the rolled section	e rolled s	ection		S - Stron	S - Strong axis of the rolled section	he rolled	section	

c) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load for the finite element analysis d) P₀ - Pre-load D) B - Buckling axis of the reinforced column

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) $\mathbf{P}_{\mathbf{ry}}$ - Yield strength of the reinforced column

g) P_{r1} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) $P_{\it t2}$ - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) P_{rcl} - Load carrying capacity of the reinforced column (predicted using the CSA curve 1) P_{rc2} - Load carrying capacity of the reinforced column (predicted using the CSA curve 2)

Table B.1 (cont'd)

FEA							Yield Strength	rength	Pre	Preload					
model	1-section	Plate	7	Bp	ئر	Area	I-section	plate	P_0^d	P_0/P_{u2}^c	Pfca/Pry	$P_{rl}/P_{ry}^{\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ $	$P_{r2}/P_{ry}^{\ \ h}$	P_{rel}/P_{ry}^{i}	$P_{1c2}/P_{1y}^{\ \ j}$
Š.						(mm ²)	(MPa)	(MPa)	(kN)						0 0 0 0 0 0 0
Ξ	(2)	(3)	€	(5)	9	(2)	æ	6	(10)	(11)	(12)	(13)	(14)	(15)	(16)
\$9	W310x179	350x25	g	≥	1.5	40300	230	350	414	0.4	0.40	0.41	0.35	0.45	0.36
9	W310x179	350x25	Ö	S	0.4	40300	300	300	3840	9.0	0.93	0.98	0.92	0.99	0.94
<i>L</i> 9	W310x179	350x25	Ö	S	0.4	40300	300	300	2560	0.4	0.94	0.98	0.92	0.09	0.94
89	W310x179	350x25	Ö	S	0.4	40300	230	350	2982	9.0	0.91	0.98	0.92	0.00	0.94
69	W310x179	350x25	Ö	S	0.4	40300	230	350	1988	0.4	0.92	0.98	0.92	0.99	0.94
70	W310x179	350x25	Ö	S	Ξ	40300	300	300	2562	9.0	09.0	89.0	0.54	99.0	0.54
71	W310x179	350x25	Ö	S	=	40300	300	300	2562	9.0	0.57	89.0	0.54	99.0	0.54
72	W310x179	350x25	Ö	S	=	40300	300	300	2562	9.0	0.54	89.0	0.54	99.0	0.54
73	W310x179	350x25	Ö	S	=	40300	300	300	2562	9.0	0.54	89.0	0.54	99:0	0.54
동 179	W310x179	350x25	Ö	S	=	40300	300	300	1708	0.4	0.57	89.0	0.54	99.0	0.54
75	W310x179	350x25	Ö	သ	-:	40300	230	350	2149	9.0	0.48	89.0	0.54	99.0	0.54
9/	W310x179	350x25	Ö	S	Ξ	40300	230	350	1433	0.4	0.51	89.0	0.54	99.0	0.54
11	W310x179	350x25	Ö	S	1.5	40300	300	300	1699	9.0	0.37	0.41	0.35	0.42	0.36
78	W310x179	350x25	Ö	S	1.5	40300	300	300	1132	0.4	0.38	0.41	0.35	0.45	0.36
79	W310x179	350x25	Ö	S	1.5	40300	230	350	1498	9.0	0.35	0.41	0.35	0.42	0.36
80	W310x179	350x25	Ð	S	1.5	40300	230	350	666	0.4	0.36	0.41	0.35	0.42	0.36
a) D - (a) D - Orientation of reinforcing plates	reinforcing	g plate	S		F - Parall	F - Parallel to the flanges	anges			G - Parall	G - Parallel to the web	web		
h) R - F	b) B - Buckling axis of the reinforced column	of the rein	forced	colum	9	W - Wea	- Weak axis of th	the rolled s	section		S - Strong	- Strong axis of the rolled	he rolled	section	
						:			•	-		:	44 - 11 - 1		6

c) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load for the finite element analysis

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) $P_{\rm ry}$ - Yield strength of the reinforced column d) P₀ - Pre-load

g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) $P_{\rm r2}$ - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

									,						
FEA	Ą						Yield Strength	rength	Preload	oad					
mo	model I-section	Plate	<u>م</u>	B	ېر	Area	I-section	plate	P_0^d	P_0/P_{u2}^c	Pfca/Pry	$P_{rl}/P_{ry}^{\ \ B}$	P_{r2}/P_{ry}^{b}	P_{rcl}/P_{ry}^{l}	$P_{rc2}/P_{ry}^{\ j}$
Š	Ġ					(mm ²)	(MPa)	(MPa)	(KN)			• • • • • • • • • • • • • • • • • • •			•
٦	(2)	(3)	€	(5)	9	(7)	(8)	(6)	(10)	(11)	(12)	(13)	(14)	(15)	(91)
∞	1 W310x179	350x25	9	S	=	40300	300	300	2562	9.0	0.54	89.0	0.54	99.0	0.54
òò	2 W310x179	350x25	Ö	S	Ξ:	40300	300	300	2562	9.0	0.54	99.0	0.54	99.0	0.54
œ	3 W310x179	350x25	ŋ	S	Ξ:	40300	300	300	2562	9.0	0.54	99.0	0.54	99.0	0.54
ò	4 W310x179	290x16	<u> </u>	≱	0.4	32080	300	300	3764	9.0	0.98	0.98	0.92	0.99	0.94
œ	S W310x179	290x16	Ľ	≱	0.4	32080	300	300	2509	0.4	0.98	0.98	0.92	0.99	0.94
98	6 W310x179	290x16	Œ,	≩	0.4	32080	230	350	2918	9.0	0.95	0.98	0.92	0.99	0.94
òò	7 W310x179	290x16	Ľ	≱	0.4	32080	230	350	1945	9.4	96.0	0.98	0.92	0.99	0.94
- 80 - 80			Œ,	≱	Ξ	32080	300	300	2168	9.0	0.64	89.0	0.54	99.0	0.54
	_		Ľ,	≱	Ξ	32080	300	300	2168	9.0	09.0	99.0	0.54	99.0	0.54
180 S	_		Œ,	≱	1:1	32080	300	300	2168	9.0	0.56	99.0	0.54	99.0	0.54
) ತ	I W310x179		Œ	≯		32080	300	300	2168	9.0	0.57	89.0	0.54	99.0	0.54
6	2 W310x179	-	ĹĽ,	≯	Ξ	32080	300	300	1445	0.4	0.58	89.0	0.54	99.0	0.54
6	3 W310x179	290x16	Œ	≯	Ξ	32080	230	350	1825	9.0	0.53	0.68	0.54	99.0	0.54
8	4 W310x179	290x16	Ľ	≱	1.1	32080	230	350	1217	0.4	0.54	89.0	0.54	99.0	0.54
5,	5 W310x179	290x16	Œ	≱	1.5	32080	300	300	1412	9.0	0.36	0.41	0.35	0.42	0.36
96	6 W310x179	290x16	Ľ.	*	1.5	32080	300	300	1412	9.0	0.36	0.41	0.35	0.42	0.36
a) D	a) D - Orientation of reinforcing plates	f reinforcing	g plate	ş		F - Paral	F - Parallel to the flanges	ınges			G - Paral	G - Parallel to the web	web		
4	b) B - Buckling axis of the reinforced colum	of the rein	forced	Colum	5	W - Wea	- Weak axis of the rolled section	e rolled se	ction		S - Stron	- Strong axis of the rolled		section	
	TOWNING AVIS					; :						,		Jass	(6,000

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) $P_{\rm ry}$ - Yield strength of the reinforced column f) P_{fea} - Load carrying capacity obtained from the finite element analysis d) P₀ - Pre-load c) λ - Slenderness parameter.

g) P₁₁ - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1) h) $P_{\it t2}$ - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

model section Plate D' B' λ Area section plate Podel Pyden Pyden	V CIE							Viold C	trongth	Drs	Peop					
model 1-section Plate D° B° A° Po_Pa_2 Po_Pa_2 Po_Pa_2 Pi_Pa_3	Z Z			:		,			u Cuigui	<u>: </u>						
No. (1) (2) (3) (4) (5) (6) (7) (8) (9) (10) (11) (12) (13) (14) (15) (15) (19) (11) (12) (13) (14) (15) (15) (19) (11) (12) (13) (14) (15) (15) (19) (11) (12) (13) (14) (15) (15) (19) (19) (11) (12) (13) (14) (15) (15) (19) (19) (19) (19) (11) (12) (13) (14) (15) (15) (14) (15) (15) (15) (15) (15) (15) (15) (15	model		Plate	<u>"</u>	æ	×	Area	I-section	plate	\mathbf{P}_0	P_0/P_{u2}^c					
(1) (2) (3) (4) (5) (6) (7) (8) (9) (10) (11) (12) (13) (14) (15) (15) (19) (19) (19) (19) (19) (19) (19) (19	N O						(mm ²)	(MPa)	(MPa)	(k R						
97 W310k179 290k16 F W 1.5 32080 300 941 0.4 0.37 0.41 0.35 0.42 98 W310k179 290k16 F W 1.5 32080 230 350 1196 0.6 0.34 0.41 0.35 0.42 99 W310k179 290k16 F W 1.5 32080 230 350 797 0.4 0.35 0.41 0.35 0.42 100 W310k179 290k16 F S 0.4 32080 300 3724 0.6 0.94 0.98 0.92 0.99 101 W310k179 290k16 F S 0.4 32080 300 300 2482 0.4 0.98 0.92 0.99 102 W310k179 290k16 F S 0.4 32080 300 300 2482 0.4 0.98 0.92 0.99 0.99 0.99 0.99 <	Ξ	(2)	(3)	<u>4</u>	3	9	6	8	6	(10)	(11)	(12)	(13)	(14)	(15)	(16)
98 W310x179 290x16 F W 1.5 32080 230 350 1196 0.6 0.34 0.41 0.35 0.42 99 W310x179 290x16 F W 1.5 32080 230 350 797 0.4 0.35 0.41 0.35 0.42 100 W310x179 290x16 F S 0.4 32080 300 3724 0.6 0.94 0.98 0.92 0.99 101 W310x179 290x16 F S 0.4 32080 300 2482 0.4 0.98 0.92 0.99 102 W310x179 290x16 F S 0.4 32080 300 1999 0.6 0.94 0.98 0.92 0.99 103 W310x179 290x16 F S 1.1 32080 300 1999 0.6 0.94 0.98 0.92 0.99 104 W310x179 290x16	97	W310x179	290x16	ഥ	≥	1.5	32080	300	300	941	0.4	0.37	0.41	0.35	0.42	0.36
99 W310x179 290x16 F W 1.5 32080 230 350 797 0.4 0.35 0.41 0.35 0.42 100 W310x179 290x16 F S 0.4 32080 300 300 2482 0.4 0.94 0.98 0.92 0.99 101 W310x179 290x16 F S 0.4 32080 300 300 2482 0.4 0.94 0.99 0.09 0.99 0.99 0.09 <	86	W310x179	290x16	ĬŢ,	≩	1.5	32080	230	350	1196	9.0	0.34	0.41	0.35	0.42	0.36
100 W310x179 290x16 F S 0.4 32080 300 3724 0.6 0.94 0.98 0.92 0.99 101 W310x179 290x16 F S 0.4 32080 300 300 2482 0.4 0.94 0.98 0.92 0.99 102 W310x179 290x16 F S 0.4 32080 230 350 1927 0.4 0.94 0.98 0.92 0.99 103 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.69 0.88 0.92 0.99 104 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.69 0.68 0.54 0.66 105 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.69 0.68 0.54 0.66 <	8	W310x179	290x16	Œ	≱	1.5	32080	230	350	797	0.4	0.35	0.41	0.35	0.42	0.36
101 W310x179 290x16 F S 0.4 32080 300 360 2482 0.4 0.94 0.98 0.92 0.99 102 W310x179 290x16 F S 0.4 32080 230 350 2890 0.6 0.93 0.98 0.92 0.99 103 W310x179 290x16 F S 0.4 32080 230 360 1999 0.6 0.69 0.68 0.54 0.66 109 104 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.64 0.68 0.54 0.66 105 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.61 0.68 0.54 0.66 107 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.61 0.68 0.54 0.66 107 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.61 0.68 0.54 0.66 109 W310x179 290x16 F S 1.1 32080 300 300 1333 0.4 0.63 0.68 0.54 0.66 110 W310x179 290x16 F S 1.1 32080 300 300 1295 0.6 0.57 0.68 0.54 0.66 110 W310x179 290x16 F S 1.1 32080 300 300 1295 0.6 0.57 0.68 0.54 0.66 111 W310x179 290x16 F S 1.5 32080 300 300 1295 0.6 0.39 0.41 0.35 0.42 0.42 111 W310x179 290x16 F S 1.5 32080 300 300 1295 0.6 0.39 0.41 0.35 0.42 0.42 0.44 0.44 0.44 0.44 0.44 0.44	8	W310x179	290x16	Ľ	S	0.4	32080	300	300	3724	9.0	0.94	0.98	0.92	0.99	0.94
102 W310x179 290x16 F S 0.4 32080 230 350 2890 0.6 0.93 0.98 0.92 0.99 103 W310x179 290x16 F S 0.4 32080 300 390 169 0.6 0.69 0.68 0.54 0.69 104 W310x179 290x16 F S 1.1 32080 300 1999 0.6 0.69 0.68 0.54 0.66 105 W310x179 290x16 F S 1.1 32080 300 1999 0.6 0.61 0.68 0.54 0.66 107 W310x179 290x16 F S 1.1 32080 300 1999 0.6 0.63 0.68 0.54 0.66 108 W310x179 290x16 F S 1.1 32080 300 300 123 0.65 0.63 0.68 0.54 0.66 110	101	W310x179	290x16	Ľ	S	0.4	32080	300	300	2482	0.4	0.94	0.98	0.92	0.99	0.94
103 W310x179 290x16 F S 0.4 32080 350 1927 0.4 0.94 0.98 0.92 0.99 104 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.69 0.68 0.54 0.66 105 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.64 0.68 0.54 0.66 107 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.61 0.68 0.54 0.66 108 W310x179 290x16 F S 1.1 32080 300 300 1333 0.4 0.63 0.68 0.54 0.66 109 W310x179 290x16 F S 1.1 32080 300 300 1295 0.6 0.57 0.68 0.54 0.66 <	102	W310x179	290x16	ï	S	0.4	32080	230	350	2890	9.0	0.93	0.98	0.92	0.99	0.94
104 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.69 0.68 0.54 0.66 105 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.64 0.68 0.54 0.66 106 W310x179 290x16 F S 1.1 32080 300 1999 0.6 0.61 0.68 0.54 0.66 107 W310x179 290x16 F S 1.1 32080 300 1685 0.6 0.57 0.68 0.54 0.66 110 W310x179 290x16 F S 1.1 32080 300 1685 0.6 0.57 0.68 0.54 0.66 110 W310x179 290x16 F S 1.1 32080 300 300 1295 0.6 0.57 0.68 0.54 0.66 111	103	W310x179	290x16	Ľ	S	0.4	32080	230	350	1927	0.4	0.94	0.98	0.92	0.99	0.94
105 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.64 0.68 0.54 0.66 106 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.61 0.68 0.54 0.66 107 W310x179 290x16 F S 1.1 32080 300 300 1939 0.6 0.62 0.68 0.54 0.66 109 W310x179 290x16 F S 1.1 32080 330 350 1685 0.6 0.57 0.68 0.54 0.66 110 W310x179 290x16 F S 1.1 32080 330 300 1295 0.6 0.57 0.68 0.54 0.66 111 W310x179 290x16 F S 1.5 32080 300 300 1295 0.6 0.51 0.68 0.54	3	W310x179	290x16	Œ	S	==	32080	300	300	1999	9.0	0.69	89.0	0.54	99.0	0.54
106 W310x179 290x16 F S 1.1 32080 300 300 1999 0.6 0.61 0.68 0.54 0.66 107 W310x179 290x16 F S 1.1 32080 300 300 1333 0.4 0.63 0.68 0.54 0.66 108 W310x179 290x16 F S 1.1 32080 230 350 1685 0.6 0.57 0.68 0.54 0.66 110 W310x179 290x16 F S 1.1 32080 330 350 1123 0.4 0.59 0.68 0.54 0.66 111 W310x179 290x16 F S 1.5 32080 300 300 1295 0.6 0.39 0.41 0.35 0.42 112 W310x179 290x16 F S 1.5 32080 300 300 863 0.4 0.40 0.41 0.35		W310x179	290x16	Ľ	S		32080	300	300	6661	9.0	0.64	99.0	0.54	99.0	0.54
107 W310x179 290x16 F S 1.1 32080 300 300 1333 0.6 0.62 0.68 0.54 0.66 108 W310x179 290x16 F S 1.1 32080 230 350 1685 0.6 0.57 0.68 0.54 0.66 110 W310x179 290x16 F S 1.1 32080 230 350 1123 0.4 0.59 0.68 0.54 0.66 111 W310x179 290x16 F S 1.5 32080 300 300 1295 0.6 0.39 0.41 0.35 0.42 112 W310x179 290x16 F S 1.5 32080 300 300 863 0.4 0.40 0.41 0.35 0.42 a) D - Orientation of reinforcing plates F - Parallel to the flanges F - Parallel section W- Weak axis of the rolled section S - Strong axis of the rolled section		W310x179	290x16	Ľ	S	=	32080	300	300	6661	9.0	0.61	0.68	0.54	99.0	0.54
1.1 32080 300 300 1333 0.4 0.63 0.68 0.54 0.66 1.1 32080 230 350 1685 0.6 0.57 0.68 0.54 0.66 1.1 32080 230 350 1123 0.4 0.59 0.68 0.54 0.66 1.5 32080 300 300 1295 0.6 0.39 0.41 0.35 0.42 1.5 32080 300 300 863 0.4 0.40 0.41 0.35 0.42 A - Parallel to the flanges G - Parallel to the web		W310x179	290x16	Ľ	S	1:1	32080	300	300	6661	9.0	0.62	99.0	0.54	99.0	0.54
1.1 32080 230 350 1685 0.6 0.57 0.68 0.54 0.66 1.1 32080 230 350 1123 0.4 0.59 0.68 0.54 0.66 1.5 32080 300 300 1295 0.6 0.39 0.41 0.35 0.42 1.5 32080 300 300 863 0.4 0.40 0.41 0.35 0.42 F - Parallel to the flanges G - Parallel to the web S - Strong axis of the rolled section	801	W310x179	290x16	ſZ,	S	Ξ	32080	300	300	1333	0.4	0.63	89.0	0.54	99.0	0.54
1.1 32080 230 350 1123 0.4 0.59 0.68 0.54 0.66 1.5 32080 300 300 1295 0.6 0.39 0.41 0.35 0.42 1.5 32080 300 863 0.4 0.40 0.41 0.35 0.42 F - Parallel to the flanges G - Parallel to the web W - Weak axis of the rolled section S - Strong axis of the rolled section	89	W310x179	290x16	Œ	S	Ξ	32080	230	350	1685	9.0	0.57	89.0	0.54	99.0	0.54
1.5 32080 300 300 1295 0.6 0.39 0.41 0.35 0.42 1.5 32080 300 863 0.4 0.40 0.41 0.35 0.42 F - Parallel to the flanges G - Parallel to the web S - Strong axis of the rolled section	110	W310x179	290x16	Œ	S	Ξ	32080	230	350	1123	0.4	0.59	99.0	0.54	99.0	0.54
F - Parallel to the flanges W - Weak axis of the rolled section S - Strong axis of the rolled section	111	W310x179	290x16	Ľ,	S	1.5	32080	300	300	1295	9.0	0.39	0.41	0.35	0.42	0.36
F - Parallel to the flanges W - Weak axis of the rolled section S - Strong axis of the rolled	112	W310x179	290x16	Ľ	S	1.5	32080	300	300	863	0.4	0.40	0.41	0.35	0.42	0.36
W. Wenk axis of the rolled section S - Strong axis of the rolled	a) D - C	Orientation of	reinforcing	g plate	s		F - Parall	el to the fl	anges	İ		G - Paral	lel to the	web		
	F) D (F	Queting oxie	of the rain	forced	miles	Ę		kaxis of th		ection		S - Stron	g axis of t	he rolled	section	

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) W - Weak axis of the rolled section d) P₀ - Pre-load

 $P_{\rm ry}$ - Yield strength of the reinforced column c) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load f) P_{fea} - Load carrying capacity obtained from the finite element analysis

g) P_{r1} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) $P_{\rm r2}$ - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2) i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

	P_{rc2}/P_{ry}^{j}	•	(16)	0.36	0.36	0.94	0.94	0.94	0.94	0.54	0.54	0.54	0.54	0.54	0.54	0.54	0.36	0.36	0.36		
	Prel/Pry	1	(15)	0.42	0.42	0.99	0.99	0.99	0.99	99.0	99'0	99.0	99.0	99.0	99.0	99.0	0.42	0.42	0.42		section
	Pr2/Pry I		(14)	0.35	0.35	0.92	0.92	0.92	0.92	0.54	0.54	0.54	0.54	0.54	0.54	0.54	0.35	0.35	0.35	ep	rolled
	Pr1/Pry		(13)	0.41	0.41	0.98	0.98	0.98	0.98	89.0	99.0	89.0	99.0	0.68	89.0	89.0	0.41	0.41	0.41	el to the w	Strong axis of the
	P _{fea} /P _{ry}		(12)	0.38	0.39	0.98	0.98	96.0	0.97	9.65	0.61	0.56	0.57	0.58	0.54	0.55	0.37	0.37	0.35	G - Parallel to the web	S - Strong
oad	P_0/P_{u2}^{c}		(11)	9.0	0.4	9.0	0.4	9.0	0.4	9.0	9.0	9.0	9.0	9.4	9:0	9.4	9.0	9.4	9.0		•2
Preload	$\mathbf{P_0}^{\mathbf{q}}$	(kN)	(10)	8601	732	3712	2474	2887	1924	1954	1954	1954	1954	1302	1667	===	1263	842	1086		section
rength	plate	(MPa)	(6)	350	350	300	300	350	350	300	300	300	300	300	350	350	300	300	350	ınges	73
Yield Strength	I-section	(MPa)	(8)	230	230	300	300	230	230	300	300	300	300	300	230	230	300	300	230	- Parallel to the flanges	- Weak axis of the rolled
•	Area	(mm ²)	6	32080	32080	34000	34000	34000	34000	34000	34000	34000	34000	34000	34000	34000	34000	34000	34000	F - Paralle	W - Weak
	ž		9	1.5	1.5	0.4	4.0	0.4	0.4	Ξ	Ξ:	=	=	=	=	Ε:	1.5	1.5	1.5		
	æ		3	S	S	≩	≥	≱	≥	≥	≩	≱	≩	≱	≱	≩	≩	≱	*	S	colum
	מי		€	ഥ	Ľ	Ľ.	Ľ,	Ľ,	Ľ.	Œ,	Ľ	ഥ	Ľ,	Ľ	Ľ	ഥ	Ľ	Ľ	F	g plate	forced
	Plate	,	(3)	290x16	290x16	350x16	350x16	350x16	350x16	350x16	350x16	350x16	reinforcin	of the rein							
	I-section		(2)	W310x179	W310x179	W310x179	W310x179	W310x179	W310x179	W310x179	a) D - Orientation of reinforcing plates	b) B - Buckling axis of the reinforced colur									
FEA	model	No.	(113	114	115	911	117	118	611	120	121	27 182	123	124	125	126	127	128	a) D - Or	b) B - Bu

W - Weak axis of the rolled section d) P₀ - Pre-load b) B - Buckling axis of the reinforced column c) λ - Slenderness parameter.

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2)

 $P_{\rm ry}$ - Yield strength of the reinforced column f) P_{fea} - Load carrying capacity obtained from the finite element analysis

g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1) h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

RR A															
ָבְ בְּ							Yield Strength	trength	Pre	Preload					
model	I-section	Plate	r O	B	γ _c	Area	1-section	plate	$\mathbf{P_0}^{\mathbf{q}}$	P_0/P_{u2}^c	Pfc./Pry	$P_{rl}/P_{ry}^{\ \ B}$	P_{r2}/P_{ry}^{h}	P_{rcl}/P_{ry}^{i}	P_{rc2}/P_{ry}
N O						(mm ²)	(MPa)	(MPa)	(kN)		į				
Ξ	(2)	(3)	₹	(5)	9	6	(8)	6	(10)	(11)	(12)	(13)	(14)	(15)	(91)
129	W310x179	350x16	뜨	≥	1.5	34000	230	350	724	0.4	0.36	0.41	0.35	0.42	0.36
130	W310x179	350x16	Ľ,	S	0.4	34000	300	300	3717	9.0	0.94	0.98	0.92	0.00	0.94
131	W310x179	350x16	Ľ	S	0.4	34000	300	300	2478	0.4	0.94	0.98	0.92	0.99	0.94
132	W310x179	350x16	<u></u>	S	0.4	34000	230	350	2890	9.0	0.93	0.98	0.92	0.99	0.94
133	W310x179	350x16	Ľ	S	0.4	34000	230	350	1927	0.4	0.94	0.98	0.92	0.99	0.94
134	W310x179	350x16	Ľ	S	Ξ	34000	300	300	1973	9.0	89.0	89.0	0.54	99.0	0.54
135	W310x179	350x16	Ľ.	S	1.1	34000	300	300	1973	9.0	0.63	89.0	0.54	99.0	0.54
136	W310x179	350x16	Ľ,	S	Ξ:	34000	300	300	1973	9.0	19.0	89.0	0.54	99.0	0.54
_	W310x179	350x16	Œ,	S	:	34000	300	300	1973	9.0	0.62	99.0	0.54	99.0	0.54
82 183	W310x179	350x16	Ľ,	S	:	34000	300	300	1315	0.4	0.63	89.0	0.54	99.0	0.54
139	W310x179	350x16	Ľ,	S	- :	34000	230	350	1683	9.0	0.56	89.0	0.54	99.0	0.54
140	W310x179	350x16	Ľ,	S	Ξ:	34000	230	350	1122	0.4	0.58	89.0	0.54	99.0	0.54
141	W310x179	350x16	Ľ.	S	1.5	34000	300	300	1276	9.0	0.39	0.41	0.35	0.42	0.36
142	W310x179	350x16	Œ	S	1.5	34000	300	300	851	0.4	0.40	0.41	0.35	0.42	0.36
143	W310x179	350x16	ഥ	S	1.5	34000	230	350	1097	9.0	0.38	0.41	0.35	0.42	0.36
44	W310x179	350×16	Œ	S	1.5	34000	230	350	732	0.4	0.39	0.41	0.35	0.42	0.36
a) D - C	a) D - Orientation of reinforcing plates	reinforcing	plate	s		F - Paral	F - Parallel to the flanges	anges			G - Paral	G - Parallel to the web	web		
b) B - E	b) B - Buckling axis of the reinforced column	of the rein	forced	colum		W - Wea	W - Weak axis of the rolled section	ne rolled s	ection		S - Stron	S - Strong axis of the rolled section	he rolled	section	

c) λ - Slenderness parameter. DUCKIIIB AAIS OF 4

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) d) P₀ - Pre-load

 $P_{\rm ry}$ - Yield strength of the reinforced column f) P_{fea} - Load carrying capacity obtained from the finite element analysis

g) Pr1 - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) $P_{\rm r2}$ - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) P_{rel} - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

FEA	A						Yield Strength	Irength	Pre	Preload					
model	lel I-section	Plate	D	B	۲,	Area	I-section	plate	Pod	P ₀ /P _{u2} °	Pfea/Pry	$P_{rl}/P_{ry}^{\ \ B}$	$P_{r2}/P_{ry}^{\ \ h}$	$P_{\rm rel}/P_{\rm ry}^{\ i}$	$P_{rc2}/P_{ry}^{\ \ j}$
Ž	ند					(mm ²)	(MPa)	(MPa)	(KN)						8 0 0 0 0
Ξ	(2)	(3)	₹	3	9	6	(8)	6	(10)	(11)	(12)	(13)	(14)	(15)	(16)
145	5 W310x179	350x16	O	≥	0.4	34000	300	300	3468	9.0	0.95	0.98	0.92	0.00	0.94
146			Ö	≩	0.4	34000	300	300	2312	0.4	0.97	0.98	0.92	0.99	0.94
147			Ö	≥	0.4	34000	230	350	2721	9.0	0.97	0.98	0.92	0.99	0.94
148	-		Ö	≩	0.4	34000	230	350	1814	0.4	0.97	0.98	0.92	0.99	0.94
149		350x16	Ö	≩	=	34000	300	300	1330	9.0	0.72	99.0	0.54	99.0	0.54
150	_	350x16	Ö	≥	=	34000	300	300	1330	9.0	99.0	99.0	0.54	99.0	0.54
151		350x16	Ö	≱	=	34000	300	300	1330	9.0	09.0	99.0	0.54	99.0	0.54
152		350x16	ŋ	≱	Ξ	34000	300	300	1330	9.0	0.61	99.0	0.54	99.0	0.54
	_	350x16	Ö	≥	Ξ	34000	300	300	887	0.4	0.62	99.0	0.54	99.0	0.54
: Š 184	-	350×16	Ö	≥	Ξ:	34000	230	350	1142	9.0	0.62	89.0	0.54	99.0	0.54
155			Ö	≱	1.1	34000	230	350	762	0.4	0.64	99.0	0.54	99.0	0.54
156			Ö	≥	1.5	34000	300	300	805	9.0	0.40	0.41	0.35	0.42	0.36
157		350x16	Ö	≥	1.5	34000	300	300	536	0.4	0.41	0.41	0.35	0.42	0.36
158	8 W310x179	350x16	Ö	≥	1.5	34000	230	350	718	9.0	0.40	0.41	0.35	0.42	0.36
159	9 W310x179	350x16	Ö	≥	1.5	34000	230	350	479	0.4	0.41	0.41	0.35	0.42	0.36
3	0 W310x179	350x16	Ö	S	0.4	34000	300	300	3824	9.0	0.93	0.98	0.92	0.99	0.94
	a) D - Orientation of reinforcing plates	reinforcin	o nlate	·		F - Parall	- Parallel to the flanges	anges			G - Paral	G - Parallel to the web	web		
2						. :	• • •		-		0		Pollo.	40:000	

S - Strong axis of the rolled section W - Weak axis of the rolled section b) B - Buckling axis of the reinforced column

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) $P_{\rm ry}$ - Yield strength of the reinforced column c) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load f) P_{fea} - Load carrying capacity obtained from the finite element analysis d) P₀ - Pre-load

g) Pr1 - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

FEA	Y.						Yield Strength	trength	Pre	Preload	-				
model	del I-section	on Plate	î O	æ	*ر	Area	I-section	plate	Pod	P_0/P_{u2}^{ϵ}	P _{fcs} /Pry	P_{rl}/P_{ry}^{8}	$P_{r2}/P_{ry}^{\ h}$	P_{rcl}/P_{ry}^{i}	$P_{rc2}/P_{ry}^{\ j}$
S _O	o.					(mm ²)	(MPa)	(MPa)	(kN)					000000000000000000000000000000000000000	
=	(2)	(3)	€	3	9	6	(8)	(6)	(10)	(11)	(12)	(13)	(14)	(15)	(16)
191	51 W310x179	179 350x16	D 9	S	0.4	34000	300	300	2549	0.4	0.94	0.98	0.92	0.99	0.94
162	52 W310x179	179 350x16	9	S	0.4	34000	230	350	2963	9.0	0.89	0.98	0.92	0.99	0.94
91	3 W310x179	179 350x16	D 9	S	0.4	34000	230	350	1975	0.4	0.00	0.98	0.92	0.99	0.94
<u>s</u>		179 350x16	9	S	Ξ	34000	300	300	2480	9.0	0.64	0.68	0.54	99.0	0.54
165	55 W310x179	179 350x16	D 9	S	=	34000	300	300	2480	9.0	0.60	0.68	0.54	99.0	0.54
991	6 W310x179	179 350x16	D 9	S	=	34000	300	300	2480	9.0	0.56	99.0	0.54	99.0	0.54
191	_	•	D 9	S	Ξ	34000	300	300	2480	9.0	0.57	0.68	0.54	99.0	0.54
168		_	D 9	S	=	34000	300	300	1653	0.4	0.59	0.68	0.54	99.0	0.54
691	69 W310x179	179 350x16	9	S	:	34000	230	350	2057	9.0	0.51	0.68	0.54	99.0	0.54
2 185	0 W310x179	179 350x16	9 9	S	1.1	34000	230	350	1371	0.4	0.53	99.0	0.54	99.0	0.54
17	1 W310x179	179 350x16	9 9	S	1.5	34000	300	300	1628	9.0	0.38	0.41	0.35	0.42	0.36
17	2 W310x179	179 350x16	9 9	S	1.5	34000	300	300	9801	0.4	0.39	0.41	0.35	0.42	0.36
17	3 W310x179	179 350x16	9 9	S	1.5	34000	230	350	1393	9.0	0.36	0.41	0.35	0.42	0.36
17.	4 W310x179	179 350x16	9 g	S	1.5	34000	230	350	929	0.4	0.37	0.41	0.35	0.42	0.36
175	75 W150x30	(30 130x5	5 F	≩	0.4	2090	300	300	627	9.0	0.97	0.98	0.92	0.00	0.94
176	6 W150x30	(30 130x5	5 F	≯	0.4	2090	300	300	418	0.4	0.98	0.98	0.92	0.99	0.94
a) D	- Orientatio	a) D - Orientation of reinforcing plates	sing plat	es		F - Paral	F - Parallel to the flanges	anges			G - Paral	G - Parallel to the web	web		
. (4 R	- Buckling	b) B - Buckling axis of the reinforced colum	einforce.	d colu	1	W - Wea	- Weak axis of the rolled	e rolled se	section		S - Stron	Strong axis of the rolled		section	
2	DUTTIL	: A: : : : : : : : : : : : : : : : : :	1		:		:	1						; i	į

W - Weak axis of the rolled section b) B - Buckling axis of the reinforced column

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) d) P₀ - Pre-load

 $P_{\boldsymbol{\eta}}$ - Yield strength of the reinforced column c) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load f) P_{rea} - Load carrying capacity obtained from the finite element analysis

g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2) i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

ı																
1	FEA							Yield Strength	rength	Pre	Preload					
=	model	I-section	Plate	۵	æ	*	Area	I-section	plate	Pod	P_0/P_{u2}^c	$P_{fca}/P_{ry}^{\ \ f}$	$P_{rl}/P_{ry}^{\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ $	$P_{r2}/P_{ry}^{\ \ h}$	$P_{\rm rcl}/P_{\rm ry}^{}$	$P_{rc2}/P_{ry}^{\ \ j}$
	ò N						(mm ²)	(MPa)	(MPa)	(kN)						000000000000000000000000000000000000000
•	Ξ	(2)	(3)	€	(5)	9	(7)	(8)	(6)	(10)	(11)	(12)	(13)	(14)	(15)	(91)
İ	177	W150x30	130x5	ഥ	≱	0.4	2090	230	350	486	9.0	0.95	0.98	0.92	0.99	0.94
	178	W150x30	130x5	Ľ	≩	0.4	2090	230	350	324	0.4	0.95	0.98	0.92	0.99	0.94
	179	W150x30	130x5	Œ	≱	Ξ	5090	300	300	370	9.0	99.0	99.0	0.54	99.0	0.54
	180	W150x30	130x5	Ľ	≱	Ξ	2090	300	300	370	9.0	0.64	0.68	0.54	99:0	0.54
	181	W150x30	130x5	뜨	≱	Ξ	5090	300	300	370	9.0	0.60	0.68	0.54	99.0	0.54
	182	W150x30	130x5	ഥ	≯	Ξ	5090	300	300	370	9.0	0.61	0.68	0.54	99.0	0.54
	183	W150x30	130x5	Ľ.	≥	Ξ	5090	300	300	247	0.4	0.62	99.0	0.54	99.0	0.54
	184	W150x30	130x5	Ľ.,	≱	Ξ	2090	230	350	308	9.0	0.57	89.0	0.54	99.0	0.54
1	185	W150x30	130x5	Ľ	≱	Ξ	5090	230	350	206	0.4	0.58	0.68	0.54	99.0	0.54
186	186	W150x30	130x5	[≱	1.5	5090	300	300	242	9.0	0.37	0.41	0.35	0.42	0.36
	187	W150x30	130x5	Œ	≱	1.5	5090	300	300	191	0.4	0.38	0.41	0.35	0.42	0.36
	188	W150x30	130x5	Ľ.	≱	1.5	5090	230	350	202	9.0	0.36	0.41	0.35	0.42	0.36
	189	W150x30	130x5	Ľ.	≯	1.5	5090	230	350	135	0.4	0.37	0.41	0.35	0.42	0.36
	061	W150x30	130x5	Œ	S	0.4	5090	300	300	621	9.0	0.95	0.98	0.92	0.99	0.94
	161	W150x30	130x5	Œ,	S	0.4	2090	300	300	621	9.0	0.95	0.98	0.92	0.99	0.94
	192	W150x30	130x5	ഥ	S	0.4	2090	300	300	621	9.0	0.95	0.98	0.92	0.99	0.94
B	D-0	a) D - Orientation of reinforcing plates	reinforcin	g plate	y _a		F - Parall	F - Parallel to the flanges	ınges			G - Parall	G - Parallel to the web	web		
9	B - B	b) B - Buckling axis of the reinforced colun	of the rein	forced	colum	- =	W - Weal	Weak axis of the rolled		section		S - Strong	- Strong axis of the rolled		section	,
					1					•						ć

d) P₀ - Pre-load c) λ - Slenderness parameter.

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) $\mathbf{P}_{\mathbf{y}}$ - Yield strength of the reinforced column

g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1) f) P_{fea} - Load carrying capacity obtained from the finite element analysis

h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) P_{rel} - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

									ì						
FEA			[] :				Yield Strength	rength	Preload	oad					
model	I-section	Plate	Τ	Bp	مر	Area	I-section	plate	P ₀ d	P_0/P_{u2}^c	Pfea/Pry	$P_{rl}/P_{ry}^{\ \ B}$	$P_{r2}/P_{ry}^{\ \ h}$	P_{rel}/P_{ry}^{i}	$P_{rc2}/P_{ry}^{\ j}$
No.						(mm ²)	(MPa)	(MPa)	(kN)						
Ξ	(2)	3)	€	(5)	9	6	(8)	6)	(10)	(11)	(12)	(13)	(14)	(15)	(16)
193	W150x30	130x5	ᄕ	S	9.0	2090	300	300	414	0.4	0.95	0.98	0.92	0.99	0.94
194	W150x30	130x5	Œ	S	9.4	5090	230	350	481	9.0	0.94	0.98	0.92	0.99	0.94
195	W150x30	130x5	Ľ,	S	0.4	2090	230	350	321	0.4	0.95	86.0	0.92	0.99	0.94
196	W150x30	130x5	Ľ,	S	Ξ	2090	300	300	341	9.0	0.71	89.0	0.54	99.0	0.54
197	W150x30	130x5	Œ,	S	Ξ	2090	300	300	341	9.0	0.67	89.0	0.54	99.0	0.54
198	W150x30	130x5	Œ	S	Ξ	2090	300	300	341	9.0	0.63	99.0	0.54	99.0	0.54
166	W150x30	130x5	ī	S	-	2090	300	300	341	9.0	0.65	89.0	0.54	99.0	0.54
200	W150x30	130x5	Œ	S	=	5090	300	300	228	0.4	0.64	89.0	0.54	99.0	0.54
	W150x30	130x5	<u> </u>	S	1.1	2090	230	350	285	9.0	0.59	89.0	0.54	99.0	0.54
202 187	W150x30	130x5	<u>17</u>	S	=	2090	230	350	961	0.4	19.0	89.0	0.54	99.0	0.54
	W150x30	130x5	Ľ.	S	1.5	2090	300	300	222	9.0	0.44	0.41	0.35	0.42	0.36
204	W150x30	130x5	Œ	S	1.5	2090	300	300	222	9.0	0.44	0.41	0.35	0.42	0.36
205	W150x30	130x5	Ľ.	S	1.5	2090	300	300	222	9.0	0.44	0.41	0.35	0.42	0.36
206	W150x30	130x5	ני	S	1.5	2090	300	300	148	0.4	0.44	0.41	0.35	0.42	0.36
207	W150x30	130x5	ഥ	S	1.5	2090	230	350	186	9.0	0.42	0.41	0.35	0.42	0.36
208	W150x30	130x5	Ľ.	S	1.5	2090	230	350	124	0.4	0.43	0.41	0.35	0.42	0.36
a) D - C	a) D - Orientation of reinforcing plates	reinforcing	g plate	s		F - Parall	F - Parallel to the flanges	ınges			G - Parall	G - Parallel to the web	web		
b) B - E	b) B - Buckling axis of the reinforced column	of the rein	forced	colum	_	W - Weal	- Weak axis of the rolled section	e rolled se	ction		S - Strong	; axis of t	- Strong axis of the rolled section	section	
•)			,				•	•	,	•		•		ć

W - Weak axis of the rolled section d) P₀ - Pre-load b) B - Buckling axis of the reinforced column c) λ - Slenderness parameter.

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) $P_{\eta y}$ - Yield strength of the reinforced column

f) P_{fea} - Load carrying capacity obtained from the finite element analysis

g) P_{r1} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2) i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

V (1)							Viold Ctronath	ronath	Dra	Drahond					
rea			á		ç			ucugui		neo!		÷			9
model	I I-section	Plate	<u>-</u>	œ	~	Area	I-section	plate	g _o	P_0/P_{u2}^c	P_{fca}/P_{ry}	P_{rl}/P_{ry}^{8}	P_{r2}/P_{ry} "	P_{rel}/P_{ry}	P_{rc2}/P_{ry}
No.						(mm ²)	(MPa)	(MPa)	(kN						
Ξ	(2)	3	€	(5)	9	(2)	(8)	(6)	(10)	(E)	(12)	(13)	(14)	(15)	(91)
209	W150x30	175x5	Ð	≱	0.4	5540	300	300	578	9.0	0.96	0.98	0.92	0.99	0.94
210	W150x30	175x5	ŋ	≩	0.4	5540	300	300	578	9.0	0.96	0.98	0.92	0.99	0.94
211	W150x30	175x5	Ö	≩	0.4	5540	300	300	578	9.0	96.0	0.98	0.92	0.99	0.94
212	W150x30	175x5	Ö	≥	0.4	5540	300	300	386	0.4	0.97	0.98	0.92	0.99	0.94
213	W150x30	175x5	Ö	≥	0.4	5540	230	350	453	9.0	0.0	0.98	0.92	0.99	0.94
214	W150x30	175x5	Ö	≩	0.4	5540	230	350	302	0.4	0.97	0.98	0.92	0.99	0.94
215	W150x30	175x5	Ö	≱	=	5540	300	300	225	9.0	0.72	99.0	0.54	99.0	0.54
216	W150x30	175x5	Ð	≩	=	5540	300	300	225	9.0	99.0	89.0	0.54	99.0	0.54
	W150x30	175x5	Ö	≩	=	5540	300	300	225	9.0	09'0	89.0	0.54	99.0	0.54
78 881 881	W150x30	175x5	Ö	≱	=	5540	300	300	225	9.0	0.61	99.0	0.54	99.0	0.54
	W150x30	175x5	Ö	≱	-:	5540	300	300	150	0.4	0.62	99.0	0.54	99.0	0.54
220	W150x30	175x5	Ö	≩	1.1	5540	230	350	193	9.0	0.63	99.0	0.54	99.0	0.54
221	W150x30	175x5	O	≱	Ξ	5540	230	350	128	0.4	0.65	99.0	0.54	99.0	0.54
222	W150x30	175x5	Ö	≩	1.5	5540	300	300	137	9.0	0.38	0.41	0.35	0.42	0.36
223	W150x30	175x5	ŋ	≥	1.5	5540	300	300	137	9.0	0.38	0.41	0.35	0.42	0.36
224	W150x30	175x5	Ð	*	1.5	5540	300	300	137	9.0	0.38	0.41	0.35	0.42	0.36
a) D -	a) D - Orientation of reinforcing plates	reinforcin	g plate	Š		F - Parall	F - Parallel to the flanges	anges			G - Paral	G - Parallel to the web	web		
. A	Duabling ovie	of the rain	formed	į	ş	W. Wes	. Weak axis of the rolled section	rolled a	Potion		S - Strong	S - Strong axis of the rolled	he rolled	section	
o) D	o) d - duckiiiig axis oi uie iciiiioiceu coini		7075	=		3) A : A	A GAIS C:	5 30101 2							

W - Weak axis of the rolled section d) P₀ - Pre-load b) B - Buckling axis of the reinforced column

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2)

 $P_{\rm ry}$ - Yield strength of the reinforced column c) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load for the finite element analysis

g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1) h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

									,						
FEA							Yield Strength	trength	Pre	Preload					
model	I-section	Plate	<u>"</u> Ω	B	ゾ	Area	I-section	plate	P_0^d	P_0/P_{u2}^c	P_{fca}/P_{ry}^{f}	$P_{rl}/P_{ry}^{\ \ B}$	$P_{r2}/P_{ry}^{\ \ h}$	P_{rcl}/P_{ry}^{i}	$P_{rc2}/P_{ry}^{\ \ j}$
Ż.						(mm ²)		(MPa)	(kN)					0 0 0 0 0 0 0	0
Ξ	(2)	(3)	₹	S	9	6		6	(10)	(11)	(12)	(13)	(14)	(15)	(91)
225	W150x30	175x5	9	≱	1.5	5540	300	300	16	0.4	0.39	0.41	0.35	0.42	0.36
226	W150x30	175x5	Ö	≱	1.5	5540	230	350	122	9.0	0.40	0.41	0.35	0.42	0.36
227	W150x30	175x5	ŋ	≯	1.5	5540	230	350	8	0.4	0.41	0.41	0.35	0.45	0.36
228	W150x30	175x5	Ö	S	0.4	5540	300	300	634	9.0	0.97	0.98	0.92	0.99	0.94
229	W150x30	175x5	Ö	S	0.4	5540	300	300	634	9.0	0.97	0.98	0.92	0.99	0.94
230	W150x30	175x5	Ö	S	0.4	5540	300	300	634	9.0	0.97	0.98	0.92	0.99	0.94
231	W150x30	175x5	Ö	S	0.4	5540	300	300	423	4.0	0.97	0.98	0.92	0.99	0.94
232	W150x30	175x5	9	S	9.4	5540	230	350	491	9.0	0.95	0.98	0.92	0.99	0.94
233	W150x30	175x5	Ö	S	0.4	5540	230	350	327	0.4	0.95	0.98	0.92	0.99	0.94
189 23 34	W150x30	175x5	ŋ	S	Ξ	5540	300	300	406	9.0	0.65	89.0	0.54	99.0	0.54
	W150x30	175x5	ŋ	S	Ξ	5540	300	300	406	9.0	0.60	99.0	0.54	99.0	0.54
236	W150x30	175x5	ŋ	S	Ξ	5540	300	300	406	9.0	0.56	99.0	0.54	99.0	0.54
237	W150x30	175x5	Ö	S	=	5540	300	300	406	9.0	0.58	99.0	0.54	99.0	0.54
238	W150x30	175x5	Ö	S	Ξ:	5540	300	300	270	0.4	0.59	89.0	0.54	99.0	0.54
239	W150x30	175x5	ŋ	S	Ξ	5540	230	350	337	9.0	0.51	99.0	0.54	99.0	0.54
240	W150x30	175x5	Ð	S	1.1	5540	230	350	225	0.4	0.53	0.68	0.54	0.66	0.54
a) D - (a) D - Orientation of reinforcing plates	reinforcin	g plate	Š		F - Paral	- Parallel to the flanges	anges			G - Paral	G - Parallel to the web	web		
` 4	b) B - Buckling axis of the reinforced column	of the rein	forced	miles	Ę	W - We	- Weak axis of the rolled section	rolled se	ction		S - Stron	S - Strong axis of the rolled	the rolled	section	
1 - m (n	DUCKIIIK AAIS		3)		=	3	: : : : : : : : : : : : : : : : : : : :								,

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) d) P₀ - Pre-load

 $P_{\rm ry}$ - Yield strength of the reinforced column c) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load from the finite element analysis

g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1) h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

No. Classicion Plate D ^a B ^b X ^c Area I-section plate P ₀ ^d P ₀ P ₀ ¹ ² P ₀ P ₀ ¹ ² P ₁ P _P ^g P ₁ P _P	<u> </u>							i icia Silcingii	ucuguı	21.2	rreioau	_				
No. (II) (2) (3) (4) (5) (6) (7) (8) (9) (10) (11) (12) (13) (14) 241 W150x30 175x3 G S 1.5 5540 300 266 0.6 0.45 0.41 0.35 242 W150x30 175x3 G S 1.5 5540 300 300 266 0.6 0.42 0.41 0.35 242 W150x30 175x3 G S 1.5 5540 300 300 266 0.6 0.42 0.41 0.35 244 W150x30 175x3 G S 1.5 5540 300 300 177 0.4 0.41 0.35 244 W150x30 175x5 G S 1.5 5540 300 300 177 0.4 0.41 0.35 245 W150x30 175x5 G S 1.5 5540 300	model		Plate	۵	æ	γد	Area	I-section	plate	P_0^d	P_0/P_{u2}°			$P_{r2}/P_{ry}^{\ \ h}$	P_{rcl}/P_{ry}^{ri	P_{rc2}/P_{ry}
(1) (2) (3) (4) (5) (6) (7) (8) (9) (11) (12) (13) (14) 241 W150x30 175x5 G S 1.5 5540 300 300 266 0.6 0.45 0.41 0.35 242 W150x30 175x5 G S 1.5 5540 300 300 266 0.6 0.42 0.41 0.35 243 W150x30 175x5 G S 1.5 5540 300 300 266 0.6 0.42 0.41 0.35 244 W150x30 175x5 G S 1.5 5540 300 300 267 0.6 0.36 0.41 0.35 245 W150x30 175x5 G S 1.5 5540 300 300 277 0.6 0.96 0.94 0.35 245 W150x30 130x8 F W 0.4 5870	Š						(mm ²)	(MPa)	(MPa)	(kN)			:	0 0 0 0		
241 W150x30 175x5 G S 1.5 5540 300 366 0.6 0.45 0.41 0.35 242 W150x30 175x5 G S 1.5 5540 300 300 266 0.6 0.42 0.41 0.35 243 W150x30 175x5 G S 1.5 5540 300 300 266 0.6 0.39 0.41 0.35 244 W150x30 175x5 G S 1.5 5540 300 300 177 0.4 0.41 0.35 245 W150x30 175x5 G S 1.5 5540 300 300 60 0.37 0.41 0.35 246 W150x30 130x8 F W 0.4 5870 300 300 418 0.4 0.35 0.41 0.35 248 W150x30 130x8 F W 0.4 5870 300 300	Ξ	(2)	3	€	(5)	9	6	(8)	(6)	(10)	(E)	(12)	(13)	(14)	(15)	(16)
242 W150x30 175x3 G S 1.5 5540 300 300 266 0.6 0.42 0.41 0.35 243 W150x30 175x5 G S 1.5 5540 300 300 266 0.6 0.39 0.41 0.35 244 W150x30 175x5 G S 1.5 5540 300 300 177 0.4 0.40 0.41 0.35 245 W150x30 175x5 G S 1.5 5540 230 300 177 0.4 0.40 0.41 0.35 245 W150x30 175x5 G S 1.5 5540 230 350 227 0.6 0.39 0.41 0.35 246 W150x30 130x8 F W 0.4 5870 300 300 487 0.6 0.95 0.98 0.92 249 W150x30 130x8 F W 0.	241	W150x30	175x5	g	S	1.5	5540	300	300	266	9.0	0.45	0.41	0.35	0.42	0.36
243 W150x30 175x5 G S 1.5 5540 300 300 266 0.6 0.39 0.41 0.35 244 W150x30 175x5 G S 1.5 5540 300 300 177 0.4 0.40 0.41 0.35 245 W150x30 175x5 G S 1.5 5540 230 350 151 0.4 0.49 0.41 0.35 245 W150x30 175x5 G S 1.5 5540 230 350 151 0.4 0.35 0.41 0.35 247 W150x30 130x8 F W 0.4 5870 300 300 47 0.6 0.95 0.98 0.92 249 W150x30 130x8 F W 0.4 5870 230 350 487 0.6 0.95 0.98 0.92 250 W150x30 130x8 F W 1.1	242	W150x30	175x5	Ö	S	1.5	5540	300	300	266	9.0	0.42	0.41	0.35	0.42	0.36
244 W150x30 175x5 G S 1.5 5540 300 300 177 0.4 0.40 0.41 0.35 245 W150x30 175x5 G S 1.5 5540 230 350 157 0.6 0.37 0.41 0.35 246 W150x30 175x5 G S 1.5 5540 230 350 151 0.4 0.37 0.41 0.35 247 W150x30 130x8 F W 0.4 5870 300 300 418 0.4 0.95 0.98 0.92 249 W150x30 130x8 F W 0.4 5870 300 350 487 0.6 0.95 0.98 0.92 250 W150x30 130x8 F W 0.4 5870 300 300 371 0.6 0.95 0.98 0.94 251 W150x30 130x8 F W 1.	243	W150x30	175x5	Ö	S	1.5	5540	300	300	266	9.0	0.39	0.41	0.35	0.42	0.36
245 W150x30 175x5 G S 1.5 5540 230 350 227 0.6 0.37 0.41 0.35 246 W150x30 175x5 G S 1.5 5540 230 350 151 0.4 0.38 0.41 0.35 247 W150x30 130x8 F W 0.4 5870 300 300 418 0.4 0.98 0.92 249 W150x30 130x8 F W 0.4 5870 230 350 487 0.6 0.95 0.98 0.92 250 W150x30 130x8 F W 0.4 5870 300 371 0.6 0.95 0.98 0.92 251 W150x30 130x8 F W 1.1 5870 300 371 0.6 0.59 0.68 0.54 252 W150x30 130x8 F W 1.1 5870 300 30	244	W150x30	175x5	g	S	1.5	5540	300	300	177	0.4	0.40	0.41	0.35	0.42	0.36
246 W150x30 175x5 G S 1.5 5540 230 350 151 0.4 0.38 0.41 0.35 247 W150x30 130x8 F W 0.4 5870 300 300 627 0.6 0.96 0.98 0.92 248 W150x30 130x8 F W 0.4 5870 230 300 418 0.0 0.99 0.98 0.92 249 W150x30 130x8 F W 0.4 5870 230 350 325 0.4 0.95 0.98 0.92 250 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.95 0.98 0.92 251 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 252 W150x30 130x8 F W 1.	245	W150x30	175x5	ŋ	S	1.5	5540	230	350	227	9.0	0.37	0.41	0.35	0.42	0.36
247 W150x30 130x8 F W 6.4 5870 300 300 627 0.6 0.96 0.98 0.92 248 W150x30 130x8 F W 0.4 5870 300 300 418 0.4 0.97 0.98 0.92 249 W150x30 130x8 F W 0.4 5870 230 350 325 0.4 0.95 0.98 0.92 250 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.68 0.58 0.54 251 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 252 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 254 W150x30 130x8 F W 1.	246	W150x30	175x5	ŋ	S	1.5	5540	230	350	151	9.4	0.38	0.41	0.35	0.42	0.36
248 W150x30 130x8 F W 6.4 5870 300 300 418 0.4 0.97 0.98 0.92 249 W150x30 130x8 F W 0.4 5870 230 350 487 0.6 0.95 0.98 0.92 250 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.68 0.68 0.54 251 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.68 0.54 253 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 254 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 255 W150x30 130x8 F W 1.1 587	247	W150x30	130x8	ഥ	≱	0.4	5870	300	300	627	9.0	96.0	0.98	0.92	0.99	0.94
249 W150x30 130x8 F W 6.4 5870 230 350 487 0.6 0.95 0.98 0.92 250 W150x30 130x8 F W 0.4 5870 230 350 371 0.6 0.95 0.98 0.92 251 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.64 0.68 0.54 253 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 254 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 255 W150x30 130x8 F W 1.1 5870 300 300 247 0.4 0.61 0.68 0.54 255 W150x30 130x8 F W 1.	248	W150x30	130x8	Œ,	≥	0.4	5870	300	300	418	0.4	0.97	0.98	0.92	0.99	0.94
250 W150x30 130x8 F W 0.4 5870 230 350 325 0.4 0.95 0.98 0.92 251 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.68 0.58 0.54 252 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 253 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 255 W150x30 130x8 F W 1.1 5870 300 300 247 0.4 0.61 0.68 0.54 255 W150x30 130x8 F W 1.1 5870 330 350 0.6 0.55 0.68 0.54 256 W150x30 130x8 F W 1.1 58		W150x30	130x8	<u> </u>	≱	0.4	5870	230	350	487	9.0	0.95	0.98	0.92	0.99	0.94
251 W150x30 130x8 F W 1.1 5870 300 300 371 0.6 0.68 0.68 0.54 252 W150x30 130x8 F W 1.1 5870 300 370 0.6 0.69 0.68 0.54 253 W150x30 130x8 F W 1.1 5870 300 371 0.6 0.59 0.68 0.54 255 W150x30 130x8 F W 1.1 5870 300 371 0.6 0.59 0.68 0.54 255 W150x30 130x8 F W 1.1 5870 300 370 0.4 0.61 0.68 0.54 256 W150x30 130x8 F W 1.1 5870 230 350 0.6 0.55 0.68 0.54 256 W150x30 130x8 F W 1.1 5870 230 350 0.6 0		W150x30	130x8	<u></u>	≩	0.4	5870	230	350	325	9.0	0.95	0.98	0.92	0.99	0.94
W 1.1 5870 300 300 371 0.6 0.64 0.68 0.54 W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 W 1.1 5870 300 300 247 0.4 0.61 0.68 0.54 W 1.1 5870 330 350 319 0.6 0.55 0.68 0.54 W 1.1 5870 230 350 319 0.6 0.55 0.68 0.54 F- Parallel to the flanges G- Parallel to the web		W150x30	130x8	江	≱	1.1	5870	300	300	371	9.0	89.0	99.0	0.54	99.0	0.54
W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 W 1.1 5870 300 300 247 0.4 0.61 0.68 0.54 W 1.1 5870 230 350 319 0.6 0.55 0.68 0.54 W 1.1 5870 230 350 319 0.6 0.55 0.68 0.54 F- Parallel to the flanges	252	W150x30	130x8	Ľ,	≥		5870	300	300	371	9.0	0.64	89.0	0.54	99.0	0.54
W 1.1 5870 300 300 371 0.6 0.59 0.68 0.54 W 1.1 5870 300 300 247 0.4 0.61 0.68 0.54 W 1.1 5870 230 350 319 0.6 0.55 0.68 0.54 F - Parallel to the flanges	253	W150x30	130x8	Ľ,	≩	=	5870	300	300	371	9.0	0.59	89.0	0.54	99.0	0.54
W 1.1 5870 300 300 247 0.4 0.61 0.68 0.54 W 1.1 5870 230 350 319 0.6 0.55 0.68 0.54 F- Parallel to the flanges G- Parallel to the web	254	W150x30	130x8	Ľ,	≱	1:1	5870	300	300	371	9.0	0.59	99.0	0.54	99.0	0.54
W 1.1 5870 230 350 319 0.6 0.55 0.68 0.54 F - Parallel to the flanges G - Parallel to the web	255	W150x30	130x8	Ľ	≩	Ξ	5870	300	300	247	9.4	0.61	99.0	0.54	99.0	0.54
F - Parallel to the flanges G - Parallel to the web	256	W150x30	130x8	Œ,	≥	1.1	5870	230	350	319	9.0	0.55	0.68	0.54	99.0	0.54
	a) D - C	rientation of	reinforcin	g plate	Š		F - Parall	el to the fl	anges			G - Paral	lel to the	web		
Lyn Derting min of the mindered actions W. Wood pair of the rolled cection C - Grong axis of the rolled section		1	of the same	62000	- 100	9		Lavie of th		uoi jos		C - Stron	o avic of	Ped	section	

c) λ - Sienderness parameter. d) P_0 - Pre-load e) P_{u2} - Load for the finite element analysis d) P₀ - Pre-load

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) P_{ry} - Yield strength of the reinforced column

g) Pr1 - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) $P_{\it l2}$ - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2) i) P_{rel} - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

j) P_{rc2} - Load carrying capacity of the reinforced column (predicted using the CSA curve 2)

Table B.1 (cont'd)

									-						
FEA							Yield Strength	trength	٦ ع	Freioad					
model	I-section	Plate	<u>"</u>	æ	مح	Arca	I-section	plate	$\mathbf{P}_{0}^{\mathbf{q}}$	P_0/P_{u2}°	Pica/Pry	$P_{rl}/P_{ry}^{\ \ B}$	P_{r2}/P_{ry}^{h}	P_{rcl}/P_{ry}^{l}	$P_{rc2}/P_{ry}^{\ j}$
Š.						(mm ²)	(MPa)	(MPa)	(kN)					0 0 0 0 0 0 0	• • • • • • • • • • • • • • • • • • •
(E)	(2)	(3)	€	(5)	9	(2)	(8)	(6)	(10)	(11)	(12)	(13)	(14)	(15)	(16)
257	W150x30	130x8	Ľ	≱	=	5870	230	350	212	0.4	0.57	89.0	0.54	99.0	0.54
258	W150x30	130x8	Œ,	≩	1.5	5870	300	300	242	9.0	0.37	0.41	0.35	0.45	0.36
259	W150x30	130x8	Ľ	≩	1.5	5870	300	300	162	0.4	0.38	0.41	0.35	0.42	0.36
260	W150x30	130x8	Ľ,	≩	1.5	5870	230	350	209	9.0	0.35	0.41	0.35	0.42	0.36
261	W150x30	130x8	íz.	≩	1.5	5870	230	350	140	0.4	0.36	0.41	0.35	0.42	0.36
262	W150x30	130x8	<u>r</u>	S	0.4	5870	300	300	617	9.0	0.94	0.98	0.92	0.00	0.94
263	W150x30	130x8	ᄄ	S	0.4	5870	300	300	412	0.4	0.95	0.98	0.92	0.00	0.94
264	W150x30	130x8	Œ	S	0.4	5870	230	350	481	9.0	0.95	0.98	0.92	0.99	0.94
	W150x30	130x8	Ľ	S	0.4	5870	230	350	320	0.4	0.95	0.98	0.92	0.09	0.94
% 191	W150x30	130x8	Ľ	S	Ξ	5870	300	300	329	9.0	0.72	99.0	0.54	99.0	0.54
	W150x30	130x8	Ľ	S	Ξ:	5870	300	300	329	9.0	0.67	89.0	0.54	99.0	0.54
268	W150x30	130x8	<u> </u>	S	=	5870	300	300	329	9.0	0.63	89.0	0.54	99.0	0.54
569	W150x30	130x8	Œ	S	Ξ	5870	300	300	329	9.0	0.65	99.0	0.54	99.0	0.54
270	W150x30	130x8	뜨	S	=	5870	300	300	219	0.4	0.65	99.0	0.54	99.0	0.54
271	W150x30	130x8	ഥ	S	-:	5870	230	320	282	9.0	0.60	89.0	0.54	99.0	0.54
272	W150x30	130x8	ഥ	S	1:1	5870	230	350	188	0.4	0.62	0.68	0.54	99.0	0.54
a) D - O	a) D - Orientation of reinforcing plates	reinforcin	g plate	S		F - Parall	F - Parallel to the flanges	ınges			G - Paral	G - Parallel to the web	web		
h) R - R	b) B - Buckling axis of the reinforced colur	of the rein	forced	colum	ş	W - Wea	- Weak axis of the rolled	_	section		S - Stron	Strong axis of the	rolled	section	
2 (2)	UCRIIIS HAIN		1		=					,				0.000	ć

c) λ - Slenderness parameter. d) P_0 - Pre-load c) P_{u2} - Load f) P_{tea} - Load carrying capacity obtained from the finite element analysis d) P₀ - Pre-load

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2)

P_{ry} - Yield strength of the reinforced column

g) Pr1 - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1) h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

FEA							Yield Strength	trength	Pre	Preload					
model	I-section	Plate	"	æ	γc	Area	I-section	plate	P_0^d	P_0/P_{u2}^c	Pfcu/Pry	$P_{rl}/P_{ry}^{\ B}$	$P_{r2}/P_{ry}^{\ h}$	P_{rel}/P_{ry}^{i}	$P_{rc2}/P_{ry}^{\ j}$
S.						(mm ²)	(MPa)	(MPa)	(kN)						5 0 0 0 0 8 8
Ξ	(2)	(3)	<u>4</u>	(5)	9	6	(8)	(6)	(10)	(11)	(12)	(13)	(14)	(15)	(10)
273	W150x30	130x8	뜨	S	1.5	5870	300	300	213	9.0	0.44	0.41	0.35	0.42	0.36
274	W150x30	130x8	Œ	S	1.5	5870	300	300	142	0.4	0.45	0.41	0.35	0.42	0.36
275	W150x30	130x8	Ľ.	S	1.5	5870	230	350	184	9.0	0.43	0.41	0.35	0.42	0.36
276	W150x30	130x8	Œ,	S	1.5	5870	230	350	123	0.4	0.44	0.41	0.35	0.45	0.36
777	W150x30	175x8	Ö	≩	0.4	6590	300	300	900	9.0	0.96	0.98	0.92	0.99	0.94
278	W150x30	175x8	ŋ	≩	0.4	6590	300	300	373	0.4	0.96	0.98	0.92	0.99	0.94
279	W150x30	175x8	Ö	≱	0.4	0659	230	350	444	9.0	96.0	0.98	0.92	0.99	0.94
280	W150x30	175x8	Ö	≩	0.4	0659	230	350	296	0.4	0.96	0.98	0.92	0.99	0.94
	W150x30	175x8	Ö	≱	==	9	300	300	961	9.0	0.72	99.0	0.54	99.0	0.54
282 192	W150x30	175x8	Ö	≩	=	6590	300	300	961	9.0	0.65	99.0	0.54	99.0	0.54
	W150x30	175x8	Ö	≩	Ξ:	6590	300	300	961	9.0	0.59	89.0	0.54	99.0	0.54
284	W150x30	175x8	Ö	≥	Ξ	6590	300	300	961	9.0	09.0	89.0	0.54	99.0	0.54
285	W150x30	175x8	Ö	≩	-:	6590	300	300	131	0.4	0.61	99.0	0.54	99.0	0.54
286	W150x30	175x8	Ö	≩	=	6590	230	350	174	9.0	0.62	89.0	0.54	99.0	0.54
287	W150x30	175x8	Ö	≱	Ξ	6590	230	350	911	0.4	0.64	99.0	0.54	99.0	0.54
288	W150x30	175x8	Ö	≩	1.5	6590	300	300	114	9.0	0.38	0.41	0.35	0.42	0.36
a) D - O	a) D - Orientation of reinforcing plates	reinforcin	g plate	دة		F - Paral	F - Parallel to the flanges	anges			G - Paral	G - Parallel to the web	web		
0 0 0	sing sailthea	aios che	60000	mil Oo	<u> </u>	W Wee	Lavie of th		action		S. Stron	Strong axis of the rol	led	section	
a - a ía	D) B - Buckling axis of the femiology colum	ol me ien	22.50		= 1	٠.	- Wear axis of the follow			,			}		6

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) $P_{\boldsymbol{\eta}}$ - Yield strength of the reinforced column f) P_{fea} - Load carrying capacity obtained from the finite element analysis d) P₀ - Pre-load c) λ - Slenderness parameter.

g) P_{r1} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1) h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2)

i) $P_{\rm rel}$ - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

	FEA							Yield Strength	Irength	Pre	Preload					
Ξ	model	I-section	Plate	D _a	B	۲ς	Area	l-section	plate	Pod	P_0/P_{u2}^c	Pfeu/Pry	P_{rl}/P_{ry}^{8}	$P_{r2}/P_{ry}^{\ \ h}$	P_{rcl}/P_{ry}^{l}	$P_{rc2}/P_{ry}^{\ j}$
	Š.						(mm ²)	(MPa)	(MPa)	(kN)						• • • • • • • • •
:	Ξ	(2)	(3)	₹	3	9	6	(8)	(6)	(10)	(11)	(12)	(13)	(14)	(15)	(16)
1,,	289	W150x30	175x8	9	≥	1.5	6590	300	300	9/	0.4	0.39	0.41	0.35	0.42	0.36
. 1	290	W150x30	175x8	ŋ	≥	1.5	9	230	350	901	9.0	0.40	0.41	0.35	0.42	0.36
. 1	167	W150x30	175x8	ŋ	≥	1.5	6590	230	350	7.1	0.4	0.41	0.41	0.35	0.42	0.36
	292	W150x30	175x8	Ö	S	0.4	6590	300	300	636	9.0	0.98	0.98	0.92	0.99	0.94
	293	W150x30	175x8	Ö	S	0.4	0629	300	300	424	0.4	0.98	0.98	0.92	0.99	0.94
	994	W150x30	175x8	Ö	S	0.4	6590	230	350	494	9.0	0.97	0.98	0.92	0.99	0.94
(1	295	W150x30	175x8	Ö	S	0.4	6590	230	350	330	0.4	0.97	0.98	0.92	0.99	0.94
(4	296	W150x30	175x8	ŋ	S	-:	6590	300	300	419	9.0	0.62	0.68	0.54	99.0	0.54
	767	W150x30	175x8	Ö	S	-:	6590	300	300	419	9.0	0.58	0.68	0.54	99.0	0.54
193	298	W150x30	175x8	ŋ	S	=	6590	300	300	419	9.0	0.55	0.68	0.54	99.0	0.54
	299	W150x30	175x8	ŋ	S	Ξ	6590	300	300	419	9.0	0.57	99.0	0.54	99.0	0.54
(~)	300	W150x30	175x8	g	S	=	6590	300	300	280	0.4	0.58	99.0	0.54	99.0	0.54
(4)	301	W150x30	175x8	Ö	S	Ξ	6590	230	350	352	9.0	0.49	99.0	0.54	99.0	0.54
. (~1	302	W150x30	175x8	Ö	S	Ξ:	6590	230	350	235	0.4	0.52	0.68	0.54	99.0	0.54
G)	303	W150x30	175x8	Ö	S	1.5	6590	300	300	277	9.0	0.38	0.41	0.35	0.42	0.36
(-)	304	W150x30	175x8	Ö	S	1.5	6590	300	300	184	0.4	0.39	0.41	0.35	0.42	0.36
a	0-0	a) D - Orientation of reinforcing plates	reinforcin	g plate	Š.		F - Parall	F - Parallel to the flanges	anges	į		G - Paral	G - Parallel to the web	web		
1		, of the sail of	of the rain	forced			W. Wen	W. West axis of the rolled section	e rolled or	rtion		S - Stron	S. Strong axis of the rolled section	he rolled	section	
õ	-	o) B - Buckling axis of the femiology column		フン 5 5			30 AA - AA	11 10 CIVE 4	らりこうこと	553		֖֭֭֓֞֝֞֜֜֝֞֝֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֡֓֓֓֡֓֓֓֓֡֓֡֓֡֓֡֓֡֓֡֓֡		11::::		

b) B - Buckling axis of the reinforced column M - M = M axis of the following c) λ - Slendemess parameter. (a) $R_0 - R$ - Pre-load (b) $R_0 - R$ - Load carrying (c)

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) nent analysis P_{ry} - Yield strength of the reinforced column

f) P_{fea} - Load carrying capacity obtained from the finite element analysis

g) Pri - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2) i) Prel - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Table B.1 (cont'd)

FEA							Yield Strength	rength	Pre	Preload					
model	model I-section	Plate	<u>م</u>	B _p	չ	Area	I-section	plate	P_0^d	P_0/P_{u2}^c	$P_{fca}/P_{ry}^{\ f}$	$P_{rl}/P_{ry}^{\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ $	$P_{r2}/P_{ry}^{\ \ h}$	P _{ICI} /P _{IY}	$P_{rc2}/P_{ry}^{\ \ j}$
No.						(mm ²)	(MPa)	(MPa)	(kN)						1 2 0 0 0
Ξ	(2)	(3)	€	(5)	9	6	(8)	<u>6</u>	(10)	(11)	(12)	(13)	(14)	(15)	(16)
305	W150x30	175x8	g	S	1.5	6590	230	350	244	9.0	0.35	0.41	0.35	0.42	0.36
306	W150x30	175x8	Ö	S	1.5	6590	230	350	162	0.4	0.37	0.41	0.35	0.45	0.36
307	W310x179	350x16	Ö	≯	0.4	34000	300	300	3468	9.0	0.97	0.98	0.92	0.99	0.94
308	W310x179	350x16	Ö	≯	:	34000	300	300	1330	9.0	0.62	0.68	0.54	99.0	0.54
300	W310x179	350x16	g	S	Ξ	34000	300	300	2480	9.0	0.59	99.0	0.54	99.0	0.54
310	W310x179	350x16	Ľ.	S	1.5	34000	300	300	1276	9.0	0.41	0.41	0.35	0.42	0.36
311	W150x30	130x5	Ľ	S	Ξ:	2090	300	300	341	9.0	0.63	99.0	0.54	99.0	0.54
	W150x30	130x5	Ľ,	S	1.5	2090	300	300	222	9.0	0.44	0.41	0.35	0.42	0.36
EIE 94	W310x179	350x25	Ö	≥	0.4	40300	300	300	3353	9.0	96.0	0.98	0.92	0.00	0.94
314	W310x179	350x25	Ö	≥	Ξ:	40300	300	300	1151	9.0	0.63	99.0	0.54	99.0	0.54
315	W310x179	350x25	Ö	≥	1.5	40300	300	300	899	9.0	0.41	0.41	0.35	0.42	0.36
316	W310x179	350x16	g	≯	1.1	34000	300	300	1330	9.0	99.0	99.0	0.54	99.0	0.54
317	W310x179	350x16	g	≯	1.5	34000	300	300	805	9.0	0.42	0.41	0.35	0.99	0.36
a) D - C	a) D - Orientation of reinforcing plates	reinforcing	g plate	S		F - Parall	F - Parallel to the flanges	anges			G - Paral	G - Parallel to the web	web		
F) D	Anothing avie	of the rain	forced.		2	W . Wea	. Weak axis of the rolled section	e rolled s	ection		S - Stron	S - Strong axis of the rolled	he rolled	section	
1 - 0 (0	o) b - bucking axis of the fellioleca coin		3015		=		1 10 01VB 4	2 20101							

e) P_{u2} - Load carrying capacity of the I-section (predicted using the SSRC curve 2) c) λ - Slenderness parameter. d) P_0 - Pre-load e) P_{u2} - Load f) P_{tca} - Load carrying capacity obtained from the finite element analysis d) P₀ - Pre-load

g) P_{rl} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 1)

h) P_{r2} - Load carrying capacity of the reinforced column (predicted using the SSRC curve 2) i) P_{rc1} - Load carrying capacity of the reinforced column (predicted using the CSA curve 1)

Appendix C

Statistical Analysis Data for the Professional Factors for the Columns from Group 2

Statistical Analysis Data for the Professional Factors for the Columns from Group 2

This appendix serves as a supplement to Chapter 5. It presents the statistical analysis data used to obtain the professional factors for the columns from group 2 (columns reinforced with plates parallel to the flanges and buckling about the weak axis of the rolled section and columns reinforced with plates parallel to the web and buckling about the strong axis of the rolled section). The statistical analysis procedures for the columns from group2 are same as those from group 1 presented in Chapter 5.

Tables C.1 to C.3 present the analysis data for the simulated professional factors for columns from group 2 for values of the slenderness ratio, λ , of 0.4, 1.1 and 1.5 respectively. Tables C.4 to C.6 present the analysis data for the normalized professional factors for columns from group 2 for values of the slenderness ratio, λ , of 0.4, 1.1 and 1.5 respectively.

Plots of the simulated professional ratio, ρ_s , versus out-of-straightness for columns from group 2 for values of the slenderness ratio, λ , of 0.4, 1.1 and 1.5 are presented in Figures C.1, C.2 and C.3 respectively. The normalized professional ratio for l = 0.4, 1.1, and 1.5 are plotted in Figures C.4, C.5, and C.6 respectively.

Table C.1 Simulated Professional Factors for Columns form Group 2 ($\lambda = 0.4$)

FEA			Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0		(P_{fea}/P_{rl})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
17	F	W	L/8000	1.005	1.025	1.094	1.012	1.069
18	F	W	L/2000	0.987	1.007	1.074	0.994	1.049
19	F	W	L/1000	0.973	0.993	1.059	0.980	1.035
20	F	W	L/1000	0.978	0.997	1.064	0.985	1.040
21	F	W	L/1000	0.958	0.978	1.043	0.965	1.019
22	F	W	L/1000	0.961	0.981	1.047	0.968	1.022
66	G	S	L/1000	0.934	0.953	1.017	0.941	0.993
67	G	S	L/1000	0.940	0.959	1.023	0.947	1.000
68	G	S	L/1000	0.911	0.930	0.992	0.918	0.969
69	G	S	L/1000	0.917	0.935	0.998	0.923	0.975
84	F	W	L/1000	0.977	0.996	1.063	0.984	1.038
85	F	W	L/1000	0.980	1.000	1.067	0.987	1.042
86	F	W	L/1000	0.952	0.971	1.036	0.959	1.012
87	F	W	L/1000	0.956	0.975	1.040	0.963	1.016
115	F	W	L/1000	0.980	1.000	1.067	0.987	1.042
116	F	W	L/1000	0.983	1.003	1.070	0.990	1.045
117	F	W	L/1000	0.963	0.983	1.049	0.971	1.025
118	F	W	L/1000	0.966	0.986	1.051	0.973	1.027
160	G	S	L/1000	0.929	0.948	1.011	0.936	0.988
161	G	S	L/1000	0.937	0.956	1.020	0.944	0.996
162	G	S	L/1000	0.894	0.912	0.973	0.900	0.950
163	G	S	L/1000	0.902	0.920	0.981	0.908	0.959
175	F	W	L/1000	0.972	0.992	1.058	0.979	1.033
176	F	W	L/1000	0.978	0.998	1.065	0.986	1.040
177	F	W	L/1000	0.947	0.966	1.031	0.954	1.007
178	F	W	L/1000	0.952	0.971	1.036	0.959	1.012
228	G	S	L/8000	0.981	1.001	1.068	0.989	1.044
229	G	S	L/2000	0.981	1.000	1.067	0.988	1.043
230	G	S	L/1000	0.968	0.987	1.053	0.975	1.029
231	G	S	L/1000	0.972	0.992	1.058	0.980	1.034
232	G	S	L/1000	0.951	0.970	1.035	0.958	1.011
233	G	S	L/1000	0.955	0.974	1.039	0.962	1.015

Note: Δ_0 - Initial imperfection

Prv - Yield strength of reinforced column

P_{fea} - Finite elemetn analysis after reinforcing

P_{rl} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

S - Strong axis of the rolled section

Table C.1 (Cont'd)

FEA	•		Out-of-		SSRC 1	SSRC 2	CSA I	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0		(P_{fea}/P_{rl})	(P_{fea}/P_{r2})	(P_{fea}/P_{rel})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
247	F	W	L/1000	0.965	0.984	1.050	0.972	1.026
248	F	W	L/1000	0.972	0.992	1.058	0.979	1.034
249	F	W	L/1000	0.949	0.968	1.033	0.956	1.009
250	F	W	L/1000	0.954	0.973	1.038	0.961	1.014
292	G	S	L/1000	0.977	0.997	1.063	0.984	1.039
293	G	S	L/1000	0.979	0.999	1.066	0.986	1.041
294	G	S	L/1000	0.970	0.990	1.056	0.978	1.032
295	G	S	L/1000	0.970	0.990	1.056	0.977	1.031

Note: Δ_0 - Initial imperfection

Prv - Yield strength of reinforced column

Pfea - Finite elemetn analysis after reinforcing

P_{r1} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{re1} - Capacity after reinforcing (CSA1)

P_{re2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

S - Strong axis of the rolled section

Table C.2 Simulated Professional Factors for Columns from Group 2 ($\lambda = 1.1$)

FEA model No.	D	В	Out-of-Straightness δ_0	P _{fea} /P _{ry}	$\begin{array}{c} \text{SSRC 1} \\ \rho_s \\ (P_{\text{fea}}/P_{\text{r1}}) \end{array}$	$\begin{array}{c} SSRC\ 2 \\ \rho_s \\ (P_{fea}/P_{r2}) \end{array}$	$\begin{array}{c} \text{CSA 1} \\ \rho_s \\ (P_{\text{fea}}/P_{\text{rc1}}) \end{array}$	$\begin{array}{c} \text{CSA 2} \\ \rho_{s} \\ (P_{\text{fea}}/P_{\text{rc2}}) \end{array}$
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
23	F	W	L/8000	0.641	0.942	1.189	0.971	1.190
24	F	W	L/2000	0.601	0.883	1.114	0.909	1.115
25	F	W	L/1000	0.556	0.818	1.032	0.842	1.033
26	F	W	L/1000	0.560	0.823	1.038	0.848	1.039
27	F	W	L/1000	0.572	0.841	1.062	0.867	1.062
28	F	W	L/1000	0.511	0.751	0.947	0.773	0.948
29	F	W	L/1000	0.538	0.791	0.998	0.814	0.999
70	G	S	L/8000	0.603	0.887	1.119	0.914	1.120

Note: Δ_0 - Initial imperfection

P_{rv} - Yield strength of reinforced column

P_{fea} - Finite elemetn analysis after reinforcing

P_{rl} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{re1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

S - Strong axis of the rolled section

Table C.2 (Cont'd)

FEA			Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0	_	(P_{fea}/P_{r1})	(P_{fea}/P_{r2})	(P_{fea}/P_{rel})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
71	G	S	L/2000	0.575	0.845	1.067	0.871	1.067
72	G	S	L/1100	0.541	0.796	1.004	0.819	1.005
73	G	S	L/1100	0.541	0.795	1.003	0.818	1.004
74	G	S	L/1100	0.573	0.842	1.062	0.867	1.063
75	G	S	L/1150	0.478	0.702	0.886	0.723	0.887
76	G	S	L/1150	0.506	0.743	0.938	0.765	0.938
81	G	S	L/1100	0.544	0.799	1.009	0.823	1.009
82	G	S	L/1100	0.537	0.789	0.996	0.813	0.996
83	G	S	L/1100	0.540	0.794	1.002	0.818	1.003
88	F	W	L/8000	0.643	0.946	1.193	0.974	1.194
89	F	W	L/2000	0.603	0.886	1.118	0.912	1.119
90	F	W	L/1000	0.564	0.829	1.045	0.853	1.046
91	F	W	L/1000	0.569	0.837	1.056	0.862	1.057
92	F	W	L/1000	0.577	0.849	1.071	0.874	1.072
93	F	W	L/1000	0.528	0.776	0.979	0.799	0.980
94	F	W	L/1000	0.544	0.800	1.009	0.823	1.010
119	F	W	L/8000	0.652	0.959	1.210	0.988	1.211
120	F	W	L/2000	0.607	0.893	1.126	0.919	1.127
121	F	W	L/1000	0.564	0.830	1.047	0.855	1.048
122	F	W	L/1000	0.571	0.839	1.058	0.864	1.059
123	F	W	L/1000	0.579	0.852	1.074	0.877	1.075
124	F	W	L/1000	0.538	0.791	0.998	0.815	0.999
125	F	W	L/1000	0.554	0.815	1.028	0.839	1.029
164	G	S	L/8000	0.641	0.942	1.189	0.970	1.190
165	G	S	L/2000	0.598	0.879	1.108	0.905	1.109
166	G	S	L/1150	0.564	0.829	1.045	0.853	1.046
167	G	S	L/1150	0.574	0.845	1.066	0.870	1.066
168	G	S	L/1150	0.589	0.865	1.092	0.891	1.093
169	G	S	L/2750	0.507	0.745	0.941	0.768	0.941
170	G	S	L/1200	0.534	0.785	0.990	0.808	0.991
179	F	W	L/8000	0.681	1.001	1.263	1.031	1.264
180	F	W	L/2000	0.642	0.944	1.191	0.972	1.191

Note: Δ_0 - Initial imperfection

Pry - Yield strength of reinforced column

P_{fea} - Finite elemetn analysis after reinforcing

P_{rl} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

S - Strong axis of the rolled section

Table 5.6 (Cont'd)

FEA			Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0	-	(P_{fea}/P_{r1})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	_ (9)
181	F	W	L/1000	0.605	0.889	1.122	0.916	1.123
182	F	W	L/1000	0.608	0.894	1.128	0.921	1.129
183	F	W	L/1000	0.616	0.906	1.143	0.933	1.143
184	F	W	L/1000	0.572	0.841	1.062	0.867	1.062
185	F	W	L/1000	0.585	0.860	1.085	0.885	1.085
234	G	S	L/8000	0.648	0.953	1.202	0.981	1.203
235	G	S	L/2000	0.604	0.887	1.120	0.914	1.120
236	G	S	L/1000	0.564	0.829	1.046	0.854	1.047
237	G	S	L/1000	0.581	0.854	1.077	0.879	1.078
238	G	S	L/1000	0.590	0.868	1.095	0.894	1.096
239	G	S	L/1000	0.511	0.751	0.947	0.773	0.948
240	G	S	L/1000	0.535	0.787	0.992	0.810	0.993
251	F	W	L/8000	0.677	0.996	1.257	1.026	1.258
252	F	W	L/2000	0.637	0.937	1.182	0.965	1.183
253	F	W	L/1000	0.594	0.873	1.102	0.899	1.103
254	F	W	L/1000	0.593	0.873	1.101	0.899	1.102
255	F	W	L/1000	0.608	0.893	1.127	0.920	1.128
256	F	W	L/1000	0.547	0.804	1.015	0.828	1.015
257	F	W	L/1000	0.566	0.832	1.049	0.857	1.050
296	G	S	L/8000	0.624	0.918	1.158	0.945	1.159
297	G	S	L/2000	0.585	0.860	1.085	0.885	1.086
298	G	S	L/1000	0.551	0.811	1.023	0.835	1.024
299	G	S	L/1000	0.566	0.832	1.049	0.856	1.050
300	G	S	L/1000	0.578	0.849	1.072	0.875	1.072
301	G	S	L/1000	0.494	0.726	0.916	0.747	0.916
302	G	S	L/1000	0.520	0.764	0.964	0.787	0.964

Note: Δ_0 - Initial imperfection

P_{rv} - Yield strength of reinforced column

P_{fea} - Finite elemetn analysis after reinforcing

P_{r1} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

S - Strong axis of the rolled section

Table C.3 Simulated Professional Factors for Columns from Group 2 ($\lambda = 1.5$)

FEA			Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0		(P_{fea}/P_{ri})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
30	F	W	L/8000	0.415	1.020	1.181	0.999	1.161
31	F	W	L/2000	0.379	0.930	1.076	0.911	1.058
32	F	W	L/1000	0.355	0.871	1.008	0.853	0.991
33	F	W	L/1000	0.362	0.889	1.029	0.870	1.011
34	F	W	L/1000	0.338	0.830	0.961	0.813	0.945
35	F	W	L/1000	0.351	0.862	0.997	0.844	0.980
77	G	S	L/1350	0.374	0.919	1.064	0.901	1.046
78	G	S	L/1350	0.383	0.942	1.090	0.922	1.071
79	G	S	L/1350	0.349	0.857	0.992	0.839	0.975
80	G	S	L/1350	0.361	0.887	1.027	0.869	1.010
95	F	W	L/1000	0.357	0.878	1.016	0.860	0.998
96	F	W	L/1000	0.360	0.885	1.024	0.866	1.006
97	F	W	L/1000	0.365	0.897	1.038	0.879	1.020
98	F	W	L/1050	0.344	0.846	0.980	0.829	0.963
99	F	W	L/1050	0.351	0.862	0.997	0.844	0.980
126	F	W	L/1400	0.356	0.875	1.013	0.857	0.995
127	F	W	L/1050	0.368	0.903	1.045	0.884	1.027
128	F	W	L/1100	0.349	0.856	0.991	0.839	0.974
129	F	W	L/1100	0.361	0.886	1.026	0.868	1.008
171	G	S	L/1350	0.384	0.943	1.091	0.924	1.073
172	G	S	L/1350	0.389	0.956	1.106	0.936	1.087
173	G	S	L/1300	0.363	0.891	1.031	0.873	1.014
174	G	S	L/1300	0.369	0.908	1.050	0.889	1.032
186	F	W	L/1000	0.374	0.918	1.062	0.899	1.044
187	F	W	L/1000	0.379	0.931	1.077	0.911	1.059
188	F	W	L/1000	0.362	0.890	1.030	0.872	1.012
189	F	W	L/1000	0.371	0.912	1.055	0.893	1.037
241	G	S	L/8000	0.452	1.111	1.286	1.088	1.264
242	G	S	L/2000	0.417	1.024	1.185	1.003	1.164
243	G	S	L/1000	0.391	0.961	1.112	0.941	1.093
244	G	S	L/1000	0.400	0.982	1.136	0.962	1.117
245	G	S	L/1000	0.369	0.907	1.050	0.889	1.032

Note: Δ_0 - Initial imperfection

Prv - Yield strength of reinforced column

P_{fea} - Finite elemetn analysis after reinforcing

P_{rl} - Capacity after reinforcing (SSRC1)

P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{re2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

S - Strong axis of the rolled section

Table C.3 (Cont'd)

FEA	-		Out-of-		SSRC 1	SSRC 2	CSA 1	CSA 2
model	D	В	Straightness	P_{fea}/P_{ry}	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$	$ ho_{s}$
No.			δ_0		(P_{fea}/P_{rl})	(P_{fea}/P_{r2})	(P_{fea}/P_{rc1})	(P_{fea}/P_{rc2})
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)
246	G	S	L/1000	0.379	0.930	1.077	0.911	1.058
258	F	\mathbf{W}	L/1000	0.368	0.904	1.046	0.886	1.029
259	F	W	L/1000	0.377	0.925	1.071	0.906	1.052
260	F	W	L/1000	0.353	0.868	1.004	0.850	0.987
261	F	W	L/1000	0.361	0.888	1.027	0.870	1.010
303	G	S	L/1000	0.381	0.937	1.085	0.918	1.066
304	G	S	L/1000	0.393	0.966	1.118	0.946	1.099
305	G	S	L/1000	0.352	0.864	1.000	0.846	0.983
306	G	S	L/1000	0.365	0.897	1.038	0.879	1.020

Note: Δ_0 - Initial imperfection

P_{rv} - Yield strength of reinforced column

P_{fea} - Finite elemetn analysis after reinforcing

P_{r1} - Capacity after reinforcing (SSRC1)

 P_{r2} - Capacity after reinforcing (SSRC2)

P_{rc1} - Capacity after reinforcing (CSA1)

P_{rc2} - Capacity after reinforcing (CSA2)

L - Column length

D - Orientation of reinforcing plates

F - Parallel to the flanges

G - Parallel to the web

B - Buckling axis

W - Weak axis of the rolled section

S - Strong axis of the rolled section

Table C.4 Normalized Professional Factors for Columns from Group 2 ($\lambda = 0.4$)

FEA			Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_s = m \delta_0/L + b$	$ ho_{\text{seq}}$	ρ_{n}
No.			δ_0	(P_{fea}/P_{r2})	SSRC 2	•	$\rho_{\rm s}/\rho_{\rm seq}$
_(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
66	G	S	L/1000	1.02	$\rho_s = -49.041 \times \delta_0 / L + 1.09$	1.041	0.977
67	G	S	L/1000	1.02	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	0.983
68	G	S	L/1000	0.99	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	0.953
69	G	S	L/1000	1.00	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	0.959
17	F	W	L/8000	1.09	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.084	1.009
18	F	W	L/2000	1.07	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.065	1.008
19	F	W	L/1000	1.06	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	1.018
20	F	W	L/1000	1.06	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	1.022
21	F	W	L/1000	1.04	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	1.002
22	F	W	L/1000	1.05	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	1.005
84	F	W	L/1000	1.06	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	1.021
85	F	W	L/1000	1.07	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	1.025
86	F	W	L/1000	1.04	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	0.995
87	F	W	L/1000	1.04	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	0.999
115	F	W	L/1000	1.07	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	1.025
116	F	W	L/1000	1.07	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	1.028
117	F	W	L/1000	1.05	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	1.008
118	F	W	L/1000	1.05	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	1.010
160	G	S	L/1000	1.01	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	0.971
161	G	S	L/1000	1.02	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	0.979
162	G	S	L/1000	0.97	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	0.934
163	G	S	L/1000	0.98	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	0.943
175	F	W	L/1000	1.06	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	1.016
176	F	W	L/1000	1.07	$\rho_{\rm s} = -49.041 {\rm k} \delta_0 / {\rm L} + 1.09$	1.041	1.023
177	F	W	L/1000	1.03	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	0.990
178	F	W	L/1000	1.04	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	0.995
228	G	S	L/8000	1.07	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.084	0.986
229	G	S	L/2000	1.07	$\rho_{\rm s} = -49.041 {\rm x} \delta_0 / {\rm L} + 1.09$	1.065	1.002
230	G	S	L/1000	1.05	$\rho_s = -49.041x\delta_0/L + 1.09$	1.041	1.012
231	G	S	L/1000	1.06	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	1.017
232	G	S	L/1000	1.03	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	0.994
233	G	S	L/1000	1.04	$\rho_{\rm s} = -49.041 \text{x} \delta_0 / \text{L} + 1.09$	1.041	0.998
247	F	W	L/1000	1.05	$\rho_s = -49.041 \times \delta_0 / L + 1.09$	1.041	1.009

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section

G - Parallel to the web P_{r2} - Capacity after reinforcing (SSRC2) P_{fea} - Finite elemeth analysis after reinforcing

Table C.4 (Cont'd)

FEA model	D	В	Out-of- Straightness	SSRC 2 _{Ps}	$\rho_{\rm s}={\rm m}\;\delta_{\rm 0}/{\rm L}+{\rm b}$	ρ_{seq}	SSRC 2
No.			δ_0	$(P_{\text{fea}}/P_{\text{r2}})$	SSRC 2	·	$ ho_{s}/ ho_{seq}$
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
248	F	W	L/1000	1.06	$\rho_s = -49.041 \times \delta_0 / L + 1.09$	1.041	1.017
249	F	\mathbf{W}	L/1000	1.03	$\rho_s = -49.041 \times \delta_0 / L + 1.09$	1.041	0.992
250	F	W	L/1000	1.04	$\rho_s = -49.041 \times \delta_0 / L + 1.09$	1.041	0.998
292	G	S	L/1000	1.06	$\rho_s = -49.041 \times \delta_0 / L + 1.09$	1.041	1.021
293	G	S	L/1000	1.07	$\rho_s = -49.041 \times \delta_0 / L + 1.09$	1.041	1.024
294	G	S	L/1000	1.06	$\rho_s = -49.041 \times \delta_0 / L + 1.09$	1.041	1.015
295	G	S	L/1000	1.06	$\rho_s = -49.041 \times \delta_0 / L + 1.09$	1.041	1.014

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section S - Strong axis of the rolled section

F - Parallel to the flanges G - Parallel to the web

P_{r2} - Capacity after reinforcing (SSRC2)

P_{fea} - Finite elemetn analysis after reinforcing

 ρ_{seq} - Professional ratio predicted by the equation

Table C.5 Normalized Professional Factors for Columns from Group 2 ($\lambda = 1.1$)

FEA			Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_{\rm s}={\rm m}\;\delta_{\rm 0}/{\rm L}+{\rm b}$	$ ho_{seq}$	ρ_{n}
No.			δ_0	(P_{fea}/P_{r2})	SSRC 2		ρ_s/ρ_{seq}
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
70	G	S	L/8000	1.12	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.18	0.95
71	G	S	L/2000	1.07	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.12	0.96
72	G	S	L/1100	1.00	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.05	0.96
73	G	S	L/1100	1.00	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.05	0.96
74	G	S	L/1100	1.06	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.05	1.01
75	G	S	L/1150	0.89	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.05	0.84
76	G	S	L/1150	0.94	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.05	0.89
81	G	S	L/1100	1.01	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.05	0.96
82	G	S	L/1100	1.00	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.05	0.95
83	G	S	L/1100	1.00	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.05	0.96
23	F	W	L/8000	1.19	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.18	1.01
24	F	W	L/2000	1.11	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.12	0.99

Note: L - Column length

B - Buckling axis

D - Orientation of reinforcing plate W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section

G - Parallel to the web

P_{r2} - Capacity after reinforcing (SSRC2)

P_{fea} - Finite elemetn analysis after reinforcing

Table C.5 (Cont'd)

model D B Straighmess δ0 ρ_s ρ_s = m $\delta_0/L + b$ $\rho_s eq$ ρ_n (1) (2) (3) (4) (5) (6) (7) (8) 25 F W L/1000 1.03 ρ_s = -168.09x δ_0 /L + 1.1992 1.03 1.00 26 F W L/1000 1.06 ρ_s = -168.09x δ_0 /L + 1.1992 1.03 1.01 27 F W L/1000 1.06 ρ_s = -168.09x δ_0 /L + 1.1992 1.03 1.03 28 F W L/1000 1.90 ρ_s = -168.09x δ_0 /L + 1.1992 1.03 0.92 29 F W L/1000 1.12 ρ_s = -168.09x δ_0 /L + 1.1992 1.13 0.97 88 F W L/2000 1.12 ρ_s = -168.09x δ_0 /L + 1.1992 1.13 1.00 90 F W L/1000 1.06 ρ_s = -168.09x δ_0 /L + 1.1992 1.03 1.01 91 F W L/1000 1.07	FEA			Out-of-	SSRC 2		····	SSRC 2
No. δ ₀ (P_{res}/P_{c2}) SSRC 2 ρ/ρ_{seq} (1) (2) (3) (4) (5) (6) (7) (8) (25 F W L/1000 1.03 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 26 F W L/1000 1.04 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.01 27 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03 28 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.92 29 F W L/1000 1.00 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.97 88 F W L/8000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.01 89 F W L/2000 1.12 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.01 90 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 90 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.01 91 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 92 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 93 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 93 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 91 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 91 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.95 92 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.95 119 F W L/2000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.95 119 F W L/2000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.95 119 F W L/2000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.95 110 F W L/2000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 112 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 112 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 112 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 112 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 112 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 112 F W L/1000 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 112 F W L/1000 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 112 F W L/1000 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 112 F W L/1000 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.01 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.19 1.10 1.00 1.01 1.01 1.01 1.01 1.01	model	D	В	Straightness	$ ho_{s}$	$\rho_s = m \delta_0 / L + b$	$ ho_{seq}$	$ ho_{\mathfrak{n}}$
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	No.			δ_0	(P_{fea}/P_{r2})	SSRC 2		$ ho_{s}/ ho_{seq}$
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	(1)	(2)	(3)	(4)	(5)		(7)	(8)
27 F W L/1000 1.06 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.03 2.8 F W L/1000 0.95 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 0.92 2.9 F W L/1000 1.00 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 0.97 8.8 F W L/8000 1.19 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.18 1.01 8.9 F W L/2000 1.12 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.18 1.01 9.0 F W L/1000 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.01 9.1 F W L/1000 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.02 9.2 F W L/1000 1.07 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.04 9.3 F W L/1000 1.07 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.04 9.3 F W L/1000 1.07 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 0.95 9.4 F W L/1000 1.01 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 0.95 1.19 F W L/2000 1.01 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 0.98 1.19 F W L/2000 1.21 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.18 1.03 1.22 F W L/1000 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.18 1.03 1.22 F W L/1000 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.02 1.22 F W L/1000 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.02 1.22 F W L/1000 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.04 1.24 F W L/1000 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.04 1.24 F W L/1000 1.07 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.04 1.24 F W L/1000 1.07 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.04 1.26 F W L/1000 1.07 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.04 1.27 F W L/1000 1.09 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.04 1.04 F W L/1000 1.07 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.05 1.04 1.06 G S L/2000 1.11 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.00 1.64 G S L/2000 1.19 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.12 0.99 1.66 G S L/1150 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.15 1.01 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.16 0.93 1.00 1.07 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.17 1.18 1.01 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.19 1.10 1.01 1.02 1.01 1.01 1.01 1.01 1.01	25	F	W	L/1000	1.03	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.00
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	26	F	W	L/1000	1.04	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.01
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	27	F	W	L/1000	1.06	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.03
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	28	F	W	L/1000	0.95	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.03	0.92
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	29	F	W	L/1000	1.00	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	0.97
90 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.01 91 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 92 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 93 F W L/1000 0.98 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.95 94 F W L/1000 1.01 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.98 119 F W L/8000 1.21 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.03 120 F W L/2000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.12 1.01 121 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 122 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03 123 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.00 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 164 G S L/8000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 165 G S L/2000 1.11 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.12 0.99 166 G S L/1150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 0.99 167 G S L/150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/1150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/1150 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/2000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/150 0.99 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 169 G S L/2750 0.94 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.15 1.07 1.08 F W L/2000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.93 1.79 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.07 1.07 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.07 1.07 F W L/8000 1.29 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.10 1.09 1.11 0.11 0.11 0.11 0.11 0.11	88	F	W	L/8000	1.19	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.18	1.01
91 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 92 F W L/1000 0.98 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.95 94 F W L/1000 1.01 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.98 119 F W L/8000 1.21 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.03 120 F W L/2000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.12 1.01 121 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 122 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03 123 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 164 G S L/8000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 165 G S L/2000 1.11 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 166 G S L/150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 0.99 167 G S L/150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/2750 0.94 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.14 0.83 170 G S L/2000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 169 G S L/2750 0.94 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.15 1.07 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.93 1.79 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.93 1.79 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.93 1.79 F W L/8000 1.29 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.07 1.80 F W L/2000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 1.82 F W L/1000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 1.82 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 1.83 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09	89	F	W	L/2000	1.12	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.12	1.00
92 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 93 F W L/1000 0.98 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.95 94 F W L/1000 1.01 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.98 119 F W L/8000 1.21 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.03 120 F W L/2000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.12 1.01 121 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 122 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03 123 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.00 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.03 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 164 G S L/8000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 165 G S L/2000 1.11 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.01 165 G S L/150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.15 0.99 167 G S L/150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 0.99 167 G S L/150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/150 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/2750 0.94 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 169 G S L/2750 0.94 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.83 170 G S L/1200 0.99 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.83 170 G S L/1200 0.99 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.83 170 G S L/1000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.17 0.06 0.93 179 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.07 181 F W L/1000 1.12 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 182 F W L/1000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 182 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.01 1.01 184 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03	90	F	W	L/1000	1.05		1.03	1.01
93 F W L/1000 0.98 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.95 94 F W L/1000 1.01 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.98 119 F W L/8000 1.21 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.03 120 F W L/2000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.12 1.01 121 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 122 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03 123 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.00 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.03 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 164 G S L/8000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 165 G S L/2000 1.11 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.01 165 G S L/1150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 0.99 167 G S L/1150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 0.99 167 G S L/1200 0.91 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/2750 0.94 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 169 G S L/2750 0.94 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.83 170 G S L/1200 0.99 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.83 170 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 1.89 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.07 181 F W L/2000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.06 0.93 179 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 182 F W L/1000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 182 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 183 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.01 1.01	91	F	W	L/1000	1.06	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.02
94 F W L/1000 1.01 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 0.98 119 F W L/8000 1.21 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.03 120 F W L/2000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.12 1.01 121 F W L/1000 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.02 122 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03 123 F W L/1000 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.00 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.04 124 F W L/1000 1.00 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 164 G S L/8000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.00 165 G S L/2000 1.11 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.12 0.99 166 G S L/1150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 0.99 167 G S L/150 1.05 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/2750 0.94 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.14 0.83 1.70 G S L/1200 0.99 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.16 0.93 1.79 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.07 1.80 F W L/2000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.12 1.07 1.81 F W L/1000 1.12 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 1.82 F W L/1000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 1.83 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.01 1.01 1.04 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.01 1.01 1.01 1.01 1.01 1.01 1.01	92	F	W	L/1000	1.07	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.04
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	93		W	L/1000	0.98	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	0.95
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	94	F	W	L/1000	1.01	• •	1.03	0.98
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	119	F	W	L/8000	1.21	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.18	1.03
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	120	F	W	L/2000	1.13	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.12	1.01
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	121	F	W	L/1000	1.05	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.02
124 F W L/1000 1.00 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 0.97 125 F W L/1000 1.03 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.00 164 G S L/8000 1.19 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.18 1.01 165 G S L/2000 1.11 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.12 0.99 166 G S L/1150 1.05 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.05 0.99 167 G S L/1150 1.07 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.05 1.01 168 G S L/1150 1.09 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.05 1.01 169 G S L/2750 0.94 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.05 1.04 169 G S L/2750 0.94 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.14 0.83 170 G S L/1200 0.99 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.06 0.93 179 F W L/8000 1.26 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.18 1.07 180 F W L/2000 1.19 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.12 1.07 181 F W L/1000 1.12 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.09 182 F W L/1000 1.13 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.09 183 F W L/1000 1.14 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.01 1.11 184 F W L/1000 1.06 $ρ_s = -168.09xδ_0/L + 1.1992$ 1.03 1.01	122	F	W	L/1000	1.06	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.03
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	123	F	W	L/1000	1.07	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.04
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	124	_	W	L/1000	1.00	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.03	0.97
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	125	F	W	L/1000	1.03	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.00
166 G S L/1150 1.05 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.05 0.99 167 G S L/1150 1.07 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.05 1.01 168 G S L/1150 1.09 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.05 1.04 169 G S L/2750 0.94 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.14 0.83 170 G S L/1200 0.99 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.06 0.93 179 F W L/8000 1.26 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.18 1.07 180 F W L/2000 1.19 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.12 1.07 181 F W L/1000 1.12 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.09 182 F W L/1000 1.13 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.09 183 F W L/1000 1.14 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.01 184 F W L/1000 1.06 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.01	164	G	S	L/8000	1.19	$\rho_{\rm s} = -168.09 \times \delta_0 / L + 1.1992$	1.18	1.01
167 G S L/1150 1.07 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.01 168 G S L/1150 1.09 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.05 1.04 169 G S L/2750 0.94 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.14 0.83 170 G S L/1200 0.99 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.06 0.93 179 F W L/8000 1.26 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.18 1.07 180 F W L/2000 1.19 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.12 1.07 181 F W L/1000 1.12 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 182 F W L/1000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 183 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.11 184 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03	165	G		L/2000	1.11	$\rho_{\rm s} = -168.09 \times \delta_0 / L + 1.1992$	1.12	0.99
168 G S L/1150 1.09 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.05 1.04 169 G S L/2750 0.94 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.14 0.83 170 G S L/1200 0.99 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.06 0.93 179 F W L/8000 1.26 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.18 1.07 180 F W L/2000 1.19 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.12 1.07 181 F W L/1000 1.12 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.03 1.09 182 F W L/1000 1.13 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.03 1.09 183 F W L/1000 1.14 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.03 1.11 184 F W L/1000 1.06 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.03 1.03	166	G		L/1150	1.05		1.05	0.99
169 G S L/2750 0.94 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.14 0.83 170 G S L/1200 0.99 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.06 0.93 179 F W L/8000 1.26 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.18 1.07 180 F W L/2000 1.19 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.12 1.07 181 F W L/1000 1.12 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.09 182 F W L/1000 1.13 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.09 183 F W L/1000 1.14 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.11 184 F W L/1000 1.06 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.03				L/1150	1.07		1.05	1.01
170 G S L/1200 0.99 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.06 0.93 179 F W L/8000 1.26 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.18 1.07 180 F W L/2000 1.19 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.12 1.07 181 F W L/1000 1.12 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.09 182 F W L/1000 1.13 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.09 183 F W L/1000 1.14 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.11 184 F W L/1000 1.06 $\rho_s = -168.09 \times \delta_0/L + 1.1992$ 1.03 1.03	168	G		L/1150	1.09	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.05	1.04
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	169	_		L/2750	0.94	$\rho_{\rm s} = -168.09 \text{ x} \delta_0 / \text{L} + 1.1992$	1.14	0.83
$\begin{array}{cccccccccccccccccccccccccccccccccccc$			_		0.99	$\rho_{\rm s} = -168.09 \times \delta_0 / L + 1.1992$	1.06	0.93
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	179	F	W	L/8000	1.26	$\rho_{\rm s} = -168.09 \times \delta_0 / L + 1.1992$	1.18	1.07
182 F W L/1000 1.13 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.09 183 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.11 184 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03	180	F	W	L/2000	1.19	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.12	1.07
183 F W L/1000 1.14 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.11 184 F W L/1000 1.06 $\rho_s = -168.09x\delta_0/L + 1.1992$ 1.03 1.03	181	F	W	L/1000	1.12	• •	1.03	1.09
184 F W L/1000 1.06 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.03 1.03	182	F	W	L/1000	1.13	• •	1.03	1.09
· •	183	F	W	L/1000	1.14	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.03	1.11
185 F W L/1000 1.08 $\rho_s = -168.09 \times \delta_0 / L + 1.1992$ 1.03 1.05	184	F	W	L/1000	1.06	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.03
Note: L. Column length R. Ruckling axis					1.08	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.05

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section

G - Parallel to the web P_{r2} - Capacity after reinforcing (SSRC2)

P_{fea} - Finite elemetn analysis after reinforcing

Table C.5 (Cont'd)

FEA			Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_s = m \delta_0/L + b$	$ ho_{seq}$	$\rho_{\mathtt{n}}$
No.			δ_0	$(P_{\text{fea}}/P_{\text{r2}})$	SSRC 2		$\rho_{\rm s}/\rho_{\rm seq}$
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
234	G	S	L/8000	1.20	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.18	1.02
235	G	S	L/2000	1.12	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.12	1.00
236	G	S	L/1000	1.05	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.03	1.01
237	G	S	L/1000	1.08	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.04
238	G	S	L/1000	1.09	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.03	1.06
239	G	S	L/1000	0.95	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	0.92
240	G	S	L/1000	0.99	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.03	0.96
251	F	S	L/8000	1.26	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.18	1.07
252	F	S	L/2000	1.18	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.12	1.06
253	F	S	L/1000	1.10	$\rho_{\rm s} = -168.09 \text{m} \delta_0 / \text{L} + 1.1992$	1.03	1.07
254	F	S	L/1000	1.10	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.07
255	F	S	L/1000	1.13	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	1.09
256	F	S	L/1000	1.01	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.03	0.98
257	F	S	L/1000	1.05	$\rho_{\rm s} = -168.09 \times \delta_0 / L + 1.1992$	1.03	1.02
296	G	S	L/8000	1.16	$\rho_{\rm s} = -168.09 \times \delta_0 / L + 1.1992$	1.18	0.98
297	G	S	L/2000	1.08	$\rho_{\rm s} = -168.09 \text{x} \delta_0 / \text{L} + 1.1992$	1.12	0.97
298	G	S	L/1000	1.02	$\rho_{\rm s} = -168.09 \times \delta_0 / L + 1.1992$	1.03	0.99
299	G	S	L/1000	1.05	$\rho_{\rm s} = -168.09 \times \delta_0 / L + 1.1992$	1.03	1.02
300	G	S	L/1000	1.07	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.03	1.04
301	G	S	L/1000	0.92	$\rho_{\rm s} = -168.09 {\rm x} \delta_0 / {\rm L} + 1.1992$	1.03	0.89
302	G	S	L/1000	0.96	$\rho_s = -168.09 \times \delta_0 / L + 1.1992$	1.03	0.93

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section

G - Parallel to the web

P_{r2} - Capacity after reinforcing (SSRC2)

P_{fea} - Finite elemetn analysis after reinforcing

Table C.6 Normalized Professional Factors for Columns from Group 2 ($\lambda = 1.5$)

FEA			Out-of-	SSRC 2			SSRC 2
model	D	В	Straightness	$ ho_{s}$	$\rho_{\rm s}={\rm m}\;\delta_0/{\rm L}+{\rm b}$	$ ho_{\text{seq}}$	$\rho_{\mathtt{n}}$
No.			δ_0	(P_{fea}/P_{r2})	SSRC 2	•	$ ho_{s}/ ho_{seq}$
_(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
77	G	S	L/1350	1.06	$\rho_s = -190.38 \times \delta_0 / L + 1.2219$	1.08	0.98
78	G	S	L/1350	1.09	$\rho_s = -190.38x\delta_0/L + 1.2219$	1.08	1.01
79	G	S	L/1350	0.99	$\rho_s = -190.38 \times \delta_0 / L + 1.2219$	1.08	0.92
80	G	S	L/1350	1.03	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.08	0.95
30	F	W	L/8000	1.18	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.20	0.99
31	F	W	L/2000	1.08	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.13	0.96
32	F	W	L/1000	1.01	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.03	0.98
33	F	W	L/1000	1.03	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.03	1.00
34	F	W	L/1000	0.96	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.04	0.93
35	F	W	L/1000	1.00	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.04	0.96
95	F	W	L/1000	1.02	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.03	0.98
96	F	W	L/1000	1.02	$\rho_{\rm s} = -190.38 \times \delta_0 / L + 1.2219$	1.03	0.99
97	F	W	L/1000	1.04	$\rho_{\rm s} = -190.38 {\rm x} \delta_0 / {\rm L} + 1.2219$	1.03	1.01
98	F	W	L/1050	0.98	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.04	0.94
99	F	W	L/1050	1.00	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.04	0.96
126	F	W	L/1400	1.01	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.04	0.97
127	F	W	L/1050	1.05	$\rho_{\rm s} = -190.38 {\rm x} \delta_0 / {\rm L} + 1.2219$	1.04	1.00
128	F	W	L/1100	0.99	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.05	0.94
129	F	W	L/1100	1.03	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.05	0.98
171	G	S	L/1350	1.09	$\rho_{\rm s} = -190.38 {\rm x} \delta_0 / {\rm L} + 1.2219$	1.08	1.01
172	G	S	L/1350	1.11	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.08	1.02
173	G	S	L/1300	1.03	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.08	0.96
174	G	S	L/1300	1.05	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.08	0.97
186	F	W	L/1000	1.06	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.03	1.03
187	F	W	L/1000	1.08	$\rho_{\rm s} = -190.38 \times \delta_0 / L + 1.2219$	1.03	1.04
188	F	W	L/1000	1.03	$\rho_{\rm s} = -190.38 \times \delta_0 / L + 1.2219$	1.03	1.00
189	F	W	L/1000	1.06	$\rho_{\rm s} = -190.38 \times \delta_0 / L + 1.2219$	1.03	1.02
241	G	S	L/8000	1.29	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.20	1.07
242	G	S	L/2000	1.18	$\rho_s = -190.38 \times \delta_0 / L + 1.2219$	1.13	1.05
243	G	S	L/1000	1.11	$\rho_{\rm s} = -190.38 \times \delta_0 / L + 1.2219$	1.03	1.08
244	G	S	L/1000	1.14	$\rho_{\rm s} = -190.38 \times \delta_0 / L + 1.2219$	1.03	1.10
245	G	S	L/1000	1.05	$\rho_s = -190.38 \times \delta_0 / L + 1.2219$	1.03	1.02
_246	<u>G</u>	<u> </u>	L/1000	1.08	$\rho_s = -190.38 \times \delta_0 / L + 1.2219$	1.03	1.04

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section

G - Parallel to the web P_{r2} - Capacity after reinforcing (SSRC2) P_{fea} - Finite elemeth analysis after reinforcing

Table C.6 (Cont'd)

FEA model No.	D	В	Out-of-Straightness δ_0	$\begin{array}{c} \text{SSRC 2} \\ \rho_{s} \\ (P_{\text{fea}}/P_{r2}) \end{array}$	$\rho_s = m \delta_0 / L + b$ SSRC 2	$ ho_{seq}$	SSRC 2 ρ_n ρ_s/ρ_{seq}
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
258	F	W	L/1000	1.05	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.03	1.01
259	F	W	L/1000	1.07	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.03	1.04
260	F	W	L/1000	1.00	$\rho_{\rm s} = -190.38 {\rm x} \delta_0 / {\rm L} + 1.2219$	1.03	0.97
261	F	W	L/1000	1.03	$\rho_{\rm s} = -190.38 {\rm x} \delta_0 / {\rm L} + 1.2219$	1.03	1.00
303	G	S	L/1000	1.08	$\rho_{\rm s} = -190.38 \times \delta_0 / L + 1.2219$	1.03	1.05
304	G	S	L/1000	1.12	$\rho_{\rm s} = -190.38 {\rm x} \delta_0 / {\rm L} + 1.2219$	1.03	1.08
305	G	S	L/1000	1.00	$\rho_{\rm s} = -190.38 \text{x} \delta_0 / \text{L} + 1.2219$	1.03	0.97
306	G	S	L/1000	1.04	$\rho_s = -190.38 \times \delta_0 / L + 1.2219$	1.03	1.01

B - Buckling axis

D - Orientation of reinforcing plate

W - Weak axis of the rolled section

F - Parallel to the flanges

S - Strong axis of the rolled section

G - Parallel to the web

P_{r2} - Capacity after reinforcing (SSRC2)

P_{fea} - Finite elemetn analysis after reinforcing

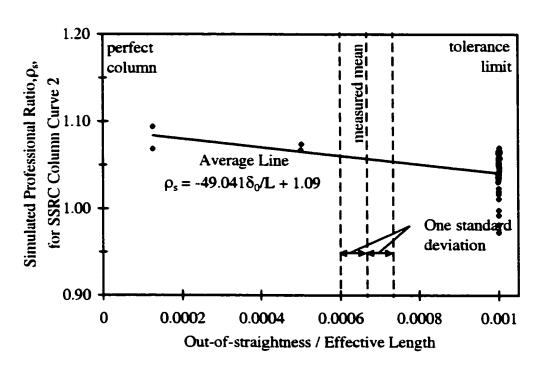


Figure C.1 Simulated Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 2 ($\lambda = 0.4$)

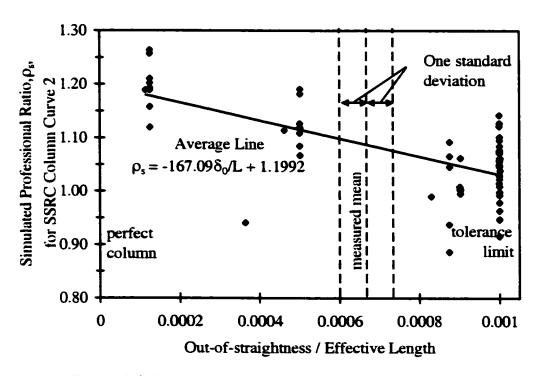


Figure C.2 Simulated Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 2 ($\lambda = 1.1$)

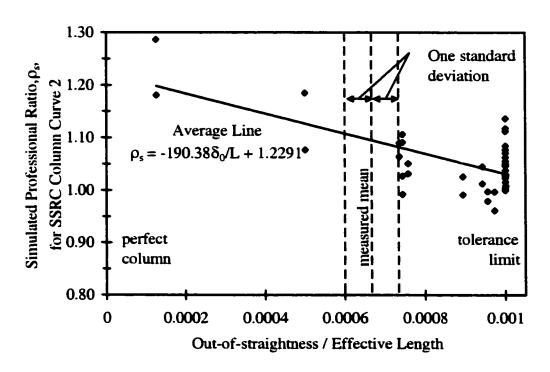


Figure C.3 Simulated Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 2 ($\lambda = 1.5$)

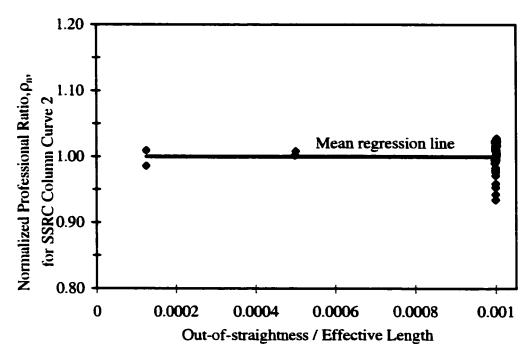


Figure C.4 Normalized Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 2 (λ =0.4)

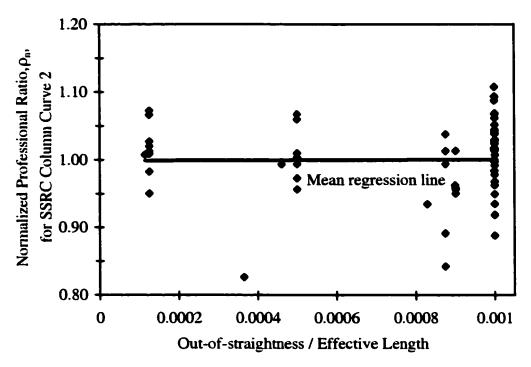


Figure C.5 Normalized Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 2 (λ = 1.1)

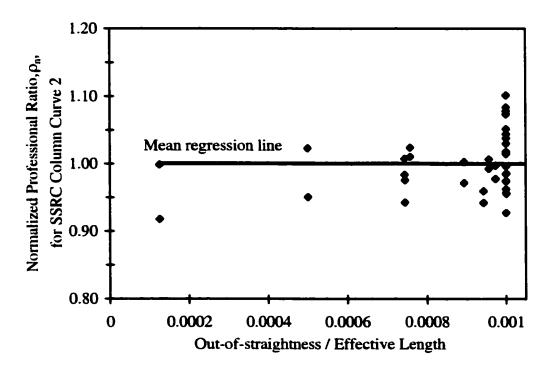


Figure C.6 Normalized Professional Ratio vs. Value of Out-ofstraightness for Columns from Group 2 ($\lambda = 1.5$)

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